

CS/B.Tech/ME/Even/Sem-6th/ME-605C/2015



**WEST BENGAL UNIVERSITY OF TECHNOLOGY**

**ME-605C**

**TURBO MACHINERY**

Time Allotted: 3 Hours

Full Marks: 70

*The questions are of equal value.*

*The figures in the margin indicate full marks.*

*Candidates are required to give their answers in their own words as far as practicable.*

*All symbols are of usual significance.*

**GROUP A**

**(Multiple Choice Type Questions)**

1. Answer any *ten* questions.

10×1 = 10

(i) At the stagnation point of a flow field, the value is zero for

(A) pressure

☒ (B) velocity

(C) temperature

(D) none of these

(ii) In all reaction turbines, for maximum efficiency

☒ (A) the velocity of swirl at entrance must be zero

(B) the velocity of flow at outlet must be zero

☒ (C) the velocity of swirl at outlet must be zero

(D) the velocity of flow at entrance must be zero

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(iii) A fast centrifugal pump impeller will have

(A) forward facing blades

☒ (B) backward facing blades

(C) radial blades

(D) propeller type blades

(iv) The mechanical efficiency takes into account the losses in

(A) pump impeller

(B) bearings and windages

(C) pipes

☒ (D) all of the above

(v) Draft tube is used for discharging water from the exit of

(A) Kaplan Turbine

☒ (B) Francis Turbine

(C) Pelton Turbine

(D) both (A) and (B)

(vi) The ratio of total useful heat drop to the total isentropic heat drop, is called

☒ (A) stage efficiency

(B) internal efficiency

(C) rankine efficiency

(D) none of these

(vii) Axial flow compressors have

(A) purely impulse stage

☒ (B) impulse and reaction stages

(C) purely reaction stage

(D) alternate stages of impulse and reaction

(viii) Stalling of the blade means

(A) loss of drag

☒ (B) separation of flow

(C) loss of lift

(D) loss of performance

(ix) Fluid is flowing through a duct with a Mach number equal to 1.2. An increase in cross-sectional area in the downstream will cause an

☒ (A) decrease in velocity

(B) choked flow situation

(C) increase in static pressure

(D) increase in velocity

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(x) Pelton turbine is used for

- (A) low head high discharge    ☒ (B) high head low discharge  
(C) medium head medium discharge    (D) high head medium discharge

(xi) Muschel curve means

- (A) constant entropy curve    (B) constant head curve  
☒ (C) constant efficiency curve    (D) constant discharge curve

(xii) Generally the efficiency of the prototype is \_\_\_\_\_ the efficiency of model of a turbomachine.

- ☒ (A) less than    ☒ (B) greater than  
(C) equal    ☒ (D) none of these

### GROUP B (Short Answer Type Questions)

Answer any *three* questions.

3×5 = 15

2. Differentiate between isentropic and adiabatic process. 5
3. Explain terms flow co-efficient; head co-efficient and power co-efficient. 5
4. Explain the functional differences between nozzle and diffuser for compressible flow. 5
5. Air flows from a large tank (P = 650 kPa(abs), T = 550°C) through converging nozzle, with a throat area of 600 mm<sup>2</sup> and discharge to the atmosphere. Determine the rate of mass flow under isentropic condition in the nozzle. 5

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Turn Over

6. Explain the hydraulic functions of spiral casing, guide vanes and the draft tube. 5
7. Differentiate between impulse turbine and a reaction turbine. Why does efficiency of pump is less than that of a turbine? 3+2

### GROUP C (Long Answer Type Questions)

Answer any *three* questions.

3×15 = 45

8. A centrifugal pump lifts water against a static head of 40 m of water. Suction and delivery pipes are both 150 mm in diameter. Head losses in the suction and delivery pipes are 2.5 m and 7 m of water respectively. The impeller is 400 mm in diameter, 25 mm wide and rotating at 1200 r.p.m. The impeller blade angle at exit is 30°. Assume radial entry at the impeller inlet, if the manometric efficiency is 85% and overall efficiency is 75%, determine the discharge and power required to drive the pump. 15
9. (a) A hydraulic turbine is to develop 1015 kW when running at 120 rpm under a net head of 12 m. Work out the maximum flow rate specific speed for the turbine if the overall efficiency at the best operating point is 92%. In order to predict its performance, a 1:10 scale model is tested under a head of 7.2 m. What would be the speed, power output and water consumption of the model if it runs under the conditions similar to the prototype? 7
- (b) Show that the pressure head rise in the impeller of a centrifugal pump with no swirl at inlet, and frictional and other losses in the impeller are neglected, is given by 8

$$\frac{1}{2g} [C_{r1}^2 + U_1^2 - C_{r2}^2 \csc^2 \beta_2]$$

where  $C_{r1}$  and  $C_{r2}$  are velocity of flow at inlet.

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- 10.(a) A Francis turbine is to be designed to develop 370 kW of power under a head of 70 m while running at a speed of 750 rpm. Following are some pertinent data of the turbine: 10+5

- (i) Ratio of width of runner to outer diameter of runner = 0.1
- (ii) Ratio of inner diameter of runner to outer diameter of runner = 0.5
- (iii) Flow ratio = 0.15
- (iv) Hydraulic efficiency = 95%
- (v) Mechanical efficiency = 84%
- (vi) Circumferential area occupied by thickness of vanes = 5%

Assuming constant flow velocity calculate (a) guide vane angle (b) runner blade angle at inlet and (c) blade angle at outlet.

- (b) The following data refer to an elbow type draft tube:  
Area of circular inlet =  $25 \text{ m}^2$ . Area of rectangular outlet =  $11 \text{ m}^2$ . Velocity of water at inlet to draft tube =  $10 \text{ m/s}$   
The frictional head loss in draft tube equals to 10% of the inlet velocity head.  
Elevation of inlet plane above tail race level =  $0.6 \text{ m}$   
Determine:  
(i) Vacuum or negative head at inlet  
(ii) Power loss in tail race

- 11.(a) The mean bucket speed of a pelton turbine is  $14 \text{ m/s}$ . It is supplied with water at the rate of  $0.8 \text{ m}^3/\text{s}$  under a head of  $45 \text{ m}$ . If the water jet is deflected by the buckets, through an angle of  $165^\circ$ , find the power and efficiency of the turbine. Assume coefficient of velocity as  $0.985$ . 7+8

- (b) A compressor cascade has the following data:  
Velocity of air at entry =  $75 \text{ m/s}$ , air angle at entry =  $48^\circ$ , air angle at exit =  $25^\circ$ , pitch chord =  $1.1$ , stagnation pressure loss =  $11 \text{ mm W.G.}$ , density of air =  $1.25 \text{ kg/m}^3$ .  
Determine loss coefficient, drag and lift coefficients, ideal and actual pressure recovery coefficients, diffuser efficiency and maximum diffuser efficiency.

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- 12.(a) A pelton wheel is to be designed for the specifications. Power =  $735.75 \text{ kW}$ , s.p head =  $200 \text{ m}$ , speed =  $800 \text{ r.p.m.}$ ,  $\eta_0 = 0.86$  and the jet diameter is not to exceed one-tenth of the wheel diameter. Determine 7+6+2

- (i) Wheel diameter
- (ii) Number of jets required
- (iii) Diameter of the jet.

Take  $C_v = 0.98$  and speed ratio =  $0.45$ .

- (b) Find the mach number when an Aeroplane is flying at  $900 \text{ km/hour}$  through still air having a pressure of  $8.0 \text{ N/cm}^2$  and temperature  $-15^\circ\text{C}$ . take  $K = 287 \text{ J/kgK}$ . Calculate the pressure, Temperature and Density of air at the stagnation point on the nose of the plane.  
(c) What do you mean by dimensional homogeneity?