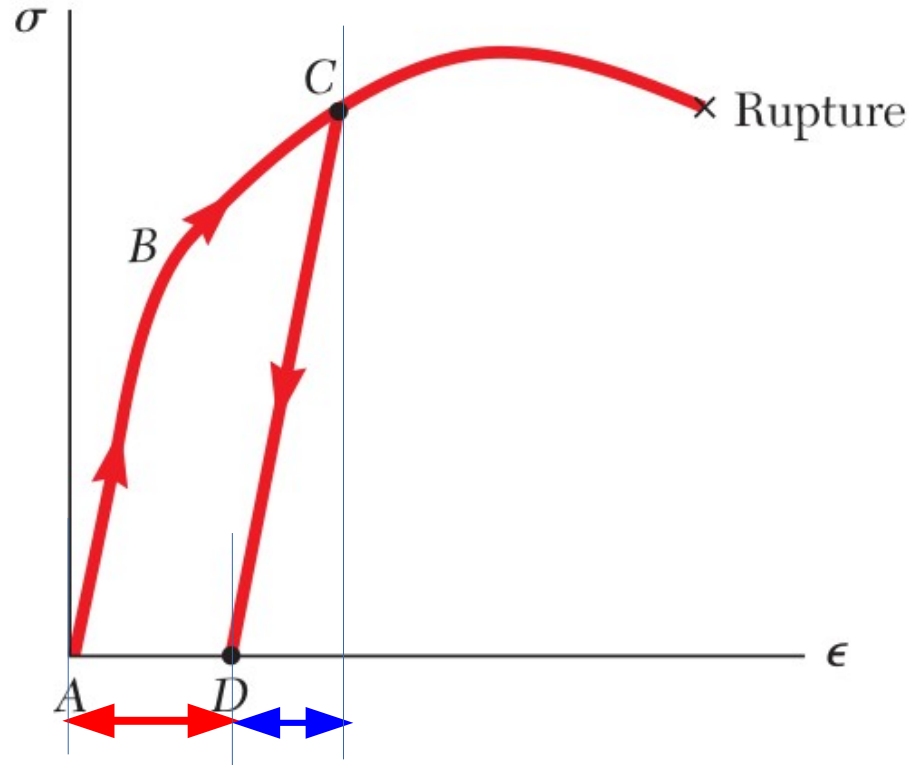


# **ME231: Solid Mechanics-I**

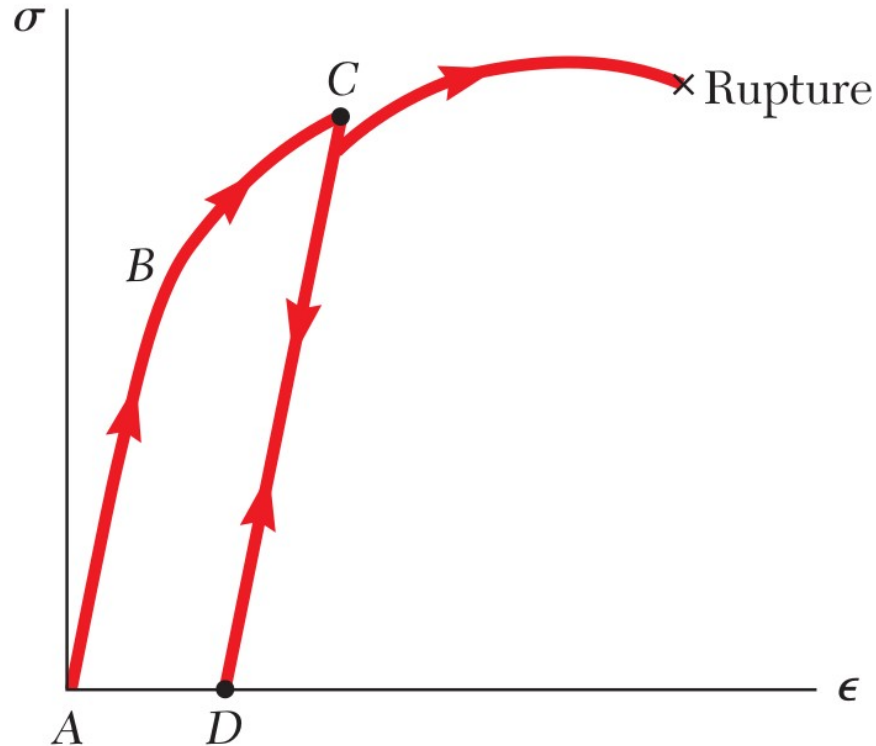
## **Stress, Strain and Temperature relationship**

# Elastic vs. Plastic Behaviour: Unloading



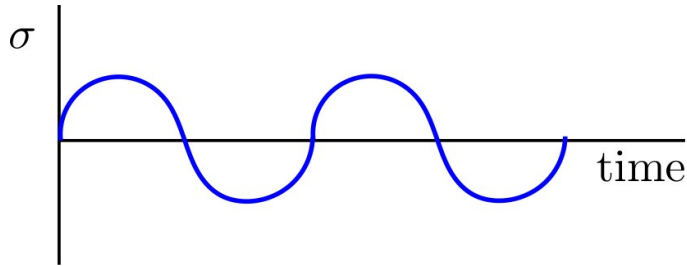
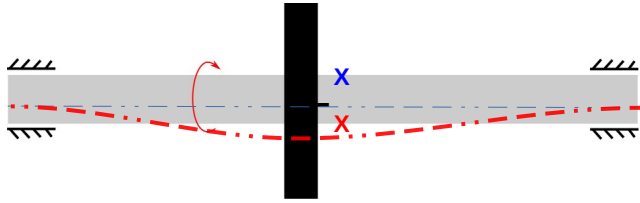
- If a material is subjected beyond its elastic limit (till point  $C$ ) and then the load is removed; the stress and strain decreases in a linear fashion (path  $CD$ ). The linear unloading path  $CD$  is parallel to the initial loading path  $AB$ .
- It should be noted that after complete unloading strain  $\epsilon$  does not return to zero, which indicates that **a permanent set or plastic deformation** of the material has taken place.
- Strain recovered during the unloading process is the **elastic strain**.

# Elastic vs. Plastic Behaviour: Reloading



- If the unloaded test specimen is reloaded under tension, stress-strain curve first follow the path DC, then it will bend to the right and connect with the curved portion of the original stress-strain diagram.
- The straight-line portion of the new loading curve is longer than the corresponding portion of the initial one.
- Thus, the proportional limit and the elastic limit have increased as a result of the strain-hardening that occurred during the earlier loading. However, since the point of rupture remains unchanged, the ductility of the specimen, which should now be measured from point  $D$ , has decreased.

# Repeated loading and fatigue

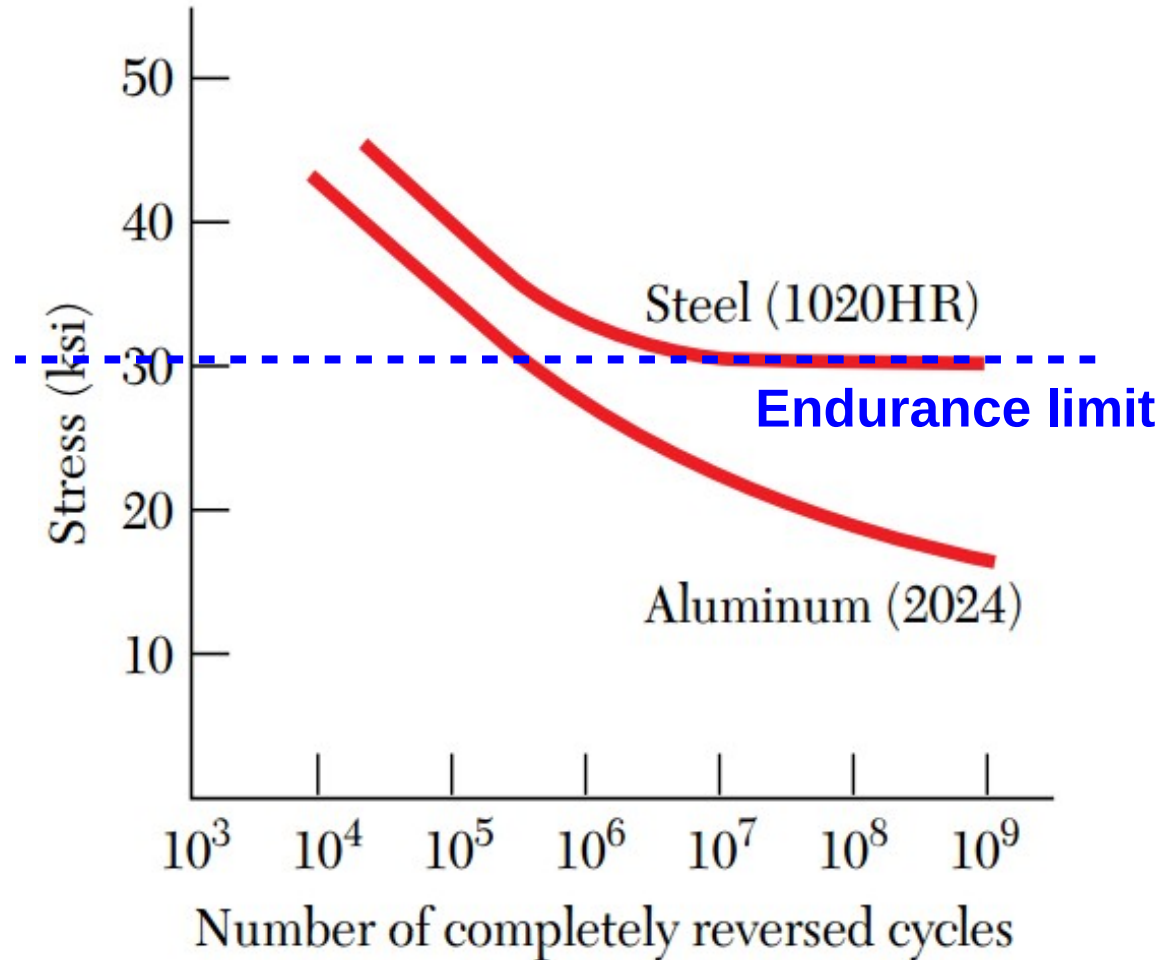


- Most engineering components experience repeated or fluctuating load
- For example,
  - A beam supporting an industrial crane can be loaded as many as two million times in 25 years (about 300 loadings per working day)
  - An automobile crankshaft is loaded about half a billion times if the automobile is driven 200,000 miles,
  - An individual turbine blade can be loaded several hundred billion times during its lifetime.

- When loadings are repeated thousands or millions of times, then the rupture can occur at a stress much lower than the static breaking strength; this phenomenon is known as **fatigue**.
- A fatigue failure is of a **brittle nature**, even for materials that are normally ductile.

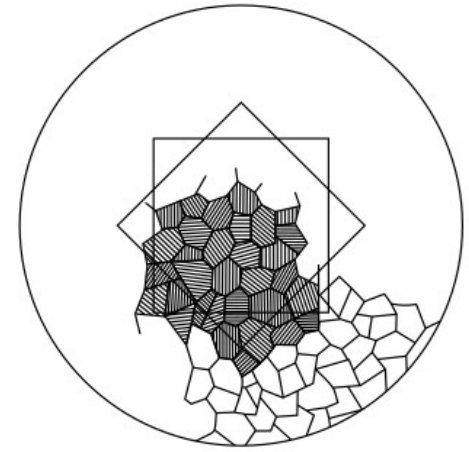
# Endurance limit

$\sigma - n$  curve  
or  
 $s - n$  curve



# Elastic stress-strain relations

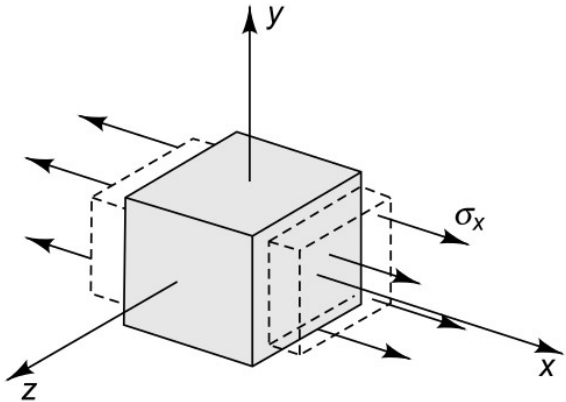
- We already developed stress-strain relationship as Hooke's law for special case of one dimensional loading.
- We will now generalize the elastic behaviour and establish the relationship between six components of stress and six components of elastic strain.
- We will restrict our self to all **linearly elastic materials**. We will also restrict our self to definitions of **strain for small deformations**.
- We will also assume materials to be **homogeneous isotropic**. An isotropic material is defined as one whose properties are **independent of orientation**.
- Materials made up of randomly oriented structural elements may be thought of as being **statistically isotropic**.
- Homogeneity implied that material properties are **independent of position**.



Statistically isotropic  
material

Consider an element on which only one component of normal stress is acting. This normal component of stress will produce a corresponding normal component of strain. Relation between the normal stress and normal strain produced is,

$$\epsilon_x = \frac{\sigma_x}{E} \dots\dots\dots(6)$$

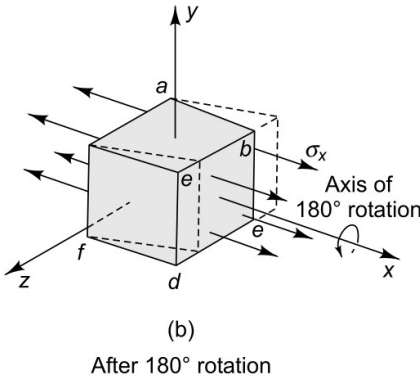
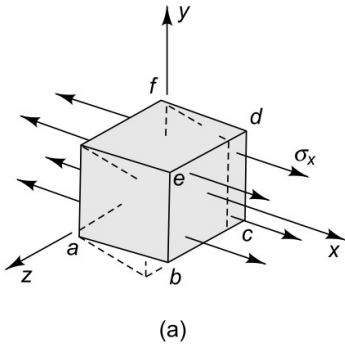


From the measurement made during the uniaxial tensile test, it is observed that there are deformations in the lateral directions also. It is found that lateral strain is a fixed fraction of the normal strain. This fixed fraction is called **Poisson's ratio** and is denoted by the symbol  $\nu$ . Thus, lateral strain can be defined as,

$$\epsilon_y = \epsilon_z = -\nu\epsilon_x = -\nu\frac{\sigma_x}{E}. \dots\dots\dots(7)$$

The possibility of occurrence of shear strain because of normal stress  $\sigma_x$  can be discarded because of isotropy.

**Thus normal stress will produce only normal strains.**



Now, if normal stress  $\sigma_y$  is considered then, normal strain in  $y$ -direction will be

$$\epsilon_y = \frac{\sigma_y}{E}, \quad \dots\dots\dots(8)$$

and corresponding lateral strains will be,  $\epsilon_x = \epsilon_z = -\nu\epsilon_y = -\nu\frac{\sigma_y}{E}$ . .....(9)

Similarly for normal stress  $\sigma_z$  corresponding strains are,

$$\epsilon_z = \frac{\sigma_z}{E}, \quad \text{and} \quad \epsilon_x = \epsilon_y = -\nu\epsilon_z = -\nu\frac{\sigma_z}{E}. \quad \dots\dots\dots(10)$$

Under the most general loading condition, **shear stresses does not affect the normal strains directly when deformations are small.** **Also shear stresses in a direction does not affect shear strains in other directions.** Hence, Hooke’s law for shear stresses is

$$\gamma_{xy} = \frac{\tau_{xy}}{G}, \quad \gamma_{xz} = \frac{\tau_{xz}}{G}, \quad \text{and} \quad \gamma_{yz} = \frac{\tau_{yz}}{G}. \quad \dots\dots\dots(11)$$

where  $G$  is called the **shear modulus**.



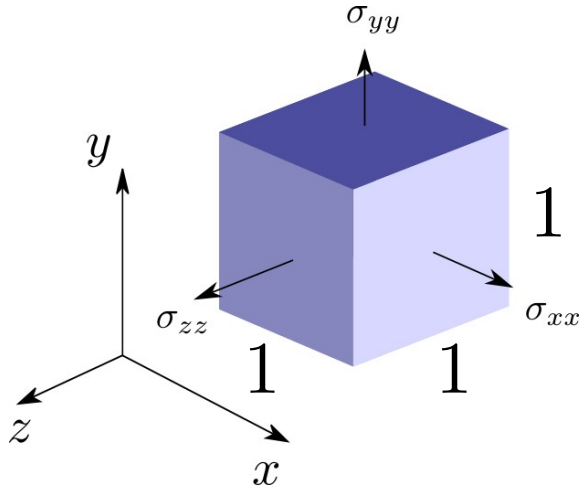
# Multi-axial loading: Generalized Hooke's Law

Consider a case where all stress components are  $\sigma_{xx}, \sigma_{yy}, \sigma_{zz}, \tau_{xy}, \tau_{xz}$  and  $\tau_{yz}$  acting simultaneously, then **within the limits of linear elasticity and small deformations** stresses and strains can be related as,

$$\begin{aligned}\epsilon_{xx} &= \frac{\sigma_{xx}}{E} - \nu \frac{\sigma_{yy}}{E} - \nu \frac{\sigma_{zz}}{E}, & \gamma_{xy} &= \frac{\tau_{xy}}{G}, \\ \epsilon_{yy} &= \frac{\sigma_{yy}}{E} - \nu \frac{\sigma_{xx}}{E} - \nu \frac{\sigma_{zz}}{E}, & \gamma_{xz} &= \frac{\tau_{xz}}{G}, \\ \epsilon_{zz} &= \frac{\sigma_{zz}}{E} - \nu \frac{\sigma_{yy}}{E} - \nu \frac{\sigma_{xx}}{E}, & \gamma_{yz} &= \frac{\tau_{yz}}{G}.\end{aligned}\dots\dots\dots(12)$$

These equations are known as the **generalized Hooke's law**. These equations involves three constants  $E$ ,  $G$  and  $\nu$ .

# Dilatation and Bulk Modulus



Consider a cubic material element having unit volume shown in its unstressed state. Under the stresses  $\sigma_{xx}$ ,  $\sigma_{yy}$  and  $\sigma_{zz}$  it deforms into a rectangular parallelepiped of volume  $v$ , where

$$v = (1 + \epsilon_x)(1 + \epsilon_y)(1 + \epsilon_z).$$

As strains are smaller than unity, we can write,

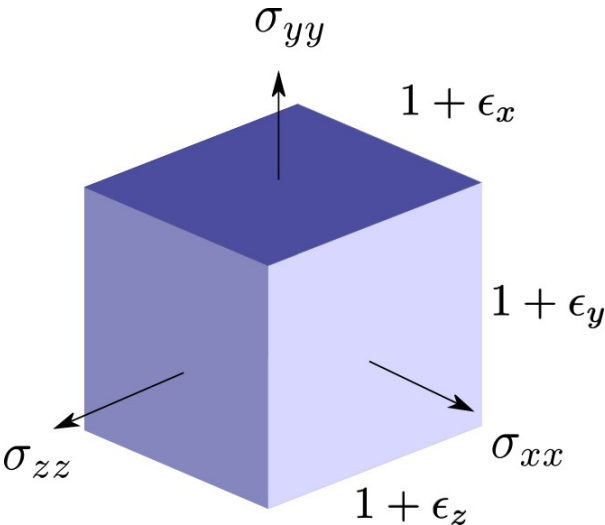
$$v \approx 1 + \epsilon_x + \epsilon_y + \epsilon_z.$$

Now the change in volume is

$$e = v - 1 = \epsilon_{xx} + \epsilon_{yy} + \epsilon_{zz} \quad \dots\dots\dots(13)$$

Here,  $e$  represents **the change in volume per unit volume** which is called **dilatation** of the material. Using (12) we can rewrite (13) as,

$$e = \frac{1 - 2\nu}{E} (\sigma_{xx} + \sigma_{yy} + \sigma_{zz}). \quad \dots\dots\dots(14)$$



If a body is subjected to uniform hydrostatic pressure, i.e.,  $\sigma_{xx} = \sigma_{yy} = \sigma_{zz} = -p$ , then (14) yields

$$e = -\frac{3p(1 - 2\nu)}{E} = -\frac{p}{k}, \quad \dots\dots\dots(15)$$

where  $k = \frac{E}{3(1 - 2\nu)}$  is a material constant, known as **bulk modulus** of the material.

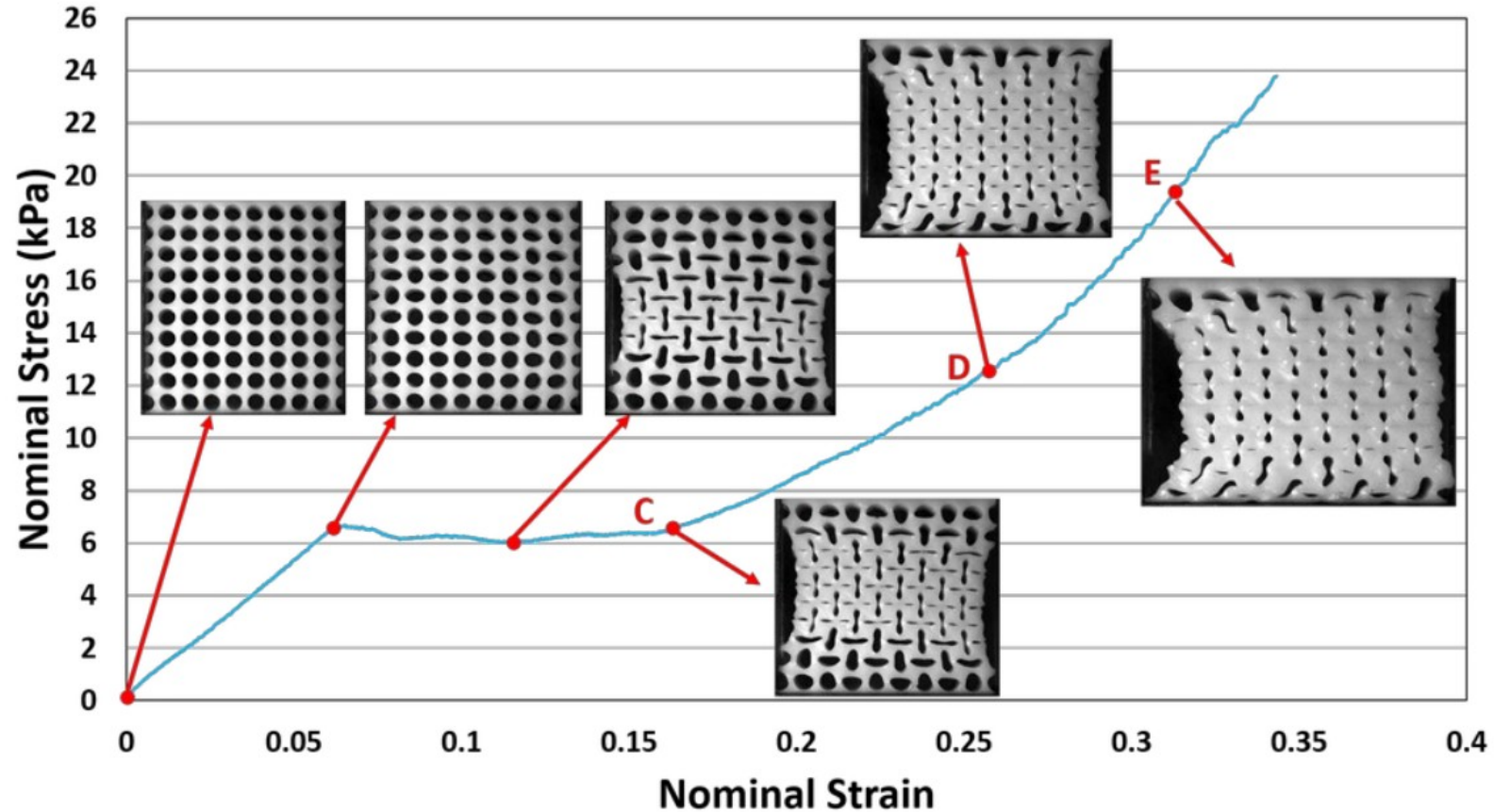
Bulk modulus is defined as the ratio of pressure to dilatation/volumetric strain ( $e$ ). Note that  $k$  is always positive, as hydrostatic pressure will always decrease the volume.

Hence,  $(1-2\nu)>0$  or  $\nu < 0.5$ .  $\nu$  is also positive, hence for any engineering material

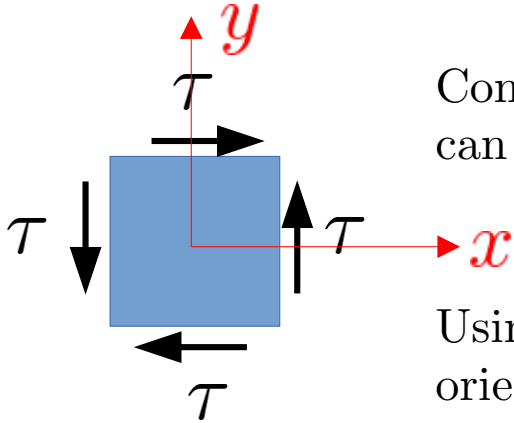
$$0 < \nu < 0.5.$$

- $\nu=0$  : Stretching is one directional without contraction in lateral direction.
- $\nu=0.5$ , i.e.,  $k=\infty$ , which means, zero dilatation or no change in volume when pressure is applied. i.e., perfectly incompressible materials.

# Structures with negative Poisson's ratio



# Relationship between $E$ , $\nu$ and $G$

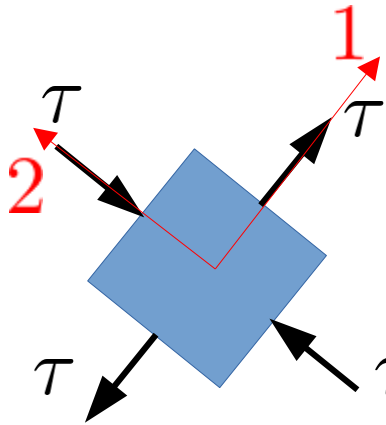


Consider a material element in pure shear loading. Using Hooke's law, we can write shear strain in the element as,

$$\gamma_{xy} = \frac{\tau}{G}. \quad \dots\dots\dots(16)$$

Using stress transformation, let us determine the state of stress at angle orientation of  $45^\circ$ . We already did this as exercise and shown that the state of stress at  $45^\circ$  orientation of the element will be as follows.

For this element, applying generalized Hooke's law yields,



$$\begin{aligned} \epsilon_1 &= \frac{\tau}{E} - \nu \frac{-\tau}{E} = \frac{(1 + \nu)\tau}{E} \\ \epsilon_2 &= \frac{-\tau}{E} - \nu \frac{\tau}{E} = -\frac{(1 + \nu)\tau}{E} \end{aligned} \quad \dots\dots\dots(17)$$

Maximum shear strain is nothing but  $\gamma_{xy}$ , which can be determined as

$$\gamma_{xy} = \epsilon_1 - \epsilon_2 = \frac{2(1 + \nu)}{E} \tau. \quad \dots\dots\dots(18)$$

Now equating (16) and (18) we can write,

$$G = \frac{E}{2(1 + \nu)}. \quad \text{.....(19)}$$

Thus for an isotropic elastic material there are just **two independent elastic constants**.