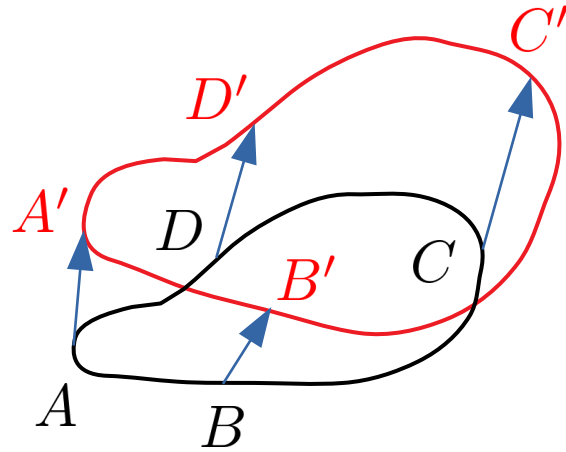


ME231: Solid Mechanics-I

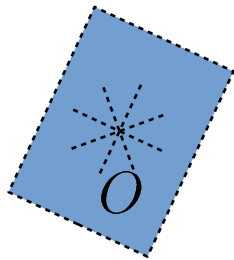
Stress and Strain

Deformation

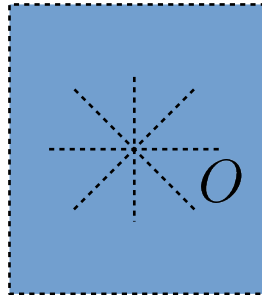


- Displacement of a continuous body may consist of,
- Rigid body displacement
 - Rigid body rotation
 - Relative displacements between points (deformation)

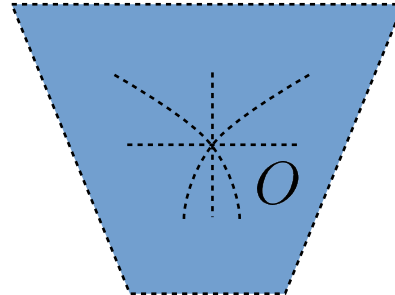
Rigid body
rotation



Uniform
deformation



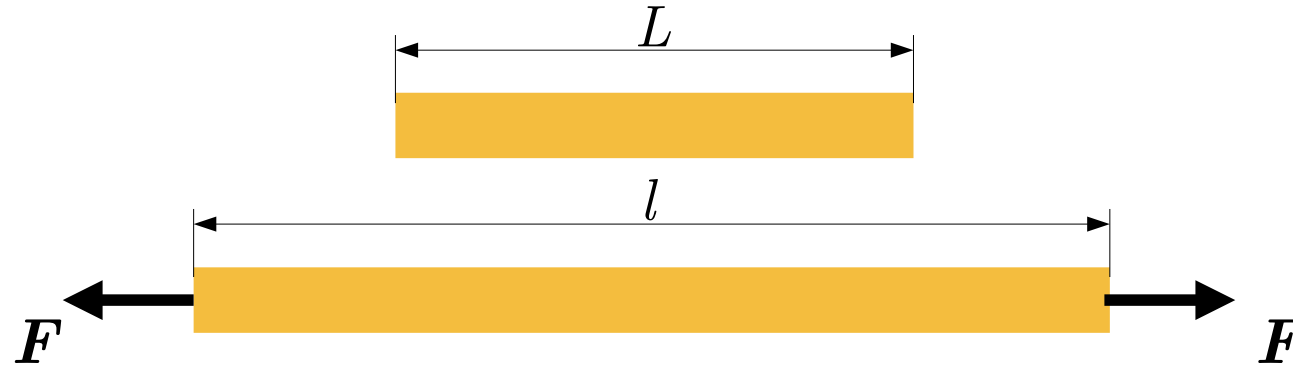
Non-uniform
deformation



For negligible area surrounding point O , deformation can be assumed to be uniform even if overall deformation is non-uniform.

- The displacements associated with rigid-body motion can be either large or small, while the displacements associated with deformation usually are small.
- The description and analysis of rigid-body motion is important in dynamics where the forces required to produce different time rates of rigid-body motion are of interest.
- The description and analysis of deformation is important in our present study of the mechanics of deformable bodies where the forces required to produce different distortions are of interest.
- To start with, we focus our attention on a body whose particles all lie in the same plane and which deforms only in this plane. This type of deformation is called **plane strain**.

Measurement of deformation: Strain



Strain is a measure of deformation. For uniaxial condition strain is defined as follows.

Engineering or nominal strain is defined as **deformation per unit original length**.

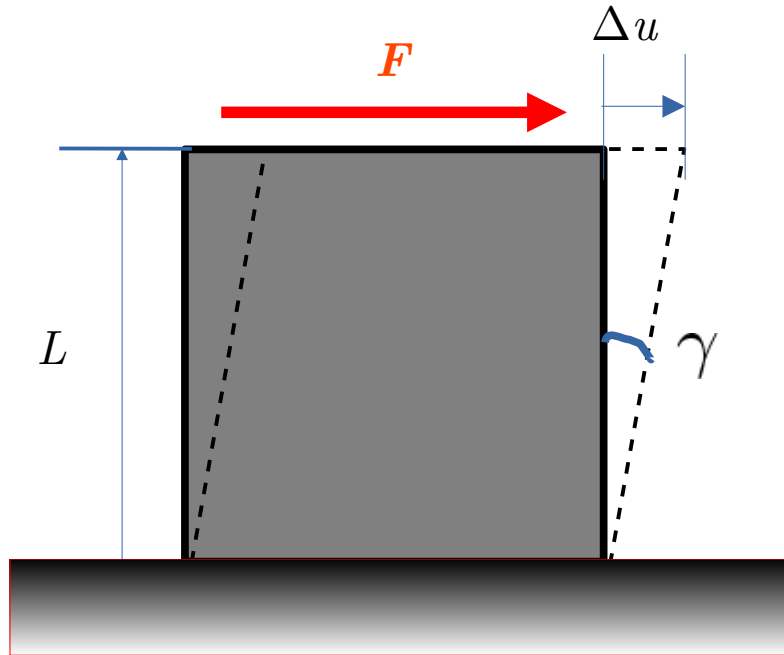
$$\varepsilon = \frac{l - L}{L} = \frac{\Delta L}{L}. \quad \text{.....(32)}$$

Another definition of strain is called **true or logarithmic strain** defined as,

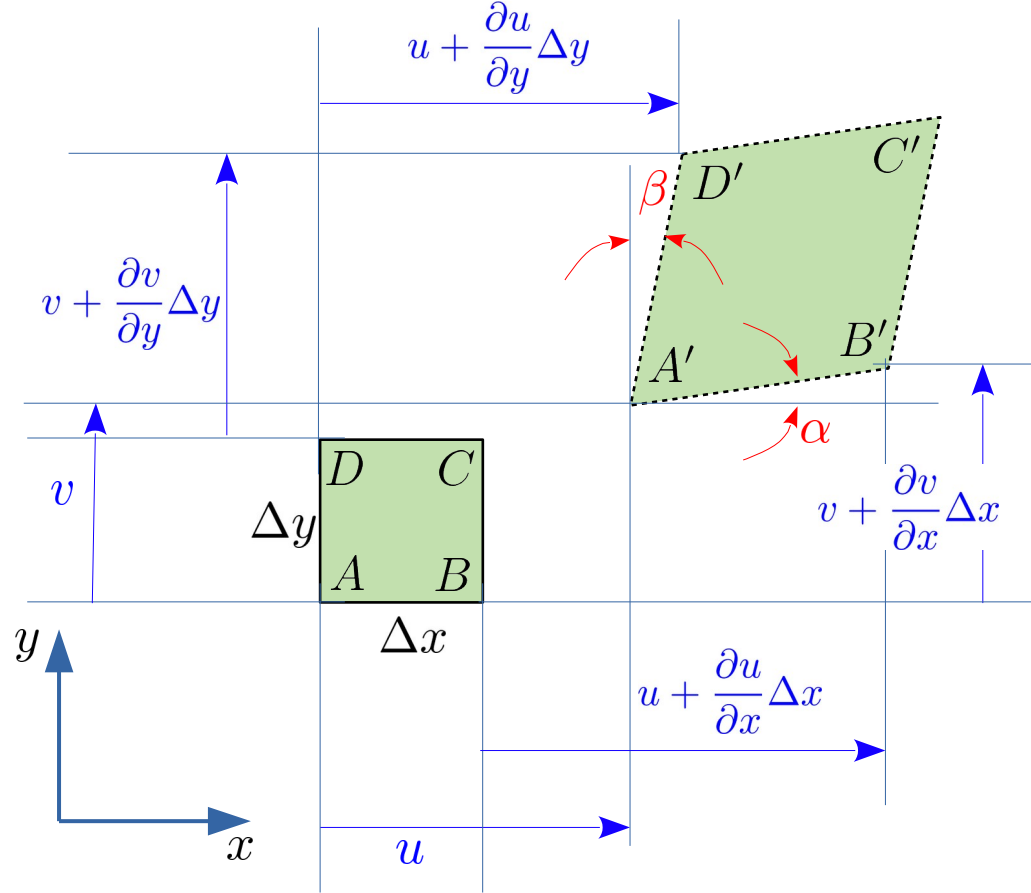
$$e = \int_L^l \frac{dl}{l} = \ln \frac{l}{L}. \quad \text{.....(33)}$$

In case of tangential force, **shear strain** is defined as

$$\tan \gamma \approx \gamma = \frac{\Delta u}{L}.$$



Plane strain in case of small deformations



$$\epsilon_x = \lim_{\Delta x \rightarrow 0} \frac{A'B' - AB}{AB} = \lim_{\Delta x \rightarrow 0} \frac{\Delta x + \frac{\partial u}{\partial x} \Delta x - \Delta x}{\Delta x},$$

$$\Rightarrow \epsilon_x = \frac{\partial u}{\partial x},$$

$$\epsilon_y = \lim_{\Delta y \rightarrow 0} \frac{A'D' - AD}{AD} = \lim_{\Delta y \rightarrow 0} \frac{\Delta y + \frac{\partial v}{\partial y} \Delta y - \Delta y}{\Delta y},$$

$$\Rightarrow \epsilon_y = \frac{\partial v}{\partial y},$$

$$\gamma_{xy} = \lim_{\substack{\Delta x \rightarrow 0 \\ \Delta y \rightarrow 0}} \frac{\pi}{2} - \angle B'A'D' = \alpha + \beta$$

$$\Rightarrow \gamma_{xy} = \lim_{\substack{\Delta x \rightarrow 0 \\ \Delta y \rightarrow 0}} \frac{\frac{\partial v}{\partial x} \Delta x}{\Delta x + \frac{\partial u}{\partial x} \Delta x} + \frac{\frac{\partial u}{\partial y} \Delta y}{\Delta y + \frac{\partial v}{\partial y} \Delta y}$$

$$\Rightarrow \gamma_{xy} \approx \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}.$$

Strain-displacement relations(in 2D)

Engineering strain components,

$$\epsilon_x = \frac{\partial u}{\partial x}, \quad \epsilon_y = \frac{\partial v}{\partial y}, \quad \gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \quad \text{.....(34)}$$

Similar to stress, strain is also a second order tensor. In plane strain case, strain matrix is defined as,

$$[\epsilon] = \begin{bmatrix} \epsilon_{xx} & \gamma_{xy}/2 \\ \gamma_{yx}/2 & \epsilon_{yy} \end{bmatrix} \quad \text{.....(35)}$$

Strain transformation

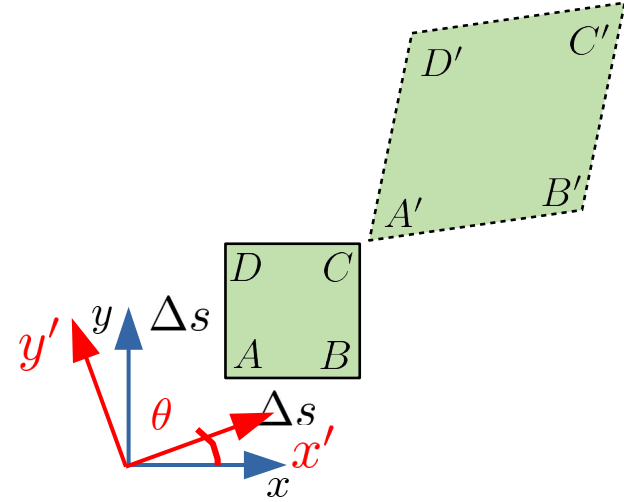
Being a second order tensor, strain tensor follows general rules of transformations.

$$[\epsilon]' = [Q]^T [\epsilon] [Q],$$

Following this equation transformed strains will be given as

$$\begin{aligned}\epsilon'_{xx} &= \frac{(\epsilon_{xx} + \epsilon_{yy})}{2} + \frac{(\epsilon_{xx} - \epsilon_{yy})}{2} \cos 2\theta + \frac{\gamma_{xy}}{2} \sin 2\theta, \\ \epsilon'_{yy} &= \frac{(\epsilon_{xx} + \epsilon_{yy})}{2} - \frac{(\epsilon_{xx} - \epsilon_{yy})}{2} \cos 2\theta - \frac{\gamma_{xy}}{2} \sin 2\theta, \\ \gamma'_{xy} &= \frac{\gamma_{xy}}{2} \cos 2\theta - \frac{(\epsilon_{xx} - \epsilon_{yy})}{2} \sin 2\theta.\end{aligned}$$

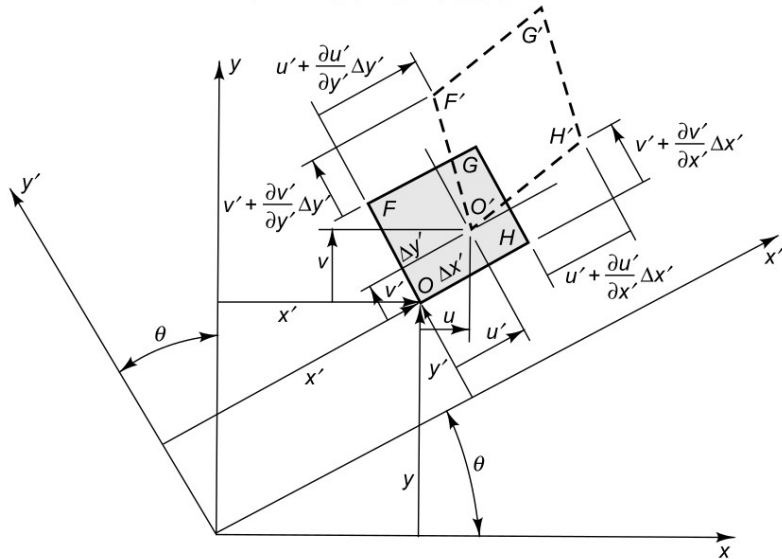
.....(36)



Strain transformation equations (36) can also be derived from purely geometrical relations. Relations between the coordinate and displacements in xy -system and $x'y'$ -system is given as

$$\begin{aligned} x &= x' \cos \theta - y' \sin \theta & u' &= u \cos \theta + v \sin \theta \\ y &= x' \sin \theta + y' \cos \theta. & v' &= -u \sin \theta + v \cos \theta. \end{aligned} \quad \dots\dots\dots(38)$$

Now, using strain-displacement relations,



$$\begin{aligned} \epsilon'_x &= \frac{\partial u'}{\partial x'} = \frac{\partial u'}{\partial x} \frac{\partial x}{\partial x'} + \frac{\partial u'}{\partial y} \frac{\partial y}{\partial x'} = \frac{\partial u'}{\partial x} \cos \theta + \frac{\partial u'}{\partial y} \sin \theta \\ &\Rightarrow \left(\frac{\partial u}{\partial x} \cos \theta + \frac{\partial v}{\partial x} \sin \theta \right) \cos \theta + \left(\frac{\partial u}{\partial y} \cos \theta + \frac{\partial v}{\partial y} \sin \theta \right) \sin \theta \\ &\Rightarrow \frac{\partial u}{\partial x} \cos^2 \theta + \frac{\partial v}{\partial y} \sin^2 \theta + \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \sin \theta \cos \theta, \\ &\Rightarrow \epsilon_x \cos^2 \theta + \epsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta. \end{aligned} \quad \dots\dots\dots(39)$$

$$\begin{aligned}
\epsilon'_y &= \frac{\partial v'}{\partial y'} = \frac{\partial v'}{\partial x} \frac{\partial x}{\partial y'} + \frac{\partial v'}{\partial y} \frac{\partial y}{\partial y'} = -\frac{\partial v'}{\partial x} \sin \theta + \frac{\partial v'}{\partial y} \cos \theta \\
&\Rightarrow -\left(-\frac{\partial u}{\partial x} \sin \theta + \frac{\partial v}{\partial x} \cos \theta\right) \sin \theta + \left(-\frac{\partial u}{\partial y} \sin \theta + \frac{\partial v}{\partial y} \cos \theta\right) \cos \theta \\
&\Rightarrow \frac{\partial v}{\partial y} \cos^2 \theta + \frac{\partial u}{\partial x} \sin^2 \theta - \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}\right) \sin \theta \cos \theta \\
&\Rightarrow \epsilon_y \cos^2 \theta + \epsilon_x \sin^2 \theta - \gamma_{xy} \sin \theta \cos \theta \quad \dots\dots\dots(40)
\end{aligned}$$

$$\begin{aligned}
\gamma'_{xy} &= \frac{\partial u'}{\partial y'} + \frac{\partial v'}{\partial x'} = \left(\frac{\partial v'}{\partial x} \frac{\partial x}{\partial y'} + \frac{\partial v'}{\partial y} \frac{\partial y}{\partial y'}\right) + \left(\frac{\partial u'}{\partial x} \frac{\partial x}{\partial y'} + \frac{\partial u'}{\partial y} \frac{\partial y}{\partial y'}\right) \\
&\Rightarrow \left(\frac{\partial v}{\partial y} - \frac{\partial u}{\partial x}\right) \sin \theta \cos \theta + \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}\right) (\cos^2 \theta - \sin^2 \theta) \\
&\Rightarrow (\epsilon_y - \epsilon_x) \sin \theta \cos \theta + \gamma_{xy} (\cos^2 \theta - \sin^2 \theta) \quad \dots\dots\dots(41)
\end{aligned}$$

Use of trigonometric identities will results in equations identical to the equations derived from tensor transformations.

Principal strains and maximum shear strain

- Planes at which **shear strain is zero** are called **principal planes** of strains and the normal strains at these planes are called **principal strains**. Inclination of principal planes from x -axis are given by

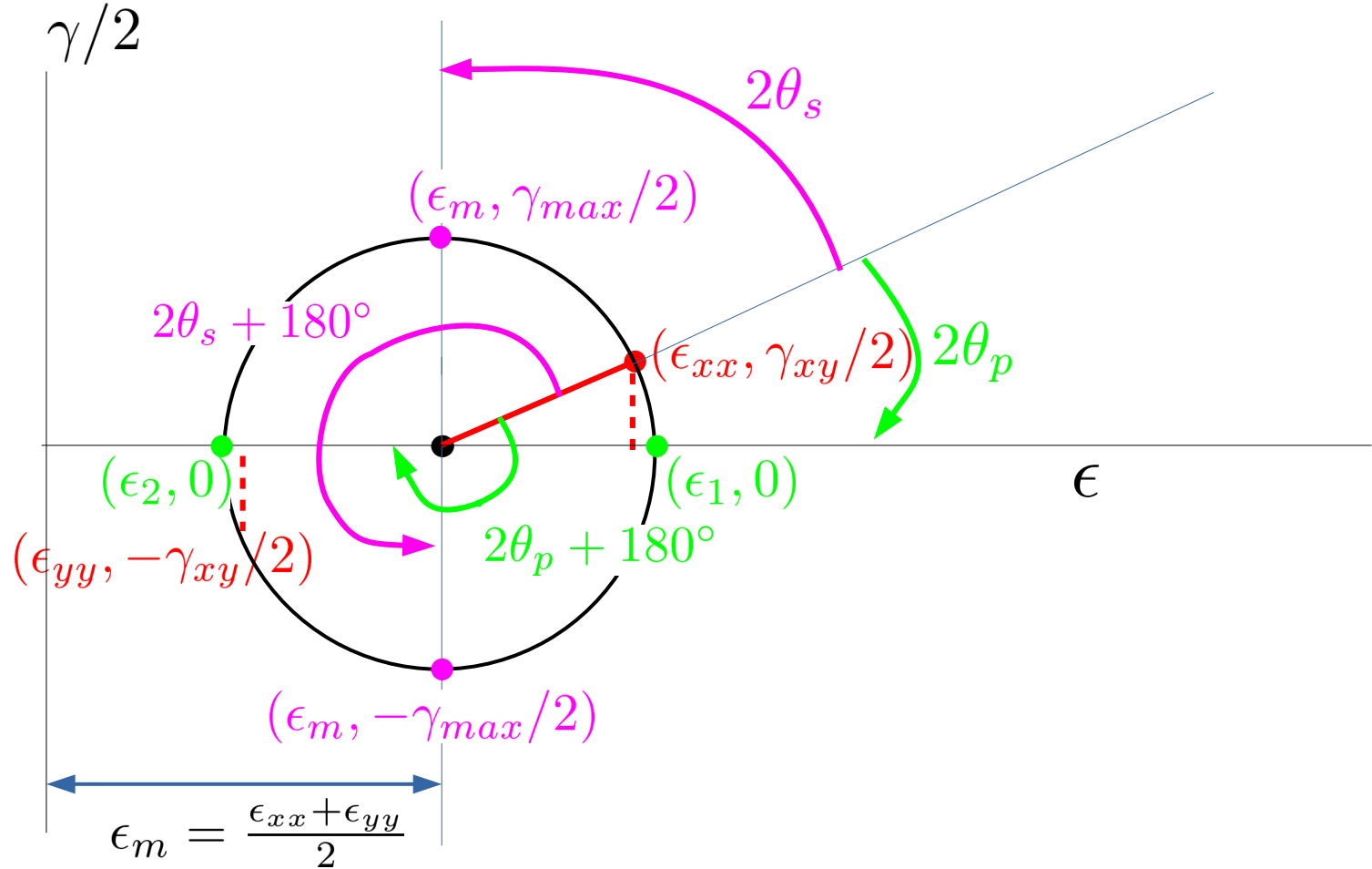
$$\tan 2\theta_p = \frac{\gamma_{xy}}{\epsilon_{xx} - \epsilon_{yy}} \quad \dots\dots\dots(42)$$

- Maximum value of shear strain is given as,

$$\gamma_{\max} = \sqrt{(\epsilon_{xx} - \epsilon_{yy})^2 + \gamma_{xy}^2}, \text{ at angle } \theta \text{ satisfying } \tan 2\theta_s = -\frac{\epsilon_{xx} - \epsilon_{yy}}{\gamma_{xy}}. \quad \dots\dots\dots(43)$$

- Similar to stresses, Mohr's circle can be drawn for strains also.

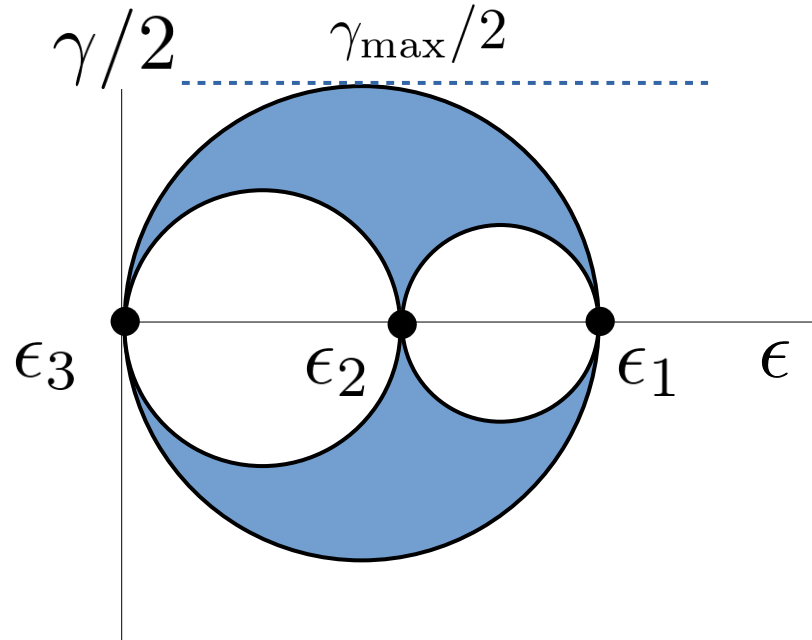
Mohr's circle for strains



Principal strains in plane strain

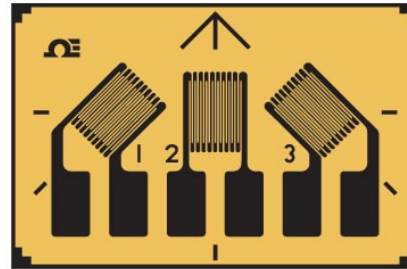
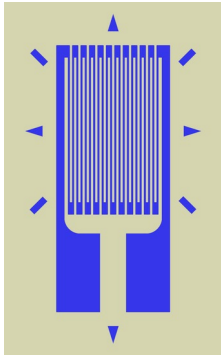
For plane strain, $\varepsilon_{zz} = 0$. In case ε_1 and, $\varepsilon_2 > 0$, then maximum stresses will be calculated as,

$$\gamma_{\max} = |\epsilon_{\max} - \epsilon_{\min}| = |\epsilon_3 - \epsilon_1|$$

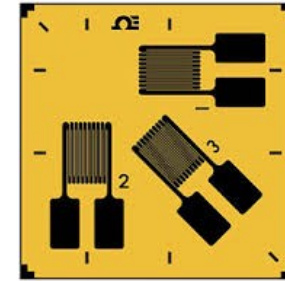


Measurement of strains

- **Electrical strain gages** are used to measure strains.
- Strain gages work on the principal that certain metals exhibit a **change in electrical resistance with change in mechanical strain**.
- Strain gages are bonded on any surface and it measure **strain the axial direction** of the gage.
- **Strain rosette** are combination of strain gages, used to measure strains in different directions and calculate principal strains.

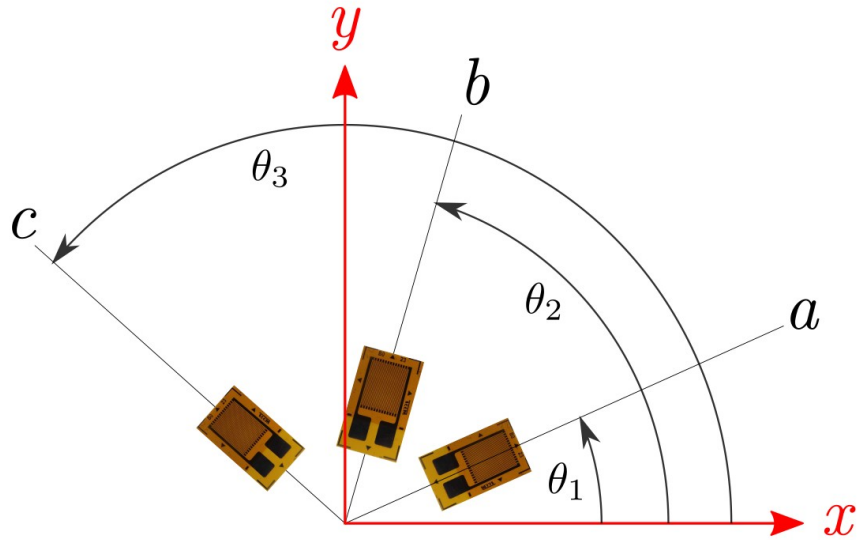


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Strain rosette



If, in-plane strain components are ϵ_x , ϵ_y , γ_{xy} , then strains in the direction of a , b , and c can be expression using in-plane strains and the inclination from x -axis, as

$$\epsilon_a = \epsilon_x \cos^2 \theta_1 + \epsilon_y \sin^2 \theta_1 + \gamma_{xy} \sin \theta_1 \cos \theta_1,$$

$$\epsilon_b = \epsilon_x \cos^2 \theta_2 + \epsilon_y \sin^2 \theta_2 + \gamma_{xy} \sin \theta_2 \cos \theta_2,$$

$$\epsilon_c = \epsilon_x \cos^2 \theta_3 + \epsilon_y \sin^2 \theta_3 + \gamma_{xy} \sin \theta_3 \cos \theta_3.$$

.....(44)

These three equations can be solved to find three unknowns i.e. ϵ_x , ϵ_y , and, γ_{xy} . With the knowledge of all in-plane strain components, principal strains and their directions can also be calculated.