

A discrete calculus approach to simulating hydraulic transients in pipe networks

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Abstract

Typically hydraulic transients in pipe networks are simulated using either the method of characteristics (MOC) or the wave characteristics method (WCM). The WCM simplifies the calculations in order to decrease the computation cost at the expense of accuracy whereas the MOC offers higher spatial resolution but may require significantly more computational resources. The time step in the MOC may need to be very small in order to ensure the numerical stability of the solution which can have a significant impact on the run time for a simulation. This paper develops a discrete calculus approach to the problem of hydraulic transients in pipe networks which aims to maintain the flexibility of the WCM without compromising the accuracy that is obtained using the MOC. The scheme employed herein allows spatial and temporal discretisation to be refined as necessary without the need to alter pipe wave speeds in order to satisfy the Courant-Friedrich-Lewy (CFL) stability condition.

1 Introduction

Alterations to steady state pipe flow, such as valve closure, may cause severe pressure fluctuations throughout the system until a new steady state flow is recovered. This transient flow can have a significant impact on the design of piping networks as the system must be made sufficiently stronger to cope with sudden surges or drops in pressure. Pressure waves travel from the source of the disturbance at the speed of sound for the fluid medium usually being dissipated by viscous effects or by the geometry of the system.

1.1 Water hammer

Typically a time varying one-dimensional model for transient flow in closed conduits is constructed from the conservation of mass and momentum and Reynolds transport theorem (see chapter 2 of [1] for details). These equations govern either the pressure p and velocity V of the fluid or the piezometric head H and the discharge Q as functions of the spatial coordinate x and time t . The flow is assumed to be slightly compressible and the conduit walls slightly deformable such that spatial variation of the fluid density ρ and the conduit cross-sectional area A are neglected. Thus the water hammer equations, in terms of H and Q , are given by

$$\frac{\partial H}{\partial t} + \frac{a^2}{gA} \frac{\partial Q}{\partial x} = C(x), \quad (1a)$$

$$\frac{\partial Q}{\partial t} + gA \frac{\partial H}{\partial x} = \hat{R}(Q), \quad (1b)$$

where g is the acceleration due to gravity, a is the velocity of the pressure wave in an elastic conduit filled with a slightly compressible fluid, $C(x)$ is the known spatially varying consumption (flow into or out of the system) and $\hat{R}(Q)$ is a resistance term which specifies the pipe friction acting on the fluid. Typically the resistance term is taken to be

$$\hat{R}(Q) = -\frac{f(Q)Q|Q|}{2DA}, \quad (1c)$$

where D is the conduit diameter and $f(Q)$ is the dimensionless Darcy-Weisbach friction factor which incorporates friction losses and is a function of the flow Reynolds number. The velocity of the pressure wave in the conduit is determined by

$$a^2 = \frac{\frac{K}{\rho}}{1 + \frac{DK}{eE}} = \frac{eEK}{\rho(eE + DK)}, \quad (2)$$

where K is the bulk modulus of elasticity for the fluid, e is the thickness of the conduit walls and E is the Young's modulus for the container material. The friction factor f depends both upon the Reynold number of the flow and the relative roughness of the pipe. There are a number of different empirical equations for determining the friction factor f , such as the implicit Colebrook-White equation (this should be looked at in more detail as there are a number of different options).

Equations (1) are the basic equations governing transient flow in a pipe, terms can be added or modified to account for more complex phenomena such as pipe inclination or a more exotic friction term (see Wylie & Streeter for details[5]). The numerical solution of (1) may be approached in a number of different ways including the method of characteristics, the wave characteristics method, explicit finite-difference schemes (and variations such as Lax-Wendroff and MacCormack), Implicit finite-difference schemes and spectral methods.

In general the flows we will be modelling exist in complex networks of interconnected components such as pipes, valves and pumps which may be modelled as resistances to the flow through the particular component. Although these networks exist as three-dimensional structures we may simplify our analysis by considering these components to be connected one-dimensional objects. This approach presents challenges for us in modelling the connections between components but significantly reduces the complexity when solving the system numerically. This approach is analogous to the idea of the circuit diagram often used in electrical engineering and will be presented in a similar way.

2 Discrete calculus

We wish to describe the temporal variation of the volume flow rate Q and the head H in a one-dimensional domain. This presents us with a bit of problem when it comes to determining flows in networks as connections between more than two pipes/components aren't one-dimensional. One way to approach this difficulty is by considering a discrete calculus [3] formulation for the governing equations.

2.1 Discrete calculus formulation

The discrete calculus approach has the advantage of directly using the network structure to calculate the derivatives. In the discrete calculus formulation a network is represented as a graph (1-complex) where the edges (1-cells) represent pipes/components and the nodes (0-cells) represent connections and boundaries. The edges of the graph consist of ordered pairs of nodes which define the orientation of the edge. The boundary of each edge is comprised of the

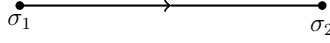


Figure 1: The edge points from σ_1 to σ_2 ; flow from σ_1 to σ_2 is considered positive.

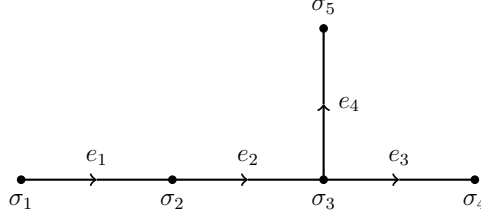


Figure 2: An example graph with 5 nodes and 4 edges.

union of nodes and the intersection of any two edges is either empty or a boundary element of both edges.

Flow passing through an edge in the same direction as its orientation is considered positive while flow in the opposite direction is considered negative. We will consider edges that are oriented but not directed since flow is permitted in both directions. An edge consists of two nodes (labelled σ_1 and σ_2) and the orientation may be defined by an ordering of these nodes as the list $\tau = \{\sigma_1, \sigma_2\}$. A node is considered to have two orientations ("sourceness" and "sinkness") but by convention all nodes will be given the same orientation, "sourceness", so that the negative end of an edge will not be coherent with the orientation of a node.

2.1.1 Incidence matrix

The structure of a graph (network) may be represented algebraically via an incidence matrix. The incidence matrix \mathbf{K}^T encodes which edges are incident to which nodes in the graph and is defined as

$$K_{ij} = \begin{cases} 0 & \text{if } \sigma_j \text{ is not on the boundary of } e_i, \\ +1 & \text{if } \sigma_j \text{ is coherent with the orientation of } e_i, \\ -1 & \text{if } \sigma_j \text{ is not coherent with the orientation of } e_i, \end{cases} \quad (3)$$

where e_i is the label of an edge in the graph. For a graph the incidence matrix \mathbf{K}^T is an $n \times m$ matrix (where n is the number of nodes and m is the number of edges). For example the graph shown in figure 2 has

$$\mathbf{K} = \begin{bmatrix} 1 & -1 & 0 & 0 & 0 \\ 0 & 1 & -1 & 0 & 0 \\ 0 & 0 & 1 & -1 & 0 \\ 0 & 0 & 1 & 0 & -1 \end{bmatrix},$$

such that the incidence matrix is given by

$$\mathbf{K}^T = \begin{bmatrix} 1 & 0 & 0 & 0 \\ -1 & 1 & 0 & 0 \\ 0 & -1 & 1 & 1 \\ 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & -1 \end{bmatrix}.$$

2.1.2 Chains

In order to define a domain of integration on the graph we must define a 1-chain. A 1-chain is an m -tuple of scalars which assigns a coefficient to each edge where m is the number of distinct edges in the graph. A 1-chain is represented by a column vector of size m with zeros in the entries corresponding to edges not included in the chain. A 1-chain may be viewed as an indicator vector representing a set of edges i.e. $\tau_1 = [1, 1, 0, 0]^T$ represents the set of edges $\{e_1, e_2\}$. Generally a 1-chain τ_1 may be expressed as

$$\tau_1 = \sum_{i=1}^m a^i \mathbf{e}_i, \quad a^i \in \mathbb{R}, \quad (4)$$

where the sign of a^i indicates the orientation and \mathbf{e}_i are basis vectors for each edge.

The incidence matrix \mathbf{K}^T maps 1-chains into their corresponding boundary elements. In other words when \mathbf{K}^T is applied to a 1-chain (edge chain) the result is a 0-chain (node chain) i.e. $\tau_0 = \mathbf{K}^T \tau_1$. The chain τ_0 represents the oriented set of nodes on the boundary of the edges represented by τ_1 . For example for the nodes on the boundary of edges e_1 and e_2 we have

$$\tau_0 = \mathbf{K}^T \begin{bmatrix} 1 \\ 1 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \\ -1 \\ 0 \\ 0 \end{bmatrix},$$

which corresponds to the nodes σ_1 and σ_3 as expected. For the boundary of the entire graph we have

$$\tau_0 = \mathbf{K}^T \begin{bmatrix} 1 \\ 1 \\ 1 \\ 1 \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \\ 1 \\ -1 \\ -1 \end{bmatrix},$$

where we see that join points (σ_3) are considered part of the boundary set. The incidence matrix \mathbf{K}^T provides both a representation of the topology of the graph and the boundary operator.

2.1.3 Discrete forms

We may define a vector space that locally maps 1-chains to scalars at each edge in the graph, called 1-cochains. A 1-cochain may be viewed as a function defined on the domain of edges. We represent a 1-cochain as an $m \times 1$ column vector \mathbf{c}^1 where the sign of each coefficient indicates the orientation. We can define integration as the pairing (inner product) of a 1-chain with a 1-cochain, $[\![\mathbf{c}^1, \tau_1]\!] = \mathbf{c}^1 \cdot \tau_1$, producing a scalar quantity.

2.1.4 Metric tensor

Given a set of basic chains σ_i we may define a metric tensor

$$g_{ij} = \langle \sigma_i, \sigma_j \rangle.$$

Typically in the discrete setting, the basis set of edges is defined to be orthogonal such that $g_{ij} = 0$ for $i \neq j$. The metric tensor thus represented by the $m \times m$ diagonal matrix \mathbf{G} where $\mathbf{G} = \text{diag}(g_{ii})$. Converting a 1-chain into its equivalent 1-cochain consists of a simple scaling of the chain coefficients $\mathbf{c}^1 = \mathbf{G} \tau_1$.

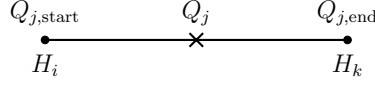


Figure 3: A general component j with an average volume flow rate Q_j and nodal heads H_i and H_k .

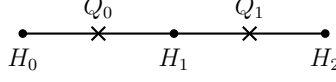


Figure 4: A (very) simple example network.

3 Component networks

In order to model the flow network we shall suppose that a generic components is represented by an edge and the connectivity of those components by nodes. A general component, as shown in 3, has an average mass flow rate defined at the center of the element and nodal pressure at either end. The mass flow rate is assumed to vary along the component so that in general the mass flow rate at each end does not equal the average mass flow rate in the component.

For simplicity we shall suppose that the variation in flow rate in an element is linear so that by combining multiple elements we will be able to approximate the true variation and improve this approximation by increasing the number of elements used in the model. Using first order forward and backward differences we have

$$\left. \frac{\partial Q}{\partial x} \right|_{j,\text{end}} = \frac{Q_{j,\text{end}} - Q_j}{\frac{1}{2}L_j}, \quad (5)$$

$$\left. \frac{\partial Q}{\partial x} \right|_{j,\text{start}} = \frac{Q_j - Q_{j,\text{start}}}{\frac{1}{2}L_j}. \quad (6)$$

Also we have

$$\left. \frac{\partial Q}{\partial x} \right|_{j,\text{end}} = -\frac{gA_j}{a_j^2} \frac{\partial H_k}{\partial t}, \quad (7)$$

$$\left. \frac{\partial Q}{\partial x} \right|_{j,\text{start}} = -\frac{gA_j}{a_j^2} \frac{\partial H_i}{\partial t}, \quad (8)$$

therefore

$$Q_{j,\text{end}} = Q_j - \frac{gA_j L_j}{2a_j^2} \frac{\partial H_k}{\partial t}, \quad (9)$$

$$Q_{j,\text{start}} = Q_j + \frac{gA_j L_j}{2a_j^2} \frac{\partial H_i}{\partial t}. \quad (10)$$

Here L_j , A_j and a_j are the component length, area and wavespeed respectively. Conservation of mass requires that the sum of the flow rates into a node equals zero. An example is useful for demonstrating how this condition can be enforced at each node.

Consider the very simple example network shown in figure 4 the continuity condition at each node is given by

$$\begin{bmatrix} Q_{0,\text{start}} \\ Q_{1,\text{start}} - Q_{0,\text{end}} \\ -Q_{1,\text{end}} \end{bmatrix} = \begin{bmatrix} Q_0 + \frac{gA_0 L_0}{2a_0^2} \frac{\partial H_0}{\partial t} \\ Q_1 + \frac{gA_1 L_1}{2a_1^2} \frac{\partial H_1}{\partial t} - Q_0 + \frac{gA_0 L_0}{2a_0^2} \frac{\partial H_1}{\partial t} \\ -Q_1 + \frac{gA_1 L_1}{2a_1^2} \frac{\partial H_2}{\partial t} \end{bmatrix} = \mathbf{C},$$

where \mathbf{C} is the vector of nodal consumptions, which may be written as

$$\mathbf{K}^T \mathbf{Q} + \begin{bmatrix} \frac{gA_0L_0}{2a_0^2} & 0 & 0 \\ 0 & \frac{gA_0L_0}{2a_0^2} + \frac{gA_1L_1}{2a_1^2} & 0 \\ 0 & 0 & \frac{gA_1L_1}{2a_1^2} \end{bmatrix} \frac{\partial \mathbf{H}}{\partial t} = \mathbf{C}.$$

Let \mathbf{K}_+ be the matrix which contains a '1' in an entry only if \mathbf{K} contains '1' in the same entry (otherwise 0) and let \mathbf{K}_- be the matrix which contains a '1' in an entry only if \mathbf{K} contains a '-1', for this example network

$$\mathbf{K} = \begin{bmatrix} 1 & -1 & 0 \\ 0 & 1 & -1 \end{bmatrix}$$

therefore

$$\mathbf{K}_+ = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \quad \text{and} \quad \mathbf{K}_- = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}.$$

Using these definitions we may write the discrete continuity equation as

$$\boxed{\mathbf{K}^T \mathbf{Q} + \mathbf{D} \frac{\partial \mathbf{H}}{\partial t} = \mathbf{C}}, \quad (11)$$

where

$$\mathbf{D} = \mathbf{K}_+^T \mathbf{M} \mathbf{K}_+ + \mathbf{K}_-^T \mathbf{M} \mathbf{K}_-$$

and

$$\mathbf{M} = \text{diag} \left(\frac{gA_j L_j}{2a_j^2} \right),$$

where typically $M_{j,j} = 0$ when component j is not a pipe.

For a network with N_n nodes and N_c components we require $N_n + N_c$ equations in order to determine the flow rates and heads in the network. The system (11) gives us N_n equations for the conservation of mass at each node so we still require N_c additional equations in order to fully specify the system.

3.1 Component equations

In order to maintain generality we shall specify the momentum equation in each edge in terms of a general resistance \mathbf{R} , which is specified for each particular type of component, such that

$$\boxed{\mathbf{B} \frac{\partial \mathbf{Q}}{\partial t} = \mathbf{R}(\mathbf{Q}, \mathbf{KH})}, \quad (12)$$

or in component form

$$B_{j,j} \frac{\partial Q_j}{\partial t} = R_j(Q_j, \Delta H_j). \quad (13)$$

The diagonal matrix \mathbf{B} is a matrix of coefficients, this allows us some flexibility when specifying the momentum equation, typically $B_{j,j} = 1$. Specifying the resistance term in this way allows

for greater flexibility in defining components. The most obvious component that we might consider is a pipe, from the Darcy-Weisbach equation we have

$$\frac{\Delta H}{L} = \frac{fQ^2}{2gDA^2} \implies \frac{fQ^2}{2DA} - \frac{gA\Delta H}{L} = 0, \quad (14)$$

where $f = f(Q)$ is the friction factor. This means we may define a resistance term of the form

$$R_j = -\frac{f(Q_j)Q_j|Q_j|}{2D_jA_j} + \frac{gA_j}{L_j}\Delta H_j. \quad (15)$$

This is the correct form of the resistance term as in the continuous equation we have

$$\frac{\partial Q}{\partial t} = \hat{R}(Q) - gA \frac{\partial H}{\partial x} = R(Q, \partial H).$$

The discrete equations (11) and (12), when supplemented with appropriate boundary conditions, allow the flow rates and heads to be determined throughout the system.

4 Steady state solver

In the absence of transients the discrete continuity equation (11) reduces to

$$\mathbf{K}^T \mathbf{Q} = \mathbf{C} \quad (16)$$

and (12) reduces to

$$\mathbf{R}(\mathbf{Q}, \mathbf{KH}) = \mathbf{0}. \quad (17)$$

We wish to solve (16) along with the nonlinear momentum equations (17).

4.1 Newton iteration

Due to the nonlinear nature of (12) it is necessary to perform Newton iteration in order to converge to a solution. Let us split the flow rate and head vectors into a known/guessed part and a correction such that $\mathbf{Q} = \mathbf{Q}^g + \mathbf{Q}^c$ and $\mathbf{H} = \mathbf{H}^g + \mathbf{H}^c$ then

$$\mathbf{R}(\mathbf{Q}, \mathbf{KH}) \approx \mathbf{R}(\mathbf{Q}^g, \mathbf{KH}^g) + \left. \frac{\partial \mathbf{R}}{\partial \mathbf{Q}} \right|_{\mathbf{Q}^g} \mathbf{Q}^c + \left. \frac{\partial \mathbf{R}}{\partial \mathbf{KH}} \right|_{\mathbf{KH}^g} \mathbf{KH}^c.$$

The continuity equation (16) is now given by

$$\mathbf{K}^T \mathbf{Q}^c = \mathbf{C} - \mathbf{K}^T \mathbf{Q}^g, \quad (18)$$

and the momentum equation (17) is given by

$$-\left. \frac{\partial \mathbf{R}}{\partial \mathbf{KH}} \right|_{\mathbf{KH}^g} \mathbf{KH}^c - \left. \frac{\partial \mathbf{R}}{\partial \mathbf{Q}} \right|_{\mathbf{Q}^g} \mathbf{Q}^c = \mathbf{R}(\mathbf{Q}^g, \mathbf{KH}^g). \quad (19)$$

These equations may be combined into a single system of the form

$$\begin{bmatrix} \mathbf{K}^T & \mathbf{0} \\ -\mathbf{J} & -\mathbf{G} \end{bmatrix} \begin{bmatrix} \mathbf{Q}^c \\ \mathbf{H}^c \end{bmatrix} = \begin{bmatrix} \mathbf{C} - \mathbf{K}^T \mathbf{Q}^g \\ \mathbf{R}(\mathbf{Q}^g, \mathbf{KH}^g) \end{bmatrix}, \quad (20)$$

where $\mathbf{J} = \left. \frac{\partial \mathbf{R}}{\partial \mathbf{Q}} \right|_{\mathbf{Q}^g}$ is the flow Jacobian matrix and $\mathbf{G} = \left. \frac{\partial \mathbf{R}}{\partial \mathbf{KH}} \right|_{\mathbf{KH}^g} \mathbf{K}$. Boundary conditions specifying either the head or consumption at a node will modify this system of equations.

4.1.1 An example of the discrete calculus approach

Consider the simple network shown in figure 4 which has $\mathbf{Q} = [Q_0, Q_1]^T$ and $\mathbf{H} = [H_0, H_1, H_2]^T$, along with

$$\mathbf{K} = \begin{bmatrix} 1 & -1 & 0 \\ 0 & 1 & -1 \end{bmatrix} \quad \text{and} \quad \mathbf{K}^T = \begin{bmatrix} 1 & 0 \\ -1 & 1 \\ 0 & -1 \end{bmatrix}.$$

We can see how the matrix product

$$\mathbf{K}^T \mathbf{Q} = \begin{bmatrix} 1 & 0 \\ -1 & 1 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} Q_0 \\ Q_1 \end{bmatrix} = \begin{bmatrix} Q_0 \\ Q_1 - Q_0 \\ -Q_1 \end{bmatrix} = \mathbf{C}$$

enforces that the sum of the flow rates into a node is equal to the consumption at that node.

The system of equations (without boundary conditions) (20) is given by

$$\begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ -1 & 1 & 0 & 0 & 0 \\ 0 & -1 & 0 & 0 & 0 \\ -\frac{\partial R}{\partial Q}|_{Q_0^g} & 0 & -\frac{\partial R}{\partial \Delta H}|_{H_0^g-H_1^g} & \frac{\partial R}{\partial \Delta H}|_{H_0^g-H_1^g} & 0 \\ 0 & -\frac{\partial R}{\partial Q}|_{Q_1^g} & 0 & -\frac{\partial R}{\partial \Delta H}|_{H_1^g-H_2^g} & \frac{\partial R}{\partial \Delta H}|_{H_1^g-H_2^g} \end{bmatrix} \begin{bmatrix} Q_0^c \\ Q_1^c \\ H_0^c \\ H_1^c \\ H_2^c \end{bmatrix} = \begin{bmatrix} C_0 - Q_0^g \\ C_1 - Q_1^g + Q_0^g \\ C_2 + Q_1^g \\ R(Q_0^g, H_0^g - H_1^g) \\ R(Q_1^g, H_1^g - H_2^g) \end{bmatrix}.$$

If we wished to specify head boundary conditions ($H_0 = H_{known} = 20$) at the first node and ($H_2 = P_{atm}/\rho g = 101325/(999.7 \cdot 9.80665) = 10.33537514$) we would modify the system of equations to be

$$\begin{bmatrix} 0 & 0 & 1 & 0 & 0 \\ -1 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 \\ -\frac{\partial R}{\partial Q}|_{Q_0^g} & 0 & -\frac{\partial R}{\partial \Delta H}|_{H_0^g-H_1^g} & \frac{\partial R}{\partial \Delta H}|_{H_0^g-H_1^g} & 0 \\ 0 & -\frac{\partial R}{\partial Q}|_{Q_1^g} & 0 & -\frac{\partial R}{\partial \Delta H}|_{H_1^g-H_2^g} & \frac{\partial R}{\partial \Delta H}|_{H_1^g-H_2^g} \end{bmatrix} \begin{bmatrix} Q_0^c \\ Q_1^c \\ H_0^c \\ H_1^c \\ H_2^c \end{bmatrix} = \begin{bmatrix} H_{known} - H_0^g \\ C_1 - Q_1^g + Q_0^g \\ (P_{atm}/\rho g) - H_2^g \\ R(Q_0^g, H_0^g - H_1^g) \\ R(Q_1^g, H_1^g - H_2^g) \end{bmatrix}.$$

The system of equations is then solved for the corrections, \mathbf{Q}^c and \mathbf{H}^c , these are then used to update the guess $\mathbf{Q}^g \rightarrow \mathbf{Q}^g + \mathbf{Q}^c$ and $\mathbf{H}^g \rightarrow \mathbf{H}^g + \mathbf{H}^c$ and then process is repeated until the corrections are sufficiently small. Calculating the above example with two resistances being perfectly smooth pipes of lengths $L_0 = 100\text{m}$, $L_1 = 200\text{m}$ and diameters $D_0 = D_1 = 50\text{mm}$, with $C_1 = 0$, we obtain the solution $\mathbf{Q} = [0.00239623, 0.00239623]^T$ and $\mathbf{H} = [20, 16.77845838, 10.33537514]^T$. For such a simple example the system of equations is equivalent to the finite-difference form however for more complex networks, with multiple branches, the discrete calculus formulation allows the continuity condition to be satisfied without adding an additional constraint.

4.2 Resistances

Each particular component in the network will have a different resistance term and different models of the same component may also produce different resistances. The typical resistance term for pipes is based on the Darcy-Weisbach equation and is shown in (15). We will now outline some other resistance terms for common components.

4.2.1 Valves

A valve has a discharge relationship defined by

$$Q = C_d A_v \sqrt{2g\Delta H}, \quad (21)$$

where C_d is the discharge coefficient and A_v is the valve opening area. Therefore

$$\Delta H = \frac{Q^2}{2g(C_d A_v)^2} = \frac{kQ^2}{2gA^2} \implies \frac{kQ^2}{2A} - gA\Delta H = 0, \quad (22)$$

where A is the cross-sectional area and k is the loss coefficient which varies depending upon the percentage opening of the valve. This means that the resistance term for valves may be written as

$$R_j = -\frac{k_j Q_j |Q_j|}{2A_j} + gA_j \Delta H_j. \quad (23)$$

Alternatively we could write this resistance as

$$\boxed{R_j = -\frac{Q_j |Q_j|}{2A_j} + k_j^{-1} g A_j \Delta H_j}, \quad (24)$$

but we must remember that in this form the transient coefficient in the matrix \mathbf{B} is given by

$$\boxed{B_{j,j} = k_j^{-1}}. \quad (25)$$

The loss coefficient k may be determined from the discharge coefficient and the valve opening area using the relation

$$k = \frac{A^2}{(C_d A_v)^2} \implies k^{-1} = \frac{(C_d A_v)^2}{A^2} = k^{-1}(\tau), \quad (26)$$

where τ is the valve opening ratio. Writing the resistance function in the form (24), using k^{-1} , is more numerically stable as when the valve is fully closed, $A_v = 0$, $k \rightarrow \infty$ but $k^{-1} = 0$.

For example suppose we have a single valve resistance element with $H_0 = 20$, $H_1 = P_{atm}/\rho g$ with $k = 7$ and diameter $D = 50\text{mm}$. Then

$$\begin{aligned} Q &= \sqrt{\frac{2gA^2\Delta H}{k}} = \pi D^2 \sqrt{\frac{g(H_0 - H_1)}{8k}} \\ &= \pi \cdot 0.05^2 \cdot \sqrt{\frac{9.80665 \cdot (20 - (\frac{101325}{997.0 \cdot 9.80665}))}{56}} \approx 0.010203. \end{aligned} \quad (27)$$

This solution agrees with the solution found from iteratively solving the linear system of equations.

TODO inverse pressure loss coefficient k^{-1} vs valve open percentage ϕ in a table.

4.2.2 Pumps

In order to define the resistance term for a given pump we must know the relationship between the flow rate Q and the pumping head ΔH . It is necessary to know this relationship for both positive and negative heads and forward and reverse flow, i.e. in all four quadrants of the $(Q, \Delta H)$ diagram. Typically data for a given pump is only available in the first quadrant where

$Q > 0$ and $\Delta H > 0$ so it often necessary to extrapolate the data or use four-quadrant data from a similar pump.

The discharge of a pump Q is a function of the rotational speed N and the pumping head ΔH . The rotational speed of a pump during power failure is dependent upon the net torque T and the combined moment of inertia of the rotating parts of the pump and the liquid entrained in the impeller. The values of these four variables at the best efficiency point are known as the rated conditions, denoted by a subscript R . Using the rated conditions as a reference we may define the non-dimensional variables

$$q = \frac{Q}{Q_R}, \quad \Delta h = \frac{\Delta H}{\Delta H_R}, \quad n = \frac{N}{N_R} \quad \text{and} \quad \tau = \frac{T}{T_R}.$$

During normal pumping $q, \Delta h, n$ and τ are all positive, when one or more of these variables becomes negative the pump is in an abnormal operating zone. **TODO Table of different zones + diagram - use Chaudry p119-120.**

For pumps with similar geometry and flow profiles

$$\frac{\Delta H}{N^2 D^2} = \text{Constant} \quad \text{and} \quad \frac{N}{Q D^3} = \text{Constant},$$

where D is the diameter of the pump impeller. The impeller diameter D is constant for a particular pump and so may be included in the constants therefore we may define the non-dimensional constants.

$$\frac{\Delta h}{n^2} = \text{Constant} \quad \text{and} \quad \frac{n}{q} = \text{Constant}.$$

Let

$$F_h = \frac{\Delta h}{n^2 + q^2}, \quad F_\tau = \frac{\tau}{n^2 + q^2} \quad \text{and} \quad \theta = \arctan\left(\frac{n}{q}\right),$$

then we may define four-quadrant characteristic curves for the head and torque for a particular pump. These curves define the functions $F_h(\theta)$ and $F_\tau(\theta)$ and are usually approximated using tabulated values given at equal intervals of θ **TODO appendix containing example data table or reference table in Chaudry p523-524.**

TODO discuss specific speed and how it can be used to describe similar pumps.

So since $\Delta h = (n^2 + q^2) F_h(\theta)$ it follows that

$$\Delta H = -\Delta H_R (n^2 + q^2) F_h(\theta) \implies gA\Delta H_R (n^2 + q^2) F_h(\theta) + gA\Delta H = 0, \quad (28)$$

therefore the resistance term for a pump defined by a four-quadrant characteristic curve is given by

$$R_j = gA_j \left[(\Delta H_R)_j (n_j^2 + q_j^2) F_h(\theta_j) + \Delta H_j \right], \quad (29)$$

where $n_j = N_j/N_R$, $q_j = Q_j/Q_R$ and

$$\theta = \arctan\left(\frac{n_j}{q_j}\right) = \text{atan2}(n_j, q_j),$$

where if $\theta < 0$ then $\theta \rightarrow \theta + 2\pi$.

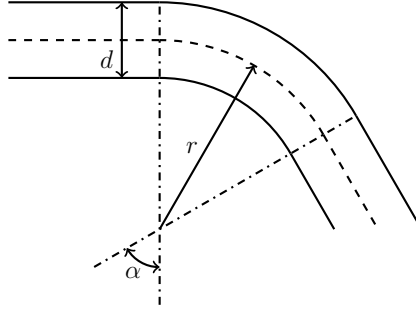


Figure 5: A bend in a pipe of diameter d with a bend radius r and bend angle α .

4.2.3 Bends

Flow in curved pipes may be significantly impeded due to the effects of surface friction, secondary flow and flow separation. The resistance in a pipe bend may be modelled in a similar way to a valve but with the loss coefficient modified to account for the bend geometry and flow conditions. For a bend in a pipe j of diameter d_j , bend angle $0 < \alpha_j < \pi$ and bend radius r_j , as shown in figure 5, the resistance may be defined by as

$$R_j = -\frac{k_j Q_j |Q_j|}{2A_j} + gA_j \Delta H_j \quad \text{where} \quad (30)$$

$$k_j = f_j \alpha_j \frac{r_j}{d_j} + (0.1 + 2.4 f_j) \sin\left(\frac{\alpha_j}{2}\right) + \frac{6.6 f_j \left[\sqrt{\sin\left(\frac{\alpha_j}{2}\right)} + \sin\left(\frac{\alpha_j}{2}\right) \right]}{\left(\frac{r_j}{d_j}\right)^{\frac{4\alpha_j}{\pi}}}.$$

Here $f_j = f(Q_j)$ is the friction factor which may also be a function of the pipe roughness. This empirical equation, see [4], is one of a number of possible methods for determining loss coefficients for pipe bends; different methods may be preferred in certain situations. For equation (30) to be valid it is required that $r_j/d_j \geq 0.5$.

4.2.4 Size changes

Flow through contractions/expansions cause the flow to accelerate/decelerate so it is important to consider size changes when pipes of different diameter are connected. Contractions and expansions are really two sides of the same coin, a contraction with flow in the opposite direction is an expansion just as an expansion with reverse flow is a contraction. Lets consider contractions and expansions separately before their action with reverse flow and combining them into a single component.

4.2.4.1 Contraction

Let us consider a contraction where the flow is from inlet node i to outlet node k , as shown in figure 6. The head loss due to the contraction [4] is given by

$$H_L = \frac{K_c V_i^2}{2g}, \quad (31)$$

where the contraction loss coefficient is

$$K_c = 0.0696 (1 - \beta^5) \lambda^2 + (\lambda - 1)^2$$

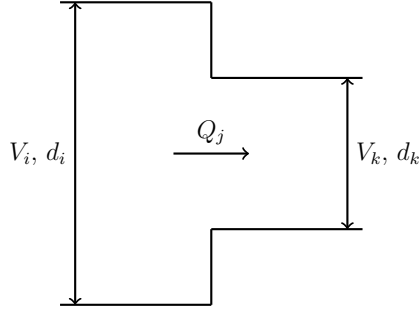


Figure 6: An abrupt contraction where positive flow is from i to k . The size ratio is $\beta = d_k/d_i$.

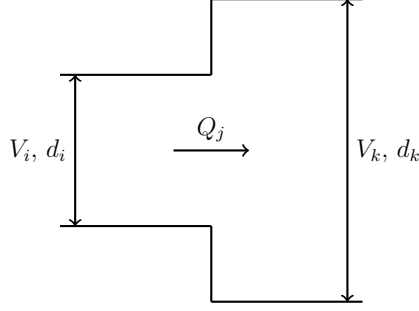


Figure 7: An abrupt expansion where positive flow is from i to k . The size ratio is $\hat{\beta} = d_i/d_k$.

and

$$\lambda = 1 + 0.622 (1 - 0.215\beta^2 - 0.785\beta^5).$$

Here $\beta = d_k/d_i$ is the ratio of the smaller diameter to the larger diameter and $V_i = Q_j/A_i$ is the inlet flow velocity.

4.2.4.2 Expansion

For an expansion where the flow is from inlet node i to outlet node k , as shown in figure 7 the head loss due to the expansion [4] is given by

$$H_L = \frac{\hat{K}_e V_i^2}{2g}, \quad (32)$$

where

$$\hat{K}_e = (1 - \hat{\beta}^2)^2$$

and $\hat{\beta} = d_i/d_k$.

4.2.4.3 Reverse flow

For positive flows, $Q_j > 0$, if $\beta < 1$ or $\hat{\beta} > 1$ then we have a contraction but if $\beta > 1$ or $\hat{\beta} < 1$ then we have an expansion. For negative flows, $Q_j < 0$, the situation is reversed if $\beta < 1$ or $\hat{\beta} > 1$ then we have an expansion but if $\beta > 1$ or $\hat{\beta} < 1$ then we have a contraction.

Reverse flow in a contraction, see figure 6, is actually an expansion with a head loss given by

$$H_L = \frac{K_e V_k^2}{2g}, \quad (33)$$

where

$$K_e = (1 - \beta^2)^2.$$

Similarly reverse flow in an expansion, see figure 7, is actually a contraction with a head loss given by

$$H_L = \frac{\hat{K}_c V_k^2}{2g}, \quad (34)$$

where

$$\hat{K}_c = 0.0696 (1 - \hat{\beta}^5) \hat{\lambda}^2 + (\hat{\lambda} - 1)^2$$

and

$$\hat{\lambda} = 1 + 0.622 (1 - 0.215 \hat{\beta}^2 - 0.785 \hat{\beta}^5).$$

4.2.4.4 Resistance for abrupt size changes

These four different scenarios for forward/reverse flow and $\beta = \hat{\beta}^{-1}$ greater/less than 1 means the resistance term for an abrupt size changes must be split into four parts.

$$\boxed{R_j = -\frac{KQ_j^2}{2A} + gA\Delta H_j, \quad \text{where}$$

$$A = \begin{cases} A_i & \text{if } Q_j > 0, \\ A_k & \text{if } Q_j < 0, \end{cases}$$

$$K = \begin{cases} K_c, & \text{if } Q_j > 0, \beta < 1, \\ \hat{K}_e, & \text{if } Q_j > 0, \beta > 1, \\ K_e, & \text{if } Q_j < 0, \beta < 1, \\ \hat{K}_c, & \text{if } Q_j < 0, \beta > 1. \end{cases}$$

4.2.5 Generic resistance

Many components may be represented by a generic resistance of the form

$$\Delta H = A + BQ^n + CQ^m, \quad (35)$$

where A , B , C , n , and m are constants. In order to preserve the directionality of the resistance, when the flow is reversed, we must replace Q^n and Q^m with $Q|Q|^{n-1}$ and $Q|Q|^{m-1}$, respectively. The resistance function for such a generic resistance may be written as

$$\boxed{R_j = -gA_j [A + BQ_j|Q_j|^{n-1} + CQ_j|Q_j|^{m-1}] + gA_j\Delta H_j.} \quad (36)$$

This form of resistance is fairly general and includes many common components when the coefficients and exponents are chosen appropriately.

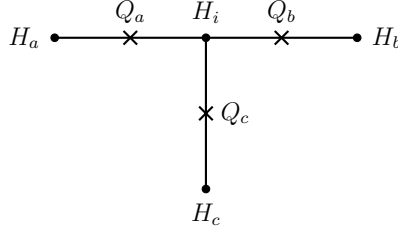


Figure 8: A T-junction is a three way pipe junction with a side branch c connected at 90° to the main pipe between a and b .

4.2.6 Open pipes

Open pipes are pipes that are open to the atmosphere at one end. The discharge relationship of an open pipe is given by

$$\Delta H = \frac{KQ|Q|}{2gA^2}, \quad (37)$$

such that the resistance function is given by

$$R_j = -\frac{KQ_j|Q_j|}{2A_j} + gA_j\Delta H. \quad (38)$$

Here the loss coefficient K is a constant that depends on the geometry of the pipe at the exit. See [4] p.131 for details of different K for various geometries.

4.2.7 Sprinklers

4.2.8 Flow controllers

4.2.9 T junctions

Consider the T junction shown in figure 8. The main pipe between a and b has a branch c connected at 90° . The flow rates through each part of the junction, Q_a , Q_b and Q_c , are considered positive towards the internal junction node i .

The resistance function for each edge in the T junction is dependent upon the flow rates in the other two edges. Generally the resistance function for each edge is of the form

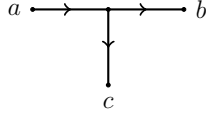
$$R_j = -\frac{K_jQ_j|Q_j|}{2A_j} + gA_j\Delta H_j, \quad j \in \{a, b, c\}, \quad (39)$$

where the loss coefficients K_a , K_b and K_c are functions of the flow rates Q_a , Q_b and Q_c . The form of the loss coefficients depends on the flow regime in the junction. The flow regime is determined by the sign of the flow rate in each edge [4].

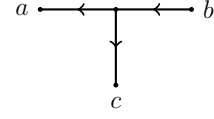
4.2.9.1 Diverging flow on main branch

Flow diverging from the main branch is shown in figure 9 where 9a shows the flow diverging from node a and 9b shows the flow diverging from node b . The loss coefficient for the main branch is given by

$$K_{\text{main,div}}(Q_{\text{in}}, Q_{\text{out}}) = 0.62 - 0.98\gamma + 0.36\gamma^2 + 0.03\gamma^{-6}, \quad (40)$$



(a) Flow diverging from node a has flow rates $Q_a > 0$, $Q_b < 0$ and $Q_c < 0$.



(b) Flow diverging from node b has flow rates $Q_a < 0$, $Q_b > 0$ and $Q_c < 0$.

Figure 9: Diverging flow on the main branch of a T junction.

where $\gamma = Q_{\text{in}}/Q_{\text{out}}$. For flow diverging from node a we have $Q_{\text{in}} = Q_a$ and $Q_{\text{out}} = Q_b$ and for flow diverging from node b we have $Q_{\text{in}} = Q_b$ and $Q_{\text{out}} = Q_a$.

For the side branch the loss coefficient is given by

$$K_{\text{side,div}}(Q_{\text{in}}, Q_c) = (0.81 - 1.13\beta + \beta^2) \hat{d}^4 + 1.12\hat{d} - 1.08\hat{d}^3 + K_*, \quad (41)$$

where $\beta = Q_{\text{in}}/Q_c$ and $\hat{d} = d_c/d_{\text{main}}$. The constant K_* is given by

$$K_* = 0.57 - 1.07r_c^{\frac{1}{2}} - 2.13r_c + 8.24r_c^{\frac{3}{2}} - 8.48r_c^2 + 2.90r_c^{\frac{5}{2}}, \quad (42)$$

where $r_c = r/d_c$. Here r is the radius of curvature of the junction and d_c is the diameter of the side branch. For flow diverging from node a we have $Q_{\text{in}} = Q_a$ and for flow diverging from node b we have $Q_{\text{in}} = Q_b$.

4.2.9.2 Converging flow on main branch

Flow converging to the main branch is shown in figure 10 where 10a shows flow converging to node a and 10b shows flow converging to node b . The loss coefficient for the main branch is given by

$$K_{\text{main,conv}}(Q_{\text{in}}, Q_{\text{out}}) = \gamma^2 - 0.95 - 2C_{xC}(\gamma - 1) - 2C_M\gamma(\gamma - 1), \quad (43)$$

where $\gamma = Q_{\text{in}}/Q_{\text{out}}$ and C_{xC} and C_M are given by

$$C_M = 0.23 + 1.46r_c - 2.75r_c^2 + 1.65r_c^3, \quad (44)$$

$$C_{xC} = 0.08 + 0.56r_c - 1.75r_c^2 + 1.83r_c^3. \quad (45)$$

For flow converging to node a we have $Q_{\text{in}} = Q_b$ and $Q_{\text{out}} = Q_a$ and for flow converging to node b we have $Q_{\text{in}} = Q_a$ and $Q_{\text{out}} = Q_b$.

For the side branch the loss coefficient is given by

$$K_{\text{side,conv}}(Q_{\text{in}}, Q_c) = 2C_{yC} - 1 + 2[(C_{xC} - 1) + (2 - C_{xC} - C_M)\beta - 0.46\beta^2] \hat{d}^4, \quad (46)$$

where $\beta = Q_{\text{in}}/Q_c$ and C_{yC} is given by

$$C_{yC} = 1 - 0.25\hat{d}^{1.3} - [0.11r_c - 0.65r_c^2 + 0.83r_c^3] \hat{d}^2. \quad (47)$$

4.2.9.3 Converging or diverging flow from the side branch

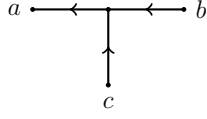
Flow on the side branch is shown in figure 11 where 11a shows flow converging to the side branch and 11b shows flow diverging from the side branch. For flow converging to the side branch we have the loss coefficient

$$K_{\text{convto}}(Q_{\text{in}}, Q_{\text{out}}) = (0.81 - 1.16\sqrt{r_c} + 0.5r_c) \gamma^2 - (0.95 - 1.65r_c) \gamma + 1.34 - 1.69r_c, \quad (48)$$

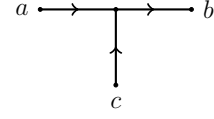
where $\gamma = Q_{\text{in}}/Q_{\text{out}} = Q_a/Q_c$. For flow diverging from the side branch we have the loss coefficient

$$K_{\text{divfrom}}(Q_{\text{in}}, Q_{\text{out}}) = 0.59\gamma^2 + (1.18 - 1.84\sqrt{r_c} + 1.16r_c) \gamma - 0.68 + 1.04\sqrt{r_c} - 1.16r_c, \quad (49)$$

where $\gamma = Q_{\text{in}}/Q_{\text{out}} = Q_c/Q_a$.

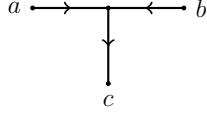


(a) Flow converging to node a has flow rates $Q_a < 0$, $Q_b > 0$ and $Q_c > 0$.

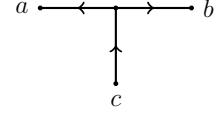


(b) Flow converging to node b has flow rates $Q_a > 0$, $Q_b < 0$ and $Q_c > 0$.

Figure 10: Converging flow on the main branch of a T junction.



(a) Flow converging to node c has flow rates $Q_a > 0$, $Q_b > 0$ and $Q_c < 0$.



(b) Flow diverging from node c has flow rates $Q_a < 0$, $Q_b < 0$ and $Q_c > 0$.

Figure 11: Converging/diverging flow to the side branch of a T junction.

4.2.9.4 Loss coefficients

For each of the six possible flow regimes the loss coefficient in each edge, K_a , K_b and K_c , takes on a different value. The loss coefficient K_a is given by

$$K_a(Q_a, Q_b, Q_c) = \begin{cases} 0 & \text{if } Q_a > 0, Q_b < 0, Q_c < 0, \\ K_{\text{main,div}}(Q_b, Q_a) & \text{if } Q_a < 0, Q_b > 0, Q_c < 0, \\ 0 & \text{if } Q_a < 0, Q_b > 0, Q_c > 0, \\ K_{\text{main,conv}}(Q_a, Q_b) & \text{if } Q_a > 0, Q_b < 0, Q_c > 0, \\ K_{\text{convto}}(Q_a, Q_c) & \text{if } Q_a > 0, Q_b > 0, Q_c < 0, \\ K_{\text{divfrom}}(Q_c, Q_a) & \text{if } Q_a < 0, Q_b < 0, Q_c > 0, \end{cases} \quad (50)$$

K_b is given by

$$K_b(Q_a, Q_b, Q_c) = \begin{cases} K_{\text{main,div}}(Q_a, Q_b) & \text{if } Q_a > 0, Q_b < 0, Q_c < 0, \\ 0 & \text{if } Q_a < 0, Q_b > 0, Q_c < 0, \\ K_{\text{main,conv}}(Q_b, Q_a) & \text{if } Q_a < 0, Q_b > 0, Q_c > 0, \\ 0 & \text{if } Q_a > 0, Q_b < 0, Q_c > 0, \\ K_{\text{convto}}(Q_a, Q_c) & \text{if } Q_a > 0, Q_b > 0, Q_c < 0, \\ K_{\text{divfrom}}(Q_c, Q_a) & \text{if } Q_a < 0, Q_b < 0, Q_c > 0, \end{cases} \quad (51)$$

and K_c is given by

$$K_c(Q_a, Q_b, Q_c) = \begin{cases} K_{\text{side,div}}(Q_a, Q_c) & \text{if } Q_a > 0, Q_b < 0, Q_c < 0, \\ K_{\text{side,div}}(Q_b, Q_c) & \text{if } Q_a < 0, Q_b > 0, Q_c < 0, \\ K_{\text{side,conv}}(Q_b, Q_c) & \text{if } Q_a < 0, Q_b > 0, Q_c > 0, \\ K_{\text{side,conv}}(Q_a, Q_c) & \text{if } Q_a > 0, Q_b < 0, Q_c > 0, \\ 0 & \text{if } Q_a > 0, Q_b > 0, Q_c < 0, \\ 0 & \text{if } Q_a < 0, Q_b < 0, Q_c > 0. \end{cases} \quad (52)$$

Substituting these loss coefficients into the function (39) we are able to calculate the resistance of the T junction for any flow regime.

4.2.9.5 Jacobian

Special care needs to be taken with the Jacobian matrix for the T junction since off-diagonal elements are non-zero. The Jacobian matrix, for the T junction shown in figure 8, is given by

$$\mathbf{J} = \begin{bmatrix} \frac{\partial R_a}{\partial Q_a} & \frac{\partial R_a}{\partial Q_b} & \frac{\partial R_a}{\partial Q_c} \\ \frac{\partial R_b}{\partial Q_a} & \frac{\partial R_b}{\partial Q_b} & \frac{\partial R_b}{\partial Q_c} \\ \frac{\partial R_c}{\partial Q_a} & \frac{\partial R_c}{\partial Q_b} & \frac{\partial R_c}{\partial Q_c} \end{bmatrix}. \quad (53)$$

The off-diagonal elements are due to the dependence of the branch resistances (39) on the flow rates in the other branches. For example, the resistance of the main branch, R_a , depends on the flow rate in the side branch, Q_c , and the flow rate in the other part of the main branch Q_b not just on Q_a .

4.2.10 Cross junctions

The solution of the steady state problem for flow networks has been thoroughly explored and many other solution methods exist. It is illustrative to see how the water-hammer equations collapse to the steady state problem in the absence of time variation but we must continue towards our original goal, to devise a method for calculating fluid transients in pipe networks.

Appendix on calculating the laminar initial guess.

5 Transient solver

The full transient equations are more difficult to solve than the steady state problem as the resistance term in the momentum is now time dependent and we must retain all the terms in the discrete continuity equation (11). The equations must be stepped forward in time using a scheme which is both accurate and numerically stable.

5.1 Theta method

The θ -method is a generalisation of the Crank-Nicolson method which allows the scheme to be made more or less implicit. In general the scheme takes the form

$$\frac{u_i^{n+1} - u_i^n}{\Delta t} = \theta F_i^{n+1} \left(u, x, t, \frac{\partial u}{\partial x}, \dots \right) + (1 - \theta) F_i^n \left(u, x, t, \frac{\partial u}{\partial x}, \dots \right), \quad (54)$$

where $\theta \in [0, 1]$. If $\theta = 0$ then the scheme is the explicit forward Euler method, if $\theta = 1$ then it is the fully implicit backward Euler method and if $\theta = 1/2$ then it is the Crank-Nicolson method. Usually the Crank-Nicolson scheme is preferred to both the forward and backward Euler methods as it is second order accurate whereas they are both only first order accurate. However the approximate solutions found using the Crank-Nicolson scheme sometimes include spurious oscillations so it may be preferable to use the less accurate implicit scheme in certain situations.

Using the θ -method on the continuity equation (11) we have

$$\mathbf{K}^T (\theta \mathbf{Q}^{n+1} + (1 - \theta) \mathbf{Q}^n) + \mathbf{D} \left(\frac{\mathbf{H}^{n+1} - \mathbf{H}^n}{\Delta t} \right) = \mathbf{C}. \quad (55)$$

Then for the momentum equation (12) we have

$$\mathbf{B} \frac{\mathbf{Q}^{n+1} - \mathbf{Q}^n}{\Delta t} = \mathbf{R} (\theta \mathbf{Q}^{n+1} + (1 - \theta) \mathbf{Q}^n, \theta \mathbf{K} \mathbf{H}^{n+1} + (1 - \theta) \mathbf{K} \mathbf{H}^n). \quad (56)$$

At time-step n the pipe flow rates \mathbf{Q}^n and nodal heads \mathbf{H}^n are known and we would like to estimate the solution at the subsequent time-step $n + 1$. The steady state solution \mathbf{Q}^0 and \mathbf{H}^0 is used to initialise the scheme.

5.1.1 Newton iteration

Since the solution is known at time-step n only the solution at time-step $n + 1$ needs to be split into a known/guessed part and a correction i.e. $\mathbf{Q}^{n+1} = \mathbf{Q}^{n+1,g} + \mathbf{Q}^{n+1,c}$ and $\mathbf{H}^{n+1} = \mathbf{H}^{n+1,g} + \mathbf{H}^{n+1,c}$. The resistance term is then approximated as

$$\mathbf{R}(\mathbf{Q}, \mathbf{KH}) \approx \mathbf{R}(\bar{\mathbf{Q}}, \mathbf{K}\bar{\mathbf{H}}) + \left. \frac{\partial \mathbf{R}}{\partial \mathbf{Q}} \right|_{\bar{\mathbf{Q}}} \theta \mathbf{Q}^{n+1,c} + \left. \frac{\partial \mathbf{R}}{\partial \mathbf{KH}} \right|_{\mathbf{K}\bar{\mathbf{H}}} \theta \mathbf{KH}^{n+1,c},$$

where $\bar{\mathbf{Q}} = \theta \mathbf{Q}^{n+1,g} + (1 - \theta) \mathbf{Q}^n$, $\bar{\mathbf{H}} = \theta \mathbf{H}^{n+1,g} + (1 - \theta) \mathbf{H}^n$. Splitting into a known part and correction at the next time step the continuity equation is given by

$$\theta \mathbf{K}^T \mathbf{Q}^{n+1,c} + \frac{1}{\Delta t} \mathbf{D} \mathbf{H}^{n+1,c} = \mathbf{C} - \mathbf{K}^T \bar{\mathbf{Q}} - \frac{1}{\Delta t} \mathbf{D} (\mathbf{H}^{n+1,g} - \mathbf{H}^n), \quad (57)$$

and the momentum equation is given by

$$\left(\frac{1}{\Delta t} \mathbf{B} - \theta \left. \frac{\partial \mathbf{R}}{\partial \mathbf{Q}} \right|_{\bar{\mathbf{Q}}} \right) \mathbf{Q}^{n+1,c} - \theta \left. \frac{\partial \mathbf{R}}{\partial \mathbf{KH}} \right|_{\mathbf{K}\bar{\mathbf{H}}} \mathbf{KH}^{n+1,c} = \mathbf{R}(\bar{\mathbf{Q}}, \mathbf{K}\bar{\mathbf{H}}) - \frac{1}{\Delta t} \mathbf{B} (\mathbf{Q}^{n+1,g} - \mathbf{Q}^n). \quad (58)$$

Equations (57) and (58) must be solved together as the single system of equations

$$\begin{bmatrix} \theta \mathbf{K}^T & \frac{1}{\Delta t} \mathbf{D} \\ \frac{1}{\Delta t} \mathbf{B} - \theta \bar{\mathbf{J}} & -\theta \bar{\mathbf{G}} \end{bmatrix} \begin{bmatrix} \mathbf{Q}^{n+1,c} \\ \mathbf{H}^{n+1,c} \end{bmatrix} = \begin{bmatrix} \mathbf{C} - \mathbf{K}^T \bar{\mathbf{Q}} - \frac{1}{\Delta t} \mathbf{D} (\mathbf{H}^{n+1,g} - \mathbf{H}^n) \\ \mathbf{R}(\bar{\mathbf{Q}}, \mathbf{K}\bar{\mathbf{H}}) - \frac{1}{\Delta t} \mathbf{B} (\mathbf{Q}^{n+1,g} - \mathbf{Q}^n) \end{bmatrix}, \quad (59)$$

where $\bar{\mathbf{J}} = \left. \frac{\partial \mathbf{R}}{\partial \mathbf{Q}} \right|_{\bar{\mathbf{Q}}}$ and $\bar{\mathbf{G}} = \left. \frac{\partial \mathbf{R}}{\partial \mathbf{KH}} \right|_{\mathbf{K}\bar{\mathbf{H}}} \mathbf{K}$. In a similar manner to the steady solver the system of equations (59) would be solved for the corrections, $\mathbf{Q}^{n+1,c}$ and $\mathbf{H}^{n+1,c}$, these would then be used to update the guess $\mathbf{Q}^{n+1,g} \rightarrow \mathbf{Q}^{n+1,g} + \mathbf{Q}^{n+1,c}$ and $\mathbf{H}^{n+1,g} \rightarrow \mathbf{H}^{n+1,g} + \mathbf{H}^{n+1,c}$ and then the process would be repeated until the corrections are sufficiently small. After updating the simulation time $t \rightarrow t + \Delta t$, this time-stepping is repeated until the simulation is complete.

5.2 Boundary conditions

Boundary conditions may be applied at the nodes in the system by replacing the continuity equation at that node with an appropriate boundary condition. For example if the head is specified to be H_{known} at a given node i then we have the condition

$$\theta H_i^{n+1,c} = H_{known} - (\theta H_i^{n+1,g} + (1 - \theta) H_i^n), \quad (60)$$

which can be used to replace row i in the system of equations (59). Here the boundary head H_{known} may vary over time. Known flow rates at the nodes are specified using the consumption vector \mathbf{C} . The consumption vector \mathbf{C} in the system of equations may also vary with time.

5.2.1 Surge tanks

A surge tank is an open topped vessel, filled with fluid upto some predetermined height, as shown in figure 12. By damping the pressure response and supplying/storing fluid a surge tank

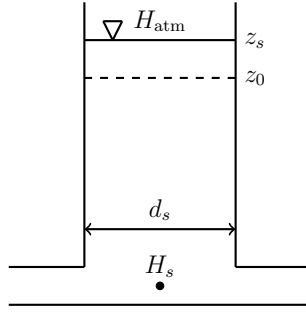


Figure 12: A surge tank boundary at node s with an a fluid level z_s , an initial fluid level z_0 and tank diameter d_s .

is often able to alleviate the ill effects of transient flows in a system [Reference Chaudry p349-350](#). Surge tanks are able to be modelled by modifying the consumption term in the continuity equation at a node thereby modifying the rate at which the nodal head changes with time.

Suppose that a surge tank is placed in the network at node s , at this node we have the continuity equation

$$(\mathbf{K}^T)_s \mathbf{Q} + D_{s,s} \frac{\partial H_s}{\partial t} = C_s, \quad (61)$$

where $(\mathbf{K}^T)_s$ indicates the s th row of the matrix \mathbf{K}^T and $D_{s,s}$ indicates the element of the matrix D at row s and column s . The flow which leaves the node and fills the surge tank may be modelled by choosing an appropriate expression for the consumption term C_s .

The nodal head H_s is equal to the head due to atmospheric pressure plus the fluid column height i.e.

$$H_s = H_{\text{atm}} + z_s. \quad (62)$$

The rate at which fluid flows into the surge tank is

$$A_s \frac{dz_s}{dt},$$

where $A_s = \pi d_s^2/4$ is the cross-sectional area of the tank. If we assume that the fluid in the tank moves as a slug then we may say that

$$A_s \frac{dz_s}{dt} = A_s \frac{\partial z_s}{\partial t} = A_s \frac{\partial H_s}{\partial t}.$$

Therefore, since flow into the tank is equal to flow out of the node ($C_s < 0$), we may rewrite the continuity equation as

$$(\mathbf{K}^T)_s \mathbf{Q} + D_{s,s} \frac{\partial H_s}{\partial t} = -A_s \frac{\partial H_s}{\partial t}.$$

Splitting into a known part and correction at the next time step the continuity equation, at node s , is given by

$$\theta (\mathbf{K}^T)_s \mathbf{Q}^{n+1,c} + \frac{(D_{s,s} + A_s)}{\Delta t} H_s^{n+1,c} = -(\mathbf{K}^T)_s \bar{\mathbf{Q}} - \frac{(D_{s,s} + A_s)}{\Delta t} (H_s^{n+1,g} - H_s^n). \quad (63)$$

The equation (63) then substitutes the s th row of the system of governing equations (59). It may be observed that by adding the surge tank we are effectively modifying the coefficient $D_{s,s}$ such that $D_{s,s} \rightarrow D_{s,s} + A_s$ thereby reducing the rate at which H_s changes with time for a given set of flow rates through the edges connected to node s .

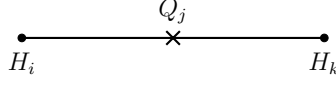


Figure 13: General component j with an average flow rate Q_j and nodal heads H_i and H_k .

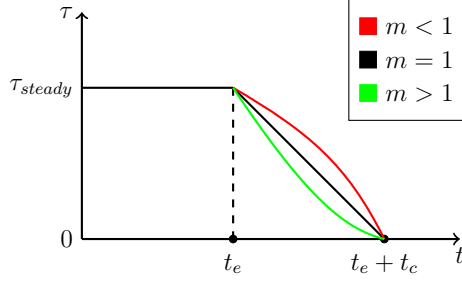


Figure 14: Valve closure. The open ratio $\tau(t)$ goes from τ_{steady} at time t_e to zero at time $t_e + t_c$, where t_c is the time taken for the valve to close. The exponent m modifies the closure profile, as shown by the red and green curves.

5.3 Components

Components other than pipes may be specified in equation (12). For a general component, of the form shown in figure 13, a relationship between the nodal heads and the pipe flow rate must be specified.

5.3.1 Valves

Transient flows in pipe networks may be caused by the closing and opening of valves. In order to describe the transient behaviour of a valves we must define the valve open ratio $\tau(t)$ as a function of time. The valve open percentage is then used to determine the loss coefficient k in the valve resistance equation (24).

5.3.1.1 Closing

The function $\tau(t)$, as shown in figure 14, for a valve closure is given by

$$\tau(t) = \begin{cases} \tau_{steady}, & \text{if } t \leq t_e \\ \tau_{steady} \left(1 - \frac{t-t_e}{t_c}\right)^m, & \text{if } t_e \leq t \leq t_e + t_c \\ 0, & \text{if } t \geq t_e + t_c, \end{cases} \quad (64)$$

where t_e is the event time, t_c is the time taken for the valve to close, τ_{steady} is the steady state valve open percentage. The exponent $m > 0$ modifies the shape of the closure profile with $m = 1$ corresponding to a linear profile.

5.3.1.2 Opening

The function $\tau(t)$, as shown in figure 15, for a valve opening is given by

$$\tau(t) = \begin{cases} \tau_{steady}, & \text{if } t \leq t_e \\ \tau_{steady} + (1 - \tau_{steady}) \left(\frac{t-t_e}{t_c}\right)^m, & \text{if } t_e \leq t \leq t_e + t_c \\ 1, & \text{if } t \geq t_e + t_c, \end{cases} \quad (65)$$

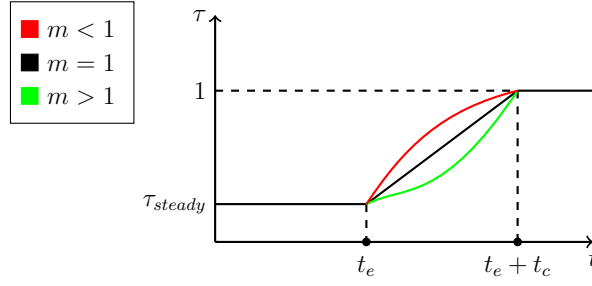


Figure 15: Valve opening. The open ratio $\tau(t)$ goes from τ_{steady} at time t_e to one at time $t_e + t_c$, where t_c is the time taken for the valve to close. The exponent m modifies the closure profile, as shown by the red and green curves.

where t_e is the event time, t_c is the time taken for the valve to close, τ_{steady} is the steady state valve open percentage. The exponent $m > 0$ modifies the shape of the opening profile with $m = 1$ corresponding to a linear profile.

5.3.2 Pumps

The starting and stopping of pumps are a significant cause of transient flows in pumping systems. Starting of a pump is usually controlled by keeping the discharge valve close until the pump reaches the rated speed and then slowly opening the valve. However sometimes pumps are started without using valves by controlling the speed electronically. Similarly pump shutdown may be assisted by valve closure or by external control. Another common source of pump transients is power failure, in this situation the speed of the pump is no longer controlled and is instead determined by the torque on the pump and the flow conditions.

5.3.2.1 Shutdown

The simplest way to shutdown a pump in a controlled manner is a reduction of the rotational speed, at a specified event time t_e , to zero at time $t_e + t_c$, where t_c is stopping time. The rotational speed N as a function of time t , as shown in figure 16, is defined by the piecewise function

$$N(t) = \begin{cases} N_{steady}, & \text{if } t \leq t_e \\ N_{steady} \left(1 - \frac{t-t_e}{t_c}\right)^m, & \text{if } t_e \leq t \leq t_e + t_c \\ 0, & \text{if } t \geq t_e + t_c. \end{cases} \quad (66)$$

The exponent $m > 0$ modifies the shape of the closure profile with $m = 1$ corresponding to a linear profile. Changing the rotational speed alters both n and θ in the pump resistance equation (29) so that the resistance varies with time too.

5.3.2.2 Startup

The time varying function describing the startup of a pump, as shown in figure 17, is very similar to the function describing a shutdown and is given by

$$N(t) = \begin{cases} 0, & \text{if } t \leq t_e \\ N_{\infty} \left(\frac{t-t_e}{t_s}\right)^m, & \text{if } t_e \leq t \leq t_e + t_s \\ N_{\infty}, & \text{if } t \geq t_e + t_s, \end{cases} \quad (67)$$

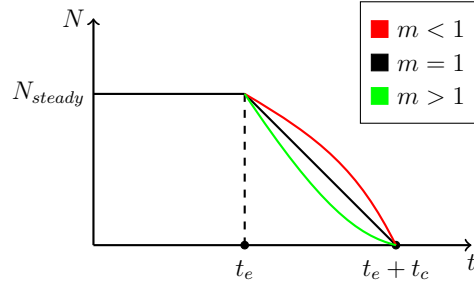


Figure 16: Pump shutdown. The rotational speed $N(t)$ goes from N_{steady} at time t_e to zero at time $t_e + t_c$, where t_c is the time taken for the pump to stop. The exponent m modifies the closure profile, as shown by the red and green curves.

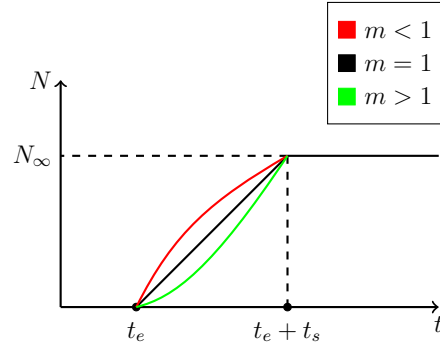


Figure 17: Pump startup. The rotational speed $N(t)$ goes from zero at time t_e to N_∞ at time $t_e + t_s$, where t_s is the time taken for the pump to startup. The exponent m modifies the closure profile, as shown by the red and green curves.

where again t_e is the event time, t_s is startup time and N_∞ is the pump speed once the startup procedure has been completed. The exponent $m > 0$ modifies the shape of the closure profile with $m = 1$ corresponding to a linear profile.

5.3.3 Check / Non-return valves

A check or non-return valve is a valve that allows fluid to flow through it in one direction only. The resistance of a check valve has the same form as for a standard valve, see equation (24), however the inverse loss coefficient k_j^{-1} is given by

$$k_j^{-1} = \begin{cases} 0, & \text{if } \Delta H_j \leq 0, \\ k_0^{-1}, & \text{if } \Delta H_j > 0. \end{cases} \quad (68)$$

This means that the check valve is open when the pressure difference across it is positive and closed when the pressure difference is negative. The value k_0 is the loss coefficient of the valve when it is fully open. For steady flow, the check valve is considered to be open ($\tau = 1$), so the inverse loss coefficient is k_0^{-1} .

5.3.4 Relief devices

5.3.4.1 Safety valves

A safety valve is a valve that opens automatically when the pressure in the system exceeds a certain threshold. The resistance of a safety valve has the same form as for a standard valve,

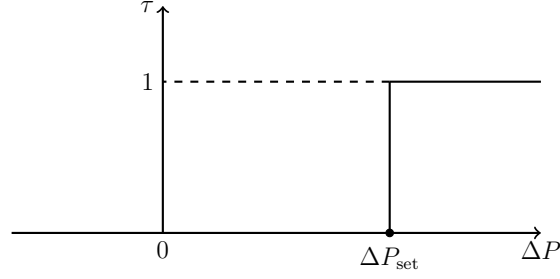


Figure 18: Safety valve open ratio as a function of pressure difference across the valve.

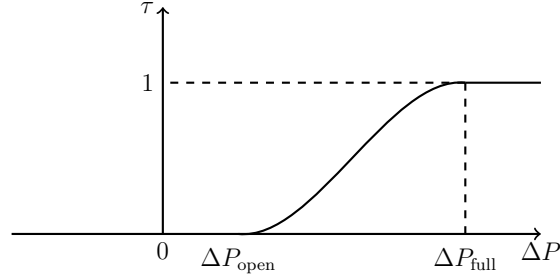


Figure 19: Pressure relief valve open ratio as a function of pressure difference across the valve.

see equation (24), however the inverse loss coefficient k_j^{-1} is given by

$$k_j^{-1} = \begin{cases} 0, & \text{if } \Delta P_j \leq \Delta P_{\text{set}}, \\ k_0^{-1}, & \text{if } \Delta P_j > \Delta P_{\text{set}}, \end{cases} \quad (69)$$

where ΔP_{set} is the set pressure difference at which the valve opens. This means that the safety valve is open when the pressure difference across it is greater than the set pressure difference and closed when the pressure difference is less than the set pressure difference. The value k_0 is the loss coefficient of the valve when it is fully open. For steady flow, the safety valve is considered to be closed ($\tau = 0$), so the inverse loss coefficient is zero.

5.3.4.2 Pressure relief valves

Pressure relief valves are similar to safety valves, but they open more gradually as the pressure in the system increases. The resistance of a pressure relief valve has the same form as for a standard valve, see equation (24). For a pressure relief valve, the open ratio τ is a function of the pressure difference across the valve and is given by

$$\tau = \begin{cases} 0, & \text{if } \Delta P_j \leq \Delta P_{\text{open}}, \\ f(\Delta P_j), & \text{if } \Delta P_{\text{open}} < \Delta P_j < \Delta P_{\text{full}}, \\ 1, & \text{if } \Delta P_j \geq \Delta P_{\text{full}}, \end{cases} \quad (70)$$

where ΔP_{open} is the pressure difference at which the valve starts to open, ΔP_{full} is the pressure difference at which the valve is fully open, and f is a function that describes the profile of the valve opening.

5.3.4.3 Bursting disks

A bursting disk is a thin membrane that ruptures when the pressure in the system exceeds a certain threshold. The resistance of a bursting disk is very similar to a safety valve, where the

inverse loss coefficient is given by

$$k_j^{-1} = \begin{cases} 0, & \text{if } \Delta P_j \leq \Delta P_{\text{burst}}, \\ k_0^{-1}, & \text{if } \Delta P_j > \Delta P_{\text{burst}}, \end{cases} \quad (71)$$

however once the bursting disk has ruptured it remains open. The value k_0 is the loss coefficient of the bursting disk when it is fully open. So if $t > t_{\text{burst}}$, then $k_j^{-1} = k_0^{-1}$. Initially the bursting disk is considered to be closed ($\tau = 0$), so the inverse loss coefficient is zero. If the pressure difference across the bursting disk remains lower than the bursting pressure throughout the entire simulation time, then $k_j^{-1} = 0$ at all times.

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