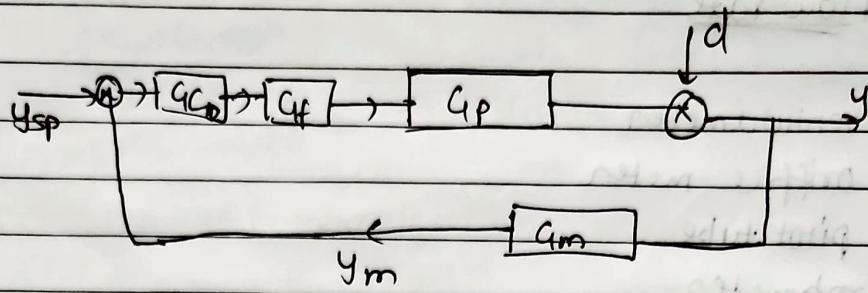


#

Instrumentation:



- 1) Generalized description of measuring instruments
- 2) Measuring instruments : selection and functioning
- 3) Hardware of pneumatic control system.
- 4) control valve : final control element
- 5) symbols for P and ID

#

Temperature:

- 1) thermocouple
- 2) liq filled thermometers.
- 3) Radiation pyrometer
- 4) Resistance thermometer
- 5) Bimetallic

#

pressure:

- ① manometer
- ② Riezo
- ③ Diaphragm
- ④ Bellows
- ⑤ Bourdon Tube

Measurement System Application and Design

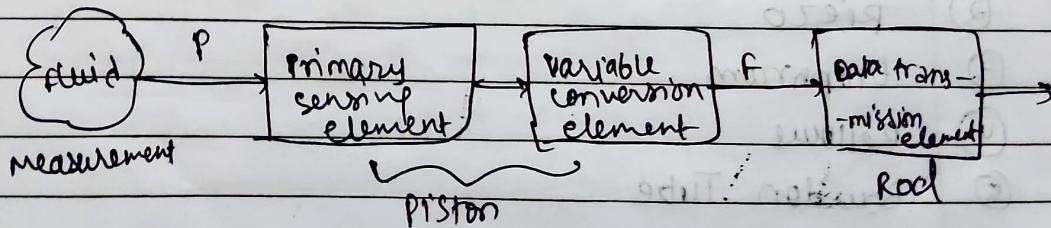
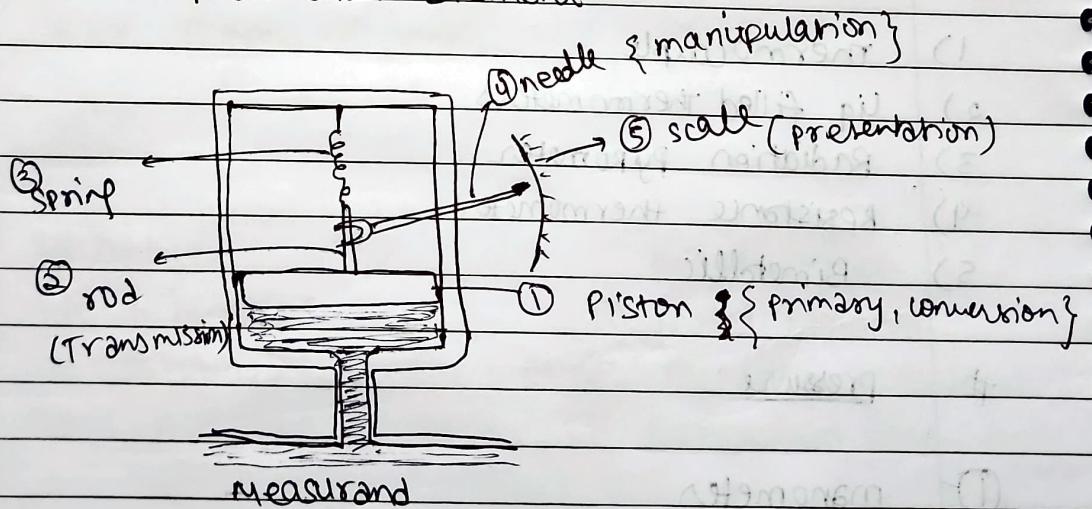
Ernest O. Slobelkin

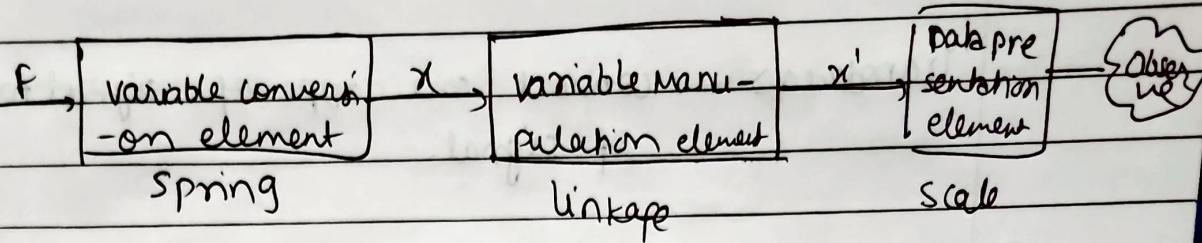
Flow rate:

- 1) venturimeter
- 2) orifice meter
- 3) pitot tube
- 4) rotameter
- 5) voriolli's meter
- 6) ultra-sonic flowmeter

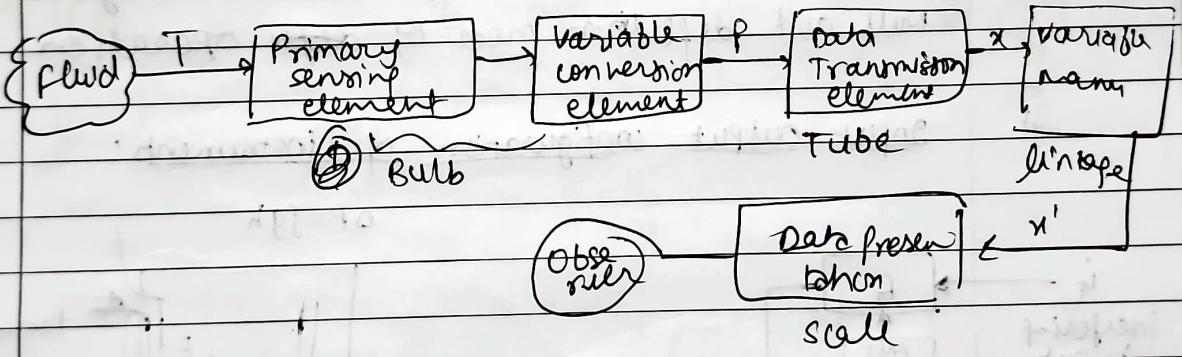
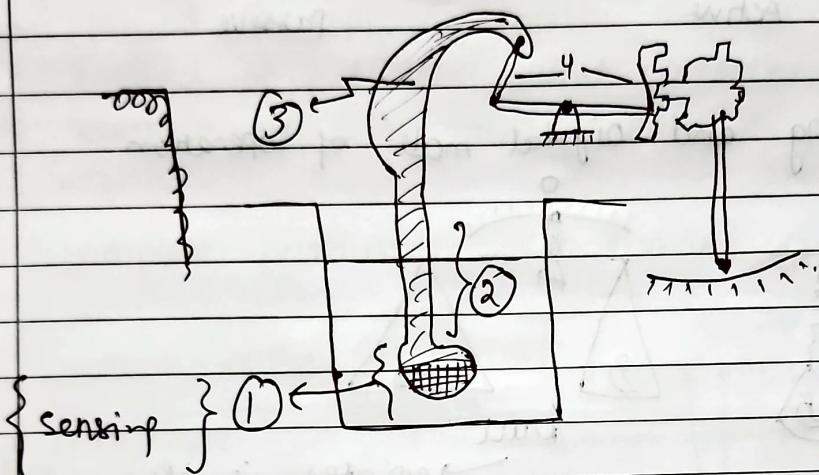
Functional elements:

- ① Primary sensing Element: { extracts energy from the system}
- ② variable conversion Element
- ③ variable manipulation Element
- ④ Data transmission Element
- ⑤ Data presentation Element

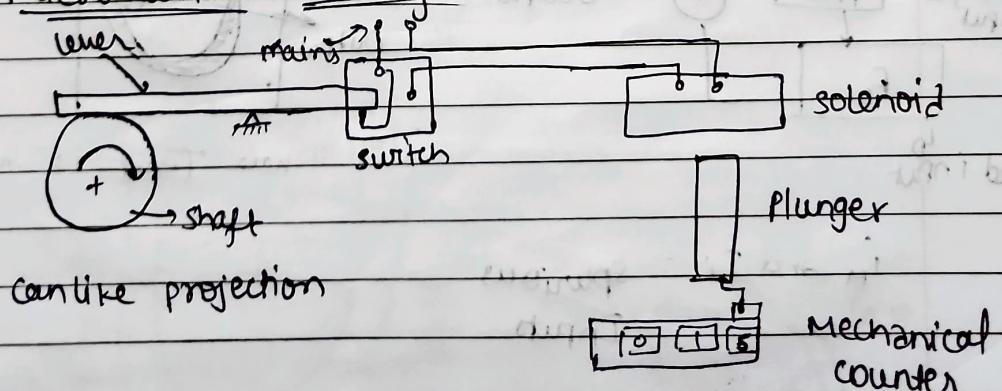




Pressure thermometer:



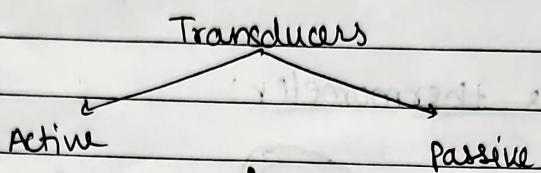
Microswitch sensing arm: {counting the no. of revolutions}



Different function of elements:

Transducers → convert one type of signal to other type of signal.

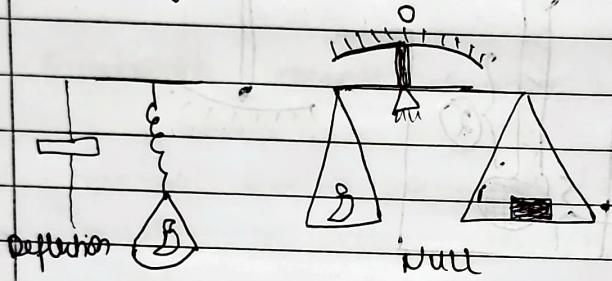
①



②

Analog and digital mode of operation

③



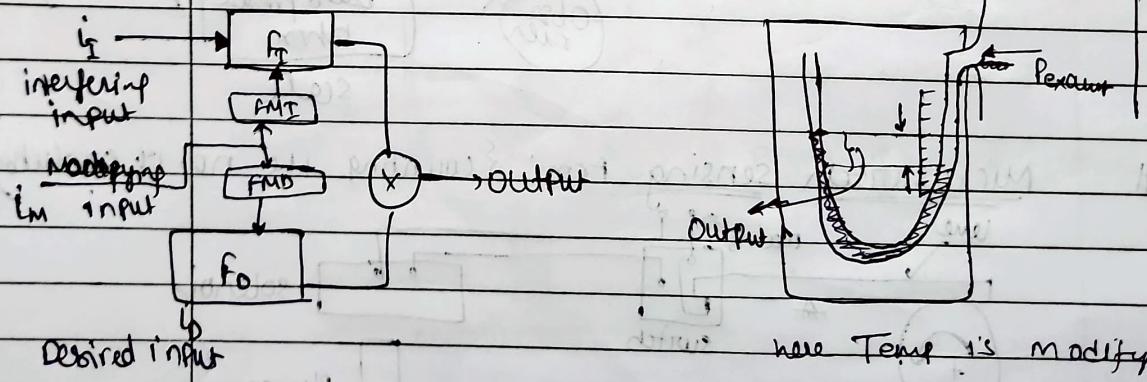
→ pointer is always at null point

Null and deflection mode of ~~def~~ operation.

#

Input output configuration of instruments:

$$\Delta P = \rho g h$$



Desired input

here Temp is modifying input

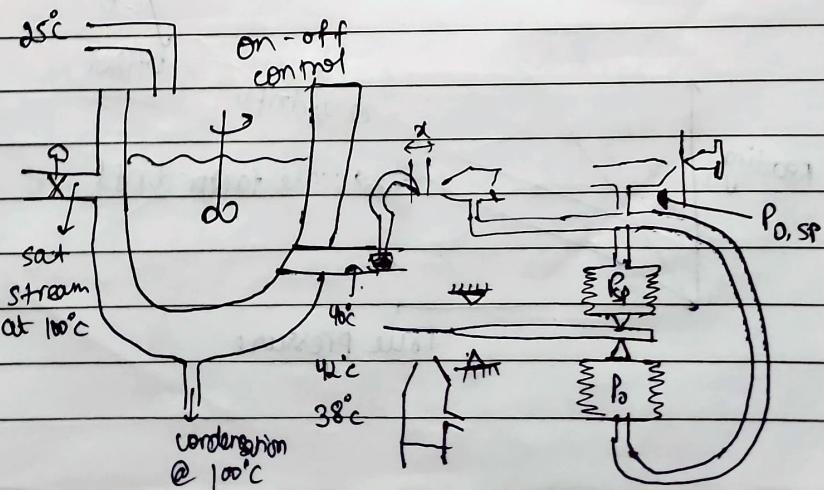
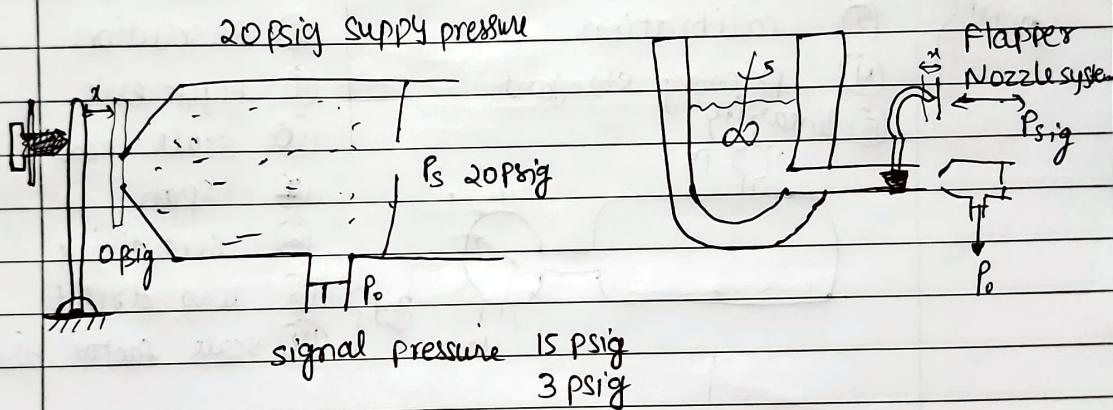
i_m and T_i = spurious inputs

— / —

Methods for correction of interfering and modifying inputs:

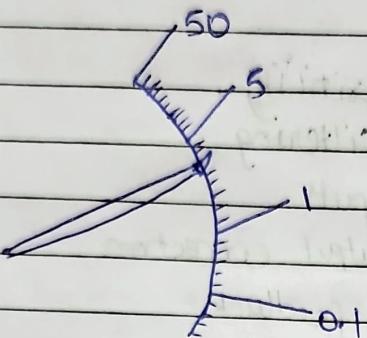
- ① Method of inherent sensitivity.
- ② Method of signal filtering
- ③ Method of opposing inputs
- ④ Method of calculated output correction
- ⑤ Method of high gain feedback.

complete hardware of pneumatic control system.



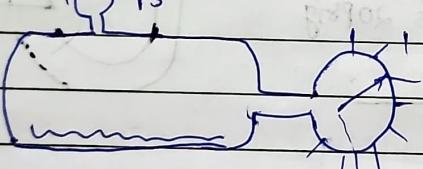
When $P_0 > P_{0,sp}$: $P_{0, \text{controller}} = 15 \text{ psig}$

When $P_0 < P_{0,sp}$ $P_{0, \text{controller}} = 3 \text{ psig}$

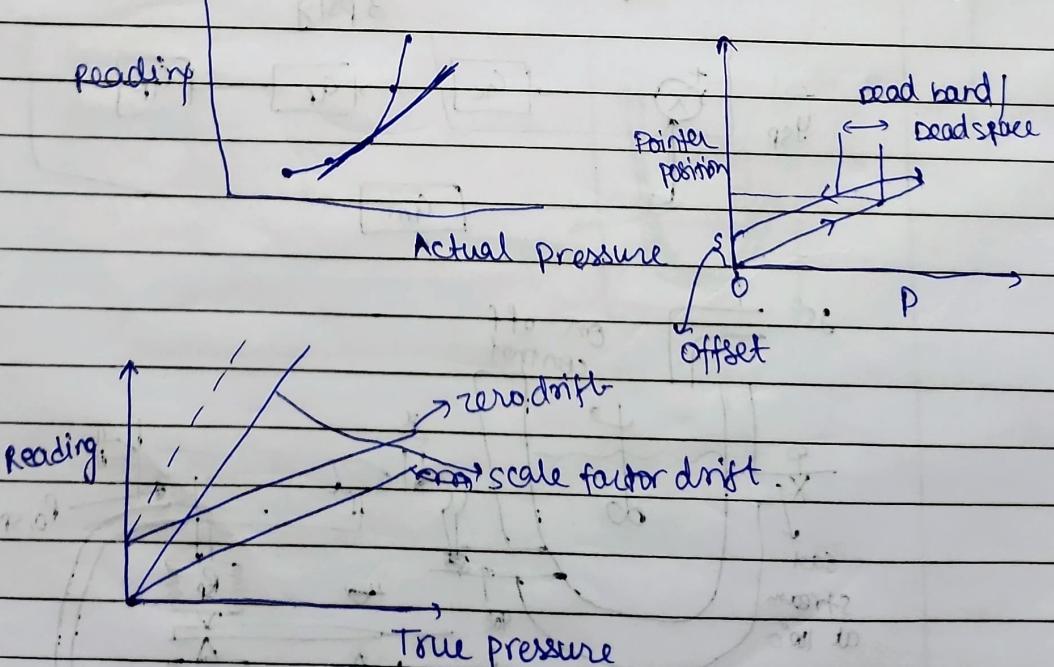


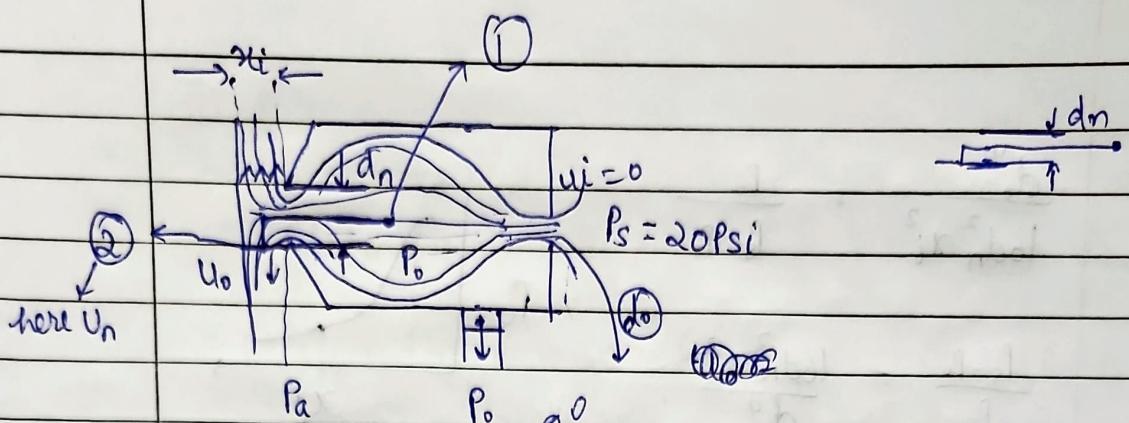
Generalized performance characteristics of instruments

- ① static characterization
- ② Dynamic "
- ③ calibration
- ④ primary standard
- ⑤ linearity



- ⑥ threshold
- ⑦ noise floor
- ⑧ resolution
- ⑨ hysteresis
- ⑩ scale readability
- ⑪ span
- ⑫ sensitivity
- ⑬ zero drift
- ⑭ scale factor drift





$$p_s + \frac{1}{2} \rho u_i^2 = p_o + \frac{1}{2} \rho u_o^2$$

$$c_0 \sqrt{\frac{2(p_s - p_o)}{\rho}} = u_o$$

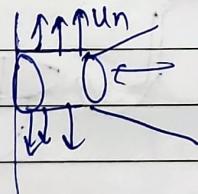
Bernoulli b/w ① and ②

$$p_o + \frac{1}{2} \rho u_i^2 + \rho g \frac{d_n}{2} = p_a + \frac{1}{2} \rho u_n^2$$

$$p_o - p_a + \frac{1}{2}$$

$$u_n = \sqrt{\frac{2p_o}{\rho}}$$

Now mass balance.



$$u_n \times \frac{\pi}{4} \times 2\pi x_i \left(\frac{d_o}{2} \right) = \frac{\pi d_o^2}{4} u_o \times \frac{\rho}{g}$$

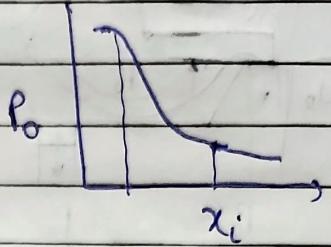
$$Q \int \frac{\pi p_o}{g} \times \frac{\pi}{4} \times 2\pi x_i \frac{d_n}{2} = \frac{\pi d_o^2}{4} c_0 \int \frac{2(p_s - p_o)}{g} \times \frac{\rho}{g}$$

$$\frac{d_o^2}{4d_n x_i} = \sqrt{\frac{p_o}{p_s - p_o}}$$

$$\frac{d_o^4}{16d_n^2 x_i^2} = \frac{P_0}{P_s - P_0}$$

$$\frac{P_s - P_0}{P_0} = \frac{16d_n^2 x_i^2}{d_o^4}$$

$$\frac{P_s}{P_0} = \frac{16d_n^2 x_i^2}{d_o^4} + 1$$



$$\frac{dP_o}{dx_i} = -k_{fn}$$

Pneumatic controller: proportional control

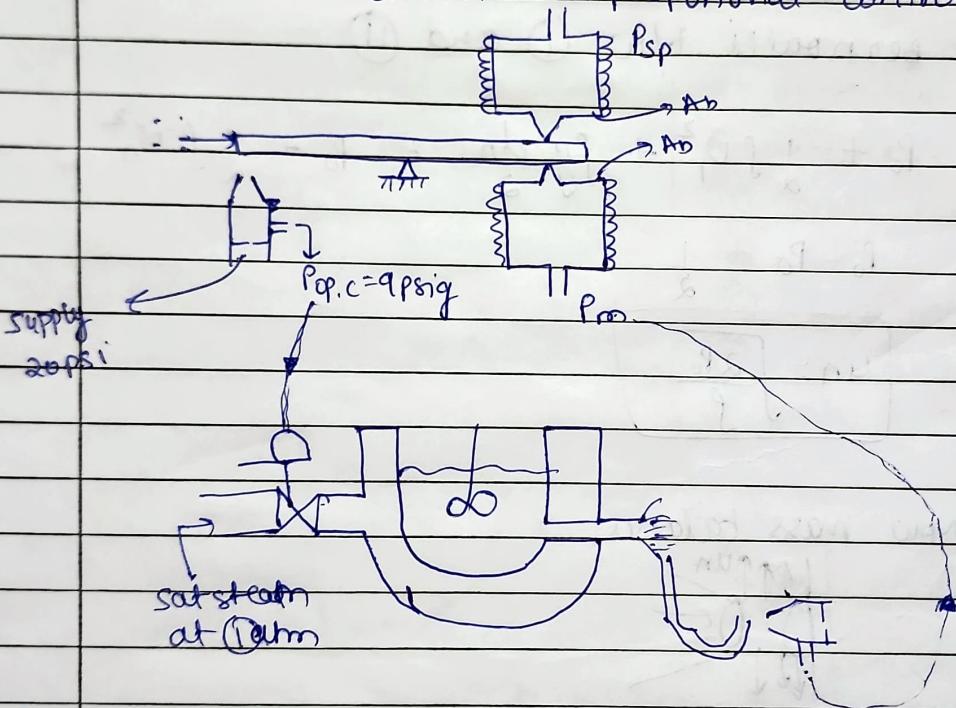
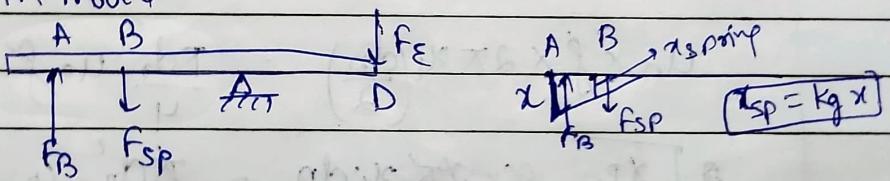


photo in motion



$$-ACFB,0 + BC.Fsp,0 - CD.FE,0 = 0 \quad \left\{ \begin{array}{l} \text{true about} \\ \text{when no} \\ \text{error in} \\ \text{position} \end{array} \right.$$

When there is an Error:

$$-ACFB + BC.Fsp - CD.FE = 0$$

— / —

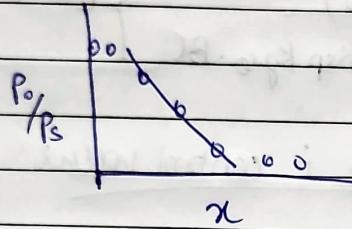
(1) - (1)

$$-AC(\cancel{AB} - \cancel{AC})$$

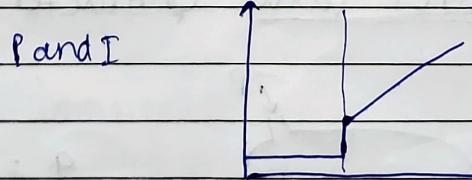
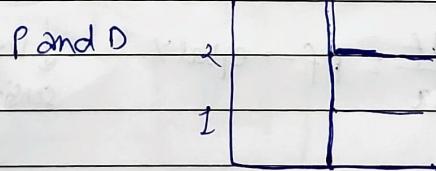
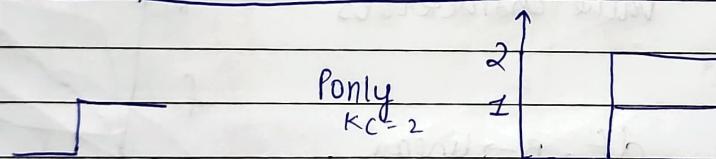
$$-AC(F_B - F_{B,0}) + BC(F_{sp} - F_{sp,0}) - CD(F_E - F_{E,0}) = 0$$

$$-AC(A_{fb,0}P_0 - A_{fb,0}P_{sp}) + BC \cdot K_{sp} \Delta x_{sp} - CD A_b (P_{sp} - P_m) = 0$$

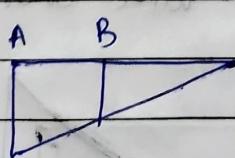
$$P_0 = -K_{fn} \cdot x \quad -AC \cdot A_{fb,0} (-K_{fn} x + K_{fn} x)$$



$$\text{so, } -AC \cdot A_{fb} (-K_{fn})(\Delta x) + BC \cdot K_{sp} \Delta x_{sp} - CD \cdot A_b \cdot E = 0$$



$$E = P_{sp} - P_m$$

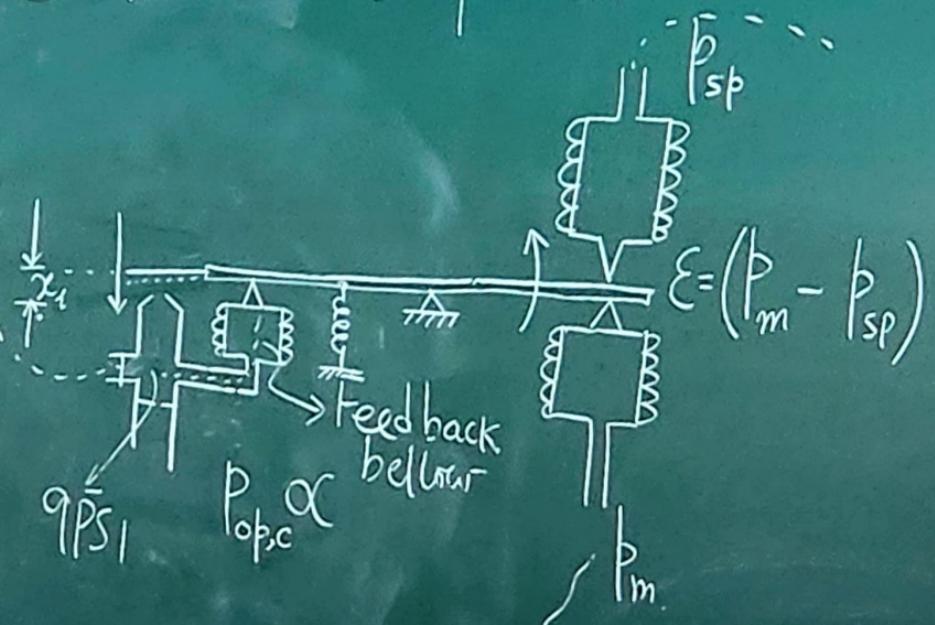
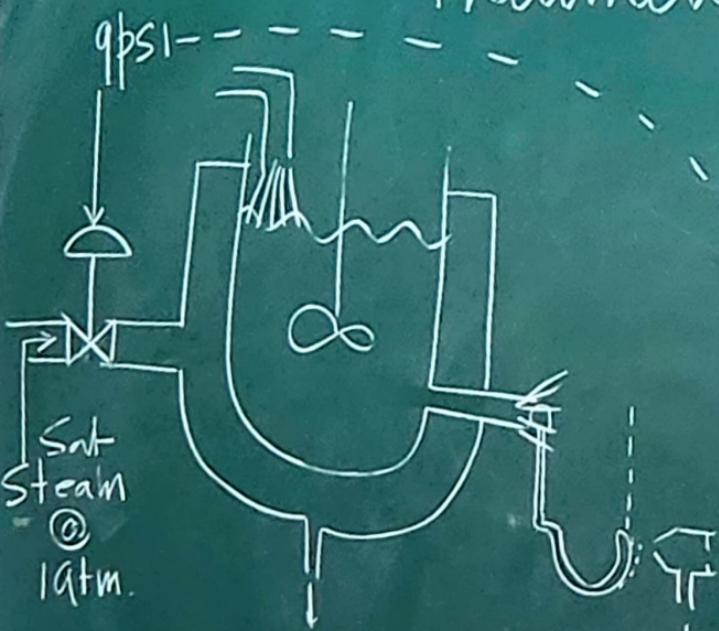


$$\begin{aligned} F_B &= F_{B,0} - F_{B,1} \\ &= P_{o,0} A_b - P_{o,1} A_b \\ &= (P_{o,0} - P_{o,1}) A_b \\ &= -K_{fn} x A_b \end{aligned}$$

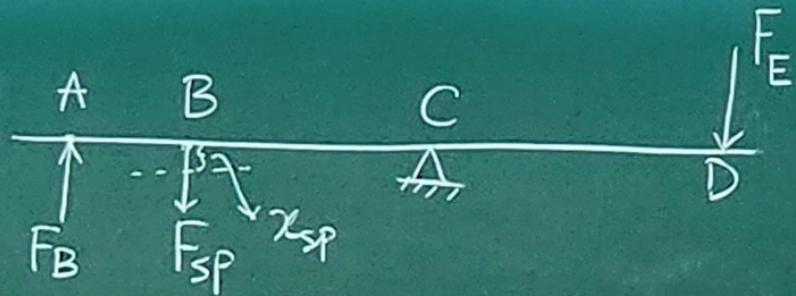
$$\Delta x_{sp} = k g x$$

$$P_o = -K_{fn} x$$

Pneumatic Controller: Proportional Control



control.



$$-P_{SP})$$

$$-AC \cdot F_{B_0} + BC \cdot F_{SP,0} - CD F_{E,0} = 0$$

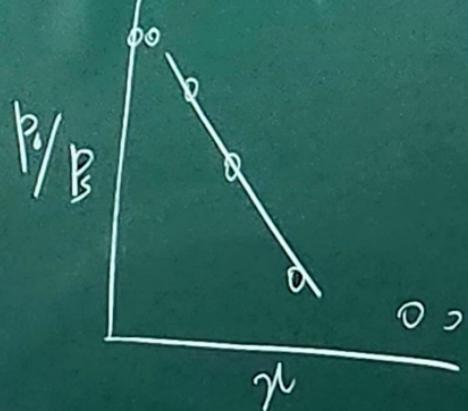
$$-AC F_B + BC F_{SP} - CD F_E = 0$$

$$-AC(F_B - F_{B_0}) + BC(F_{SP} - F_{SP,0}) - CD(F_E - F_{E,0}) = 0$$

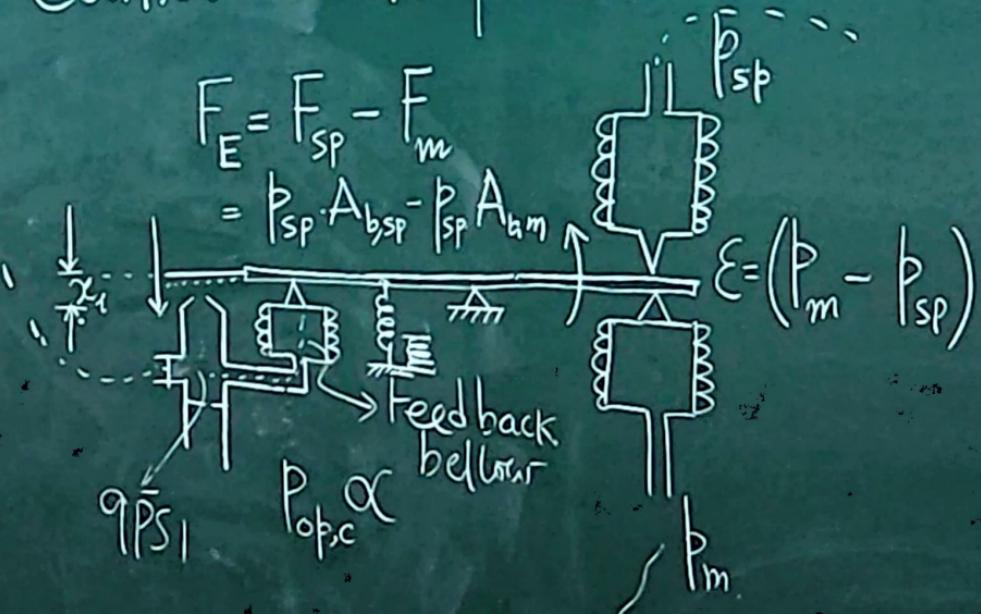
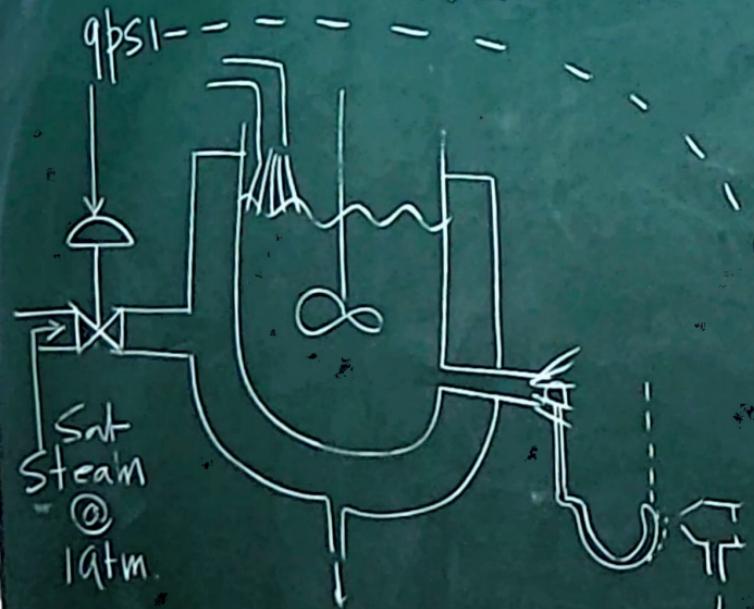
$$-AC(A_{Sb,b} P_0 - A_{Sb,b} P_{0,0}) + BC \cdot k_{SP} \cdot \chi_{SP} - CD \cdot A_b (P_{SP} - P_m) = 0$$

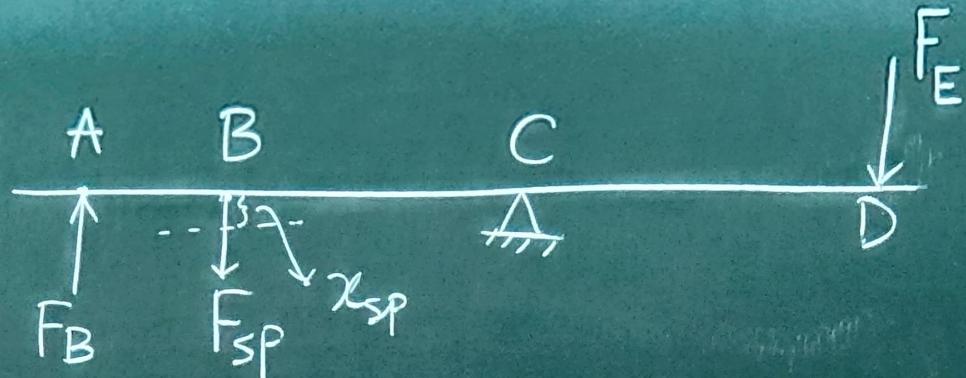
$$-AC \cdot A_{Sb,b} (-k_{fn} \chi + k_{fn} \chi_0) - CD \cdot A_b (P_{SP} - P_m) = 0$$

$$P_0 = -k_{fn} \cdot \chi$$



Pneumatic Controller: Proportional Control





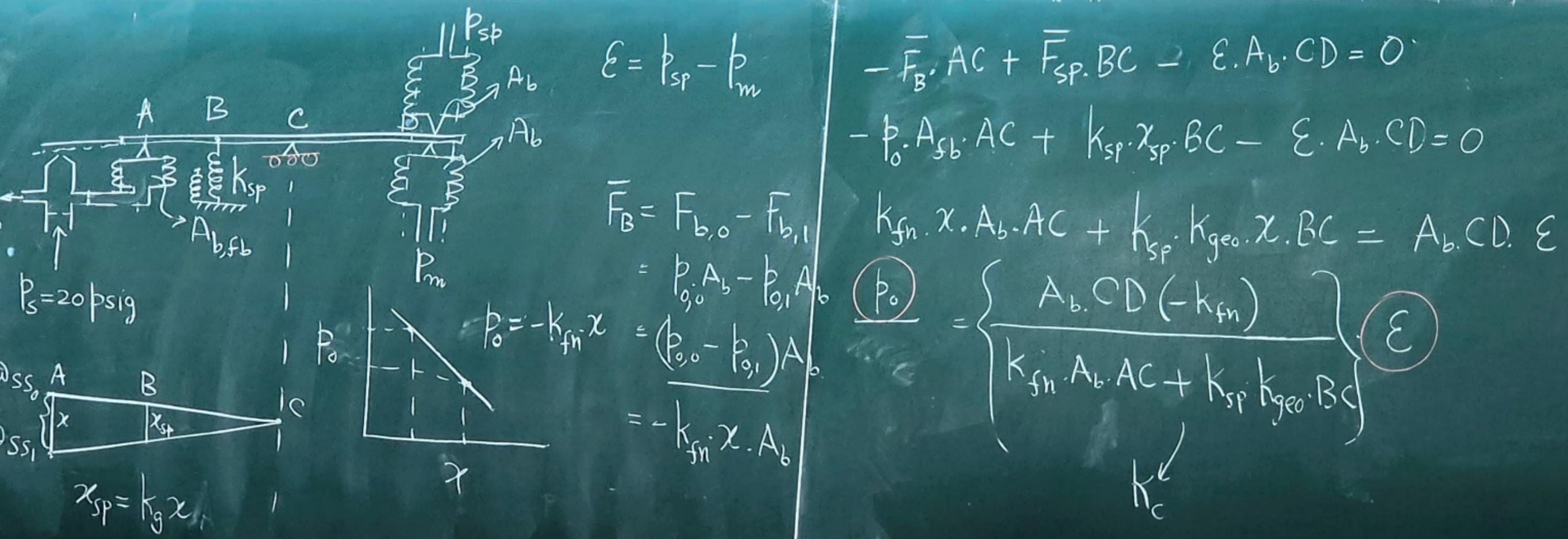
$$-AC \cdot F_{B_0} + BC \cdot F_{SP,0} - CD \cdot F_{E,0} = 0$$

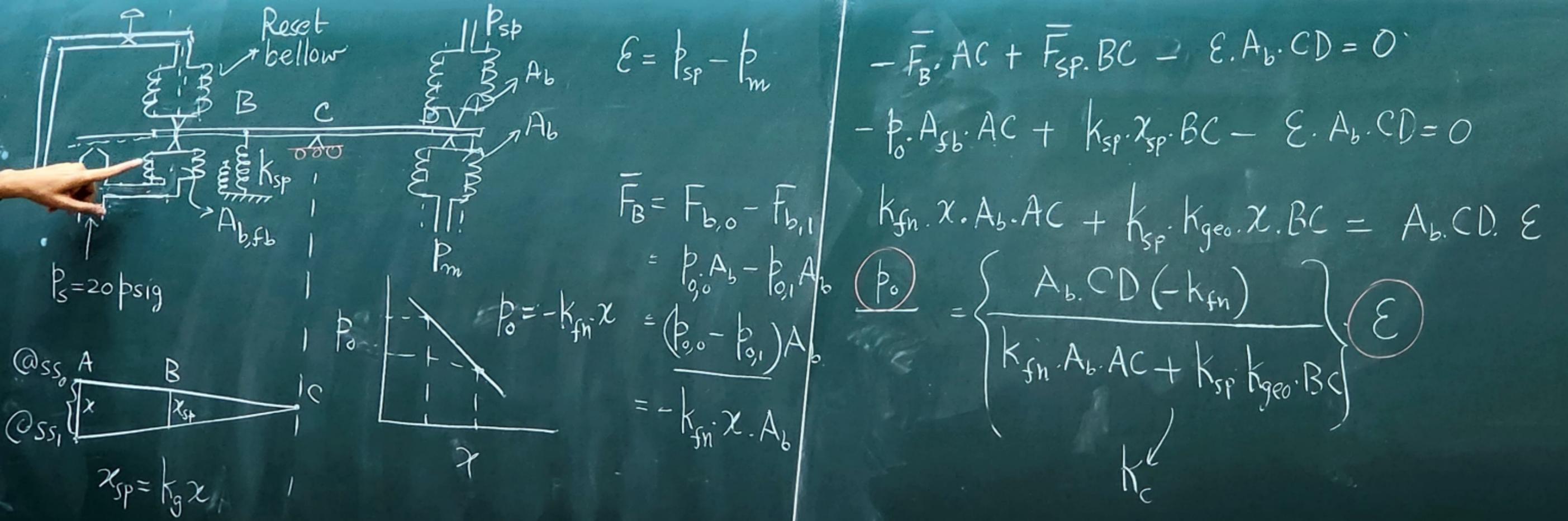
$$-AC \cdot F_B + BC \cdot F_{SP} - CD \cdot F_E = 0$$

$$-AC(F_B - F_{B_0}) + BC(F_{SP} - F_{SP,0}) - CD(F_E - F_{E,0}) = 0$$

$$-AC(A_{f_{b,b}} \cdot \beta_0 - A_{f_{b,b}} \cdot \beta_{0,0}) + BC \cdot k_{SP} \cdot x_{SP} - CD \cdot A_b (\beta_{SP} - \beta_m) = 0$$

$$-AC \cdot A_{f_{b,b}} (-k_{fh} x + k_{fh} x_0)$$





$$\epsilon = p_{sp} - p_m$$

$$\begin{aligned}
 \bar{F}_B &= F_{b,0} - F_{b,1} \\
 &= p_{o,0} A_b - p_{o,1} A_b \\
 &= (p_{o,0} - p_{o,1}) A_b \\
 &= -k_{fn} \chi \cdot A_b
 \end{aligned}$$

$$\begin{aligned}
 -\bar{F}_B \cdot AC + \bar{F}_{sp} \cdot BC - \epsilon \cdot A_b \cdot CD &= 0 \\
 -p_o \cdot A_{sb} \cdot AC + k_{sp} \cdot \chi_{sp} \cdot BC - \epsilon \cdot A_b \cdot CD &= 0 \\
 k_{fn} \cdot \chi \cdot A_b \cdot AC + k_{sp} \cdot k_{geo} \cdot \chi \cdot BC &= A_b \cdot CD \cdot \epsilon \\
 \underline{p_o} &= \frac{A_b \cdot CD \cdot (-k_{fn})}{k_{fn} \cdot A_b \cdot AC + k_{sp} \cdot k_{geo} \cdot BC} \cdot \underline{\epsilon} \\
 k_c
 \end{aligned}$$

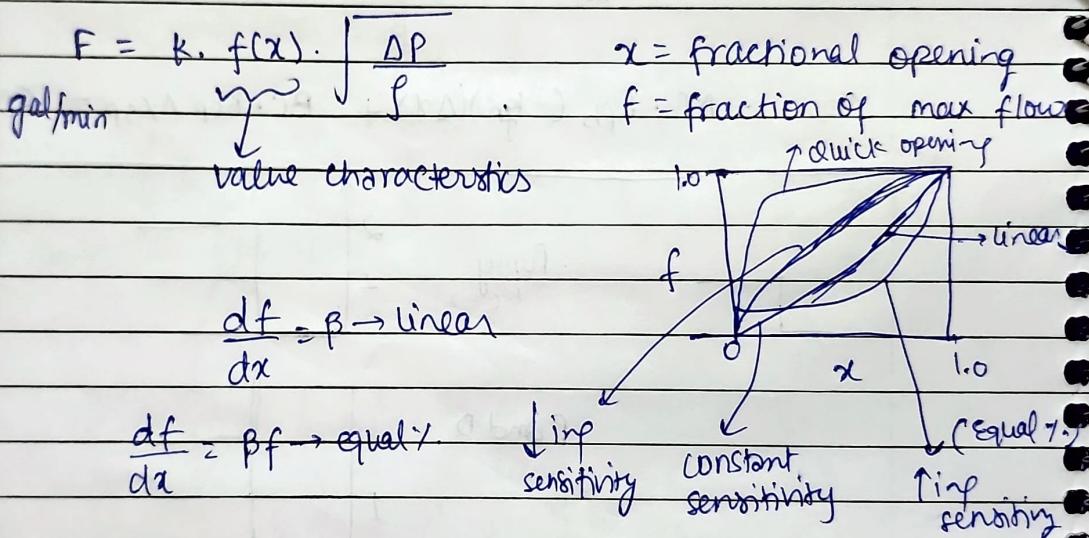
$$-F_{BAC} + \bar{F}_{sp, BC} - EA_b \cdot CD = 0$$

$$-P_0 A_{fb} AC + k_{sp} \chi_{sp} BC - EA_b CD = 0$$

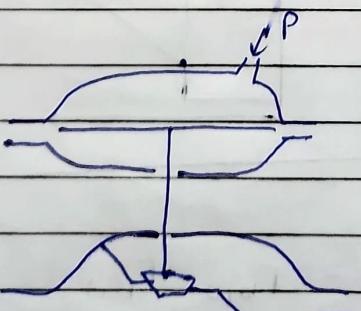
$$k_{fn} \chi_{Ab} AC + k_{sp} k_{geo} \chi_{bc} BC = A_b CDE$$

$$P_0 = \begin{cases} A_b CD (-k_{fn}) \\ k_{fn} A_b \cdot AC + k_{sp} k_{geo} \cdot BC \end{cases} E$$

Final control element i control value:



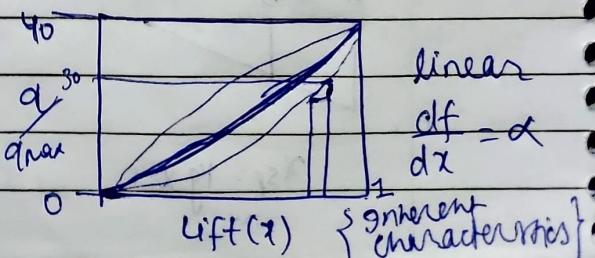
control value characteristics:



$$q = a \cdot f(x) \cdot \int \frac{\Delta Pv}{(s \cdot g)}$$

$f(x) = \text{value characteristics}$

$$q_{max} = C_v \sqrt{\frac{\Delta Pv}{(s \cdot g)}} \cdot f(x)$$



control valve characteristics

inherent
(constant ΔP
across the
valve)

installed
(under actual
operation: ΔP across
value changes as flow
changes)

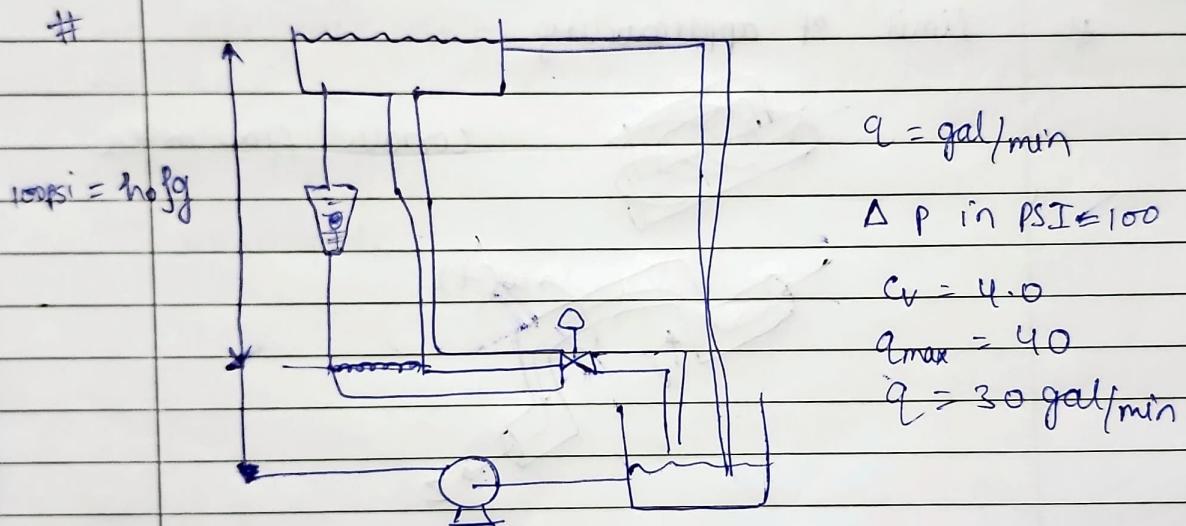
For equal y -value:

$$\frac{df}{dx} = \beta f$$

$$\frac{df_f}{dx} = \beta$$

$$\int_{f_0}^f \ln f = \int_{x_0}^x dx$$

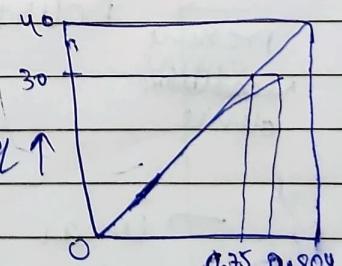
$$\frac{\ln f}{f_0} = x - x_0$$



Case a) line losses negligible

$$x=1 \quad q = 40 \text{ gal/min}$$

$$x=0.75 \quad q = 30 \text{ gal/min}$$



Case b) significant line loss

$$q = 30 \text{ gal/min}$$

$$\Delta P = 24 \text{ psi}$$

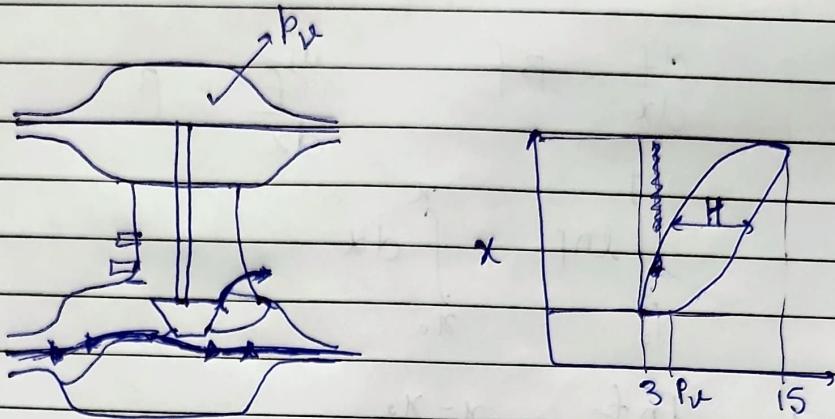
$$q_{\max} = 37.3 \text{ gal/min}$$

$$x = \frac{30}{37.3} = 0.804 \quad x(\text{lift})$$

Ch-6 sensors for measurement control

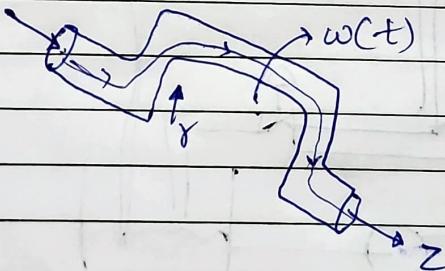
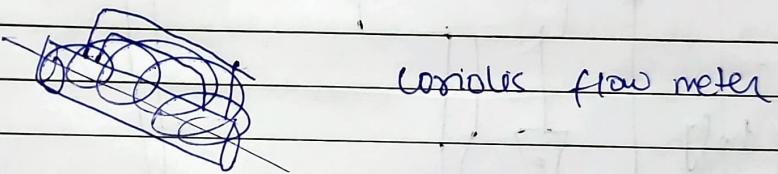
— / —

control valve positioner)



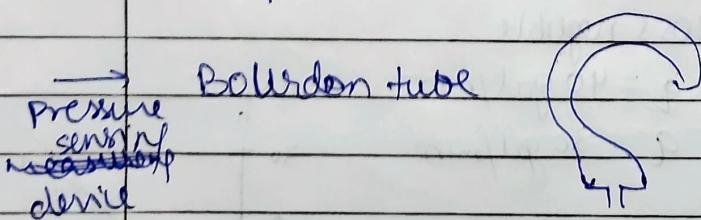
Dead band

limit of applicability.



Time of flight meter

Doppler



→
pressure
sensor
device

→ linear variable differential transformer

→ Diaphragm and Bellows

→ level monitoring pressure

Assorted instruments : Piezoelectric

→ composition monitoring

chromatography

Refractometry

→ temperature sensor:

a) thermo electric effect

→ Reference junction ~~for~~ correction

→ J-type, K-type thermocouples { Range, voltage response }

→ Resistance temp. (RTD)

→ thermistors

→ ~~radiation~~ radiation detectors

→ thermopile