HEAT TRANSFER [CH21204]

January 25, 2023

$$\begin{array}{c}
T_{\infty} \\
u_{\infty}
\end{array}$$

$$\begin{array}{c}
U \\
x \\
u(x, 0) = 0 \\
v(x, 0) = 0 \\
T(x, 0) = T_{s}
\end{array}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$u\,\frac{\partial u}{\partial x} + v\,\frac{\partial u}{\partial y} = v\,\frac{\partial^2 u}{\partial y^2}$$

$$u\frac{\partial T}{\partial x} + v\frac{\partial T}{\partial y} = \alpha \frac{\partial^2 T}{\partial y^2}$$

At
$$x = 0$$
:

$$u(0, y) = u_{\infty}, \quad T(0, y) = T_{\infty}$$

At
$$y = 0$$
:

$$u(x, 0) = 0,$$

At
$$y = 0$$
: $u(x, 0) = 0$, $v(x, 0) = 0$, $T(x, 0) = T_s$

As
$$y \to \infty$$
:

As
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: $u(x, \infty) = u_{\infty}$, $T(x, \infty) = T_{\infty}$

$$T(x, \infty) = T_{\infty}$$

$$\delta = \frac{5.0}{\sqrt{u_{\infty}/vx}} = \frac{5.0x}{\sqrt{Re_x}}$$

$$\delta = \frac{5.0}{\sqrt{u_{\infty}/vx}} = \frac{5.0x}{\sqrt{\text{Re}_x}} \qquad C_{f,x} = \frac{\tau_w}{\rho^{\text{V}^2/2}} = \frac{\tau_w}{\rho u_{\infty}^2/2} = 0.664 \text{ Re}_x^{-1/2}$$

$$Nu_x = \frac{h_x x}{k} = 0.332 \text{ Pr}^{1/3} \text{Re}_x^{1/2}$$
 $Pr > 0.6$ $T_f = (T_s + T_\infty)/2.$

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$$\delta_t = \frac{\delta}{\Pr^{1/3}} = \frac{5.0x}{\Pr^{1/3} \sqrt{\operatorname{Re}_x}}$$

 $\delta_t = \frac{\delta}{\Pr^{1/3}} = \frac{5.0x}{\Pr^{1/3} \sqrt{\Re e}}$ laminar flow over an isothermal flat plate

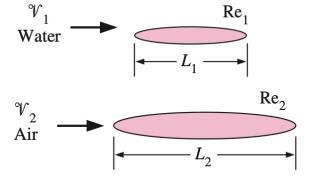
$$x^* = \frac{x}{L}$$
, $y^* = \frac{y}{L}$, $u^* = \frac{u}{v}$, $v^* = \frac{v}{v}$, $P^* = \frac{P}{\rho v^2}$, and $T^* = \frac{T - T_s}{T_{\infty} - T_s}$

Continuity:
$$\frac{\partial u^*}{\partial x^*} + \frac{\partial v^*}{\partial y^*} = 0$$

$$u^* \frac{\partial u^*}{\partial x^*} + v^* \frac{\partial u^*}{\partial y^*} = \frac{1}{\text{Re}_L} \frac{\partial^2 u^*}{\partial y^{*2}} - \frac{dP^*}{dx^*}$$

$$e^* \frac{\partial T^*}{\partial x^*} + v^* \frac{\partial T^*}{\partial y^*} = \frac{1}{\text{Re}_L \text{Pr}} \frac{\partial^2 T^*}{\partial y^{*2}}$$

$$u^*(0, y^*) = 1$$
, $u^*(x^*, 0) = 0$, $u^*(x^*, \infty) = 1$, $v^*(x^*, 0) = 0$, $T^*(0, y^*) = 1$, $T^*(x^*, 0) = 0$, $T^*(x^*, \infty) = 1$



Parameters before nondimensionalizing

$$L, \mathcal{V}, T_{\infty}, T_{s}, \nu, \alpha$$

Parameters after nondimensionalizing:

Re, Pr

If
$$Re_1 = Re_2$$
, then $C_{f1} = C_{f2}$

Local Nusselt number:

$$Nu_x = function (x^*, Re_L, Pr)$$

Average Nusselt number:

$$Nu = function (Re_L, Pr)$$

A common form of Nusselt number:

$$Nu = C \operatorname{Re}_L^m \operatorname{Pr}^n$$

 $C_f = f_4(Re_L)$

Momentum: Reynolds analogy

Heat: Chilton-Colburn analogy

$$u^* \frac{\partial u^*}{\partial x^*} + v^* \frac{\partial u^*}{\partial y^*} = \frac{1}{\operatorname{Re}_L} \frac{\partial^2 u^*}{\partial y^{*2}} - \frac{dP^*}{dx^*}$$

$$u^* \frac{\partial T^*}{\partial x^*} + v^* \frac{\partial T^*}{\partial y^*} = \frac{1}{\operatorname{Re}_L} \frac{\partial^2 u^*}{\partial y^{*2}}$$

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$$\tau_s = \left. \mu \frac{\partial u}{\partial y} \right|_{y=0} = \frac{\mu \mathcal{V}}{L} \frac{\partial u^*}{\partial y^*} \right|_{y^*=0} = \frac{\mu \mathcal{V}}{L} f_2(x^*, \operatorname{Re}_L)$$

$$C_{f,x} = \frac{\tau_s}{\rho V^2/2} = \frac{\mu V/L}{\rho V^2/2} f_2(x^*, \text{Re}_L) = \frac{2}{\text{Re}_L} f_2(x^*, \text{Re}_l) = f_3(x^*, \text{Re}_L)$$

$$\operatorname{Nu}_{x} = \frac{hL}{k} = \frac{\partial T^{*}}{\partial y^{*}}\Big|_{y^{*}=0} = g_{2}(x^{*}, \operatorname{Re}_{L}, \operatorname{Pr})$$

$$C_{f,x} \frac{\text{Re}_L}{2} = \text{Nu}_x \qquad (\text{Pr} = 1)$$

Profiles:
$$u^* = T$$

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Gradients: $\frac{\partial u^*}{\partial y^*}\Big|_{y^*=0} = \frac{\partial T^*}{\partial y^*}\Big|_{y^*=0}$

Analogy: $C_{f,x}\frac{\operatorname{Re}_L}{2} = \operatorname{Nu}_x$
 $C_{f,x} = \operatorname{St}_x$ (Pr = 1)

$$St = \frac{h}{\rho C_p \mathcal{V}} = \frac{\operatorname{Nu}}{\operatorname{Re}_L \operatorname{Pr}}$$

Analogy:
$$C_{f,x} \frac{\text{Re}_L}{2} = \text{Nu}$$

$$\frac{C_{f,x}}{2} = \operatorname{St}_x \qquad (\operatorname{Pr} = 1)$$

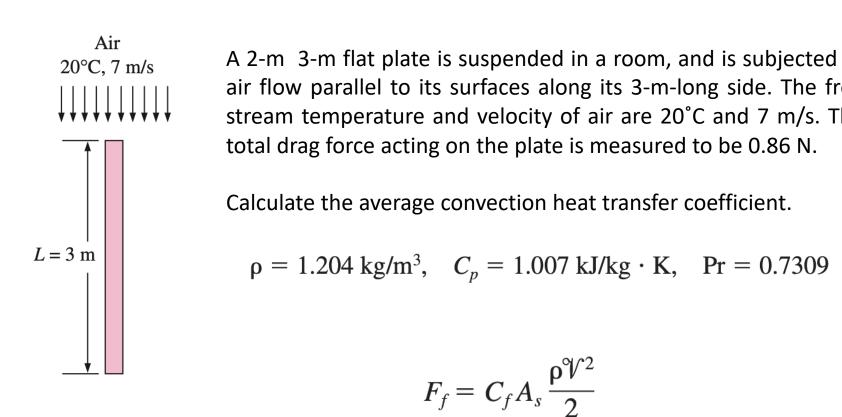
$$St = \frac{h}{\rho C_p \text{ of }} = \frac{\text{Nu}}{\text{Re}_L \text{Pr}}$$

$$C_{f, x} = 0.664 \text{ Re}_x^{-1/2}$$
 and $Nu_x = 0.332 \text{ Pr}^{1/3} \text{ Re}_x^{1/2}$

Modified Reynolds analogy or Chilton–Colburn analogy:

$$C_{f,x} \frac{Re_L}{2} = \text{Nu}_x \text{Pr}^{-1/3} \qquad \text{or} \qquad \frac{C_{f,x}}{2} = \frac{h_x}{\rho C_p \mathcal{V}} \text{Pr}^{2/3} \equiv j_H \qquad 0.6 < \text{Pr} < 60.$$

$$Colburn j\text{-factor}$$



A 2-m 3-m flat plate is suspended in a room, and is subjected to air flow parallel to its surfaces along its 3-m-long side. The free stream temperature and velocity of air are 20°C and 7 m/s. The total drag force acting on the plate is measured to be 0.86 N.

$$\rho = 1.204 \text{ kg/m}^3$$
, $C_p = 1.007 \text{ kJ/kg} \cdot \text{K}$, $Pr = 0.7309$

$$F_f = C_f A_s \frac{\rho V^2}{2}$$

$$A_s = 2WL = 2(2 \text{ m})(3 \text{ m}) = 12 \text{ m}^2$$

$$C_f = \frac{F_f}{\rho A_s V^2/2} = \frac{0.86 \text{ N}}{(1.204 \text{ kg/m}^3)(12 \text{ m}^2)(7 \text{ m/s})^2/2} \left(\frac{1 \text{kg} \cdot \text{m/s}^2}{1 \text{ N}}\right) = 0.00243$$

$$h = \frac{C_f \, \text{pVC}_p}{2 \, \text{Pr}^{2/3}} = \frac{0.00243}{2} \, \frac{(1.204 \, \text{kg/m}^3)(7 \, \text{m/s})(1007 \, \text{J/kg} \cdot ^{\circ}\text{C})}{0.7309^{2/3}} = 12.7 \, \text{W/m}^2 \cdot ^{\circ}\text{C}$$