Heat, Mass and Momentum Transfer Analogy

Introduction to Heat and Mass Transfer Incropera and Dewitt

Continuity equation and x-momentum equation (B.L.)

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad \tag{I}$$

$$u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y} = -\frac{1}{\rho}\frac{\partial p}{\partial x} + \underbrace{\gamma}\frac{\partial^2 u}{\partial y^2} \dots (II) \qquad \underbrace{\frac{\mu}{\rho}}$$

Energy equation:

$$u\frac{\partial T}{\partial x} + v\frac{\partial T}{\partial y} = \alpha \frac{\partial^2 T}{\partial y^2} + q + \mu \phi \dots \text{(III)} \quad \left(\frac{\kappa}{\rho C_P}\right)$$

Concentration BL:

$$u\frac{\partial C_A}{\partial x} + v\frac{\partial C_A}{\partial y} = D_{AB}\frac{\partial^2 C_A}{\partial y^2} + R_A$$
....(IV)



THE CONVECTION TRANSFER EQUATIONS

2-D Steady flow

Approximations and special considerations:

$$\frac{v_x}{U} \quad \frac{T_s - T}{T_s - T_{\infty}} \quad \frac{C_{AS} - C_A}{C_{AS} - C_{A\infty}}$$

Incompressible, Constant properties, Negligible body forces, non-reacting $N_A=0$, No energy generation q=0, negligible viscous dissipation $\phi=0$

B.L. approximations:

$$\frac{u \gg v}{\frac{\partial u}{\partial y} \gg \frac{\partial u}{\partial x}, \frac{\partial v}{\partial x}, \frac{\partial v}{\partial y}} \qquad \text{Velocity B.L.}$$

$$\frac{\partial T}{\partial y} \gg \frac{\partial T}{\partial x} \qquad \text{Thermal B.L.}$$

$$\frac{\partial C_A}{\partial y} \gg \frac{\partial C_A}{\partial x} \qquad \text{Conc. B.L.}$$

BOUNDARY LAYER SIMILARITY – THE NORMALIZED CONVECTION TRANSFER EQUATIONS

Non-dimensionalizing parameters

$$x^* = \frac{x}{L}$$
 $y^* = \frac{y}{L}$ $u^* = \frac{u}{V}$ $v^* = \frac{v}{V}$ $T^* = \frac{T - T_s}{T_\infty - T_s}$ $C_A^* = \frac{C_A - C_{AS}}{C_{A\infty} - C_{AS}}$

Wall

Free stream

$$u^{*} \frac{\partial u^{*}}{\partial x^{*}} + v^{*} \frac{\partial u^{*}}{\partial y^{*}} = -\frac{dp^{*}}{dx^{*}} + \underbrace{v^{*} \frac{\partial^{2} u^{*}}{\partial y^{*2}}}_{VL} \underbrace{v^{*}(x^{*}, 0) = 0}_{v^{*}(x^{*}, 0) = 0}$$

$$u^{*}(x^{*}, \infty) = \frac{u_{\infty}(x^{*})}{V}$$

$$\frac{1}{Re}$$

$$u^*(x^*,0) = 0$$

 $v^*(x^*,0) = 0$

$$u^*(x^*,\infty) = \frac{u_\infty(x^*)}{V}$$

$$Re_L$$

$$u^* \frac{\partial T^*}{\partial x^*} + v^* \frac{\partial T^*}{\partial y^*} = \underbrace{\frac{\alpha}{VL}} \frac{\partial^2 T^*}{\partial y^{*2}} \qquad T^*(x^*, 0) = 0 \qquad T^*(x^*, \infty) = 1$$

$$\frac{1}{\text{Re.Pr}}$$

$$T^*(x^*,0)=0$$

$$T^*(x^*,\infty)=1$$

$$Re_L, Pr$$

$$u^* \frac{\partial C_A^*}{\partial x^*} + v^* \frac{\partial C_A^*}{\partial y^*} = \underbrace{D_{AB}}_{VL} \underbrace{\partial^2 C_A^*}_{\partial y^{*2}} \qquad C_A^*(x^*, 0) = 0 \qquad C_A^*(x^*, \infty) = 1$$

$$C_A^*(x^*,0)=0$$

$$C_A^*(x^*,\infty)=1$$

$$Re_L, Sc$$

FUNCTIONAL FORM OF THE SOLUTIONS

$$u^* = f_1\left(x^*, y^*, \operatorname{Re}_L, \frac{dp^*}{dx^*}\right)$$

Shear stress at the surf. $y^* = 0$

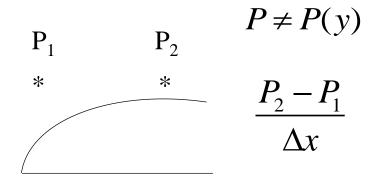
$$\tau_{s} = \mu \frac{\partial u}{\partial y} \bigg]_{y=0} = \frac{\mu V}{L} \frac{\partial u^{*}}{\partial y^{*}} \bigg]_{y^{*}=0}$$

$$C_f = \frac{\tau_s}{\frac{1}{2}\rho V^2} = \frac{2}{\operatorname{Re}_L} \frac{\partial u^*}{\partial y^*} \bigg]_{y^*=0}$$

$$C_f \frac{\mathrm{Re}_L}{2} = f_2(x^*, \mathrm{Re}_L)$$

$$x^* = \frac{x}{L} \qquad u^* = \frac{u}{V}$$
$$y^* = \frac{y}{L} \qquad v^* = \frac{v}{V}$$

For a prescribed geometry



$$T^* = f_3\left(x^*, y^*, \operatorname{Re}_L, \operatorname{Pr}, \frac{dp^*}{dx^*}\right)$$

$$q_{s} = -k_{f} \frac{\partial T}{\partial y} \bigg]_{y=0} \Rightarrow h = \frac{-k_{f} \frac{\partial T}{\partial y} \bigg]_{y=0}}{T_{s} - T_{\infty}}$$

Convective heat transfer coefficient
$$h = -\frac{k_f}{L} \frac{\left(T_{\infty} - T_{S}\right)}{\left(T_{S} - T_{\infty}\right)} \frac{\partial T^*}{\partial y^*} \bigg]_{y^* = 0}$$

$$Nu = \frac{hL}{k_f} = \frac{\partial T^*}{\partial y^*} \bigg]_{y^* = 0}$$

$$Nu = f_4(x^*, \text{Re}, \text{Pr})$$

 $Nu = f_4(x^*, \text{Re}, \text{Pr})$ For a prescribed geometry

Average Nu number
$$\overline{Nu} = \frac{\overline{h}L}{\kappa_f} = f_5 (\text{Re, Pr})$$

$$C_A^* = f_6\left(x^*, y^*, \operatorname{Re}_L, Sc, \frac{dp^*}{dx^*}\right)$$

$$N_A^{"} = -D_{AB} \frac{\partial C_A}{\partial y} \bigg]_{y=0} = h_m (C_{AS} - C_{A\infty})$$
 Convective mass transfer coefficient

$$h_{m} = \frac{-D_{AB} \frac{\partial C_{A}}{\partial y}}{C_{AS} - C_{AS}}$$

$$h_{m} = \frac{D_{AB}}{L} \frac{\partial C_{A}^{*}}{\partial y^{*}} \bigg]_{y^{*}=0}$$

$$\frac{h_m}{L}$$
 = Sherwood No = $\frac{\partial C_A^*}{\partial y^*} \bigg]_{y^*=0}$

$$Sh = f_7(x^*, Re_L, Sc)$$
 For a prescribed geometry

$$\overline{Sh} = \frac{\overline{h_m}L}{D_{AR}} = f_8 \left(\text{Re}_L, Sc \right)$$

Wall

Free stream

$$u^* \frac{\partial u^*}{\partial x^*} + v^* \frac{\partial u^*}{\partial y^*} = -\frac{dp^*}{dx^*} + \underbrace{\begin{pmatrix} v \\ VL \end{pmatrix}} \frac{\partial^2 u^*}{\partial y^{*2}} \qquad u^*(x^*, 0) = 0$$

$$v^*(x^*, 0) = 0$$

$$v^*(x^*, 0) = 0$$

$$u^*(x^*, \infty) = \frac{u_{\infty}(x^*)}{V}$$

$$u^*(x^*, 0) = 0$$
$$v^*(x^*, 0) = 0$$

$$u^*(x^*,\infty) = \frac{u_\infty(x^*)}{V}$$

 Re_L

$$u^* \frac{\partial T^*}{\partial x^*} + v^* \frac{\partial T^*}{\partial y^*} = \underbrace{\left(\frac{\alpha}{VL}\right)}_{Oy^{*2}} \frac{\partial^2 T^*}{\partial y^{*2}} \qquad T^*(x^*, 0) = 0 \qquad T^*(x^*, \infty) = 1$$

$$\frac{1}{\text{Re.Pr}}$$

$$T^*(x^*,0)=0$$

$$T^*(x^*,\infty)=1$$

$$Re_L, Pr$$

$$u^* \frac{\partial C_A^*}{\partial x^*} + v^* \frac{\partial C_A^*}{\partial y^*} = \underbrace{D_{AB}}_{VL} \underbrace{\partial^2 C_A^*}_{\partial y^{*2}} \qquad C_A^*(x^*, 0) = 0 \qquad C_A^*(x^*, \infty) = 1$$

$$Re.Sc$$

$$C_A^*(x^*,0)=0$$

$$C_A^*(x^*,\infty)=1$$

$$Re_L, Sc$$

THE REYNOLD'S ANALOGY

$$\frac{dp^*}{dx^*} = 0$$

 $\frac{dp^*}{dx^*} = 0 \qquad \text{Pr} = Sc = 1 \Rightarrow \text{ Solutions of } u^* \ T^* \text{ and } C_A^* \text{ must be equivalent.}$

$$f_1 = f_3 = f_6$$

Geometric similarity

Same is true for C_f , Nu and Sh

Dynamic similarity

$$f_2 = f_4 = f_7$$

$$C_f \frac{\text{Re}_L}{2} = Nu = Sh$$

$$\frac{Nu}{\text{Re.Pr}}$$
 - Stanton no for heat transfer

$$\frac{C_f}{2} = \frac{Nu}{\text{Re}_L.\text{Pr}} = \frac{Sh}{\text{Re}_L.Sc}$$

$$\frac{Sh}{\text{Re.}Sc}$$
 - Stanton no for mass transfer

$$\frac{C_f}{2} = St = St_m$$

$$u^* = f_1\left(x^*, y^*, \operatorname{Re}_L, \frac{dp^*}{dx^*}\right)$$

$$T^* = f_3\left(x^*, y^*, \operatorname{Re}_L, \operatorname{Pr}, \frac{dp^*}{dx^*}\right)$$

$$C_A^* = f_6\left(x^*, y^*, \operatorname{Re}_L, \operatorname{Sc}, \frac{dp^*}{dx^*}\right)$$

$$C_f \frac{\mathrm{Re}_L}{2} = f_2(x^*, \mathrm{Re}_L)$$

$$Nu = f_4(x^*, \text{Re}, \text{Pr})$$
 For a prescribed geometry

$$Sh = f_7(x^*, Re_L, Sc)$$
 For a prescribed geometry

Modified Reynold's analogy

Chilton Colburn analogy

$$\frac{C_f}{2} = St. \Pr^{\frac{2}{3}} = j_H$$

$$\frac{C_f}{2} = St_m.Sc^{\frac{2}{3}} = j_m$$

$$\frac{C_f}{2} = St. \Pr^{\frac{2}{3}} = St_m. Sc^{\frac{2}{3}}$$

$$\frac{C_f}{2} = j_H = j_m$$

j =Colburn 'j' factor

BOUNDARY LAYER ANALOGIES

$$Pr = \frac{v}{\alpha} = \frac{Momentum diffusivity}{Thermal diffusivity}$$

For laminar B.L.
$$\frac{\delta}{\delta_t} = \Pr^n$$

For gas: $\delta_t \approx \delta$ For liquid metal: $\delta_t \gg \delta$

For oil: $\delta_t \ll \delta$

$$Sc = \frac{v}{D_{AB}} = \frac{\text{Momentum diffusivity}}{\text{Mass diffusivity}} \qquad \frac{\delta}{\delta_c} = Sc^n$$

Lewis number Le =
$$\frac{\text{Thermal diffusivity}}{\text{Mass diffusivity}} = \frac{\alpha}{D_{AB}} = \frac{\nu}{D_{AB}} \cdot \frac{\alpha}{\nu} = \frac{Sc}{Pr}$$

For simultaneous heat and mass transfer: $\frac{\delta_t}{\delta_c} \approx Le^n$ For most applications: $n = \frac{1}{3}$

Chilton Colburn analogy (Modified Reynold's analogy)

$$St = \frac{Nu}{\text{Re.Pr}}$$

$$\frac{C_f}{2} = St.\Pr^{\frac{2}{3}} = j_H = St_m.Sc^{\frac{2}{3}} = j_m$$

$$St_m = \frac{Sh}{\text{Re.}Sc}$$

	Momentum	Heat	Mass
Laminar	$C_{f,x} = 0.664 \mathrm{Re}_{x}^{-1/2}$ $\mathrm{Re}_{x} < 5 \times 10^{5}$	$Nu_x = 0.332 \text{Re}_x^{1/2} \text{Pr}^{1/3}$ $\overline{Nu}_L = 0.664 \text{Re}_L^{1/2} \text{Pr}^{1/3}$ $\text{Re}_x < 5 \times 10^5, 0.6 < \text{Pr} < 50$	$Sh_x = 0.332 \text{Re}_x^{1/2} Sc^{1/3}$ $\overline{Sh}_L = 0.664 \text{Re}_L^{1/2} Sc^{1/3}$ $\text{Re}_x < 5 \times 10^5, 0.6 < Sc < 300$
Mixed / Turbulent	$C_{f,L} = \frac{0.074}{\text{Re}_L^{1/5}} - \frac{1742}{\text{Re}_L}$ $5 \times 10^5 < \text{Re}_L < 10^8$ $\text{Re}_{x,c} = 5 \times 10^5$	$\overline{Nu_L} = (0.037R^{4/5} - 871) Pr^{1/3}$ $0.6 \le Pr \le 50, Re_{x,c} = 5 \times 10^5$ $5 \times 10^5 < Re_L < 10^8$	$\overline{Sh_L} = (0.037 \text{Re}^{4/5} - 871) Sc^{1/3}$ $0.6 \le Sc \le 300, \text{Re}_{x,c} = 5 \times 10^5$ $5 \times 10^5 < \text{Re}_L < 10^8$