OLITE

Differentiable Lighting Informed Trajectory Evaluation for On-Orbit Inspection



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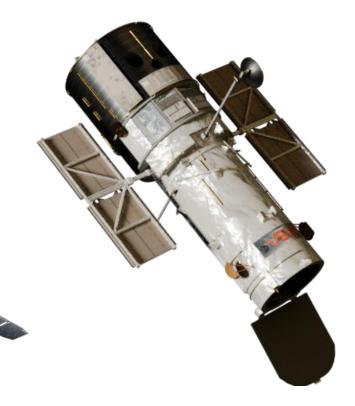




76TH
INTERNATIONAL
ASTRONAUTICAL
CONGRESS



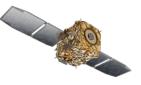




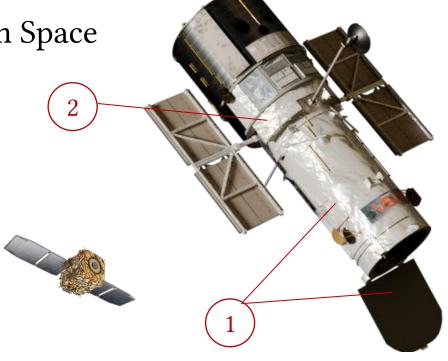


1. High-dynamic range



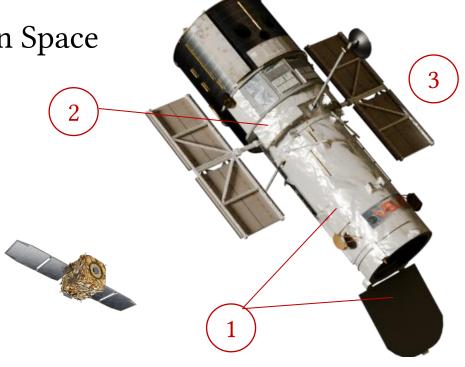


- 1. High-dynamic range
- 2. Specular objects

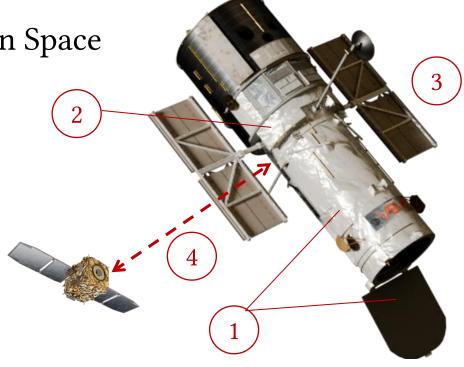




- 1. High-dynamic range
- 2. Specular objects
- 3. Self-shadowing

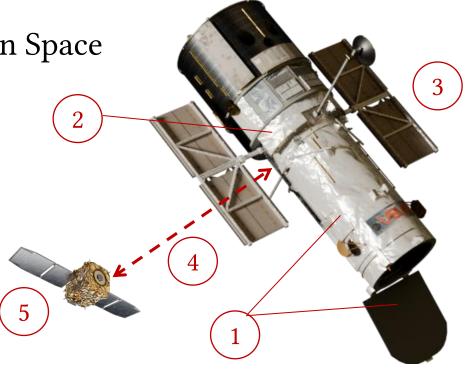


- 1. High-dynamic range
- 2. Specular objects
- 3. Self-shadowing
- 4. Dynamic lighting/imaging distances



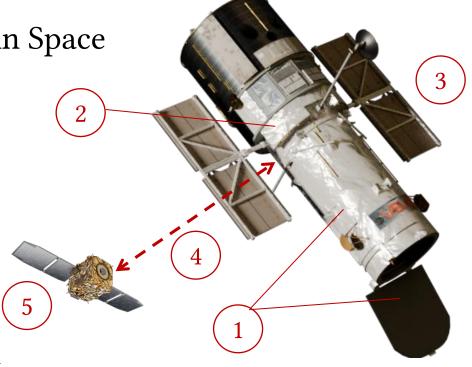


- 1. High-dynamic range
- 2. Specular objects
- 3. Self-shadowing
- 4. Dynamic lighting/imaging distances
- 5. Cannot just move anywhere in an orbit to avoid these conditions





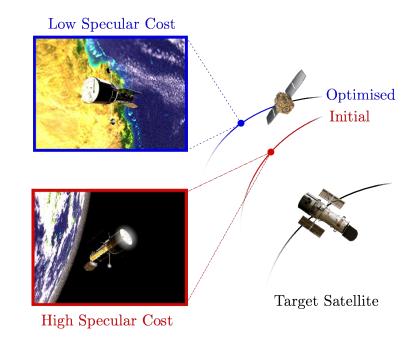
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But what if you could plan ahead and choose your orbit?

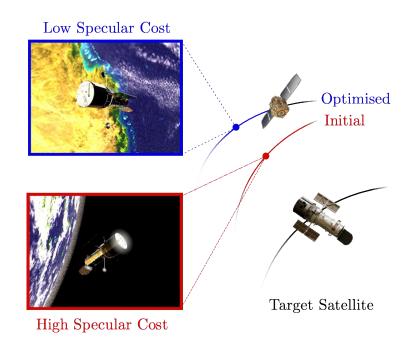


Our Approach



Our Approach

Optimise orbit of inspection "chaser" satellite to minimise visual costs relative to a "target" satellite.

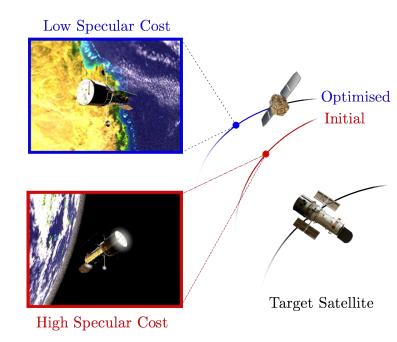


Our Approach

Optimise orbit of inspection "chaser" satellite to minimise visual costs relative to a "target" satellite.

How?

- Build an end-to-end differentiable simulator for on-orbit inspection
 - Differentiable orbit propagation
 - o Differentiable rendering
- Minimise arbitrary costs via simple gradient descent.



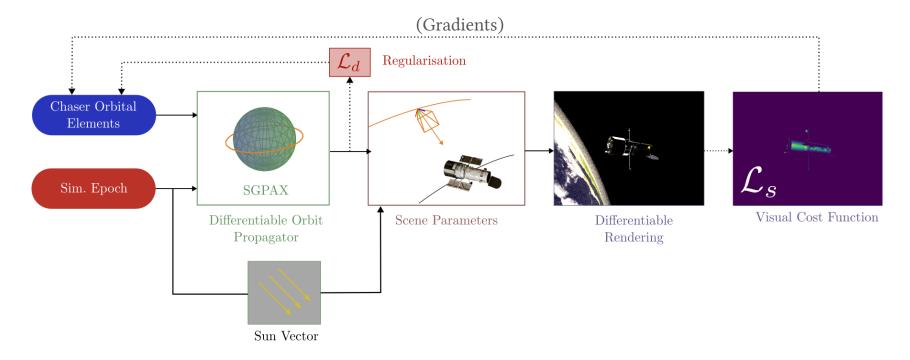


Related Work

NAME	Diff. Orbits?	Diff. Rendering?	Photometric?	Optical Effects?	Photorealistic?
ALL-STAR [Li et al. 2024]					
SPIN [Montalvo et al. 2024]					
SISPO [Pajusalu et al. 2022]					
HySIM [Felicetti et al. 2024]					
ƏLITE [Ours]					



End-to-End Differentiable Inspection

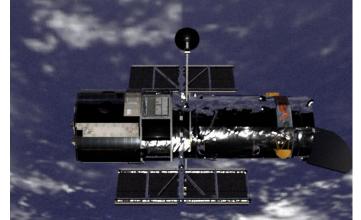






 Differentiable orbit propagation in JAX + Mitsuba 3 rendering



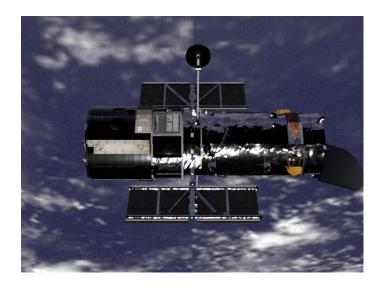




- Differentiable orbit propagation in JAX + Mitsuba 3 rendering
- Assume state & geometry are known (requires depths and surface normals)





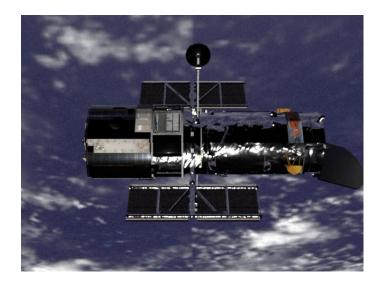




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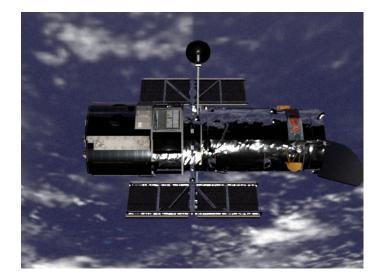




- Differentiable orbit propagation in JAX +
 Mitsuba 3 rendering
- Assume state & geometry are known (requires depths and surface normals)
- Assume always pointing at target
- Simulate passive inspections e.g.,
 circular "football" orbits







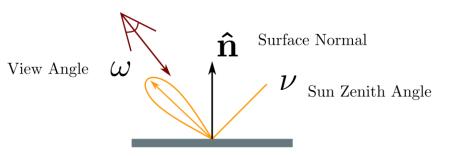




- Main cost: reduce direct specular reflections seen by the camera

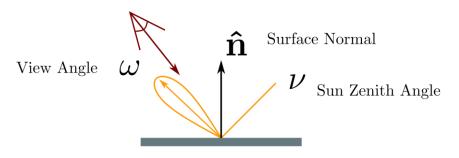


- Main cost: reduce direct specular reflections seen by the camera
 - Model sun illumination direction, use known geometry
 - Use physically-based reflection model, model strength of reflection from surface seen by the camera

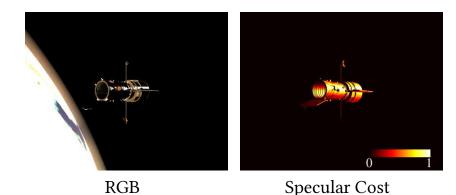


Phong Reflection Model

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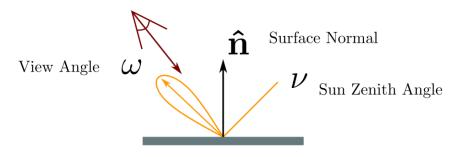


Phong Reflection Model

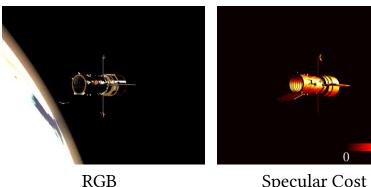




- Main cost: reduce direct specular reflections seen by the camera
 - Model sun illumination direction, use known geometry
 - Use physically-based reflection model, model strength of reflection from surface seen by the camera
- Additional cost: relative distance between chaser and target to avoid drift



Phong Reflection Model





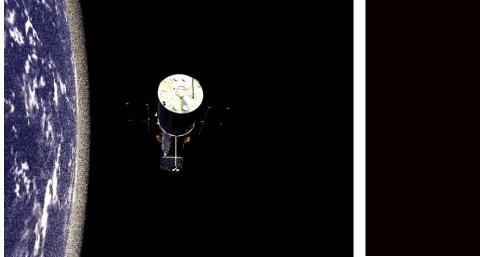
Specular Cost

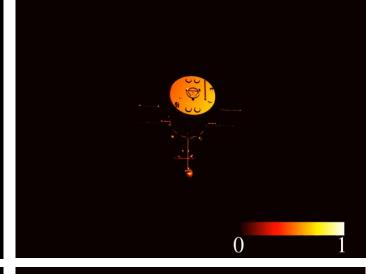
Before Optimisation



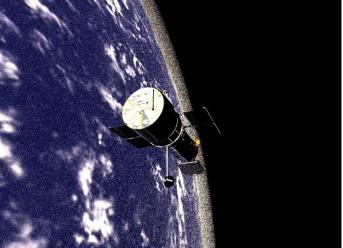


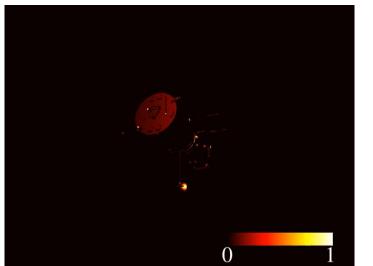
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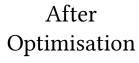


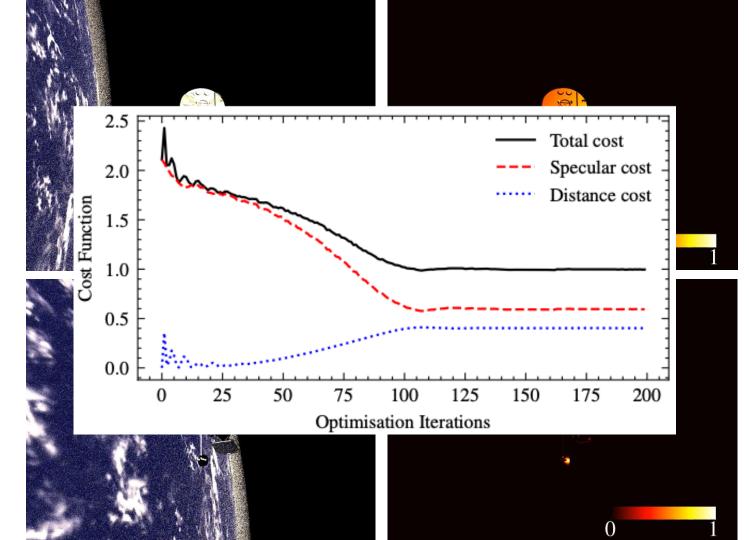
After Optimisation





Before Optimisation

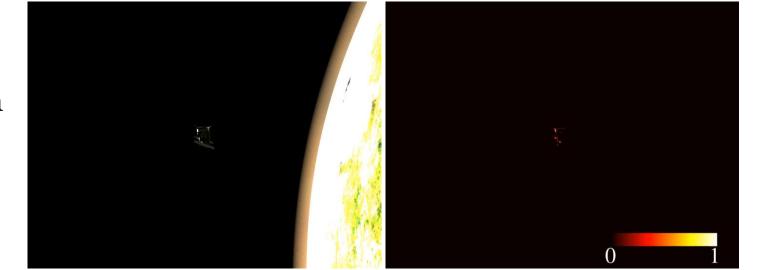




Pixel Saturation

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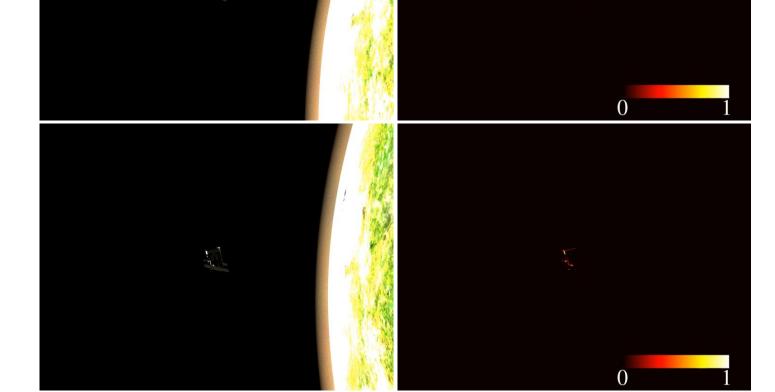
Before Optimisation



Pixel Saturation

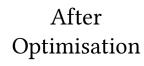
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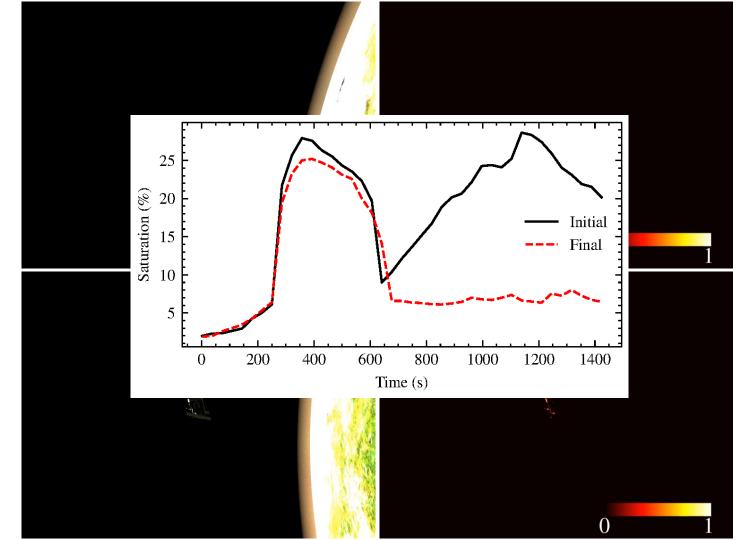
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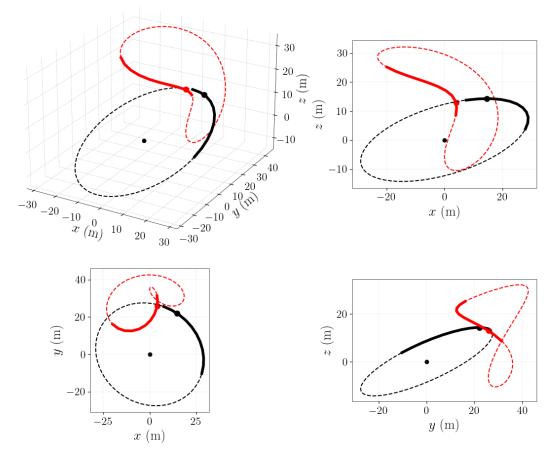


Pixel Saturation

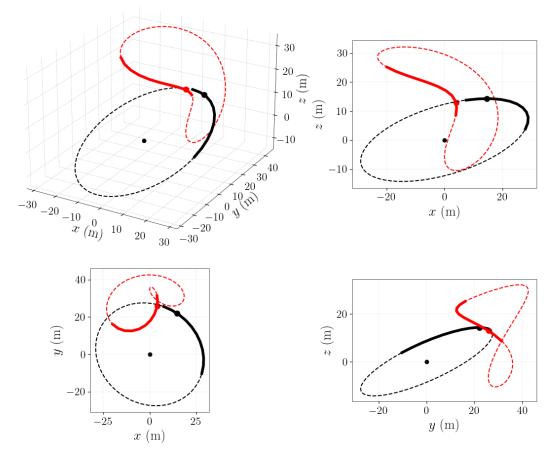
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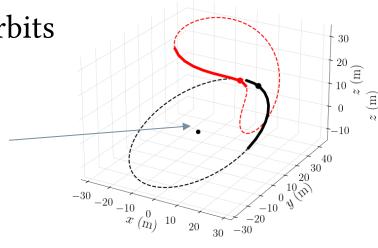


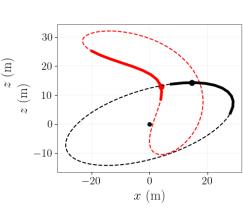


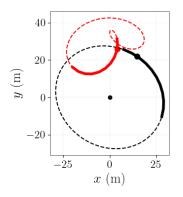


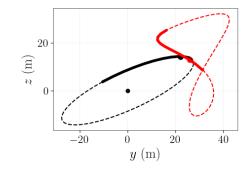


Target satellite (e.g., Hubble)











Comparing Relative Orbits 30 20 E 20 z (m) z10 -10Target satellite (e.g., Hubble) -10 $^{-30}_{-20}_{-10}_{-10}_{x \ (m)}^{0}_{10}$ -2020 Original relative orbit x (m) 20 30 20 20 y (m) z (m) -20-2020 40

-25

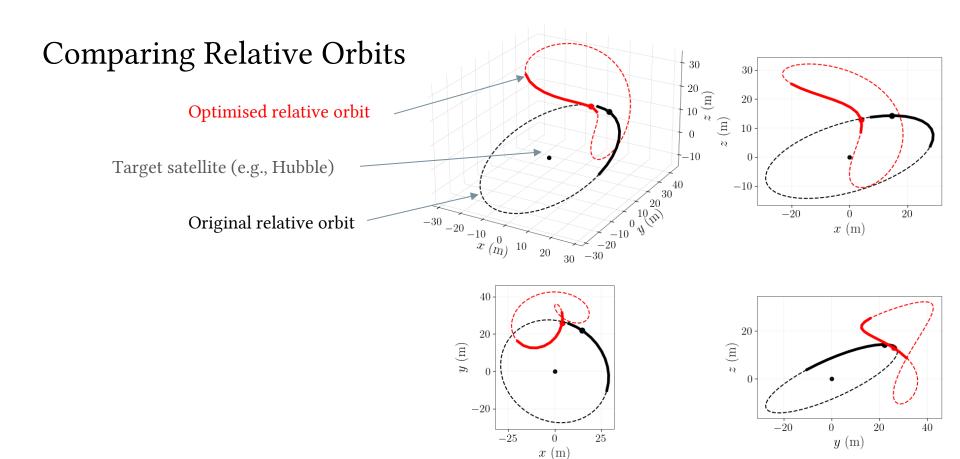
0

x (m)

25



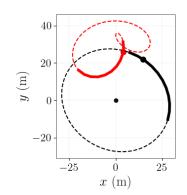
y (m)



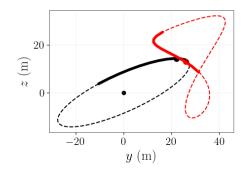


Comparing Relative Orbits 30 20 Optimised relative orbit 10 -10Target satellite (e.g., Hubble) 30 -10 $^{-30}_{-20}_{-10}_{-10}_{x(m)}^{0}_{10}$ -2020 Original relative orbit x (m)

Optimising the orbital elements to improve visual costs leads to new, specialised inspection trajectories in the relative frame.



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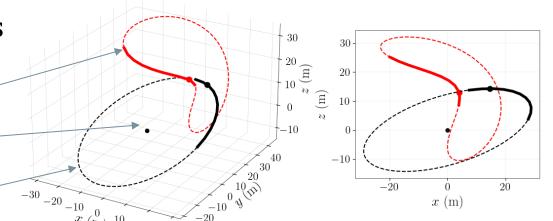




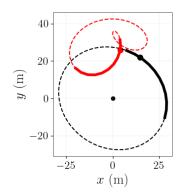
Optimised relative orbit

Target satellite (e.g., Hubble)

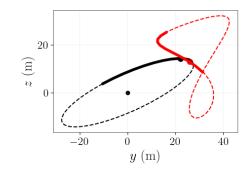
Original relative orbit



- Optimising the orbital elements to improve visual costs leads to new, specialised inspection trajectories in the relative frame.
- Deviation from the initial orbit is very small, so delta V costs are low.



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- Fully-differentiable vision-to-planning pipeline.
- Demonstrated capability in optimising inspection trajectories.



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Future Work



- Fully-differentiable vision-to-planning pipeline.
- Demonstrated capability in optimising inspection trajectories.

Future Work

- Integrate attitude dynamics & materials, add uncertainty quantification
- Apply to mission planning & operations.
- Other applications combining imaging & orbits (e.g., Earth observation tasking).



Q & A

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Supported by:











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