ME 280a: HW 7

April Novak

November 30, 2016

1 Introduction and Objectives

The purpose of this study is to describe the process for solving a 3-D diffusion-reaction equation with no time dependence, and then to describe the differences in the solution method once the problem is allowed to have time dependence. Then, a simple 1-D diffusion-reaction equation is solved as a transient case with Backward Euler (BE) time stepping.

2 Procedure

This section details the problem statement and mathematical method used for solving the problem.

2.1 Theoretical Problem Statement

Before deriving the governing equation, some preliminaries are necessary. The deformation gradient tensor \mathbf{F} is used to map between two different coordinate systems. One of these coordinates frames is defined by the coordinates x_i (the present coordinates), and the other by X_i (the reference coordinates):

$$\bar{dx} = \mathbf{F}\bar{d\xi} \tag{1}$$

The Jacobian \mathcal{J} is defined as the determinant of the deformation gradient tensor, and is required to transform integrals over the physical domain to the master element for application of quadrature rules:

$$\mathcal{J} \equiv \det \mathbf{F} \tag{2}$$

Finally, to have a physically meaningful transformation between two coordinate frames, the Jacobian must be positive. The diffusion-reaction equation to be solved in this assignment represents a balance between the time rate of change of the concentration in a control volume with the diffusion of mass into the control volume, a reaction that either produces or removes mass from the volume, and any sources of mass:

$$\dot{c} = \nabla \cdot (D\nabla c) - \tau c + f \tag{3}$$

where c is the concentration, an overhead dot represents differentiation in time, D the diffusion coefficient, τ the reaction rate frequency, and f the concentration source per unit time. This problem is to be solved to determine the concentration. In general, the diffusion coefficient can be an n-th order tensor, where n is the number of spatial dimensions, but for this assignment, is assumed to be a scalar, such that the medium is isotropic.

2.2 The Weak Form

The weak form of Eq. (3) is obtained by multiplying through by a test function v. Then, integrating over the body:

$$\int_{\Omega} \dot{c}v d\Omega = \int_{\Omega} \nabla \cdot (D\nabla c)v d\Omega - \int_{\Omega} \tau cv d\Omega + \int_{\Omega} fv d\Omega \tag{4}$$

where Ω is the entire domain. Applying the product rule to the diffusion term:

$$\int_{\Omega} \dot{c}v d\Omega = -\int_{\Omega} (\nabla v) D(\nabla c) d\Omega + \int_{\Gamma} D\nabla c \cdot \hat{n}v d\Gamma - \int_{\Omega} \tau cv d\Omega + \int_{\Omega} fv d\Omega$$
 (5)

where Γ is the boundary area of the domain Ω with unit normal vector \hat{n} . The boundary integral applies over the entire domain $\Gamma = \Gamma_q \cup \Gamma_d$, where Γ_q is the boundary on which the mass flux is specified (Neumann boundary condition) and Γ_d is the boundary on which the concentration is specified (Dirichlet boundary condition). Because the value of the mass flux is unknown on the Dirichlet boundaries, specifying the third term above on the boundaries would be difficult, and hence for simplicity, the shape functions are assumed to equal zero on the Dirichlet boundaries. With this simplification:

$$\int_{\Omega} \dot{c}v d\Omega = -\int_{\Omega} (\nabla v) D(\nabla c) d\Omega - \int_{\Gamma_g} qv d\Gamma - \int_{\Omega} \tau cv d\Omega + \int_{\Omega} fv d\Omega \tag{6}$$

where $q \equiv -D(\nabla c) \cdot \hat{n}$. Hence, the weak form can be stated as:

Find
$$c \in H^{c}(\Omega) \subset H^{1}(\Omega)$$
 so that $c|_{\Gamma_{d}} = \bar{c}$ and so that $\forall v \in H^{v}(\Omega) \subset H^{1}(\Omega), v|_{\Gamma_{d}} = 0$,
and for $q \in L^{2}(\Gamma_{q})$ and $f \in L^{2}(\Omega)$

$$\int_{\Omega} \dot{c}v d\Omega = -\int_{\Omega} (\nabla v) D(\nabla c) d\Omega - \int_{\Gamma_{q}} qv d\Gamma - \int_{\Omega} \tau cv d\Omega + \int_{\Omega} fv d\Omega$$
(7)

where \bar{c} is the vector of known concentrations on the displacement boundary. This weak form is more general than the strong form because it does not assume twice-differentiability of the concentration. The particular weighted residual method to be applied is the Bubnov-Galerkin method, where both the solution and the weight function are expanded in the same basis functions. Hence, both the displacement and the weight function are in $H^1(\Omega)$, the space necessary to ensure finite integrals in the weak form above.

The space $H^1(\Omega)$ is a Hilbert-space norm, where the 1 superscript indicates that it contains all functions whose highest finite derivative is the first derivative. This is the space from which the shape and weight functions must come because at most a first derivative is required in the weak form. For other applications, where for example, the highest derivative present in the weak form is a second derivative, then the weight and shape functions would need to be in $H^2(\Omega)$ in order for all integrals to remain finite. In other words, c is in $H^1(\Omega)$ if the following statement is true:

$$||c||_{H^1(\Omega)}^2 = \int_{\Omega} \left(\frac{\partial c}{\partial x}\right)^2 d\Omega + \int_{\Omega} u^2 d\Omega < \infty$$
 (8)

Because the flux and source also appear in the integrals in Eq. (7), there is a requirement on the space of functions from which they can inhabit. From the weak form, no differentiation of these functions is required, so they must be within $H^0(\Omega)$, sometimes referred to as $L^2(\Omega)$. The weak form in Eq. (7) is equivalent to the strong form provided that the solution is sufficiently differentiable that the higher derivatives required in the strong form are defined. Next, the specifics of the finite element implementation are given in order to specify the above to the finite element method.

2.2.1 The Finite Element Weak Form

This section covers the details regarding finite element implementation of Eq. (7). To implement this weak form, first the solution for the concentration c and the weight function v are expanded in a series of shape functions:

$$c^{h} = \sum_{j=1}^{n_{en}} a_{j} \phi_{j} = \mathbf{N}\Phi$$

$$v^{h} = \sum_{i=1}^{n_{en}} b_{i} \phi_{i} = \mathbf{N}\Psi$$

$$(9)$$

where a are the expansion coefficients to be solved for, ϕ are the expansion functions defined in the physical domain, and the h superscript indicates that this approximation occurs over each element, with n_{en} nodes per element. Because c is a scalar, there are n_{en} unknowns per element. The number and order of these shape functions determines the order of the finite element approximation. For convenience, the shape functions and unknowns are grouped into vectors:

$$\mathbf{N} \equiv \begin{bmatrix} \phi_1 & \phi_2 & \phi_3 & \phi_4 & \phi_5 & \phi_6 & \phi_7 & \phi_8 \end{bmatrix} \tag{10}$$

$$\Phi \equiv \begin{bmatrix} a_1 & a_2 & a_3 & a_4 & a_5 & a_6 & a_7 & a_8 \end{bmatrix}^T
\Psi \equiv \begin{bmatrix} b_1 & b_2 & b_3 & b_4 & b_5 & b_6 & b_7 & b_8 \end{bmatrix}^T$$
(11)

where a trilinear approximation has been assumed, such that in 3-D there are 8 nodes per element, with each node having one associated unknown. The gradient of the solution and the weight function are required in the weak form. This can be expressed conveniently as:

$$\nabla c = \frac{\partial c}{\partial x_1} \hat{x_1} + \frac{\partial c}{\partial x_2} \hat{x_2} + \frac{\partial c}{\partial x_3} \hat{x_3}$$

$$= \begin{bmatrix}
\frac{\partial \phi_1}{\partial x_1} & \frac{\partial \phi_2}{\partial x_1} & \frac{\partial \phi_3}{\partial x_1} & \frac{\partial \phi_4}{\partial x_1} & \frac{\partial \phi_5}{\partial x_1} & \frac{\partial \phi_6}{\partial x_1} & \frac{\partial \phi_7}{\partial x_1} & \frac{\partial \phi_8}{\partial x_1} \\
\frac{\partial \phi_1}{\partial x_2} & \frac{\partial \phi_2}{\partial x_2} & \frac{\partial \phi_3}{\partial x_2} & \frac{\partial \phi_4}{\partial x_2} & \frac{\partial \phi_5}{\partial x_2} & \frac{\partial \phi_6}{\partial x_2} & \frac{\partial \phi_7}{\partial x_2} & \frac{\partial \phi_8}{\partial x_2} \\
\frac{\partial \phi_1}{\partial x_3} & \frac{\partial \phi_2}{\partial x_3} & \frac{\partial \phi_3}{\partial x_3} & \frac{\partial \phi_4}{\partial x_3} & \frac{\partial \phi_5}{\partial x_3} & \frac{\partial \phi_6}{\partial x_3} & \frac{\partial \phi_7}{\partial x_3} & \frac{\partial \phi_8}{\partial x_3}
\end{bmatrix} \Phi$$

$$= \mathbf{B}\Phi$$
(12)

The gradient of the shape function is defined similarly. Inserting these approximations into the weak form:

$$\int_{\Omega} \mathbf{N}\dot{\Phi} \cdot (\mathbf{N}\Psi) d\Omega = -\int_{\Omega} D(\mathbf{B}\Psi) \cdot (\mathbf{B}\Phi) d\Omega - \int_{\Gamma_{\sigma}} q(\mathbf{N}\Psi) d\Gamma - \int_{\Omega} \tau(\mathbf{N}\Phi) \cdot (\mathbf{N}\Psi) d\Omega + \int_{\Omega} f(\mathbf{N}\Psi) d\Omega$$
(13)

where it has been assumed that the shape functions are not functions of time, such that the time derivative acts only on the vector of unknowns Φ . The dot product is defined such that $\mathbf{a} \cdot \mathbf{b} = \mathbf{a}^T \mathbf{b}$, and noting the equivalency between $\mathbf{N}\Psi$ and $(\mathbf{N}\Psi)^T$

$$\int_{\Omega} (\mathbf{N}\Psi)^T \mathbf{N} \dot{\Phi} d\Omega = -\int_{\Omega} (\mathbf{B}\Psi)^T D(\mathbf{B}\Phi) d\Omega - \int_{\Gamma_a} q(\mathbf{N}\Psi)^T d\Gamma - \int_{\Omega} \tau(\mathbf{N}\Psi)^T (\mathbf{N}\Phi) d\Omega + \int_{\Omega} f(\mathbf{N}\Psi)^T d\Omega \quad (14)$$

Then, because $(\mathbf{ab})^T = \mathbf{b}^T \mathbf{a}^T$:

$$\int_{\Omega} \Psi^{T} \mathbf{N}^{T} \mathbf{N} \dot{\Phi} d\Omega = -\int_{\Omega} \Psi^{T} \mathbf{B}^{T} D(\mathbf{B} \Phi) d\Omega - \int_{\Gamma_{q}} q \Psi^{T} \mathbf{N}^{T} d\Gamma - \int_{\Omega} \tau \Psi^{T} \mathbf{N}^{T} (\mathbf{N} \Phi) d\Omega + \int_{\Omega} f \Psi^{T} \mathbf{N}^{T} d\Omega \quad (15)$$

Then, because Ψ^T appears in every term, it can effectively be cancelled (the above could be rearranged so that Ψ^T multiplies a large term, and then if that entire integral must be zero, then the integrand must also be zero).

$$\int_{\Omega} \mathbf{N}^{T} \mathbf{N} \dot{\Phi} d\Omega + \int_{\Omega} \mathbf{B}^{T} D \mathbf{B} \Phi d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} \Phi d\Omega = -\int_{\Gamma_{d}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega$$
(16)

At this point, it is assumed that the system is in steady state. Then, the above reduces to the following:

$$\int_{\Omega} \mathbf{B}^{T} D \mathbf{B} \Phi d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} \Phi d\Omega = -\int_{\Gamma_{q}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega$$
(17)

For simplicity, the above terms can be defined as matrices:

$$\mathbf{K} \equiv \int_{\Omega} \mathbf{B}^{T} D \mathbf{B} \Phi d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} \Phi d\Omega$$

$$\mathbf{R} \equiv -\int_{\Gamma_{q}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega$$
(18)

to give the matrix system:

$$\mathbf{K}\Phi = \mathbf{R} \tag{19}$$

where Φ represents what is to be solved for over the domain. So, the weak form for the time-independent FE method (with Dirichlet boundary conditions applied using static condensation by removing them from the matrix system) is:

Find
$$\Phi \in \mathbf{H}^{\phi}(\Omega) \subset \mathbf{H}^{1}(\Omega)$$
 so that $\Phi|_{\Gamma_{d}} = \bar{\Phi}$ and so that $\forall \Psi \in \mathbf{H}^{\psi}(\Omega) \subset \mathbf{H}^{1}(\Omega), \Psi|_{\Gamma_{d}} = \mathbf{0}$,
and for $q \in L^{2}(\Gamma_{q})$ and $f \in L^{2}(\Omega)$
$$\int_{\Omega} \mathbf{B}^{T} D \mathbf{B} \Phi d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} \Phi d\Omega = -\int_{\Gamma_{q}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega$$
(20)

The definitions for all the terms that appear above have been given previously. Section 2.2.2 will show the weak form with the penalty method, while Section 2.3 will show how the above is applied element-by-element.

2.2.2 The Finite Element Weak Form - Penalty Method

The penalty method is a means by which to apply Dirichlet boundary conditions without the tedious need to separate rows and columns from the stiffness and loading vectors. From Eq. (20), the weak form for a finite element implementation requires that the weight functions are zero on essential boundaries. The penalty method relaxes this requirement, and adds a term to the weak form to account for violation of a Dirichlet boundary condition on a Dirichlet boundary. This method is widely-used, but is not strictly required to apply Dirichlet boundary conditions - the alternative of separating rows and columns containing known quantities, and subtracting from the load vector, can always be performed. The penalty method adds a term to the weak form of the form $P^* \int_{\Gamma_d} (\mathbf{N} \Psi) \cdot (\bar{c} - \mathbf{N} \Phi) d\Gamma$ into Eq. (20):

Find
$$\Phi \in \mathbf{H}^{\phi}(\Omega) \subset \mathbf{H}^{1}(\Omega)$$
 so that $\Phi|_{\Gamma_{d}} = \bar{\Phi}$ and so that $\forall \Psi \in \mathbf{H}^{\psi}(\Omega) \subset \mathbf{H}^{1}(\Omega)$
and for $q \in L^{2}(\Gamma_{q})$ and $f \in L^{2}(\Omega)$
$$\int_{\Omega} \mathbf{B}^{T} D \mathbf{B} \Phi d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} \Phi d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \mathbf{N} \Phi d\Gamma = -\int_{\Gamma_{q}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \bar{c} d\Gamma$$
(21)

where \bar{c} is a vector of known concentrations on the Dirichlet boundary and P^* represents something like a spring constant, and is a large, positive number. A high value of this artificial spring constant will apply a traction to "force" the displacement boundary to be satisfied on Γ_d , the displacement boundary. In other words, the penalty term represents a traction that enforces the Dirichlet boundary condition. This term, however, would never be applied if we still required $\Psi|_{\Gamma_d} = \mathbf{0}$, and so the kinematic restrictions on the weight functions are dropped, and they do not need to be zero on the Dirichlet boundaries. With this weak form, the matrix equation $\mathbf{K}\Phi = \mathbf{R}$ remains the same, but the contents of \mathbf{K} and \mathbf{R} change to:

$$\mathbf{K} \equiv \int_{\Omega} \mathbf{B}^{T} D \mathbf{B} d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \mathbf{N} d\Gamma$$

$$\mathbf{R} \equiv -\int_{\Gamma_{g}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \bar{c} d\Gamma$$
(22)

These additional penalty terms are only added for elements that are on Dirichlet boundaries, which is evident from the Γ_d appearing in the integral bounds.

2.3 Finite Element Implementation

2.3.1 Element-by-Element Matrices and Vectors

This section provides in explicit detail the forms of the element stiffness matrices and load vectors using the penalty method (the penalty terms could simply be dropped if needed, and hence the penalty method is discussed here to be as complete as possible). While Eq. (56) holds over the entire domain, the strength of the finite element method is that the integrals in Eq. (56) can be performed over each element, since the shape functions are a nodal basis such that they are only nonzero at a single node. Once assembling into the global stiffness matrix, this gives a sparse system. So, for an element e, the element stiffness matrix and load vector are:

$$\mathbf{K}^{e} \equiv \int_{\Omega_{e}} \mathbf{B}^{T} D \mathbf{B} d\Omega + \int_{\Omega_{e}} \tau \mathbf{N}^{T} \mathbf{N} d\Omega + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \mathbf{N} d\Gamma$$

$$\mathbf{R}^{e} \equiv -\int_{\Gamma_{q,e}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega_{e}} f \mathbf{N}^{T} d\Omega + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \bar{c} d\Gamma$$
(23)

where $\Gamma_{d,e}$ is the intersection of the boundary of element e with the Dirichlet boundary and $\Gamma_{q,e}$ is the intersection of the boundary of element e with the Neumann boundary. All of these integrals are performed in the master domain using quadrature rules. This master domain is a cube defined over $-1 \le \xi_1 \le 1, -1 \le \xi_2 \le 1, -1 \le \xi_3 \le 1$. In order to transform between the physical and master domain, a transformation rule is needed to map between x, y, z and ξ_1, ξ_2, ξ_3 . This transformation rule can take many forms, but a convenient one is to simply use the shape function expansion:

$$x_i = \sum_{j=1}^{n_{en}} X_{i,j} \phi_j(\xi_1, \xi_2, \xi_3)$$
 (24)

where the shape functions from here forward are implied to be defined over the master element, $X_{i,j}$ are the physical (real) coordinates, and i refers to the fact that the above expansion is assumed to apply equally for x_1 , x_2 and x_3 . This mapping, which uses the shape functions as the basis for the mapping, is called a parametric map. So, all the integrals in \mathbf{K}^e and \mathbf{R}^e are performed over a single element by transforming the integrals to the master domain. The integrals in the physical domain are with respect to $d\bar{x}$, and to transform them to the master domain, the deformation gradient tensor \mathbf{F} defined by Eq. (1) is used. Using the chain rule reveals the form of \mathbf{F} :

$$\begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} = \begin{bmatrix} \frac{\partial x_1}{\partial \xi_1} & \frac{\partial x_1}{\partial \xi_2} & \frac{\partial x_1}{\partial \xi_3} \\ \frac{\partial x_2}{\partial \xi_1} & \frac{\partial x_2}{\partial \xi_2} & \frac{\partial x_2}{\partial \xi_3} \\ \frac{\partial x_3}{\partial \xi_1} & \frac{\partial x_3}{\partial \xi_2} & \frac{\partial x_3}{\partial \xi_3} & \frac{\partial x_3}{\partial \xi_3} \end{bmatrix} \begin{bmatrix} \xi_1 \\ \xi_2 \\ \xi_3 \end{bmatrix}$$
 (25)

Or, in shorthand notation:

$$dx_i = F_{ii}\xi_i \tag{26}$$

where suffix notation is implied. The inverse relationship is:

$$d\xi_j = F_{ji}dx_i = F_{ij}^{-1}dx_i \tag{27}$$

The above transformation rule holds for the volume integral Jacobian - Nanson's formula must be used to transform the area integrals appearing in the element stiffness matrix and load vector to area integrals in the master domain. There is no reason that the Jacobian for the volume transformation be the same as that for the surface transformation, which is why more care must be taken here to use the correct transformation. In this case, area integrals are transformed by Nanson's rule:

$$dA_e = (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_e \tag{28}$$

where \hat{n} is a unit normal to the master element surface (must be determined for each element on the surface), \hat{N} the normal to the physical element surface (must be determined), \mathcal{J} is the Jacobian defined in Eq. (2), and **F** is the deformation gradient tensor defined in Eq. (25). Then, all the integrals over areas must set one of ξ_i to ± 1 to be consistent with the fact that on a surface, one of the master coordinates is held constant. So, in order to perform computations over each element, Eq. (29) becomes:

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \mathbf{B}^{T} D\mathbf{B} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \mathbf{N} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$\mathbf{R}^{e} \equiv -\int_{\Gamma_{q,e}} q \mathbf{N}^{T} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \bar{c} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$(29)$$

where it is implied in the area integrals that one of the spatial coordinates is set to ± 1 according to the orientation of the surface. To accurately perform the integration in the master element, everything within the integrand must also be converted to the master coordinate frame. Hence, **B** becomes:

$$\begin{bmatrix} \frac{\partial \xi_1}{\partial x} & \frac{\partial \xi_2}{\partial x} & \frac{\partial \xi_3}{\partial x} \\ \frac{\partial \xi_1}{\partial y} & \frac{\partial \xi_2}{\partial y} & \frac{\partial \xi_3}{\partial y} \\ \frac{\partial \xi_1}{\partial z} & \frac{\partial \xi_2}{\partial z} & \frac{\partial \xi_3}{\partial z} \end{bmatrix} \begin{bmatrix} \frac{\partial \phi_1}{\partial \xi_1} & \frac{\partial \phi_2}{\partial \xi_1} & \frac{\partial \phi_3}{\partial \xi_1} & \frac{\partial \phi_4}{\partial \xi_1} & \frac{\partial \phi_5}{\partial \xi_1} & \frac{\partial \phi_6}{\partial \xi_1} & \frac{\partial \phi_7}{\partial \xi_1} & \frac{\partial \phi_8}{\partial \xi_1} \\ \frac{\partial \phi_1}{\partial \xi_2} & \frac{\partial \phi_2}{\partial \xi_2} & \frac{\partial \phi_3}{\partial \xi_2} & \frac{\partial \phi_4}{\partial \xi_2} & \frac{\partial \phi_5}{\partial \xi_2} & \frac{\partial \phi_6}{\partial \xi_2} & \frac{\partial \phi_7}{\partial \xi_2} & \frac{\partial \phi_8}{\partial \xi_2} \\ \frac{\partial \phi_1}{\partial \xi_3} & \frac{\partial \phi_2}{\partial \xi_3} & \frac{\partial \phi_3}{\partial \xi_3} & \frac{\partial \phi_4}{\partial \xi_3} & \frac{\partial \phi_5}{\partial \xi_3} & \frac{\partial \phi_6}{\partial \xi_3} & \frac{\partial \phi_7}{\partial \xi_3} & \frac{\partial \phi_8}{\partial \xi_2} \\ \frac{\partial \phi_1}{\partial \xi_3} & \frac{\partial \phi_2}{\partial \xi_3} & \frac{\partial \phi_3}{\partial \xi_3} & \frac{\partial \phi_4}{\partial \xi_3} & \frac{\partial \phi_5}{\partial \xi_3} & \frac{\partial \phi_6}{\partial \xi_3} & \frac{\partial \phi_7}{\partial \xi_3} & \frac{\partial \phi_8}{\partial \xi_3} \\ \end{bmatrix} = \mathbf{B}$$
 (30)

As mentioned previously, it is implied that all shape functions appearing in this section are defined over the master domain. So, \mathbf{B} is replaced by $\mathbf{F}^{-1}\mathcal{B}$, where \mathcal{B} is the second matrix on the LHS above. This leads to the following element-wise stiffness matrix and load vector:

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \mathbf{N} (\partial \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$\mathbf{R}^{e} \equiv - \int_{\Gamma_{q,e}} q \mathbf{N}^{T} (\partial \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \bar{c} (\partial \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$(31)$$

The trilinear brick shape functions (in the master domain) are defined in Eq. (39). The unit normals are of length 3, and are oriented such that the correct dimensions are obtained in the matrix and load integrals above. All the integrals above are performed using quadrature. To integrate in higher than a single dimension using a Gaussian quadrature rule, simply apply the rule in each direction. So, in 1-D, where a single loop sums over all the quadrature points, three loops are needed to sum over the ξ_1, ξ_2, ξ_3 directions. For instance, the integrals above, in quadrature form, are:

$$\mathbf{K}^{e} = \sum_{q=1}^{g} \sum_{r=1}^{g} \sum_{s=1}^{g} w_{q} w_{g} w_{s} \left\{ (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| + \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| \right\} +$$

$$\sum_{r=1}^{g} \sum_{s=1}^{g} w_{g} w_{s} \left\{ P^{*} \mathbf{N}^{T} \mathbf{N} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} \right\}$$

$$\mathbf{R}^{e} \equiv \sum_{q=1}^{g} \sum_{r=1}^{g} \sum_{s=1}^{g} w_{q} w_{g} w_{s} f \mathbf{N}^{T} |\mathbf{F}| +$$

$$\sum_{s=1}^{g} \sum_{s=1}^{g} w_{g} w_{s} \left\{ -q \mathbf{N}^{T} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} + P^{*} \mathbf{N}^{T} \bar{c} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} \right\}$$

$$(32)$$

where g is the number of quadrature points, where the same quadrature rule has been assumed to be applied in each spatial dimension. It is implied that in the area integrals, $d\hat{A}_e$ refers to one of $d\xi_1 d\xi_2$, $d\xi_2 d\xi_3$, $d\xi_3 d\xi_1$, depending on the particular surface. To be as explicit as possible, there are essentially four types of elements. The element stiffness matrix and load vector for each of these possible combinations is shown below for completeness.

1. an element on the interior (no displacement or traction boundaries)

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3}$$

$$\mathbf{R}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3}$$
(33)

2. an element on the boundary with only displacement boundaries

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \mathbf{N} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$(34)$$

$$\mathbf{R}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \bar{c} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

3. an element on the boundary with only traction boundaries

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3}$$

$$\mathbf{R}^{e} \equiv -\int_{\Gamma_{q,e}} q \mathbf{N}^{T} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3}$$
(35)

4. an element on the boundary with both displacement and traction boundaries

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \mathbf{N} (\partial \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$\mathbf{R}^{e} \equiv - \int_{\Gamma_{q,e}} q \mathbf{N}^{T} (\partial \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \bar{c} (\partial \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$(36)$$

Care must be taken for determining if an element is on a Dirichlet boundary. If it is, then penalty terms must be added, and if not, the penalty terms are absent. So, whether or not the penalty method is used, there is some amount of bookkeeping required to record which nodes correspond to Dirichlet boundaries.

2.3.2 Global-Local Transformation

After all computations over the elements are complete, all the local stiffness matrices and load vectors must be organized into the global stiffness matrix and load vector. The placement of local matrices into the global matrix is performed using a connectivity matrix that relates the local node numbers to the global node numbers. A connectivity matrix is usually organized so that each row corresponds to a single element. Then, each column in that row refers to each local node in that element, and the information held in the connectivity matrix are the global node numbers relating to the local node numbers. For example, for a 2-D domain with four elements, and global node numbering beginning in the bottom left corner and moving up to the top right corner, the connectivity matrix (also called the location matrix **LM** in this document) has the following form:

$$\mathbf{LM} = \begin{bmatrix} 1 & 2 & 5 & 4 \\ 2 & 3 & 6 & 5 \\ 4 & 5 & 8 & 7 \\ 5 & 6 & 9 & 8 \end{bmatrix} \tag{37}$$

where the local nodes are numbered in a counterclockwise manner beginning from the bottom left node. Then, for example, the second row in the global stiffness matrix would be assembled as:

$$\mathbf{K}(2,:) = \begin{bmatrix} k_{2,1}^{e=1}, & k_{2,2}^{e=1} + k_{1,1}^{e=2}, & k_{1,2}^{e=2}, & k_{2,4}^{e=1}, & k_{1,4}^{e=2} + k_{2,3}^{e=1}, & k_{1,3}^{e=2}, & 0, & 0 \end{bmatrix}$$
(38)

2.4 Shape Functions

The shape functions for 3-D finite elements are a natural extension of the shape functions used in lower dimensions. Trilinear elements, or 3-D linear elements, are used for the remainder of this assignment. This choice of shape functions represents a nodal basis, since the shape functions go to zero at all nodes except for the node for which they are defined. The trilinear shape functions are:

$$\phi_{1}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 - \xi_{1})(1 - \xi_{2})(1 - \xi_{3})$$

$$\phi_{2}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 + \xi_{1})(1 - \xi_{2})(1 - \xi_{3})$$

$$\phi_{3}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 + \xi_{1})(1 + \xi_{2})(1 - \xi_{3})$$

$$\phi_{4}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 - \xi_{1})(1 + \xi_{2})(1 - \xi_{3})$$

$$\phi_{5}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 - \xi_{1})(1 - \xi_{2})(1 + \xi_{3})$$

$$\phi_{6}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 + \xi_{1})(1 - \xi_{2})(1 + \xi_{3})$$

$$\phi_{7}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 + \xi_{1})(1 + \xi_{2})(1 + \xi_{3})$$

$$\phi_{8}(\xi_{1}, \xi_{2}, \xi_{3}) = \frac{1}{8}(1 - \xi_{1})(1 + \xi_{2})(1 + \xi_{3})$$

3 Mesh Generator

This section discusses the mesh generator used to mesh the tubular "S" structure given in the assignment, and is largely a repeat of the mesh used for assignment 5, except that it has been modified to work for an unknown that is a scalar quantity, rather than a vector quantity. While the mesh is the same as that for assignment 5, the location matrix now does not need to have an associated matrix that relates the three unknowns for each node to their position in the global matrix (here, there is only one unknown per node). The meshing begins in each θ -chunk. For the circular cross-section structure, the x and y coordinates are related to each other by:

$$x^2 + y^2 = r \tag{40}$$

where r is the inner radius of the tubular structure. Each piece in the circumferential direction is defined according to θ , where $0 \le \theta \le 2\pi$ defines the "slice" parallel to \overrightarrow{e}_r . The generation of the mesh is based on determining the coordinates of each node. The first node is assigned to the first "slice" for $\theta = 0$. In the discussion of the mesh generator, "slice" refers to each plane for which there exists a hollowed annulus (that is meshed according to the number of layers and circumferential points). Θ is the angle referring to the circumferential angle, while θ refers to the angle in each slice. The overall algorithm for generating the coordinates for the mesh is as follows:

- 1. Begin with slice for $\Theta = 0$. Beginning then for $\theta = 0$, move counterclockwise around the first layer (inner surface of the tube). Increment θ in units of $2\pi/N_c$, and for each θ , assign the x and y coordinates according to Eq. (40).
- 2. After finishing the inner layer, increase r by dt, where dt is the thickness of each layer, to move to the next layer, then repeat step 1.
- 3. Repeat steps 1 and 2 until all layers in each slice have been meshed, where for each layer, dt is added.
- 4. Now that a slice has been meshed, repeat for all slices. This requires determining the x, y, z centers for each new slice. This is performed by sweeping through Θ . The y coordinate is y = 0 for all points, while the x coordinate continually increases moving from slice to slice, and the z coordinate is positive for the left half of the tube, and negative for the right half of the tube.
- 5. Now that all slices have been meshed, the mesh looks like as follows for $N_c = 8, N_\theta = 8, N_t = 3$.

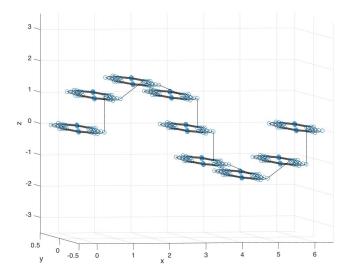


Figure 1. Mesh for $N_c = 8, N_{\theta} = 8, N_t = 3$ with no tilt to the slices. Lines connect each coordinate for better visuality.

The next step is to rotate the slices appropriately through the angle Θ for each slice so that a tubular structure is formed. Only the x and z coordinates must be modified. To perform the tilt, the following quantities are computed using trigonometry:

$$w = \sin(\pi/2 - \Theta)(r + dt)\cos(\theta)/\sin(pi/2)$$

$$h = w\sin(\Theta)$$

$$p = w\cos(\Theta)$$
(41)

To tilt the z-coordinate, for points in the first and fourth quadrant of each slice:

$$z_{new} = z - (-1)^{tube}h\tag{42}$$

And for points in the second and third quadrant of each slice:

$$z_{new} = z + (-1)^{tube}h (43)$$

where tube is a variable indicating whether or not the slice is in the left or right half of the tube. For the left half of the tube, tube = 1, and in the right half, tube = 2.

6. Then, tilt the x-coordinates. For points in the first and fourth quadrants of each slice:

$$x_{new} = x + p \tag{44}$$

And for points in the second and third quadrants of each slice:

$$x_{new} = x - p \tag{45}$$

7. Finally, tilt the slices that exactly align with the peak and valley of the tube (for odd numbers of N_{θ} , this would not be performed). For the slices that align with the peaks and for nodes in the first and fourth quadrants, adjust the z coordinates according to:

$$z_{new} = z - (r + dt)\cos(\theta) \tag{46}$$

And for points in the second and third quadrants:

$$z_{new} = z + (r + dt)\cos(\theta) \tag{47}$$

The x-coordinates are adjusted by simply setting all of them to the centroid coordinate for that slice.

This process is fairly complicated, and reveals why meshing software is so valuable. The process here is left fairly general that it can apply for any values of N_{θ} , N_{c} , N_{t} , but any slight change in the geometry completely invalidates the program. The final mesh for $N_{c}=8$, $N_{\theta}=8$, $N_{t}=3$ is shown below. This mesh is shown because the requested mesh with only $N_{c}=4$ is relatively difficult to perceive in a 3-D plot in Matlab, so the following plot better reveals the mesh. Lines are drawn between each coordinate.

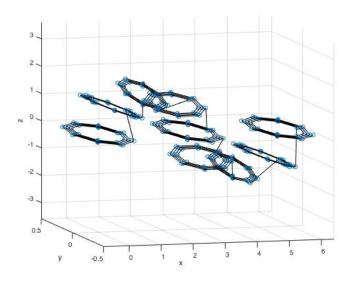


Figure 2. Mesh for $N_c = 8$, $N_{\theta} = 8$, $N_t = 3$. Lines connect each coordinate for better visuality.

The coarser mesh, for $N_c = 4$, $N_{\theta} = 8$, $N_t = 3$ is shown below, again with lines connecting each coordinate for better visibility.

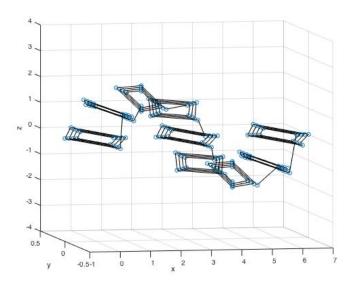


Figure 3. Mesh for $N_c = 4$, $N_{\theta} = 8$, $N_t = 3$. Lines connect each coordinate for better visuality.

The mesh in Fig. 3 is shown below without the lines connecting each coordinate.

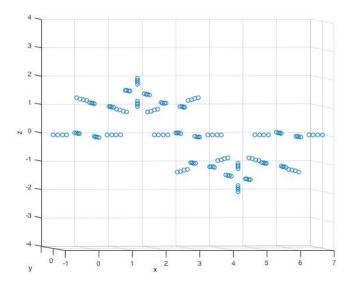


Figure 4. Mesh for $N_c = 4, N_{\theta} = 8, N_t = 3$.

Note that this assignment did not provide the dimensions of the tube, so I assumed that the inner radius of each arch was 1, the radius of the inner hole of the tube 0.3, and the thickness of the tube 0.2.

3.1 The Connectivity Matrix

In order for this mesh to be useful for finite element implementation, a connectivity function must be defined to relate the local node numbering to the global node numbering. The mesh generated numbers the global nodes according to the order in which they were generated. For instance, the first 8 nodes are in the inner layer of the first slice, the next 8 are in the second layer of the first slice, and so on for the first slice. Then, moving to the next Θ slice, the node numbering again begins on the inside of the tube and moves counterclockwise in layers until reaching the outside of the tube. This is shown schematically in the figures above by the black lines connecting the coordinates in the order in which the coordinates are generated.

The connectivity matrix is an $N \times 8$ matrix, where N is the total number of elements and 8 is the number of local nodes per element (linear elements are assumed). The local node numbering is performed according to a clockwise fashion. The following schematic shows the node numbering, where the left portion shows the front face, and the right portion shows the back face, all while looking at the front face (i.e. the back face is not written with the perspective of looking at the outward-facing portion of the back face).

$$\begin{array}{ccc}
4 - -3 & 8 - -7 \\
2 - -1 & 6 - -5
\end{array} \tag{48}$$

So, for each slice, the nodes on the face of each element can be determined using a numbering scheme that follows the order in which the nodes were defined. Beginning with $\theta = 0$, and moving counterclockwise, the local nodes are numbered, moving progressively outwards in the layers until reaching the last node for a particular slice.

There are $N_t \cdot N_c \cdot N_\theta$ total elements. For each slice, the nodes are numbered moving counterclockwise, beginning at the same node that is meshed first. After each layer is complete, the numbering moves to the next layer in the same fashion. Once an entire slice is complete, the next slice is also meshed. This defines only the *frontal* node numberings shown in the above equation. For example, for $N_\theta = 2, N_t = 3, N_c = 4$, the connectivity matrix **LM** looks like the following *before* the nodes on the backs of the first 12 elements are related to the nodes on the fronts of the next 12 elements.

```
0
                                        0
                    5
                                 0
               3
                    6
                         7
                                    0
                                        0
                                 0
          3
               4
                    7
                         8
                                 0
                                    0
                                        0
          4
                         5
               1
                                    0
          5
               6
                    9
                        10
                                    0
                             0
                                 0
                                        0
          6
               7
                   10
                        11
                             0
                                    0
                                        0
          7
                        12
               8
                   11
                             0
                                 0
                                    0
                                        0
          8
               5
                   12
                         9
                             0
                                 0
                                    0
                                        0
          9
                   13
                                    0
              10
                        14
                             0
                                 0
                                        0
         10
              11
                   14
                        15
                                 0
                                    0
                             0
                                        0
         11
              12
                   15
                        16
                             0
                                 0
                                    0
         12
               9
                   16
                        13
                                 0
                                    0
                             0
                                        0
                   21
                        22
                                    0
         17
              18
                             0
                                 0
                                        0
              19
         18
                   22
                        23
                             0
                                 0
                                    0
                                        0
         19
              20
                   23
                        24
                             0
                                 0
                                    0
                                        0
         20
              17
                   24
                        21
                             0
                                 0
                                    0
                                        0
         21
              22
                   25
                        26
                                    0
                             0
                                 0
         22
              23
                   26
                        27
                                 0
                                    0
                             0
LM =
                                                                           (49)
         23
              24
                   27
                        28
                                 0
                                    0
         24
              21
                   28
                        25
                             0
                                 0
                                    0
                                        0
         25
              26
                   29
                        30
                             0
                                 0
                                    0
         26
              27
                   30
                        31
                                 0
                                    0
                             0
                                        0
         27
              28
                   31
                        32
                             0
                                 0
                                    0
         28
              25
                   32
                        29
                             0
                                 0
                                    0
                                        0
         33
              34
                   37
                        38
                                 0
                                    0
                             0
                                        0
         34
              35
                        39
                                    0
                   38
                                 0
         35
              36
                   39
                        40
                             0
                                 0
                                    0
                                        0
         36
              33
                   40
                        37
                             0
                                 0
                                    0
                                        0
         37
              38
                   41
                        42
                                 0
                                    0
                                        0
                             0
         38
              39
                   42
                        43
                             0
                                 0
                                    0
                                        0
         39
                   43
                                    0
              40
                        44
                             0
                                 0
                                        0
         40
              37
                   44
                                 0
                                    0
                                        0
                        41
                             0
         41
              42
                   45
                        46
                                 0
                                    0
                                        0
                             0
         42
              43
                   46
                        47
                                 0
                                    0
                                        0
         43
                                 0
                                    0
              44
                   47
                        48
                             0
                                        0
         44
              41
                                 0
                                    0
                   48
                        45
                             0
```

This is not the final form for the connectivity matrix (a.k.a. location matrix). Because the slices lay exactly on top of one another, the front nodes of the second slice are exactly the back nodes on the previous slice. With this knowledge, the back nodes for each element can be assigned based on the frontal nodes of the following slice. Then, the last $N_c N_t$ rows in the location matrix above can be deleted, since they refer to the frontal nodes of a slice that does not technically exist (there are only 2 slices, but 3 planes defining those slices). With this information, the final form of the location matrix becomes, for $N_{\theta} = 2$, $N_t = 3$, $N_c = 4$ for example:

$$\mathbf{LM} = \begin{bmatrix} 1 & 2 & 5 & 6 & 17 & 18 & 21 & 22 \\ 2 & 3 & 6 & 7 & 18 & 19 & 22 & 23 \\ 3 & 4 & 7 & 8 & 19 & 20 & 23 & 24 \\ 4 & 1 & 8 & 5 & 20 & 17 & 24 & 21 \\ 5 & 6 & 9 & 10 & 21 & 22 & 25 & 26 \\ 6 & 7 & 10 & 11 & 22 & 23 & 26 & 27 \\ 7 & 8 & 11 & 12 & 23 & 24 & 27 & 28 \\ 8 & 5 & 12 & 9 & 24 & 21 & 28 & 25 \\ 9 & 10 & 13 & 14 & 25 & 26 & 29 & 30 \\ 10 & 11 & 14 & 15 & 26 & 27 & 30 & 31 \\ 11 & 12 & 15 & 16 & 27 & 28 & 31 & 32 \\ 12 & 9 & 16 & 13 & 28 & 25 & 32 & 29 \\ 17 & 18 & 21 & 22 & 33 & 34 & 37 & 38 \\ 18 & 19 & 22 & 23 & 34 & 35 & 38 & 39 \\ 19 & 20 & 23 & 24 & 35 & 36 & 39 & 40 \\ 20 & 17 & 24 & 21 & 36 & 33 & 40 & 37 \\ 21 & 22 & 25 & 26 & 37 & 38 & 41 & 42 \\ 22 & 23 & 26 & 27 & 38 & 39 & 42 & 43 \\ 23 & 24 & 27 & 28 & 39 & 40 & 43 & 44 \\ 24 & 21 & 28 & 25 & 40 & 37 & 44 & 41 \\ 25 & 26 & 29 & 30 & 41 & 42 & 45 & 46 \\ 26 & 27 & 30 & 31 & 42 & 43 & 46 & 47 \\ 27 & 28 & 31 & 32 & 43 & 44 & 47 & 48 \\ 28 & 25 & 32 & 29 & 44 & 41 & 48 & 45 \end{bmatrix}$$

Each row in the location matrix corresponds to an elements, and each column to a local node number, so that LM(1,4) indicates the global node number of local node number 4 in element 1. This method is extended to the case for $N_{\theta} = 8$, $N_c = 4$, $N_t = 3$, where the purpose of the previous discussion for a fewer number of circumferential elements was simply to illustrate the process by which the location matrix is generated. So, for the problem statement in this homework assignment $(N_{\theta} = 8, N_c = 4, N_t = 3)$:

$$\mathbf{LM} = \begin{bmatrix} LM_1 \\ LM_2 \end{bmatrix} \tag{51}$$

where, in order to be able to print the matrix, the following components are defined to simply be stacked on top of each other as in Eq. (51).

```
2
                            17
                                18
                                     21
                                          22
                   5
                        6
          2
                                          23
              3
                        7
                                     22
                   6
                            18
                                19
         3
                   7
                        8
                                20
                                     23
                                          24
              4
                            19
         4
                   8
                        5
                            20
                                 17
                                     24
                                          21
         5
              6
                   9
                       10
                            21
                                 22
                                     25
                                          26
          6
              7
                                23
                                     26
                                          27
                  10
                       11
                            22
          7
              8
                  11
                       12
                            23
                                24
                                     27
                                          28
                                          25
          8
              5
                  12
                        9
                            24
                                21
                                     28
         9
                  13
                            25
                                26
                                     29
                                          30
              10
                       14
         10
              11
                  14
                       15
                            26
                                27
                                     30
                                          31
              12
                  15
                                28
                                          32
         11
                       16
                            27
                                     31
         12
              9
                  16
                       13
                            28
                                25
                                     32
                                          29
                  21
         17
              18
                       22
                            33
                                     37
                                          38
                                34
         18
             19
                  22
                       23
                            34
                                35
                                     38
                                          39
         19
             20
                  23
                            35
                                     39
                       24
                                36
                                          40
         20
             17
                  24
                       21
                            36
                                33
                                     40
                                          37
         21
              22
                  25
                       26
                            37
                                          42
                                38
                                     41
         22
             23
                  26
                       27
                            38
                                39
                                     42
                                          43
         23
             24
                  27
                       28
                            39
                                40
                                     43
                                          44
         24
             21
                  28
                       25
                            40
                                37
                                     44
                                          41
         25
              26
                  29
                       30
                            41
                                42
                                     45
                                          46
         26
             27
                  30
                       31
                            42
                                43
                                     46
                                          47
         27
              28
                  31
                       32
                            43
                                44
                                     47
                                          48
         28
              25
                  32
                       29
                            44
                                41
                                     48
                                          45
LM_1 =
         33
              34
                  37
                       38
                            49
                                50
                                     53
                                          54
              35
         34
                  38
                       39
                            50
                                51
                                     54
                                          55
         35
              36
                  39
                       40
                            51
                                52
                                     55
                                          56
         36
             33
                  40
                       37
                            52
                                49
                                     56
                                          53
         37
              38
                  41
                       42
                            53
                                54
                                     57
                                          58
         38
             39
                  42
                       43
                            54
                                55
                                     58
                                          59
         39
             40
                  43
                                     59
                                          60
                       44
                            55
                                56
         40
             37
                  44
                       41
                            56
                                53
                                     60
                                          57
         41
              42
                  45
                                          62
                       46
                            57
                                58
                                     61
         42
              43
                  46
                       47
                            58
                                59
                                     62
                                          63
         43
              44
                  47
                       48
                            59
                                60
                                     63
                                          64
         44
              41
                  48
                            60
                                     64
                                          61
                       45
                                57
         49
             50
                  53
                       54
                            65
                                66
                                     69
                                          70
         50
             51
                  54
                                     70
                                          71
                       55
                            66
                                67
         51
             52
                  55
                            67
                                          72
                       56
                                68
                                     71
         52
             49
                  56
                       53
                            68
                                65
                                     72
                                          69
         53
             54
                  57
                       58
                            69
                                70
                                     73
                                          74
         54
             55
                  58
                       59
                            70
                                71
                                     74
                                          75
         55
             56
                  59
                                72
                                     75
                       60
                            71
                                          76
         56
             53
                  60
                       57
                            72
                                69
                                     76
                                          73
         57
                            73
                                74
                                     77
             58
                  61
                       62
                                          78
         58
             59
                  62
                       63
                            74
                                75
                                     78
                                          79
         59
              60
                  63
                            75
                                76
                                     79
                                          80
                       64
         60
              57
                  64
                       61
                            76
                                73
                                     80
                                          77
         65
             66
                  69
                       70
                                82
                            81
                                     85
                                          86
```

(52)

```
66
                 67
                       70
                             71
                                                      87
                                   82
                                          83
                                                86
          67
                 68
                       71
                             72
                                   83
                                          84
                                                87
                                                      88
          68
                 65
                       72
                             69
                                   84
                                          81
                                                88
                                                      85
          69
                 70
                       73
                             74
                                   85
                                          86
                                                89
                                                      90
          70
                 71
                       74
                             75
                                   86
                                          87
                                                90
                                                      91
          71
                 72
                       75
                                                91
                                                      92
                             76
                                   87
                                          88
          72
                 69
                       76
                             73
                                   88
                                          85
                                                92
                                                      89
          73
                 74
                       77
                             78
                                   89
                                          90
                                                93
                                                      94
                 75
          74
                       78
                             79
                                   90
                                          91
                                                94
                                                      95
                 76
                       79
                                   91
                                                95
          75
                             80
                                          92
                                                      96
          76
                 73
                       80
                             77
                                   92
                                          89
                                                96
                                                      93
          81
                 82
                       85
                             86
                                   97
                                          98
                                               101
                                                      102
          82
                 83
                       86
                             87
                                   98
                                         99
                                               102
                                                     103
          83
                 84
                       87
                                   99
                                         100
                                               103
                                                     104
                             88
          84
                 81
                       88
                             85
                                   100
                                         97
                                               104
                                                     101
          85
                 86
                       89
                             90
                                   101
                                         102
                                               105
                                                     106
          86
                 87
                       90
                             91
                                   102
                                         103
                                               106
                                                     107
                                               107
          87
                 88
                       91
                             92
                                   103
                                         104
                                                     108
          88
                 85
                       92
                             89
                                   104
                                         101
                                               108
                                                     105
          89
                 90
                       93
                                   105
                                         106
                                               109
                             94
                                                     110
          90
                 91
                       94
                             95
                                   106
                                         107
                                               110
                                                     111
          91
                 92
                       95
                             96
                                   107
                                         108
                                               111
                                                     112
          92
                 89
                       96
                             93
                                         105
                                               112
                                                     109
                                   108
LM_2 =
          97
                 98
                      101
                             102
                                   113
                                         114
                                               117
                                                     118
                                                                                   (53)
                 99
                                               118
          98
                      102
                             103
                                   114
                                         115
                                                     119
          99
                100
                      103
                                   115
                                         116
                                               119
                                                     120
                             104
                97
          100
                      104
                             101
                                   116
                                         113
                                               120
                                                     117
          101
                102
                      105
                             106
                                               121
                                                     122
                                   117
                                         118
                                               122
          102
                103
                      106
                             107
                                   118
                                         119
                                                     123
                104
                                         120
                                               123
          103
                      107
                             108
                                   119
                                                     124
                101
                      108
                                   120
                                         117
                                               124
                                                     121
          104
                             105
                                               125
          105
                106
                      109
                            110
                                   121
                                         122
                                                     126
                107
                            111
                                   122
                                         123
                                               126
          106
                      110
                                                     127
          107
                108
                      111
                             112
                                   123
                                         124
                                               127
                                                     128
          108
                105
                      112
                             109
                                   124
                                         121
                                               128
                                                     125
          113
                114
                      117
                             118
                                   129
                                         130
                                               133
                                                     134
                                               134
                115
                      118
                            119
                                   130
                                         131
                                                     135
          114
                116
                      119
                             120
                                   131
                                         132
                                               135
                                                     136
          115
          116
                113
                      120
                            117
                                   132
                                         129
                                               136
                                                     133
          117
                118
                      121
                            122
                                   133
                                         134
                                               137
                                                     138
                119
                      122
                             123
                                   134
                                         135
                                               138
                                                     139
          118
          119
                120
                      123
                             124
                                   135
                                         136
                                               139
                                                     140
                      124
                             121
                                   136
          120
                117
                                         133
                                               140
                                                     137
                122
                      125
                             126
          121
                                   137
                                         138
                                               141
                                                     142
          122
                123
                      126
                             127
                                   138
                                         139
                                               142
                                                     143
          123
                124
                      127
                             128
                                   139
                                         140
                                               143
                                                     144
          124
                121
                      128
                            125
                                                     141
                                   140
                                         137
                                               144
```

4 Computational Cost

The cost of a 3-D finite element simulation of a scalar-valued equation such as the diffusion-reaction equation results in element stiffness matrices of size 8×8 when using trilinear shape functions. For a cubical mesh, with $M \times M$ elements, there are $(M+1)^3$ total unknowns in the mesh, so the global stiffness matrix

is of size $((M+1)^3) \times ((M+1)^3)$. If the symmetry of the local stiffness matrices is taken into account (they will always be symmetric for linear differential equations so long as the Bubnov-Galerkin approach is used), then the required storage per element drops from 64 to 28.

If the Conjugate Gradient (CG) method is used to solve the matrix system $\mathbf{Ka} = \mathbf{R}$, then this represents repeated application of a 8×8 matrix and a 8×1 vector, which is an $\mathcal{O}(N)$ operation. This is performed for each iteration, so the cost for the CG method scales as $\mathcal{O}(IN)$, where I is the number of iterations needed to reach a particular tolerance in the solve.

The cost of a solve refers to both the storage required and the number of floating point operations required. There are three ways to perform the storage for a mesh consisting of $M \times M \times M$ linear elements:

- 1. Direct storage all zeros are stored, and no shortcuts are made by saving element-by-element. This requires $((M+1)^3) \times ((M+1)^3) \approx M^6$
- 2. Element-by-element storage: no zeros are stored. This requires $8 \times 8 \times M^3 = 64 M^3$
- 3. Element-by-element storage, taking advantage of the symmetry of the element stiffness matrices. This requires $28M^3$

So, the storage scales cubically when using element-by-element storage as opposed to direct storage. The number of floating point operations is also significantly reduced when using an iterative solver. The total number of unknowns is $(M+1)^3$, so the ratio of the direct solver (Gaussian elimination) to the CG method is:

$$\frac{\mathcal{O}(N^3)}{I\mathcal{O}(N)} = \frac{\left((M+1)^3\right)^2}{I} = \frac{(M+1)^6}{I} \tag{54}$$

where $N=(M+1)^3$ is the total number of unknowns. For $N_t \times N_c \times N_\theta$ elements, there are $(N_t+1)N_c(N_\theta+1)$ total nodes. So, if there are 3 unknowns per node, and if a CG solver takes IO(N) operations, then the total number of operations required is:

Operations with CG method =
$$I(N_t + 1)N_c(N_\theta + 1)$$
 (55)

where I is the number of iterations.

5 Time-Dependent Implementation

This section will describe the FE solution to the transient problem in Eq. (3), beginning from Eq. (56). The time term that was last included in Eq. (16) can simply be added back into the stiffness matrix portion:

Find
$$\Phi \in \mathbf{H}^{\phi}(\Omega) \subset \mathbf{H}^{1}(\Omega)$$
 so that $\Phi|_{\Gamma_{d}} = \bar{\Phi}, \Phi(t = 0) = \bar{c_{0}}$
and so that $\forall \ \Psi \in \mathbf{H}^{\psi}(\Omega) \subset \mathbf{H}^{1}(\Omega)$ and for $q \in L^{2}(\Gamma_{q})$ and $f \in L^{2}(\Omega)$
$$\int_{\Omega} \mathbf{N}^{T} \mathbf{N} \dot{\Phi} d\Omega + \int_{\Omega} \mathbf{B}^{T} D \mathbf{B} \Phi d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} \Phi d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \mathbf{N} \Phi d\Gamma =$$
$$- \int_{\Gamma_{q}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \bar{c} d\Gamma$$
(56)

where now the only difference is that the time term is included in the weak form, and included in the weak form is the requirement that the solution satisfy the initial condition. All computations occur over each individual element, and the weak form above can be transformed to integrals in the master domain following the procedure described previously. Then, the matrix system to be solved becomes:

$$\mathbf{M}\dot{\Phi} + \mathbf{K}\Phi = \mathbf{R} \tag{57}$$

where these matrices are defined as follows over the entire domain:

$$\mathbf{M} \equiv \int_{\Omega} \mathbf{N}^{T} \mathbf{N} d\Omega$$

$$\mathbf{K} \equiv \int_{\Omega} \mathbf{B}^{T} D \mathbf{B} d\Omega + \int_{\Omega} \tau \mathbf{N}^{T} \mathbf{N} d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \mathbf{N} d\Gamma$$

$$\mathbf{R} \equiv -\int_{\Gamma_{q}} q \mathbf{N}^{T} d\Gamma + \int_{\Omega} f \mathbf{N}^{T} d\Omega + P^{*} \int_{\Gamma_{d}} \mathbf{N}^{T} \bar{c} d\Gamma$$
(58)

Over each element, these matrices are as follows:

$$\mathbf{M}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3}$$

$$\mathbf{K}^{e} \equiv \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} (\mathbf{F}^{-1} \mathcal{B})^{T} D(\mathbf{F}^{-1} \mathcal{B}) |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} + \int_{-1}^{1} \int_{-1}^{1} \int_{-1}^{1} \tau \mathbf{N}^{T} \mathbf{N} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} +$$

$$P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \mathbf{N} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$\mathbf{R}^{e} \equiv - \int_{\Gamma_{q,e}} q \mathbf{N}^{T} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e} + \int_{-1}^{1} \int_{-1}^{1} f \mathbf{N}^{T} |\mathbf{F}| d\xi_{1} d\xi_{2} d\xi_{3} +$$

$$P^{*} \int_{\Gamma_{d,e}} \mathbf{N}^{T} \bar{c} (\mathcal{J} \mathbf{F}^{-T} \cdot \hat{N}) \cdot \hat{n} d\hat{A}_{e}$$

$$(59)$$

As can be seen, the element stiffness matrix and load vectors do not change, but rather a mass matrix \mathbf{M} must now be computed for each element. The local mass matrices are assembled into this matrix by the same procedure as described in Section 2.3.2, and the location matrix that is used to populate the global stiffness matrix is equivalently used to populate the global mass matrix. However, the procedure up until now is not fully complete, since an approximation is needed to describe how the value of the FE solution changes in time, since separable space-time finite elements are used. Taylor series are used to derive several common finite difference approximations to the first derivative that appears in this equation. Expanding a quantity a in a Taylor series about $t + \Delta t$:

$$a(t + \Delta t) \approx a(t) + \frac{da}{dt} \Delta t + \cdots$$
 (60)

If the higher order terms indicates by the ellipses above are neglected, an approximation to the first derivative is obtained:

$$\frac{da}{dt} \approx \frac{a(t + \Delta t) - a(t)}{\Delta t} \tag{61}$$

Because the highest neglected term is of order Δt^2 , this discretization scheme is a first-order scheme. The above discretization scheme is termed an Euler method, and can be either implicit or explicit, depending on the point in time at which da/dt is to be evaluated. If da/dt is evaluated at $t+\Delta t$, then the scheme is implicit, and is unconditionally stable. If evaluated at t, then the scheme is explicit, which is conditionally stable for sufficiently small time steps, which may be a large limitation on the computational cost of a simulation. Using this first-order approximation to the first derivative in the governing equations, some options for time discretization are:

$$\mathbf{M} \frac{\Phi^{k+1} - \Phi^{k}}{\Delta t} + \mathbf{K} \Phi^{k} = \mathbf{R} \qquad \text{explicit}$$

$$\mathbf{M} \frac{\Phi^{k+1} - \Phi^{k}}{\Delta t} + \mathbf{K} \Phi^{k+1} = \mathbf{R} \qquad \text{implicit}$$
(62)

where k indicates the time step index. The implicit method is to be used in this assignment, so rearranging the second line above in order to isolate Φ^{k+1}

$$\left(\frac{1}{\Delta t}\mathbf{M} + \mathbf{K}\right)\phi^{k+1} = R + \frac{1}{\Delta t}\mathbf{M}\Phi^k$$
(63)

This allows a time-marching scheme, where it is assumed that Φ^k is always known because an initial condition is given. So, the general algorithm is a small modification to the assignment developed for the first homework. First, apply the initial condition to find Φ^1 . Then, solve the above equation in a loop to find the Φ corresponding to each subsequent time step, until the end of the simulation. The above equation is solved for the global matrices, so for each time step, the entire finite element implementation is performed.

For the 1-D problem for which this is to be applied, the element mass and stiffness matrices and the element load vector simplify considerably. For simplicity, this problem is not solved with the penalty method (static condensation, or the process of removing Dirichlet nodes from the matrix system and load vector, is performed instead). For 1-D and with no forcing term, Eq. (59) simplifies to:

$$\mathbf{M}_{ij}^{e} \equiv \int_{-1}^{1} \phi_{i} \phi_{j} \partial d\xi$$

$$\mathbf{K}_{ij}^{e} \equiv \int_{-1}^{1} \frac{d\phi_{i}}{d\xi} D \frac{d\phi_{j}}{d\xi} \frac{1}{\vartheta} d\xi + \int_{-1}^{1} \tau \phi_{i} \phi_{j} \partial d\xi$$

$$\mathbf{R}_{j}^{e} \equiv -q\phi_{j} \Big|_{0}^{L}$$
(64)

where the Jacobian $\mathcal{J} = dx/d\xi$. When not using the penalty method, the Dirichlet nodes must be removed from the solve, or else a singular system will result. So, the above is solved as follows, where "u" indicates a node whose value is unknown (not a Dirichlet node), while "k" indicates a node whose value is known (a Dirichlet node):

$$\left(\frac{1}{\Delta t}M_{uu} + K_{uu}\right)\phi_u^{k+1} + \left(\frac{1}{\Delta t}M_{uk} + K_{uk}\right)\phi_k^{k+1} = R_u + \frac{1}{\Delta t}\left(M_{uu}\phi_u^k + M_{uk}\phi_k^k\right)$$
(65)

where the k that appears in the superscript indicates iteration number. Rearranging the above gives the actual equation that is solved at each time step:

$$\phi_u^{k+1} = \left(\frac{1}{\Delta t}M_{uu} + K_{uu}\right)^{-1} \left[R_u + \frac{1}{\Delta t}\left(M_{uu}\phi_u^k + M_{uk}\phi_k^k\right) - \left(\frac{1}{\Delta t}M_{uk} + K_{uk}\right)\phi_k^{k+1}\right]$$
(66)

Because the material properties are constant in time, and because separable space-time finite elements are used, the matrices above only need to be computed once, and then simply are re-applied in a loop until reaching the final time.

6 Results

This section shows the results for the 1-D diffusion-reaction equation for an end time of 6.5e3 seconds. The three plots that follow show the time evolution. Fig. 5 shows the results for 100 time steps, Fig. 6 for 1000 time steps, and Fig. 7 for 10000 time steps. The time progression is shown beginning with red color early in the transient and progressing to purple/blue towards the end of the time period. The blue line shows the initial condition, and the black line shows the solution at the last time step (the final time solution). Also plotted is the flux -Ddc/dx as a function of time to show how the Neumann boundary condition is satisfied.

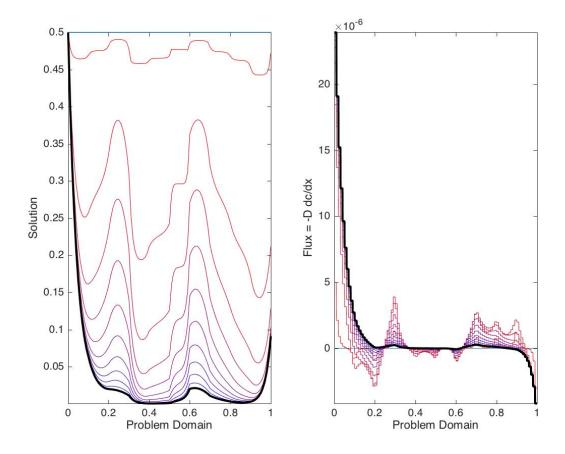


Figure 5. Time-dependent solution for concentration (red = early in transient, purple/blue = late in transient, black = at final time step) for 100 time steps for every 10 time steps.

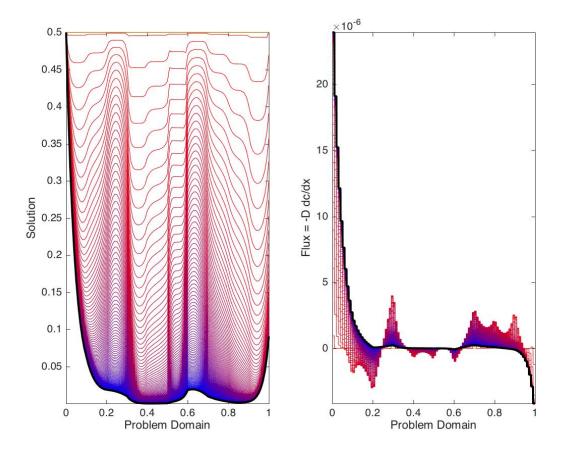


Figure 6. Time-dependent solution for concentration (red = early in transient, purple/blue = late in transient, black = at final time step) for 1000 time steps for every 10 time steps.

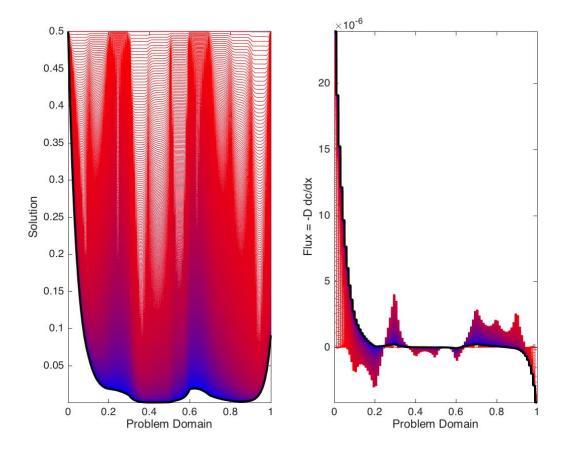


Figure 7. Time-dependent solution for concentration (red = early in transient, purple/blue = late in transient, black = at final time step) for 10000 time steps for every 10 time steps.

Because there are so many time steps for the 10000 time step case, Fig. 7 is re-plotted every 100 times steps instead of every 10 time steps in Fig. 8.

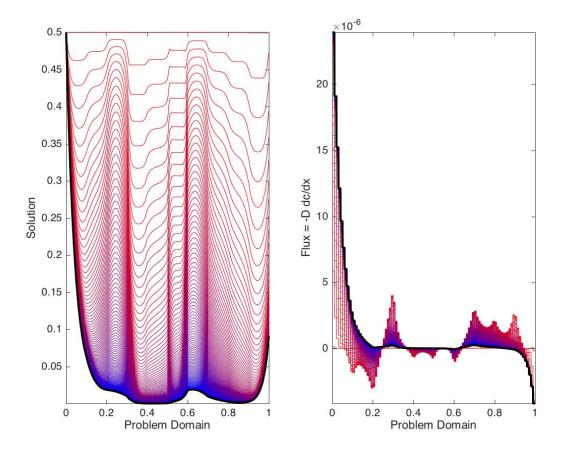


Figure 8. Time-dependent solution for concentration (red = early in transient, purple/blue = late in transient, black = at final time step) for 10000 time steps for every 100 time steps.

As can be seen, there is relatively little difference in the results obtained for the three different time steps, except that with a finer time step, the solution results are obtained at more instances in time. This is beneficial early in the transient, especially because, as can be seen from Fig. 5, in the first few time steps the solution changes very rapidly, but as time progresses, begins to approach a steady state. Adaptive time stepping would therefore be very beneficial for this approach to equilibrium case, since then large numbers of time steps could be used early in the transient, and fewer later as the solution stops changing as much.

7 Appendix

This section contains the complete code used in this assignment.

7.1 FEProgram.m

This program contains the master program for running the FE problem.

```
fontsize = 16;
                                  % fontsize for plots
num_elem = 100;
                                  % number of finite elements
                                  % linear elements
shape\_order = 2;
end_time = 6.5e3;
                                 % end simulation time
                                  % number of time steps
num_steps = 100;
dt = end_time / num_steps;
                                  % time step size
ic = 0.5:
                                  % initial condition
discr = 100;
                                   % plot every discr time steps
\% specify D and Tau over the domain in a block structure
D_blocks = \begin{bmatrix} 2.4 & 2.0 & 1.5 & 0.6 & 1.3 & 0.14 & 1.1 & 2.2 & 2.0 & 1.5 \end{bmatrix} \cdot * (10^{\circ}(-6));
Tau_blocks = \begin{bmatrix} 1.2 & 0.8 & 0.3 & 1.4 & 1.15 & 0.75 & 0.35 & 0.85 & 1.25 & 2.0 \end{bmatrix} * (10^{\circ}(-3));
space_blocks = 0.1:0.1:L;
% form the permutation matrix for assembling the global matrices
[permutation] = permutation(shape_order);
parent_domain = -1:0.1:1;
physical_domain = linspace(0, L, num_elem * length(parent_domain) - (num_elem
   \hookrightarrow - 1);
% define the quadrature rule
[wt, qp] = quadrature(shape_order);
% interpolate D and Tau into the physical domain
[D-physical-domain] = PhysicalInterpolation(physical-domain, space-blocks,
   \hookrightarrow D_blocks);
[Tau_physical_domain] = PhysicalInterpolation(physical_domain, space_blocks,
   → Tau_blocks);
% perform the meshing
[num\_nodes, num\_nodes\_per\_element, LM, coordinates] = mesh(L, num\_elem,

→ shape_order);
% specify the boundary conditions
[dirichlet_nodes, neumann_nodes, a_k] = BCnodes(left, right, left_value,

→ right_value, num_nodes);
num_dirichlet = length(dirichlet_nodes(1,:));
n = 1; % index for the time step
% apply the initial condition
soln_condensed_cell = cell([1, num_steps]);
soln_condensed_cell{1, n} = ic .* ones(1, num_nodes - num_dirichlet);
% initialize the solution and derivative cells
soln_FE_cell = cell([1, length(physical_domain)]);
soln_derivative_FE_cell = cell([1, length(physical_domain)]);
soln_FE_cell{1, n} = ic .* ones(1, length(physical_domain));
soln_derivative_FE_cell {1, n} = zeros(1, length(physical_domain));
% interpolate D and Tau into the an elemental basis
[D_elem, right_endpoint_index, right_endpoint_coordinate] =
```

```
→ ElementInterpolation (coordinates, num_elem, num_nodes_per_element,
   → space_blocks , D_blocks );
[Tau_elem, right_endpoint_index, right_endpoint_coordinate] =
   → ElementInterpolation (coordinates, num_elem, num_nodes_per_element,

→ space_blocks , Tau_blocks );
K_{cell} = cell([1, num_elem]);
M_{cell} = cell([1, num_{elem}]);
F_cell = cell([1, num_elem]);
K = zeros(num\_nodes, num\_nodes);
M = zeros(num\_nodes, num\_nodes);
F = zeros(num\_nodes, 1);
for elem = 1:num_elem
    k = zeros(num_nodes_per_element);
    m = zeros(num_nodes_per_element);
    f = zeros (num_nodes_per_element, 1);
     for l = 1: length(qp)
         for i = 1:num_nodes_per_element
              [N, dN, x_xe, dx_dxe] = shapefunctions(qp(1), shape_order,
                 \hookrightarrow coordinates, LM, elem);
              % assemble the (elemental) forcing vector
              if strcmp(right , 'Neumann')
                  if ((neumann_nodes(1,1) = (elem + 1)) & (i = 
                     → num_nodes_per_element) && (l == 1))
                     f(i) = f(i) - neumann_nodes(2, 1);
                  end
              end
              for j = 1:num_nodes_per_element
                  \% assemble the (elemental) stiffness matrix
                  k(i,j) = k(i,j) + wt(1) * (D_elem(elem) * dN(i) * dN(i) /
                     \rightarrow dx_dxe + Tau_elem(elem) * N(i) * N(j) * dx_dxe);
                  % assemble the (elemental) mass matrix
                  m(i,j) = m(i,j) + wt(1) * N(i) * N(j) * dx_dxe;
              end
         end
     end
     % store elemental values into cells
     K_{\text{cell}}\{1, \text{ elem}\} = k;
     M_{cell}\{1, elem\} = m;
     F_{cell} \{1, elem\} = f;
end
for elem = 1:num_elem % assemble into the global matrices
     m = 1;
     for m = 1: length(permutation(:,1))
        i = permutation(m, 1);
        j = permutation(m, 2);
```

```
K(LM(elem , i), LM(elem , j)) = K_cell\{1, elem\}(i, j) + K(LM(elem , i),
                            \hookrightarrow LM(elem, j);
                   M(LM(elem, i), LM(elem, j)) = M_cell\{1, elem\}(i, j) + M(LM(elem, i),
                           \hookrightarrow LM(elem, j));
            end
            for i = 1:length(f)
                   F(LM(elem, i)) = F((LM(elem, i))) + F_cell\{1, elem\}(i);
            end
\mathbf{end}
% perform static condensation to remove known Dirichlet nodes from solve
[K_uu, K_uk, F_u, F_k] = condensation(K, F, num_nodes, dirichlet_nodes);
[M_uu, M_uk, F_u, F_k] = condensation(M, F, num_nodes, dirichlet_nodes);
for n = 1:num\_steps
         \% perform the very first solve using Gaussian elimination (time-dependent
                 \hookrightarrow case
          A_{mat} = (1 / dt) * M_{uu} + K_{uu};
          b_mat = F_u + (1 / dt) * (M_uu * soln_condensed_cell \{1, n\} + M_uk * (M_uu * soln_condensed_cell \{1, n\} + M_
                 \rightarrow dirichlet_nodes (2,:) ') - ((1 / dt) * M_u k + K_u k) * dirichlet_nodes
                  \hookrightarrow (2,:) ';
          a_u\_condensed = A\_mat \setminus b\_mat;
          soln\_condensed\_cell\{1, n + 1\} = a\_u\_condensed;
         % expand a_condensed to include the Dirichlet nodes
          a = zeros(num\_nodes, 1);
          a_row = 1;
                                       % index for dirichlet_nodes
          i = 1;
          j = 1;
                                      % index for expanded row
          for a_row = 1:num_nodes
                    if (find(dirichlet_nodes(1, :) = a_row))
                             a(a_row) = dirichlet_nodes(2,i);
                              i = i + 1;
                    else
                             a(a_row) = a_u\_condensed(j);
                              j = j + 1;
                   end
         end
         % assemble the solution in the physical domain
          [solution_FE, solution_derivative_FE] = postprocess(num_elem,
                 → parent_domain, a, LM, num_nodes_per_element, shape_order,

→ coordinates , physical_domain );
          soln_FE_cell\{n+1\} = solution_FE;
          soln_derivative_FE_cell{n+1} = solution_derivative_FE;
         n = n + 1;
end
[none] = PlotInTime(1, soln_FE_cell, physical_domain, 'Solution', discr,
        \rightarrow num_steps);
```

```
[none] = PlotInTime(-D\_physical\_domain, soln\_derivative\_FE\_cell, \\ \hookrightarrow physical\_domain, 'Flux\_=\_-D\_dc/dx', discr, num\_steps); \\ %saveas(gcf, '10000\_100', 'jpeg')
```

7.2 MeshGenerator.m

This program generates the mesh and connectivity matrix.

```
% Mesh generator, ME 280a HW 5
clear all
Nt = 3;
                                      % number of layers
No = 8;
                                      % number of elements in theta
Nc = 4;
                                      % number of elements in circum
\mathbf{if} \mod(\mathrm{No}, 2) = 0
    disp ('No_must_be_even!')
end
                                      % linear elements
num\_nodes\_per\_elem = 8;
R = 1;
                                      % radius of each arch
r = 0.3;
                                      % radius of inner hole
t = 0.2;
                                      % thickness of the tube wall
layer\_thickness = t / (Nt);
                                    % thickness of each ring
angle = (2*pi) / Nc;
                                      % angle in horizontal plane
% each row represents one coordinate of a global node
coordinates = zeros(Nt * Nc, 3);
Angle = \mathbf{pi} / (No / 2);
k = 1; % index for coordinate row
x = 0;
y = 0;
z = 0;
Theta = 0; % angle in each plane
\% find the x-coordinates of the No + 1 slices
x_{centers} = zeros(1, No + 1);
x_{centers}(1) = x;
theta_inc = \mathbf{pi} / (No / 2);
theta = theta_inc;
% in the first half of the tube
for i = 2:((No/2) + 1)
    x_{centers}(i) = x_{centers}(1) + (R + t + r) * (1 - cos(theta));
    theta = theta + theta_inc;
end
% in the second half of the tube
j = i + 1;
second_part_start = x_centers(i);
```

```
for l = 2:((No/2) + 1)
    x_centers(j) = second_part_start + x_centers(1);
    j = j + 1;
end
% create a vector of the y-coordinates
z_{\text{centers}} = \mathbf{zeros}(1, \text{No} + 1);
z_centers(1) = z;
theta = theta_inc;
% in the first half of the tube
for i = 2:((No/2) + 1)
    z_{\text{centers}}(i) = (R + t + r) * sin(theta);
    theta = theta + theta_inc;
end
% in the second half of the tube
j = i + 1;
for l = 2:((No/2) + 1)
    z_{centers}(j) = -z_{centers}(l);
    j = j + 1;
end
for l = 1:(No + 1) % mesh in the theta direction
    % for each plane
    theta = 0;
    dt = 0;
    % meshes in a plane perpendicular to tube axis
    for j = 1:(Nt + 1) % create all layers of rings
         for i = 1:Nc % create a single ring
             % x-coordinate
             coordinates (k, 1) = x_{-centers}(1) + (r + dt) * cos(theta);
             % compute tilting parameters
             w = \sin(pi/2 - Theta) * ((r + dt) * cos(theta)) / sin(pi/2);
             h = w * sin(Theta);
             p = w * cos(Theta);
             \% y-coordinate
             coordinates (k, 2) = y + (r + dt) * sin(theta);
             \% z-coordinate
             coordinates(k,3) = z_centers(1);
             % tilt for each half of the tube
             if l > No/2
                 \operatorname{sn} = -1;
             else
                 \operatorname{sn} = 1;
             end
```

```
% tilt the z-coordinate for off-symmetric planes
            if find ([1, 2, 8], i)
                 coordinates(k,3) = coordinates(k,3) - sn*h;
             else
                 coordinates(k,3) = coordinates(k,3) + sn*h;
            end
            % tilt the symmetric planes (the peaks)
            if (1 = 3) \mid | (1 = 7)
                 if find ([1, 2, 8], i)
                     coordinates(k,3) = coordinates(k,3) - ((r + dt) * cos(
                        \hookrightarrow theta));
                 else
                     coordinates(k,3) = coordinates(k,3) + ((r + dt) * cos(
                        \hookrightarrow theta));
                 end
                 coordinates(k,1) = x_centers(1);
            end
            % tilt the x-coordinate
            if find ([1, 2, 8], i)
                 coordinates(k,1) = coordinates(k,1) + p;
            else
                 coordinates(k,1) = coordinates(k,1) - p;
            end
            k = k + 1;
             theta = theta + angle;
        end
        dt = dt + layer_thickness; % reset radius
        theta = 0; % reset angle
    end
    % move the angle (along length) and centers for the next plane
    Theta = Theta + Angle;
    x = x + (R + t + r) * (1 - \cos(Theta));
    y = y;
    z = z + (R + t + r) * sin(Theta);
end
X = coordinates(:,1);
Y = coordinates(:,2);
Z = coordinates(:,3);
% generate the connectivity matrix
num_elem = No * Nc * Nt;
LM = zeros (num_elem, num_nodes_per_elem);
% apply in a single slice
j = 1;
k = 1;
```

```
e = 1:
for l = 1:(No + 1) % for each slice, assign the front node values
   for elem = e:(e + num\_elem / No - 1) % for each element in the slice
       LM(elem, 1) = j;
       LM(elem, 3) = j + Nc;
       if (\text{mod}(\text{elem}, \text{Nc}) = 0)
           LM(elem, 2) = k;
           LM(elem, 4) = k + Nc;
           k = k + Nc;
       else
           LM(elem, 2) = j + 1;
           LM(elem, 4) = j + Nc + 1;
       end
       j = j + 1;
   end
   % update for next slice frontal values
  k = k + Nc;
   j = j + Nc;
   e = e + Nc * Nt;
% assign the back face values - front values for second slice are
\%\ back\ values\ for\ first\ slice , etc.
i = 1;
for j = 1:No
    for elem = i:(i + Nc*Nt - 1)
        LM(elem, (Nc + 1) : end) = LM(elem + Nc * Nt, 1:Nc);
    end
    i = i + Nc*Nt;
end
% delete the unecessary last "chunk" in the LM
LM = LM(1:num\_elem, :);
```

7.3 BCnodes.m

This function applies the boundary conditions.

```
case 'Dirichlet'
        dirichlet_nodes(1, i) = 1;
        dirichlet_nodes(2, i) = left_value;
        i = i + 1;
    case 'Neumann'
        neumann_nodes(1, j) = 1;
        j = j + 1;
    otherwise
        disp ('You_entered_an_incorrect_field_for_the_type_of_BC_on_the_left_
           → boundary.');
end
switch right
    case 'Dirichlet'
        dirichlet_nodes(1, i) = num_nodes;
        dirichlet_nodes(2, i) = right_value;
    case 'Neumann'
        neumann\_nodes(1, j) = num\_nodes;
        neumann_nodes(2, j) = right_value;
    otherwise
        disp ('You_entered_an_incorrect_field_for_the_type_of_BC_on_the_right_
           → boundary.');
end
a_{-k} = [];
if isempty(dirichlet_nodes)
    disp('no_dirichlet_nodes')
else
    a_k = dirichlet_nodes(2,:);
end
```

7.4 ElementInterpolation.m

This function interpolates the material property data into an element basis.

7.5 PhysicalInterpolation.m

This function interpolates the material property data into the physical domain.

7.6 PlotInTime.m

This function performs the plotting for this assignment.

```
if minimum < current_min
         current_min = minimum;
    end
end
plot(physical_domain, param .* func{1, 1})
ylim ([minimum, maximum])
ylabel (y_label)
xlabel('Problem_Domain')
hold on
dc = 0.0;
for n = [2:discr:num_steps, num_steps]
    if n == num_steps
         plot(physical_domain, param .* func{1, n}, 'k-', 'LineWidth',
            \hookrightarrow linewidth)
    else
         plot (physical_domain, param .* func {1, n}, 'Color', [1.0 - dc, 0.0, dc
            \hookrightarrow ])
    end
    drawnow
    dc = dc + 0.01;
end
none = 0;
end
```

7.7 condensation.m

This function performs the static condensation to remove Dirichlet nodes from the solve.

```
% Performs static condensation and removes Dirichlet nodes from the global
\% \ matrix \ solve \ K * a = F
% To illustrate the process here, assume that the values at the first and
% last nodes (1 and 5) are specified. The other nodes (2, 3, and 4) are
\% unknown. For a 5x5 node system, the following matrices are defined:
\% K =
                 K(1,1)
                         K(1,2)
                                  K(1,3)
                                           K(1,4)
                                                   K(1,5)
%
                         K(2,2)
                                  K(2,3)
                                           K(2,4)
                 K(2,1)
                                                   K(2,5)
%
                 K(3,1)
                         K(3,2)
                                  K(3,3)
                                           K(3,4)
                                                   K(3,5)
%
                                  K(4,3)
                 K(4,1)
                         K(4,2)
                                           K(4,4)
                                                   K(4,5)
                                  K(5,3)
%
                         K(5,2)
                                          K(5,4)
                 K(5,1)
                                                   K(5,5)
\% K_{-}uu_{-}rows =
                 K(2,1)
                         K(2,2)
                                  K(2,3)
                                           K(2,4)
                                                   K(2,5)
%
                 K(3,1)
                         K(3,2)
                                  K(3,3)
                                           K(3,4)
                                                   K(3,5)
%
                 K(4,1)
                         K(4,2)
                                  K(4,3)
                                          K(4,4)
                                                  K(4,5)
% K_{-}uu =
                         K(2,2)
                                  K(2,3)
                                          K(2,4)
%
                         K(3,2)
                                 K(3,3) \quad K(3,4)
%
                         K(4,2) \quad K(4,3) \quad K(4,4)
```

```
% K_{-}uk =
                K(2,1)
                                                  K(2,5)
                K(3,1)
                                                  K(3,5)
%
                K(4,1)
                                                  K(4,5)
% K_{-}ku =
                        K(1,2) K(1,3) K(1,4)
%
%
                         K(5,2) K(5,3) K(5,4)
%
               K(1,1)
                                                  K(1,5)
% K_{-}kk =
%
%
%
                K(5,1)
                                                  K(5,5)
function [K_uu, K_uk, F_u, F_k] = condensation(K, F, num_nodes,

    dirichlet_nodes)
K_uu_rows = zeros(num_nodes - length(dirichlet_nodes(1, :)), num_nodes);
K_uk = zeros(num_nodes - length(dirichlet_nodes(1,:)), length(dirichlet_nodes
   \hookrightarrow (1,:));
F_u = zeros(num\_nodes - length(dirichlet\_nodes(1, :)), 1);
F_k = zeros(length(dirichlet_nodes(1,:)), 1);
K_{row} = 1;
            \% index for dirichlet_nodes
i = 1;
            % index for condensed row
j = 1;
            % index for unknown condensed row
1 = 1;
            % index for known condensed row
for K_row = 1:num_nodes
    if (find(dirichlet_nodes(1, :) == K_row))
        F_k(m) = F(K_row);
        m = m + 1;
        i = i + 1;
    else
        K_uu_rows(j,:) = K(K_row,:);
        F_u(1) = F(K_row);
        j = j + 1;
        1 = 1 + 1;
    end
end
% perform static condensation to remove Dirichlet node columns from solve
K_uu = zeros(num_nodes - length(dirichlet_nodes(1, :)), num_nodes - length(
```

```
\hookrightarrow dirichlet_nodes (1, :));
K_{\text{-}}column = 1;
i = 1;
                  % index for dirichlet nodes
j = 1;
                  % index for condensed column
m = 1;
                 % index for K_uk column
for K_column = 1:num_nodes
    if (find(dirichlet_nodes(1, :) = K_column))
         K_uk(:,m) = K_uu_rows(:, K_column);
        m = m + 1;
         i = i + 1;
    else
         K_{uu}(:,j) = K_{uu}rows(:, K_{column});
         j = j + 1;
    end
end
%K_{-}uu = sparse(K_{-}uu);
```

7.8 mesh.m

This function meshes the domain.

```
function [num_nodes, num_nodes_per_element, LM, coordinates] = mesh(L,
   → num_elem, shape_order)
num\_nodes = (shape\_order - 1) * num\_elem + 1;
% for evenly-spaced nodes, on a 3-D mesh. Each row corresponds to a node.
coordinates = zeros(num_nodes, 3);
\% in 1-D, the first node starts at (0,0), and the rest are evenly-spaced
for i = 2:num\_nodes
   coordinates(i,:) = [coordinates(i-1, 1) + L/(num\_nodes - 1), 0, 0];
\mathbf{end}
% Which nodes correspond to which elements depends on the shape function
% used. Each row in the LM corresponds to one element.
num_nodes_per_element = shape_order;
LM = zeros (num_elem, num_nodes_per_element);
for i = 1:num_elem
    for j = 1:num_nodes_per_element
        LM(i,j) = num\_nodes\_per\_element * (i - 1) + j - (i - 1);
    end
end
end
```

7.9 permutation.m

This function meshes the domain.

```
function [permutation] = permutation(num_nodes_per_element)

permutation = zeros(num_nodes_per_element ^ 2, 2);

r = 1;
c = 1;
c = 1:num_nodes_per_element^2
    permutation(i,:) = [r, c];
    if c == num_nodes_per_element
        c = 1;
        r = r + 1;
    else
        c = c + 1;
    end
end
```

7.10 postprocess.m

This function postprocesses the results into the physical domain.

```
function [solution_FE, solution_derivative_FE] = postprocess(num_elem,
   → parent_domain, a, LM, num_nodes_per_element, shape_order, coordinates,
   → physical_domain)
b = zeros(1, shape\_order);
A = zeros(shape\_order);
m = length(parent\_domain) + 1;
p = 1;
u\_sampled\_solution\_matrix = zeros(num\_elem, length(parent\_domain));
u_sampled_solution_derivative_matrix = zeros(num_elem, length(parent_domain));
for elem = 1:num_elem
    % over each element, figure out the polynomial by solving a linear
    \% system, Ax = b, where A depends on the order of the shape functions
    for i = 1:num_nodes_per_element
        b(i) = a(LM(elem, i));
    end
    for j = 1:shape_order % loop over the rows of A
        coordinate = coordinates (LM(elem, j));
        for l = 1: shape-order % loop over the columns of A
            A(j,l) = coordinate  (l - 1);
        end
    end
    % solve for the coefficients on the actual polynomial
    coefficients = A \setminus (b');
    % determine the solution over the element
    solution_over_element = zeros(1, length(parent_domain));
    element_domain = linspace (coordinates (LM(elem, 1)), coordinates (LM(elem,
       → num_nodes_per_element)), length(parent_domain));
```

```
for i = 1:num_nodes_per_element
        solution_over_element = solution_over_element + coefficients(i) .* (
            \rightarrow element_domain . (i-1);
    end
    % determine the derivative over the element
    derivative_over_element = zeros(1, length(parent_domain));
    for i = 2:num_nodes_per_element % the derivative of the constant is zero
        derivative_over_element = derivative_over_element + coefficients(i) .*
            \hookrightarrow (i - 1) .* (element_domain .^ (i - 2));
    end
    \% put into a matrix
    u_sampled_solution_matrix(p,:) = solution_over_element;
    u_sampled_solution_derivative_matrix(p,:) = derivative_over_element;
    p = p + 1;
end
% assemble solution and derivative into a single vector
solution_FE = zeros(1, length(physical_domain));
solution_derivative_FE = zeros(1, length(physical_domain));
for i = 1:length(u_sampled_solution_matrix(:,1))
    if i == 1
        solution_FE(1:length(u_sampled_solution_matrix(i,:))) =

    u_sampled_solution_matrix(i,:);
        solution_derivative_FE (1:length (u_sampled_solution_derivative_matrix (i
            \rightarrow ,:))) = u_sampled_solution_derivative_matrix(i,:);
    else
        solution_FE(m:(m + length(u_sampled_solution_matrix(1,:)) - 2)) =
            \hookrightarrow u_sampled_solution_matrix(i,2:end);
        solution_derivative_FE (m: (m + length (
            \hookrightarrow u_sampled_solution_derivative_matrix (1,:) - 2)) =
            → u_sampled_solution_derivative_matrix(i,2:end);
        m = m + length(u\_sampled\_solution\_matrix(1,:)) - 1;
    end
end
```

7.11 quadrature.m

This function selects the quadrature rule.

```
function [wt, qp] = quadrature(shape_order)
shape_order = 4; % forces a five-point quadrature rule for this problem
switch shape_order
case 2
    wt = [1.0, 1.0];
    qp = [-sqrt(1/3), sqrt(1/3)];
case 3
    wt = [5/9, 8/9, 5/9];
    qp = [-sqrt(3/5), 0, sqrt(3/5)];
case 4
    wt = [(322-13*sqrt(70))/900, (322+13*sqrt(70))/900, 128/225, (322+13*sqrt(70))/900];
```

```
\begin{array}{l} \operatorname{qp} = \left[ -(1/3) * \mathbf{sqrt} (5 + 2 * \mathbf{sqrt} (10/7)) \right], \quad -(1/3) * \mathbf{sqrt} (5 - 2 * \mathbf{sqrt} (10/7)), \quad 0.0, \\ & \hookrightarrow (1/3) * \mathbf{sqrt} (5 - 2 * \mathbf{sqrt} (10/7)), \quad (1/3) * \mathbf{sqrt} (5 + 2 * \mathbf{sqrt} (10/7)) \right]; \\ \operatorname{otherwise} \\ & \operatorname{\mathbf{disp}} ( \text{'You\_entered\_an\_unsupported\_shape\_function\_order\_for\_the\_} \\ & \hookrightarrow \operatorname{quadrature\_rule.'} ); \\ \mathbf{end} \end{array}
```

7.12 shapefunctions.m

This function provides the shape functions.

```
\% N
             : shape functions in the master domain
% dN
             : derivative of the shape functions with respect to xe
% x_{-}xe
             : x as a function of xe
% dx_{-}dxe
            : derivative of x with respect to xe
function [N, dN, x_xe, dx_dxe] = shapefunctions (xe, shape_order, coordinates,
   \hookrightarrow LM, elem)
% shape functions and their derivatives
N = zeros(shape\_order, 1);
dN = zeros (shape_order, 1);
switch shape_order
    case 2
        N(1) = (1 - xe) ./ 2;
        N(2) = (1 + xe) ./ 2;
        dN(1) = -1/2;
        dN(2) = 1/2;g
    otherwise \\
        disp('You_entered_an_unsupported_shape_function_order.');
end
% check that the sum of the shape functions adds up to 1
sum = 0;
for j = 1: shape\_order
    \mathbf{sum} = \mathbf{sum} + \mathrm{N}(\mathrm{j});
end
if (abs(sum - 1.0) > 1e-10)
    disp ('Sum_of_the_shape_functions_does_not_add_up_to_1.');
\% x(xe) transformation to the parametric domain
x_x = 0.0;
dx_dxe = 0.0;
for i = 1:shape_order
    x_x = x_x + coordinates(LM(elem, i)) * N(i);
    dx_{dx} = dx_{dx} + coordinates(LM(elem, i)) * dN(i);
end
```