# Assignment #1 - Astronomia de Sistemas Planetários

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Disponível em: https://github.com/araujoarthur/psastronomyclass/tree/main/A1

### Packages and Constant Definition

```
In [76]: using DataFrames, PlotlyJS
    const possibleBodies = ["Mercury", "Venus", "Earth", "Mars", "Jupyter", "
    const au = 149597870700

WARNING: redefinition of constant possibleBodies. This may fail, cause in
    correct answers, or produce other errors.
Out[76]:
```

#### **Functions definition**

Orbital Element Calculation Functions:

```
In [53]:
         function \Omega(body::String, t::Float64) # \0mega
             if !(body ∈ possibleBodies)
                  return nothing
             elseif body == "Mercury"
                  return 48.3333 + 3.24587E-5 * t
             elseif body == "Venus"
                  return 76.6799 + 2.46590E-5 * t
             elseif body == "Earth"
                  return 0
             elseif body == "Mars"
                  return 49.5574 + 2.11081E-5 * t
             elseif body == "Jupyter"
                  return 100.4541 + 2.76854E-5 * t
             elseif body == "Saturn"
                  return 113.6634 + 2.38980E-5 * t
             elseif body == "Uranus"
                  return 74.0005 + 1.3978E-5 * t
             elseif body == "Neptune"
                  return 131.7806 + 3.0173E-5 * t
             end
         end
         function i(body::String, t::Float64) # \isansi
             if !(body ∈ possibleBodies)
                  return nothing
             elseif body == "Mercury"
                  return 7.0047 + 5E-8 * t
             elseif body == "Venus"
                  return 3.3946 + 2.75E-8 * t
             elseif body == "Earth"
                  return 0
             elseif body == "Mars"
                  return 1.8497 - 1.78E-8 * t
             elseif body == "Jupyter"
                  return 1.3030 - 1.557E-7 * t
             elseif body == "Saturn"
                  return 2.4886 - 1.081E-7 * t
             elseif body == "Uranus"
                  return 0.7733 + 1.9E-8 * t
             elseif body == "Neptune"
                  return 1.7700 - 2.55E-7 * t
             end
         end
         function ω(body::String, t::Float64) # \omega
             if !(body ∈ possibleBodies)
                  return nothing
             elseif body == "Mercury"
                  return 29.1241 + 1.01444E-5 * t
             elseif body == "Venus"
                  return 54.8910 + 1.38374E-5 * t
             elseif body == "Earth"
                  return 282.9404 + 4.70935E-5 * t
             elseif body == "Mars"
                  return 286.5016 + 2.92961E-5 * t
             elseif body == "Jupyter"
                  return 273.8777 + 1.64505E-5 * t
             elseif body == "Saturn"
                  return 339.3939 + 2.97661E-5 * t
             elseif body == "Uranus"
                  return 96.6612 + 3.0565E-5 * t
             elseif body == "Neptune"
                  raturn 272 0/61 _ 6 027E 6 * +
```

```
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    end
end
function a(body::String, t=0::Float64) # \scra
    if !(body ∈ possibleBodies)
        return nothing
    elseif body == "Mercury"
        return 0.387098*au
    elseif body == "Venus"
        return 0.723330*au
    elseif body == "Earth"
        return 1*au
    elseif body == "Mars"
        return 1.523688*au
    elseif body == "Jupyter"
        return 5.20256*au
    elseif body == "Saturn"
        return 9.55475*au
    elseif (body == "Uranus") && (t ≠ 0)
        return (19.18171 - 1.55E-8 * t) * au
    elseif (body == "Neptune") && (t ≠ 0)
        return (30.05826 + 3.313E-8 * t) * au
        return nothing
    end
end
function e(body::String, t::Float64) # \bfe
    if !(body ∈ possibleBodies)
        return nothing
    elseif body == "Mercury"
        return 0.205635 + 5.59E-10 * t
    elseif body == "Venus"
        return 0.006773 - 1.302E-9 * t
    elseif body == "Earth"
        return 0.016709 - 1.151E-9 * t
    elseif body == "Mars"
        return 0.093405 + 2.516E-9 * t
    elseif body == "Jupyter"
        return 0.048498 + 4.469E-9 * t
    elseif body == "Saturn"
        return 0.055546 - 9.499E-9 * t
    elseif body == "Uranus"
        return 0.047318 + 7.45E-9 * t
    elseif body == "Neptune"
        return 0.008606 + 2.15E-9 * t
    end
end
function M(body::String, t::Float64) # \bisansM
    if !(body ∈ possibleBodies)
        return nothing
    elseif body == "Mercury"
        return 168.6562 + 4.0923344368 * t
    elseif body == "Venus"
        return 48.0052 + 1.6021302244 * t
    elseif body == "Earth"
        return 356.0470 + 0.9856002585 * t
    elseif body == "Mars"
        return 18.6021 + 0.5240207766 * t
    elseif body == "Jupyter"
        return 19.8950 + 0.0830853001 * t
```

```
elseif body == "Saturn"
                  return 316.9670 + 0.0334442282 * t
              elseif body == "Uranus"
                  return 142.5905 + 0.011725806 * t
              elseif body == "Neptune"
                  return 260.2471 + 0.005995147 * t
              end
          end
          M (generic function with 1 method)
Out[531:
          Arc Correction for E and M
In [65]: | function arcCorrection(ARC::Float64)
              println("ARC: ", ARC)
              if ARC >= 360.0
                  println("CORR: ", mod(ARC,360.0))
                  return mod(ARC,360.0)
              else
                  println("NOT CORR")
                  return ARC
              end
          end
         arcCorrection (generic function with 1 method)
Out[65]:
          Time Function:
In [55]:
          function t(d::Integer, m::Integer, y::Integer, h::Integer, mm::Integer) #
              return 367 * y - floor(7*(y + ((m+9)/12))/4) + floor(275*m/9) + d -
          end
Out[55]: t (generic function with 1 method)
          Eccencentric Anomaly and True Anomaly
In [56]:
          function E(M::Float64, ec::Float64, E=M::Float64) # \bisansE
              \Delta E = (M - \tilde{E} + (ec * sind(\tilde{E})))/(1 - (ec * cosd(\tilde{E})))
              E = \hat{E} + \Delta E;
              if (\Delta E > 5E-6)
                  return E(M, ec, E)
              else
                  return E
              end
          end
          function v(ec::Float64, E::Float64) #\nu
              return atand(sqrt((1 + ec)/(1 - ec)) * tand(E/2))
          end
         ν (generic function with 1 method)
Out[56]:
          Distance from planet to Sun
         function r(axis::Float64, ec::Float64, nu::Float64)
In [57]:
              return (axis * (1 - (ec^2)))/(1 + (ec * cosd(nu)))
          end
```

```
Out[57]: r (generic function with 1 method)
         Cartesian Coordinates
In [63]:
         function cartesian x(r):Float64, \Omega::Float64, \omega::Float64, \nu::Float64,
              return r * ((\cos d(\Omega) * \cos d(\omega + v)) - (\sin d(\Omega) * \sin d(\omega + v))
          end
          function cartesian y( r::Float64, \Omega::Float64, \omega::Float64, \nu::Float64,
              return r * ((\sin d(\Omega) * \cos d(\omega + \nu)) + (\cos d(\Omega) * \sin d(\omega + \nu))
          function cartesian z( r::Float64, ω::Float64, ν::Float64, ί::Float64)
              return r * sind(\omega + v) * sind(i)
          end
         cartesian z (generic function with 1 method)
Out[63]:
         Heliocentric Ecliptic Coordinates
         function \ell(X::Float64, Y::Float64) # \ell
In [59]:
              return atand(X/Y)
         end
          function b(X::Float64, Y::Float64, Z::Float64) # \bscrb
              return Z/(sqrt((X^2) + (Y^2)))
         end
         b (generic function with 1 method)
Out[591:
         Calculation of time for my date (01/07/2053 00:35)
         givenTime = t(1,7,2053, 0, 35)
In [60]:
         19540.024305555555
Out[60]:
         Orbital Elements of Mercury
In [70]: orbitalElements = Dict()
          for planet ∈ possibleBodies
              orbitalElements[planet] = Dict()
              orbitalElements[planet]["\Omega"] = Float64(\Omega(planet, givenTime))
              orbitalElements[planet]["i"] = Float64(i(planet, givenTime))
              orbitalElements[planet]["\omega"] = Float64(\omega(planet, givenTime))
              orbitalElements[planet]["\alpha"] = Float64(\alpha(planet, givenTime))
              orbitalElements[planet]["e"] = e(planet, givenTime)
              orbitalElements[planet]["M"] = arcCorrection(M(planet, givenTime))
              orbitalElements[planet]["E"] = E(orbitalElements[planet]["M"], orbitalElements[planet]["E"]
              orbitalElements[planet]["v"] = v(orbitalElements[planet]["e"], orbita
              orbitalElements[planet]["X"] = cartesian x(orbitalElements[planet]["r
              orbitalElements[planet]["Y"] = cartesian_y(orbitalElements[planet]["r
              orbitalElements[planet]["Z"] = cartesian z(orbitalElements[planet]["r
              orbitalElements[planet]["\ell"] = \ell (orbitalElements[planet]["X"], orbita
              orbitalElements[planet]["b"] = b(orbitalElements[planet]["X"], orbita
          end
```

```
ARC: 80132.97056153399
CORR: 212.9705615339917
ARC: 31353.668725441174
CORR: 33.66872544117359
ARC: 19614.700006651838
CORR: 174.7000066518376
ARC: 10257.980811380097
CORR: 177.9808113800973
ARC: 1643.3837833883772
CORR: 203.3837833883772
ARC: 970.4680319085465
CORR: 250.46803190854655
ARC: 371.7130342422291
CORR: 11.713034242229128
ARC: 377.39241809537845
CORR: 17.392418095378446
```

```
In [72]: DataTable = DataFrame(PLANET=possibleBodies)
DataTable[!, "Ω (°)"] = [orbitalElements[planet]["Ω"] for planet in possible DataTable[!, "ሬ (°)"] = [orbitalElements[planet]["ሬ"] for planet in possible DataTable[!, "ሬ (m)"] = [orbitalElements[planet]["ሬ"] for planet in posside DataTable[!, "ሬ (m)"] = [orbitalElements[planet]["ሬ"] for planet in possibleBodies]
DataTable[!, "Μ (°)"] = [orbitalElements[planet]["Μ"] for planet in posside DataTable[!, "Ε (°)"] = [orbitalElements[planet]["Ε"] for planet in posside DataTable[!, "ν (°)"] = [orbitalElements[planet]["ν"] for planet in posside DataTable[!, "r (m)"] = [orbitalElements[planet]["r"] for planet in posside DataTable[!, "X (m)"] = [orbitalElements[planet]["X"] for planet in posside DataTable[!, "Z (m)"] = [orbitalElements[planet]["Y"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] = [orbitalElements[planet]["ℓ"] for planet in posside DataTable[!, "ℓ (°)"] for planet in posside DataTable[!, "ℓ"] for planet in
```

```
Row | PLANET \Omega (°) i (°) \omega (°) \alpha (m)
                                                \mathbf{e}
                                                            М
    E (°)
                                             Y (m)
             ν (°) r (m)
                                  X (m)
      ь (°)
    | String Float64 Float64 Float64 Float64
                                                           Flo
at64 Float64 Float64 Float64 Float64
                                                         Float64
        Float64
Float64
  1 | Mercury 48.9675 7.00568 29.3223 5.7909e10 0.205646 21
2.971 212.875 -76.5327 5.29253e10 5.26817e10 1.81325e9 -4.737
      88.0287 -0.089867
15e9
  2 | Venus 77.1617 3.39514 55.1614 1.08209e11 0.00674756 3
3.6687 33.6725 16.9437 1.0751e11 -9.22356e10 5.4902e10 6.058
89e9 -59.2374 0.0564463
3 | Earth 0.0 0.0 283.861 1.49598e11 0.0166865 17
4.7 174.702 87.3946 1.49443e11 1.46569e11 2.91681e10 0.0
78.7448 0.0
4 | Mars 49.9699 1.84935 287.074 2.2794e11 0.0934542 17
7.981 177.984 89.0822 2.25612e11 9.13358e10 2.06287e11 2.025 96e9 23.8819 0.00898021
  5 | Jupyter 100.995 1.29996 274.199 7.78292e11 0.0485853 20
3.384 203.365 -78.8576 7.69232e11 3.41214e11 -6.89398e11 -4.617
12e9 -26.3329 -0.00600235
 6 | Saturn 114.13 2.48649 339.976 1.42937e12 0.0553604 25
0.468 250.417 -56.2729 1.38249e12 1.09074e12 8.47453e11 -5.827
07e10 52.1546 -0.0421865
 7 | Uranus 74.2736 0.773671 97.2584 2.8695e12 0.0474636 1
      11.7227 6.14432 2.73401e12 -2.73153e12 1.10781e11 3.591
11e10 -87.6776 0.0131361
  8 | Neptune 132.37 1.76502 272.728 4.49675e12 0.00864801
7.3924 17.395 8.77188 4.45831e12 2.62715e12 3.59951e12 -1.345
61e11 36.1243 -0.0301959
```

## New DataFrame to Graph in Polar Coordinates

Generating dataframe

8×15 DataFrame

In [82]: graphData = DataFrame(PLANET=possibleBodies, RADIUS=[i for i in 1:length(

 $Out[82]: 8 rows \times 3 columns$ 

	PLANET	RADIUS	POLARCOORD
	String	Int64	Float64
1	Mercury	1	88.0287
2	Venus	2	-59.2374
3	Earth	3	78.7448
4	Mars	4	23.8819
5	Jupyter	5	-26.3329
6	Saturn	6	52.1546
7	Uranus	7	-87.6776
8	Neptune	8	36.1243

## Graphing

In [83]: plot(scatterpolar(graphData, r=:RADIUS, theta=:POLARCOORD, color=:PLANET,

Out[83]:

WebIO not detected.

Please read the troubleshooting guide for more information on how to resolve this issue.

https://juliagizmos.github.io/WebIO.jl/latest/troubleshooting/not-detected/