

Project : Building a Controller

Project Rubric : WRITE UP

1) Provide a Writeup / README that includes all the rubric points and how you addressed each one. You can submit your writeup as markdown or pdf.

This would be the document where I explain the rubric points

Project Rubric : Implemented Controller

1) Implemented body rate control in C++.

- The controller is proportional to the error in body rate command.
- The controller takes into account the moments of inertia of the drone when calculating the commanded moments.

```
107 ////////////////////////////////////////////////// BEGIN STUDENT CODE //////////////////////////////////////
108 V3F error = pqrCmd - pqr;
109 momentCmd = kpPQR * error * V3F(Ixx,Iyy,Izz);
110 ////////////////////////////////////////////////// END STUDENT CODE //////////////////////////////////////
```

2) Implement roll pitch control in C++.

- The controller uses acceleration and thrust commands.
- The controller uses the rotation matrix to account the non-linear transformation from local accelerations to body rates.

```
137 ////////////////////////////////////////////////// BEGIN STUDENT CODE //////////////////////////////////////
138 float c_d,target_R13,target_R23;
139 if (collThrustCmd > 0.0)
140 {
141     c_d = collThrustCmd/mass;
142     target_R13 = -CONSTRAIN(accelCmd.x/c_d, -maxTiltAngle, maxTiltAngle);
143     target_R23 = -CONSTRAIN(accelCmd.y/c_d, -maxTiltAngle, maxTiltAngle);
144
145     pqrCmd.x = kpBank * ((R(1, 0) * (target_R13 - R(0, 2))) - (R(0, 0) * (target_R23 - R(1, 2)))) / R(2, 2);
146     pqrCmd.y = kpBank * ((R(1, 1) * (target_R13 - R(0, 2))) - (R(0, 1) * (target_R23 - R(1, 2)))) / R(2, 2);
147 }
148 else
149 {
150     pqrCmd.x = 0.0;
151     pqrCmd.y = 0.0;
152 }
153
154 pqrCmd.z = 0;
155
156 ////////////////////////////////////////////////// END STUDENT CODE //////////////////////////////////////
157
```

3) Implement altitude controller in C++.

- The controller uses both the down position and the down velocity to command thrust.
- The controller also contains **integratedAltitudeError** Variable which stores the integrated values over time.

```

185 /////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
186 float posZ_error = posZCmd - posZ;
187 integratedAltitudeError += posZ_error * dt;
188
189 velZCmd = velZCmd + kpPosZ*posZ_error + KiPosZ*integratedAltitudeError;
190 velZCmd = CONSTRAIN(velZCmd, -maxDescentRate, maxAscentRate);
191
192 accelZCmd = accelZCmd + kpVelZ*(velZCmd - velZ);
193 thrust = mass * (9.81f - (accelZCmd / R(2,2)));
194
195 /////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

4)Implement lateral position control in C++.

```

229 /////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
230
231 velCmd.constrain(-maxSpeedXY,maxSpeedXY);
232 V3F pos_Error = posCmd - pos;
233 V3F vel_Error = velCmd - vel;
234 accelCmd = accelCmdFF + kpPosXY*pos_Error + kpVelXY * vel_Error;
235 accelCmd.z = 0;
236 accelCmd.constrain(-maxAccelXY, maxAccelXY);
237
238 /////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

5)Implement yaw control in C++.

```

257 /////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
258 float yaw_Error = fmodf(yawCmd - yaw, F_PI*2.f);
259 if (yaw_Error > F_PI)
260 |   yaw_Error += - 2.0f*F_PI;
261 else if (yaw_Error < -M_PI)
262 |   yaw_Error += 2.0f*F_PI;
263
264 yawRateCmd = kpYaw * yaw_Error;
265
266
267
268 /////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

6)Implement calculating the motor commands given commanded thrust and moments in C++.

```

71 /////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
72
73 float a = momentCmd.x*sqrtf(2.f)/L;
74 a = a/(4.f);
75 float b = momentCmd.y*sqrtf(2.f)/L;
76 b = b/(4.f);
77 float c = momentCmd.z/kappa;
78 c = c/(4.f);
79 float d = collThrustCmd;
80 d = d/(4.f);
81
82 cmd.desiredThrustsN[0] = a+b+c+d;
83 cmd.desiredThrustsN[1] = -a+b-c+d;
84 cmd.desiredThrustsN[3] = -a-b+c+d;
85 cmd.desiredThrustsN[2] = a-b-c+d;
86 /////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

Project Rubric : Flight Evaluation

Test Results of my controller

```
File Edit View Search Terminal Help
SIMULATOR!
Select main window to interact with keyboard/mouse:
LEFT DRAG / X+LEFT DRAG / Z+LEFT DRAG = rotate, pan, zoom camera
W/S/UP/LEFT/DOWN/RIGHT - apply Force
C - clear all graphs
R - reset simulation
Space - pause simulation
Simulation #1 (../config/5_TrajectoryFollow.txt)
Simulation #2 (../config/1_Intro.txt)
Simulation #3 (../config/1_Intro.txt)
PASS: ABS(Quad.PosFollowErr) was less than 0.500000 for at least 0.800000 seconds
Simulation #4 (../config/1_Intro.txt)
PASS: ABS(Quad.PosFollowErr) was less than 0.500000 for at least 0.800000 seconds
Simulation #5 (../config/1_Intro.txt)
PASS: ABS(Quad.PosFollowErr) was less than 0.500000 for at least 0.800000 seconds
Simulation #6 (../config/1_Intro.txt)
PASS: ABS(Quad.PosFollowErr) was less than 0.500000 for at least 0.800000 seconds
Simulation #7 (../config/1_Intro.txt)
PASS: ABS(Quad.PosFollowErr) was less than 0.500000 for at least 0.800000 seconds
Simulation #8 (../config/2_AttitudeControl.txt)
Simulation #9 (../config/2_AttitudeControl.txt)
PASS: ABS(Quad.Roll) was less than 0.025000 for at least 0.750000 seconds
PASS: ABS(Quad.Omega.X) was less than 2.500000 for at least 0.750000 seconds
Simulation #10 (../config/2_AttitudeControl.txt)
PASS: ABS(Quad.Roll) was less than 0.025000 for at least 0.750000 seconds
PASS: ABS(Quad.Omega.X) was less than 2.500000 for at least 0.750000 seconds
Simulation #11 (../config/3_PositionControl.txt)
Simulation #12 (../config/3_PositionControl.txt)
PASS: ABS(Quad1.Pos.X) was less than 0.100000 for at least 1.250000 seconds
PASS: ABS(Quad2.Pos.X) was less than 0.100000 for at least 1.250000 seconds
PASS: ABS(Quad2.Yaw) was less than 0.100000 for at least 1.000000 seconds
Simulation #13 (../config/4_NonIdealities.txt)
Simulation #14 (../config/4_NonIdealities.txt)
PASS: ABS(Quad1.PosFollowErr) was less than 0.100000 for at least 1.500000 seconds
PASS: ABS(Quad2.PosFollowErr) was less than 0.100000 for at least 1.500000 seconds
PASS: ABS(Quad3.PosFollowErr) was less than 0.100000 for at least 1.500000 seconds
Simulation #15 (../config/4_NonIdealities.txt)
PASS: ABS(Quad1.PosFollowErr) was less than 0.100000 for at least 1.500000 seconds
PASS: ABS(Quad2.PosFollowErr) was less than 0.100000 for at least 1.500000 seconds
PASS: ABS(Quad3.PosFollowErr) was less than 0.100000 for at least 1.500000 seconds
Simulation #16 (../config/5_TrajectoryFollow.txt)
Simulation #17 (../config/5_TrajectoryFollow.txt)
PASS: ABS(Quad2.PosFollowErr) was less than 0.250000 for at least 3.000000 seconds
Press <RETURN> to close this window...
```