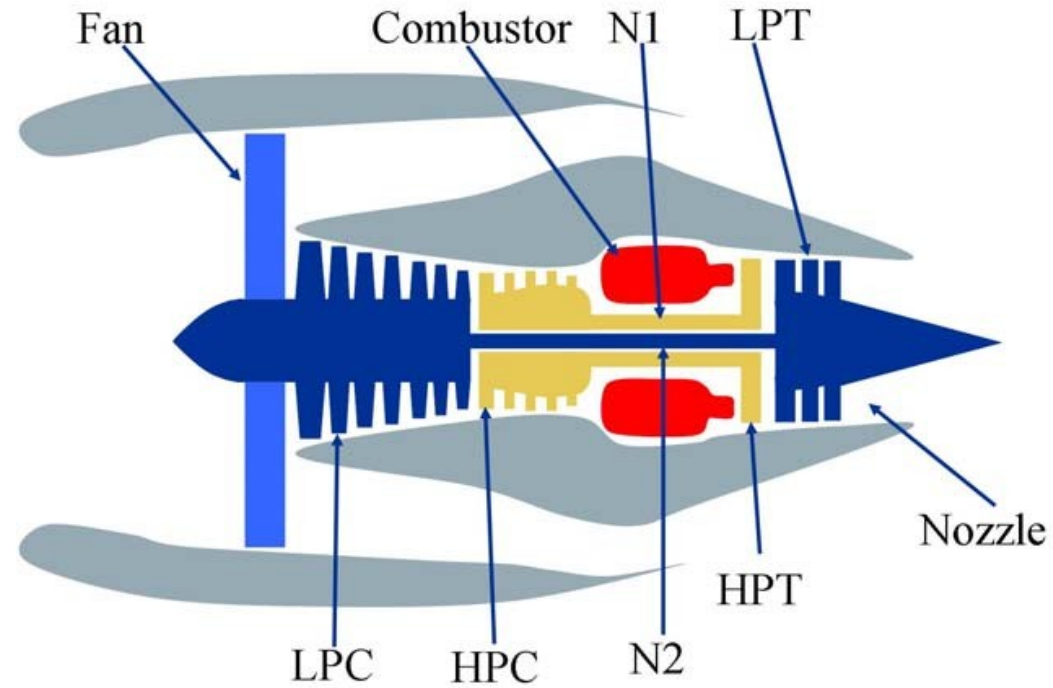


# A Study on Predictive Maintenance for Turbofan Aircraft Engines

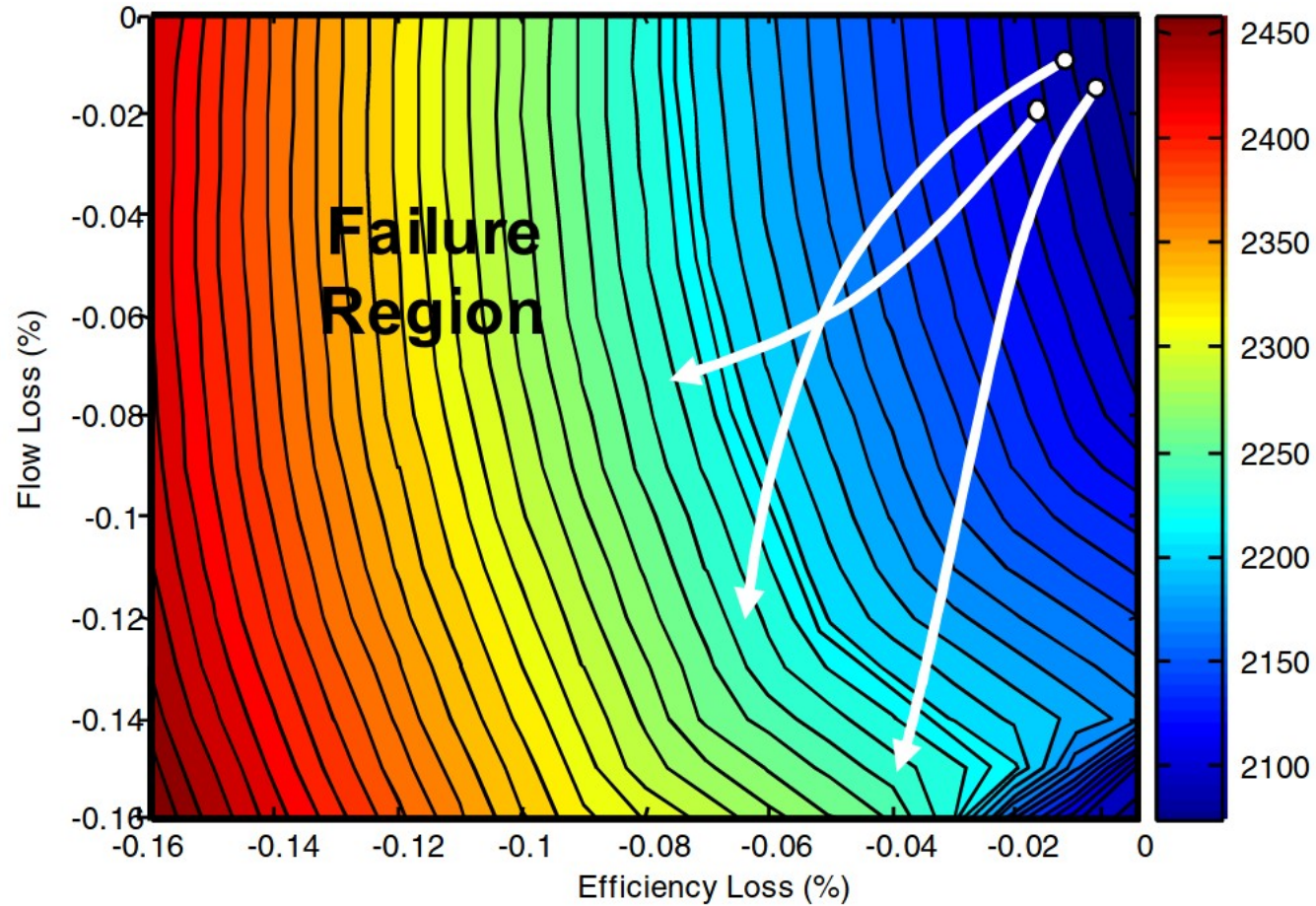
# Basic Structure of a Turbofan Engine



# The Problem

- Normal, progressive degradation of engine operations over time
- Failure = Reduced margins
  - Failure under study is NOT anomaly or catastrophe prediction

# Failure Model



# The Dataset

- Source: NASA
  - Integrated Vehicle Health Management (IVHM) program
  - Ames Research Center
- (Simulated) 90,000 lb thrust class high-bypass turbofan engine
- Training set: 100 simulated engines run to failure
- 24 features: 3 operational settings, 21 sensors

# Outcomes to Predict

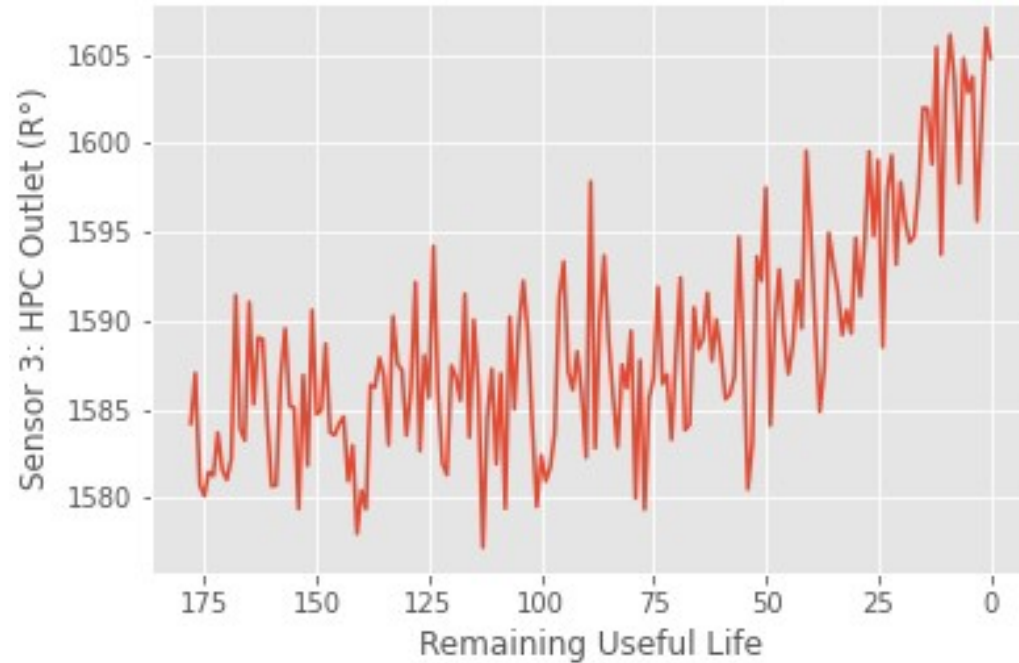
Based on current and historical operations and sensor data...

- Classification:
  - Binary - predict failure within  $x$  cycles
  - Multiclass – predict failure within  $x$  cycles, then within  $y < x$  cycles
- Regression: predict *remaining useful life*, or **RUL**
  - *Initial RUL in dataset ranged from 128 to 362 cycles*

*“Early is Better than Late”*

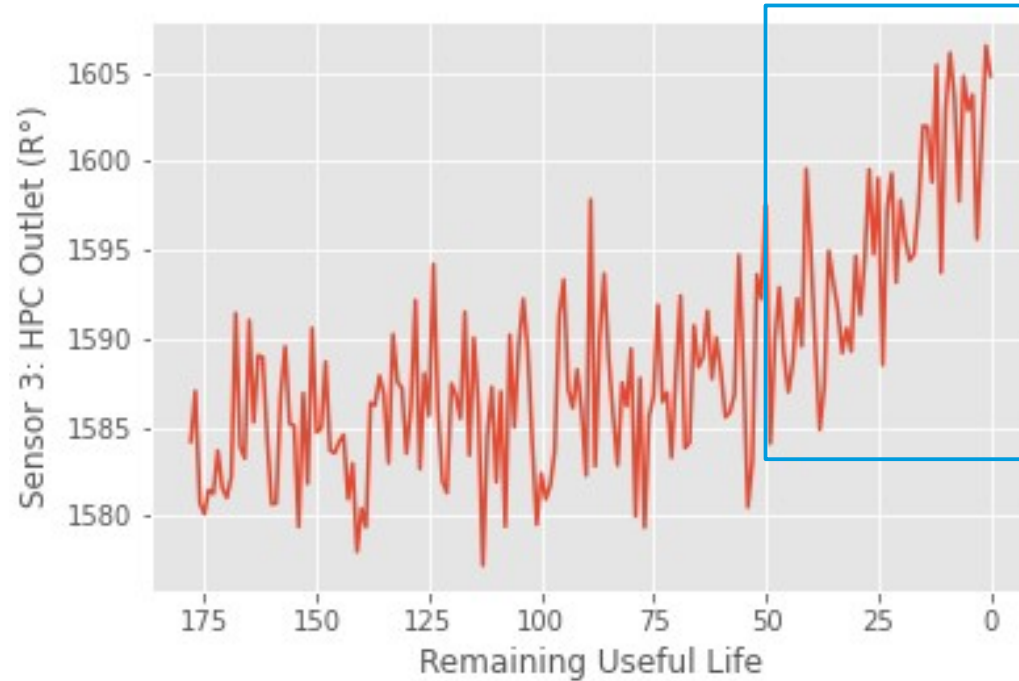
# Sensor Data Trending

Engine 3 High Pressure Compressor Outlet Temperature (Rankine)



# Sensor Data Trending

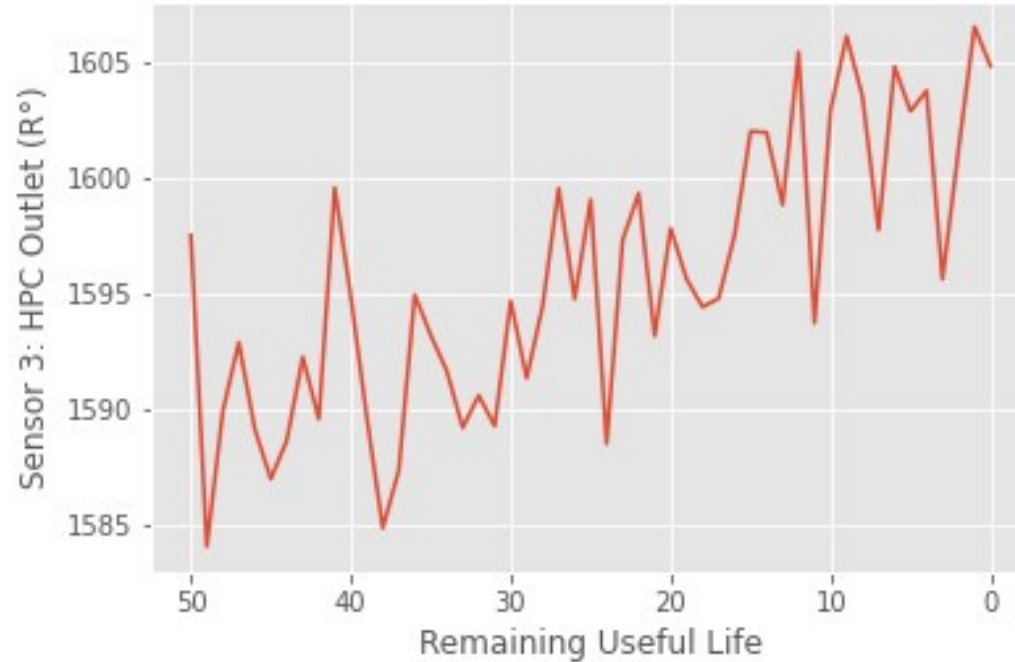
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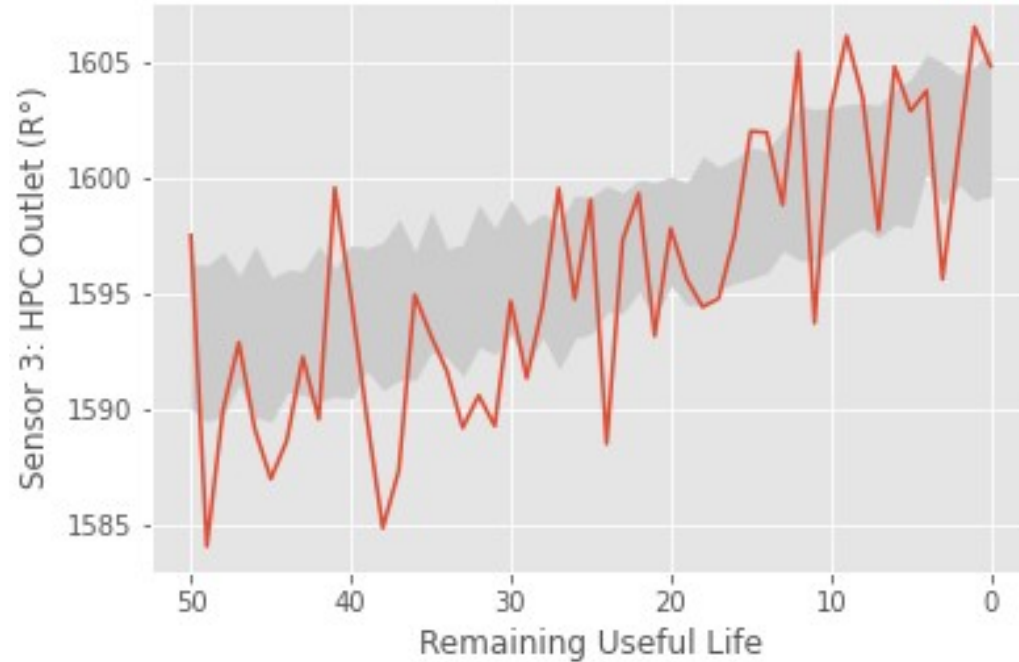
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# Sensor Data Trending

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# Models

- Logistics Regression
- Linear Regression
- Random Forest
  - Classifier
  - Regressor
- Neural Network
  - LSTM

# Models

- Logistics Regression
- Linear Regression
- Random Forest
  - Classifier
  - Regressor
- Neural Network
  - LSTM
- Regression:  
RMSE = 35 cycles
- Classification:  
recall = 80%
- Neural Networks:  
single response for all inputs

# Way Forward

- Score Weights
  - Create different penalties for 'Late' predictions
- Smooth data and amplify the signal in the noise

# Image Credits

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## Slide 2:

- Left image: Hideyuki KAMON, CC BY-SA 2.0 , <http://www.flickr.com/photos/hyougushi/66755262/>
- Right image: Saxena, A., Goebel, K., Simon, D., & Eklund, N. (2008). Damage propagation modeling for aircraft engine run-to-failure simulation. 2008 International Conference on Prognostics and Health Management, 1–9.  
<https://doi.org/10.1109/PHM.2008.4711414>
- 

## Slide 4:

- Saxena, A., Goebel, K., Simon, D., & Eklund, N. (2008). Damage propagation modeling for aircraft engine run-to-failure simulation. 2008 International Conference on Prognostics and Health Management, 1–9.  
<https://doi.org/10.1109/PHM.2008.4711414>

# Questions