4.4 Closed-Loop System's Response to Disturbances (Answers)

4.4.1 Transfer Function for Disturbance Response with PI Controller

1. In the case where the reference signal r(t) = 0, we focus on the disturbance torque T_d applied to the inertial load, and the PI controller is used. The control signal is:

$$u_m(t) = k_p(0-\omega_m(t)) + k_i \int_0^t (0-\omega_m(au)) d au$$

- 2. In the Laplace domain: $U_m(s) = -k_p\Omega_m(s) rac{k_i}{s}\Omega_m(s)$
- 3. The open-loop dynamics of the motor are given by:

$$\Omega_m(s) = rac{K}{ au s + 1} U_m(s) + rac{1}{J_{ea}} T_d(s)$$

4. Substitute the expression for $U_m(s)$ into this equation:

$$\Omega_m(s) = rac{K}{ au s + 1} \left(-k_p \Omega_m(s) - rac{k_i}{s} \Omega_m(s)
ight) + rac{1}{J_{eq}} T_d(s)$$

5. Rearrange this to express $\Omega_m(s)$ as a function of $T_d(s)$:

$$\Omega_m(s) \left(1 + rac{K}{ au s + 1} \left(k_p + rac{k_i}{s}
ight)
ight) = rac{1}{J_{eq}} T_d(s)$$

$$or,\; \Omega_m(s)\left(au s^2+s+Kk_ps+Kk_i
ight)=rac{s(au s+1)}{J_{eq}}T_d(s)$$

The transfer function $G_D(s)=rac{\Omega_m(s)}{T_d(s)}$ is:

$$G_D(s) = rac{s(au s + 1)}{J_{eq}(au s^2 + s + K k_p s + K k_i)}$$

This expresses how the motor speed $\Omega_m(s)$ responds to the disturbance torque $T_d(s)$, and depends on the parameters k_p , k_i , K, τ , and J_{eq} .

4.4.2 Steady-State Velocity with Proportional Control

When a proportional controller is used (i.e., $k_i=0$ and $k_p
eq 0$), the control signal is:

$$u_m(t) = -k_n \omega_m(t)$$

The closed-loop transfer function becomes: $\Omega_m(s) = rac{K}{ au s + 1 + K k_p} U_m(s) + rac{1}{J_{eq}} T_d(s)$

Using the Final Value Theorem for a step disturbance torque $T_d(s) = rac{T_{d0}}{s}$:

$$\omega_{ss,P}=\lim_{s\to 0} s\cdot\Omega_m(s)=\lim_{s\to 0} \frac{s\cdot\frac{1}{J_{eq}}\cdot\frac{T_{d0}}{s}}{\tau s^2+(1+Kk_p)s}. \text{ At steady-state: } \omega_{ss,P}=\frac{T_{d0}}{J_{eq}(1+Kk_p)}.$$

Thus, the steady-state velocity $\omega_{ss,P}$ is directly proportional to the disturbance torque T_{d0} , and proportional control alone cannot eliminate the steady-state error caused by the disturbance.

4.4.3 Steady-State Velocity with Integral Control

When an integral controller is used (i.e., $k_p=0$ and $k_i
eq 0$), the control signal is:

$$u_m(t) = -rac{k_i}{s}\Omega_m(s)$$

The closed-loop transfer function becomes:

$$\Omega_m(s) = rac{K}{ au s + 1 + rac{K k_i}{s}} U_m(s) + rac{1}{J_{eq}} T_d(s)$$

Using the Final Value Theorem for a step disturbance torque $T_d(s) = rac{T_{d0}}{s}$:

$$\omega_{ss,I} = \lim_{s o 0} s \cdot \Omega_m(s) = \lim_{s o 0} rac{s \cdot rac{1}{J_{eq}} \cdot rac{T_{d0}}{s}}{ au s^2 + s + rac{Kk_i}{s}}$$

Since the denominator has $\frac{Kk_i}{s}$, the steady-state velocity $\omega_{ss,I}=0$. Thus, integral control ensures that the steady-state velocity error due to the disturbance torque is zero, meaning that integral control can eliminate steady-state error from disturbances.