# First results of GLOBUS-M2 ASCOT5 modelling Bakharev N.N.



#### Motivation

#### Tests and benchmarks for future use in

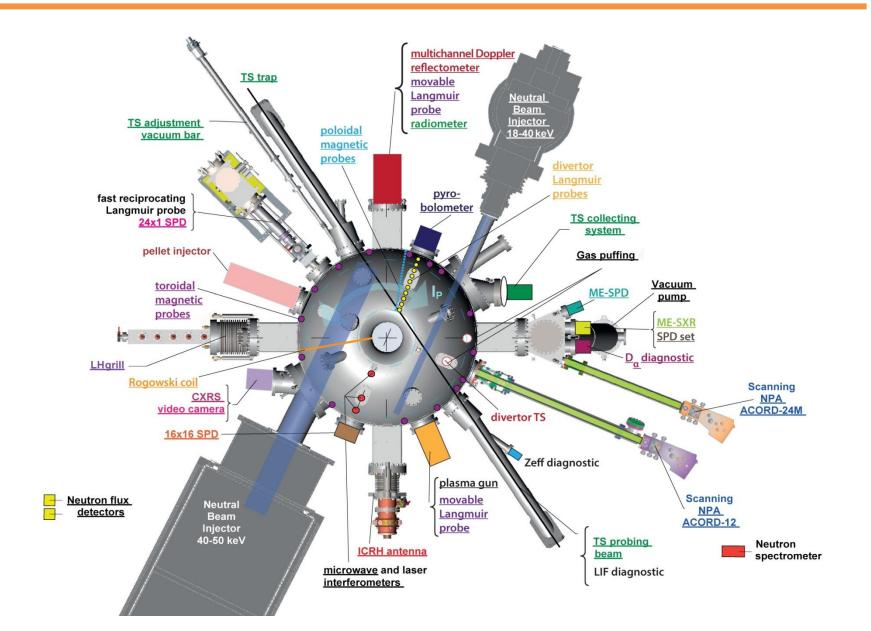
 Modeling of FI distribution in the NBI-heated experiments -calc. of P<sub>abs</sub>, losses, neutron rate, NPA signals, CD etc. (CX required)

 Modeling of the TAE-induced losses/redistribution. (MHD required)

#### Globus-M2

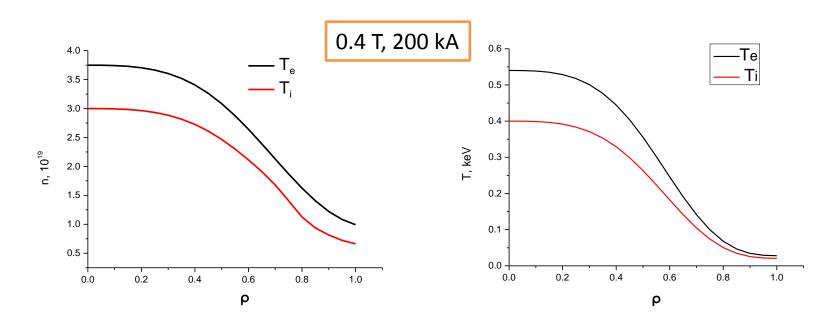
- New machine (first campaign 2019)
- R [cm]/a [cm]= 36/24 = 1.5
- $B_T = 1T$ ,  $I_p = 500 \text{ kA}$
- Extreme P<sub>heat</sub> /V = 6 MW/m<sup>3</sup>
- Diverse diagnostics, heating and CD systems, including 2 x 1 MW NBI, ICRH, LHCD, plasma gun

### Globus-M2

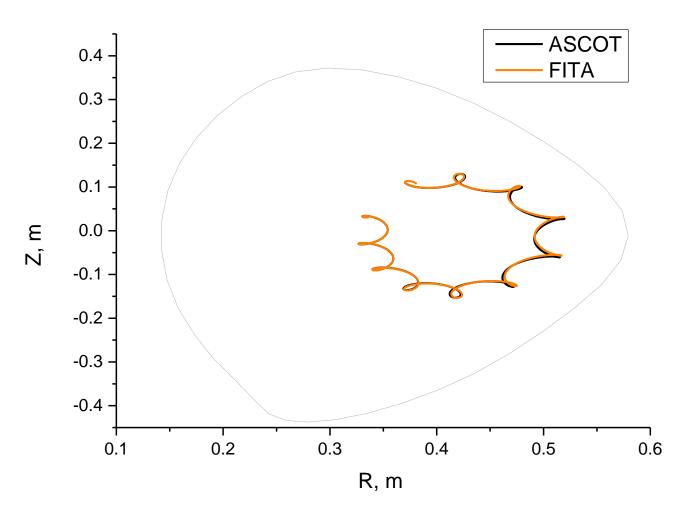


# Modelling overwiev

- 18 keV H (#32994-98)and () (#32952-57) NBI is considered.
- MHD-stable, neutron and NPA measurements exist (CX needed).
- Comparison with NUBEAM and homegrown GO simulations – fast in tracking algorithm(FITA).



# Orbit modelling



✓ Sanity check of Magnetic eq. and marker inputs

## Comparison with NUBEAM

#### **NUBEAM**

- uses GC approx. + Finite Larmor radius adjustment
- Slightly different options

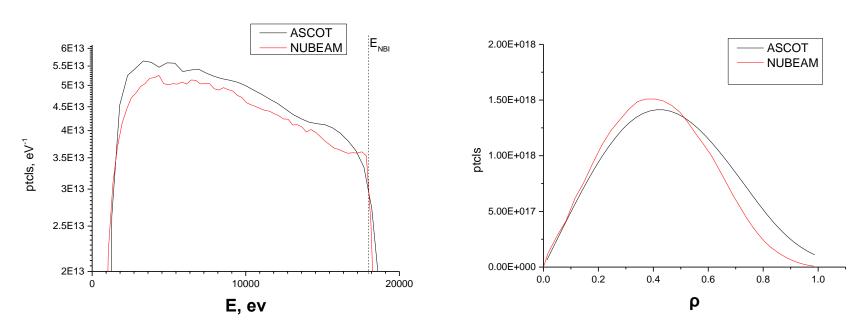
#### **NUBEAM markers -> ASCOT**

#### **Problems:**

- Represented in GC approximation
- Not completely compatible with ASCOT:
- e. g. first order GCTRANS on leads to mu < 0 error and other errors ( $\sim$ 3% of markers)
- NUBEAM uses irregular grid and increased central statistics.
- Not very precise  $\sigma_{CX}$  for such low energies.

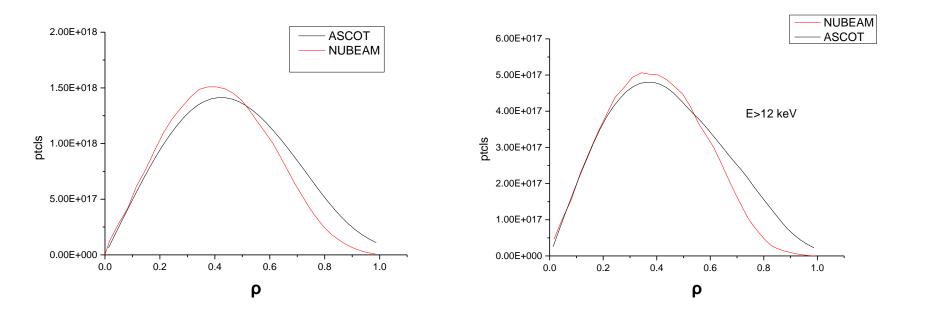
## Comparison with NUBEAM

#### We will consider monoenergetic 18 keV beam



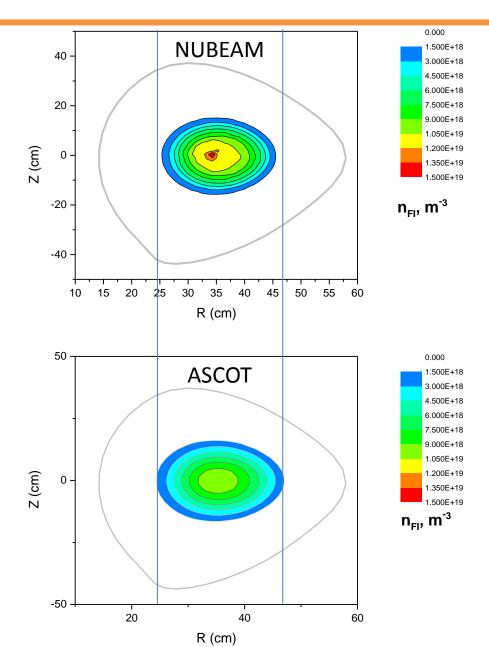
- ✓ Almost similar slowing-down.
- ✓ Density, predicted by NYBEAM is shifted inwards.
- ? Density, predicted by NYBEAM is lower.
- ? Different FI acceleration near E<sub>NBI</sub>

## Comparison with NUBEAM

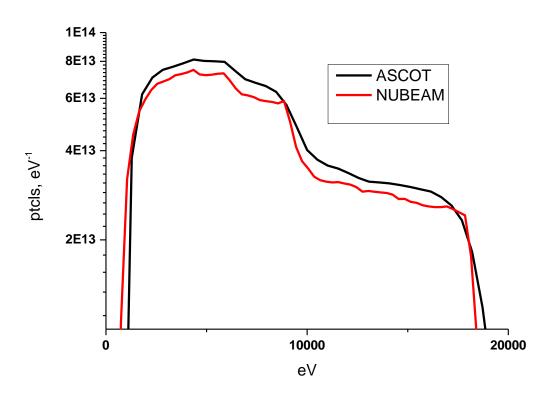


The difference near LCFS is even higher -> GC approx. plays the main role

## Comparison with NUBEAM distributions

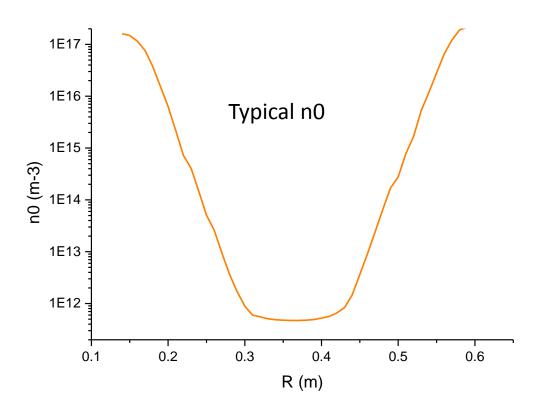


# 3-component beam



Same features.

## **CX** losses



 CX and increase in E<sub>beam</sub> will result in even higher differences with NUBEAM which fails to reproduce the experiments.

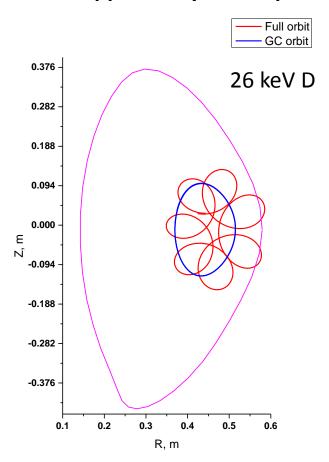
## **FI** losess

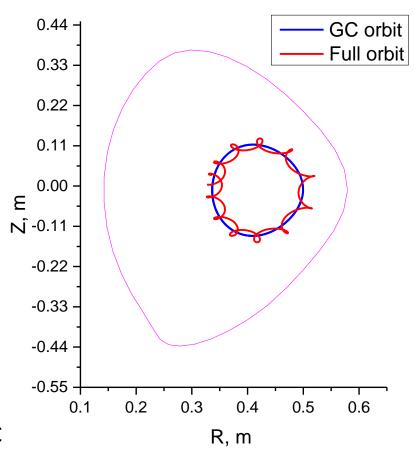
Losses	ASCOT (NO n0)	FITA	NUBEAM (NO n0)
Shine-through, %	-	8	6
wall	5% of markers	-	6% of P
CX, %	-	-	
FO(rho > 1), %	13(NUBEAM markers) 12 (FITA markers)	16	15

FO losses in ASCOT are a bit lower than in other codes

## GC approximation in ASCOT

#### Good opportunity to compare GO and GC, implemented in one code

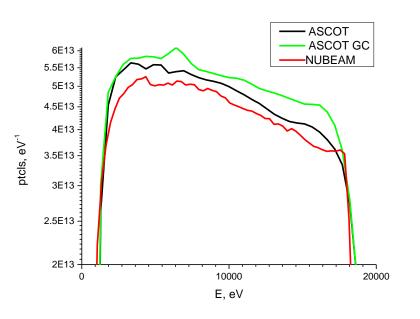


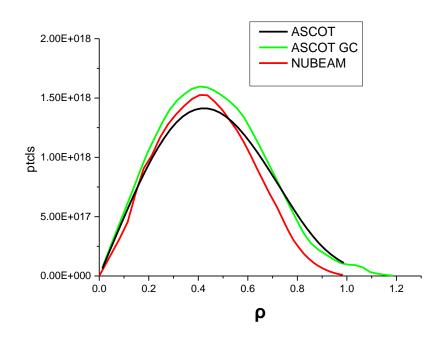


Not all orbits are easy to describe with GC

## GC approximation in ASCOT

#### Good opportunity to compare GO and GC, implemented in one code





- ✓ Higher FI density with GC approximation.
- ✓ NUBEAM is somewhere between ASCOT's GO and GC.

#### **SUMMARY**

- Comparison with NUBEAM and FITA was performed.
- Modelling shows reasonable results, however some peculiarities are unclear
  - -> "smoke" test of input scripts is needed.
- After that we are ready for the real-world simulations:
  - -FI distribution (CX is needed)
  - -TAE or other MHD(MHD is needed)
- Consultations are needed.

#### ASCOT user-experience:

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Great design – transparent (almost a "white-box") and flexible. :3 No backward compatibility. (^{J} \circ \Box \circ) ^{J} (^{\bot} \Box \bot)
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Thanks to the ASCOT team and especially Konsta for his time

