



# ES221 : Mechanics of Solids

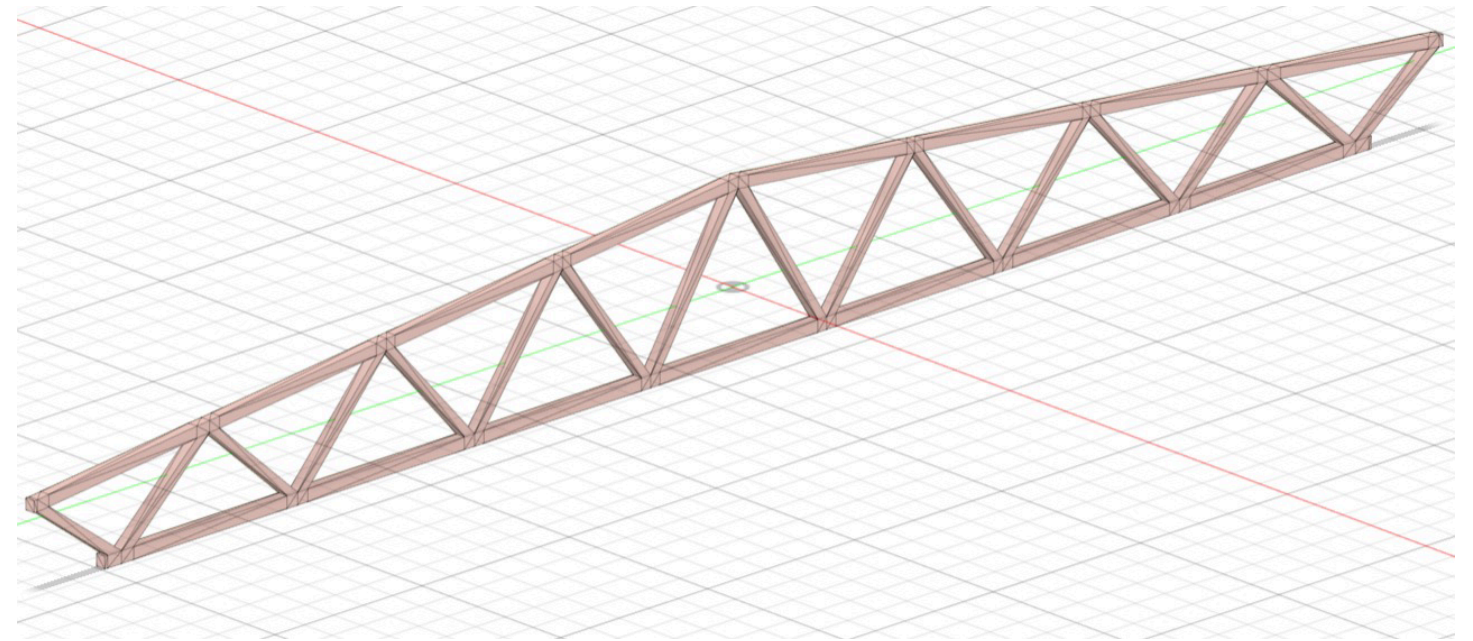
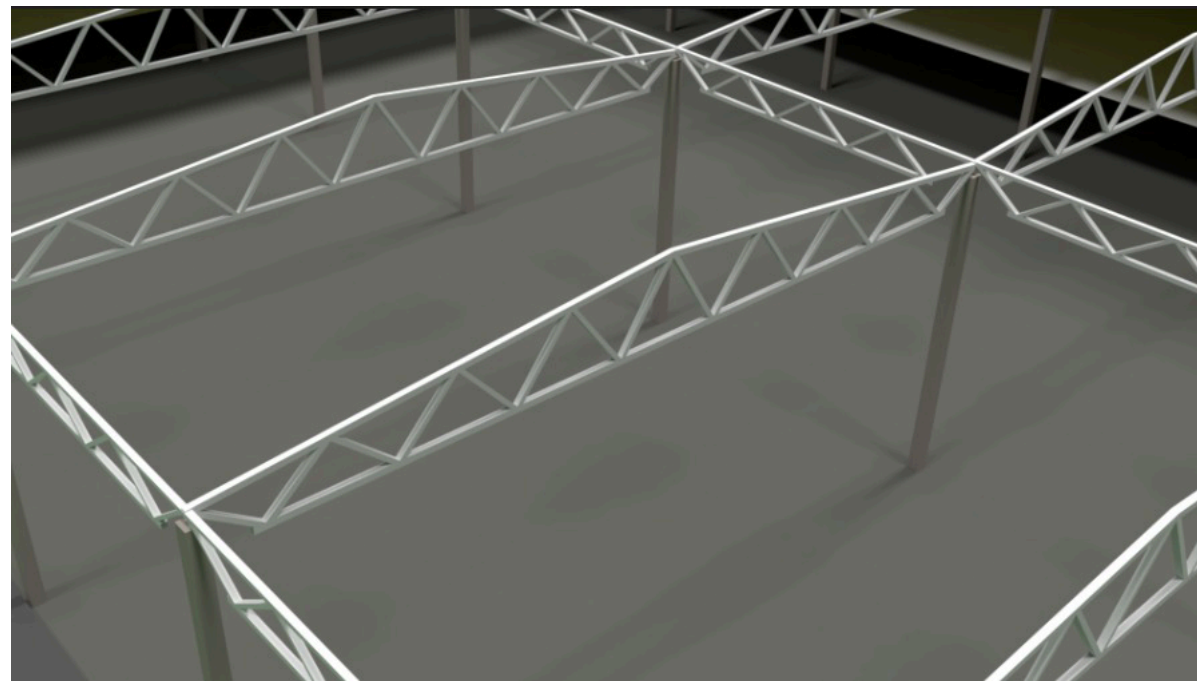
Project : Modeling of Internal Forces and Deflections in Roof Trusses Using FEM

Prof. Harmeet  
Singh

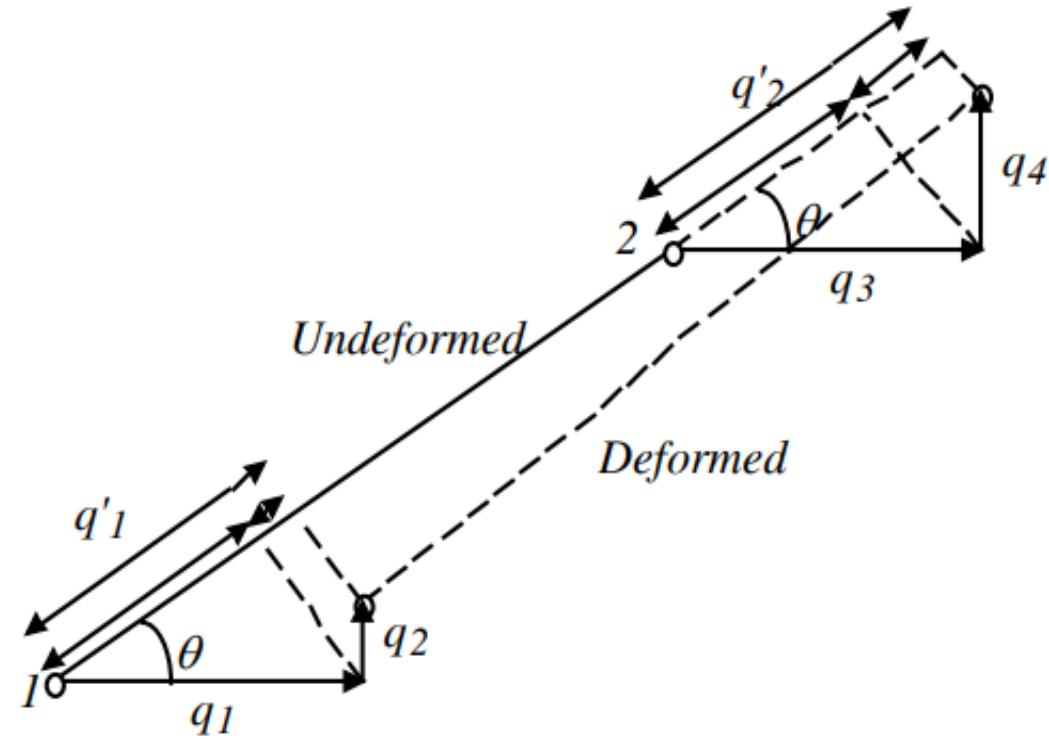
Kavya Shah (23110163)  
Apoorv Rane (23110036)  
Arnav Gogate (23110043)  
Goraksh Bendale (23110118)

# Project Objectives

- Perform 2D truss analysis using FEM in Python
- Compute nodal displacements and axial stresses
- Validate results with ANSYS Workbench
- Understand structural behaviour under static loads



# FEM Background



So final global stiffness matrix for a truss element is:

$$\mathbf{k} = \frac{AE}{L} \begin{bmatrix} c^2 & cs & -c^2 & -cs \\ cs & s^2 & -cs & -s^2 \\ -c^2 & -cs & c^2 & cs \\ -cs & -s^2 & cs & s^2 \end{bmatrix}$$

$$\mathbf{q}' = \begin{bmatrix} q'_1 \\ q'_2 \end{bmatrix}, \quad \mathbf{k}' = \frac{AE}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

Global displacements:

$$\mathbf{q} = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \end{bmatrix}$$

Transformation matrix:

$$\mathbf{M} = \begin{bmatrix} c & s & 0 & 0 \\ 0 & 0 & c & s \end{bmatrix}, \quad \text{with } c = \cos \theta, \quad s = \sin \theta$$

Solve this equation to get the unknown displacements

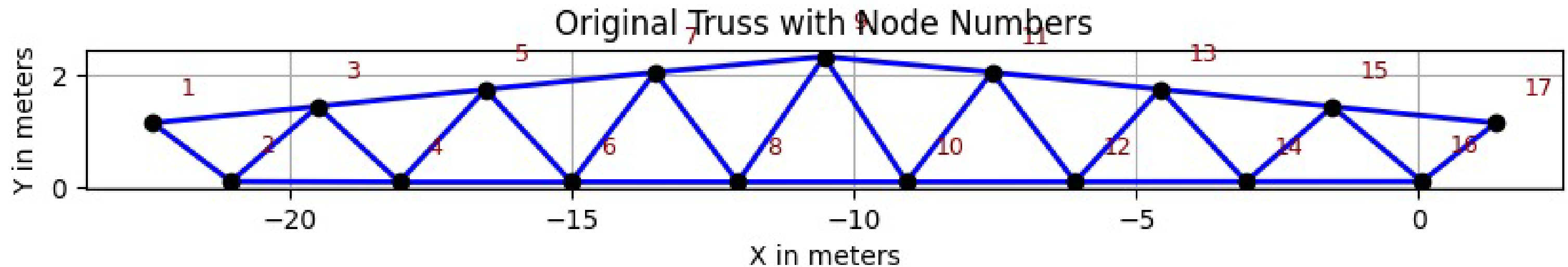
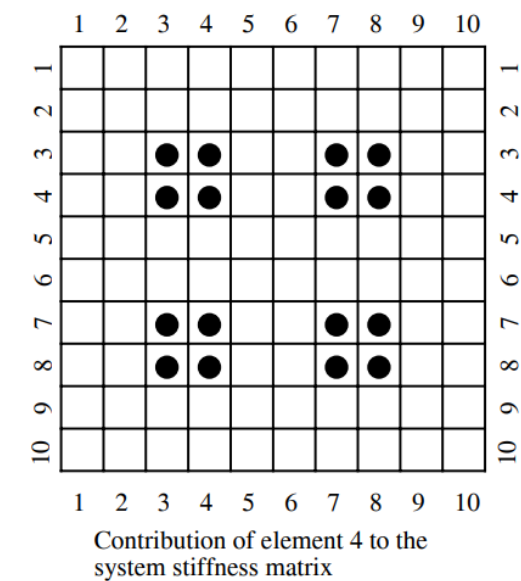
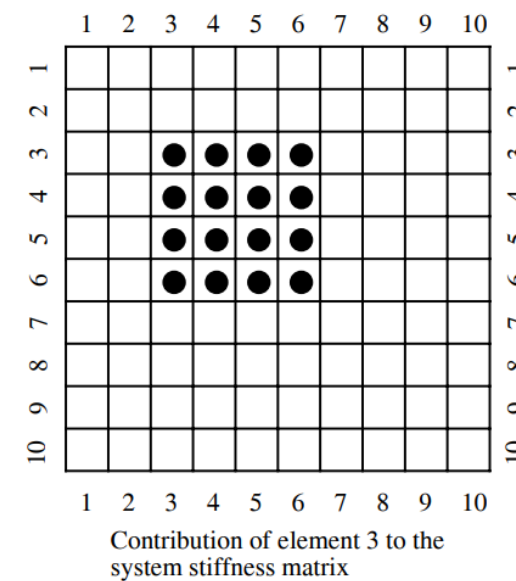
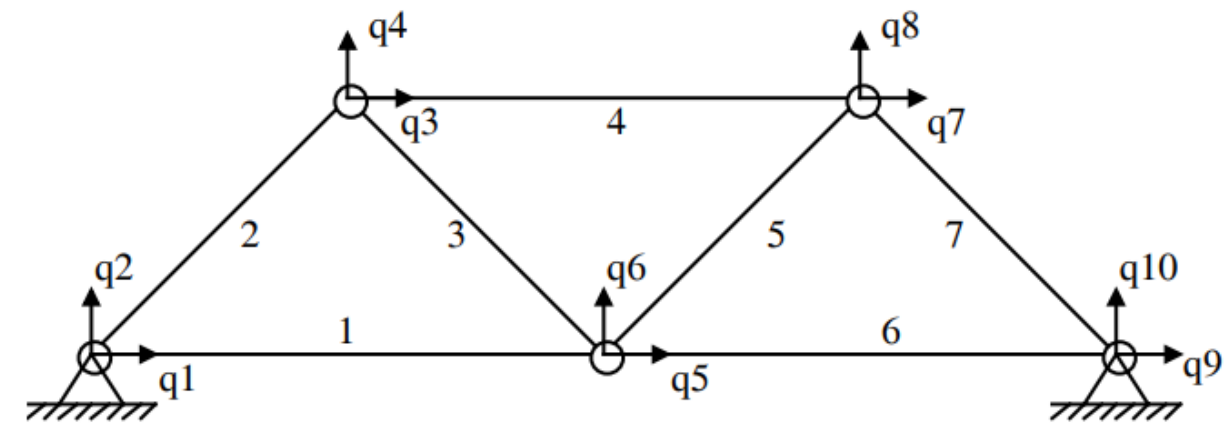
$$\mathbf{KQ} = \mathbf{F}$$

For stresses:

$$\sigma = \frac{E}{L} [-1 \quad 1] \mathbf{q}' = \frac{E}{L} [-1 \quad 1] \mathbf{M} \mathbf{q}$$

# Methodology

- Compute element stiffness matrices
- Assemble global stiffness matrix
- Apply boundary conditions
- Input external forces
- Solve for displacements
- Compute stresses and internal forces



# Results

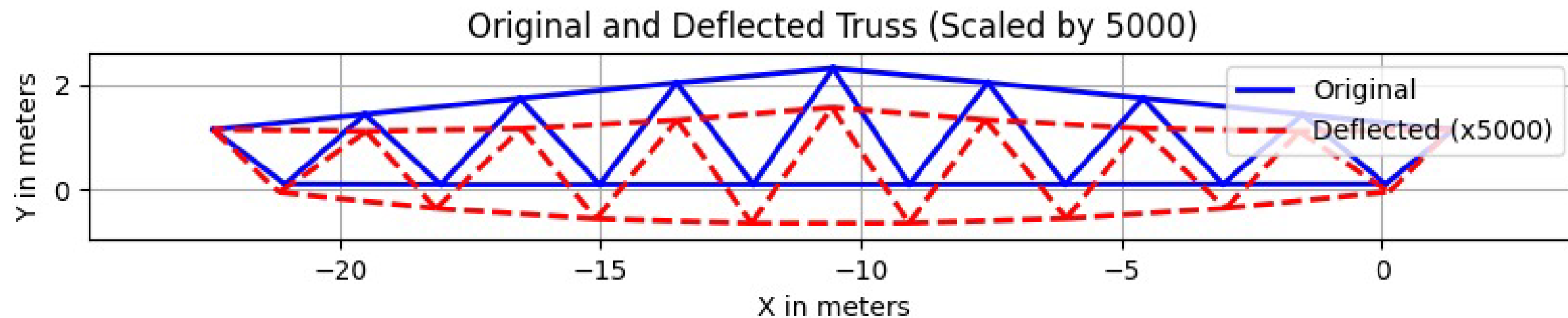
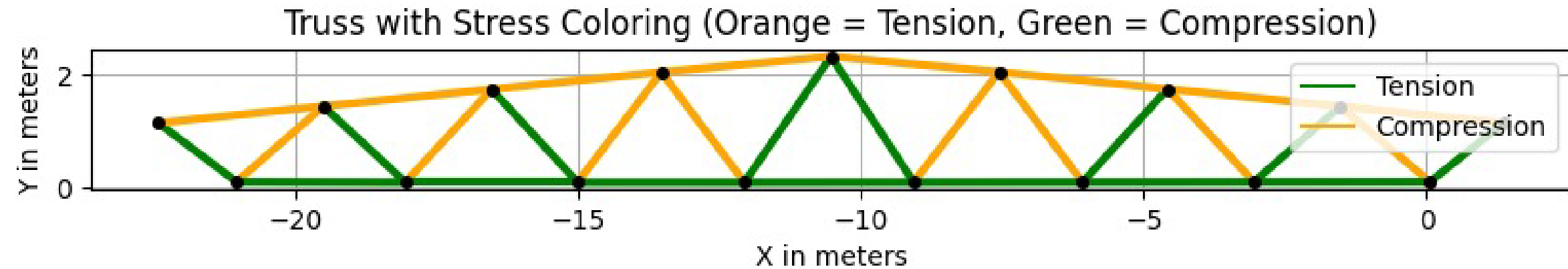
Node	Force (in N)
3	1000
5	2000
7	2000
9	2000
11	2000
13	1000
15	2000

Youngs Modulus: 200E9 GPa

Cross Section Area: 0.05 m<sup>2</sup>

Boundary Conditions : Node 1 (Fixed Support)

Node 17 (Roller Support)



# Results

Node	X Displacement (m)	Y Displacement (m)
1	0.000000	0.000000
2	-0.000022	-0.000032
3	0.000005	-0.000066
4	-0.000019	-0.000092
5	0.000005	-0.000115
6	-0.000013	-0.000132
7	0.000001	-0.000143
8	-0.000007	-0.000149
9	-0.000004	-0.000150
10	-0.000001	-0.000149
11	-0.000009	-0.000143
12	0.000005	-0.000131
13	-0.000012	-0.000115
14	0.000011	-0.000093
15	-0.000012	-0.000065
16	0.000015	-0.000030
17	-0.000008	0.000000

Element	Connection	Stress (Pa)	Force (N)
1	1 ↔ 2	172240.70	8612.04
2	2 ↔ 3	-156878.21	-7843.91
3	3 ↔ 4	94232.89	4711.64
4	4 ↔ 5	-88320.51	-4416.03
5	5 ↔ 6	23308.06	1165.40
6	6 ↔ 7	-21320.65	-1066.03
7	7 ↔ 8	-24346.03	-1217.30
8	8 ↔ 9	23107.48	1155.37
9	9 ↔ 10	17057.89	852.90
10	10 ↔ 11	-18053.61	-902.68
11	11 ↔ 12	-28685.83	-1434.29
12	12 ↔ 13	30784.35	1539.22
13	13 ↔ 14	-67793.40	-3389.67
14	14 ↔ 15	75662.34	3783.12
15	15 ↔ 16	-168758.69	-8437.94
16	16 ↔ 17	174156.58	8707.83
17	17 ↔ 15	-136746.24	-6837.31
18	15 ↔ 13	-323845.93	-16192.30
19	13 ↔ 11	-391111.35	-19555.57
20	11 ↔ 9	-397086.47	-19854.32
21	9 ↔ 7	-400835.32	-20041.77
22	7 ↔ 5	-402923.84	-20146.19
23	5 ↔ 3	-326663.86	-16333.19
24	3 ↔ 1	-138265.69	-6913.29
25	2 ↔ 4	255929.28	12796.46
26	4 ↔ 6	385060.45	19253.02
27	6 ↔ 8	413798.89	20689.94
28	8 ↔ 10	386044.04	19302.20
29	10 ↔ 12	406467.67	20323.38
30	12 ↔ 14	368400.48	18420.02
31	14 ↔ 16	265270.69	13263.54

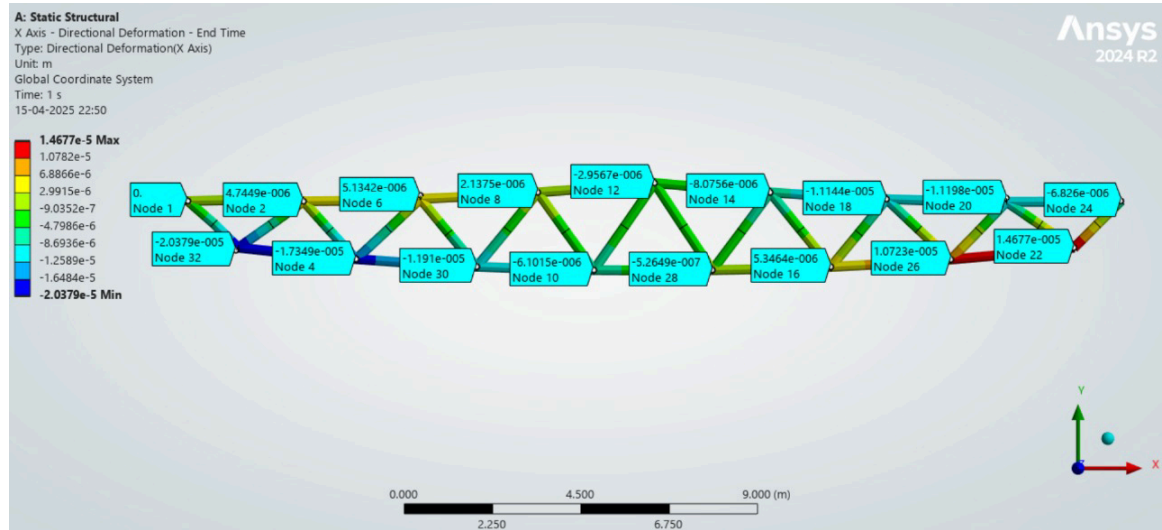
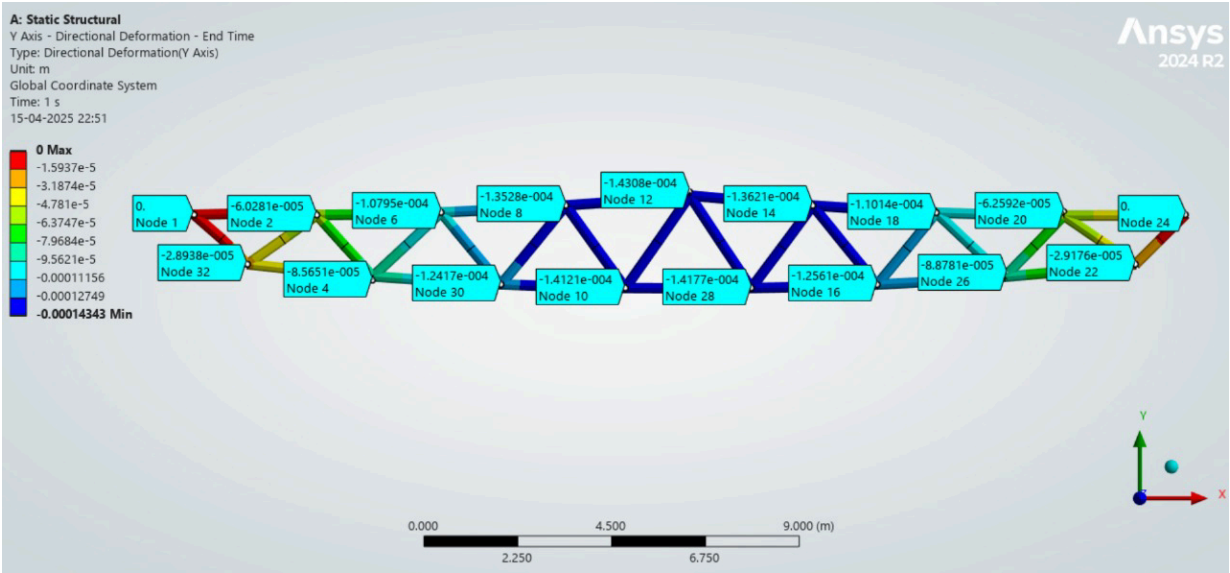
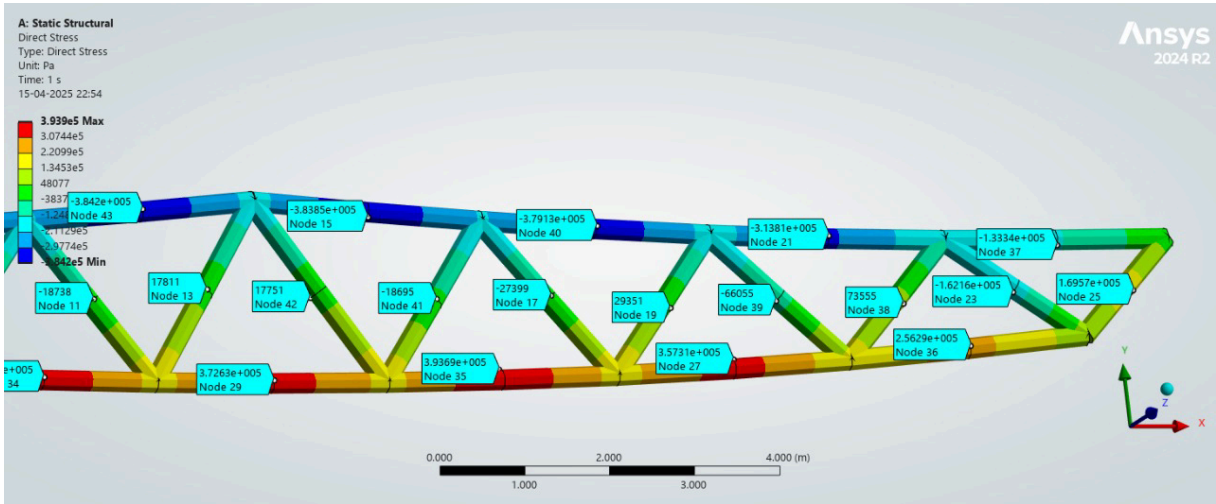
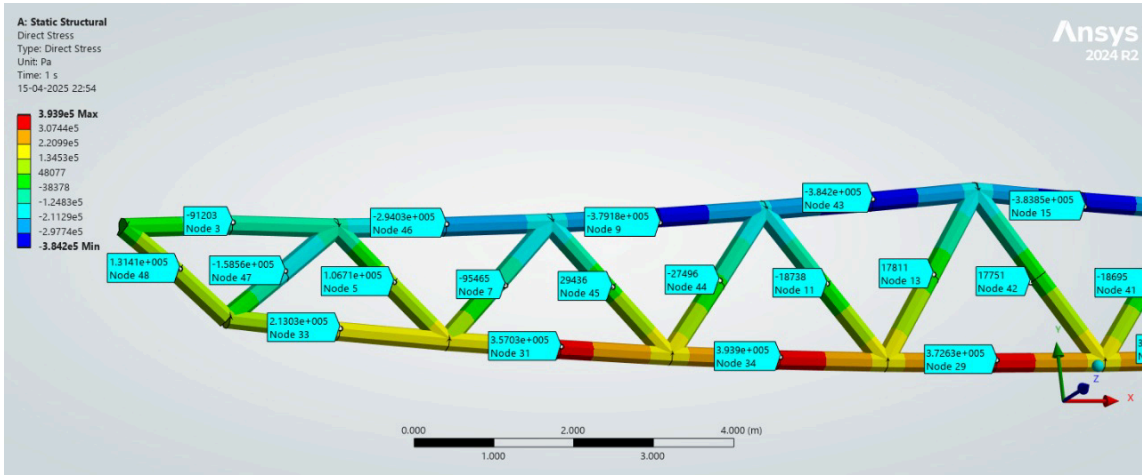


# Validation with ANSYS

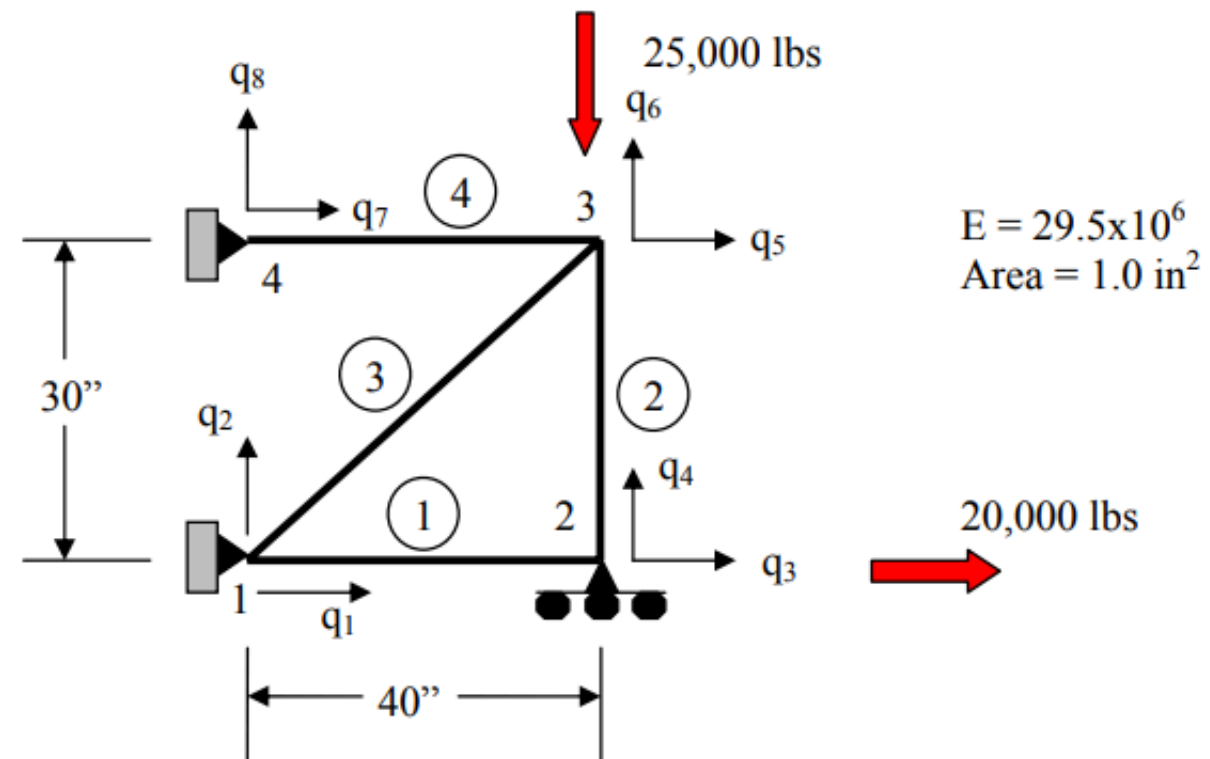
X- deformation deviation: ~15.6%

Y- deformation deviation: ~5.2%

Stress deviation: ~10.1%



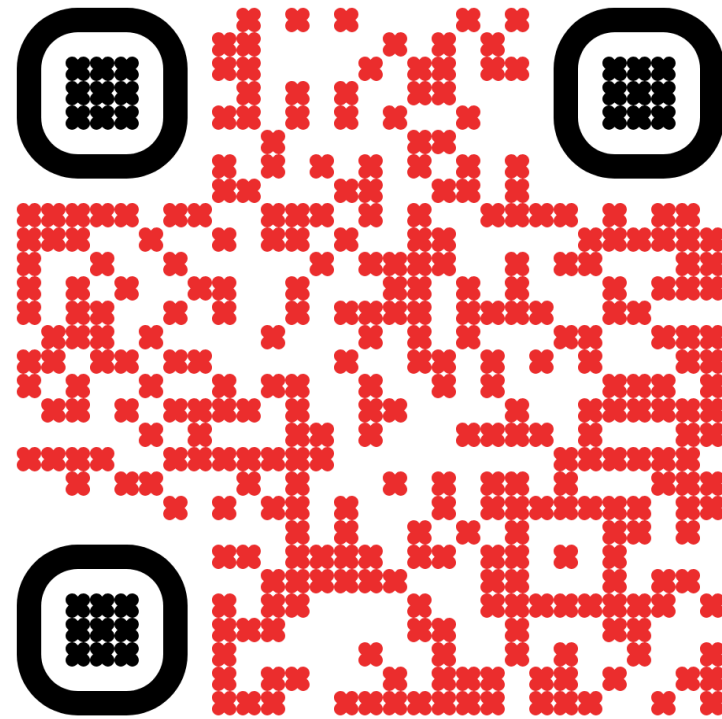
# Demo



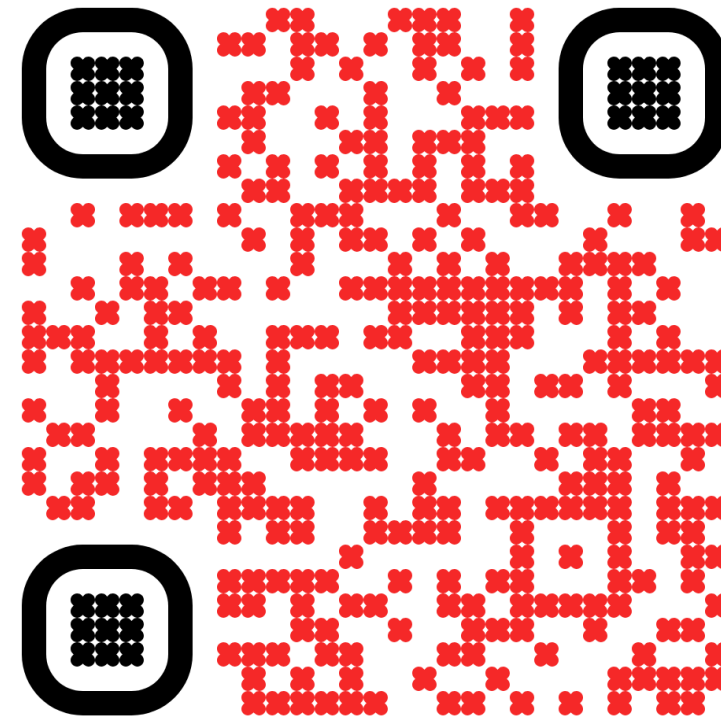
$$\begin{Bmatrix} q_3 \\ q_5 \\ q_6 \end{Bmatrix} = \begin{Bmatrix} 27.12 \times 10^{-3} \\ 5.65 \times 10^{-3} \\ -22.25 \times 10^{-3} \end{Bmatrix} \quad \text{inches}$$



# Questions?



Source Code



Project Report