



01OAIQD – Dynamic Design of Machines

Academic year 2019-2020

Discrete linear systems

Emission date: October 18, 2019

GROUP:

EXERCISE 1

An undamped system (presented in Figure 1) is made with two torsion bars (negligible inertia) and two flywheels. One of the flywheels (number 2) is loaded with a harmonic excitation $M(t) = M_0 \sin(\omega t)$. Analyze the dynamic behavior of the system by computing:

- natural frequencies and corresponding mode shapes;
- potential and kinetic energy associated to the different subsystems for each mode.
- maximum shear stresses in the torsion bars using the configuration space and the state space approach.

Data: $G = 7.7 \cdot 10^{10} \text{ N/m}^2$, $l_I = 0.5 \text{ m}$, $l_{II} = 1.0 \text{ m}$, $d_I = 12.8 \text{ mm}$, $d_{II} = 14.1 \text{ mm}$, $J_{p1} = 6 \text{ kgm}^2$, $J_{p2} = 4 \text{ kgm}^2$, $M_0 = 100 \text{ Nm}$, $\omega = 8 \text{ rad/s}$.

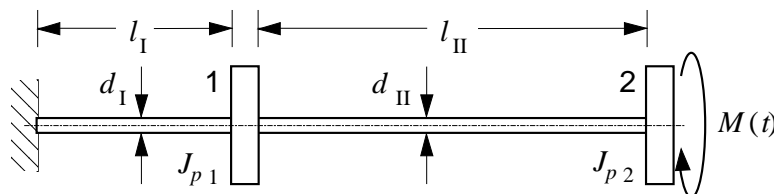


Figure 1

$\omega_1 = 5.6 \text{ rad/s}$, $\omega_2 = 12.7 \text{ rad/s}$, $\{\theta\}_1 = \{0.58 \ 1\}^T$, $\{\theta\}_2 = \{1 \ -0.87\}^T$, $\tau_I = 390 \text{ MPa}$, $\tau_{II} = 16 \text{ MPa}$.

EXERCISE 2

Consider a 1 dof system having a stiffness $k = 10 \text{ kN/m}$ and a mass $m = 4 \text{ kg}$. A logarithmic decrement $\delta = 0.3$ was measured on it. Additionally the system is subject to a harmonic excitation $F(t) = F_0 \sin(\omega t)$ where $F_0 = 20 \text{ N}$.

Compute the:

amplitude, phase angle, real part, imaginary part relative to an excitation frequency

1) $\omega = 0.9\omega_n$

2) $\omega = 2.5\omega_n$

assuming a viscous (a) and hysteretic (b) damping model.

a) $\omega = 0.9 \omega_n$: $x_0 = 9.59 \cdot 10^{-3} \text{ m}$, $\Phi = -24.3^\circ$, $\Re = 8.74 \cdot 10^{-3} \text{ m}$, $\Im = -3.95 \cdot 10^{-3} \text{ m}$;

$$\omega = 2.5\omega_n : x_0 = 3.81 \cdot 10^{-4} \text{ m}, \Phi = -177.4^\circ, \Re = -3.80 \cdot 10^{-4} \text{ m}, \Im = -1.73 \cdot 10^{-5} \text{ m};$$

$$\text{b) } \omega = 0.9\omega_n : x_0 = 9.41 \cdot 10^{-3} \text{ m}, \Phi = -26.7^\circ, \Re = 8.40 \cdot 10^{-3} \text{ m}, \Im = -4.22 \cdot 10^{-3} \text{ m};$$

$$\omega = 2.5\omega_n : x_0 = 3.81 \cdot 10^{-4} \text{ m}, \Phi = -179.0^\circ, \Re = -3.81 \cdot 10^{-4} \text{ m}, \Im = -6.93 \cdot 10^{-6} \text{ m}.$$

EXERCISE 3

The undamped system represented in Figure 2 is characterized by two bars having a diameter d (with negligible inertia) and two masses. A harmonic excitation $F(t) = F_0 \sin(\omega t)$ is applied on the mass 2.

It is requested to:

- compute the amplitude of the oscillations by adopting the modal analysis,
- compute the maximum stress in bar 1 and bar 2.

Data: $l_1 = 0.4 \text{ m}$, $l_2 = 0.2 \text{ m}$, $d = 0.02 \text{ m}$, $E = 2.06 \cdot 10^{11} \text{ Pa}$, $m_1 = 50 \text{ kg}$, $m_2 = 20 \text{ kg}$, $\omega = 1700 \text{ rad/s}$, $F_0 = 2000 \text{ N}$.

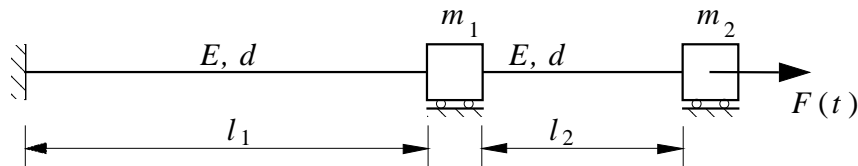


Figure 2

$$x_{0,1} = -4.59 \cdot 10^{-5} \text{ m}, x_{0,2} = -4.89 \cdot 10^{-5} \text{ m}.$$

EXERCISE 4

Consider the belt drive system represented in Figure 3. The main data are reported in Table 1.

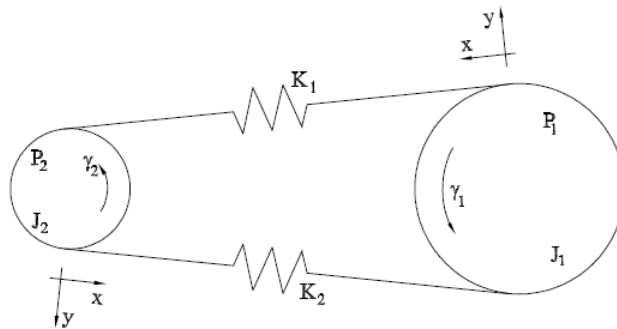


Figure 3.

$J_1 = 0.2$	Kgm^2
$J_2 = 0.15$	Kgm^2
$r_1 = 0.07$	m
$r_2 = 0.04$	m
$l = 0.5$	m

Table 1.

The Young modulus E of the belt in the axial direction is equal to $2.02 \cdot 10^9 \text{ N/m}^2$ while the transversal section area A is equal to $4.26 \cdot 10^{-5} \text{ m}^2$.

It is requested to compute the two natural frequencies of the systems.

$$\omega_1 = 0 \text{ Hz}, \omega_2 = 17.5 \text{ Hz},$$

EXERCISE 5

For the two degree of freedom system depicted in Figure 4

- determine the equations of dynamic equilibrium using the Lagrangian approach,
- arrange the equations in matrix form,
- find the natural frequencies and the mode shapes of the system,
- compute the modal matrices (modal mass and modal stiffness),
- mass-normalize the mode shapes,
- Check the mass-orthogonality of the eigenvectors.

Make the calculation using the following two sets of data

$$k_1 = k_2 = 400000 \text{ N/m} \quad k_3 = 800000 \text{ N/m}; \quad m_1 = m_2 = 2 \text{ kg}$$

$$k_1 = 400000 \text{ N/m} \quad k_2 = 200000 \text{ N/m} \quad k_3 = 800000 \text{ N/m}; \quad m_1 = 2 \text{ kg} \quad m_2 = 2.5 \text{ kg}$$

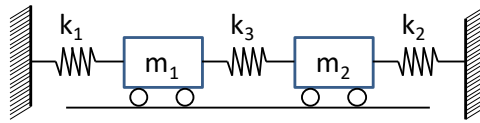


Figure 4

EXERCISE 6

A dynamic system with three flywheels is depicted in Figure 5. Neglecting the inertia of the shafts

- write the dynamic equations of equilibrium and
- determine natural frequencies and mode shapes of the system (torsional motion only)
-

$$d_1 = 30 \text{ mm} \quad d_2 = 20 \text{ mm}$$

$$L_1 = 100 \text{ mm} \quad L_2 = 80 \text{ mm}$$

$$m_1 = 1 \text{ kg} \quad m_2 = 0.5 \text{ kg}$$

$$D_1 = 120 \text{ mm} \quad D_2 = 60 \text{ mm} \quad D_3 = 100 \text{ mm}$$

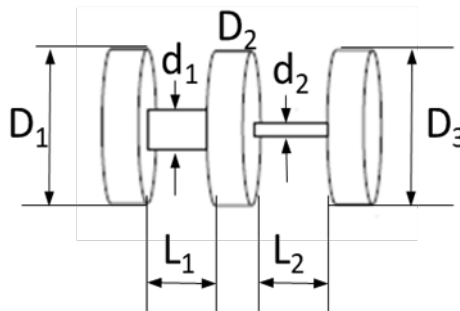


Figure 5

EXERCISE 7

For the two degree of freedom depicted in the Figure 6

- determine the equations of dynamic equilibrium (Lagrange approach).
- Find natural frequencies and modal shapes of the system.
- Use the eigenvectors to compute the modal matrices.

Data

$$k_1 = k_2 = 20000 \text{ N/m}$$

$$m = 950 \text{ kg} \quad IC = 1400 \text{ kg m}^2$$

$$L_1 = 1.0 \text{ m} \quad L_2 = 1.5 \text{ m}$$

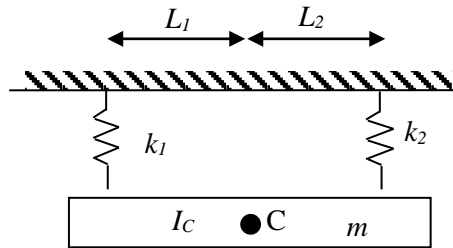


Figure 6.

$\omega_1=5.9 \text{ rad/s}, \omega_2=7.3 \text{ rad/s}, (Y/A)_1=-1.55, (Y/A)_2=0.95]$

EXERCISE 8

The output to a harmonic excitation of a vibrating structure is described only by the first mode:

$\{q\}_1 = \{1/\sqrt{5} \quad 2/\sqrt{5}\}^T$. Take into account that the modal mass is 2 kg, the modal stiffness is 5000 N/m, the modal force is $10+(2U+P)/3$ N. The excitation frequency is equal to $0.8\omega_1$, and the damping is negligible. Compute the amplitude of oscillation of the first degree of freedom.

$x_{01} = \dots\dots\dots \text{ m}$

EXERCISE 9

A two degrees of freedom undamped system has the following modal shapes:

$\{q\}_1 = \{1 \quad 0.8165\}^T, \{q\}_2 = \{1 \quad -0.8165\}^T$.

The stiffness matrix, the mass matrix and the vector of the excitation forces are:

$$[K] = 10^3 \begin{bmatrix} 10 & -8 \\ -8 & 15 \end{bmatrix} \frac{\text{N}}{\text{m}}, \quad [M] = \begin{bmatrix} 4 & 0 \\ 0 & 6 \end{bmatrix} \text{ kg}, \quad \{F\} = 10^3 \left(1 + \frac{2U+P}{20}\right) \begin{Bmatrix} 5 \\ 8 \end{Bmatrix} \text{ N}$$

Compute the modal force corresponding to the first mode.

$\bar{F}_1 = \dots\dots\dots \text{ N}$

EXERCISE 10

Consider a vibrating system. The dynamic response is characterized using only the first mode

$\{q\}_1 = \{1/\sqrt{7} \quad 2/\sqrt{7}\}^T$. The modal mass is equal to 3 kg, the modal stiffness is equal to 3000 N/m, the modal force is equal to $10+(2U+P)/3$ N. The excitation frequency is equal to $0.8\omega_1$, the damping is negligible.

Compute the amplitude of oscillation of the degree of freedom n° 1.

$x_{01} = \dots\dots\dots \text{ m}$

MULTIPLE CHOICE PROBLEMS: select the right answer. Explain in few sentences (eventually with the support of graphs, schems and formulas) the reason of the correct answer.

- 1) The Lagrange equations can be written
 - [A] only for undamped systems,
 - [B] only for damped and undamped systems subject to vibration
 - [C]** also for non conservative systems.
- 2) For a damped multi degree of freedom system it is not possible to obtain a complete modal decoupling. This sentence is
 - [A] Always correct
 - [B] Never correct
 - [C]** Not correct only in some cases.
- 3) An external excitation applied to a dynamic system is considered quasi static if
 - [A] Its frequency is much lower with respect to the lowest natural frequency of the dynamic system.
 - [B]** Its frequency is equal or lower with respect to the lowest natural frequency of the dynamic system.
 - [C] Its frequency is equal or higher with respect to the lowest natural frequency of the dynamic system.
- 4) A gyroscopic matrix present in the equations of motion of a mechanical system
 - [A] Is symmetric
 - [B]** Is skew symmetric and may never cause instability.
 - [C] Is skew symmetric and may cause instability.
- 5) A mechanical system is characterized by a damping factor $\xi=0.6$. This means that its structure is characterized mainly by
 - [A] light alloys,
 - [B] steel alloys,
 - [C]** a dedicated damping device.