

Controllability & Observability

EPD 30.114 ADVANCED FEEDBACK & CONTROL

Important Properties of LTI Systems

- Controllability and Observability represent 2 major concepts of Modern Control System theory. They were introduced by R. E. Kalman in 1959/60.
- **CONTROLLABILITY**
 - In order to be able to do whatever we want with the dynamic system under control input, the system must be controllable
- **OBSERVABILITY**
 - In order to see what is going on inside the system under observation (the internal states), the system must be observable
- Concepts of controllability and observability are derived from linear system theory, so knowledge of linear algebra is key
- Controllability and observability are dual concepts and should be understood as a whole rather than separate properties.

R. Kalman, "On the general theory of control systems," *IRE Transactions on Automatic Control*, vol. 4, no. 3, pp. 110-110, Dec 1959.



Complete State Controllability

- Consider the state-space system (scalar input with n states):

$$\mathbf{A} \text{ is an } n \times n \text{ matrix} \quad \dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$$

- We know the solution is: $\mathbf{x}(t) = e^{\mathbf{A}t}\mathbf{x}(0) + \int_0^t e^{\mathbf{A}(t-\tau)}\mathbf{B}u(\tau)d\tau$
- A system is said to be state controllable at $t=t_0$ if it is possible to construct an unconstrained control signal that will transfer an initial state to any final state in a finite time interval $t_0 \leq t \leq t_f$. If every state is controllable, the system is said to be **completely state controllable**.
- Without loss of generality, let's assume we are going from an unknown initial state location to the origin (zero) at $t=t_f$

$$\mathbf{x}(t_f) = e^{\mathbf{A}t_f}\mathbf{x}(0) + \int_0^{t_f} e^{\mathbf{A}(t_f-\tau)}\mathbf{B}u(\tau)d\tau = \mathbf{0}$$

$$\mathbf{x}(0) = -e^{-\mathbf{A}t_f} \int_0^{t_f} e^{\mathbf{A}(t_f-\tau)}\mathbf{B}u(\tau)d\tau$$

$$= -\int_0^{t_f} e^{-\mathbf{A}\tau}\mathbf{B}u(\tau)d\tau$$

$$e^{-\mathbf{A}\tau} = \mathbf{I} - \mathbf{A}\tau + \frac{\mathbf{A}^2\tau^2}{2!} - \frac{\mathbf{A}^3\tau^3}{3!} + \frac{\mathbf{A}^4\tau^4}{4!} - \dots = \sum_{k=0}^{\infty} \frac{(-\mathbf{A})^k \tau^k}{k!} = \boxed{\sum_{k=0}^{\infty} \beta_k(\tau) \mathbf{A}^k} = \boxed{\sum_{k=0}^{n-1} \alpha_k(\tau) \mathbf{A}^k}$$

Cayley-Hamilton Theorem

- The Cayley-Hamilton theorem states that the matrix \mathbf{A} satisfies its own characteristic equation (\mathbf{A} is an $n \times n$ matrix)

- c.e:
$$|\lambda \mathbf{I} - \mathbf{A}| = \lambda^n + a_1 \lambda^{n-1} + \dots + a_{n-1} \lambda + a_n = 0$$

- By C-H theorem:
$$\mathbf{A}^n + a_1 \mathbf{A}^{n-1} + \dots + a_{n-1} \mathbf{A} + a_n \mathbf{I} = 0$$

- Illustration:
$$\mathbf{A} = \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix}$$

$$|\lambda \mathbf{I} - \mathbf{A}| = \mathbf{A}^2 - 2\mathbf{A} + \mathbf{I} = 0$$

$$\begin{aligned} \mathbf{A}^2 - 2\mathbf{A} + \mathbf{I} &= \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} - 2 \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \\ &= \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix} - 2 \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} 0 & 0 \\ 0 & 0 \end{bmatrix} \end{aligned}$$

Cayley-Hamilton Theorem

- So why is $\mathbf{A}^n + a_1\mathbf{A}^{n-1} + \dots + a_{n-1}\mathbf{A} + a_n\mathbf{I} = 0$ helpful?

- Let's consider a 2 x 2 \mathbf{A} matrix

$$\lambda^2 + b\lambda + c = 0$$

- The c.e. is

$$\mathbf{A}^2 + b\mathbf{A} + c\mathbf{I} = 0$$

- By C-H theorem:

$$\mathbf{A}^2 = -b\mathbf{A} - c\mathbf{I}$$

- Extending this:

$$\mathbf{A}^3 = \mathbf{A}^2\mathbf{A} = (-b\mathbf{A} - c\mathbf{I})\mathbf{A}$$

$$= -b\mathbf{A}^2 - c\mathbf{A}$$

$$= -b(-b\mathbf{A} - c\mathbf{I}) - c\mathbf{A} = (b^2 - c)\mathbf{A} + bc\mathbf{I}$$

- And so on.... You can always express higher powers of \mathbf{A} in terms of \mathbf{A} and \mathbf{I}

- For an $n \times n$ \mathbf{A} matrix

- Powers of \mathbf{A} n and above can be expressed as a linear combinations of \mathbf{I} , \mathbf{A} , $\mathbf{A}^2, \dots, \mathbf{A}^{n-1}$.

- This helps for matrix exponentials:

$$e^{\mathbf{A}t} = \mathbf{I} + \mathbf{A}t + \frac{\mathbf{A}^2 t^2}{2!} + \frac{\mathbf{A}^3 t^3}{3!} + \frac{\mathbf{A}^4 t^4}{4!} + \dots = \sum_{k=0}^{\infty} \frac{\mathbf{A}^k t^k}{k!} = \sum_{k=0}^{n-1} \alpha_k(t) \mathbf{A}^k$$

Complete State Controllability

- Back to Controllability: $\mathbf{x}(0) = -\int_0^{t_f} e^{-\mathbf{A}\tau} \mathbf{B}u(\tau) d\tau$
- Using C-H theorem: $e^{-\mathbf{A}\tau} = \sum_{k=0}^{\infty} \frac{(-\mathbf{A})^k \tau^k}{k!} = \sum_{k=0}^{n-1} \alpha_k(\tau) \mathbf{A}^k$
- Combining:
$$\begin{aligned} \mathbf{x}(0) &= -\int_0^{t_f} \sum_{k=0}^{n-1} \alpha_k(\tau) \mathbf{A}^k \mathbf{B}u(\tau) d\tau \\ &= -\sum_{k=0}^{n-1} \left[\mathbf{A}^k \mathbf{B} \int_0^{t_f} \alpha_k(\tau) u(\tau) d\tau \right] \end{aligned}$$
- Defining: $\int_0^{t_f} \alpha_k(\tau) u(\tau) d\tau = \beta_k$, $\mathbf{x}(0) = -\sum_{k=0}^{n-1} \mathbf{A}^k \mathbf{B} \beta_k = \begin{bmatrix} \mathbf{B} & \mathbf{AB} & \cdots & \mathbf{A}^{n-1} \mathbf{B} \end{bmatrix} \begin{bmatrix} \beta_0 \\ \beta_1 \\ \vdots \\ \beta_{n-1} \end{bmatrix}$
- If the system is completely state controllable, then given any (non-zero) initial state $\mathbf{x}(0)$, the above equation must be satisfied
 - This requires the rank of $\begin{bmatrix} \mathbf{B} & \mathbf{AB} & \cdots & \mathbf{A}^{n-1} \mathbf{B} \end{bmatrix}$ ($n \times n$ matrix) to be n (full rank)
 - The vectors of $\mathbf{B}, \mathbf{AB}, \dots, \mathbf{A}^{n-1} \mathbf{B}$ are linearly independent
- Result also holds when control input is an r -dimension vector $\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu}$

$\mathbf{C}_o = \begin{bmatrix} \mathbf{B} & \mathbf{AB} & \cdots & \mathbf{A}^{n-1} \mathbf{B} \end{bmatrix}_{n \times nr}$ ($n \times nr$ matrix) is called the **Controllability Matrix**

Exercise on Controllability

- For the 2 following systems, comment on state controllability.

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 1 & 1 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$$

$$\begin{aligned} \mathbf{C}_o &= [\mathbf{B} \quad \mathbf{AB}] \\ &= \begin{bmatrix} 1 & 1 \\ 0 & 0 \end{bmatrix} \end{aligned}$$

\mathbf{C}_o is singular. System not completely state controllable.

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 1 & 1 \\ 2 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$

$$\begin{aligned} \mathbf{C}_o &= [\mathbf{B} \quad \mathbf{AB}] \\ &= \begin{bmatrix} 0 & 1 \\ 1 & -1 \end{bmatrix} \end{aligned}$$

\mathbf{C}_o is non-singular. System completely state controllable.

Extra Exercise on Controllability

- If the state and input matrices of a system are given as follows, can you comment on the state controllability?

$$\mathbf{A} = \begin{bmatrix} -2 & 0 & 0 \\ 0 & -3 & 0 \\ 1 & 1 & -1 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} -1 \\ 2 \\ 0 \end{bmatrix}$$

$$\mathbf{C}_o = \begin{bmatrix} -1 & 2 & -4 \\ 2 & -6 & 18 \\ 0 & 1 & -5 \end{bmatrix}$$

\mathbf{C}_o is rank 2 (not full rank =3) . System not completely state controllable.

Output Controllability

- In practical design of control systems, it is usually desired to control the output rather than the internal states of the system
- Complete state controllability is neither necessary nor sufficient for controlling the output of the system. A separate criterion is needed.
- For a system:
 $\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu}$
 $\mathbf{y} = \mathbf{Cx} + \mathbf{Du}$
 \mathbf{x} : n vector
 \mathbf{u} : r -vector
 \mathbf{y} : m -vector
 \mathbf{A} : $n \times n$ matrix
 \mathbf{B} : $n \times r$ matrix
 \mathbf{C} : $m \times n$ matrix
 \mathbf{D} : $m \times r$ matrix
- System is said to be completely output controllable if it is possible to construct an unconstrained control vector that will transfer any given output to final output in a finite time interval
- It can be proven that if the following condition is met, the system is completely output controllable

$$\mathbf{O}_C = \begin{bmatrix} \mathbf{CB} & \mathbf{CAB} & \mathbf{CA}^2\mathbf{B} & \cdots & \mathbf{CA}^{n-1}\mathbf{B} & \mathbf{D} \end{bmatrix}_{m \times (n+1)r} \quad \text{is rank } m.$$

Uncontrollable System & Stabilizability

- An uncontrollable system has a subsystem that is physically disconnected from the input
 - No matter that the input and whatever finite time required, there is no way to affect the state(s)
- For partially controllable systems, if the uncontrollable modes are stable and the unstable modes are controllable, the system is said to be **stabilizable**
- Consider:
$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$$
 - The system is not state controllable.
 - Stable mode that corresponds to the pole/eigenvalue at -1 is not controllable
 - Unstable mode that corresponds to the pole/eigenvalue at 1 is controllable
 - System can be made stable by a suitable feedback controller
 - System is said to be **stabilizable**

Complete Observability

- Consider the state-space system: $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$
 $\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u}$
- We know the solution is: $\mathbf{y}(t) = \mathbf{C}e^{\mathbf{A}t}\mathbf{x}(0) + \mathbf{C}\int_0^t e^{\mathbf{A}(t-\tau)}\mathbf{B}\mathbf{u}(\tau)d\tau + \mathbf{D}\mathbf{u}$
- A system is said to be **completely observable** if every state at $t=t_0$ can be determined from the observation of $y(t)$ over a finite time interval $t_0 \leq t \leq t_f$
 - Every transition of the state eventually affects every element of the output vector
- Concept of observability is important in practice where there is difficulty during state feedback control when some of the state variables are not accessible for direct measurement
 - Necessary to estimate the unmeasurable states in order to construct control signals

Complete Observability

- From the general solution of $y(t)$:

$$\mathbf{y}(t) = \mathbf{C}e^{\mathbf{A}t}\mathbf{x}(0) + \mathbf{C}\int_0^t e^{\mathbf{A}(t-\tau)}\mathbf{B}\mathbf{u}(\tau)d\tau + \mathbf{D}\mathbf{u}$$

- $\mathbf{A}, \mathbf{B}, \mathbf{C}$ and \mathbf{D} are known and so is $\mathbf{u}(t)$
- Last 2 terms of the solution are known and can be accounted for
- Hence to investigate complete observability, it is sufficient to consider

$$\mathbf{y}(t) = \mathbf{C}e^{\mathbf{A}t}\mathbf{x}(0)$$

- Recall C-H theorem: $e^{\mathbf{A}t} = \mathbf{I} + \mathbf{A}t + \frac{\mathbf{A}^2 t^2}{2!} + \frac{\mathbf{A}^3 t^3}{3!} + \frac{\mathbf{A}^4 t^4}{4!} + \dots = \sum_{k=0}^{\infty} \frac{\mathbf{A}^k t^k}{k!} = \sum_{k=0}^{n-1} \alpha_k(t) \mathbf{A}^k$

$$\mathbf{y}(t) = \sum_{k=0}^{n-1} \alpha_k(t) \mathbf{C} \mathbf{A}^k \mathbf{x}(0)$$

- Or: $\mathbf{y}(t) = \alpha_0(t) \mathbf{C} \mathbf{x}(0) + \alpha_1(t) \mathbf{C} \mathbf{A} \mathbf{x}(0) + \dots + \alpha_{n-1}(t) \mathbf{C} \mathbf{A}^{n-1} \mathbf{x}(0)$

Complete Observability

- If the system is completely observable, then given the output $\mathbf{y}(t)$ over time interval $t_0 \leq t \leq t_f$ $\mathbf{x}(0)$ can be uniquely determined.

$$\mathbf{y}(t) = [\alpha_0(t) \quad \alpha_1(t) \quad \cdots \quad \alpha_{n-1}(t)] \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \\ \vdots \\ \mathbf{CA}^{n-1} \end{bmatrix} \mathbf{x}(0)$$

- For this to be possible the rank of $\begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \\ \vdots \\ \mathbf{CA}^{n-1} \end{bmatrix}_{nm \times n}$ must be n

- This condition can be rewritten. For complete observability, the following matrix must be of rank n or has at least n linearly independent column vectors.

$$\mathbf{O}_B = \begin{bmatrix} \mathbf{C}^* & \mathbf{A}^* \mathbf{C}^* & \cdots & (\mathbf{A}^*)^{n-1} \mathbf{C}^* \end{bmatrix}_{n \times nm} \quad (n \times nm \text{ matrix}) \text{ is called the } \mathbf{Observability Matrix}$$

$$* : \text{Conjugate Transpose} \quad \mathbf{A}^* = (\bar{\mathbf{A}})^T = \overline{\mathbf{A}^T}$$

Exercise

- For the following system, is the system completely observable?

$$\begin{aligned}\dot{\mathbf{x}} &= \mathbf{A}\mathbf{x} + \mathbf{B}u \\ \mathbf{y} &= \mathbf{C}\mathbf{x}\end{aligned}\quad \mathbf{x} = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}, \quad \mathbf{A} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -6 & -11 & -6 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}, \quad \mathbf{C} = \begin{bmatrix} 4 & 5 & 1 \end{bmatrix}$$

$$\mathbf{O}_B = \begin{bmatrix} \mathbf{C}^* & \mathbf{A}^* \mathbf{C}^* & (\mathbf{A}^*)^2 \mathbf{C}^* \end{bmatrix} \quad \mathbf{A}^* = \begin{bmatrix} 0 & 0 & -6 \\ 1 & 0 & -11 \\ 0 & 1 & -6 \end{bmatrix}, \quad \mathbf{C}^* = \begin{bmatrix} 4 \\ 5 \\ 1 \end{bmatrix}$$

$$\mathbf{A}^* \mathbf{C}^* = \begin{bmatrix} 0 & 0 & -6 \\ 1 & 0 & -11 \\ 0 & 1 & -6 \end{bmatrix} \begin{bmatrix} 4 \\ 5 \\ 1 \end{bmatrix} = \begin{bmatrix} -6 \\ -7 \\ -1 \end{bmatrix} \quad \mathbf{A}^* \mathbf{A}^* \mathbf{C}^* = \begin{bmatrix} 0 & 0 & -6 \\ 1 & 0 & -11 \\ 0 & 1 & -6 \end{bmatrix} \begin{bmatrix} -6 \\ -7 \\ -1 \end{bmatrix} = \begin{bmatrix} 6 \\ 5 \\ -1 \end{bmatrix}$$

$$\mathbf{O}_B = \begin{bmatrix} 4 & -6 & 6 \\ 5 & -7 & 5 \\ 1 & -1 & -1 \end{bmatrix} \quad \det \mathbf{O}_B = \det \begin{bmatrix} 4 & -6 & 6 \\ 5 & -7 & 5 \\ 1 & -1 & -1 \end{bmatrix} = 0$$

Rank of Observability Matrix is less than 3. System not completely observable

Principle of Duality

- Controllability and Observability share an intricate relationship. The principle of duality was conceived by Kalman to connect the analogies between controllability & observability
- Consider a system S_1 and its dual system S_2 :

$$\begin{array}{llll}
 \dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} & \mathbf{x}: n \text{ vector} & \mathbf{A}: n \times n \text{ matrix} & \dot{\mathbf{z}} = \mathbf{A}^* \mathbf{z} + \mathbf{C}^* \mathbf{v} & \mathbf{z}: n \text{ vector} & \mathbf{A}^*: n \times n \text{ matrix} \\
 \mathbf{y} = \mathbf{C}\mathbf{x} & \mathbf{u}: r\text{-vector} & \mathbf{B}: n \times r \text{ matrix} & \mathbf{w} = \mathbf{B}^* \mathbf{z} & \mathbf{v}: m\text{-vector} & \mathbf{B}^*: r \times n \text{ matrix} \\
 & \mathbf{y}: m\text{-vector} & \mathbf{C}: m \times n \text{ matrix} & & \mathbf{w}: r\text{-vector} & \mathbf{C}^*: n \times m \text{ matrix}
 \end{array}$$

- The principle of duality states S_1 is completely state controllable (or observable) if and only if system S_2 is completely observable (state-controllable).

$$\begin{array}{ll}
 \mathbf{C}_O = \begin{bmatrix} \mathbf{B} & \mathbf{AB} & \dots & \mathbf{A}^{n-1}\mathbf{B} \end{bmatrix}_{n \times nr} & \mathbf{C}_O = \begin{bmatrix} \mathbf{C}^* & \mathbf{A}^* \mathbf{C}^* & \dots & (\mathbf{A}^*)^{n-1} \mathbf{C}^* \end{bmatrix}_{n \times nm} \\
 \mathbf{O}_B = \begin{bmatrix} \mathbf{C}^* & \mathbf{A}^* \mathbf{C}^* & \dots & (\mathbf{A}^*)^{n-1} \mathbf{C}^* \end{bmatrix}_{n \times nm} & \mathbf{O}_B = \begin{bmatrix} \mathbf{B} & \mathbf{AB} & \dots & \mathbf{A}^{n-1}\mathbf{B} \end{bmatrix}_{n \times nr}
 \end{array}$$

- For a partially observable system, if the unobservable modes are stable and the observable modes are unstable, the system is said to be detectable.
 - Concept of detectability is dual to the concept of stabilizability

Illustrative Example

- Consider the following system: $G(s) = \frac{1}{s^2 + 2s + 1}$

- It can be expressed in CCF (S_1),

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 \\ -1 & -2 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$

$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \mathbf{x}$$

- or also in OCF (S_2),

$$\dot{\mathbf{z}} = \begin{bmatrix} 0 & -1 \\ 1 & -2 \end{bmatrix} \mathbf{z} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$$

$$y = \begin{bmatrix} 0 & 1 \end{bmatrix} \mathbf{z}$$

- If we defined: $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$, you will note that: $\dot{\mathbf{z}} = \mathbf{A}^* \mathbf{z} + \mathbf{C}^* u$
 $y = \mathbf{C}\mathbf{x}$ $y = \mathbf{B}^* \mathbf{z}$
- In other words, S_1 is completely state controllable (or observable) if and only if system S_2 is completely observable (state-controllable).

$$\mathbf{C}_O = [\mathbf{B} \quad \mathbf{A}\mathbf{B}]$$

$$\mathbf{C}_O = [\mathbf{C}^* \quad \mathbf{A}^* \mathbf{C}^*]$$

$$\mathbf{O}_B = [\mathbf{C}^* \quad \mathbf{A}^* \mathbf{C}^*]$$

$$\mathbf{O}_B = [\mathbf{B} \quad \mathbf{A}\mathbf{B}]$$