

# EN671: Solar Energy Conversion Technology

## Flat Plate Collectors



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# Collector Losses and loss estimation

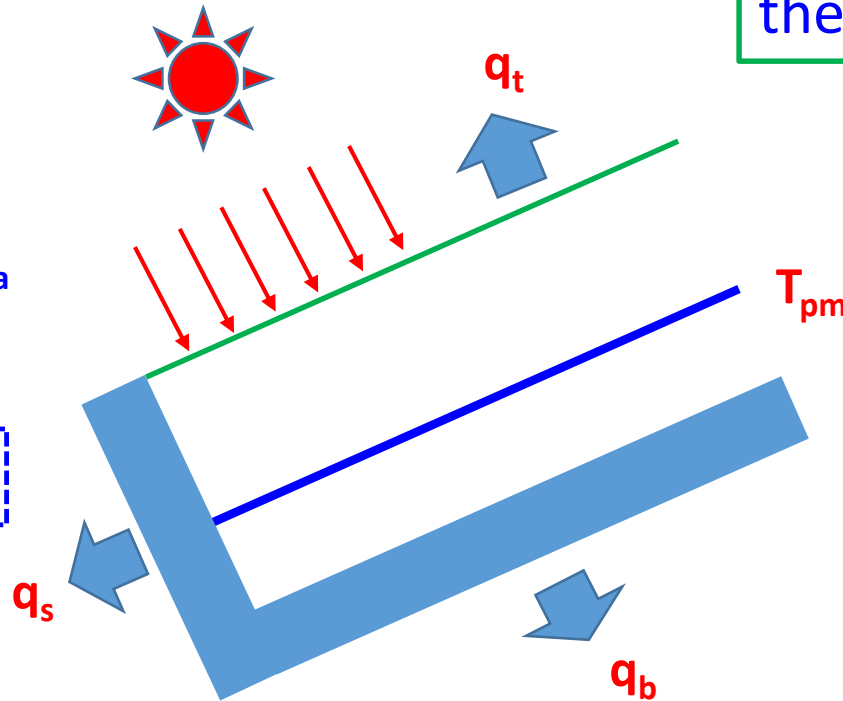
# Collector losses

Energy Balance on the absorber plate

$$q_u = A_p S - q_l$$

$T_a$

$$S = I_b r_b (\tau\alpha)_b + \{I_d r_d + (I_b + I_d) r_r\} (\tau\alpha)_d$$



Heat lost from the collector

$$q_l = U_l A_p (T_{pm} - T_a)$$

$U_l$  = Over all loss coefficient

$A_p$  = Area of the absorber plate

$T_{pm}$  = Average temperature of the absorber plate

$T_a$  = Temperature of the surrounding air

Heat lost from the collector is the sum of heat lost from the top, the bottom and the sides

$$q_l = q_t + q_b + q_s$$

# Loss coefficients

Each of the losses is also expressed in terms of coefficients

- ✓ Rate at which heat is lost from the top
- ✓ Rate at which heat is lost from the bottom
- ✓ Rate at which heat is lost from the side

Definition of each of the coefficients is based on the area  $A_p$  and the temperature difference  $(T_{pm} - T_a)$

$$q_t = U_t A_p (T_{pm} - T_a)$$

$$q_b = U_b A_p (T_{pm} - T_a)$$

$$q_s = U_s A_p (T_{pm} - T_a)$$

$$\begin{matrix} U_t \\ U_b \\ U_s \end{matrix}$$

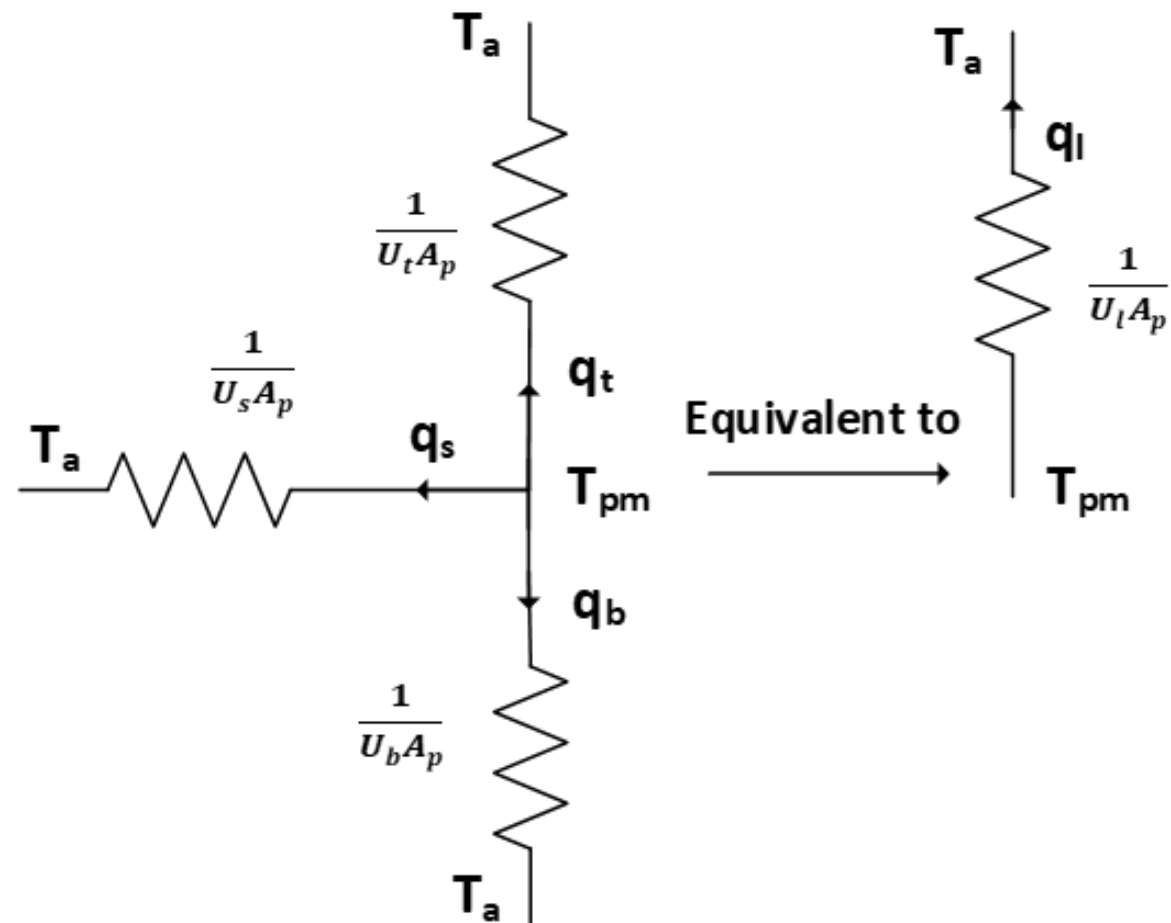
- ✓ Top loss coefficient
- ✓ Bottom loss coefficient
- ✓ Side loss coefficient

$$q_l = q_t + q_b + q_s$$

The overall loss coefficient :  $U_l = U_t + U_b + U_s$

- ✓ The overall loss coefficient is a measure of all the losses.
- ✓ Its value should be in the range of 2 to 10 W/m<sup>2</sup>-K

# Thermal Resistance Network

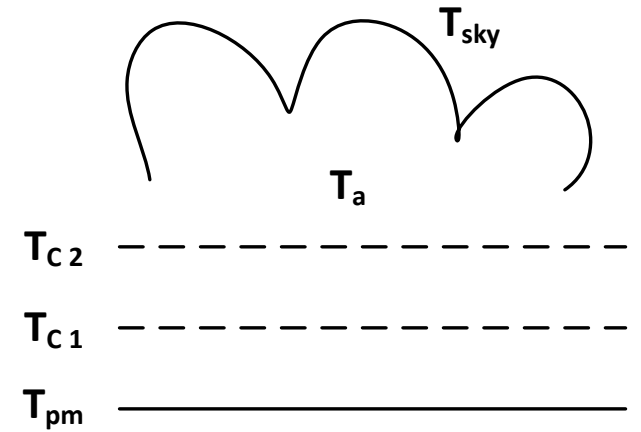


# Top loss coefficient

- Evaluated by considering convection and re-radiation losses from the absorber plate in the upward direction.

## Assumption:

- The transparent covers and the absorber plate constitute a system of infinite parallel surfaces.
- The flow of heat is one dimensional and steady.
- Temperature drop across the thickness of the covers is negligible
- The interaction between the incoming solar radiation absorbed by the covers and the outgoing loss may be neglected.
- The transparent cover is assumed to be opaque.



Heat transferred by convection and radiation absorber plate and first cover

$$\frac{q_t}{A_p} = h_{p-c_1} (T_{pm} - T_{c1}) + \frac{\sigma(T_{pm}^4 - T_{c1}^4)}{\left( \frac{1}{\epsilon_p} + \frac{1}{\epsilon_c} - 1 \right)}$$

First cover and 2<sup>nd</sup> cover

$$\frac{q_t}{A_p} = h_{c1-c2} (T_{c1} - T_{c2}) + \frac{\sigma(T_{c1}^4 - T_{c2}^4)}{\left( \frac{1}{\epsilon_c} + \frac{1}{\epsilon_c} - 1 \right)}$$

2<sup>nd</sup> cover and surroundings

$$\frac{q_t}{A_p} = h_w (T_{c2} - T_{sky}) + \sigma \epsilon_c (T_{c2}^4 - T_{sky}^4)$$

## Heat Transfer coefficient between inclined parallel surfaces

- Buchberg *et al.* developed the following correlations based on the experimental investigation of natural convection heat transfer coefficient for the enclosed space (*between the absorber plate to the first cover and the first cover to the second cover*).

$$Nu = 1; \text{ for } Ra \cos \beta < 1708$$

$$Nu = 1 + 1.446 \left[ 1 - \frac{1708}{Ra \cos \beta} \right] \quad \text{for } 1708 < Ra \cos \beta < 5900$$

$$Nu = 0.229(Ra \cos \beta)^{0.252} \quad \text{for } 5900 < Ra \cos \beta < 9.23 \times 10^4$$

$$Nu = 0.157(Ra \cos \beta)^{0.285} \quad \text{for } 9.23 \times 10^4 < Ra \cos \beta < 10^6$$



Properties are to be evaluated at the arithmetic mean of the surface temperatures

# Heat Transfer coefficient at the top cover (wind heat transfer coefficient)

- McAdams correlations:

$$h_w = 5.7 + 3.8V_\infty$$

- Correlations developed by Test et al. :

$$h_w = 8.55 + 2.56V_\infty$$

- Sky temperature:

$$T_{sky} = T_a - 6$$

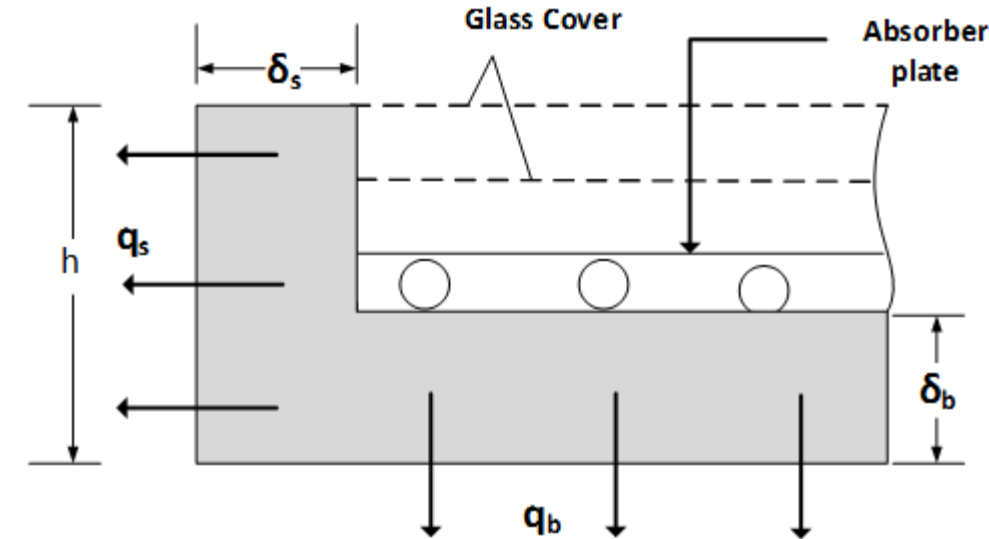


# Bottom Loss coefficient

- Evaluated by considering conduction convection losses from the absorber plate towards downward
- Assumptions: (1) Flow of heat is one dimensional and steady, (2) Thermal resistance associated with conduction dominates

$$U_b = \frac{k_i}{\delta_b} = \frac{\text{Thermal conductivity of the insulation}}{\text{Thickness of insulation}}$$

$k_i$  = Thermal conductivity of the insulation  
 $\delta_b$  = Thickness of Insulation



# Side loss coefficient

- ✓ Conduction resistance dominates
- ✓ Flow of heat is one dimensional and steady

$$q_s = U_s A_p (T_{pm} - T_a)$$

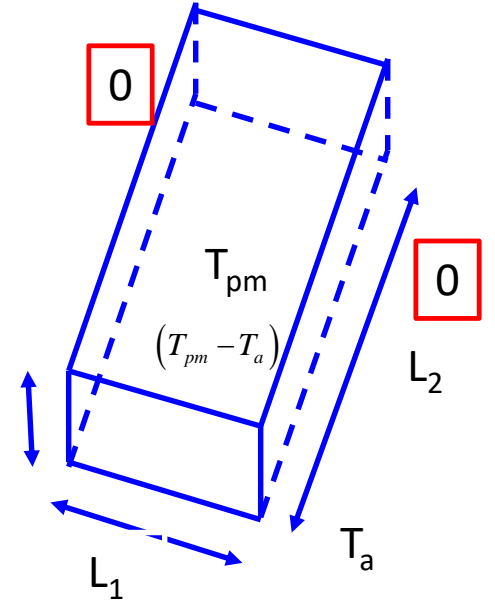
Average temperature drop across the side insulation

$$q_s = 2 \times h \times (L_1 + L_2) \times k_i \frac{(T_{pm} - T_a)}{2\delta_s}$$

$$A_p = L_1 \times L_2$$

$$U_s = \frac{q_s}{A_p (T_{pm} - T_a)} = \frac{2h(L_1 + L_2)k_i \frac{(T_{pm} - T_a)}{2\delta_s}}{L_1 \times L_2 (T_{pm} - T_a)}$$

$$\Rightarrow U_s = \frac{(L_1 + L_2)hk_i}{L_1 L_2 \delta_s}$$



$$A = 2L_2 \cdot h + 2L_1 \cdot h = 2(L_2 \cdot h + L_1 \cdot h)$$

**Ex.1:** For a FPC with a top-loss coefficient of  $6.6 \text{ W/m}^2 \text{ }^\circ\text{C}$ , determine the overall loss coefficient by using following data:

- Back insulation thickness =  $0.045 \text{ m}$
- Thermal conductivity of insulation =  $0.04 \text{ W/m}^2 \text{ }^\circ\text{C}$
- Collector bank length =  $8 \text{ m}$
- Collector bank width =  $2.5 \text{ m}$
- Collector thickness =  $0.08 \text{ m}$
- Edge insulation thickness =  $0.02 \text{ m}$

$$U_b = \frac{k_i}{\delta_b} = \frac{0.04}{0.045} = 0.889 \text{ W/m}^2 \text{ }^\circ\text{C}$$

$$U_s = \frac{(L_1 + L_2)hk_i}{L_1L_2\delta_s} = \frac{(8+2.5)0.08 \times 0.04}{8 \times 2.5 \times 0.02} = 0.084 \text{ W/m}^2 \text{ }^\circ\text{C}$$

$$U_l = U_t + U_b + U_s = 6.6 + 0.889 + 0.084 = 7.573 \text{ W/m}^2 \text{ }^\circ\text{C}$$

Ex.1: Calculate the **overall loss coefficient for a flat-plate collector with two glass covers**. The following data is given: (a) size of the absorber plate: 1.90 m x 0.90 m, (b) spacing between plate and first glass cover: 5 cm, (c) spacing between first and second glass cover: 5 cm, (d) plate emissivity : 0.90, (e) glass cover emissivity: 0.85, (f) collector tilt: 23 °, (g) Mean plate temperature: 73 °C, (h) Ambient air temperature: 25 °C, (i) wind speed : 2.7 m/s, (j) Back insulation thickness: 10 cm, (k) side insulation thickness: 5 cm, (l) thermal conductivity of insulation: 0.07 W/m-K. Use the appropriate correlation from the correlation Table-1. The properties of air is given in Table-2.

Table-1 Correlation table

$$Nu = 1 + 1.446 \left[ 1 - \frac{1708}{Ra \cos \beta} \right] \quad \text{for } 1708 < Ra \cos \beta < 5900$$

$$Nu = 0.229(Ra \cos \beta)^{0.252} \quad \text{for } 5900 < Ra \cos \beta < 9.23 \times 10^4$$

$$Nu = 0.157(Ra \cos \beta)^{0.285} \quad \text{for } 9.23 \times 10^4 < Ra \cos \beta < 10^6$$

$$T_{sky} = T_a - 6$$

$$h_w = 5.7 + 3.8V_\infty$$

Table 2 Properties of air

Properties of air at 1 atm pressure							
Temp. <i>T</i> , °C	Density <i>ρ</i> , kg/m <sup>3</sup>	Specific Heat <i>c<sub>p</sub></i> , J/kg·K	Thermal Conductivity <i>k</i> , W/m·K	Thermal Diffusivity <i>α</i> , m <sup>2</sup> /s	Dynamic Viscosity <i>μ</i> , kg/m·s	Kinematic Viscosity <i>ν</i> , m <sup>2</sup> /s	Prandtl Number Pr
-150	2.866	983	0.01171	4.158 × 10 <sup>-6</sup>	8.636 × 10 <sup>-6</sup>	3.013 × 10 <sup>-6</sup>	0.7246
-100	2.038	966	0.01582	8.036 × 10 <sup>-6</sup>	1.189 × 10 <sup>-6</sup>	5.837 × 10 <sup>-6</sup>	0.7263
-50	1.582	999	0.01979	1.252 × 10 <sup>-5</sup>	1.474 × 10 <sup>-5</sup>	9.319 × 10 <sup>-6</sup>	0.7440
-40	1.514	1002	0.02057	1.356 × 10 <sup>-5</sup>	1.527 × 10 <sup>-5</sup>	1.008 × 10 <sup>-5</sup>	0.7436
-30	1.451	1004	0.02134	1.465 × 10 <sup>-5</sup>	1.579 × 10 <sup>-5</sup>	1.087 × 10 <sup>-5</sup>	0.7425
-20	1.394	1005	0.02211	1.578 × 10 <sup>-5</sup>	1.630 × 10 <sup>-5</sup>	1.169 × 10 <sup>-5</sup>	0.7408
-10	1.341	1006	0.02288	1.696 × 10 <sup>-5</sup>	1.680 × 10 <sup>-5</sup>	1.252 × 10 <sup>-5</sup>	0.7387
0	1.292	1006	0.02364	1.818 × 10 <sup>-5</sup>	1.729 × 10 <sup>-5</sup>	1.338 × 10 <sup>-5</sup>	0.7362
5	1.269	1006	0.02401	1.880 × 10 <sup>-5</sup>	1.754 × 10 <sup>-5</sup>	1.382 × 10 <sup>-5</sup>	0.7350
10	1.246	1006	0.02439	1.944 × 10 <sup>-5</sup>	1.778 × 10 <sup>-5</sup>	1.426 × 10 <sup>-5</sup>	0.7336
15	1.225	1007	0.02476	2.009 × 10 <sup>-5</sup>	1.802 × 10 <sup>-5</sup>	1.470 × 10 <sup>-5</sup>	0.7323
20	1.204	1007	0.02514	2.074 × 10 <sup>-5</sup>	1.825 × 10 <sup>-5</sup>	1.516 × 10 <sup>-5</sup>	0.7309
25	1.184	1007	0.02551	2.141 × 10 <sup>-5</sup>	1.849 × 10 <sup>-5</sup>	1.562 × 10 <sup>-5</sup>	0.7296
30	1.164	1007	0.02588	2.208 × 10 <sup>-5</sup>	1.872 × 10 <sup>-5</sup>	1.608 × 10 <sup>-5</sup>	0.7282
35	1.145	1007	0.02625	2.277 × 10 <sup>-5</sup>	1.895 × 10 <sup>-5</sup>	1.655 × 10 <sup>-5</sup>	0.7268
40	1.127	1007	0.02662	2.346 × 10 <sup>-5</sup>	1.918 × 10 <sup>-5</sup>	1.702 × 10 <sup>-5</sup>	0.7255
45	1.109	1007	0.02699	2.416 × 10 <sup>-5</sup>	1.941 × 10 <sup>-5</sup>	1.750 × 10 <sup>-5</sup>	0.7241
50	1.092	1007	0.02735	2.487 × 10 <sup>-5</sup>	1.963 × 10 <sup>-5</sup>	1.798 × 10 <sup>-5</sup>	0.7228
60	1.059	1007	0.02808	2.632 × 10 <sup>-5</sup>	2.008 × 10 <sup>-5</sup>	1.896 × 10 <sup>-5</sup>	0.7202
70	1.028	1007	0.02881	2.780 × 10 <sup>-5</sup>	2.052 × 10 <sup>-5</sup>	1.995 × 10 <sup>-5</sup>	0.7177
80	0.9994	1008	0.02953	2.931 × 10 <sup>-5</sup>	2.096 × 10 <sup>-5</sup>	2.097 × 10 <sup>-5</sup>	0.7154
90	0.9718	1008	0.03024	3.086 × 10 <sup>-5</sup>	2.139 × 10 <sup>-5</sup>	2.201 × 10 <sup>-5</sup>	0.7132
100	0.9458	1009	0.03095	3.243 × 10 <sup>-5</sup>	2.181 × 10 <sup>-5</sup>	2.306 × 10 <sup>-5</sup>	0.7111
120	0.8977	1011	0.03235	3.565 × 10 <sup>-5</sup>	2.264 × 10 <sup>-5</sup>	2.522 × 10 <sup>-5</sup>	0.7073
140	0.8542	1013	0.03374	3.898 × 10 <sup>-5</sup>	2.345 × 10 <sup>-5</sup>	2.745 × 10 <sup>-5</sup>	0.7041
160	0.8148	1016	0.03511	4.241 × 10 <sup>-5</sup>	2.420 × 10 <sup>-5</sup>	2.975 × 10 <sup>-5</sup>	0.7014
180	0.7788	1019	0.03646	4.593 × 10 <sup>-5</sup>	2.504 × 10 <sup>-5</sup>	3.212 × 10 <sup>-5</sup>	0.6992
200	0.7459	1023	0.03779	4.954 × 10 <sup>-5</sup>	2.577 × 10 <sup>-5</sup>	3.455 × 10 <sup>-5</sup>	0.6974
250	0.6746	1033	0.04104	5.890 × 10 <sup>-5</sup>	2.760 × 10 <sup>-5</sup>	4.091 × 10 <sup>-5</sup>	0.6946
300	0.6158	1044	0.04418	6.871 × 10 <sup>-5</sup>	2.934 × 10 <sup>-5</sup>	4.765 × 10 <sup>-5</sup>	0.6935
350	0.5664	1056	0.04721	7.892 × 10 <sup>-5</sup>	3.101 × 10 <sup>-5</sup>	5.475 × 10 <sup>-5</sup>	0.6937
400	0.5243	1069	0.05015	8.951 × 10 <sup>-5</sup>	3.261 × 10 <sup>-5</sup>	6.219 × 10 <sup>-5</sup>	0.6948
450	0.4880	1081	0.05298	1.004 × 10 <sup>-4</sup>	3.415 × 10 <sup>-5</sup>	6.997 × 10 <sup>-5</sup>	0.6965
500	0.4565	1093	0.05572	1.117 × 10 <sup>-4</sup>	3.563 × 10 <sup>-5</sup>	7.806 × 10 <sup>-5</sup>	0.6986
600	0.4042	1115	0.06093	1.352 × 10 <sup>-4</sup>	3.846 × 10 <sup>-5</sup>	9.515 × 10 <sup>-5</sup>	0.7037

**Top loss coefficient:**

$$\frac{q_t}{A_p} = h_{p-c_1} (T_{pm} - T_{c1}) + \frac{\sigma(T_{pm}^4 - T_{c1}^4)}{\left( \frac{1}{\varepsilon_p} + \frac{1}{\varepsilon_c} - 1 \right)}$$

$$\frac{q_t}{A_p} = h_{c1-c2} (T_{c1} - T_{c2}) + \frac{\sigma(T_{c1}^4 - T_{c2}^4)}{\left( \frac{1}{\varepsilon_c} + \frac{1}{\varepsilon_c} - 1 \right)}$$

$$\frac{q_t}{A_p} = h_w (T_{c2} - T_{sky}) + \sigma \varepsilon_c (T_{c2}^4 - T_{sky}^4)$$





$T_{c1}$ (K)	$T_{c2}$ (K)	$q_t/A_p$ Eq.(1)	$q_t/A_p$ Eq.(2)	$q_t/A_p$ Eq.(3)	Average $W/m^2$
328	306	167.498	176.803	196.749	180.350
327	305	176.925	175.391	175.293	175.869



# Empirical equation for top loss coefficient

**Klein**

$$U_t = \left[ \frac{M}{\left( \frac{C}{T_{pm}} \right) \left( \frac{T_{pm} - T_a}{M + f} \right)^{0.33}} + \frac{1}{h_w} \right]^{-1} + \left[ \frac{\sigma(T_{pm}^2 - T_a^2)(T_{pm} - T_a)}{\frac{1}{\varepsilon_p + 0.005M(1 - \varepsilon_p)} + \frac{(2M + f - 1)}{\varepsilon_c} - M} \right]$$

$$320 \leq T_{pm} \leq 420\text{K}$$

$$260 \leq T_a \leq 310\text{K}$$

$$0.1 \leq \varepsilon_p \leq 0.95$$

$$0 \leq V_\infty \leq 10 \text{ m/s}$$

$$1 \leq M \leq 3$$

$$0 \leq \beta \leq 90^\circ$$

$$\text{where } f = (1 - 0.04h_w + 0.0005h_w^2)(1 + 0.091M)$$

$$C = 365.9(1 - 0.00883\beta + 0.0001298\beta^2)$$

$M$  = number of glass covers

**Malhotra et al.**

$$U_t = \left[ \frac{M}{\left( \frac{C}{T_{pm}} \right) \left( \frac{T_{pm} - T_a}{M + f} \right)^{0.252}} + \frac{1}{h_w} \right]^{-1} + \left[ \frac{\sigma(T_{pm}^2 - T_a^2)(T_{pm} - T_a)}{\frac{1}{\varepsilon_p + 0.0425M(1 - \varepsilon_p)} + \frac{(2M + f - 1)}{\varepsilon_c} - M} \right]$$

$$\text{where } f = \left( \frac{9}{h_w} - \frac{30}{h_w^2} \right) \left( \frac{T_a}{316.9} \right) (1 + 0.091M)$$

$$C = 204.429 \cos \beta)^{0.25} / L^{0.24}$$

$L$  = Spacing (m)

# Summary

- Heat lost from the collector
- Loss coefficients
  - Top loss
  - Bottom loss
  - Side loss
  - Overall loss coefficient
- Solved problems how to calculate losses associated with FPC
  - Iterative technique
  - Use of correlations

**Thank you**