Feedback control and state estimation

Master Lab in Autonomous driving - Motion planning and control

Paolo Falcone

Dipartimento di Ingegneria "Enzo Ferrari" Università di Modena e Reggio Emilia



• A general control structure. State- and output-feedback control

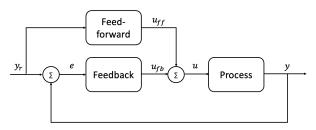
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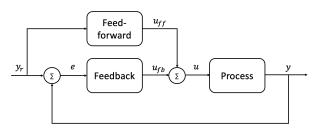
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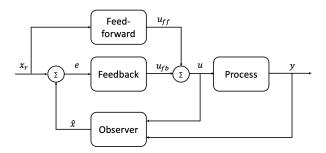
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How do we design *K*?

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K is to be designed such that the roots of p(s) are placed as desired.

Consider the double integrator

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 $K_r = k_1$ leads to unitary steady state gain. That is, y(t) = r, $t \to \infty$.

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The feedback gain is given by

$$K = [p_1 - a_1 \ p_2 - a_2 \ \dots \ p_n - a_n] \tilde{W}_r W_r^{-1},$$

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with

$$W_r = [B \ AB \ A^2B \dots A^{n-1}B],$$

$$\tilde{W}_r = \begin{bmatrix} 1 & a_1 & a_2 & \dots & a_{n-1} \\ 0 & 1 & a_1 & \dots & a_{n-1} \\ \vdots & & & & \\ 0 & 0 & \dots & 0 & 1 \end{bmatrix}^{-1}$$

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- *Step 1.* Check the reachability of the system (column rank of W_r).
- *Step* 2. Introduce the elements k_i of K as unknown variables.
- *Step 3.* Calculate the characteristic polynomials and equate the coefficients of equal powers of *s* to the desired polynomials

$$p(s) = s^n + p_1 s_{n-1} + \ldots + p_{n-1} s + p_n$$

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$$P: \min_{u(0:N-1)} V_N(x(0), u(0:N-1))$$

where the minimization is with respect to the sequence of control inputs

$$u(0:N-1) = \{u(0), u(1), \dots, u(N-1)\}\$$

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The *objective* or *criterion* or *cost function* V_N is given by

$$V_N(x(0), u(0:N-1)) = \sum_{i=0}^{N-1} (x^{\top}(i)Qx(i) + u^{\top}(i)Ru(i)) + x^{\top}(N)P_fx(N)$$

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Remark 1: All x(i) are functions of x(0) and u(0:N-1) via the model (1)! Remark 2: The first term $x^{T}(0)Qx(0)$ in the objective is really redundant but is kept for notational convenience.

Batch solution

Repeated use of $x^+ = Ax + Bu$ gives

$$\begin{bmatrix} x(1) \\ x(2) \\ \vdots \\ x(N) \end{bmatrix} = \begin{bmatrix} A \\ A^2 \\ \vdots \\ A^N \end{bmatrix} x(0) + \begin{bmatrix} B & 0 & \cdots & 0 \\ AB & B & \ddots & 0 \\ \vdots & \ddots & \ddots & \vdots \\ A^{N-1}B & A^{N-2}B & \cdots & B \end{bmatrix} \begin{bmatrix} u(0) \\ u(1) \\ \vdots \\ u(N-1) \end{bmatrix}$$
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The LQ criterion (2) can now be written

$$V_{N}(x(0), \boldsymbol{u}) = x^{\top}(0)Qx(0) + \boldsymbol{x}^{\top}\bar{Q}\boldsymbol{x} + \boldsymbol{u}^{\top}\bar{R}\boldsymbol{u}$$

$$= x^{\top}(0)Qx(0) + (\Omega x(0) + \Gamma \boldsymbol{u})^{\top}\bar{Q}(\Omega x(0) + \Gamma \boldsymbol{u}) + \boldsymbol{u}^{\top}\bar{R}\boldsymbol{u}$$

$$= \boldsymbol{u}^{\top}(\Gamma^{\top}\bar{Q}\Gamma + \bar{R})\boldsymbol{u} + 2\boldsymbol{x}^{\top}(0)\Omega^{\top}\bar{Q}\Gamma\boldsymbol{u} + \boldsymbol{x}^{\top}(0)(Q + \Omega^{\top}\bar{Q}\Omega)x(0)$$
 (5)

where $\bar{Q} = \text{diag}(Q, \dots, Q, P_f)$ and $\bar{R} = \text{diag}(R, \dots, R)$.

Dynamic programming solution to the LQ problem

Sequence of optimal control laws (*control policy*):

$$u^{0}(k;x) = K(k)x, \quad k = 0,..., N-1$$

 $K(k) = -(R + B^{T}P(k+1)B)^{-1}B^{T}P(k+1)A$

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Riccati equation:

$$P(k-1) = Q + A^{\top}P(k)A - A^{\top}P(k)B(R + B^{\top}P(k)B)^{-1}B^{\top}P(k)A, \quad P(N) = P_f$$

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Optimal cost-to-go (from time *k* to time *N*):

$$V^0_{k \to N}(x) = x^\top P(k) x$$

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If the pair (A, B) is controllable and Q, R > 0, then the solution to this infinite horizon LQ control problem gives a stable closed-loop system

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$$K = -(B^{T}PB + R)^{-1}B^{T}PA$$
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The latter equation is called the *algebraic Riccati equation*. The optimal cost is given by

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That is, if the system is *reachable* and *observable* the control and observer dynamics can be arbitrarily assigned (*separation principle*).

System model:

$$x(k+1) = Ax(k) + Bu(k) + w(k), \quad x(0) \sim \mathcal{N}(x_0, P_0), w \sim \mathcal{N}(0, Q)$$

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State estimator:

Correction:
$$\hat{x}(k|k) = \hat{x}(k|k-1) + L(k)[y(k) - C\hat{x}(k|k-1)], \ \hat{x}(0|0) = x_0$$

Prediction: $\hat{x}(k+1|k) = A\hat{x}(k|k) + Bu(k)$

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Correction:
$$\hat{x}(k|k) = \hat{x}(k|k-1) + L(k)[y(k) - C\hat{x}(k|k-1)], \ \hat{x}(0|0) = x_0$$

Prediction: $\hat{x}(k+1|k) = A\hat{x}(k|k) + Bu(k)$

Kalman filter gain:

$$L(k) = P(k)C^{\mathsf{T}}[CP(k)C^{\mathsf{T}} + R]^{-1}$$

System model:

$$x(k+1) = Ax(k) + Bu(k) + w(k), \quad x(0) \sim \mathcal{N}(x_0, P_0), w \sim \mathcal{N}(0, Q)$$

$$y(k) = Cx(k) + v(k), \quad v \sim \mathcal{N}(0, R)$$

State estimator:

Correction:
$$\hat{x}(k|k) = \hat{x}(k|k-1) + L(k)[y(k) - C\hat{x}(k|k-1)], \ \hat{x}(0|0) = x_0$$

Prediction: $\hat{x}(k+1|k) = A\hat{x}(k|k) + Bu(k)$

Kalman filter gain:

$$L(k) = P(k)C^{\mathsf{T}}[CP(k)C^{\mathsf{T}} + R]^{-1}$$

State estimation error covariance update (note that P(k) = P(k|k-1)):

Estimation error:
$$P(k|k) = P(k) - P(k)C^{T}[CP(k)C^{T} + R]^{-1}CP(k), P(0|0) = P_{0}$$

Prediction error: $P(k+1) = AP(k|k)A^{T} + Q$

Stationary Kalman filter

System:

$$x^{+} = Ax + Bu + w, \quad w \sim \mathcal{N}(0, Q)$$
$$y = Cx + v, \quad v \sim \mathcal{N}(0, R)$$

If the pair (C, A) is observable and Q, R > 0, then the Kalman filter gain L(k) and the prediction error covariance P(k) converge to the solution of the (filtering) algebraic Riccati equation

$$L = PC^{\top}[CPC^{\top} + R]^{-1}$$

$$P = APA^{\top} - APC^{\top}[CPC^{\top} + R]^{-1}CPA^{\top} + Q$$