```
function E = EccAnomFromMeanAnom( Me, e, tol )
% Use the Newton Raphson method to solve Keplers equation: E - e*sinE
= Me
% Inputs:
% Me Mean anomaly (rad)
        Eccentricity (0 <= e < 1)</pre>
  tol Error tolerance. Optional.
% Outputs:
% E Eccentric anomaly (rad)
if nargin<3</pre>
 tol = 1e-8;
end
% initial guess
if( Me>pi )
 E = Me - e/2;
else
 E = Me + e/2;
end
dist = 1;
while dist>tol
  % compute function value and function derivative value
  f = E-e.*sin(E)-Me;
  fp = 1-e.*cos(E);
  % distance between current and next iterate
 dist = max(abs(f./fp));
 E = E-f./fp;
end
```

Published with MATLAB® R2019b