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EX. 4.2

```
clear all
close all
Q1=120000000;
                                 %heat of reaction (hydrogen) [J/kg]
Q2=50010000;
                                 %heat of reaction (methane) [J/kg]
f=0.0291;
                                 %fuel/air ratio
q0=9.81;
                                 %acceleration of gravity [m/s^2]
Ve1=[1600:100:3000];
                                  %exit velocity [m/s]
M=5;
                                 %mach numbers
a = 303.1;
                                 %speed of sound at 30,000 feet [m/s]
V0=M*a;
                                 %flight speed
```

HYDROGEN

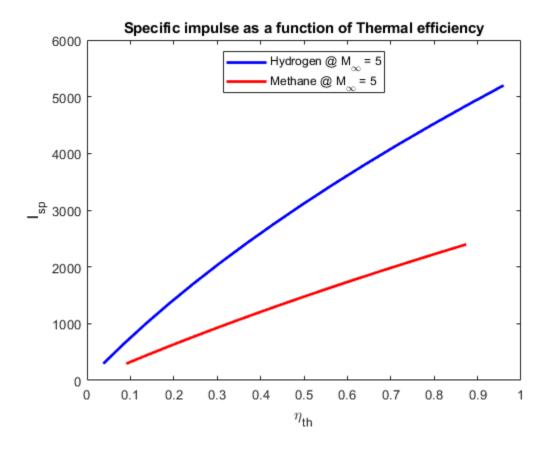
METHANE

```
Ve2=linspace(1600,2200,length(Ve1));
for j=1:length(Ve1)
   k2(j)=(f*Q2)/(V0^2/2);
   etath2(j)=((Ve2(j)^2/2)-(V0^2/2))/(f*Q2);
   eta02(j)=(2*(sqrt(etath2(j)*k2(j)+1)-1))/k2(j);
   I2(j)=(Q2*eta02(j))/(g0*V0);
end
```

PLOTS

```
figure(1)
plot (etath1,I1,'-b','LineWidth',2)
hold on
```

```
plot(etath2,I2,'-r','LineWidth',2)
legend('Hydrogen @ M_\infty = 5','Methane @ M_\infty = 5')
legend('Location','north')
title('Specific impulse as a function of Thermal efficiency')
xlabel('\eta_t_h')
ylabel('I_s_p')
```



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