
```
function Me = MeanAnomFromEccAnomE( E, e )
%
% Compute mean anomaly for a given eccentric anomaly in an eccentric
% orbit
%
% Inputs:
%   E      Eccentric anomaly
%   e      Eccentricity (0 <= e < 1)
%
% Outputs:
%   Me     Mean anomaly
%
Me = E - e*sin(E);
```

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