



ISTANBUL TECHNICAL UNIVERSITY
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**Afterburning Turbofan Engine With Seperate
Exhaust System**

INSTRUCTOR

Prof. Dr. Fırat Oğuz EDİS

Group 4

Adnan Basri GÜLTEKİN – 110150014

Beytullah GÜNGÖR – 110150042

Emrecan ÖDEMİŞ – 110150040

Onur KADIOĞLU – 110150018

Ozan İBRAHİMAĞAOĞLU – 110160001

Berkay SOYLUOĞLU – 110160301

Ahmet Talha ÇETİN – 110160007

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1. Introduction

Propulsion systems have needed to move an aircraft through the air. Firstly, propeller engines are invented. During WW2 there was a need for more efficient propulsion engines. Because the propeller engines' fuel efficiencies and their thrusts were not enough for aviation. From the 1950s turbofan engines have modified and evolved. Today most modern aircrafts use turbofan engines because of their good fuel consumption and high thrust.

A turbofan engine is the most modern variation of the basic gas turbine engine. Likely the other gas turbines, turbofan engine has a core engine. In turbofan engines the core engine is surrounded by a fan in the front and an additional turbine at the rear. In the fan and fan turbines there are various blades.

1.1. Turbofan Engine's Working Principles

A turbofan can be thought of as a turbojet being used to drive a ducted fan, with both of those contributing to the thrust. Whereas all the air taken in by a turbojet passes through the turbine (through the combustion chamber), in a turbofan some of that air bypasses the turbine. Because the turbine has to additionally drive the fan, the turbine is larger and has larger pressure and temperature drops, and so the nozzles are smaller. This means that the exhaust velocity of the core is reduced. The fan also has lower exhaust velocity, giving much more thrust per unit energy.

In turbofans there is a bypass ratio. Bypass ratio means the ratio between the mass flow rate of the bypass stream to the mass flow rate entering the core. According to bypass ratio there are two types of turbofan engines; those are low-bypass turbofans and high-bypass turbofans. In high bypass turbofans the air ratio in the combustion chamber is bigger than high bypass turbofans. On the other hand in the light of this information it can be easily said in the low bypass ratio turbofans, engines use more jet thrust relative to fan thrust. If it is desired to see the working principles it can be seen below the link easily.

1.2. Common Types of Turbofan Engines

1.2.1. Low-Bypass Turbofan

A high-specific-thrust/low-bypass-ratio turbofan normally has a multi-stage fan, developing a relatively high pressure ratio and, thus, yielding a high (mixed or cold) exhaust velocity. The core airflow needs to be large enough to give sufficient core power to drive the fan. A smaller core flow/higher bypass ratio cycle can be achieved by raising the high-pressure turbine rotor inlet temperature. In Figure 1 there is an example of this type of turbofan.

When calculating the drag of an aircraft with a jet engine, we need to take into account the drag in the region with the jet outlet as well as many different factors. In particular, we have to calculate the forces acting on the nozzle on the regions we define as Boat Tail and

Base. The CFD analysis in these regions to calculate the forces and flow-related values after jet outlet is called Afterbody CFD Analysis.

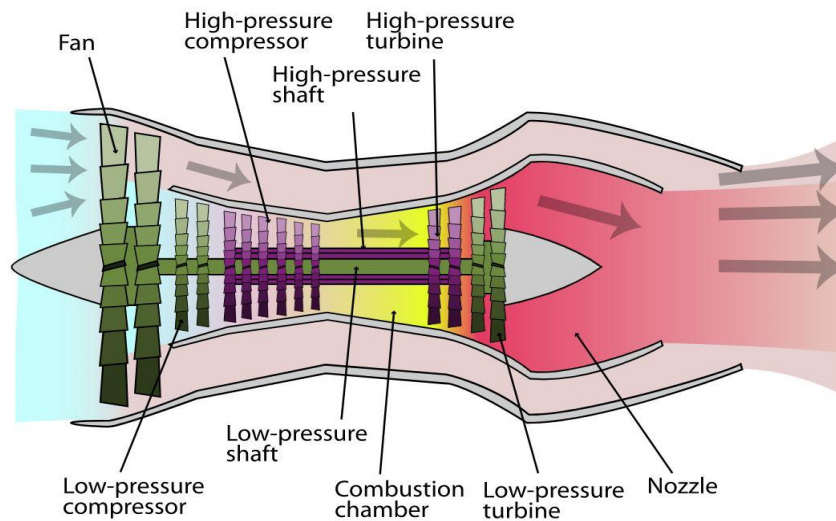


Figure 1 – Low Bypass Turbofan

1.2.2. Afterburning Turbofan

Since the 1970s, most jet fighter engines have been low/medium bypass turbofans with a mixed exhaust, afterburner and variable area final nozzle. An afterburner is a combustor located downstream of the turbine blades and directly upstream of the nozzle, which burns fuel from afterburner-specific fuel injectors. When lit, prodigious amounts of fuel are burnt in the afterburner, raising the temperature of exhaust gases by a significant degree, resulting in a higher exhaust velocity/engine specific thrust. The variable geometry nozzle must open to a larger throat area to accommodate the extra volume flow when the afterburner is lit. Afterburning is often designed to give a significant thrust boost for take off, transonic acceleration and combat maneuvers, but is very fuel intensive. Consequently, afterburning can be used only for short portions of a mission. In Figure 2 there is an example of this motor type.

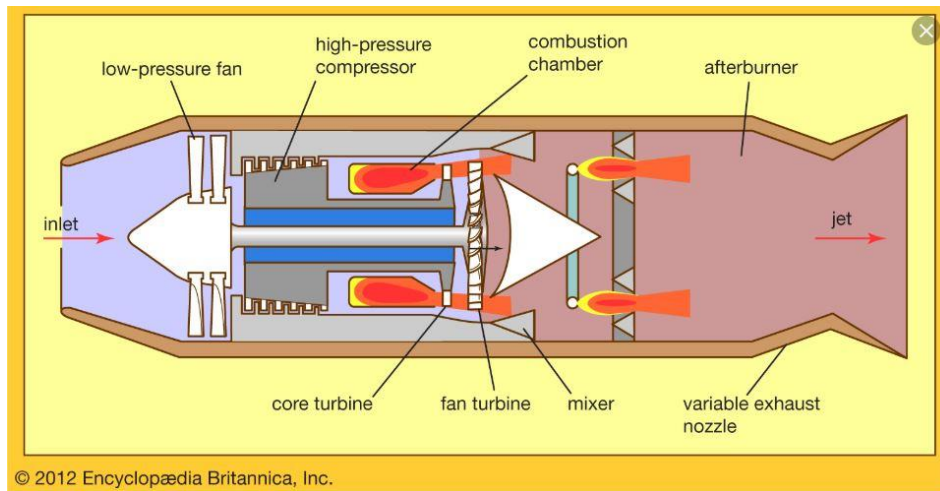


Figure 2 – Afterburning Turbofan

1.2.3. High-bypass Turbofan

To boost fuel economy and reduce noise, almost all of today's jet airliners and most military transport aircraft are powered by low-specific-thrust/high-bypass-ratio turbofans. These engines evolved from the high-specific-thrust/low-bypass-ratio turbofans used in such aircraft in the 1960s. Modern combat aircraft tend to use low-bypass ratio turbofans, and some military transport aircraft use turboprops.

Low specific thrust is achieved by replacing the multi-stage fan with a single-stage unit. Unlike some military engines, modern civil turbofans lack stationary inlet guide vanes in front of the fan rotor. The fan is scaled to achieve the desired net thrust. In Figure 3 there is an example of this motor type.

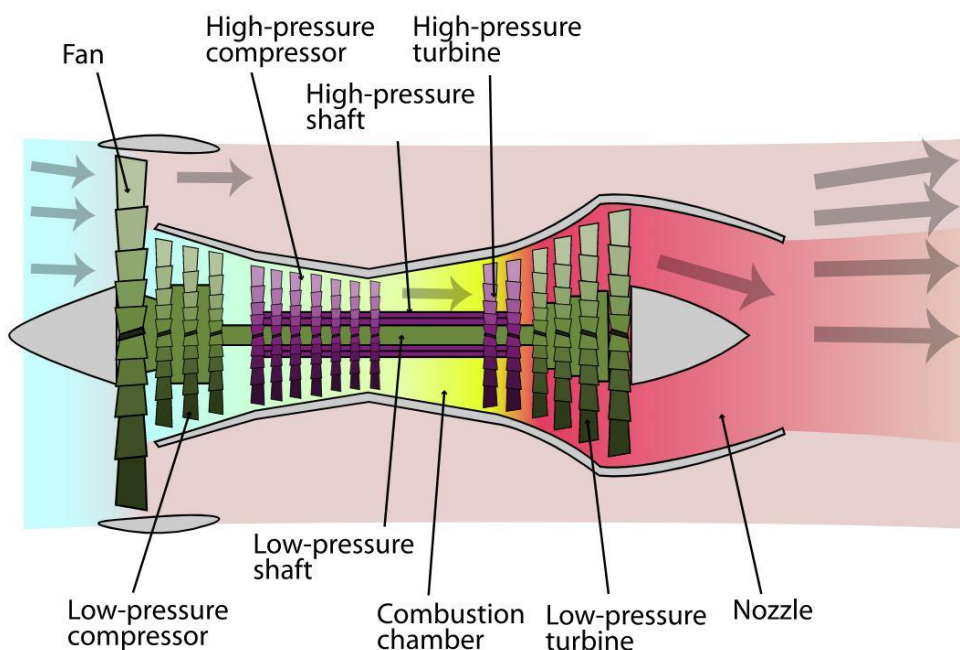


Figure 3 - High-Bypass Turbofan

1.2.4. Seperated Stream Turbofans with Afterburners

Our project group's topic is this type of turbofan engine. In this engine contains some parts of other type of turbofan engines. Because in this engine type it can be high or low bypass turbofan, it has seperated stream and 2 different type nozzle, and in it's two regions there are two types of fterburners whose names are duct burner and afterburner. This engine type has various advantages; however, with this advantages there are disadvantages to. For example; during take off after burner and duct burners are activated, this shorten the runway of aircraft, but fuel consumption is increase. During cruise afer burner and duct burners are closed and there is no need to them. This cause mechanical weight on plane. However, battle combats and sudden maneuver are needed afterburner. Because of after burner and duct burner this type of turbofan's thermodynamic cycles are more complex than the other. In Figure 4 there is an example of this motor type.

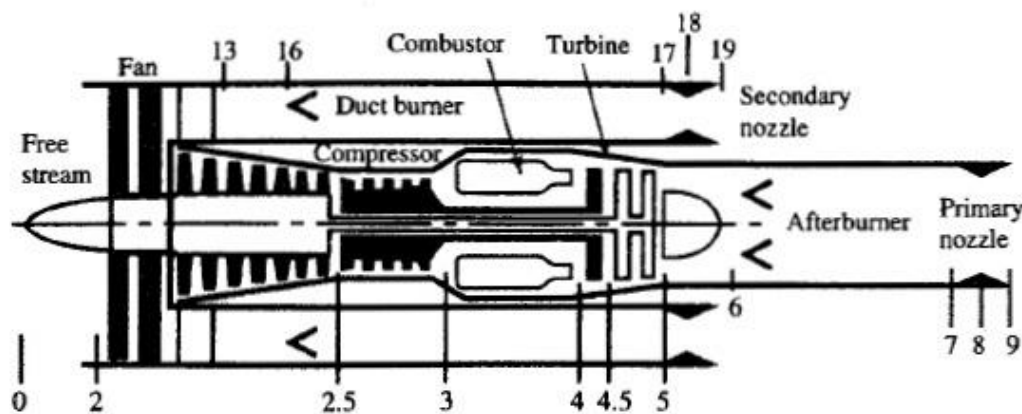


Figure 4 - Seperated Stream Turbofans with Afterburners

1.2.4.1. Spesific Pictures and Examples of Seperated Stream Turbofans with Afterburners

Our project group research this type of motor approximateliy two months. However, we could not find any examples or schematic of this type turbofan. Our class book has only this type turbofan but only it was only a description. Because of these stuations we could not find examples of motor and we could not do this part. Our research motor type websites are included after below and added to references.

<http://jet-engine.net/index.html>

https://en.wikipedia.org/wiki/Category:Turbofan_engines

https://en.wikipedia.org/wiki/Category:High-bypass_turbofan_engines

<https://www.flightglobal.com/the-power-list-top-10-delivered-commercial-turbofans/116576.article>

<https://www.geaviation.com/commercial/engines>

2. Flight Condition Parameters

Separate stream afterburning turbofan engine never produced in real life as we mentioned before. For the flight parameters of our engine, there was only one example in the textbook (Mattingly, 1996, p.416). We investigate the internet sites (airliners.net, theflyingengineer.com, en.wikipedia.org etc.), some research articles and some university archives to collect data for the similar turbofan engine types. Finally, we have analyzed the parameter values and decided the ranges of our flight condition parameters.

ENGINE	type	BPR	OPR	pif	t4	t7	t17	effi comp	effi burn	eff turb	ef	pidmax	pih	nb/nab	pin	um	pifn
leap-1a26	ss		11	40													
leap-1b25	ss		9	43,68													
pw1524g	ss		12	50													
pw1525g	ss		12	50													
pw1922g	ss		12	50													
pw1923g	ss		12	50													
leap-1b28	ss		9	43,68													
leap-1d28	ss		7	43,68													
pw1615g	ss		9	50													
v2527-a5	ss		4,8	32,8													
cfm56-5b4	ss		5,7	32,6													
cfm56-7b27	ss		5,5	32,7													
cfm leap-1b28	ss		9	43,68													
cfm56-7c27	ss		5,8	38,5													
cfm leap-1d28	ss		7	43,68													
pw j18d-17a	mf		1,6	19													
volvo rm8	mfab		0,97	16,5													
cfm56-3c-1	ss		6	30,6													
pw1127g-jm	ss		12,5	50													
GEEnX-1B-70	ss		9,1	43,5													
GE cf34-8c1	ss		5	16,5													
cfm56-7b18	ss		5,6	32,3													
pw4462	ss		4,8	32,3													
trent800	ss		6,5	42													
rb211-524g	ss		4,3	25,8													
ge90-76b	ss	-		40													
f100-229	mfab		0,35	32													
ej200	mfab		0,4	26	4,2												
f404/rm112	mfab		0,34	27													
baseline TF-CLAWS	mfab		0	6,2668		1111	1522		0,9026	0,995	0,8678						
TF-CLAWS	mfab		0,564	26,1	2,359	1723	-		-	-		0,9					
trent900	ss		8,02	41,1													
gp7000	ss		8,17	43,9													
LVL4 BOOK EoGT mat ssab		0.5-3.0	10-30 (Pic)	2.0-5.0	2000	2200	2200	0,9	0,99	0,9	0,89	0,98/0.95-0.98	0,99	0,98	0,99	0,98	0,98
baseline CJ 3000	mf		1,91	36,8	2,276	1818											
CJ 3000	mf		2,75	45	2,44	1826											

Table 1 - Flight Condition Parameter Values of Similar Real Turbofan Engines

2.1. Parameters

M₀: Local mach number is a dimensionless ratio of the speed of the aircraft to the local speed of sound.

T₀: Local temperature of the aircraft which is changing with the height.

γ_c: Specific heat ratio at the upstream of main burner.

C_{pc}: Specific heat capacity at the upstream of main burner at constant pressure.

P₀/P₉: Ratio of the pressures at the freestream and at the exit of the afterburner.

P₀/P₁₉: Ratio of the pressures at the freestream and at the exit of the duct burner.

M_0	T_0	γ_c	C_{pc}	p_0/p_9	p_0/p_{19}
0 - 1.6	216.7	1.4	1.005	1	1

Table 2 - Flight Condition Parameters

2.2. Design Constraints

T_{t4} : Maximum turbine inlet temperature which depends the material.

T_{t7} : Maximum afterburner exit temperature which depends the material.

T_{t17} : Maximum duct burner exit temperature which depends the material.

$T_{t4}(\text{max})$	$T_{t7}(\text{max})$	$T_{t17}(\text{max})$
1700-2000	2200	2200

Table 3 - Design Constraints

2.3. Gas Properties

γ_t : Specific heat ratio at the downstream of main burner.

C_{pt} : Specific heat capacity at the downstream of main burner at constant pressure.

γ_{AB} : Specific heat ratio at the downstream of afterburner.

C_{pAB} : Specific heat capacity at the downstream of afterburner at constant pressure.

γ_{DB} : Specific heat ratio at the downstream duct burner.

C_{pDB} : Specific heat capacity at the downstream of duct burner at constant pressure.

γ_t	C_{pt}	γ_{AB}	C_{pAB}	γ_{DB}	C_{pDB}
1.3	1.235	1.3	1.235	1.3	1.235

Table 4 - Gas Properties

2.4. Fuel Properties

h_{pr} : Lower heating value of fuel is the amount of heat released by combusting a specified quantity of fuel.

h_{pr}
4.28e+04

Table 5 - Fuel Properties

2.5. Component Performances

$\pi_{d \max}$: Total pressure ratio of the inlet with the effects of wall friction

π_{AB} : Total pressure ratio of the afterburner

π_n : Total pressure ratio of the nozzle

π_b : Total pressure ratio of the main burner

π_{fn} : Total pressure ratio of the fan nozzle

π_{DB} : Total pressure ratio of the duct burner

η_b : Isentropic efficiency of the main burner

η_{AB} : Isentropic efficiency of the afterburner

η_{DB} : Isentropic efficiency of the duct burner

η_m : Mechanical efficiency of the coupling between compressor, turbine and fan

e_c : Polytropic efficiency of the compressor

e_f : Polytropic efficiency of the fan

e_t : Polytropic efficiency of the turbine

$\pi_{d \max}$	π_{AB}	π_n	e_c	e_t	η_{AB}	η_m	π_b	π_{DB}	π_{fn}	e_f	η_b	η_{DB}
0.98	0.94	0.98	0.9	0.91	0.95	0.99	0.98	0.94	0.98	0.89	0.99	0.95

Table 6 – Components Performances

2.6. Design Choice Parameters

π_c : Compressor pressure ratio

π_f : Fan pressure ratio

α : Bypass ratio

π_c	π_f	α
5 - 30	2 - 5	0.5 - 5

Table 7 - Design Choice Parameters

3. Cycle analysis

We now consider a turbofan engine cycle with separate exhaust streams in which afterburning may operate in the core stream and duct burning may operate in the fan stream. When both the afterburner and duct burner are in operation, this engine cycle will give substantially higher specific thrust than the basic turbofan cycle, while still providing the low fuel consumption of the basic turbofan engine cycle when both the afterburner and duct burner are turned off.

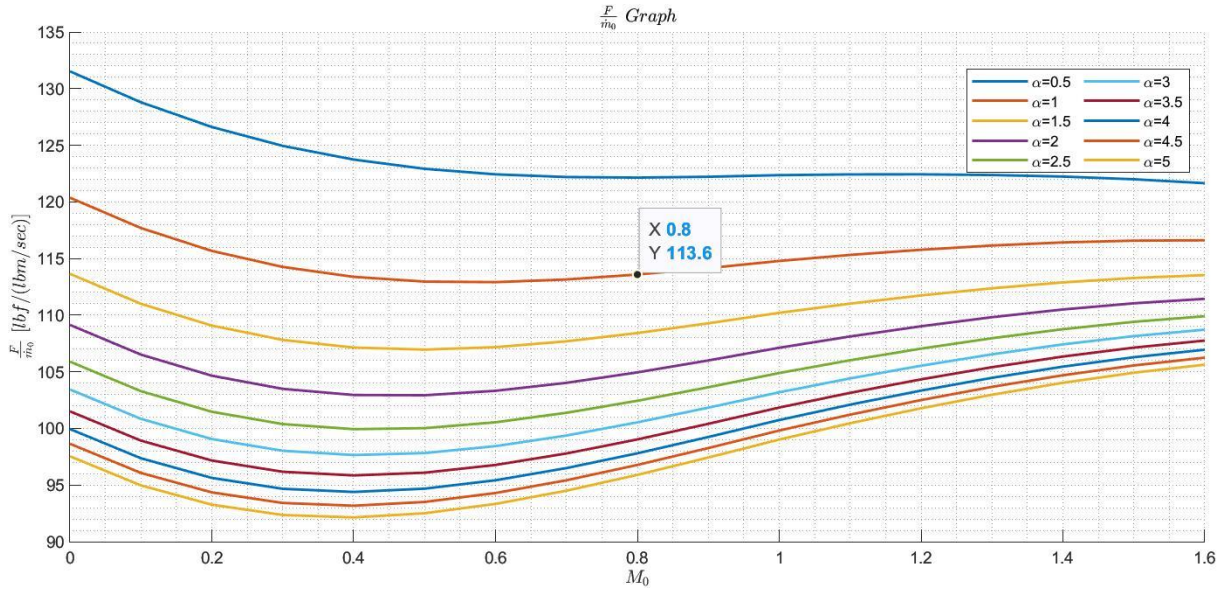


Figure 5 Specific thrust for varying bypass and Mach number

Assumptions:

- Perfect gas upstream of main burner with constant properties
- Perfect gas downstream of main burner with constant properties
- Perfect gas downstream of afterburner with constant properties
- Perfect gas downstream of duct burner with constant properties
- All components adiabatic (no turbine cooling)
- The efficiencies of the compressor, fan, and turbine described through the use of (constant) polytropic efficiencies.

Inputs: $M_0, T_0 (\text{K}, ^\circ \text{R}), \gamma_c, c_{pc} \left(\frac{\text{kJ}}{\text{kg} \cdot \text{K}}, \frac{\text{Btu}}{\text{lbm} \cdot ^\circ \text{R}} \right), \gamma_n, c_{pt} \left(\frac{\text{kJ}}{\text{kg} \cdot \text{K}}, \frac{\text{Btu}}{\text{lbm} \cdot ^\circ \text{R}} \right) \setminus$

$$h_{PR} \left(\frac{\text{kJ}}{\text{kg}}, \frac{\text{Btu}}{\text{lbm}} \right), \gamma_{AB}, c_{pAB} \left(\frac{\text{kJ}}{\text{kg} \cdot \text{K}}, \frac{\text{Btu}}{\text{lbm} \cdot ^\circ \text{R}} \right), \gamma_{DB}, c_{pDB} \left(\frac{\text{kJ}}{\text{kg} \cdot \text{K}}, \frac{\text{Btu}}{\text{lbm} \cdot ^\circ \text{R}} \right) \setminus$$

$$\pi_{dmax}, \pi_b, \pi_{AB}, \pi_{DB}, \pi_n, \pi_{fn}, e_c, e_f, e_t, \eta_b, \eta_{AB}, \eta_{DB}, \eta_m, P_0/P_9$$

$$P_0/P_{19}, T_{t4}(K, ^\circ R), T_{t7}(K, ^\circ R), T_{t17}(K, ^\circ R), \pi_c, \pi_f$$

Outputs:

$$\frac{F}{\dot{m}0} \left(\frac{N}{kg/sec}, \frac{lbf}{lbm/sec} \right), f, f_{AB}, f_{DB}, S \left(\frac{mg/sec}{N}, \frac{lbm/hr}{lbf} \right), \eta_T, \eta_P, \eta_O, \eta_c, \eta_t, \text{ etc.}$$

Equations:

$$R_c = \frac{\gamma_c - 1}{\gamma_c} c_{pc}$$

$$R_t = \frac{\gamma_t - 1}{\gamma_t} c_{pt}$$

$$R_{AB} = \frac{\gamma_{AB} - 1}{\gamma_{AB}} c_{pAB}$$

$$R_{DB} = \frac{\gamma_{DB} - 1}{\gamma_{DB}} c_{pDB}$$

$$a_0 = \sqrt{\gamma_c R_c g_c T_0}$$

$$V_0 = a_0 M_0$$

$$\tau_r = 1 + \frac{\gamma_c - 1}{2} M_0^2$$

$$\pi_r = \tau_r^{\gamma_r/(\gamma_r-1)}$$

$$\eta_r = 1 \quad \text{for } M_0 \leq 1$$

$$\eta_r = 1 - 0.075(M_0 - 1)^{1.35} \quad \text{for } M_0 > 1$$

$$\pi_d = \pi_{dmax} \eta_r$$

$$\tau_\lambda = \frac{c_{\rho t} T_{t4}}{c_{pc} T_0}$$

$$\tau_{\lambda AB} = \frac{c_{pAB} T_{i7}}{c_{pc} T_0}$$

$$\tau_{\lambda DB} = \frac{c_{\rho DB} T_{t17}}{c_{pc} T_0}$$

$$\tau_c = \pi_c^{(\gamma_c-1)/(\gamma_c e_c)}$$

$$\eta_c = \frac{\pi_c^{(\gamma_c-1)/\gamma_c} - 1}{\tau_c - 1}$$

$$\tau_f = \pi_f^{(\gamma_c-1)/(\gamma_c e_f)}$$

$$\eta_f = \frac{\pi_f^{(\gamma_c-1)/\gamma_c} - 1}{\tau_f - 1}$$

Application of the first law of thermodynamics to the burner and solving for the fuel/air ratio gives

$$f = \frac{\tau_\lambda - \tau_r \tau_c}{\eta_b h_{PR} / (c_{pc} T_0) - \tau_\lambda}$$

Solving the power balance between the turbine, compressor, and fan, with a mechanical efficiency of the coupling between the turbine and compressor and fan for the turbine temperature ratio gives

$$\tau_t = 1 - \frac{1}{\eta_m(1+f)} \frac{\tau_r}{\tau_\lambda} [\tau_c - 1 + \alpha(\tau_f - 1)]$$

$$\pi_t = \tau_t^{\gamma_0} \{((\gamma_r - 1)e_\epsilon]$$

$$\eta_t = \frac{1 - \tau_t}{1 - \tau_t^{1/\epsilon_t}}$$

$$\frac{P_{r9}}{P_9} = \frac{P_0}{P_9} \pi_r \pi_d \pi_c \pi_b \pi_t \pi_{AB} \pi_n$$

$$M_9 = \sqrt{\frac{2}{\lambda - 1} \left[\left(\frac{P_{r9}}{P_9} \right)^{(\gamma_{AB}-1)/\gamma_{AB}} - 1 \right]}$$

$$\frac{T_0}{T_0} = \frac{T_n/T_0}{(P_0/P_0)^{(\gamma_{AB}-1)/\gamma_{AB}}}$$

$$\frac{V_0}{a_0} = M_9 \sqrt{\frac{\gamma_{AB} R_{AB} T_9}{\gamma_c R_c T_0}}$$

$$\frac{P_{t19}}{P_{19}} = \frac{P_0}{P_{19}} \pi_r \pi_d \pi_f \pi_{DB} \pi_{fn}$$

In a manner completely similar to the afterburning turbojet, we have

$$M_{19} = \sqrt{\frac{2}{\gamma_{DB} - 1} \left[\left(\frac{P_{t19}}{P_{19}} \right)^{(\gamma_{DB}-1)/\gamma_{DB}} - 1 \right]}$$

$$\frac{T_{19}}{T_0} = \frac{T_{t17}/T_0}{(P_{t19}/P_{19})^{(\gamma_{DB}-1)/\gamma_{DB}}}$$

$$\frac{V_{19}}{a_0} = M_{19} \sqrt{\frac{\gamma_{DB} R_{DB} T_{19}}{\gamma_c R_c T_0}}$$

$$f_{AB} = (1 + f) \frac{\tau_{\lambda AB} - \tau_{\lambda} \tau_t}{\eta_{AB} h_{PR} / (c_{pc} T_0) - \tau_{\lambda AB}}$$

The energy balance across the duct burner gives

$$f_{DB} = \frac{\tau_{\lambda DB} - \tau_r \tau_f}{\eta_{DB} h_{PR} / (c_{pc} T_0) - \tau_{\lambda DB}}$$

The expression for the specific thrust of this engine cycle is

$$\frac{F}{\dot{m}_0} = \frac{1}{1 + \alpha} \frac{a_0}{g_c} \left[(1 + f + f_{AB}) \frac{V_9}{a_0} - M_0 + (1 + f + f_{AB}) \times \frac{R_{AB}}{R_c} \frac{T_9/T_0}{V_0/a_0} \frac{1 - P_0/P_0}{\gamma_c} \right] \\ + \frac{\alpha}{1 + \alpha} \frac{a_0}{g_c} \left[(1 + f_{DB}) \frac{V_{19}}{a_0} - M_0 + (1 + f_{DB}) \times \frac{R_{DB}}{R_c} \frac{T_{19}/T_0}{V_{19}/a_0} \frac{1 - P_0/P_{19}}{\gamma_c} \right] \\ S = \frac{f + f_{AB} + \alpha f_{DB}}{(1 + \alpha)(F/\dot{m}_0)}$$

4. Software

As a software, MATLAB based UI (User Interface) that can be used in parametric cycle analysis of the Separated Stream Afterburner Turbofan is designed. When software is opened, first image of the software is given in Figure 6.

Table	F/m_Odot	S	Fuel-Air Ratio	Eta_c	Eta_t	Propulsive Efficiency	Thermal Efficiency	Overall Efficiency	
M_0	pi_c	pi_f	alpha	T_t4	T_t7	T_t17	F__mdot_0	f	f_AB
1.6000	22	2	0.5000	1.9444e+03	2.2222e+03	2.2222e+03	123.9855	0.0381	0.0251

Figure 6- User Interface

Input parameters and options are grouped into separate panels according to their specification. Output panel is at the bottom right. Output panel has multiple tabs that contains a table and plots of the various output parameters. For every computation, input and output

parameters tabulated together at the Table Tab. At the Figure 7 input panels and output panel are highlighted in red and blue respectively.

The screenshot displays a software interface for engine design, titled "UI Figure". It is divided into several input panels and one output panel.

Input Panels (Highlighted in Red):

- Case Selection:** A dropdown menu set to "Single Condition".
- Flight Conditions:** Includes Mach Number (1.6), Mach No Increment (0.1), Max Mach Number (2.5), T₀ (K) (216.7), gamma_c (1.4), c_{pc} (kJ/(kg K)) (1.005), P_{0/P₉} (1), and P_{0/P₁₉} (1).
- Design Choices:** Includes pi_c (22), pi_c Increment (1), Max pi_c Value (30), pi_f (2), pi_f Increment (0.2), Max pi_f Value (5), By-pass Ratio (0.5), By-Pass Ratio Incr... (0.5), Max By-Pass Ratio (3), and X Axis (alpha, pi_f).
- Design Constraints:** Includes T_{t4} (K) (1944), T_{t4} Increment (50), Max T_{t4} (K) (2100), T_{t7} (K) (2222), T_{t7} Increment (50), Max T_{t7} (K) (2350), T_{t17} (K) (2222), T_{t17} Increment (50), Max T_{t17} (K) (2350), gamma_t (1.3), c_{pt} (kJ/(kg K)) (1.235), gamma_{AB} (1.3), c_{pAB} (kJ/(kg K)) (1.235), gamma_{DB} (1.3), c_{pDB} (kJ/(kg K)) (1.235), and h_{PR} (kJ/kg) (4.28e+04).
- Gas Properties:** Includes gamma_t (1.3), c_{pt} (kJ/(kg K)) (1.235), gamma_{AB} (1.3), c_{pAB} (kJ/(kg K)) (1.235), gamma_{DB} (1.3), and c_{pDB} (kJ/(kg K)) (1.235).
- Fuel Properties:** Includes h_{PR} (kJ/kg) (4.28e+04).
- Component Performances:** Includes pi_{d_max} (0.98), pi_b (0.98), pi_{AB} (0.94), pi_{DB} (0.94), pi_n (0.98), pi_{fn} (0.98), e_c (0.9), e_f (0.89), e_t (0.91), eta_b (0.99), eta_{AB} (0.95), eta_{DB} (0.95), and eta_m (0.99).

Output Panel (Highlighted in Blue):

Table

Table	F/m _{0dot}	S	Fuel-Air Ratio	Eta _c	Eta _t	Propulsive Efficiency	Thermal Efficiency	Overall Efficiency	
M ₀	pi _c	pi _f	alpha	T _{t4}	T _{t7}	T _{t17}	F _{mdot_0}	f	f _{AB}
1.6000	22	2	0.5000	1.9444e+03	2.2222e+03	2.2222e+03	123.9855	0.0381	0.0251

Figure 7 – Input and Output Panels

At the upper left Case Selection Panel can be seen (highlighted in purple) at Figure 8. There is a dropdown menu for the case selection and available solver options are seen in the menu. When case is selected, required edit fields are enabled and unnecessary fields are disabled.

Case Selection

Case: **Varying Bypass and $\pi_{i,f}$**

Design Constraints

T_{i4} (K): 1944
 T_{i4} Increment: 50
 Max T_{i4} (K): 2100
 T_{i7} (K): 2222
 T_{i7} Increment: 50
 Max T_{i7} (K): 2350
 T_{t17} (K): 2222
 T_{t17} Increment: 50
 Max T_{t17} (K): 2350

Gas Properties

$\gamma_{i,t}$: 1.3
 $c_{p,t}$ (kJ/(kg K)): 1.235
 γ_{AB} : 1.3
 c_{pAB} (kJ/(kg K)): 1.235
 γ_{DB} : 1.3
 c_{pDB} (kJ/(kg K)): 1.235

Fuel Properties

h_{PR} (kJ/kg): 4.28e+04

Component Performances

$\pi_{i,d,max}$: 0.98
 $\pi_{i,AB}$: 0.94
 $\pi_{i,n}$: 0.98
 e_c : 0.9
 e_t : 0.91
 η_{AB} : 0.95
 η_m : 0.99
 $\pi_{i,b}$: 0.98
 $\pi_{i,DB}$: 0.94
 $\pi_{i,fn}$: 0.98
 e_f : 0.89
 η_b : 0.99
 η_{DB} : 0.95

Table	F/m ₀ dot	S	Fuel-Air Ratio	Eta _c	Eta _t	Propulsive Efficiency	Thermal Efficiency	Overall Efficiency	
M ₀	$\pi_{i,c}$	$\pi_{i,f}$	α	T_{i4}	T_{i7}	T_{t17}	F _{mdot,0}	f	τ_{AB}
1.6000	22	2	0.5000	1.9444e+03	2.2222e+03	2.2222e+03	123.9855	0.0381	0.0251

Design Choices

$\pi_{i,c}$: 22
 $\pi_{i,c}$ Increment: 1
 Max $\pi_{i,c}$ Value: 30
 Min $\pi_{i,f}$ Value: 2
 $\pi_{i,f}$ Increment: 0.2
 Max $\pi_{i,f}$ Value: 5
 Min By-pass Ratio: 0.5
 By-Pass Ratio Incr...: 0.5
 Max By-Pass Ratio: 3
 X Axis: ☒ alpha ☐ $\pi_{i,f}$
 Compute

Figure 8 - Case Selection

When all of the parameters are entered, by clicking the Compute button program computes the outputs, tabulates and plots the result for the selected X-Axis (Figure 9).

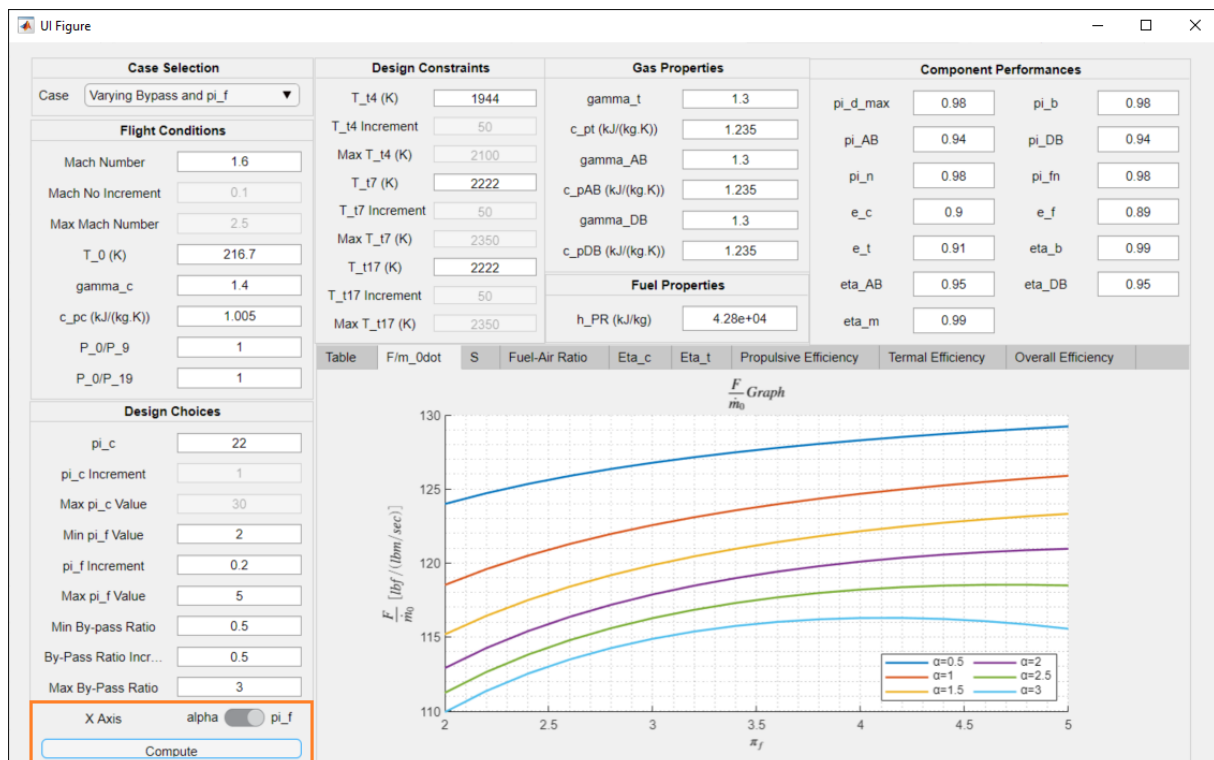


Figure 9 - Computing Program

Source code of the software is given in the Appendix. Software can be run MATLAB 2019b or later versions.

5. Validation

In this topic, validation study was carried out to confirm the accuracy of the data of the Parametric Cycle analysis program. Since an error was detected in the solution of the sample question of Afterburning Turbofan with Separate Exhaust (SSATF) in the reference book, the validation study was done with Example 7-6, which is the example of Separated Exhaust Stream Turbofan Engine (SSTF). Example 7-6 Unlike our analysis program, since it does not contain an afterburner, it is necessary to make adjustments in SSATF cycle analysis formulas and parametric analysis inputs.

First, the heat energy added to the flow in the duct burner and afterburner regions in Figure 4 is reduced to zero value in order to neutralize the afterburner event in the analysis. As seen in Equation (1), the fuel spent during the afterburner functions to increase the temperature of the air from T_{t6} to T_{t7} using the energy balance across the duct burner. For this reason, when the T_{t6} temperature is written instead of the T_{t7} value in the analysis calculations, the fuel will not be consumed in the afterburner part and the flow will progress to the nozzle at the same temperature. T_{t6} is calculated through equation (2). Since the T_{t4} value in the equation is known as the input depressor in the analysis, the T_{t6} value was obtained by taking into account the temperature drop in the turbine and the burning in the afterburner was prevented by writing the T_{t7} temperature instead. In this way, the flow temperature remained the same and the gas characteristics at the afterburner station were taken with the same values as the turbine gas characteristics shown in Figure 10.

$$(\dot{m}_0 + \dot{m}_f)c_{pt}T_{t6} + \eta_{AB}\dot{m}_{fAB}h_{PR} = (\dot{m}_0 + \dot{m}_f + \dot{m}_{fAB})c_{pAB}T_{t17} \quad (1)$$

$$T_{t6} = T_{t4} * \frac{T_{t5}}{T_{t4}} * \frac{T_{t6}}{T_{t5}} = T_{t4} * \tau_t * 1 = 3000 * 0.54866 * 1 = 1645.82 \text{ } ^\circ R = 914.3465 \text{ } K \quad (2)$$

The same method will be examined in the duct burner section. When the equation (7-64) is examined, it is seen that the fuel spent during the duct burner increases the air from T_{t16} temperature to T_{t17} temperature. In the cycle analysis formulas, if T_{t16} value is written to T_{t17} temperature, there will be no burning in the duct burner part of the engine. T_{t16} is calculated by equation (4). As a result, since there will be no combustion in the duct burner part of the engine, the flow in this region will have the same gas characteristics as the flow in the fan part. Gas properties are shown in Figure 10.

$$\dot{m}_F c_{pC} T_{t16} + \eta_{DB} \dot{m}_{fDB} h_{PR} = (\dot{m}_F + \dot{m}_{fDB}) c_{pDB} T_{t17} \quad (3)$$

$$T_{t16} = T_0 * \frac{T_{t2}}{T_0} * \frac{T_{t13}}{T_{t2}} = T_0 * \tau_r * \tau_f = 390 * 1.128 * 1.1857 = 1645.82 \text{ } ^\circ R = 289.8 \text{ } K \quad (4)$$

Afterburner and duct burner efficiencies were taken as 1 to avoid loss of efficiency in the combustion parts. Finally, other input parameters shown in Figure 10 are entered into the program and the results are obtained by operating with the single condition case option. The program results are compared with reference book solutions in Table 8.

Group 4	$\frac{F}{\dot{m}_0}$ (lbf/(lbm/sec))	f	S (lbm/hr)/lbf)	η_c	η_f	η_t	η_P	η_T	η_o
Book	18.02	0.02868	0.6366	0.842	0.882	0.92	0.66124*	0.409	0.271*
Program	18.0248	0.0287	0.6364	0.8417	0.8815	0.92	0.6627	0.409	0.271

* There is an error in the calculation in the book, the value of V_{19}/a_0 is wrong

Table 8 Comparison of reference book and program results

The screenshot shows a software interface for a single condition program. It includes several input sections: Case Selection (Single Condition), Flight Conditions (Mach Number 0.8, T0 216.7 K, gamma_c 1.4, etc.), Design Constraints (T14 1667 K, T17 914.3 K, etc.), Gas Properties (gamma_t 1.33, c_pt 1.156, etc.), Component Performances (pi_d_max 0.99, pi_b 0.96, etc.), and Design Choices (pi_c 36, pi_f 1.7, etc.). A 'Compute' button is at the bottom left. On the right, a table displays the results of the simulation.

Propulsive Termal and Overall Efficiency															
T17	T117	Fmdot_0	S	Fuel	Eta_c	Eta_t	r_AB	r_DB	S	eta_c	eta_t	eta_f	eta_P	eta_T	eta_O
914.3465	289.7888	18.0248	0.0287	-2.5065e-06	-1.0094e-09	0.6364	0.8417	0.9200	0.8815	0.6627	0.4099	0.2716			

Figure 10 Single condition program interface and results

When the results of Figure 10 are examined, it is seen that the parameters f_{AB} and f_{DB} are equal to zero. As a result, the afterburner was successfully disabled. When Table 8 is examined, it is observed that the results are consistent and the validation study is completed.

6. Parameter Effects and Plots

6.1. Outputs with Respect to Varying Bypass Ratio (α) and Compressor Pressure Ratio (π_c)

This screenshot shows the same program interface as Figure 10, but with different input values. The Flight Conditions section has Mach Number 1.6, T0 216.7 K, gamma_c 1.4, c_pc 1.005, P_0/P_9 1, and P_0/P_19 1. The Design Constraints section has T14 2000 K, T17 2200 K, etc. The Gas Properties section has gamma_t 1.3, c_pt 1.235, etc. The Component Performances section has pi_d_max 0.98, pi_b 0.98, etc. The Design Choices section has pi_c 5, pi_f 1, etc. The 'Compute' button is still present.

Figure 11 - Outputs with Respect to Bypass Ratio and Compressor Pressure Ratio

6.1.1. Fuel/Air Ratio

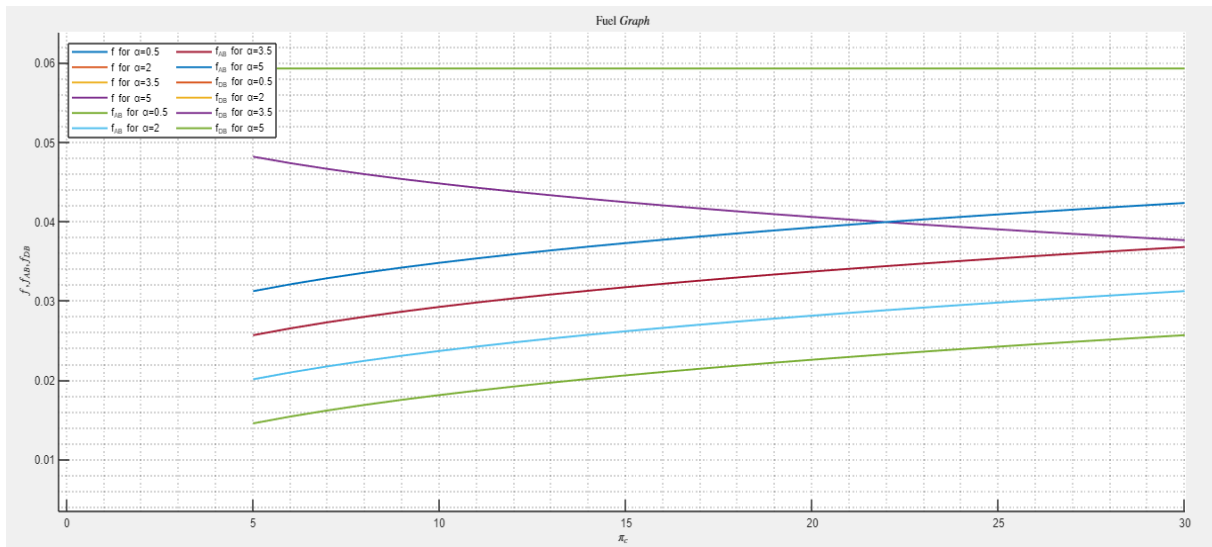


Figure 12 - Fuel/Air Ratio

Duct Burner Fuel/Air Ratio:

The green curve on the top shows the fuel/air ratio in Duct Burner. It seems to be overlapped because it is equal for each bypass ratio. Against the change of bypass ratio and compressor compression ratio, the fuel-air ratio of the Duct burner did not give a response and remained stable. As the limiting factor of the duct burner fuel-air ratio is the duct burner outlet temperature, it is normal that it does not respond to these two variables. This value is fixed at around 0.06. The differentiation of this value is related to the energy of the air coming to the Duct burner. That is, the parameters that can change this value can be specified as the T_0 M_0 values in the flight condition and the fan compression ratio. Since the total temperature at the duct burner outlet is limited from the top; Since the incoming air has a high temperature, the Mach number is high, and the fan compression ratio is high, the energy of the air entering the duct burner will also be high. Less fuel will be sufficient to reach the limited value.

Combustion Chamber Fuel/Air Ratio:

The purple curve, just below the previous curve (2nd curve on the graph), shows the fuel/air ratio in the core combustion chamber. This value also seems to be overlapped because it does not change with the bypass ratio. The bypass ratio is ineffective for the internal energy of the air entering the core combustion chamber. The compressor pressure ratio shows the work done in the compressor to air. In other words, as the compressor pressure ratio increases, the energy of air at the core combustion chamber entrance increases. For this reason, depending on the limited turbine inlet temperature, the required fuel will decrease as the compressor pressure ratio increases.

Core Afterburner Fuel/Air Ratio:

The remaining 4 curves show the fuel/air ratio of the afterburner in the core. The bypass ratios are as follows, from top to bottom; 1_Dark Blue: 5, 2_Red: 3.5, 3_Light Blue: 2,

4_Green: 0.5. As can be seen here unlike the others; The fuel that can be burnt increased as the bypass ratio increased and the compressor pressure ratio increased. If the energy of the air coming into the afterburner inlet is low, to reach the limited afterburner outlet temperature; more fuel can be burned. The increasing compressor pressure ratio causes a decrease in the temperature of the air coming to the afterburner inlet because the energy required to reach a high compression ratio is taken from the turbine. Taking more energy from the turbine while the turbine inlet temperature is constant at its upper limit means to lower the turbine outlet temperature more. In other words, getting more energy from the turbine to increase the compressor pressure ratio will decrease the temperature of the air coming to the core afterburner. In this way, more fuel can be burned. It also appears that for a fixed compressor pressure ratio, the fuel/air ratio increases as the bypass ratio increases. Because the total temperature in the turbine decreases as the bypass ratio increases and the temperature of the air entering the afterburner decreases; In this way, more fuel can be burned to reach the maximum afterburner outlet temperature.

6.1.2. Specific Thrust

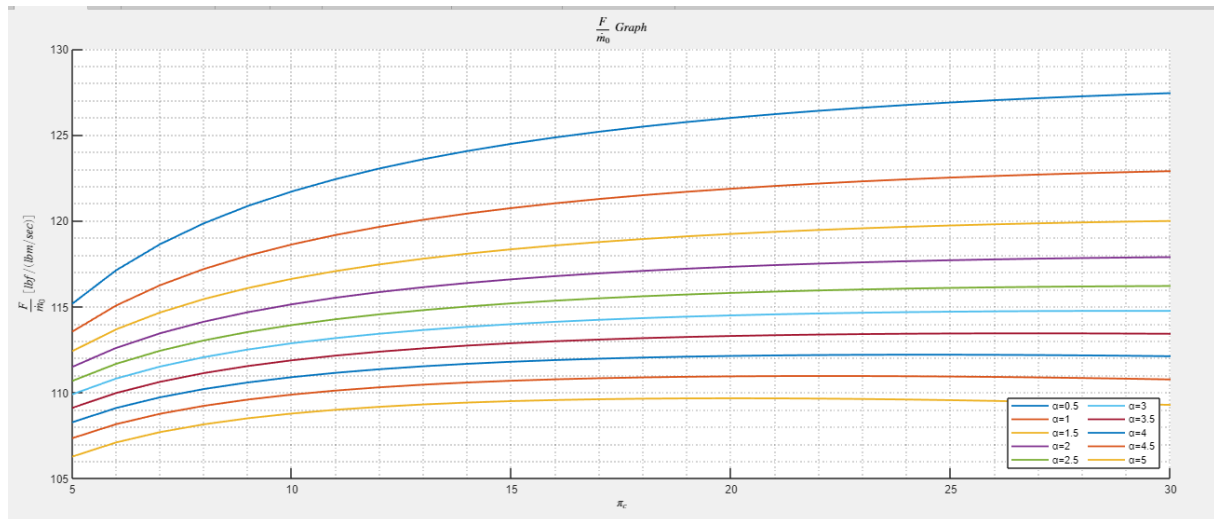


Figure 13 - Specific Thrust

This graph shows compressor pressure ratio and specific thrust values from 0.5 bypass ratio in ascending order to the top, from 0.5 increments to 5 bypass ratio. In the first reading of the graph, the increased compressor pressure ratio increased the specific thrust for a fixed bypass ratio. The rate of this increase decreased as the pressure ratio increased, and even became negative for low bypass ratios. That is, the compressor compression ratio has an optimum for specific thrust. Although we know that high compression ratios will benefit in ideal cycles, the energy drawn from the turbine, whose material inlet temperature is constant, may not meet the very high compressor pressure ratios. Also in a real situation; both to give higher energy to the air and to take higher energy from the air can decrease the efficiency, and the structural weight may increase to meet these loads as the loads on the parts will increase. For these reasons, while determining the optimum; As the compressor pressure ratio increases, the increase rate of the specific thrust decreases, and the values at reasonable levels where it starts to decrease should be selected.

In the second reading of the graph, at a fixed compressor pressure ratio, the specific thrust value decreases as the bypass ratio increases. This does not mean that the total thrust is reduced. Because this value is proportional to the mass flow. The main purpose of keeping the bypass ratio high in Turbofan engines is to accelerate the air with more mass to lower speeds and produce thrust, and the specific thrust of Turbofan engines is generally low. This value gives us information about its size due to the size of the air intake of the engine. That is, as the bypass ratio increases, a larger motor is required for a constant thrust.

6.1.3. Specific Fuel Consumption

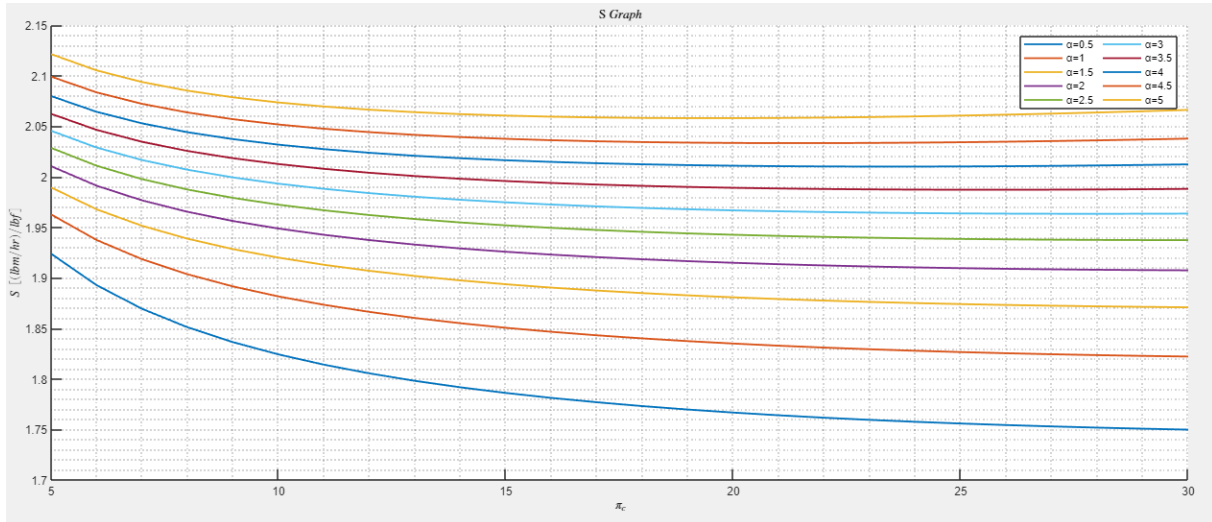


Figure 14 - Specific Fuel Consumption

In this graph, the curves started from 0.5 in order from bottom to top according to the bypass ratio and reached 5 in 0.5 increments. In other words, the bottom blue curve shows a 0.5 bypass ratio, while the top yellow curve shows a 5 bypass ratio. We can see that the specific fuel consumption decreases as the compressor pressure ratio increases. From the graph, we can see the philosophy of the cycles as the rate of compression increases, the energy received increases compared to the energy given. However, this situation turns into a disadvantage when high compressor pressure ratios are reached. In this graph, an optimum with reasonable values should be sought.

In the secondary reading of the graph, while the bypass ratio increases for a fixed compressor pressure ratio, the specific fuel consumption increases. This is a contrast to the separate flow turbofan. Because the main purpose of increasing the bypass ratio is the reduction in fuel consumption. The contradiction in this engine is due to the core afterburner. As the first graphic interprets, the core afterburner can burn more fuel as the bypass ratio increases. Because the total temperature in the turbine decreases as the bypass ratio increases and the temperature of the air entering the afterburner decreases; In this way, more fuel can be burned to reach the maximum afterburner outlet temperature. This situation increases the specific fuel consumption, contrary to what is expected, instead of decreasing, as the bypass ratio increases.

6.2. Efficiency

6.2.1. Propulsive Efficiency

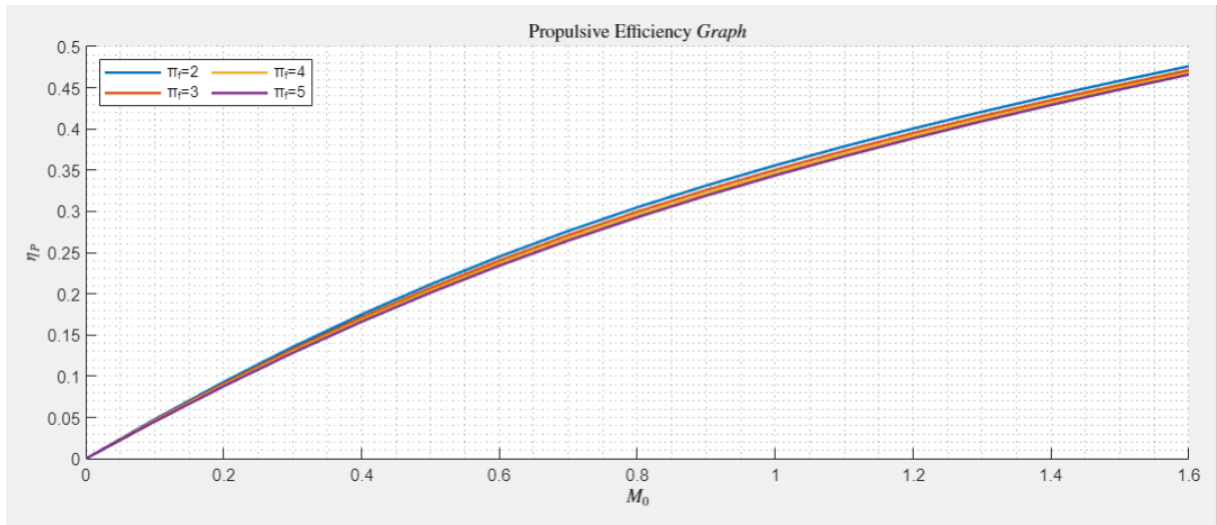


Figure 15 - Propulsive Efficiency with Varying Different Fan Pressure Ratios

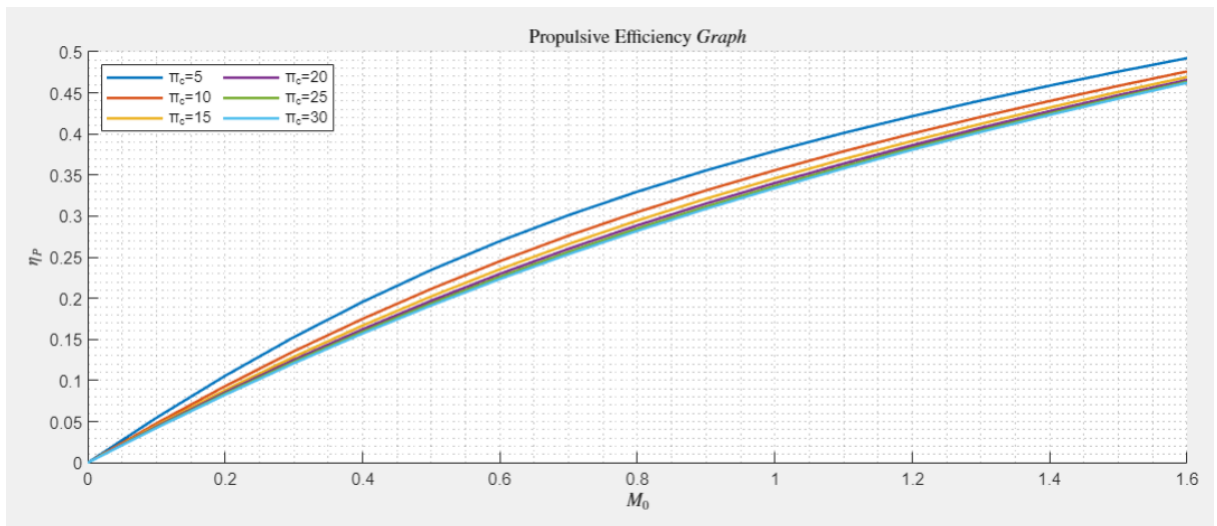


Figure 16 - Propulsive Efficiency with Varying Different Compressor Pressure Ratios

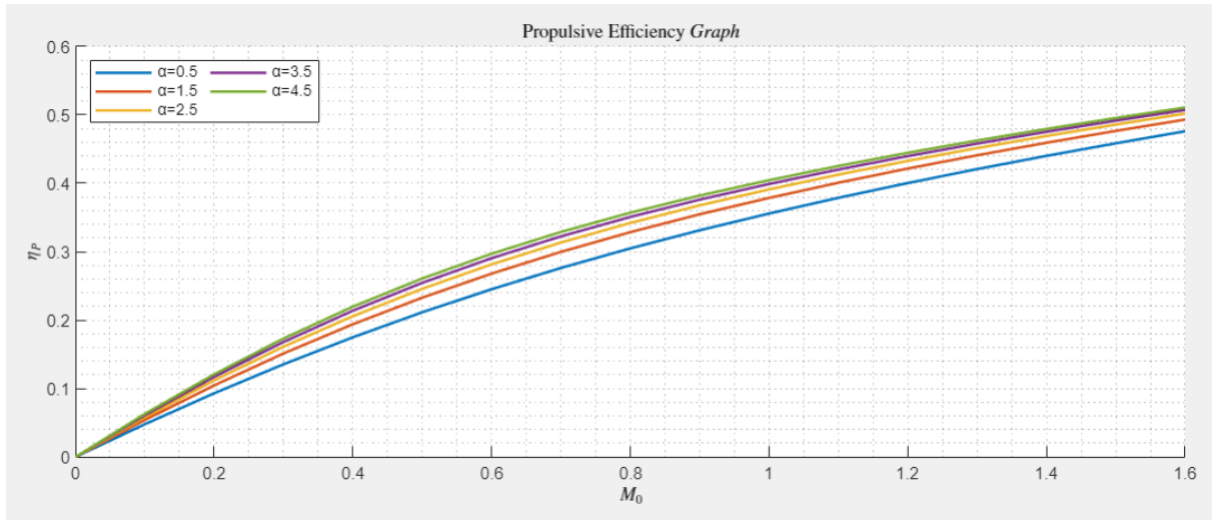


Figure 17 - Propulsive Efficiency with Varying Different Bypass Ratios

The graphs illustrate the propulsive efficiency with varying Mach number, fan pressure ratio, compressor pressure ratio and by-pass ratio. With increasing Mach number the propulsive efficiency is also increasing as expected in all engines. But any increase in fan pressure ratio and compressor pressure ratio has a slight negative effect on the propulsive efficiency of the engine because of the extra losses in compressor and fan. Contrarily, the augmentation of the by-pass number increases the propulsive efficiency.

6.2.2. Thermal Efficiency

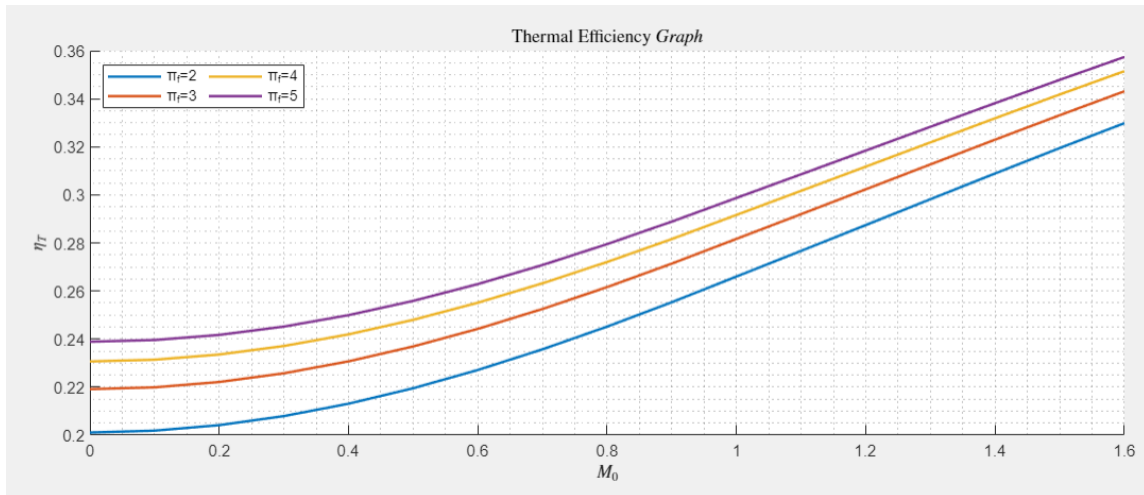


Figure 18 - Thermal Efficiency with Varying Different Fan Pressure Ratios

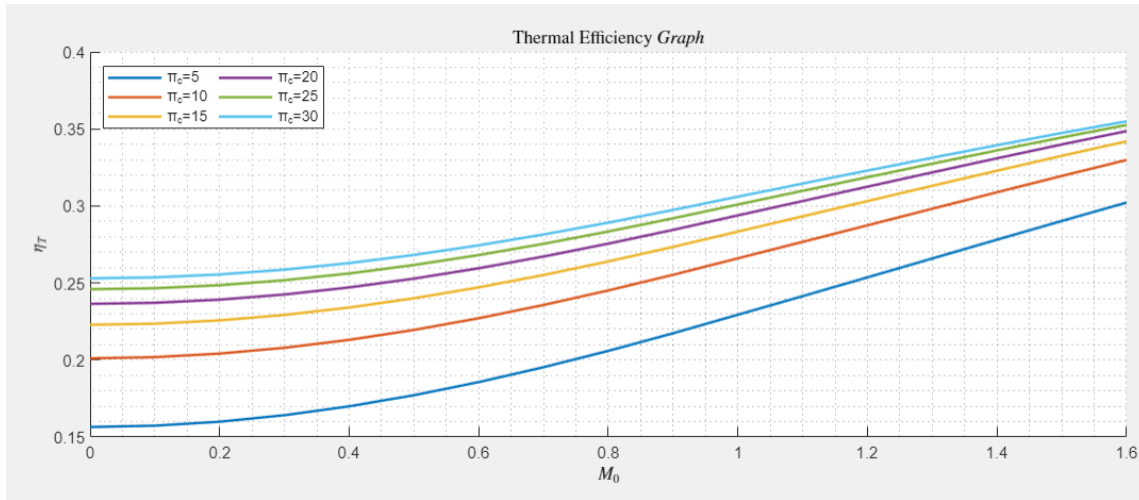


Figure 19 - Thermal Efficiency with Varying Different Compressor Pressure Ratios

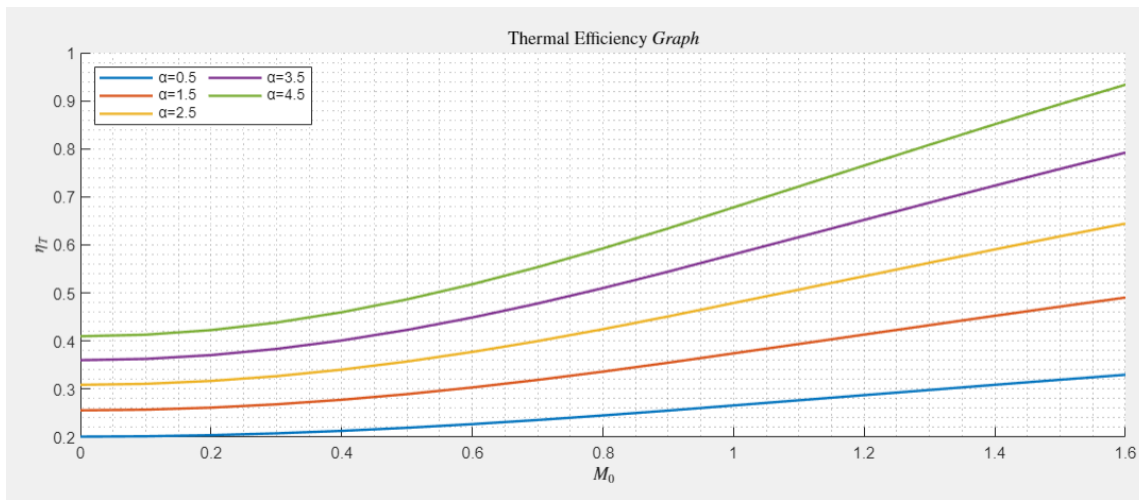


Figure 20 - Thermal Efficiency with Varying Different Bypass Ratios

In these graphs, the change of the thermal efficiency with varying Mach number, fan pressure ratio, compressor pressure ratio and by-pass number can be observed. With increasing Mach number the thermal efficiency is rising. The biggest increase in the thermal efficiency occurs with the rising by-pass number. Also, fan pressure ratio and compressor pressure ratio has a positive effect on the thermal efficiency due to less fuel need.

6.2.3. Overall Efficiency

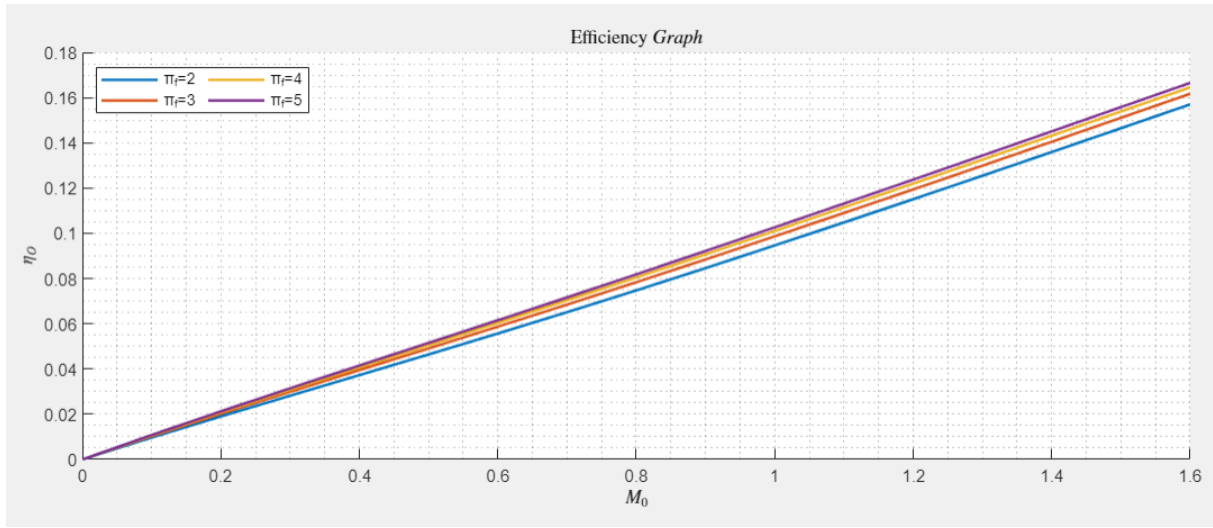


Figure 21 – Overall Efficiency For Different Fan Pressure Ratios

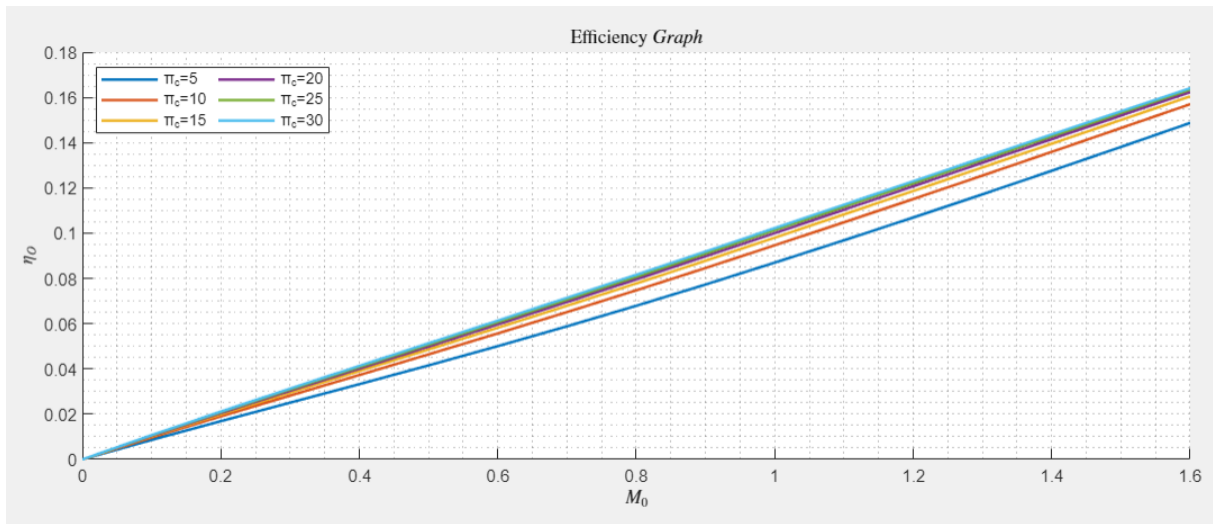


Figure 22 - Overall Efficiency For Different Compressor Pressure Ratios

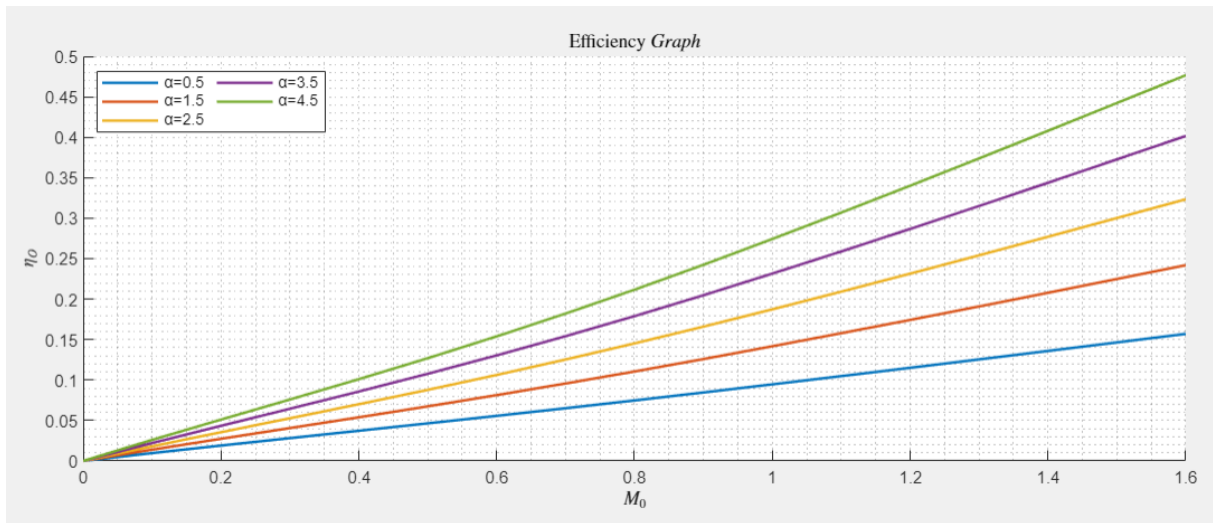


Figure 23 - Overall Efficiency For Different Bypass Ratios

The overall efficiency is mostly dominated by the propulsive efficiency and the Mach number. In the first two graphs, the maximum efficiency is about 0.16 for all fan pressure and compressor pressure ratios. But the by-pass number has a very significant effect on the overall efficiency because of the effects on the thermal efficiency and the overall efficiency reaches to 0.45 with a 4.5 by-pass ratio.

6.3. Outputs with Respect to Varying Bypass Ratio (α) and Fan Pressure Ratio (π_f)

Design Choices

pi_c	<input type="text" value="15"/>
pi_c Increment	<input type="text" value="1"/>
Max pi_c Value	<input type="text" value="30"/>
Min pi_f Value	<input type="text" value="1"/>
pi_f Increment	<input type="text" value="0.2"/>
Max pi_f Value	<input type="text" value="5"/>
Min By-pass Ratio	<input type="text" value="0.5"/>
By-Pass Ratio Increment	<input type="text" value="0.5"/>
Max By-Pass Ratio	<input type="text" value="5"/>
X Axis alpha <input checked="" type="checkbox"/> pi_f <input type="checkbox"/>	
<input type="button" value="Compute"/>	

Figure 24 Design Choices

6.3.1. Specific Thrust

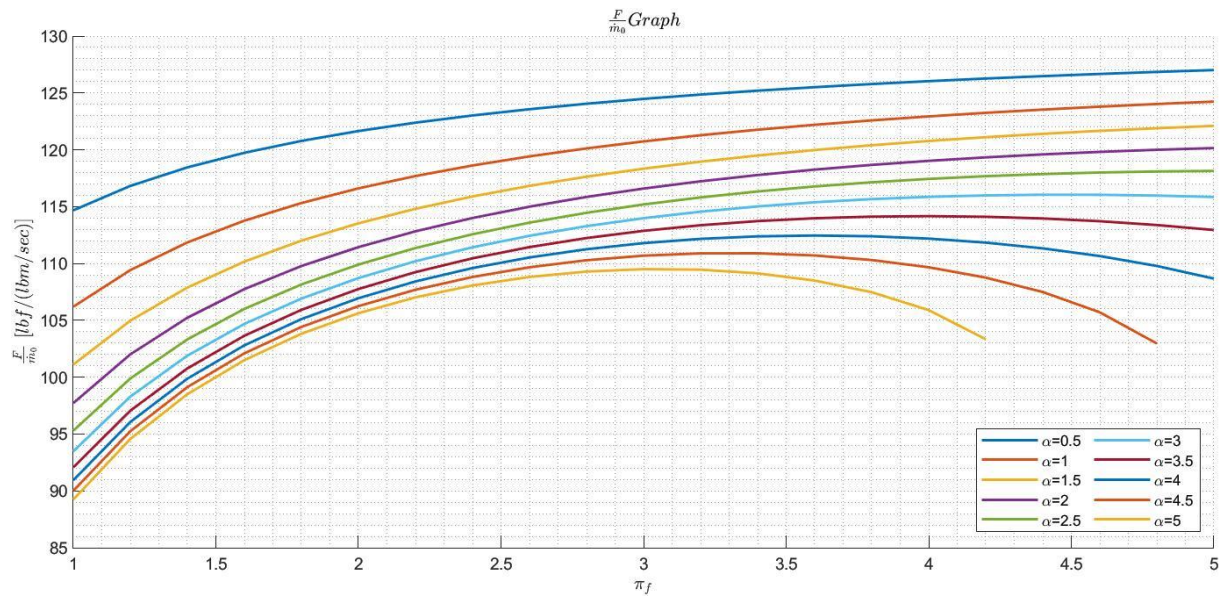


Figure 25 Specific Thrust

For lower bypass ratios, with rising π_f the air can be burned better due to better compression and the specific thrust obtained increases. At high bypass ratios, the power needed for the fan increases with rising fan pressure ratio, and the power which provides the propulsive force decreases. This power reduction causes the specific thrust to decrease.

For constant fan pressure ratio, specific thrust decreases with rising bypass ratio. Specific thrust is related to the engine size and generally has lower values.

6.3.2. Specific Fuel Consumption

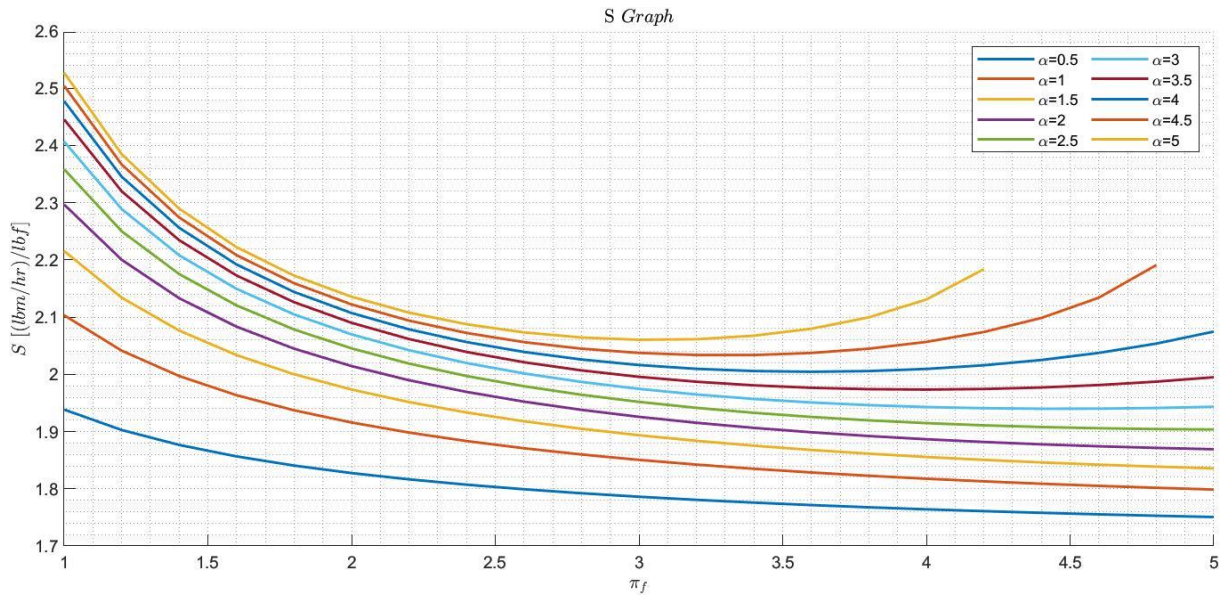


Figure 26 Specific Fuel Consumption

Specific fuel consumption is related to the specific thrust and behaves in the opposite way. The value of the specific thrust increases with the rising bypass ratio and decreasing fan pressure ratio.

We see that the SFC and specific thrust graphs have opposing characteristics as expected. Both graphs represent an optimum fan pressure ratio. This ratio is lower than the ones in ideal engine. Higher bypass ratios correspond to higher values of SFC and lower values of specific thrust. This contradicts in theory because the graphs that are presented at this section belong to a separate flow turbofan without an afterburner. In the case of a separate flow turbofan engine with afterburner, the afterburner can burn higher amounts of fuel by the increment of bypass ratio.

If there is any odd behavior or imaginary solutions, the system indicates that a turbine fails to provide enough power to fan.

6.4. Outputs with Respect to Varying Bypass Ratio (α) and Mach Number (M_0)

Design Choices	
pi_c Value	15
pi_c Increment	1
Max pi_c Value	30
pi_f Value	2
pi_f Increment	0.2
Max pi_f Value	5
Min By-pass Ratio	0.5
By-Pass Ratio Increment	0.5
Max By-Pass Ratio	5
X Axis	M_0 <input type="checkbox"/> alpha <input checked="" type="checkbox"/>
Compute	

Figure 27 Design Choices

6.4.1. Specific Thrust

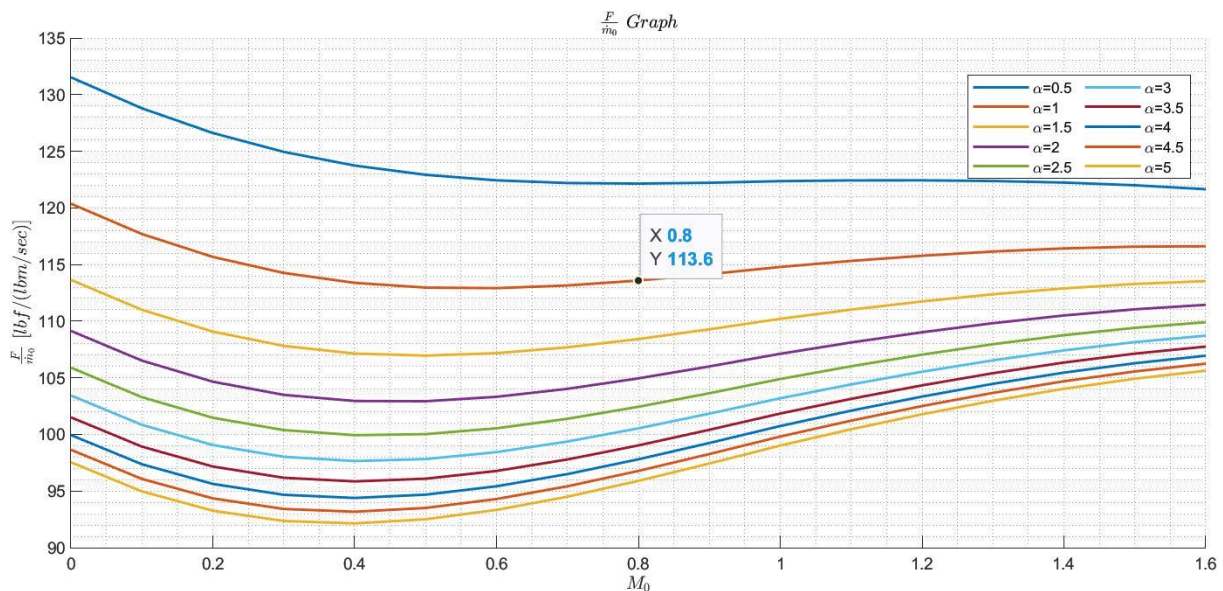


Figure 28 Specific Thrust

At constant bypass ratio, between 0 - 0.4 Mach, specific thrust decreases because flight Mach number rises and the exit velocity remains constant. After 0.4 Mach, ram pressure ratio increases with rising Mach number, and the total temperature of the air increases. The augmentation of the total temperature causes the specific thrust to increase.

At constant Mach number, with a rising bypass ratio, the specific thrust decreases because of the growing engine size.

6.4.2. Specific Fuel Consumption

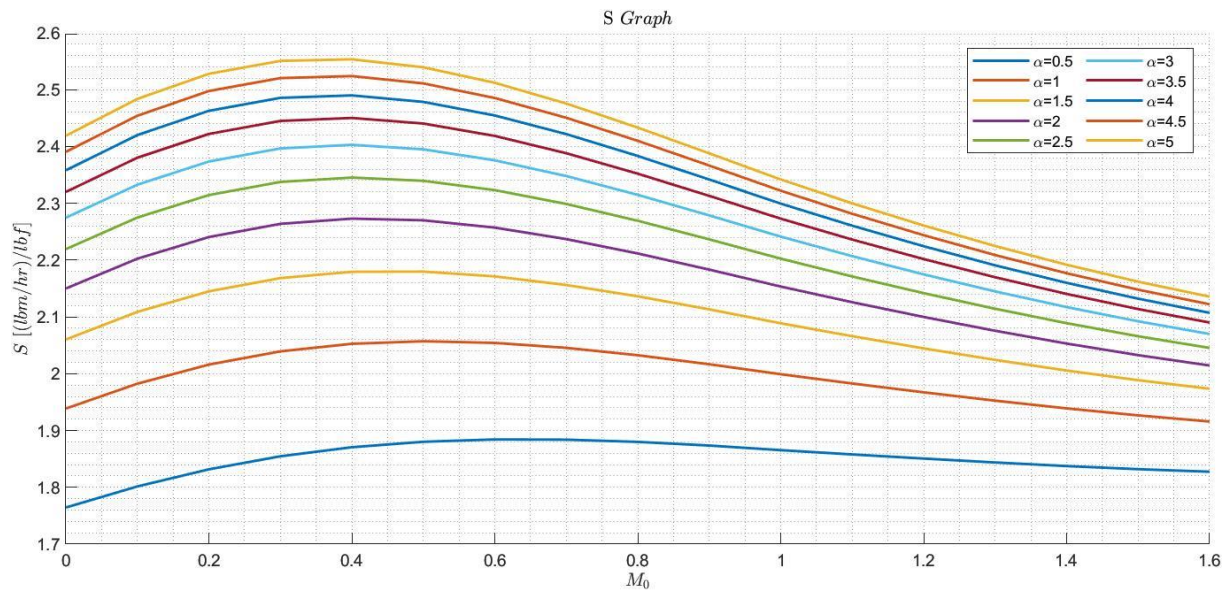


Figure 29 Specific Fuel Consumption

Specific fuel consumption is related to the specific thrust and behaves in the opposite way.

6.4. Outputs with Respect to Tt4, Tt7, Tt17 and Mach Number (M_0)

Design Choices		Design Constraints	
pi_c Value	15	Min T_t4 (K)	1570
pi_c Increment	1	T_t4 Increment	86
Max pi_c Value	30	Max T_t4 (K)	2000
pi_f Value	3	T_t7 (K)	1920
pi_f Increment	0.2	T_t7 Increment	86
Max pi_f Value	5	Max T_t7 (K)	2350
By-pass Ratio	0.5	T_t17 (K)	2200
By-Pass Ratio Increm	0.5	T_t17 Increment	86
Max By-Pass Ratio	3	Max T_t17 (K)	2350
X Axis	M_0 <input type="checkbox"/> T_t4 <input checked="" type="checkbox"/>		
<input type="button" value="Compute"/>			

Figure 30 Design Choices

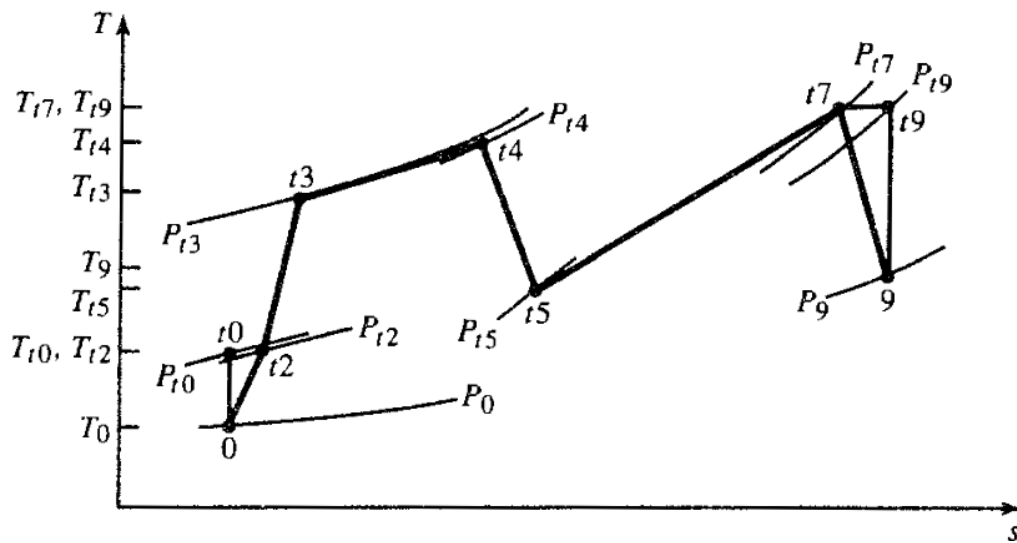


Figure 31 Brayton Cycle

The engine gains the force it needs mainly from the T_{t4} and T_{t7} temperature offset as it is plotted in Brayton Cycle. T_{t4} is temperature after the combustion chamber and T_{t7} is the temperature after the afterburner. If T_{t4} temperature increases more fuel is burned and more thrust is generated as a result. T_{t4} determines how much fuel that can be burned in the combustion chamber, hence it is a direct limitation on how much fuel that can be burned. Therefore, S decreases with increased T_{t4} . This can be seen in the following figures:

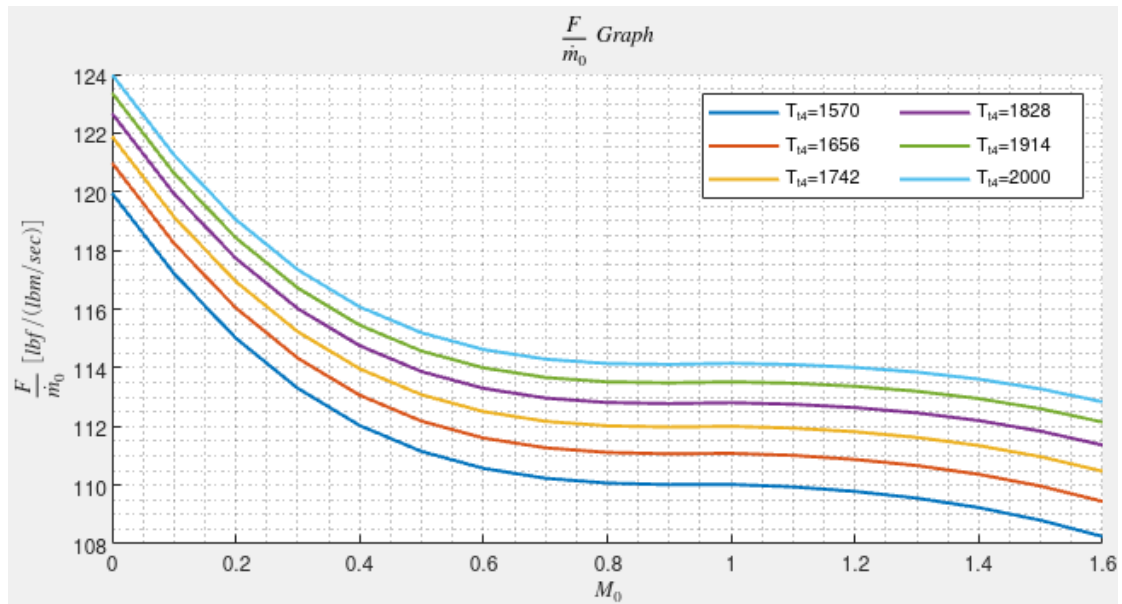


Figure 32 Specific Thrust

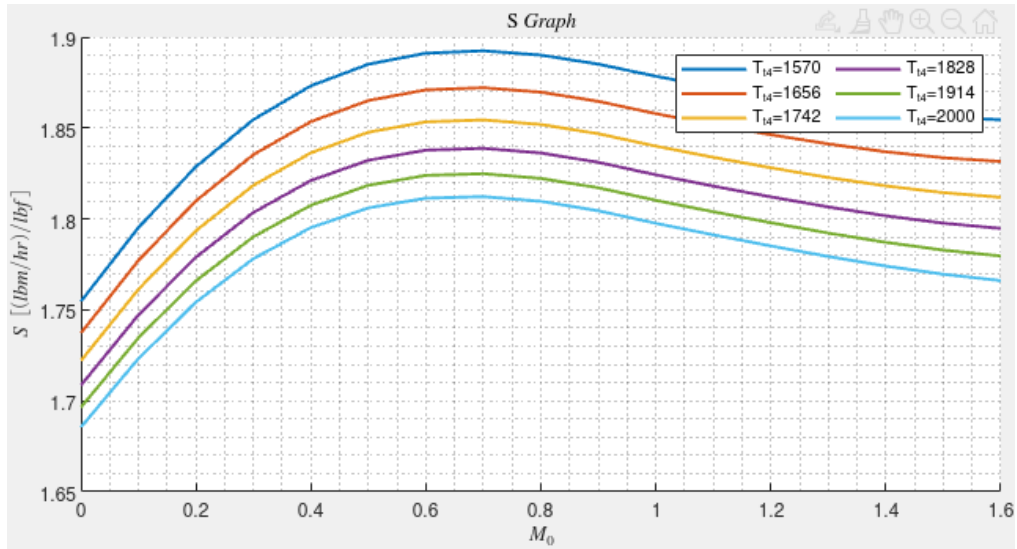


Figure 33 Specific Fuel Consumption

Design Constraints	
T ₁₄ (K)	2000
T ₁₄ Increment	86
Max T ₁₄ (K)	2000
Min T ₁₇ (K)	1920
T ₁₇ Increment	86
Max T ₁₇ (K)	2350
T _{t17} (K)	2200
T _{t17} Increment	86
Max T _{t17} (K)	2350

Figure 34 Design Choices

In the case of increased Tt7 temperature, the force is obviously increased. This ideology is related to the same Brayton Cycle principle explained above. To achieve more thrust, more fuel needs to be burned in the afterburner zone.

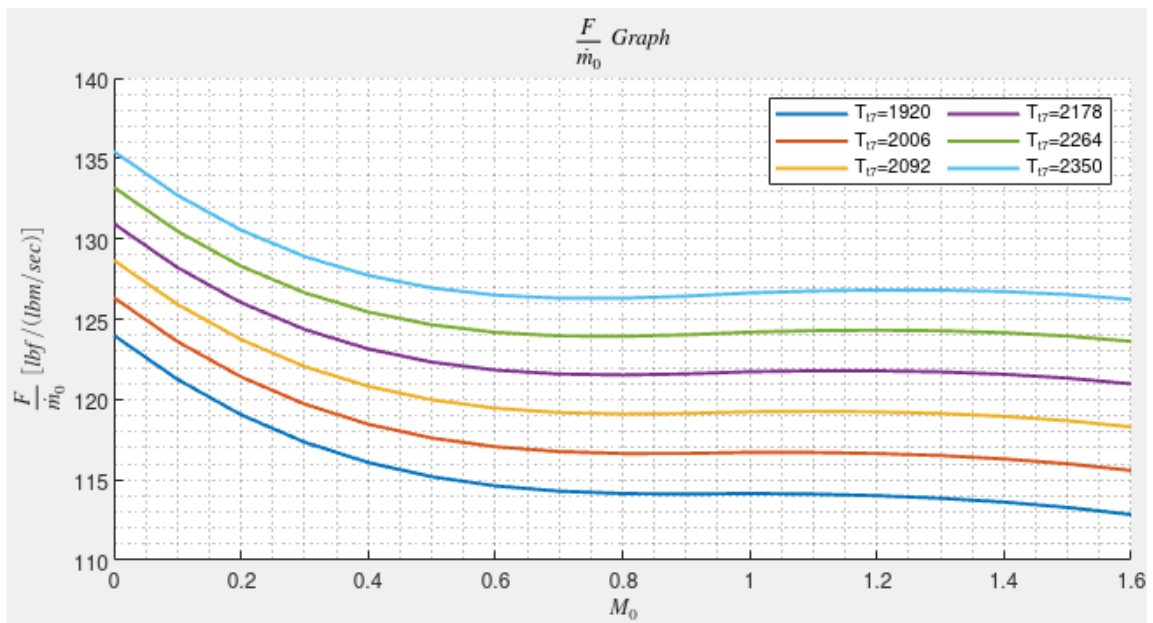


Figure 35 Specific Thrust

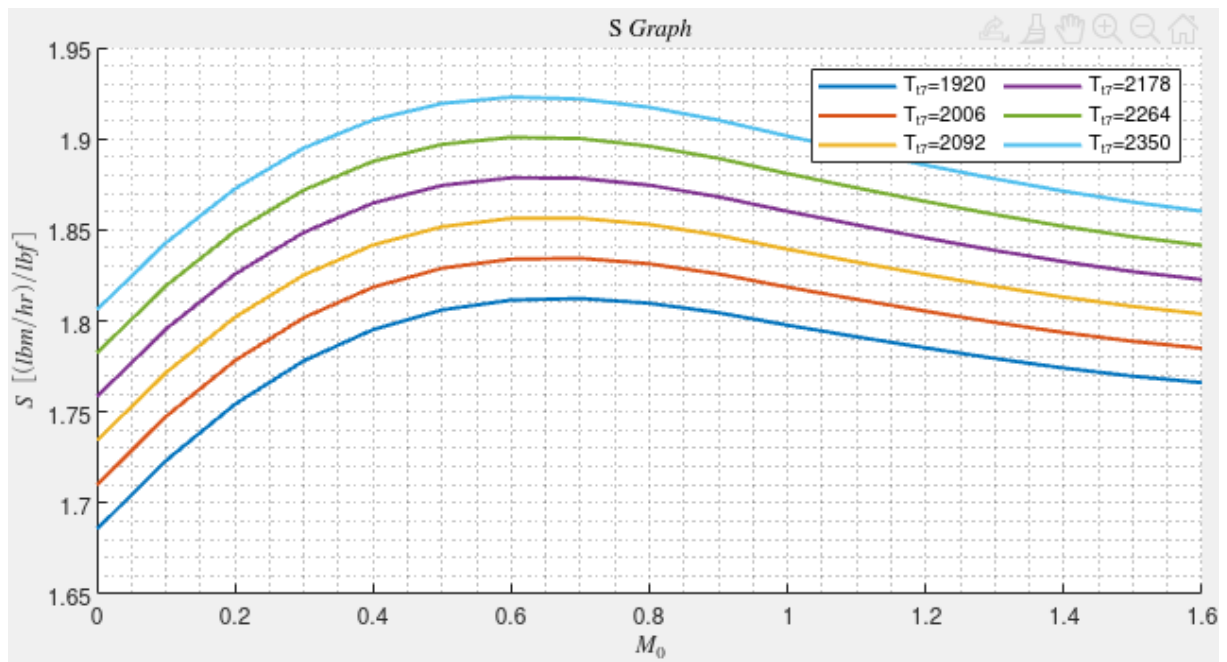
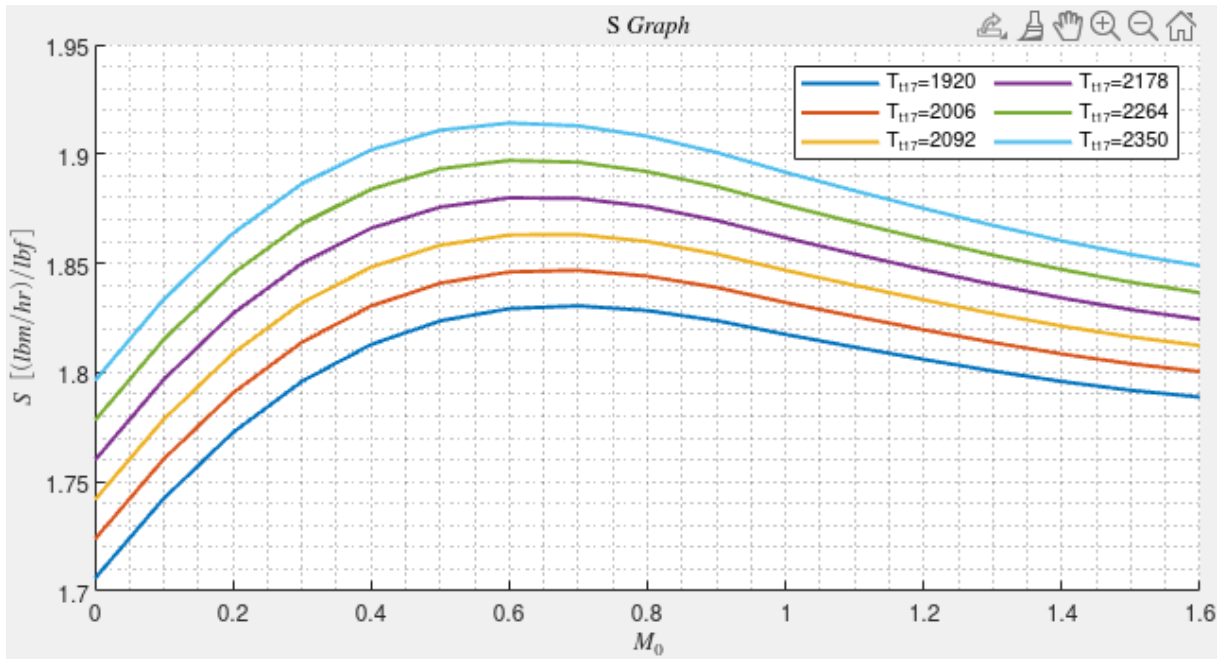
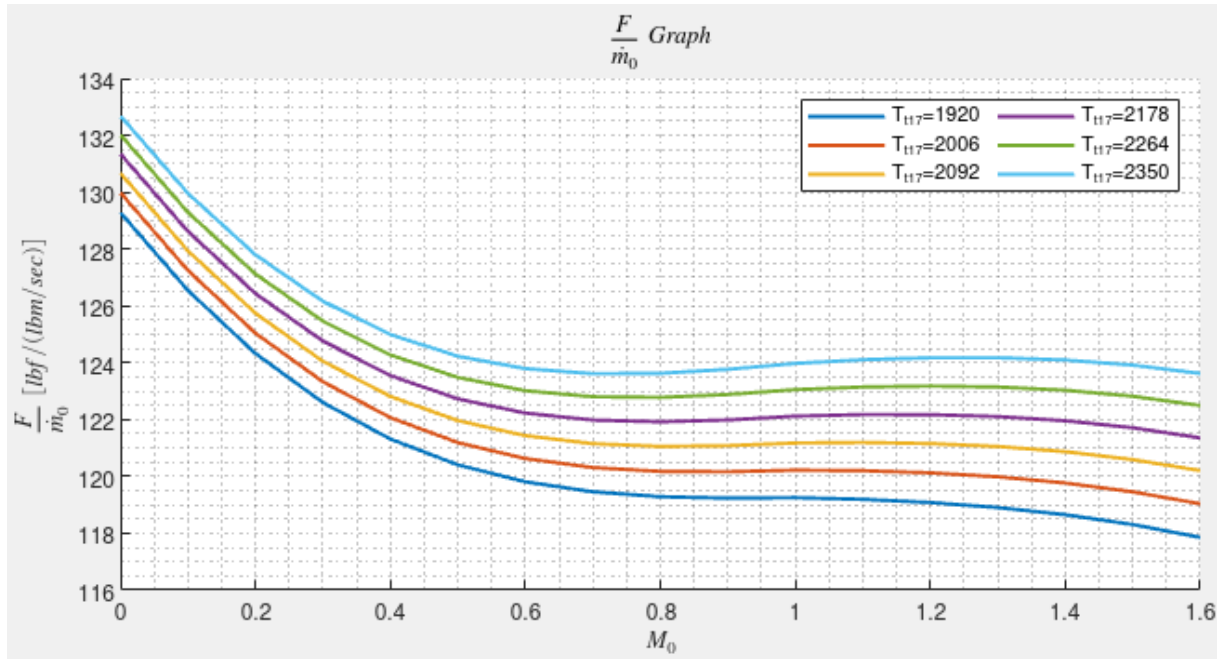


Figure 36 Specific Fuel Consumption

T_{t17} belongs to the air that is just about to enter the duct nozzle area. For an increased T_{t17} temperature, more fuel needs to be burned using the duct burner.

Design Constraints	
T_{t4} (K)	2000
T_{t4} Increment	86
Max T_{t4} (K)	2000
T_{t7} (K)	2200
T_{t7} Increment	86
Max T_{t7} (K)	2350
Min T_{t17} (K)	1920
T_{t17} Increment	86
Max T_{t17} (K)	2350

Figure 37 Design Choices



7. Conclusion

In this study Separated Stream Afterburner Turbofan Engine is researched. MATLAB based Graphical User Interface for parametric cycle analysis is designed and validated by comparing the Reference Book. Ranges of design parameters and component performances of the engine is determined by past-designed engine specifications. Parametric cycle analysis is done via designed User Interface. Effects of the design parameters on engine performance are investigated and the results are discussed with detailed graphs.

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APPENDIX

Source code:

```
classdef SSATFv2_5 < matlab.apps.AppBase
```

```
% Properties that correspond to app components
properties (Access = public)
    AfterburnerTurbofan          matlab.ui.Figure
    FlightConditionsPanel        matlab.ui.container.Panel
    GridLayout                   matlab.ui.container.GridLayout
    T_0KEditFieldLabel           matlab.ui.control.Label
    T_0KEditField                matlab.ui.control.NumericEditField
    MachNumberEditFieldLabel     matlab.ui.control.Label
    MachNumberMinEditField       matlab.ui.control.NumericEditField
    P_0P_9EditFieldLabel         matlab.ui.control.Label
    P_0P_9EditField              matlab.ui.control.NumericEditField
    P_0P_19EditFieldLabel        matlab.ui.control.Label
```

P_0P_19EditField	matlab.ui.control.NumericEditField
gamma_cEditFieldLabel	matlab.ui.control.Label
gamma_cEditField	matlab.ui.control.NumericEditField
c_pckJkgKLabel	matlab.ui.control.Label
c_pckJkgKEditField	matlab.ui.control.NumericEditField
MaxMachNumberEditFieldLabel	matlab.ui.control.Label
MaxMachNumberEditField	matlab.ui.control.NumericEditField
MachNoIncrementEditFieldLabel	matlab.ui.control.Label
MachNoIncrementEditField	matlab.ui.control.NumericEditField
TabGroup	matlab.ui.container.TabGroup
Table	matlab.ui.container.Tab
UITable	matlab.ui.control.Table
Fm_0dotTab	matlab.ui.container.Tab
FovermdotPlot	matlab.ui.control.UIAxes
STab	matlab.ui.container.Tab
SPlot	matlab.ui.control.UIAxes
FuelAirRatioTab	matlab.ui.container.Tab
FuelPlot	matlab.ui.control.UIAxes
Eta_cTab	matlab.ui.container.Tab
eta_cPlot	matlab.ui.control.UIAxes
Eta_tTab	matlab.ui.container.Tab
eta_tPlot	matlab.ui.control.UIAxes
PropulsiveEfficiencyTab	matlab.ui.container.Tab
GridLayout7	matlab.ui.container.GridLayout
eta_pPlot	matlab.ui.control.UIAxes
TermalEfficiencyTab	matlab.ui.container.Tab
eta_thermoPlot	matlab.ui.control.UIAxes
OverallEfficiencyTab	matlab.ui.container.Tab
eta_oPlot	matlab.ui.control.UIAxes
DesignConstraintsPanel	matlab.ui.container.Panel
GridLayout2	matlab.ui.container.GridLayout
MinT_t4KLabel	matlab.ui.control.Label
MinT_t4KEditField	matlab.ui.control.NumericEditField
MinT_t17KLabel	matlab.ui.control.Label
MinT_t17KEditField	matlab.ui.control.NumericEditField
MinT_t7KLabel	matlab.ui.control.Label
MinT_t7KEditField	matlab.ui.control.NumericEditField
T_t4IncrementEditFieldLabel	matlab.ui.control.Label
T_t4IncrementEditField	matlab.ui.control.NumericEditField
MaxT_t4KEditFieldLabel	matlab.ui.control.Label
MaxT_t4KEditField	matlab.ui.control.NumericEditField
T_t7IncrementEditFieldLabel	matlab.ui.control.Label
T_t7IncrementEditField	matlab.ui.control.NumericEditField
MaxT_t7KEditFieldLabel	matlab.ui.control.Label
MaxT_t7KEditField	matlab.ui.control.NumericEditField
T_t17IncrementEditFieldLabel	matlab.ui.control.Label
T_t17IncrementEditField	matlab.ui.control.NumericEditField
MaxT_t17KEditFieldLabel	matlab.ui.control.Label
MaxT_t17KEditField	matlab.ui.control.NumericEditField
ComponentPerformancesPanel	matlab.ui.container.Panel
GridLayout3	matlab.ui.container.GridLayout
pi_bEditFieldLabel	matlab.ui.control.Label
pi_bEditField	matlab.ui.control.NumericEditField
eta_ABEditFieldLabel	matlab.ui.control.Label
eta_ABEditField	matlab.ui.control.NumericEditField
eta_DBEditFieldLabel	matlab.ui.control.Label
eta_DBEditField	matlab.ui.control.NumericEditField
eta_mEditFieldLabel	matlab.ui.control.Label
eta_mEditField	matlab.ui.control.NumericEditField
eta_bEditFieldLabel	matlab.ui.control.Label
eta_bEditField	matlab.ui.control.NumericEditField
pi_DBEditFieldLabel	matlab.ui.control.Label
pi_DBEditField	matlab.ui.control.NumericEditField
pi_d_maxEditFieldLabel	matlab.ui.control.Label
pi_d_maxEditField	matlab.ui.control.NumericEditField

```

e_cEditFieldLabel      matlab.ui.control.Label
e_cEditField           matlab.ui.control.NumericEditField
e_tEditFieldLabel      matlab.ui.control.Label
e_tEditField           matlab.ui.control.NumericEditField
e_fEditFieldLabel      matlab.ui.control.Label
e_fEditField           matlab.ui.control.NumericEditField
pi_fnEditFieldLabel    matlab.ui.control.Label
pi_fnEditField         matlab.ui.control.NumericEditField
pi_nEditFieldLabel     matlab.ui.control.Label
pi_nEditField          matlab.ui.control.NumericEditField
pi_ABEditFieldLabel    matlab.ui.control.Label
pi_ABEditField         matlab.ui.control.NumericEditField
CaseSelectionPanel     matlab.ui.container.Panel
CaseDropDownLabel      matlab.ui.control.Label
CaseDropDown           matlab.ui.control.DropDown
DesignChoicesPanel     matlab.ui.container.Panel
GridLayout4            matlab.ui.container.GridLayout
ComputeButton          matlab.ui.control.Button
Minpi_cValueLabel      matlab.ui.control.Label
Minpi_cValueEditField  matlab.ui.control.NumericEditField
Minpi_fValueEditFieldLabel matlab.ui.control.Label
Minpi_fEditField       matlab.ui.control.NumericEditField
Maxpi_fValueEditFieldLabel matlab.ui.control.Label
Maxpi_fEditField       matlab.ui.control.NumericEditField
MinByPassRatioValueEditFieldLabel matlab.ui.control.Label
MinByPassRatioEditField matlab.ui.control.NumericEditField
MaxByPassRatioEditFieldLabel matlab.ui.control.Label
MaxByPassRatioEditField matlab.ui.control.NumericEditField
pi_fIncrementEditFieldLabel matlab.ui.control.Label
pi_fIncrementEditField matlab.ui.control.NumericEditField
ByPassRatioIncrementEditFieldLabel matlab.ui.control.Label
ByPassRatioIncrementEditField matlab.ui.control.NumericEditField
pi_cIncrementEditField_2Label matlab.ui.control.Label
pi_cIncrementEditField matlab.ui.control.NumericEditField
Maxpi_cValueEditField_2Label matlab.ui.control.Label
Maxpi_cValueEditField matlab.ui.control.NumericEditField
XAxisSwitchLabel      matlab.ui.control.Label
XAxisSwitch           matlab.ui.control.Switch
GasPropertiesPanel     matlab.ui.container.Panel
GridLayout5            matlab.ui.container.GridLayout
c_ptkJkgKEEditFieldLabel matlab.ui.control.Label
c_ptkJkgKEEditField   matlab.ui.control.NumericEditField
c_pABkJkgKLabel       matlab.ui.control.Label
c_pABkJkgKEEditField  matlab.ui.control.NumericEditField
c_pDBkJkgKEEditFieldLabel matlab.ui.control.Label
c_pDBkJkgKEEditField  matlab.ui.control.NumericEditField
gamma_ABEditFieldLabel matlab.ui.control.Label
gamma_ABEditField     matlab.ui.control.NumericEditField
gamma_tEditFieldLabel matlab.ui.control.Label
gamma_tEditField       matlab.ui.control.NumericEditField
gamma_DBEditFieldLabel matlab.ui.control.Label
gamma_DBEditField     matlab.ui.control.NumericEditField
FuelPropertiesPanel    matlab.ui.container.Panel
GridLayout6            matlab.ui.container.GridLayout
h_PRkJkgEditFieldLabel matlab.ui.control.Label
h_PRkJkgEditField     matlab.ui.control.NumericEditField
end

```

end

methods (Access = public)

```

function [F_mdots_0,f,f_AB,f_DB,S,eta_c,eta_t,eta_f,eta_P,eta_T,eta_0] = ...
    ATFWithSeperateExhausts(app,M_0,T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...
        gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

```



```

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0_P_9,P_0_P_19,...
T_t4,T_t7,T_t17,pi_c,pi_f,alpha)

% Calculations
g_c = 1;
R_c = ((gamma_c-1)./gamma_c).*c_pc; %
R_t = ((gamma_t-1)./gamma_t).*c_pt; %
R_AB = ((gamma_AB-1)./gamma_AB).*c_p_AB; %
R_DB = ((gamma_DB-1)./gamma_DB).*c_p_DB; %
a_0 = sqrt(gamma_c.*R_c.*g_c.*T_0*1000); % kJ den Joule'e gecis yapildi
V_0 = a_0.*M_0;
tau_r = 1 + (gamma_c-1)./2.*M_0.^2;
pi_r = tau_r.^(gamma_c./(gamma_c-1));
if M_0 <= 1
    eta_r = 1;
else
    eta_r = 1 - 0.075*(M_0-1).^(1.35);
end
pi_d = pi_d_max.*eta_r;
tau_lambda = (c_pt.*T_t4)./(c_pc.*T_0);
tau_lambda_AB = (c_p_AB.*T_t7)./(c_pc.*T_0);
tau_lambda_DB = (c_p_DB.*T_t17)./(c_pc.*T_0);
tau_c = pi_c.^((gamma_c-1)./(gamma_c.*e_c));
eta_c = (pi_c.^((gamma_c-1)./(gamma_c)-1)./(tau_c-1);
tau_f = pi_f.^((gamma_c-1)./(gamma_c.*e_f));
eta_f = (pi_f.^((gamma_c-1)./(gamma_c)-1)./(tau_f-1); %%%
f = (tau_lambda-tau_r.*tau_c)./(eta_b.*h_PR./(c_pc.*T_0)-tau_lambda);
tau_t = 1 - 1./(eta_m.*(1+f)).*(tau_r./tau_lambda).*(tau_c-1+alpha.*(tau_f-1));
pi_t = tau_t.^(gamma_t./((gamma_t-1).e_t));
eta_t = (1-tau_t)./(1-tau_t.^(1./e_t));
P_t9_P_9 = P_0_P_9.*pi_r.*pi_d.*pi_c.*pi_b.*pi_t.*pi_AB.*pi_n;
M_9 = sqrt(2./((gamma_AB-1) .* (P_t9_P_9.^((gamma_AB-1)./(gamma_AB) - 1)));
T_9_T_0 = T_t7./T_0./(P_t9_P_9.^((gamma_AB-1)./(gamma_AB)));
V_9_a_0 = M_9.*sqrt(gamma_AB.*R_AB./gamma_c./R_c.*T_9_T_0);
P_t19_P_19 = P_0_P_19.*pi_r.*pi_d.*pi_f.*pi_DB.*pi_fn;
M_19 = sqrt(2./((gamma_DB-1) .* (P_t19_P_19.^((gamma_DB-1)./(gamma_DB) - 1)));
T_19_T_0 = T_t17./T_0./(P_t19_P_19.^((gamma_DB-1)./(gamma_DB)));
V_19_a_0 = M_19.*sqrt(gamma_DB.*R_DB./gamma_c./R_c.*T_19_T_0);
f_AB = (1+f).*(tau_lambda_AB-tau_lambda.*tau_t)./(eta_AB.*h_PR./(c_pc.*T_0)-
tau_lambda_AB);
f_DB = (tau_lambda_DB-tau_r.*tau_f)./(eta_DB.*h_PR./(c_pc.*T_0)-tau_lambda_DB);
F__mdot_0 = 1./((1+alpha).*(a_0/g_c).*((1+f+f_AB).*(V_9_a_0-M_0+(1+f+f_AB).*...
R_AB./R_c.*(T_9_T_0)./(V_9_a_0).*(1-P_0_P_9)./(gamma_c))....
+alpha./((1+alpha).*(a_0/g_c).*((1+f_DB).*(V_19_a_0-M_0+(1+f_DB).*...
R_DB./R_c.*(T_19_T_0)./(V_19_a_0).*(1-P_0_P_19)./(gamma_c)));
S = 10^6*(f+f_AB+alpha*f_DB)./(1+alpha)./(F__mdot_0);
F__mdot_0 = F__mdot_0*0.224809/2.20462;
S = S/0.2248089*3600*2.20462*10^(-6);
eta_P = 2.*M_0.*((1+f+f_AB).*(V_9_a_0+alpha.*(1+f_DB).*(V_19_a_0-
(1+alpha).*(M_0)./...
((1+f+f_AB).*(V_9_a_0.^2+alpha.*(1+f_DB).*(V_19_a_0.^2-(1+alpha).*(M_0.^2);
eta_T = (a_0.^2).*((1+f+f_AB).*(V_9_a_0.^2+alpha.*(1+f_DB).*(V_19_a_0.^2-
(1+alpha).*(M_0.^2)./...
(2.*g_c.*(f+f_AB+f_DB).*h_PR.*1000);
eta_0 = eta_T.*eta_P;
if not(isreal(F__mdot_0)) && not(isreal(S))
    F__mdot_0 = NaN;
    f = NaN;
    f_AB = NaN;
    f_DB = NaN;
    S = NaN;
    eta_c = NaN;
    eta_t = NaN;
    eta_f = NaN;
    eta_P = NaN;

```

```

        eta_T = NaN;
        eta_O = NaN;
    end
end
end

% Callbacks that handle component events
methods (Access = private)

% Code that executes after component creation
function startupFcn(app)
    set(app.MachNoIncrementEditField,'enable','off');
    set(app.MaxMachNumberEditField,'enable','off');
    set(app.pi_cIncrementEditField,'enable','off');
    set(app.Maxpi_cValueEditField,'enable','off');
    set(app.pi_fIncrementEditField,'enable','off');
    set(app.Maxpi_fEditField,'enable','off');
    set(app.ByPassRatioIncrementEditField,'enable','off');
    set(app.MaxByPassRatioEditField,'enable','off');
    set(app.T_t4IncrementEditField,'enable','off');
    set(app.MaxT_t4KEditField,'enable','off');
    set(app.T_t7IncrementEditField,'enable','off');
    set(app.MaxT_t7KEditField,'enable','off');
    set(app.T_t17IncrementEditField,'enable','off');
    set(app.MaxT_t17KEditField,'enable','off');
    app.Minpi_fValueEditFieldLabel.Text = 'pi_f';
    app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
    app.Minpi_cValueLabel.Text = 'pi_c';
    app.MinT_t4KLabel.Text = 'T_t4 (K)';
    app.MinT_t7KLabel.Text = 'T_t7 (K)';
    app.MinT_t17KLabel.Text = 'T_t17 (K)';
    set(app.XAxisSwitch,'enable','off');
end

% Button pushed function: ComputeButton
function ComputeButtonPushed(app, event)
    solverval = app.CaseDropDown.Value;
    xaxisval = app.XAxisSwitch.Value;
    % Resetting Graphs
    cla(app.FovermdotPlot,'reset');
    cla(app.SPlot,'reset');
    cla(app.FuelPlot,'reset');
    cla(app.eta_cPlot,'reset');
    cla(app.eta_tPlot,'reset');
    cla(app.eta_pPlot,'reset');
    cla(app.eta_thermoPlot,'reset');
    cla(app.eta_oPlot,'reset');

    % Taking Edit Field Value
    M_0_min = app.MachNumberMinEditField.Value;
    M_0_increment = app.MachNoIncrementEditField.Value;
    M_0_max = app.MaxMachNumberEditField.Value;
    T_0 = app.T_0KEditField.Value;
    gamma_c = app.gamma_cEditField.Value;
    c_pc = app.c_pckJkgKEditField.Value;
    gamma_t = app.gamma_tEditField.Value;
    c_pt = app.c_ptkJkgKEditField.Value;
    h_PR = app.h_PRkJkgKEditField.Value;
    gamma_AB = app.gamma_ABEditField.Value;
    c_p_AB = app.c_pABkJkgKEditField.Value;

```



```

gamma_DB = app.gamma_DBEditField.Value;
c_p_DB = app.c_pDBkJkgKEditField.Value;
pi_d_max = app.pi_d_maxEditField.Value;
pi_b = app.pi_bEditField.Value;
pi_AB = app.pi_ABEditField.Value;
pi_DB = app.pi_DBEditField.Value;
pi_n = app.pi_nEditField.Value;
pi_fn = app.pi_fnEditField.Value;
e_c = app.e_cEditField.Value;
e_f = app.e_fEditField.Value;
e_t = app.e_tEditField.Value;
eta_b = app.eta_bEditField.Value;
eta_AB = app.eta_ABEditField.Value;
eta_DB = app.eta_DBEditField.Value;
eta_m = app.eta_mEditField.Value;
P_0__P_9 = app.P_0P_9EditField.Value;
P_0__P_19 = app.P_0P_19EditField.Value;

T_t4_min = app.MinT_t4KEditField.Value;
T_t4_increment = app.T_t4IncrementEditField.Value;
T_t4_max = app.MaxT_t4KEditField.Value;

T_t7_min = app.MinT_t7KEditField.Value;
T_t7_increment = app.T_t7IncrementEditField.Value;
T_t7_max = app.MaxT_t7KEditField.Value;

T_t17_min = app.MinT_t17KEditField.Value;
T_t17_increment = app.T_t17IncrementEditField.Value;
T_t17_max = app.MaxT_t17KEditField.Value;

alpha_min = app.MinByPassRatioEditField.Value;
alpha_max = app.MaxByPassRatioEditField.Value;
alpha_increment = app.ByPassRatioIncrementEditField.Value;
pi_f_min = app.Minpi_fEditField.Value;
pi_f_increment = app.pi_fIncrementEditField.Value;
pi_f_max = app.Maxpi_fEditField.Value;
pi_c_min = app.Minpi_cValueEditField.Value;
pi_c_increment = app.pi_cIncrementEditField.Value;
pi_c_max = app.Maxpi_cValueEditField.Value;

switch solverval
case 'Single Condition'
    M_0 = M_0_min;
    pi_c = pi_c_min;
    pi_f = pi_f_min;
    alpha = alpha_min;
    T_t4 = T_t4_min;
    T_t7 = T_t7_min;
    T_t17 = T_t17_min;

    [F__mdot_0,f,f_AB,f_DB,S,eta_c,eta_t,eta_f,eta_P,eta_T,eta_0] = ...

ATFwithSeperateExhausts(app,M_0,T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...
    gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...
    pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0__P_9,P_0__P_19,...
    T_t4,T_t7,T_t17,pi_c,pi_f,alpha);

    % Creating Table
    t =
table(M_0,pi_c,pi_f,alpha,T_t4,T_t7,T_t17,F__mdot_0,f,f_AB,f_DB,S,eta_c,eta_t,eta_f,eta_P,e
ta_T,eta_0);

    app.UITable.Data = t;

```

```

vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c',
',eta_t','eta_f','eta_P','eta_T','eta_0'};
app.UITable.ColumnName = vars;

case 'Varying Bypass and pi_f'
M_0 = M_0_min;
pi_c = pi_c_min;
T_t4 = T_t4_min;
T_t7 = T_t7_min;
T_t17 = T_t17_min;
pi_f = pi_f_min:pi_f_increment:pi_f_max;
alpha = alpha_min:alpha_increment:alpha_max;
tablo = zeros(length(pi_f)*length(alpha),18);

for i = 1:length(pi_f)
for j=1:length(alpha)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_0(i,j))] = ...

ATFwithSeperateExhausts(app,M_0,T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0__P_9,P_0__P_19,...
T_t4,T_t7,T_t17,pi_c,pi_f(i),alpha(j));

tablo(length(alpha)*(i-1)+j,:) = ...
[M_0,pi_c,pi_f(i),alpha(j),T_t4,T_t7,T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_0(i,j)]];

end
end
% Creating Table
t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18)));
app.UITable.Data = t;
vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c',
',eta_t','eta_f','eta_P','eta_T','eta_0'};
app.UITable.ColumnName = vars;

% Handling Plots
switch xaxisval
case 'pi_f'

plot(app.FovermdotPlot,pi_f,F__mdot_0(:,:),"LineWidth",1.4)
title(app.FovermdotPlot,"$\frac{F}{\dot{m}}_0$
Graph$","Interpreter","Latex")
grid(app.FovermdotPlot,'minor')
xlabel(app.FovermdotPlot,"$\pi_f$","Interpreter","Latex");
ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}}_0\backslash$
[lbf/(lbm/sec)]$","Interpreter","Latex");

legend(app.FovermdotPlot,'\alpha='+string(alpha),'Location','southeast','NumColumns',2);

plot(app.SPlot,pi_f,S(:,:),"LineWidth",1.4)
title(app.SPlot,"S Graph$","Interpreter","Latex")

```

```

        grid(app.SPlot, 'minor')
        xlabel(app.SPlot, "$\pi_f$", "Interpreter", "Latex");
        ylabel(app.SPlot, "$S \ [(lbm/hr)/lbf]$", "Interpreter", "Latex");

legend(app.SPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns', 2);

plot(app.FuelPlot, pi_f, f(:,1), pi_f, f_AB(:,1), pi_f, f_DB(:,1), "LineWidth", 1.4);
title(app.FuelPlot, "Fuel $Graph$", "Interpreter", "Latex")
grid(app.FuelPlot, 'minor')
xlabel(app.FuelPlot, "$\pi_f$", "Interpreter", "Latex");
ylabel(app.FuelPlot, "$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
leg = ['f for all \alpha', 'f_AB for
\alpha='+string(alpha), 'f_DB for all \alpha'];
legend(app.FuelPlot, leg, 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_cPlot, pi_f, eta_c(:,1), "LineWidth", 1.4)
title(app.eta_cPlot, "$\eta_c$ $Graph$", "Interpreter", "Latex")
grid(app.eta_cPlot, 'minor')
xlabel(app.eta_cPlot, "$\pi_f$", "Interpreter", "Latex");
ylabel(app.eta_cPlot, "$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_tPlot, pi_f, eta_t(:,1), "LineWidth", 1.4)
title(app.eta_tPlot, "$\eta_t$", "Interpreter", "Latex")
grid(app.eta_tPlot, 'minor')
xlabel(app.eta_tPlot, "$\pi_f$", "Interpreter", "Latex");
ylabel(app.eta_tPlot, "$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_pPlot, pi_f, eta_P(:,1), "LineWidth", 1.4);
title(app.eta_pPlot, " Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_pPlot, 'minor')
xlabel(app.eta_pPlot, "$ \pi_f $", "Interpreter", "Latex");
ylabel(app.eta_pPlot, "$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_thermoPlot, pi_f, eta_T(:,1), "LineWidth", 1.4);
title(app.eta_thermoPlot, "Thermal Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_thermoPlot, 'minor')
xlabel(app.eta_thermoPlot, "$ \pi_f $", "Interpreter", "Latex");
ylabel(app.eta_thermoPlot, "$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_oPlot, pi_f, eta_O(:,1), "LineWidth", 1.4);
title(app.eta_oPlot, "Efficiency $Graph$", "Interpreter", "Latex")
grid(app.eta_oPlot, 'minor')
xlabel(app.eta_oPlot, "$ \pi_f $", "Interpreter", "Latex");
ylabel(app.eta_oPlot, "$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns', 2);

case 'alpha'
plot(app.FovermdotPlot, alpha, F__mdot_0(:,1), "LineWidth", 1.4)
title(app.FovermdotPlot, "$\frac{F}{\dot{m}}_0$
Graph$", "Interpreter", "Latex")
grid(app.FovermdotPlot, 'minor')
xlabel(app.FovermdotPlot, "$\alpha$", "Interpreter", "Latex");

```

```

        ylabel(app.FovermdotPlot, "$\frac{F}{\dot{m}_0} \backslash$
[1bf/(lbm/sec)]$", "Interpreter", "Latex");

legend(app.FovermdotPlot, '\pi_f='+string(pi_f), 'Location', 'southeast', 'NumColumns', 2);

        plot(app.SPlot, alpha, S(:, :), "LineWidth", 1.4)
        title(app.SPlot, "S $Graph$", "Interpreter", "Latex")
        grid(app.SPlot, 'minor')
        xlabel(app.SPlot, "$\alpha$", "Interpreter", "Latex");
        ylabel(app.SPlot, "$S \backslash [(lbm/hr)/1bf]$", "Interpreter", "Latex");

legend(app.SPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

plot(app.FuelPlot, alpha, f(:, :), alpha, f_AB(:, :), alpha, f_DB(:, :), "LineWidth", 1.4);
title(app.FuelPlot, "Fuel $Graph$", "Interpreter", "Latex")
grid(app.FuelPlot, 'minor')
xlabel(app.FuelPlot, "$\alpha$", "Interpreter", "Latex");
ylabel(app.FuelPlot, "$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
leg = ['f for \pi_f='+string(pi_f), 'f_AB for
\pi_f='+string(pi_f), 'f_DB for \pi_f='+string(pi_f)];
legend(app.FuelPlot, leg, 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_cPlot, alpha, eta_c(:, :), "LineWidth", 1.4)
        title(app.eta_cPlot, "$\eta_c$ $Graph$", "Interpreter", "Latex")
        grid(app.eta_cPlot, 'minor')
        xlabel(app.eta_cPlot, "$\alpha$", "Interpreter", "Latex");
        ylabel(app.eta_cPlot, "$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

        plot(app.eta_tPlot, alpha, eta_t(:, :), "LineWidth", 1.4)
        title(app.eta_tPlot, "$\eta_t$", "Interpreter", "Latex")
        grid(app.eta_tPlot, 'minor')
        xlabel(app.eta_tPlot, "$\alpha$", "Interpreter", "Latex");
        ylabel(app.eta_tPlot, "$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

        plot(app.eta_pPlot, alpha, eta_P(:, :), "LineWidth", 1.4);
        title(app.eta_pPlot, " Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_pPlot, 'minor')
        xlabel(app.eta_pPlot, "$ \alpha $", "Interpreter", "Latex");
        ylabel(app.eta_pPlot, "$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_thermoPlot, alpha, eta_T(:, :), "LineWidth", 1.4);
        title(app.eta_thermoPlot, "Thermal Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_thermoPlot, 'minor')
        xlabel(app.eta_thermoPlot, "$ \alpha $", "Interpreter", "Latex");
        ylabel(app.eta_thermoPlot, "$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_oPlot, alpha, eta_O(:, :), "LineWidth", 1.4);
        title(app.eta_oPlot, "Efficiency $Graph$", "Interpreter", "Latex")
        grid(app.eta_oPlot, 'minor')
        xlabel(app.eta_oPlot, "$ \alpha $", "Interpreter", "Latex");
        ylabel(app.eta_oPlot, "$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

```

```

end
case 'Varying pi_f and pi_c'
    alpha = alpha_min;
    M_0 = M_0_min;
    T_t4 = T_t4_min;
    T_t7 = T_t7_min;
    T_t17 = T_t17_min;
    pi_f = pi_f_min:pi_f_increment:pi_f_max;
    pi_c = pi_c_min:pi_c_increment:pi_c_max;
    tablo = zeros(length(pi_f)*length(pi_c),18);

    for i = 1:length(pi_f)
        for j=1:length(pi_c)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_O(i,j)] = ...

ATFwithSeperateExhausts(app,M_0,T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0__P_9,P_0__P_19,...
            T_t4,T_t7,T_t17,pi_c(j),pi_f(i),alpha);

            tablo(length(pi_c)*(i-1)+j,:) =...
                [M_0,pi_c(j),pi_f(i),alpha,T_t4,T_t7,T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_O(i,j)];

        end
    end

    % Creating Table
    t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
        tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18));
    app.UITable.Data = t;
    vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c
','eta_t','eta_f','eta_P','eta_T','eta_O'};
    app.UITable.ColumnName = vars;
    switch xaxisval
        case 'pi_c'
            % Handling Plots

            plot(app.FovermdotPlot,pi_c,F__mdot_0(:,:),"LineWidth",1.4)
            title(app.FovermdotPlot," $\frac{F}{\dot{m}_0}$ \
Graph$", "Interpreter", "Latex")
            grid(app.FovermdotPlot,'minor')
            xlabel(app.FovermdotPlot," $\pi_c$ ", "Interpreter", "Latex");
            ylabel(app.FovermdotPlot," $\frac{F}{\dot{m}_0}$ \
[lbf/(lbm/sec)]$", "Interpreter", "Latex");

            legend(app.FovermdotPlot,'\pi_f'+string(pi_f),'Location','southeast','NumColumns',2);

            plot(app.SPlot,pi_c,S(:,:),"LineWidth",1.4)
            title(app.SPlot,"S $Graph$", "Interpreter", "Latex")
            grid(app.SPlot,'minor')
            xlabel(app.SPlot," $\pi_c$ ", "Interpreter", "Latex");
            ylabel(app.SPlot," $S \left[ \frac{lbm}{hr} \right] / lbf$ ", "Interpreter", "Latex");

```

```

legend(app.SPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

p =
plot(app.FuelPlot, pi_c, f(1,:), pi_c, f_AB(:, :), pi_c, f_DB(:, :), "LineWidth", 1.4);
title(app.FuelPlot, "Fuel $Graph$", "Interpreter", "Latex")
grid(app.FuelPlot, 'minor')
xlabel(app.FuelPlot, "$\pi_c$", "Interpreter", "Latex");
ylabel(app.FuelPlot, "$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
leg = ['f for all \pi_f', 'f_{AB} for
\pi_f='+string(pi_f), 'f_{DB} for \pi_f='+string(pi_f)];
legend(app.FuelPlot, leg, 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_cPlot, pi_c, eta_c(:, :), "LineWidth", 1.4)
title(app.eta_cPlot, "$\eta_c$ $Graph$", "Interpreter", "Latex")
grid(app.eta_cPlot, 'minor')
xlabel(app.eta_cPlot, "$\pi_c$", "Interpreter", "Latex");
ylabel(app.eta_cPlot, "$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_tPlot, pi_c, eta_t(:, :), "LineWidth", 1.4)
title(app.eta_tPlot, "$\eta_t$", "Interpreter", "Latex")
grid(app.eta_tPlot, 'minor')
xlabel(app.eta_tPlot, "$\pi_c$", "Interpreter", "Latex");
ylabel(app.eta_tPlot, "$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_pPlot, pi_c, eta_P(:, :), "LineWidth", 1.4);
title(app.eta_pPlot, " Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_pPlot, 'minor')
xlabel(app.eta_pPlot, "$ \pi_c $", "Interpreter", "Latex");
ylabel(app.eta_pPlot, "$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_thermoPlot, pi_c, eta_T(:, :), "LineWidth", 1.4);
title(app.eta_thermoPlot, "Thermal Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_thermoPlot, 'minor')
xlabel(app.eta_thermoPlot, "$ \pi_c $", "Interpreter", "Latex");
ylabel(app.eta_thermoPlot, "$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_oPlot, pi_c, eta_0(:, :), "LineWidth", 1.4);
title(app.eta_oPlot, "Efficiency $Graph$", "Interpreter", "Latex")
grid(app.eta_oPlot, 'minor')
xlabel(app.eta_oPlot, "$ \pi_c $", "Interpreter", "Latex");
ylabel(app.eta_oPlot, "$\eta_0$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

case 'pi_f'
plot(app.FovermdotPlot, pi_f, F__mdot_0(:, :), "LineWidth", 1.4)
title(app.FovermdotPlot, "$\frac{F}{\dot{m}_0}$
Graph$", "Interpreter", "Latex")
grid(app.FovermdotPlot, 'minor')
xlabel(app.FovermdotPlot, "$\pi_f$", "Interpreter", "Latex");
ylabel(app.FovermdotPlot, "$\frac{F}{\dot{m}_0}$\
[lbf/(lbm/sec)]$", "Interpreter", "Latex");

legend(app.FovermdotPlot, '\pi_c='+string(pi_c), 'Location', 'southeast', 'NumColumns', 2);

```

```

        plot(app.SPlot,pi_f,S(:,:),"LineWidth",1.4)
        title(app.SPlot,"S $Graph$", "Interpreter", "Latex")
        grid(app.SPlot,'minor')
        xlabel(app.SPlot,"$\pi_f$", "Interpreter", "Latex");
        ylabel(app.SPlot,"$S \ [(lbm/hr)/lbf]$", "Interpreter", "Latex");

legend(app.SPlot, '\pi_c='+string(pi_c), 'Location', 'northeast', 'NumColumns', 2);

plot(app.FuelPlot,pi_f,f(:,1),pi_f,f_AB(:,:),pi_f,f_DB(:,1),"LineWidth",1.4);
        title(app.FuelPlot,"Fuel $Graph$", "Interpreter", "Latex")
        grid(app.FuelPlot,'minor')
        xlabel(app.FuelPlot,"$\pi_f$", "Interpreter", "Latex");
        ylabel(app.FuelPlot,"$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
        leg = ['f for \pi_c='+string(pi_c), 'f_AB for
\pi_c='+string(pi_c), 'f_DB for \pi_c='+string(pi_c)];
        legend(app.FuelPlot,leg, 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_cPlot,pi_f,eta_c(:,:),"LineWidth",1.4)
        title(app.eta_cPlot,"$\eta_c$ $Graph$", "Interpreter", "Latex")
        grid(app.eta_cPlot,'minor')
        xlabel(app.eta_cPlot,"$\pi_f$", "Interpreter", "Latex");
        ylabel(app.eta_cPlot,"$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\pi_c='+string(pi_c), 'Location', 'northeast', 'NumColumns', 2);

        plot(app.eta_tPlot,pi_f,eta_t(:,:),"LineWidth",1.4)
        title(app.eta_tPlot,"$\eta_t$", "Interpreter", "Latex")
        grid(app.eta_tPlot,'minor')
        xlabel(app.eta_tPlot,"$\pi_f$", "Interpreter", "Latex");
        ylabel(app.eta_tPlot,"$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\pi_c='+string(pi_c), 'Location', 'northeast', 'NumColumns', 2);

        plot(app.eta_pPlot,pi_f,eta_P(:,:),"LineWidth",1.4);
        title(app.eta_pPlot," Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_pPlot,'minor')
        xlabel(app.eta_pPlot,"$ \pi_f $", "Interpreter", "Latex");
        ylabel(app.eta_pPlot,"$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\pi_c='+string(pi_c), 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_thermoPlot,pi_f,eta_T(:,:),"LineWidth",1.4);
        title(app.eta_thermoPlot,"Thermal Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_thermoPlot,'minor')
        xlabel(app.eta_thermoPlot,"$ \pi_f $", "Interpreter", "Latex");
        ylabel(app.eta_thermoPlot,"$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\pi_c='+string(pi_c), 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_oPlot,pi_f,eta_O(:,:),"LineWidth",1.4);
        title(app.eta_oPlot,"Efficiency $Graph$", "Interpreter", "Latex")
        grid(app.eta_oPlot,'minor')
        xlabel(app.eta_oPlot,"$ \pi_f $", "Interpreter", "Latex");
        ylabel(app.eta_oPlot,"$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\pi_c='+string(pi_c), 'Location', 'northwest', 'NumColumns', 2);
    end

    case 'Varying Bypass and pi_c'
        alpha = alpha_min:alpha_increment:alpha_max;
        T_t4 = T_t4_min;

```

```

T_t7 = T_t7_min;
T_t17 = T_t17_min;
M_0 = M_0_min;
pi_f = pi_f_min;
pi_c = pi_c_min:pi_c_increment:pi_c_max;
tablo = zeros(length(alpha)*length(pi_c),18);

for i = 1:length(pi_c)
    for j=1:length(alpha)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_O(i,j)] = ...

ATFwithSeperateExhausts(app,M_0,T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0_P_9,P_0_P_19,...
        T_t4,T_t7,T_t17,pi_c(i),pi_f,alpha(j));

        tablo(length(alpha)*(i-1)+j,:) = ...
            [M_0,pi_c(i),pi_f,alpha(j),T_t4,T_t7,T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_O(i,j)];

        end
    end

% Creating Table
t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
        tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18));
app.UITable.Data = t;
vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c
','eta_t','eta_f','eta_P','eta_T','eta_O'};
app.UITable.ColumnName = vars;

switch xaxisval

    case 'pi_c'
        % Handling Plots

        plot(app.FovermdotPlot,pi_c,F__mdot_0(:,:),"LineWidth",1.4)
        title(app.FovermdotPlot," $\frac{F}{\dot{m}_0}$ \
Graph$","Interpreter","Latex")
        grid(app.FovermdotPlot,'minor')
        xlabel(app.FovermdotPlot," $\pi_c$ ","Interpreter","Latex");
        ylabel(app.FovermdotPlot," $\frac{F}{\dot{m}_0}$ \
[lbf/(lbm/sec)]$","Interpreter","Latex");

        legend(app.FovermdotPlot,'\alpha='+string(alpha),'Location','southeast','NumColumns',2);

        plot(app.SPlot,pi_c,S(:,:),"LineWidth",1.4)
        title(app.SPlot,"S Graph$","Interpreter","Latex")
        grid(app.SPlot,'minor')
        xlabel(app.SPlot," $\pi_c$ ","Interpreter","Latex");
        ylabel(app.SPlot," $S \ [(\text{lbm/hr})/\text{lbf}]$ ","Interpreter","Latex");

```



```

legend(app.SPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns', 2);

p =
plot(app.FuelPlot, pi_c, f(:, :), pi_c, f_AB(:, :), pi_c, f_DB(:, :), "LineWidth", 1.4);
title(app.FuelPlot, "Fuel $Graph$", "Interpreter", "Latex")
grid(app.FuelPlot, 'minor')
xlabel(app.FuelPlot, "$\pi_c$", "Interpreter", "Latex");
ylabel(app.FuelPlot, "$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
leg = ['f for \alpha='+string(alpha), 'f_{AB} for
\alpha='+string(alpha), 'f_{DB} for \alpha='+string(alpha)];
legend(app.FuelPlot, leg, 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_cPlot, pi_c, eta_c(:, :), "LineWidth", 1.4)
title(app.eta_cPlot, "$\eta_c$ $Graph$", "Interpreter", "Latex")
grid(app.eta_cPlot, 'minor')
xlabel(app.eta_cPlot, "$\pi_c$", "Interpreter", "Latex");
ylabel(app.eta_cPlot, "$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_tPlot, pi_c, eta_t(:, :), "LineWidth", 1.4)
title(app.eta_tPlot, "$\eta_t$", "Interpreter", "Latex")
grid(app.eta_tPlot, 'minor')
xlabel(app.eta_tPlot, "$\pi_c$", "Interpreter", "Latex");
ylabel(app.eta_tPlot, "$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_pPlot, pi_c, eta_P(:, :), "LineWidth", 1.4);
title(app.eta_pPlot, " Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_pPlot, 'minor')
xlabel(app.eta_pPlot, "$ \pi_c $", "Interpreter", "Latex");
ylabel(app.eta_pPlot, "$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_thermoPlot, pi_c, eta_T(:, :), "LineWidth", 1.4);
title(app.eta_thermoPlot, "Thermal Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_thermoPlot, 'minor')
xlabel(app.eta_thermoPlot, "$ \pi_c $", "Interpreter", "Latex");
ylabel(app.eta_thermoPlot, "$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_oPlot, pi_c, eta_0(:, :), "LineWidth", 1.4);
title(app.eta_oPlot, "Efficiency $Graph$", "Interpreter", "Latex")
grid(app.eta_oPlot, 'minor')
xlabel(app.eta_oPlot, "$ \pi_c $", "Interpreter", "Latex");
ylabel(app.eta_oPlot, "$\eta_0$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns', 2);

case 'alpha'
plot(app.FovermdotPlot, alpha, F__mdot_0(:, :), "LineWidth", 1.4)
title(app.FovermdotPlot, "$\frac{F}{\dot{m}_0}$
Graph$", "Interpreter", "Latex")
grid(app.FovermdotPlot, 'minor')
xlabel(app.FovermdotPlot, "$\alpha$", "Interpreter", "Latex");
ylabel(app.FovermdotPlot, "$\frac{F}{\dot{m}_0}$\
[lbf/(lbm/sec)]$", "Interpreter", "Latex");

legend(app.FovermdotPlot, '\pi_c='+string(pi_c), 'Location', 'southeast', 'NumColumns', 2);

```

```

        plot(app.SPlot,alpha,S(:,:),"LineWidth",1.4)
        title(app.SPlot,"S $Graph$", "Interpreter", "Latex")
        grid(app.SPlot,'minor')
        xlabel(app.SPlot,"$\alpha$", "Interpreter", "Latex");
        ylabel(app.SPlot,"$S\ [(lbm/hr)/lb]$ ", "Interpreter", "Latex");

legend(app.SPlot, '\pi_c='+string(pi_c), 'Location', 'northeast', 'NumColumns', 2);

plot(app.FuelPlot,alpha,f(:,:),alpha,f_AB(:,:),alpha,f_DB(:,:),"LineWidth",1.4);
title(app.FuelPlot,"Fuel $Graph$", "Interpreter", "Latex")
grid(app.FuelPlot,'minor')
xlabel(app.FuelPlot,"$\alpha$", "Interpreter", "Latex");
ylabel(app.FuelPlot,"$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
leg = ['f for \pi_c='+string(pi_c), 'f_AB for
\pi_c='+string(pi_c), 'f_DB for \pi_c='+string(pi_c)];
legend(app.FuelPlot,leg, 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_cPlot,alpha,eta_c(:,:),"LineWidth",1.4)
title(app.eta_cPlot,"$\eta_c$ $Graph$", "Interpreter", "Latex")
grid(app.eta_cPlot,'minor')
xlabel(app.eta_cPlot,"$\alpha$", "Interpreter", "Latex");
ylabel(app.eta_cPlot,"$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\pi_c='+string(pi_c), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_tPlot,alpha,eta_t(:,:),"LineWidth",1.4)
title(app.eta_tPlot,"$\eta_t$", "Interpreter", "Latex")
grid(app.eta_tPlot,'minor')
xlabel(app.eta_tPlot,"$\alpha$", "Interpreter", "Latex");
ylabel(app.eta_tPlot,"$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\pi_c='+string(pi_c), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_pPlot,alpha,eta_P(:,:),"LineWidth",1.4);
title(app.eta_pPlot," Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_pPlot,'minor')
xlabel(app.eta_pPlot,"$ \alpha $", "Interpreter", "Latex");
ylabel(app.eta_pPlot,"$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\pi_c='+string(pi_c), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_thermoPlot,alpha,eta_T(:,:),"LineWidth",1.4);
title(app.eta_thermoPlot,"Thermal Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_thermoPlot,'minor')
xlabel(app.eta_thermoPlot,"$ \alpha $", "Interpreter", "Latex");
ylabel(app.eta_thermoPlot,"$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\pi_c='+string(pi_c), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_oPlot,alpha,eta_O(:,:),"LineWidth",1.4);
title(app.eta_oPlot,"Efficiency $Graph$", "Interpreter", "Latex")
grid(app.eta_oPlot,'minor')
xlabel(app.eta_oPlot,"$ \alpha $", "Interpreter", "Latex");
ylabel(app.eta_oPlot,"$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\pi_c='+string(pi_c), 'Location', 'northwest', 'NumColumns', 2);
end

case 'Varying Mach No and pi_c'
    alpha = alpha_min;
    T_t4 = T_t4_min;

```

```

T_t7 = T_t7_min;
T_t17 = T_t17_min;
M_0 = M_0_min:M_0_increment:M_0_max;
pi_f = pi_f_min;
pi_c = pi_c_min:pi_c_increment:pi_c_max;
tablo = zeros(length(M_0)*length(pi_c),18);

for i = 1:length(M_0)
    for j=1:length(pi_c)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_O(i,j)] = ...

ATFwithSeperateExhausts(app,M_0(i),T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0_P_9,P_0_P_19,...
T_t4,T_t7,T_t17,pi_c(j),pi_f,alpha);

        tablo(length(pi_c)*(i-1)+j,:) = ...
            [M_0(i),pi_c(j),pi_f,alpha,T_t4,T_t7,T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_O(i,j)];

        end
    end

    % Creating Table
    t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
        tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18)));
    app.UITable.Data = t;
    vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c
','eta_t','eta_f','eta_P','eta_T','eta_O'};
    app.UITable.ColumnName = vars;

    % Handling Plots
    plot(app.FovermdotPlot,M_0,F__mdot_0(:,:),"LineWidth",1.4)
    title(app.FovermdotPlot,"$\frac{F}{\dot{m}}_0 \backslash$
Graph$","Interpreter","Latex")
    grid(app.FovermdotPlot,'minor')
    xlabel(app.FovermdotPlot,"$M_0$","Interpreter","Latex");
    ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}}_0 \backslash$
[lbf/(lbfm/sec)]$","Interpreter","Latex");

    legend(app.FovermdotPlot,'\pi_c='+string(pi_c),'Location','southeast','NumColumns',2);

    plot(app.SPlot,M_0,S(:,:),"LineWidth",1.4)
    title(app.SPlot,"S $Graph$","Interpreter","Latex")
    grid(app.SPlot,'minor')
    xlabel(app.SPlot,"$M_0$","Interpreter","Latex");
    ylabel(app.SPlot,"$S \ [(lbfm/hr)/lbf]$", "Interpreter","Latex");

    legend(app.SPlot,'\pi_c='+string(pi_c),'Location','northeast','NumColumns',2);

    plot(app.FuelPlot,M_0,f(:,:),M_0,f_AB(:,:),M_0,f_DB(:,:),"LineWidth",1.4);
    title(app.FuelPlot,"Fuel $Graph$","Interpreter","Latex")
    grid(app.FuelPlot,'minor')

```

```

xlabel(app.FuelPlot,"$M_0$","Interpreter","Latex");
ylabel(app.FuelPlot,"$f,f_{AB},f_{DB}$","Interpreter","Latex");
leg = ['f for \pi_c='+string(pi_c), 'f_{AB} for
\pi_c='+string(pi_c), 'f_{DB} for \pi_c='+string(pi_c)];
legend(app.FuelPlot,leg,'Location','northwest','NumColumns',2);

plot(app.eta_cPlot,M_0,eta_c(:,:),"LineWidth",1.4)
title(app.eta_cPlot,"$\eta_c$ Graph$","Interpreter","Latex")
grid(app.eta_cPlot,'minor')
xlabel(app.eta_cPlot,"$M_0$","Interpreter","Latex");
ylabel(app.eta_cPlot,"$\eta_c$","Interpreter","Latex");

legend(app.eta_cPlot,'\pi_c='+string(pi_c),'Location','northeast','NumColumns',2);

plot(app.eta_tPlot,M_0,eta_t(:,:),"LineWidth",1.4)
title(app.eta_tPlot,"$\eta_t$","Interpreter","Latex")
grid(app.eta_tPlot,'minor')
xlabel(app.eta_tPlot,"$M_0$","Interpreter","Latex");
ylabel(app.eta_tPlot,"$\eta_t$ Graph$","Interpreter","Latex");

legend(app.eta_tPlot,'\pi_c='+string(pi_c),'Location','northeast','NumColumns',2);

plot(app.eta_pPlot,M_0,eta_P(:,:),"LineWidth",1.4);
title(app.eta_pPlot," Propulsive Efficiency
$Graph$","Interpreter","Latex")
grid(app.eta_pPlot,'minor')
xlabel(app.eta_pPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.eta_pPlot,"$\eta_P$","Interpreter","Latex");

legend(app.eta_pPlot,'\pi_c='+string(pi_c),'Location','northwest','NumColumns',2);

plot(app.eta_thermoPlot,M_0,eta_T(:,:),"LineWidth",1.4);
title(app.eta_thermoPlot,"Thermal Efficiency
$Graph$","Interpreter","Latex")
grid(app.eta_thermoPlot,'minor')
xlabel(app.eta_thermoPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.eta_thermoPlot,"$\eta_T$","Interpreter","Latex");

legend(app.eta_thermoPlot,'\pi_c='+string(pi_c),'Location','northwest','NumColumns',2);

plot(app.eta_oPlot,M_0,eta_O(:,:),"LineWidth",1.4);
title(app.eta_oPlot,"Efficiency $Graph$","Interpreter","Latex")
grid(app.eta_oPlot,'minor')
xlabel(app.eta_oPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.eta_oPlot,"$\eta_O$","Interpreter","Latex");

legend(app.eta_oPlot,'\pi_c='+string(pi_c),'Location','northwest','NumColumns',2);

case 'Varying Mach No and Bypass'
alpha = alpha_min:alpha_increment:alpha_max;
M_0 = M_0_min:M_0_increment:M_0_max;
pi_f = pi_f_min;
pi_c = pi_c_min;
T_t4 = T_t4_min;
T_t7 = T_t7_min;
T_t17 = T_t17_min;
tablo = zeros(length(M_0)*length(alpha),18);

for i = 1:length(M_0)
    for j=1:length(alpha)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_O(i,j))] = ...

ATFwithSeperateExhausts(app,M_0(i),T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

```

```

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0__P_9,P_0__P_19,...
    T_t4,T_t7,T_t17,pi_c,pi_f,alpha(j));

    tablo(length(alpha)*(i-1)+j,:) = ...
        [M_0(i),pi_c,pi_f,alpha(j),T_t4,T_t7,T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j)
,eta_T(i,j),eta_0(i,j)];

    end
end

% Creating Table
t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
    tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
        tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18));
app.UITable.Data = t;
vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c
','eta_t','eta_f','eta_P','eta_T','eta_0'};
app.UITable.ColumnName = vars;

% Handling Plots
plot(app.FovermdotPlot,M_0,F__mdot_0(:,:),"LineWidth",1.4)
title(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
Graph$","Interpreter","Latex")
grid(app.FovermdotPlot,'minor')
xlabel(app.FovermdotPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
[lbf/(lbfm/sec)]$","Interpreter","Latex");

legend(app.FovermdotPlot,'\alpha='+string(alpha),'Location','northeast','NumColumns',2);

plot(app.SPlot,M_0,S(:,:),"LineWidth",1.4)
title(app.SPlot,"$ S $Graph$","Interpreter","Latex")
grid(app.SPlot,'minor')
xlabel(app.SPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.SPlot,"$S \ [(lbfm/hr)/lbf]$", "Interpreter","Latex");

legend(app.SPlot,'\alpha='+string(alpha),'Location','northeast','NumColumns',2);

plot(app.FuelPlot,M_0,f(:,1),M_0,f_AB(:,:),M_0,f_DB(:,1),"LineWidth",1.4);
title(app.FuelPlot,"Fuel $Graph$","Interpreter","Latex")
grid(app.FuelPlot,'minor')
xlabel(app.FuelPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.FuelPlot,"$f,f_{AB},f_{DB}$","Interpreter","Latex");
leg = ['f for all \alpha=','f_{AB} for \alpha='+string(alpha),'f_{DB}
for all \alpha='];
legend(app.FuelPlot,leg,'Location','northwest','NumColumns',2);

plot(app.eta_cPlot,M_0,eta_c(:,:),"LineWidth",1.4)
title(app.eta_cPlot,"$\eta_c$ $Graph$","Interpreter","Latex")
grid(app.eta_cPlot,'minor')
xlabel(app.eta_cPlot,"$M_0$","Interpreter","Latex");
ylabel(app.eta_cPlot,"$\eta_c$","Interpreter","Latex");

legend(app.eta_cPlot,'\alpha='+string(alpha),'Location','northeast','NumColumns',2);

```

```

        plot(app.eta_tPlot,M_0,eta_t(:,:),"LineWidth",1.4)
        title(app.eta_tPlot,"$\eta_t$", "Interpreter", "Latex")
        grid(app.eta_tPlot,'minor')
        xlabel(app.eta_tPlot,"$M_0$", "Interpreter", "Latex");
        ylabel(app.eta_tPlot,"$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\alpha='+string(alpha), 'Location', 'northeast', 'NumColumns',2);

        plot(app.eta_pPlot,M_0,eta_P(:,:),"LineWidth",1.4);
        title(app.eta_pPlot," Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_pPlot,'minor')
        xlabel(app.eta_pPlot,"$ M_0 $", "Interpreter", "Latex");
        ylabel(app.eta_pPlot,"$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns',2);

        plot(app.eta_thermoPlot,M_0,eta_T(:,:),"LineWidth",1.4);
        title(app.eta_thermoPlot,"Thermal Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_thermoPlot,'minor')
        xlabel(app.eta_thermoPlot,"$ M_0 $", "Interpreter", "Latex");
        ylabel(app.eta_thermoPlot,"$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns',2);

        plot(app.eta_oPlot,M_0,eta_O(:,:),"LineWidth",1.4);
        title(app.eta_oPlot,"Efficiency $Graph$", "Interpreter", "Latex")
        grid(app.eta_oPlot,'minor')
        xlabel(app.eta_oPlot,"$ M_0 $", "Interpreter", "Latex");
        ylabel(app.eta_oPlot,"$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\alpha='+string(alpha), 'Location', 'northwest', 'NumColumns',2);

        case 'Varying Mach No and pi_f'
            alpha = alpha_min;
            M_0 = M_0_min:M_0_increment:M_0_max;
            pi_f = pi_f_min:pi_f_increment:pi_f_max;
            pi_c = pi_c_min;
            T_t4 = T_t4_min;
            T_t7 = T_t7_min;
            T_t17 = T_t17_min;
            tablo = zeros(length(pi_f)*length(M_0),18);

            for i = 1:length(M_0)
                for j=1:length(pi_f)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_O(i,j)] = ...

ATFwithSeperateExhausts(app,M_0(i),T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0_P_9,P_0_P_19,...
                T_t4,T_t7,T_t17,pi_c,pi_f(j),alpha);

                tablo(length(pi_f)*(i-1)+j,:) = ...
                [M_0(i),pi_c,pi_f(j),alpha,T_t4,T_t7,T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_O(i,j)];

            end
        end
end

```

```

% Creating Table
t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
    tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18));
app.UITable.Data = t;
vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c',
'eta_t','eta_f','eta_P','eta_T','eta_0'};
app.UITable.ColumnName = vars;

% Handling Plots
plot(app.FovermdotPlot,M_0,F__mdot_0(:,:),"LineWidth",1.4);
%
    hold(app.FovermdotPlot,'on');
%
plot(app.FovermdotPlot,S(:,:),'F__mdot_0(:,:)', "Marker", 'o', "LineWidth",1.4); % degisiklik
yapilacak
    title(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
Graph$", "Interpreter", "Latex")
    grid(app.FovermdotPlot, 'minor')
    xlabel(app.FovermdotPlot,"$ M_0 $", "Interpreter", "Latex");
    ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
[lbf/(lbm/sec)]$", "Interpreter", "Latex");

legend(app.FovermdotPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

    plot(app.SPlot,M_0,S(:,:),"LineWidth",1.4)
    title(app.SPlot,"S $Graph$", "Interpreter", "Latex")
    grid(app.SPlot, 'minor')
    xlabel(app.SPlot,"$ M_0 $", "Interpreter", "Latex");
    ylabel(app.SPlot,"$S\ [(lbm/hr)/lbf]$", "Interpreter", "Latex");

legend(app.SPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

plot(app.FuelPlot,M_0,f(:,1),M_0,f_AB(:,:),M_0,f_DB(:,:),"LineWidth",1.4);%%
    title(app.FuelPlot,"Fuel $Graph$", "Interpreter", "Latex")
    grid(app.FuelPlot, 'minor')
    xlabel(app.FuelPlot,"$ M_0 $", "Interpreter", "Latex");
    ylabel(app.FuelPlot,"$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
    leg = ['f for all \pi_f', 'f_{AB} for \pi_f='+string(pi_f), 'f_{DB} for
\pi_f='+string(pi_f)];
    legend(app.FuelPlot,leg, 'Location', 'northwest', 'NumColumns', 2);

    plot(app.eta_cPlot,M_0,eta_c(:,:),"LineWidth",1.4)
    title(app.eta_cPlot,"$\eta_c$ $Graph$", "Interpreter", "Latex")
    grid(app.eta_cPlot, 'minor')
    xlabel(app.eta_cPlot,"$M_0$", "Interpreter", "Latex");
    ylabel(app.eta_cPlot,"$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

    plot(app.eta_tPlot,M_0,eta_t(:,:),"LineWidth",1.4)
    title(app.eta_tPlot,"$\eta_t$", "Interpreter", "Latex")
    grid(app.eta_tPlot, 'minor')
    xlabel(app.eta_tPlot,"$M_0$", "Interpreter", "Latex");
    ylabel(app.eta_tPlot,"$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, '\pi_f='+string(pi_f), 'Location', 'northeast', 'NumColumns', 2);

    plot(app.eta_pPlot,M_0,eta_P(:,:),"LineWidth",1.4);

```



```

        title(app.eta_pPlot, " Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_pPlot, 'minor')
        xlabel(app.eta_pPlot, "$ M_0 $", "Interpreter", "Latex");
        ylabel(app.eta_pPlot, "$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_thermoPlot, M_0, eta_T(:, :), "LineWidth", 1.4);
        title(app.eta_thermoPlot, "Thermal Efficiency
$Graph$", "Interpreter", "Latex")
        grid(app.eta_thermoPlot, 'minor')
        xlabel(app.eta_thermoPlot, "$ M_0 $", "Interpreter", "Latex");
        ylabel(app.eta_thermoPlot, "$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

        plot(app.eta_oPlot, M_0, eta_O(:, :), "LineWidth", 1.4);
        title(app.eta_oPlot, "Efficiency $Graph$", "Interpreter", "Latex")
        grid(app.eta_oPlot, 'minor')
        xlabel(app.eta_oPlot, "$ M_0 $", "Interpreter", "Latex");
        ylabel(app.eta_oPlot, "$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, '\pi_f='+string(pi_f), 'Location', 'northwest', 'NumColumns', 2);

        case 'Varying Mach No and T_t4'
            alpha = alpha_min;
            M_0 = M_0_min:M_0_increment:M_0_max;
            pi_f = pi_f_min;
            pi_c = pi_c_min;
            T_t4 = T_t4_min:T_t4_increment:T_t4_max;
            T_t7 = T_t7_min;
            T_t17 = T_t17_min;
            tablo = zeros(length(M_0)*length(T_t4), 18);

            for i = 1:length(M_0)
                for j=1:length(T_t4)

[F__mdot_0(i,j), f(i,j), f_AB(i,j), f_DB(i,j), S(i,j), eta_c(i,j), eta_t(i,j), eta_f(i,j), eta_P(i,
j), eta_T(i,j), eta_O(i,j)] = ...

ATFwithSeperateExhausts(app, M_0(i), T_0, gamma_c, c_pc, gamma_t, c_pt, h_PR, ...

gamma_AB, c_p_AB, gamma_DB, c_p_DB, pi_d_max, pi_b, pi_AB, pi_DB, pi_n, ...

pi_fn, e_c, e_f, e_t, eta_b, eta_AB, eta_DB, eta_m, P_0_P_9, P_0_P_19, ...
                T_t4(j), T_t7, T_t17, pi_c, pi_f, alpha);

                tablo(length(T_t4)*(i-1)+j, :) = ...
                    [M_0(i), pi_c, pi_f, alpha, T_t4(j), T_t7, T_t17, ...

F__mdot_0(i,j), f(i,j), f_AB(i,j), f_DB(i,j), S(i,j), eta_c(i,j), eta_t(i,j), eta_f(i,j), eta_P(i, j
), eta_T(i,j), eta_O(i,j)];

            end
        end

        % Creating Table
        t =
        table(tablo(:, 1), tablo(:, 2), tablo(:, 3), tablo(:, 4), tablo(:, 5), tablo(:, 6), tablo(:, 7), ...
        tablo(:, 8), tablo(:, 9), tablo(:, 10), tablo(:, 11), tablo(:, 12), tablo(:, 13), tablo(:, 14), ...
            tablo(:, 15), tablo(:, 16), tablo(:, 17), tablo(:, 18));
        app.UITable.Data = t;

```

```

vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c',
', 'eta_t','eta_f','eta_P','eta_T','eta_0'};
app.UITable.ColumnName = vars;

% Handling Plots
plot(app.FovermdotPlot,M_0,F__mdot_0(:,:),"LineWidth",1.4);
title(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
Graph$","Interpreter","Latex")
grid(app.FovermdotPlot,'minor')
xlabel(app.FovermdotPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
[lbf/(lbm/sec)]$","Interpreter","Latex");

legend(app.FovermdotPlot,'T_{t4}='+string(T_t4),'Location','northeast','NumColumns',2);

plot(app.SPlot,M_0,S(:,:),"LineWidth",1.4)
title(app.SPlot,"S $Graph$","Interpreter","Latex")
grid(app.SPlot,'minor')
xlabel(app.SPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.SPlot,"$S$ [(lbf/hr)/lbf]$","Interpreter","Latex");

legend(app.SPlot,'T_{t4}='+string(T_t4),'Location','northeast','NumColumns',2);

plot(app.FuelPlot,M_0,f(:,:),M_0,f_AB(:,:),M_0,f_DB(:,:),"LineWidth",1.4);%%
title(app.FuelPlot,"Fuel $Graph$","Interpreter","Latex")
grid(app.FuelPlot,'minor')
xlabel(app.FuelPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.FuelPlot,"$f,f_{AB},f_{DB}$","Interpreter","Latex");
leg = ['f for all T_{t4}'+string(T_t4),'f_{AB} for
T_{t4}='+string(T_t4),'f_{DB} for T_{t4}='+string(T_t4)];
legend(app.FuelPlot,leg,'Location','northwest','NumColumns',2);

plot(app.eta_cPlot,M_0,eta_c(:,:),"LineWidth",1.4)
title(app.eta_cPlot,"$\eta_c$ $Graph$","Interpreter","Latex")
grid(app.eta_cPlot,'minor')
xlabel(app.eta_cPlot,"$M_0$","Interpreter","Latex");
ylabel(app.eta_cPlot,"$\eta_c$","Interpreter","Latex");

legend(app.eta_cPlot,'T_{t4}='+string(T_t4),'Location','northeast','NumColumns',2);

plot(app.eta_tPlot,M_0,eta_t(:,:),"LineWidth",1.4)
title(app.eta_tPlot,"$\eta_t$","Interpreter","Latex")
grid(app.eta_tPlot,'minor')
xlabel(app.eta_tPlot,"$M_0$","Interpreter","Latex");
ylabel(app.eta_tPlot,"$\eta_t$ Graph$","Interpreter","Latex");

legend(app.eta_tPlot,'T_{t4}='+string(T_t4),'Location','northeast','NumColumns',2);

plot(app.eta_pPlot,M_0,eta_P(:,:),"LineWidth",1.4);
title(app.eta_pPlot," Propulsive Efficiency
$Graph$","Interpreter","Latex")
grid(app.eta_pPlot,'minor')
xlabel(app.eta_pPlot,"$ M_0 $","Interpreter","Latex");
ylabel(app.eta_pPlot,"$\eta_P$","Interpreter","Latex");

legend(app.eta_pPlot,'T_{t4}='+string(T_t4),'Location','northwest','NumColumns',2);

plot(app.eta_thermoPlot,M_0,eta_T(:,:),"LineWidth",1.4);
title(app.eta_thermoPlot,"Thermal Efficiency
$Graph$","Interpreter","Latex")
grid(app.eta_thermoPlot,'minor')
xlabel(app.eta_thermoPlot,"$ M_0 $","Interpreter","Latex");

```

```

ylabel(app.eta_thermoPlot,"$\eta_T$","Interpreter","Latex");

legend(app.eta_thermoPlot,'T_{t4}='+string(T_t4),'Location','northwest','NumColumns',2);

plot(app.eta_oPlot,M_0,eta_0(:,:),"LineWidth",1.4);
title(app.eta_oPlot,"Efficiency $Graph$","Interpreter","Latex")
grid(app.eta_oPlot,'minor')
xlabel(app.eta_oPlot,"$M_0$","Interpreter","Latex");
ylabel(app.eta_oPlot,"$\eta_0$","Interpreter","Latex");

legend(app.eta_oPlot,'T_{t4}='+string(T_t4),'Location','northwest','NumColumns',2);

case 'Varying Mach No and T_t7'
    alpha = alpha_min;
    M_0 = M_0_min:M_0_increment:M_0_max;
    pi_f = pi_f_min;
    pi_c = pi_c_min;
    T_t4 = T_t4_min;
    T_t7 = T_t7_min:T_t7_increment:T_t7_max;
    T_t17 = T_t17_min;
    tablo = zeros(length(M_0)*length(T_t7),18);

    for i = 1:length(M_0)
        for j=1:length(T_t7)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_0(i,j)] = ...

ATFwithSeperateExhausts(app,M_0(i),T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0__P_9,P_0__P_19,...
            T_t4,T_t7(j),T_t17,pi_c,pi_f,alpha);

            tablo(length(T_t7)*(i-1)+j,:) = ...
            [M_0(i),pi_c,pi_f,alpha,T_t4,T_t7(j),T_t17,...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_0(i,j)];

        end
    end

% Creating Table
t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
        tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18)));
app.UITable.Data = t;
vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c
','eta_t','eta_f','eta_P','eta_T','eta_0'};
app.UITable.ColumnName = vars;

% Handling Plots
plot(app.FovermdotPlot,M_0,F__mdot_0(:,:),"LineWidth",1.4);
title(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
Graph$","Interpreter","Latex")
grid(app.FovermdotPlot,'minor')
xlabel(app.FovermdotPlot,"$M_0$","Interpreter","Latex");
ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
[lbf/(lbm/sec)]$","Interpreter","Latex");

```

```

legend(app.FovermdotPlot, 'T_{t7}='+string(T_t7), 'Location', 'northeast', 'NumColumns', 2);

plot(app.SPlot, M_0, S(:, :), "LineWidth", 1.4)
title(app.SPlot, "S $Graph$", "Interpreter", "Latex")
grid(app.SPlot, 'minor')
xlabel(app.SPlot, "$ M_0 $", "Interpreter", "Latex");
ylabel(app.SPlot, "$S \ [(\text{lbfm/hr})/\text{lbf}]$", "Interpreter", "Latex");

legend(app.SPlot, 'T_{t7}='+string(T_t7), 'Location', 'northeast', 'NumColumns', 2);

plot(app.FuelPlot, M_0, f(:, :), M_0, f_AB(:, :), M_0, f_DB(:, :), "LineWidth", 1.4); %%
title(app.FuelPlot, "Fuel $Graph$", "Interpreter", "Latex")
grid(app.FuelPlot, 'minor')
xlabel(app.FuelPlot, "$ M_0 $", "Interpreter", "Latex");
ylabel(app.FuelPlot, "$f, f_{AB}, f_{DB}$", "Interpreter", "Latex");
leg = ['f for all T_{t7}'+string(T_t7), 'f_{AB} for
T_{t7}='+string(T_t7), 'f_{DB} for T_{t7}='+string(T_t7)];
legend(app.FuelPlot, leg, 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_cPlot, M_0, eta_c(:, :), "LineWidth", 1.4)
title(app.eta_cPlot, "$\eta_c$ $Graph$", "Interpreter", "Latex")
grid(app.eta_cPlot, 'minor')
xlabel(app.eta_cPlot, "$M_0$", "Interpreter", "Latex");
ylabel(app.eta_cPlot, "$\eta_c$", "Interpreter", "Latex");

legend(app.eta_cPlot, 'T_{t7}='+string(T_t7), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_tPlot, M_0, eta_t(:, :), "LineWidth", 1.4)
title(app.eta_tPlot, "$\eta_t$", "Interpreter", "Latex")
grid(app.eta_tPlot, 'minor')
xlabel(app.eta_tPlot, "$M_0$", "Interpreter", "Latex");
ylabel(app.eta_tPlot, "$\eta_t$ Graph$", "Interpreter", "Latex");

legend(app.eta_tPlot, 'T_{t7}='+string(T_t7), 'Location', 'northeast', 'NumColumns', 2);

plot(app.eta_pPlot, M_0, eta_P(:, :), "LineWidth", 1.4);
title(app.eta_pPlot, " Propulsive Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_pPlot, 'minor')
xlabel(app.eta_pPlot, "$ M_0 $", "Interpreter", "Latex");
ylabel(app.eta_pPlot, "$\eta_P$", "Interpreter", "Latex");

legend(app.eta_pPlot, 'T_{t7}='+string(T_t7), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_thermoPlot, M_0, eta_T(:, :), "LineWidth", 1.4);
title(app.eta_thermoPlot, "Thermal Efficiency
$Graph$", "Interpreter", "Latex")
grid(app.eta_thermoPlot, 'minor')
xlabel(app.eta_thermoPlot, "$ M_0 $", "Interpreter", "Latex");
ylabel(app.eta_thermoPlot, "$\eta_T$", "Interpreter", "Latex");

legend(app.eta_thermoPlot, 'T_{t7}='+string(T_t7), 'Location', 'northwest', 'NumColumns', 2);

plot(app.eta_oPlot, M_0, eta_O(:, :), "LineWidth", 1.4);
title(app.eta_oPlot, "Efficiency $Graph$", "Interpreter", "Latex")
grid(app.eta_oPlot, 'minor')
xlabel(app.eta_oPlot, "$ M_0 $", "Interpreter", "Latex");
ylabel(app.eta_oPlot, "$\eta_O$", "Interpreter", "Latex");

legend(app.eta_oPlot, 'T_{t7}='+string(T_t7), 'Location', 'northwest', 'NumColumns', 2);

case 'Varying Mach No and T_t17'
alpha = alpha_min;

```

```

M_0 = M_0_min:M_0_increment:M_0_max;
pi_f = pi_f_min;
pi_c = pi_c_min;
T_t4 = T_t4_min;
T_t7 = T_t7_min;
T_t17 = T_t17_min:T_t17_increment:T_t17_max;
tablo = zeros(length(M_0)*length(T_t7),18);

for i = 1:length(M_0)
    for j=1:length(T_t17)

[F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,
j),eta_T(i,j),eta_O(i,j))] = ...

ATFwithSeperateExhausts(app,M_0(i),T_0,gamma_c,c_pc,gamma_t,c_pt,h_PR,...

gamma_AB,c_p_AB,gamma_DB,c_p_DB,pi_d_max,pi_b,pi_AB,pi_DB,pi_n,...

pi_fn,e_c,e_f,e_t,eta_b,eta_AB,eta_DB,eta_m,P_0__P_9,P_0__P_19,...
        T_t4,T_t7,T_t17(j),pi_c,pi_f,alpha);

        tablo(length(T_t17)*(i-1)+j,:) = ...
            [M_0(i),pi_c,pi_f,alpha,T_t4,T_t7,T_t17(j),...

F__mdot_0(i,j),f(i,j),f_AB(i,j),f_DB(i,j),S(i,j),eta_c(i,j),eta_t(i,j),eta_f(i,j),eta_P(i,j
),eta_T(i,j),eta_O(i,j)];

        end
    end

    % Creating Table
    t =
table(tablo(:,1),tablo(:,2),tablo(:,3),tablo(:,4),tablo(:,5),tablo(:,6),tablo(:,7),...
tablo(:,8),tablo(:,9),tablo(:,10),tablo(:,11),tablo(:,12),tablo(:,13),tablo(:,14),...
        tablo(:,15),tablo(:,16),tablo(:,17),tablo(:,18)));
    app.UITable.Data = t;
    vars =
{'M_0','pi_c','pi_f','alpha','T_t4','T_t7','T_t17','F__mdot_0','f','f_AB','f_DB','S','eta_c
','eta_t','eta_f','eta_P','eta_T','eta_O'};
    app.UITable.ColumnName = vars;

    % Handling Plots

    plot(app.FovermdotPlot,M_0,F__mdot_0(:,:),"LineWidth",1.4);
    title(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
Graph$","Interpreter","Latex")
    grid(app.FovermdotPlot,'minor')
    xlabel(app.FovermdotPlot,"$ M_0 $","Interpreter","Latex");
    ylabel(app.FovermdotPlot,"$\frac{F}{\dot{m}_0}$\
[lbf/(lbm/sec)]$","Interpreter","Latex");

    legend(app.FovermdotPlot,'T_{t17}='+string(T_t17),'Location','northeast','NumColumns',2);
    set(app.FovermdotPlot,'Colormap',[1,1,1]);
    plot(app.SPlot,M_0,S(:,:),"LineWidth",1.4)
    title(app.SPlot,"S $Graph$","Interpreter","Latex")
    grid(app.SPlot,'minor')
    xlabel(app.SPlot,"$ M_0 $","Interpreter","Latex");
    ylabel(app.SPlot,"$S \ [(lbm/hr)/lbf] $","Interpreter","Latex");

    legend(app.SPlot,'T_{t17}='+string(T_t17),'Location','northeast','NumColumns',2);

    plot(app.FuelPlot,M_0,f(:,:),M_0,f_AB(:,:),M_0,f_DB(:,:),"LineWidth",1.4);%

```

```

        title(app.FuelPlot,"Fuel $Graph$","Interpreter","Latex")
        grid(app.FuelPlot,'minor')
        xlabel(app.FuelPlot,"$ M_0 $","Interpreter","Latex");
        ylabel(app.FuelPlot,"$f_{AB}$, $f_{DB}$","Interpreter","Latex");
        leg = ['f for all T_{t17}'+string(T_t17), 'f_{AB} for
T_{t17}='+string(T_t17), 'f_{DB} for T_{t17}='+string(T_t17)];
        legend(app.FuelPlot,leg,'Location','northwest','NumColumns',2);

        plot(app.eta_cPlot,M_0,eta_c(:,:),"LineWidth",1.4)
        title(app.eta_cPlot,"$\eta_c$ $Graph$","Interpreter","Latex")
        grid(app.eta_cPlot,'minor')
        xlabel(app.eta_cPlot,"$M_0$","Interpreter","Latex");
        ylabel(app.eta_cPlot,"$\eta_c$","Interpreter","Latex");

        legend(app.eta_cPlot,'T_{t17}='+string(T_t17),'Location','northeast','NumColumns',2);

        plot(app.eta_tPlot,M_0,eta_t(:,:),"LineWidth",1.4)
        title(app.eta_tPlot,"$\eta_t$ \ Graph$","Interpreter","Latex")
        grid(app.eta_tPlot,'minor')
        xlabel(app.eta_tPlot,"$M_0$","Interpreter","Latex");
        ylabel(app.eta_tPlot,"$\eta_t$","Interpreter","Latex");

        legend(app.eta_tPlot,'T_{t17}='+string(T_t17),'Location','northeast','NumColumns',2);

        plot(app.eta_pPlot,M_0,eta_P(:,:),"LineWidth",1.4);
        title(app.eta_pPlot," Propulsive Efficiency
$Graph$","Interpreter","Latex")
        grid(app.eta_pPlot,'minor')
        xlabel(app.eta_pPlot,"$ M_0 $","Interpreter","Latex");
        ylabel(app.eta_pPlot,"$\eta_P$","Interpreter","Latex");

        legend(app.eta_pPlot,'T_{t17}='+string(T_t17),'Location','northwest','NumColumns',2);

        plot(app.eta_thermoPlot,M_0,eta_T(:,:),"LineWidth",1.4);
        title(app.eta_thermoPlot,"Thermal Efficiency
$Graph$","Interpreter","Latex")
        grid(app.eta_thermoPlot,'minor')
        xlabel(app.eta_thermoPlot,"$ M_0 $","Interpreter","Latex");
        ylabel(app.eta_thermoPlot,"$\eta_T$","Interpreter","Latex");

        legend(app.eta_thermoPlot,'T_{t17}='+string(T_t17),'Location','northwest','NumColumns',2);

        plot(app.eta_oPlot,M_0,eta_O(:,:),"LineWidth",1.4);
        title(app.eta_oPlot,"Efficiency $Graph$","Interpreter","Latex")
        grid(app.eta_oPlot,'minor')
        xlabel(app.eta_oPlot,"$ M_0 $","Interpreter","Latex");
        ylabel(app.eta_oPlot,"$\eta_O$","Interpreter","Latex");

        legend(app.eta_oPlot,'T_{t17}='+string(T_t17),'Location','northwest','NumColumns',2);
    end

end

% Value changed function: CaseDropDown
function CaseDropDownValueChanged(app, event)
    value = app.CaseDropDown.Value;

    switch value
    case 'Single Condition'
        set(app.MachNoIncrementEditField,'enable','off');
        set(app.MaxMachNumberEditField,'enable','off');
        set(app.pi_cIncrementEditField,'enable','off');
        set(app.Maxpi_cValueEditField,'enable','off');
        set(app.pi_fIncrementEditField,'enable','off');

```

```

set(app.Maxpi_fEditField,'enable','off');
set(app.ByPassRatioIncrementEditField,'enable','off');
set(app.MaxByPassRatioEditField,'enable','off');
set(app.T_t4IncrementEditField,'enable','off');
set(app.MaxT_t4KEditField,'enable','off');
set(app.T_t7IncrementEditField,'enable','off');
set(app.MaxT_t7KEditField,'enable','off');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
app.MachNumberEditFieldLabel.Text = 'Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'pi_f';
app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
app.Minpi_cValueLabel.Text = 'pi_c';
app.MinT_t4KLabel.Text = 'T_t4 (K)';
app.MinT_t7KLabel.Text = 'T_t7 (K)';
app.MinT_t17KLabel.Text = 'T_t17 (K)';
set(app.XAxisSwitch,'enable','off');

case 'Varying Bypass and pi_f'
set(app.MachNoIncrementEditField,'enable','off');
set(app.MaxMachNumberEditField,'enable','off');
set(app.pi_cIncrementEditField,'enable','off');
set(app.Maxpi_cValueEditField,'enable','off');
set(app.pi_fIncrementEditField,'enable','on');
set(app.Maxpi_fEditField,'enable','on');
set(app.ByPassRatioIncrementEditField,'enable','on');
set(app.T_t4IncrementEditField,'enable','off');
set(app.MaxT_t4KEditField,'enable','off');
set(app.T_t7IncrementEditField,'enable','off');
set(app.MaxT_t7KEditField,'enable','off');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
set(app.MaxByPassRatioEditField,'enable','on');
set(app.XAxisSwitch,'enable','on');
app.MachNumberEditFieldLabel.Text = 'Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'Min pi_f Value';
app.MinByPassRatioValueEditFieldLabel.Text = 'Min By-pass Ratio';
app.Minpi_cValueLabel.Text = 'pi_c';
app.MinT_t4KLabel.Text = 'T_t4 (K)';
app.MinT_t7KLabel.Text = 'T_t7 (K)';
app.MinT_t17KLabel.Text = 'T_t17 (K)';
app.XAxisSwitch.Items = {'alpha','pi_f'};

case 'Varying pi_f and pi_c'
set(app.MachNoIncrementEditField,'enable','off');
set(app.MaxMachNumberEditField,'enable','off');
set(app.pi_cIncrementEditField,'enable','on');
set(app.Maxpi_cValueEditField,'enable','on');
set(app.pi_fIncrementEditField,'enable','on');
set(app.Maxpi_fEditField,'enable','on');
set(app.ByPassRatioIncrementEditField,'enable','off');
set(app.MaxByPassRatioEditField,'enable','off');
set(app.T_t4IncrementEditField,'enable','off');
set(app.MaxT_t4KEditField,'enable','off');
set(app.T_t7IncrementEditField,'enable','off');
set(app.MaxT_t7KEditField,'enable','off');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
set(app.XAxisSwitch,'enable','on');
app.MachNumberEditFieldLabel.Text = 'Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'Min pi_f Value';
app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
app.Minpi_cValueLabel.Text = 'Min pi_c Value';
app.MinT_t4KLabel.Text = 'T_t4 (K)';
app.MinT_t7KLabel.Text = 'T_t7 (K)';

```



```

app.MinT_t17KLabel.Text = 'T_t17 (K)';
app.XAxisSwitch.Items = {'pi_f', 'pi_c'};
case 'Varying Bypass and pi_c'

    set(app.MachNoIncrementEditField, 'enable', 'off');
    set(app.MaxMachNumberEditField, 'enable', 'off');
    set(app.pi_cIncrementEditField, 'enable', 'on');
    set(app.Maxpi_cValueEditField, 'enable', 'on');
    set(app.pi_fIncrementEditField, 'enable', 'off');
    set(app.Maxpi_fEditField, 'enable', 'off');
    set(app.ByPassRatioIncrementEditField, 'enable', 'on');
    set(app.MaxByPassRatioEditField, 'enable', 'on');
    set(app.T_t4IncrementEditField, 'enable', 'off');
    set(app.MaxT_t4KEditField, 'enable', 'off');
    set(app.T_t7IncrementEditField, 'enable', 'off');
    set(app.MaxT_t7KEditField, 'enable', 'off');
    set(app.T_t17IncrementEditField, 'enable', 'off');
    set(app.MaxT_t17KEditField, 'enable', 'off');
    set(app.XAxisSwitch, 'enable', 'on');
    app.MachNumberEditFieldLabel.Text = 'Mach Number';
    app.Minpi_fValueEditFieldLabel.Text = 'pi_f Value';
    app.MinByPassRatioValueEditFieldLabel.Text = 'Min By-pass Ratio';
    app.Minpi_cValueLabel.Text = 'Min pi_c Value';
    app.MinT_t4KLabel.Text = 'T_t4 (K)';
    app.MinT_t7KLabel.Text = 'T_t7 (K)';
    app.MinT_t17KLabel.Text = 'T_t17 (K)';
    app.XAxisSwitch.Items = {'pi_c', 'alpha'};

case 'Varying Mach No and pi_c'
    set(app.MachNoIncrementEditField, 'enable', 'on');
    set(app.MaxMachNumberEditField, 'enable', 'on');
    set(app.pi_cIncrementEditField, 'enable', 'on');
    set(app.Maxpi_cValueEditField, 'enable', 'on');
    set(app.pi_fIncrementEditField, 'enable', 'off');
    set(app.Maxpi_fEditField, 'enable', 'off');
    set(app.ByPassRatioIncrementEditField, 'enable', 'off');
    set(app.MaxByPassRatioEditField, 'enable', 'off');
    set(app.T_t4IncrementEditField, 'enable', 'off');
    set(app.MaxT_t4KEditField, 'enable', 'off');
    set(app.T_t7IncrementEditField, 'enable', 'off');
    set(app.MaxT_t7KEditField, 'enable', 'off');
    set(app.T_t17IncrementEditField, 'enable', 'off');
    set(app.MaxT_t17KEditField, 'enable', 'off');
    set(app.XAxisSwitch, 'enable', 'off');
    app.MachNumberEditFieldLabel.Text = 'Min Mach Number';
    app.Minpi_fValueEditFieldLabel.Text = 'pi_f Value';
    app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
    app.Minpi_cValueLabel.Text = 'Min pi_c Value';
    app.MinT_t4KLabel.Text = 'T_t4 (K)';
    app.MinT_t7KLabel.Text = 'T_t7 (K)';
    app.MinT_t17KLabel.Text = 'T_t17 (K)';
    app.XAxisSwitch.Items = {'M_0', 'pi_c'};

case 'Varying Mach No and Bypass'
    set(app.MachNoIncrementEditField, 'enable', 'on');
    set(app.MaxMachNumberEditField, 'enable', 'on');
    set(app.pi_cIncrementEditField, 'enable', 'off');
    set(app.Maxpi_cValueEditField, 'enable', 'off');
    set(app.pi_fIncrementEditField, 'enable', 'off');
    set(app.Maxpi_fEditField, 'enable', 'off');
    set(app.ByPassRatioIncrementEditField, 'enable', 'on');
    set(app.MaxByPassRatioEditField, 'enable', 'on');
    set(app.T_t4IncrementEditField, 'enable', 'off');
    set(app.MaxT_t4KEditField, 'enable', 'off');
    set(app.T_t7IncrementEditField, 'enable', 'off');

```

```

set(app.MaxT_t7KEditField,'enable','off');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
set(app.XAxisSwitch,'enable','off');
app.MachNumberEditFieldLabel.Text = 'Min Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'pi_f Value';
app.MinByPassRatioValueEditFieldLabel.Text = 'Min By-pass Ratio';
app.Minpi_cValueLabel.Text = 'pi_c Value';
app.MinT_t4KLabel.Text = 'T_t4 (K)';
app.MinT_t7KLabel.Text = 'T_t7 (K)';
app.MinT_t17KLabel.Text = 'T_t17 (K)';
app.XAxisSwitch.Items = {'M_0','alpha'};

case 'Varying Mach No and pi_f'
set(app.MachNoIncrementEditField,'enable','on');
set(app.MaxMachNumberEditField,'enable','on');
set(app.pi_cIncrementEditField,'enable','off');
set(app.Maxpi_cValueEditField,'enable','off');
set(app.pi_fIncrementEditField,'enable','on');
set(app.Maxpi_fEditField,'enable','on');
set(app.ByPassRatioIncrementEditField,'enable','off');
set(app.MaxByPassRatioEditField,'enable','off');
set(app.T_t4IncrementEditField,'enable','off');
set(app.MaxT_t4KEditField,'enable','off');
set(app.T_t7IncrementEditField,'enable','off');
set(app.MaxT_t7KEditField,'enable','off');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
set(app.XAxisSwitch,'enable','off');
app.MachNumberEditFieldLabel.Text = 'Min Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'Min pi_f Value';
app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
app.Minpi_cValueLabel.Text = 'pi_c Value';
app.MinT_t4KLabel.Text = 'T_t4 (K)';
app.MinT_t7KLabel.Text = 'T_t7 (K)';
app.MinT_t17KLabel.Text = 'T_t17 (K)';
app.XAxisSwitch.Items = {'M_0','pi_F'};

case 'Varying Mach No and T_t4'
set(app.MachNoIncrementEditField,'enable','on');
set(app.MaxMachNumberEditField,'enable','on');
set(app.pi_cIncrementEditField,'enable','off');
set(app.Maxpi_cValueEditField,'enable','off');
set(app.pi_fIncrementEditField,'enable','off');
set(app.Maxpi_fEditField,'enable','off');
set(app.ByPassRatioIncrementEditField,'enable','off');
set(app.MaxByPassRatioEditField,'enable','off');
set(app.T_t4IncrementEditField,'enable','on');
set(app.MaxT_t4KEditField,'enable','on');
set(app.T_t7IncrementEditField,'enable','off');
set(app.MaxT_t7KEditField,'enable','off');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
set(app.XAxisSwitch,'enable','off');
app.MachNumberEditFieldLabel.Text = 'Min Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'pi_f Value';
app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
app.Minpi_cValueLabel.Text = 'pi_c Value';
app.MinT_t4KLabel.Text = 'Min T_t4 (K)';
app.MinT_t7KLabel.Text = 'T_t7 (K)';
app.MinT_t17KLabel.Text = 'T_t17 (K)';
app.XAxisSwitch.Items = {'M_0','T_t4'};

case 'Varying Mach No and T_t7'
set(app.MachNoIncrementEditField,'enable','on');

```

```

set(app.MaxMachNumberEditField,'enable','on');
set(app.pi_cIncrementEditField,'enable','off');
set(app.Maxpi_cValueEditField,'enable','off');
set(app.pi_fIncrementEditField,'enable','off');
set(app.Maxpi_fEditField,'enable','off');
set(app.ByPassRatioIncrementEditField,'enable','off');
set(app.MaxByPassRatioEditField,'enable','off');
set(app.T_t4IncrementEditField,'enable','off');
set(app.MaxT_t4KEditField,'enable','off');
set(app.T_t7IncrementEditField,'enable','on');
set(app.MaxT_t7KEditField,'enable','on');
set(app.T_t17IncrementEditField,'enable','off');
set(app.MaxT_t17KEditField,'enable','off');
set(app.XAxisSwitch,'enable','off');
app.MachNumberEditFieldLabel.Text = 'Min Mach Number';
app.Minpi_fValueEditFieldLabel.Text = 'pi_f Value';
app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
app.Minpi_cValueLabel.Text = 'pi_c Value';
app.MinT_t4KLabel.Text = 'T_t4 (K)';
app.MinT_t7KLabel.Text = 'Min T_t7 (K)';
app.MinT_t17KLabel.Text = 'T_t17 (K)';
app.XAxisSwitch.Items = {'M_0','T_t7'};

case 'Varying Mach No and T_t17'
    set(app.MachNoIncrementEditField,'enable','on');
    set(app.MaxMachNumberEditField,'enable','on');
    set(app.pi_cIncrementEditField,'enable','off');
    set(app.Maxpi_cValueEditField,'enable','off');
    set(app.pi_fIncrementEditField,'enable','off');
    set(app.Maxpi_fEditField,'enable','off');
    set(app.ByPassRatioIncrementEditField,'enable','off');
    set(app.MaxByPassRatioEditField,'enable','off');
    set(app.T_t4IncrementEditField,'enable','off');
    set(app.MaxT_t4KEditField,'enable','off');
    set(app.T_t7IncrementEditField,'enable','off');
    set(app.MaxT_t7KEditField,'enable','off');
    set(app.T_t17IncrementEditField,'enable','on');
    set(app.MaxT_t17KEditField,'enable','on');
    set(app.XAxisSwitch,'enable','off');
    app.MachNumberEditFieldLabel.Text = 'Min Mach Number';
    app.Minpi_fValueEditFieldLabel.Text = 'pi_f Value';
    app.MinByPassRatioValueEditFieldLabel.Text = 'By-pass Ratio';
    app.Minpi_cValueLabel.Text = 'pi_c Value';
    app.MinT_t4KLabel.Text = 'T_t4 (K)';
    app.MinT_t7KLabel.Text = 'T_t7 (K)';
    app.MinT_t17KLabel.Text = 'Min T_t17 (K)';
    app.XAxisSwitch.Items = {'M_0','T_t17'};
end
end
end

% Component initialization
methods (Access = private)

% Create UIFigure and components
function createComponents(app)

% Create AfterburnerTurbofan and hide until all components are created
app.AfterburnerTurbofan = uifigure('Visible','off');
app.AfterburnerTurbofan.Position = [100 100 1130 708];
app.AfterburnerTurbofan.Name = 'UI Figure';

```

```

% Create FlightConditionsPanel
app.FlightConditionsPanel = uipanel(app.AfterburnerTurbofan);
app.FlightConditionsPanel.TitlePosition = 'centertop';
app.FlightConditionsPanel.Title = 'Flight Conditions';
app.FlightConditionsPanel.FontWeight = 'bold';
app.FlightConditionsPanel.Position = [21 376 264 261];

% Create GridLayout
app.GridLayout = uigridlayout(app.FlightConditionsPanel);
app.GridLayout.RowHeight = {'1x', '1x', '1x', '1x', '1x', '1x', '1x', '1x'};

% Create T_0KEditFieldLabel
app.T_0KEditFieldLabel = uilabel(app.GridLayout);
app.T_0KEditFieldLabel.HorizontalAlignment = 'center';
app.T_0KEditFieldLabel.Layout.Row = 4;
app.T_0KEditFieldLabel.Layout.Column = 1;
app.T_0KEditFieldLabel.Text = 'T_0 (K)';

% Create T_0KEditField
app.T_0KEditField = uieditfield(app.GridLayout, 'numeric');
app.T_0KEditField.HorizontalAlignment = 'center';
app.T_0KEditField.Layout.Row = 4;
app.T_0KEditField.Layout.Column = 2;
app.T_0KEditField.Value = 216.6667;

% Create MachNumberEditFieldLabel
app.MachNumberEditFieldLabel = uilabel(app.GridLayout);
app.MachNumberEditFieldLabel.HorizontalAlignment = 'center';
app.MachNumberEditFieldLabel.Layout.Row = 1;
app.MachNumberEditFieldLabel.Layout.Column = 1;
app.MachNumberEditFieldLabel.Text = 'Mach Number';

% Create MachNumberMinEditField
app.MachNumberMinEditField = uieditfield(app.GridLayout, 'numeric');
app.MachNumberMinEditField.HorizontalAlignment = 'center';
app.MachNumberMinEditField.Layout.Row = 1;
app.MachNumberMinEditField.Layout.Column = 2;
app.MachNumberMinEditField.Value = 1.6;

% Create P_0P_9EditFieldLabel
app.P_0P_9EditFieldLabel = uilabel(app.GridLayout);
app.P_0P_9EditFieldLabel.HorizontalAlignment = 'center';
app.P_0P_9EditFieldLabel.Layout.Row = 7;
app.P_0P_9EditFieldLabel.Layout.Column = 1;
app.P_0P_9EditFieldLabel.Text = 'P_0/P_9';

% Create P_0P_9EditField
app.P_0P_9EditField = uieditfield(app.GridLayout, 'numeric');
app.P_0P_9EditField.HorizontalAlignment = 'center';
app.P_0P_9EditField.Layout.Row = 7;
app.P_0P_9EditField.Layout.Column = 2;
app.P_0P_9EditField.Value = 1;

% Create P_0P_19EditFieldLabel
app.P_0P_19EditFieldLabel = uilabel(app.GridLayout);
app.P_0P_19EditFieldLabel.HorizontalAlignment = 'center';

```

```

app.P_0P_19EditFieldLabel.Layout.Row = 8;
app.P_0P_19EditFieldLabel.Layout.Column = 1;
app.P_0P_19EditFieldLabel.Text = 'P_0/P_19';

% Create P_0P_19EditField
app.P_0P_19EditField = uieditfield(app.GridLayout, 'numeric');
app.P_0P_19EditField.HorizontalAlignment = 'center';
app.P_0P_19EditField.Layout.Row = 8;
app.P_0P_19EditField.Layout.Column = 2;
app.P_0P_19EditField.Value = 1;

% Create gamma_cEditFieldLabel
app.gamma_cEditFieldLabel = uilabel(app.GridLayout);
app.gamma_cEditFieldLabel.HorizontalAlignment = 'center';
app.gamma_cEditFieldLabel.Layout.Row = 5;
app.gamma_cEditFieldLabel.Layout.Column = 1;
app.gamma_cEditFieldLabel.Text = 'gamma_c';

% Create gamma_cEditField
app.gamma_cEditField = uieditfield(app.GridLayout, 'numeric');
app.gamma_cEditField.HorizontalAlignment = 'center';
app.gamma_cEditField.Layout.Row = 5;
app.gamma_cEditField.Layout.Column = 2;
app.gamma_cEditField.Value = 1.4;

% Create c_pckJkgKLabel
app.c_pckJkgKLabel = uilabel(app.GridLayout);
app.c_pckJkgKLabel.HorizontalAlignment = 'center';
app.c_pckJkgKLabel.Layout.Row = 6;
app.c_pckJkgKLabel.Layout.Column = 1;
app.c_pckJkgKLabel.Text = 'c_pc (kJ/(kg.K))';

% Create c_pckJkgKEditField
app.c_pckJkgKEditField = uieditfield(app.GridLayout, 'numeric');
app.c_pckJkgKEditField.HorizontalAlignment = 'center';
app.c_pckJkgKEditField.Layout.Row = 6;
app.c_pckJkgKEditField.Layout.Column = 2;
app.c_pckJkgKEditField.Value = 1.0048;

% Create MaxMachNumberEditFieldLabel
app.MaxMachNumberEditFieldLabel = uilabel(app.GridLayout);
app.MaxMachNumberEditFieldLabel.HorizontalAlignment = 'center';
app.MaxMachNumberEditFieldLabel.Layout.Row = 3;
app.MaxMachNumberEditFieldLabel.Layout.Column = 1;
app.MaxMachNumberEditFieldLabel.Text = 'Max Mach Number';

% Create MaxMachNumberEditField
app.MaxMachNumberEditField = uieditfield(app.GridLayout, 'numeric');
app.MaxMachNumberEditField.HorizontalAlignment = 'center';
app.MaxMachNumberEditField.Layout.Row = 3;
app.MaxMachNumberEditField.Layout.Column = 2;
app.MaxMachNumberEditField.Value = 2.5;

% Create MachNoIncrementEditFieldLabel
app.MachNoIncrementEditFieldLabel = uilabel(app.GridLayout);
app.MachNoIncrementEditFieldLabel.HorizontalAlignment = 'center';

```

```

app.MachNoIncrementEditFieldLabel.Layout.Row = 2;
app.MachNoIncrementEditFieldLabel.Layout.Column = 1;
app.MachNoIncrementEditFieldLabel.Text = 'Mach No Increment';

% Create MachNoIncrementEditField
app.MachNoIncrementEditField = uieditfield(app.GridLayout, 'numeric');
app.MachNoIncrementEditField.HorizontalAlignment = 'center';
app.MachNoIncrementEditField.Layout.Row = 2;
app.MachNoIncrementEditField.Layout.Column = 2;
app.MachNoIncrementEditField.Value = 0.1;

% Create TabGroup
app.TabGroup = uitabgroup(app.AfterburnerTurbofan);
app.TabGroup.Position = [287 30 816 398];

% Create Table
app.Table = uitab(app.TabGroup);
app.Table.Title = 'Table';

% Create UITable
app.UITable =uitable(app.Table);
app.UITable.ColumnName = {'F/mdot'; 'S'; 'f'; 'f_AB'; 'f_DB'; 'eta_c'; 'eta_t';
'eta_f'};
app.UITable.RowName = {};
app.UITable.FontWeight = 'bold';
app.UITable.Position = [4 4 750 367];

% Create Fm_0dotTab
app.Fm_0dotTab = uitab(app.TabGroup);
app.Fm_0dotTab.Title = 'F/m_0dot';

% Create FovermdotPlot
app.FovermdotPlot = uiaxes(app.Fm_0dotTab);
title(app.FovermdotPlot, 'F/m_0dot')
xlabel(app.FovermdotPlot, 'pi_f')
ylabel(app.FovermdotPlot, 'F/m_0dot')
app.FovermdotPlot.Position = [59 18 651 352];

% Create STab
app.STab = uitab(app.TabGroup);
app.STab.Title = 'S';

% Create SPlot
app.SPlot = uiaxes(app.STab);
title(app.SPlot, 'Title')
xlabel(app.SPlot, 'X')
ylabel(app.SPlot, 'Y')
app.SPlot.Position = [64 17 641 347];

% Create FuelAirRatioTab
app.FuelAirRatioTab = uitab(app.TabGroup);
app.FuelAirRatioTab.Title = 'Fuel-Air Ratio';

% Create FuelPlot

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```

app.FuelPlot = uiaxes(app.FuelAirRatioTab);
title(app.FuelPlot, 'Title')
xlabel(app.FuelPlot, 'X')
ylabel(app.FuelPlot, 'Y')
app.FuelPlot.Position = [58 7 721 358];

% Create Eta_cTab
app.Eta_cTab = uitab(app.TabGroup);
app.Eta_cTab.Title = 'Eta_c';

% Create eta_cPlot
app.eta_cPlot = uiaxes(app.Eta_cTab);
title(app.eta_cPlot, 'Title')
xlabel(app.eta_cPlot, 'X')
ylabel(app.eta_cPlot, 'Y')
app.eta_cPlot.Position = [32 10 744 341];

% Create Eta_tTab
app.Eta_tTab = uitab(app.TabGroup);
app.Eta_tTab.Title = 'Eta_t';

% Create eta_tPlot
app.eta_tPlot = uiaxes(app.Eta_tTab);
title(app.eta_tPlot, 'Title')
xlabel(app.eta_tPlot, 'X')
ylabel(app.eta_tPlot, 'Y')
app.eta_tPlot.PlotBoxAspectRatio = [2.52747252747253 1 1];
app.eta_tPlot.YTick = [0 0.1 0.2 0.3 0.4 0.5 0.6 0.7 0.8 0.9 1];
app.eta_tPlot.Position = [44 23 719 325];

% Create PropulsiveEfficiencyTab
app.PropulsiveEfficiencyTab = uitab(app.TabGroup);
app.PropulsiveEfficiencyTab.Title = 'Propulsive Efficiency';

% Create GridLayout7
app.GridLayout7 = uigridlayout(app.PropulsiveEfficiencyTab);
app.GridLayout7.ColumnWidth = {'1x'};
app.GridLayout7.RowHeight = {'1x'};
app.GridLayout7.Padding = [10 11.004638671875 10 11.004638671875];

% Create eta_pPlot
app.eta_pPlot = uiaxes(app.GridLayout7);
title(app.eta_pPlot, 'Title')
xlabel(app.eta_pPlot, 'X')
ylabel(app.eta_pPlot, 'Y')
app.eta_pPlot.PlotBoxAspectRatio = [2.50815217391304 1 1];
app.eta_pPlot.Layout.Row = 1;
app.eta_pPlot.Layout.Column = 1;

% Create TermalEfficiencyTab
app.TermalEfficiencyTab = uitab(app.TabGroup);
app.TermalEfficiencyTab.Title = 'Termal Efficiency';

% Create eta_thermoPlot
app.eta_thermoPlot = uiaxes(app.TermalEfficiencyTab);

```



```

title(app.eta_thermoPlot, 'Title')
xlabel(app.eta_thermoPlot, 'X')
ylabel(app.eta_thermoPlot, 'Y')
app.eta_thermoPlot.Position = [17 12 788 346];

% Create OverallEfficiencyTab
app.OverallEfficiencyTab = uitab(app.TabGroup);
app.OverallEfficiencyTab.Title = 'Overall Efficiency';

% Create eta_oPlot
app.eta_oPlot = uiaxes(app.OverallEfficiencyTab);
title(app.eta_oPlot, 'Title')
xlabel(app.eta_oPlot, 'X')
ylabel(app.eta_oPlot, 'Y')
app.eta_oPlot.Position = [8 12 797 346];

% Create DesignConstraintsPanel
app.DesignConstraintsPanel = uipanel(app.AfterburnerTurbofan);
app.DesignConstraintsPanel.TitlePosition = 'centertop';
app.DesignConstraintsPanel.Title = 'Design Constraints';
app.DesignConstraintsPanel.FontWeight = 'bold';
app.DesignConstraintsPanel.Position = [286 428 215 267];

% Create GridLayout2
app.GridLayout2 = uigridlayout(app.DesignConstraintsPanel);
app.GridLayout2.RowHeight = {'1x', '1x', '1x', '1x', '1x', '1x', '1x', '1x',
'1x'};

app.GridLayout2.ColumnSpacing = 7.33443196614583;
app.GridLayout2.Padding = [7.33443196614583 10 7.33443196614583 10];

% Create MinT_t4KLabel
app.MinT_t4KLabel = uilabel(app.GridLayout2);
app.MinT_t4KLabel.HorizontalAlignment = 'center';
app.MinT_t4KLabel.Layout.Row = 1;
app.MinT_t4KLabel.Layout.Column = 1;
app.MinT_t4KLabel.Text = 'Min T_t4 (K)';

% Create MinT_t4KEditField
app.MinT_t4KEditField = uieditfield(app.GridLayout2, 'numeric');
app.MinT_t4KEditField.HorizontalAlignment = 'center';
app.MinT_t4KEditField.Layout.Row = 1;
app.MinT_t4KEditField.Layout.Column = 2;
app.MinT_t4KEditField.Value = 1944.4;

% Create MinT_t17KLabel
app.MinT_t17KLabel = uilabel(app.GridLayout2);
app.MinT_t17KLabel.HorizontalAlignment = 'center';
app.MinT_t17KLabel.Layout.Row = 7;
app.MinT_t17KLabel.Layout.Column = 1;
app.MinT_t17KLabel.Text = 'Min T_t17 (K)';

% Create MinT_t17KEditField
app.MinT_t17KEditField = uieditfield(app.GridLayout2, 'numeric');
app.MinT_t17KEditField.HorizontalAlignment = 'center';
app.MinT_t17KEditField.Layout.Row = 7;
app.MinT_t17KEditField.Layout.Column = 2;

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app.MinT_t17KEditField.Value = 2222.2;

% Create MinT_t7KLabel
app.MinT_t7KLabel = uilabel(app.GridLayout2);
app.MinT_t7KLabel.HorizontalAlignment = 'center';
app.MinT_t7KLabel.Layout.Row = 4;
app.MinT_t7KLabel.Layout.Column = 1;
app.MinT_t7KLabel.Text = 'Min T_t7 (K)';

% Create MinT_t7KEditField
app.MinT_t7KEditField = uieditfield(app.GridLayout2, 'numeric');
app.MinT_t7KEditField.HorizontalAlignment = 'center';
app.MinT_t7KEditField.Layout.Row = 4;
app.MinT_t7KEditField.Layout.Column = 2;
app.MinT_t7KEditField.Value = 2222.2;

% Create T_t4IncrementEditFieldLabel
app.T_t4IncrementEditFieldLabel = uilabel(app.GridLayout2);
app.T_t4IncrementEditFieldLabel.HorizontalAlignment = 'center';
app.T_t4IncrementEditFieldLabel.Layout.Row = 2;
app.T_t4IncrementEditFieldLabel.Layout.Column = 1;
app.T_t4IncrementEditFieldLabel.Text = 'T_t4 Increment';

% Create T_t4IncrementEditField
app.T_t4IncrementEditField = uieditfield(app.GridLayout2, 'numeric');
app.T_t4IncrementEditField.HorizontalAlignment = 'center';
app.T_t4IncrementEditField.Layout.Row = 2;
app.T_t4IncrementEditField.Layout.Column = 2;
app.T_t4IncrementEditField.Value = 50;

% Create MaxT_t4KEditFieldLabel
app.MaxT_t4KEditFieldLabel = uilabel(app.GridLayout2);
app.MaxT_t4KEditFieldLabel.HorizontalAlignment = 'center';
app.MaxT_t4KEditFieldLabel.Layout.Row = 3;
app.MaxT_t4KEditFieldLabel.Layout.Column = 1;
app.MaxT_t4KEditFieldLabel.Text = 'Max T_t4 (K)';

% Create MaxT_t4KEditField
app.MaxT_t4KEditField = uieditfield(app.GridLayout2, 'numeric');
app.MaxT_t4KEditField.HorizontalAlignment = 'center';
app.MaxT_t4KEditField.Layout.Row = 3;
app.MaxT_t4KEditField.Layout.Column = 2;
app.MaxT_t4KEditField.Value = 2100;

% Create T_t7IncrementEditFieldLabel
app.T_t7IncrementEditFieldLabel = uilabel(app.GridLayout2);
app.T_t7IncrementEditFieldLabel.HorizontalAlignment = 'right';
app.T_t7IncrementEditFieldLabel.Layout.Row = 5;
app.T_t7IncrementEditFieldLabel.Layout.Column = 1;
app.T_t7IncrementEditFieldLabel.Text = 'T_t7 Increment';

% Create T_t7IncrementEditField
app.T_t7IncrementEditField = uieditfield(app.GridLayout2, 'numeric');
app.T_t7IncrementEditField.HorizontalAlignment = 'center';
app.T_t7IncrementEditField.Layout.Row = 5;
app.T_t7IncrementEditField.Layout.Column = 2;

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app.T_t7IncrementEditField.Value = 50;

% Create MaxT_t7KEditFieldLabel
app.MaxT_t7KEditFieldLabel = uilabel(app.GridLayout2);
app.MaxT_t7KEditFieldLabel.HorizontalAlignment = 'center';
app.MaxT_t7KEditFieldLabel.Layout.Row = 6;
app.MaxT_t7KEditFieldLabel.Layout.Column = 1;
app.MaxT_t7KEditFieldLabel.Text = 'Max T_t7 (K)';

% Create MaxT_t7KEditField
app.MaxT_t7KEditField = uieditfield(app.GridLayout2, 'numeric');
app.MaxT_t7KEditField.HorizontalAlignment = 'center';
app.MaxT_t7KEditField.Layout.Row = 6;
app.MaxT_t7KEditField.Layout.Column = 2;
app.MaxT_t7KEditField.Value = 2350;

% Create T_t17IncrementEditFieldLabel
app.T_t17IncrementEditFieldLabel = uilabel(app.GridLayout2);
app.T_t17IncrementEditFieldLabel.HorizontalAlignment = 'center';
app.T_t17IncrementEditFieldLabel.Layout.Row = 8;
app.T_t17IncrementEditFieldLabel.Layout.Column = 1;
app.T_t17IncrementEditFieldLabel.Text = 'T_t17 Increment';

% Create T_t17IncrementEditField
app.T_t17IncrementEditField = uieditfield(app.GridLayout2, 'numeric');
app.T_t17IncrementEditField.HorizontalAlignment = 'center';
app.T_t17IncrementEditField.Layout.Row = 8;
app.T_t17IncrementEditField.Layout.Column = 2;
app.T_t17IncrementEditField.Value = 50;

% Create MaxT_t17KEditFieldLabel
app.MaxT_t17KEditFieldLabel = uilabel(app.GridLayout2);
app.MaxT_t17KEditFieldLabel.HorizontalAlignment = 'center';
app.MaxT_t17KEditFieldLabel.Layout.Row = 9;
app.MaxT_t17KEditFieldLabel.Layout.Column = 1;
app.MaxT_t17KEditFieldLabel.Text = 'Max T_t17 (K)';

% Create MaxT_t17KEditField
app.MaxT_t17KEditField = uieditfield(app.GridLayout2, 'numeric');
app.MaxT_t17KEditField.HorizontalAlignment = 'center';
app.MaxT_t17KEditField.Layout.Row = 9;
app.MaxT_t17KEditField.Layout.Column = 2;
app.MaxT_t17KEditField.Value = 2350;

% Create ComponentPerformancesPanel
app.ComponentPerformancesPanel = uipanel(app.AfterburnerTurbofan);
app.ComponentPerformancesPanel.TitlePosition = 'centertop';
app.ComponentPerformancesPanel.Title = 'Component Performances';
app.ComponentPerformancesPanel.FontWeight = 'bold';
app.ComponentPerformancesPanel.Position = [747 427 356 267];

% Create GridLayout3
app.GridLayout3 = uigridlayout(app.ComponentPerformancesPanel);
app.GridLayout3.ColumnWidth = {'1x', '1x', '1x', '1x'};
app.GridLayout3.RowHeight = {'1x', '1x', '1x', '1x', '1x', '1x', '1x'};

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% Create pi_bEditFieldLabel
app.pi_bEditFieldLabel = uilabel(app.GridLayout3);
app.pi_bEditFieldLabel.HorizontalAlignment = 'center';
app.pi_bEditFieldLabel.Layout.Row = 1;
app.pi_bEditFieldLabel.Layout.Column = 3;
app.pi_bEditFieldLabel.Text = 'pi_b';

% Create pi_bEditField
app.pi_bEditField = uieditfield(app.GridLayout3, 'numeric');
app.pi_bEditField.HorizontalAlignment = 'center';
app.pi_bEditField.Layout.Row = 1;
app.pi_bEditField.Layout.Column = 4;
app.pi_bEditField.Value = 0.98;

% Create eta_ABEditFieldLabel
app.eta_ABEditFieldLabel = uilabel(app.GridLayout3);
app.eta_ABEditFieldLabel.HorizontalAlignment = 'center';
app.eta_ABEditFieldLabel.Layout.Row = 6;
app.eta_ABEditFieldLabel.Layout.Column = 1;
app.eta_ABEditFieldLabel.Text = 'eta_AB';

% Create eta_ABEditField
app.eta_ABEditField = uieditfield(app.GridLayout3, 'numeric');
app.eta_ABEditField.HorizontalAlignment = 'center';
app.eta_ABEditField.Layout.Row = 6;
app.eta_ABEditField.Layout.Column = 2;
app.eta_ABEditField.Value = 0.95;

% Create eta_DBEditFieldLabel
app.eta_DBEditFieldLabel = uilabel(app.GridLayout3);
app.eta_DBEditFieldLabel.HorizontalAlignment = 'center';
app.eta_DBEditFieldLabel.Layout.Row = 6;
app.eta_DBEditFieldLabel.Layout.Column = 3;
app.eta_DBEditFieldLabel.Text = 'eta_DB';

% Create eta_DBEditField
app.eta_DBEditField = uieditfield(app.GridLayout3, 'numeric');
app.eta_DBEditField.HorizontalAlignment = 'center';
app.eta_DBEditField.Layout.Row = 6;
app.eta_DBEditField.Layout.Column = 4;
app.eta_DBEditField.Value = 0.95;

% Create eta_mEditFieldLabel
app.eta_mEditFieldLabel = uilabel(app.GridLayout3);
app.eta_mEditFieldLabel.HorizontalAlignment = 'center';
app.eta_mEditFieldLabel.Layout.Row = 7;
app.eta_mEditFieldLabel.Layout.Column = 1;
app.eta_mEditFieldLabel.Text = 'eta_m';

% Create eta_mEditField
app.eta_mEditField = uieditfield(app.GridLayout3, 'numeric');
app.eta_mEditField.HorizontalAlignment = 'center';
app.eta_mEditField.Layout.Row = 7;
app.eta_mEditField.Layout.Column = 2;
app.eta_mEditField.Value = 0.99;

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```

% Create eta_bEditFieldLabel
app.eta_bEditFieldLabel = uilabel(app.GridLayout3);
app.eta_bEditFieldLabel.HorizontalAlignment = 'center';
app.eta_bEditFieldLabel.Layout.Row = 5;
app.eta_bEditFieldLabel.Layout.Column = 3;
app.eta_bEditFieldLabel.Text = 'eta_b';

% Create eta_bEditField
app.eta_bEditField = uieditfield(app.GridLayout3, 'numeric');
app.eta_bEditField.HorizontalAlignment = 'center';
app.eta_bEditField.Layout.Row = 5;
app.eta_bEditField.Layout.Column = 4;
app.eta_bEditField.Value = 0.99;

% Create pi_DBEditFieldLabel
app.pi_DBEditFieldLabel = uilabel(app.GridLayout3);
app.pi_DBEditFieldLabel.HorizontalAlignment = 'center';
app.pi_DBEditFieldLabel.Layout.Row = 2;
app.pi_DBEditFieldLabel.Layout.Column = 3;
app.pi_DBEditFieldLabel.Text = 'pi_DB';

% Create pi_DBEditField
app.pi_DBEditField = uieditfield(app.GridLayout3, 'numeric');
app.pi_DBEditField.HorizontalAlignment = 'center';
app.pi_DBEditField.Layout.Row = 2;
app.pi_DBEditField.Layout.Column = 4;
app.pi_DBEditField.Value = 0.94;

% Create pi_d_maxEditFieldLabel
app.pi_d_maxEditFieldLabel = uilabel(app.GridLayout3);
app.pi_d_maxEditFieldLabel.HorizontalAlignment = 'center';
app.pi_d_maxEditFieldLabel.Layout.Row = 1;
app.pi_d_maxEditFieldLabel.Layout.Column = 1;
app.pi_d_maxEditFieldLabel.Text = 'pi_d_max';

% Create pi_d_maxEditField
app.pi_d_maxEditField = uieditfield(app.GridLayout3, 'numeric');
app.pi_d_maxEditField.HorizontalAlignment = 'center';
app.pi_d_maxEditField.Layout.Row = 1;
app.pi_d_maxEditField.Layout.Column = 2;
app.pi_d_maxEditField.Value = 0.98;

% Create e_cEditFieldLabel
app.e_cEditFieldLabel = uilabel(app.GridLayout3);
app.e_cEditFieldLabel.HorizontalAlignment = 'center';
app.e_cEditFieldLabel.Layout.Row = 4;
app.e_cEditFieldLabel.Layout.Column = 1;
app.e_cEditFieldLabel.Text = 'e_c';

% Create e_cEditField
app.e_cEditField = uieditfield(app.GridLayout3, 'numeric');
app.e_cEditField.HorizontalAlignment = 'center';
app.e_cEditField.Layout.Row = 4;
app.e_cEditField.Layout.Column = 2;
app.e_cEditField.Value = 0.9;

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% Create e_tEditFieldLabel
app.e_tEditFieldLabel = uilabel(app.GridLayout3);
app.e_tEditFieldLabel.HorizontalAlignment = 'center';
app.e_tEditFieldLabel.Layout.Row = 5;
app.e_tEditFieldLabel.Layout.Column = 1;
app.e_tEditFieldLabel.Text = 'e_t';

% Create e_tEditField
app.e_tEditField = uieditfield(app.GridLayout3, 'numeric');
app.e_tEditField.HorizontalAlignment = 'center';
app.e_tEditField.Layout.Row = 5;
app.e_tEditField.Layout.Column = 2;
app.e_tEditField.Value = 0.91;

% Create e_fEditFieldLabel
app.e_fEditFieldLabel = uilabel(app.GridLayout3);
app.e_fEditFieldLabel.HorizontalAlignment = 'center';
app.e_fEditFieldLabel.Layout.Row = 4;
app.e_fEditFieldLabel.Layout.Column = 3;
app.e_fEditFieldLabel.Text = 'e_f';

% Create e_fEditField
app.e_fEditField = uieditfield(app.GridLayout3, 'numeric');
app.e_fEditField.HorizontalAlignment = 'center';
app.e_fEditField.Layout.Row = 4;
app.e_fEditField.Layout.Column = 4;
app.e_fEditField.Value = 0.89;

% Create pi_fnEditFieldLabel
app.pi_fnEditFieldLabel = uilabel(app.GridLayout3);
app.pi_fnEditFieldLabel.HorizontalAlignment = 'center';
app.pi_fnEditFieldLabel.Layout.Row = 3;
app.pi_fnEditFieldLabel.Layout.Column = 3;
app.pi_fnEditFieldLabel.Text = 'pi_fn';

% Create pi_fnEditField
app.pi_fnEditField = uieditfield(app.GridLayout3, 'numeric');
app.pi_fnEditField.HorizontalAlignment = 'center';
app.pi_fnEditField.Layout.Row = 3;
app.pi_fnEditField.Layout.Column = 4;
app.pi_fnEditField.Value = 0.98;

% Create pi_nEditFieldLabel
app.pi_nEditFieldLabel = uilabel(app.GridLayout3);
app.pi_nEditFieldLabel.HorizontalAlignment = 'center';
app.pi_nEditFieldLabel.Layout.Row = 3;
app.pi_nEditFieldLabel.Layout.Column = 1;
app.pi_nEditFieldLabel.Text = 'pi_n';

% Create pi_nEditField
app.pi_nEditField = uieditfield(app.GridLayout3, 'numeric');
app.pi_nEditField.HorizontalAlignment = 'center';
app.pi_nEditField.Layout.Row = 3;
app.pi_nEditField.Layout.Column = 2;
app.pi_nEditField.Value = 0.98;

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% Create pi_ABEditFieldLabel
app.pi_ABEditFieldLabel = uilabel(app.GridLayout3);
app.pi_ABEditFieldLabel.HorizontalAlignment = 'center';
app.pi_ABEditFieldLabel.Layout.Row = 2;
app.pi_ABEditFieldLabel.Layout.Column = 1;
app.pi_ABEditFieldLabel.Text = 'pi_AB';

% Create pi_ABEditField
app.pi_ABEditField = uieditfield(app.GridLayout3, 'numeric');
app.pi_ABEditField.HorizontalAlignment = 'center';
app.pi_ABEditField.Layout.Row = 2;
app.pi_ABEditField.Layout.Column = 2;
app.pi_ABEditField.Value = 0.94;

% Create CaseSelectionPanel
app.CaseSelectionPanel = uipanel(app.AfterburnerTurbofan);
app.CaseSelectionPanel.TitlePosition = 'centertop';
app.CaseSelectionPanel.Title = 'Case Selection';
app.CaseSelectionPanel.FontWeight = 'bold';
app.CaseSelectionPanel.Position = [22 640 263 55];

% Create CaseDropDownLabel
app.CaseDropDownLabel = uilabel(app.CaseSelectionPanel);
app.CaseDropDownLabel.HorizontalAlignment = 'right';
app.CaseDropDownLabel.Position = [1 9 34 22];
app.CaseDropDownLabel.Text = 'Case';

% Create CaseDropDown
app.CaseDropDown = uiddropdown(app.CaseSelectionPanel);
app.CaseDropDown.Items = {'Single Condition', 'Varying Bypass and pi_f',
'Varying pi_f and pi_c', 'Varying Bypass and pi_c', 'Varying Mach No and pi_c', 'Varying
Mach No and Bypass', 'Varying Mach No and pi_f', 'Varying Mach No and T_t4', 'Varying Mach
No and T_t7', 'Varying Mach No and T_t17'};
app.CaseDropDown.ValueChangedFcn = createCallbackFcn(app,
@CaseDropDownValueChanged, true);
app.CaseDropDown.Position = [49 9 201 22];
app.CaseDropDown.Value = 'Single Condition';

% Create DesignChoicesPanel
app.DesignChoicesPanel = uipanel(app.AfterburnerTurbofan);
app.DesignChoicesPanel.TitlePosition = 'centertop';
app.DesignChoicesPanel.Title = 'Design Choices';
app.DesignChoicesPanel.FontWeight = 'bold';
app.DesignChoicesPanel.Position = [20 31 265 344];

% Create GridLayout4
app.GridLayout4 = uigridlayout(app.DesignChoicesPanel);
app.GridLayout4.ColumnWidth = {'100x', '100x'};
app.GridLayout4.RowHeight = {'1x', '1x', '1x', '1x', '1x', '1x', '1x', '1x',
'1x', '1x', '1x'};

% Create ComputeButton
app.ComputeButton = uibutton(app.GridLayout4, 'push');
app.ComputeButton.ButtonPushedFcn = createCallbackFcn(app,
@ComputeButtonPushed, true);
app.ComputeButton.Layout.Row = 11;
app.ComputeButton.Layout.Column = [1 2];

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app.ComputeButton.Text = 'Compute';

% Create Minpi_cValueLabel
app.Minpi_cValueLabel = uilabel(app.GridLayout4);
app.Minpi_cValueLabel.HorizontalAlignment = 'center';
app.Minpi_cValueLabel.Layout.Row = 1;
app.Minpi_cValueLabel.Layout.Column = 1;
app.Minpi_cValueLabel.Text = 'Min pi_c Value';

% Create Minpi_cValueEditField
app.Minpi_cValueEditField = uieditfield(app.GridLayout4, 'numeric');
app.Minpi_cValueEditField.HorizontalAlignment = 'center';
app.Minpi_cValueEditField.Layout.Row = 1;
app.Minpi_cValueEditField.Layout.Column = 2;
app.Minpi_cValueEditField.Value = 22;

% Create Minpi_fValueEditFieldLabel
app.Minpi_fValueEditFieldLabel = uilabel(app.GridLayout4);
app.Minpi_fValueEditFieldLabel.HorizontalAlignment = 'center';
app.Minpi_fValueEditFieldLabel.Layout.Row = 4;
app.Minpi_fValueEditFieldLabel.Layout.Column = 1;
app.Minpi_fValueEditFieldLabel.Text = 'Min pi_f Value';

% Create Minpi_fEditField
app.Minpi_fEditField = uieditfield(app.GridLayout4, 'numeric');
app.Minpi_fEditField.HorizontalAlignment = 'center';
app.Minpi_fEditField.Layout.Row = 4;
app.Minpi_fEditField.Layout.Column = 2;
app.Minpi_fEditField.Value = 2;

% Create Maxpi_fValueEditFieldLabel
app.Maxpi_fValueEditFieldLabel = uilabel(app.GridLayout4);
app.Maxpi_fValueEditFieldLabel.HorizontalAlignment = 'center';
app.Maxpi_fValueEditFieldLabel.Layout.Row = 6;
app.Maxpi_fValueEditFieldLabel.Layout.Column = 1;
app.Maxpi_fValueEditFieldLabel.Text = 'Max pi_f Value';

% Create Maxpi_fEditField
app.Maxpi_fEditField = uieditfield(app.GridLayout4, 'numeric');
app.Maxpi_fEditField.HorizontalAlignment = 'center';
app.Maxpi_fEditField.Layout.Row = 6;
app.Maxpi_fEditField.Layout.Column = 2;
app.Maxpi_fEditField.Value = 5;

% Create MinByPassRatioValueEditFieldLabel
app.MinByPassRatioValueEditFieldLabel = uilabel(app.GridLayout4);
app.MinByPassRatioValueEditFieldLabel.HorizontalAlignment = 'center';
app.MinByPassRatioValueEditFieldLabel.Layout.Row = 7;
app.MinByPassRatioValueEditFieldLabel.Layout.Column = 1;
app.MinByPassRatioValueEditFieldLabel.Text = 'Min By-Pass Ratio Value';

% Create MinByPassRatioEditField
app.MinByPassRatioEditField = uieditfield(app.GridLayout4, 'numeric');
app.MinByPassRatioEditField.HorizontalAlignment = 'center';
app.MinByPassRatioEditField.Layout.Row = 7;
app.MinByPassRatioEditField.Layout.Column = 2;

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app.MinByPassRatioEditField.Value = 0.5;

% Create MaxByPassRatioEditFieldLabel
app.MaxByPassRatioEditFieldLabel = uilabel(app.GridLayout4);
app.MaxByPassRatioEditFieldLabel.HorizontalAlignment = 'center';
app.MaxByPassRatioEditFieldLabel.Layout.Row = 9;
app.MaxByPassRatioEditFieldLabel.Layout.Column = 1;
app.MaxByPassRatioEditFieldLabel.Text = 'Max By-Pass Ratio';

% Create MaxByPassRatioEditField
app.MaxByPassRatioEditField = uieditfield(app.GridLayout4, 'numeric');
app.MaxByPassRatioEditField.HorizontalAlignment = 'center';
app.MaxByPassRatioEditField.Layout.Row = 9;
app.MaxByPassRatioEditField.Layout.Column = 2;
app.MaxByPassRatioEditField.Value = 3;

% Create pi_fIncrementEditFieldLabel
app.pi_fIncrementEditFieldLabel = uilabel(app.GridLayout4);
app.pi_fIncrementEditFieldLabel.HorizontalAlignment = 'center';
app.pi_fIncrementEditFieldLabel.Layout.Row = 5;
app.pi_fIncrementEditFieldLabel.Layout.Column = 1;
app.pi_fIncrementEditFieldLabel.Text = 'pi_f Increment';

% Create pi_fIncrementEditField
app.pi_fIncrementEditField = uieditfield(app.GridLayout4, 'numeric');
app.pi_fIncrementEditField.HorizontalAlignment = 'center';
app.pi_fIncrementEditField.Layout.Row = 5;
app.pi_fIncrementEditField.Layout.Column = 2;
app.pi_fIncrementEditField.Value = 0.2;

% Create ByPassRatioIncrementEditFieldLabel
app.ByPassRatioIncrementEditFieldLabel = uilabel(app.GridLayout4);
app.ByPassRatioIncrementEditFieldLabel.HorizontalAlignment = 'center';
app.ByPassRatioIncrementEditFieldLabel.Layout.Row = 8;
app.ByPassRatioIncrementEditFieldLabel.Layout.Column = 1;
app.ByPassRatioIncrementEditFieldLabel.Text = 'By-Pass Ratio Increment';

% Create ByPassRatioIncrementEditField
app.ByPassRatioIncrementEditField = uieditfield(app.GridLayout4, 'numeric');
app.ByPassRatioIncrementEditField.HorizontalAlignment = 'center';
app.ByPassRatioIncrementEditField.Layout.Row = 8;
app.ByPassRatioIncrementEditField.Layout.Column = 2;
app.ByPassRatioIncrementEditField.Value = 0.5;

% Create pi_cIncrementEditField_2Label
app.pi_cIncrementEditField_2Label = uilabel(app.GridLayout4);
app.pi_cIncrementEditField_2Label.HorizontalAlignment = 'center';
app.pi_cIncrementEditField_2Label.Layout.Row = 2;
app.pi_cIncrementEditField_2Label.Layout.Column = 1;
app.pi_cIncrementEditField_2Label.Text = 'pi_c Increment';

% Create pi_cIncrementEditField
app.pi_cIncrementEditField = uieditfield(app.GridLayout4, 'numeric');
app.pi_cIncrementEditField.HorizontalAlignment = 'center';
app.pi_cIncrementEditField.Layout.Row = 2;
app.pi_cIncrementEditField.Layout.Column = 2;

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app.pi_cIncrementEditField.Value = 1;

% Create Maxpi_cValueEditField_2Label
app.Maxpi_cValueEditField_2Label = uilabel(app.GridLayout4);
app.Maxpi_cValueEditField_2Label.HorizontalAlignment = 'center';
app.Maxpi_cValueEditField_2Label.Layout.Row = 3;
app.Maxpi_cValueEditField_2Label.Layout.Column = 1;
app.Maxpi_cValueEditField_2Label.Text = 'Max pi_c Value';

% Create Maxpi_cValueEditField
app.Maxpi_cValueEditField = uieditfield(app.GridLayout4, 'numeric');
app.Maxpi_cValueEditField.HorizontalAlignment = 'center';
app.Maxpi_cValueEditField.Layout.Row = 3;
app.Maxpi_cValueEditField.Layout.Column = 2;
app.Maxpi_cValueEditField.Value = 30;

% Create XAxisSwitchLabel
app.XAxisSwitchLabel = uilabel(app.GridLayout4);
app.XAxisSwitchLabel.HorizontalAlignment = 'center';
app.XAxisSwitchLabel.Layout.Row = 10;
app.XAxisSwitchLabel.Layout.Column = 1;
app.XAxisSwitchLabel.Text = 'X Axis';

% Create XAxisSwitch
app.XAxisSwitch = uiswitch(app.GridLayout4, 'slider');
app.XAxisSwitch.Items = {'alpha', 'pi_f'};
app.XAxisSwitch.Layout.Row = 10;
app.XAxisSwitch.Layout.Column = 2;
app.XAxisSwitch.Value = 'alpha';

% Create GasPropertiesPanel
app.GasPropertiesPanel = uipanel(app.AfterburnerTurbofan);
app.GasPropertiesPanel.TitlePosition = 'centertop';
app.GasPropertiesPanel.Title = 'Gas Properties';
app.GasPropertiesPanel.FontWeight = 'bold';
app.GasPropertiesPanel.Position = [500 495 248 200];

% Create GridLayout5
app.GridLayout5 = uigridlayout(app.GasPropertiesPanel);
app.GridLayout5.RowHeight = {'1x', '1x', '1x', '1x', '1x', '1x'};

% Create c_ptkJkgKEditFieldLabel
app.c_ptkJkgKEditFieldLabel = uilabel(app.GridLayout5);
app.c_ptkJkgKEditFieldLabel.HorizontalAlignment = 'center';
app.c_ptkJkgKEditFieldLabel.Layout.Row = 2;
app.c_ptkJkgKEditFieldLabel.Layout.Column = 1;
app.c_ptkJkgKEditFieldLabel.Text = 'c_pt (kJ/(kg.K))';

% Create c_ptkJkgKEditField
app.c_ptkJkgKEditField = uieditfield(app.GridLayout5, 'numeric');
app.c_ptkJkgKEditField.HorizontalAlignment = 'center';
app.c_ptkJkgKEditField.Layout.Row = 2;
app.c_ptkJkgKEditField.Layout.Column = 2;
app.c_ptkJkgKEditField.Value = 1.2351;

```

```

% Create c_pABkJkgKLabel
app.c_pABkJkgKLabel = uilabel(app.GridLayout5);
app.c_pABkJkgKLabel.HorizontalAlignment = 'center';
app.c_pABkJkgKLabel.Layout.Row = 4;
app.c_pABkJkgKLabel.Layout.Column = 1;
app.c_pABkJkgKLabel.Text = 'c_pAB (kJ/(kg.K))';

% Create c_pABkJkgKEditField
app.c_pABkJkgKEditField = uieditfield(app.GridLayout5, 'numeric');
app.c_pABkJkgKEditField.HorizontalAlignment = 'center';
app.c_pABkJkgKEditField.Layout.Row = 4;
app.c_pABkJkgKEditField.Layout.Column = 2;
app.c_pABkJkgKEditField.Value = 1.2351;

% Create c_pDBkJkgKEditFieldLabel
app.c_pDBkJkgKEditFieldLabel = uilabel(app.GridLayout5);
app.c_pDBkJkgKEditFieldLabel.HorizontalAlignment = 'center';
app.c_pDBkJkgKEditFieldLabel.Layout.Row = 6;
app.c_pDBkJkgKEditFieldLabel.Layout.Column = 1;
app.c_pDBkJkgKEditFieldLabel.Text = 'c_pDB (kJ/(kg.K))';

% Create c_pDBkJkgKEditField
app.c_pDBkJkgKEditField = uieditfield(app.GridLayout5, 'numeric');
app.c_pDBkJkgKEditField.HorizontalAlignment = 'center';
app.c_pDBkJkgKEditField.Layout.Row = 6;
app.c_pDBkJkgKEditField.Layout.Column = 2;
app.c_pDBkJkgKEditField.Value = 1.2351;

% Create gamma_ABEditFieldLabel
app.gamma_ABEditFieldLabel = uilabel(app.GridLayout5);
app.gamma_ABEditFieldLabel.HorizontalAlignment = 'center';
app.gamma_ABEditFieldLabel.Layout.Row = 3;
app.gamma_ABEditFieldLabel.Layout.Column = 1;
app.gamma_ABEditFieldLabel.Text = 'gamma_AB';

% Create gamma_ABEditField
app.gamma_ABEditField = uieditfield(app.GridLayout5, 'numeric');
app.gamma_ABEditField.HorizontalAlignment = 'center';
app.gamma_ABEditField.Layout.Row = 3;
app.gamma_ABEditField.Layout.Column = 2;
app.gamma_ABEditField.Value = 1.3;

% Create gamma_tEditFieldLabel
app.gamma_tEditFieldLabel = uilabel(app.GridLayout5);
app.gamma_tEditFieldLabel.HorizontalAlignment = 'center';
app.gamma_tEditFieldLabel.Layout.Row = 1;
app.gamma_tEditFieldLabel.Layout.Column = 1;
app.gamma_tEditFieldLabel.Text = 'gamma_t';

% Create gamma_tEditField
app.gamma_tEditField = uieditfield(app.GridLayout5, 'numeric');
app.gamma_tEditField.HorizontalAlignment = 'center';
app.gamma_tEditField.Layout.Row = 1;
app.gamma_tEditField.Layout.Column = 2;
app.gamma_tEditField.Value = 1.3;

```

```

% Create gamma_DBEditFieldLabel
app.gamma_DBEditFieldLabel = uilabel(app.GridLayout5);
app.gamma_DBEditFieldLabel.HorizontalAlignment = 'center';
app.gamma_DBEditFieldLabel.Layout.Row = 5;
app.gamma_DBEditFieldLabel.Layout.Column = 1;
app.gamma_DBEditFieldLabel.Text = 'gamma_DB';

% Create gamma_DBEditField
app.gamma_DBEditField = uieditfield(app.GridLayout5, 'numeric');
app.gamma_DBEditField.HorizontalAlignment = 'center';
app.gamma_DBEditField.Layout.Row = 5;
app.gamma_DBEditField.Layout.Column = 2;
app.gamma_DBEditField.Value = 1.3;

% Create FuelPropertiesPanel
app.FuelPropertiesPanel = uipanel(app.AfterburnerTurbofan);
app.FuelPropertiesPanel.TitlePosition = 'centertop';
app.FuelPropertiesPanel.Title = 'Fuel Properties';
app.FuelPropertiesPanel.FontWeight = 'bold';
app.FuelPropertiesPanel.Position = [501 429 246 64];

% Create GridLayout6
app.GridLayout6 = uigridlayout(app.FuelPropertiesPanel);
app.GridLayout6.RowHeight = {'1x'};

% Create h_PRkJkgEditFieldLabel
app.h_PRkJkgEditFieldLabel = uilabel(app.GridLayout6);
app.h_PRkJkgEditFieldLabel.HorizontalAlignment = 'center';
app.h_PRkJkgEditFieldLabel.Layout.Row = 1;
app.h_PRkJkgEditFieldLabel.Layout.Column = 1;
app.h_PRkJkgEditFieldLabel.Text = 'h_PR (kJ/kg)';

% Create h_PRkJkgEditField
app.h_PRkJkgEditField = uieditfield(app.GridLayout6, 'numeric');
app.h_PRkJkgEditField.HorizontalAlignment = 'center';
app.h_PRkJkgEditField.Layout.Row = 1;
app.h_PRkJkgEditField.Layout.Column = 2;
app.h_PRkJkgEditField.Value = 42798;

% Show the figure after all components are created
app.AfterburnerTurbofan.Visible = 'on';
end
end

% App creation and deletion
methods (Access = public)

% Construct app
function app = SSATFv2_5

% Create UIFigure and components
createComponents(app)

% Register the app with App Designer

```

```

registerApp(app, app.AfterburnerTurbofan)

% Execute the startup function
runStartupFcn(app, @startupFcn)

    if nargin == 0
        clear app
    end
end

% Code that executes before app deletion
function delete(app)

    % Delete UIFigure when app is deleted
    delete(app.AfterburnerTurbofan)
end
end
end

```