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PERFORMANCE AND LOADS DATA FROM A HOVER TEST OF A FULL-SCALE XV-15 ROTOR

Fort F. Felker, Mark D. Betzina, and David B. Signor Ames Research Center

SUMMARY

A hover test of a full-scale XV-15 rotor was conducted at the Outdoor Aerodynamic Research Facility at Ames Research Center. The primary objective of the test was to obtain accurate measurements of the hover performance of the original, metal-blade XV-15 rotor system. Data were acquired for rotor tip Mach numbers ranging from 0.60 to 0.73. This report presents data on rotor performance, rotor wake downwash velocities, and rotor loads.

NOMENCLATURE

\boldsymbol{A}	rotor disc area, πR^2 , m ²
a	speed of sound, m/s
C_{P}	rotor power coefficient, $C_P = C_Q$
$C_{P,corrected}$	rotor power coefficient corrected for wind, $C_{P,corrected} = C_{Q,corrected}$
C_{PM}	rotor pitching moment coefficient, pitching moment $/ ho ARV_{tip}^{2}$
$C_{\mathcal{O}}$	rotor torque coefficient, torque/ $ ho ARV_{tip}^2$
CQ, corrected	rotor torque coefficient corrected for wind, See text for equations
C_T	rotor thrust coefficient, thrust $/\rho AV_{tip}^2$
C_{Y}	rotor side force coefficient, side force $/\rho AV_{tip}^2$
C_{YM}	rotor yawing moment coefficient, yawing moment $/\rho ARV_{tip}^2$
C_{Z}	rotor normal force coefficient, normal force $ ho AV_{tip}^2$
FM .	rotor figure of merit, $C_T^{3/2}/C_Q\sqrt{2}$
FM corrected	rotor figure of merit corrected for wind, $C_T^{3/2}/C_{Q,corrected}\sqrt{2}$
M_{tip}	rotor tip Mach number, V_{tip}/a
q	dynamic pressure, $\rho V^2/2$, N/m^2
R	rotor radius, m

<i>r</i>	blade radial station, m
V_h	ideal induced hover velocity, $V_{tip}\sqrt{C_T/2}$, m/s
$V_{\mathfrak{t}}$	ideal induced velocity, m/s
V_{tip}	rotor tip speed, ΩR , m/s
V_w	wind speed, m/s
λ_h	ideal induced hover velocity ratio, V_h/V_{tip}
λ_i	ideal induced velocity ratio, V_i/V_{tip}
μ_y	lateral wind velocity ratio, $-V_w \sin \psi_w/V_{tip}$
μ_x	axial wind velocity ratio, $V_w \cos \psi_w / V_{tip}$
ρ	air density, kg/m ³
σ	rotor solidity ratio
$oldsymbol{\psi_{oldsymbol{w}}}$	wind direction relative to rotor axis
Ω	rotor rotation speed, radians/sec

INTRODUCTION

Hovering flight is a critical operating condition for VTOL aircraft, since the hover performance usually determines the aircraft's maximum payload. The payload is typically 30% of the aircraft's gross weight, and small changes in the hover performance can have a large effect on the size of the payload. Hover performance is particularly important for tilt-rotors, since their basic rotor design (disc loading, solidity ratio, etc.) is a compromise between the requirements of hovering and cruise flight. Analytical predictions of tilt-rotor hover performance have not been sufficiently validated to provide a high level of confidence in the predicted performance.

An experimental investigation was recently conducted at Ames Research Center to accurately measure the hover performance of three tilting prop-rotors (ref. 1). The rotors tested in this investigation were the original metal blades for the XV-15 Tilt Rotor Research Aircraft; a set of composite, Advanced Technology Blades (ATB) for the XV-15; and a 0.658-scale model of the proposed V-22A Osprey (JVX) rotor. All rotors had three blades, and had a diameter of 7.62 m.

This report presents the data obtained with the XV-15 metal blades. Data is presented on rotor aerodynamic forces and moments, rotor wake downwash velocities, and rotor loads.

The authors gratefully acknowledge the efforts of the many people at Ames Research Center, Boeing Vertol Co., and Bell Helicopter, Textron, who made this test possible. Thanks are also due to Rob Faye, for his assistance in the preparation of this report.

DESCRIPTION OF TEST APPARATUS

L. T. Carrier M. Later M.

Outdoor Aerodynamic Research Facility

The test was conducted at the Ames Outdoor Aerogynamic Research Facility, which consists of a 30 m square concrete pad, a below-ground-level frame for attaching model support struts, and an underground control room with a complete data acquisition system. The facility is sufficiently remote from other buildings so that there is no aerodynamic interference (except with the ground), and accurate near- and far-field acoustic data can be obtained. An aerial photograph of the Outdoor Aerodynamic Research Facility with the Prop Test Rig installed is shown in figure 1.

Prop Test Rig

The Ames Prop Test Rig was used to power the rotors, with a maximum power output of 1864 kW at 625 rotor RPM. A three-view drawing of the Prop Test Rig with the XV-15 rotor system installed is shown in figure 2, and a photograph of the Prop Test Rig with the XV-15 rotor installed is shown in figure 3. The rotor axis of rotation was horizontal to minimize interference effects between the ground and the rotor. The rotor shaft was 6.71 m above the ground (1.76 rotor radii). Note that the Prop Test Rig and its supporting structure provide very little blockage of the rotor wake. This minimizes the influence of the test apparatus on the rotor wake, and ensures that high-quality isolated-rotor performance data can be acquired.

Balance Systems

A new rotor balance system was designed and built for this test program. The general arrangement of the balance system is shown in figure 4. This balance system was designed to be very sensitive to rotor thrust and torque, with minimal interactions on the indicated thrust and torque caused by other forces, moments, or thermal effects. An instrumented drive shaft was installed inside the rotor balance, between the gearbox and the rotor mast, to accurately measure shaft torque. This design provided two load paths for thrust: through the rotor balance, and through the instrumented drive shaft. The drive shaft was not as stiff in the axial direction as the rotor balance, and only about 3% of the rotor thrust was carried by the shaft. The shaft was instrumented to measure this axial load. The gages on the balance system were thermally-compensated to minimize errors which were due to thermal effects. (The rotor balance and instrumented drive shaft were designed by J. Mayer and H. Silcox of the Boeing Vertol Co.)

Careful laboratory calibrations were performed on the balance system. The rotor

thrust balance was accurate to within 50 N up to 50,000 N (0.1% error), with no significant interactions caused by other forces or moments. The shaft axial force gage was also accurate to within 50 N, and was corrected for interactions caused by shaft torque. The instrumented drive shaft was accurate to within 70 N-m of torque, which is less than 0.3% of the shaft's maximum capacity of 28,500 N-m. The shaft torque data were corrected for interactions caused by shaft axial load. Because there were two bearings between the instrumented drive shaft and the rotor, the bearing torque was measured by the rotor balance and subtracted from the shaft torque to obtain the actual rotor torque.

A redundant set of load cells were installed between the Prop Test Rig and its support system (see fig. 2). These loads cells were not as accurate as the primary balance system, and were used as a backup. The measurements of the two balance systems were compared throughout the test to ensure that both systems were working properly at all times.

Check loads were performed periodically during the test to assess installed balance system accuracy under simultaneous thrust and torque loading, and to check for adverse effects caused by operational thermal loads. These check loadings demonstrated that the installed balance system was accurate to within 200 N of thrust (0.3% of maximum thrust of test) and 70 N-m of torque (0.3% of maximum torque of test).

Rotor System

The rotor was tested on a Bell Helicopter Model 300 rotor mast and gimballed hub (similar to the XV-15 aircraft's mast and hub). The XV-15 rotor system had three blades with a diameter of 7.62 m. A summary of the rotor system characteristics is provided in table 1. The rotor blades were identical to the flight hardware used on the XV-15 aircraft. This rotor system had a solidity ratio of 0.0891. The twist distribution, thickness distribution, chord distribution, and airfoils used on this rotor system are shown in figures 5, 6, 7, and 8, respectively. Further information on the characteristics of this rotor system are provided in ref. 2.

Wake Rake

The distribution of total pressure and static pressure in the rotor wake was measured with a wake rake. The location of the rake relative to the rotor was chosen to be similar to the location of the wing of a typical tilt-rotor aircraft. The wake rake is visible behind the rotor in figure 3. The dynamic pressure and velocity distributions in the rotor wake were computed from the total and static pressure data. Two types of pressure probes were used on the wake rake: pitot-static probes, and 5-port directional probes. There were 13 pitot-static probes and 9 directional probes. The static pressure data obtained with the pitot-static probes are more accurate than that obtained with the directional probes.

Therefore, the dynamic pressures and velocities computed from data obtained with the pitot-static probes are more accurate than those computed from data obtained with the directional probes. Data obtained with both sets of probes are presented in this report.

TEST CONDITIONS

Data were obtained with rotor tip Mach numbers ranging from 0.6 to 0.73. Cyclic pitch was used to trim the rotor to gimbal angles of 0.1° or less for all data points. Most of the data were obtained with winds of 1.5 m/s or less, with a maximum wind speed of 3.5 m/s. The air density was computed from measured values of temperature, pressure, and humidity. A phototach was driven at the rotor speed and generate 1,024 pulses per revolution. The rotor rotation speed was computed from this signal.

WIND CORRECTIONS

Even very light winds can have significant effects on rotor hover performance (ref. 3). To minimize errors in the performance data caused by winds, all performance testing was conducted in winds of 1.5 m/s or less. Also, the measured rotor torque was corrected for the effect of the wind using a correction procedure based on momentum theory. (The correction procedure was developed by W. Johnson of Ames Research Center and M. A. McVeigh of Boeing Vertol.) The wind speed and direction were measured by a sensor located on the inflow side of the rotor plane approximately 16 rotor radii from the rotor hub at the same height as the rotor axis, and at an angle of 45° from the rotor axis. The location of the wind sensor relative to the rotor, and the sign conventions for the wind speed and direction are shown in figure 9. The following equations describe the wind correction procedure that was used:

$$C_{Q,corrected} = C_Q + (\mu_x C_T + \mu_y C_Y) - K(\lambda_i - \lambda_h) C_T$$

$$\lambda_i^2 (\mu_y^2 + (\lambda_i - \mu_x)^2) = \lambda_h^4$$

Note that μ_y is positive in the same direction as C_Y , and μ_x is positive in the same direction as C_T . K is the ratio of actual induced power to ideal induced power: 1.16 was used here.

The magnitude of the correction on C_Q was typically less than 3% for winds of less than 1.5 m/s. The correction procedure reduces scatter in the performance data caused

by winds varying from data point to data point, and reduces any bias in the performance data caused by consistent prevailing winds throughout the test. Rotor figure of merit as a function of thrust coefficient for the XV-15 rotor system, with and without wind corrections, is shown in figure 10. Data obtained with winds of 0.5 m/s or less are presented in figure 10(a), data obtained with winds of 1.5 m/s or less are presented in figure 10(b), and all the data are shown in figure 10(c). The reduction in data scatter caused by the wind corrections can be seen in these figures. Both corrected and uncorrected data are presented in this report.

RESULTS

Performance and Loads Data

Rotor performance and loads data are tabulated in Appendix A. A dictionary of the parameters in Appendix A is provided in table 2. The data are organized by run number, and an index of the test conditions in each run is provided in table 3. Figure 11 shows the orientation of balance forces and moments, and the positive directions of the forces and moments. Thrust and side force are horizontal, and normal force is vertical.

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The effect of tip Mach number on corrected rotor figure of merit is shown in figures 12 and 13. $C_{P,corrected}/\sigma$ as a function of C_T/σ is shown in figure 14. $C_{P,corrected}$ as a function of C_T is shown in figure 15. $C_{P,corrected}$ as a function of $C_T^{3/2}$ is shown in figure 16. The curves shown in these figures are polynomial curve fits of the data. There is very little effect on rotor performance due to tip Mach number variations for the range of tip Mach numbers covered in this test.

 C_T/σ as a function of collective pitch is shown in figure 17. The collective pitch data were obtained from the collective actuator position, and some errors caused by control system geometric nonlinearities are present in the data. These errors are estimated to be less than $\pm 1^{\circ}$. The effect of rotor thrust on hub spindle flap bending moment is shown in figure 18. The hub spindle flap bending moment gage was at r/R = 0.06. The effect of rotor thrust on blade flap bending moment at 0.3R is shown in figure 19. The effect of rotor thrust on pitch link load is shown in figure 20. The distance from the pitch link to the blade pitch axis was 0.24 m. The effect of rotor torque on hub spindle chord bending moment is shown in figure 21. The hub spindle chord bending moment gage was at r/R = 0.06. The effect of rotor torque on blade chord bending moment at 0.3R is shown in figure 22.

Wake Rake Data

Data obtained with the rotor wake rake are presented in Appendix B. The location of the pressure taps is presented in table 4. A dictionary of the parameters in Appendix B is provided in table 5. The data are organized by run number. Plots of wake dynamic pressure as a function of radius for several rotor thrusts are presented in figure 23.

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APPENDIX A

ROTOR PERFORMANCE AND LOADS DATA

SPND FB SPND CE Fb .3R CB .3R P LINK	-13127. -84C. -5202. -3378. -482.	-11322. -563. -4307. -3209. -332.	-10164. -596. -3698. -3247. -154.	-8877. -757. -3000. -3284. 0.000191	-7054. -1062. -2044. -3234. 149.
POMER TOPQUE, C CA/S CQ/S, C FM FM	2840. 2840. 0.00241 0.00270 0.0004	132. 2187. 0.000185 0.09206 0.1912	152. 2525. 0.000214 0.00240 0.3782	197. 3268. 0.000277 0.00311 0.4883	271. 4499. 0.00332 0.06428 0.6628
C1/S C2/S C2/S C4M/S C4M/S	1.00030 0.00051 0.00009 0.00003	0.01508 0.00017 0.00088 0.00052 00010	0.02603 0.00040 0.00047 0.00065 0.00014	0.03726 0.00044 0.00123 0.00046 00038	0.05311 00038 0.00106 0.00062 0.00918
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THRUST SIDE NORMAE PITCH YAW	-83 -142. 25. 454. 23.	4165. 46. 242. 545. -105. 2150.	7185. 113. 129. 688. 151. 2464.	10282- 122- 340- 485- -401- 3199-	14536. -103. 293. 651. 190.
F R R R Y L C C C C C C C C C C C C C C C C C C	-313. -273. -27. 415. 405.	4134. -196. -463. 1476. 548.	7294. -417. -343. 1523. 2361.	10535. -129. -787. 2530. 779.	15011. -316. -962. 3518. 1707.
PSIND PSIN HUM, R HEAP PPESS	1.2 148. 44. 12.4 101.5	140. 140. 12.4 101.5	1.5 143. 44. 12.4 101.5	1.3 139. 44. 12.5 1.25 1.235	1.5 125. 44. 12.8 101.5
POINT VATIF COLL	14 15 18 234.5 0.6922	14 16 587.8 234.5 0.6922 -5.0	587.8 234.8 0.6922	14 16 587.6 234.5 0.5921	14 587.7 734.5 0.4917

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POWER TORQUE,C CQ,C CQ/S,C FM FM,C	887. 14578. 0.001242 0.01394 0.7544	1018. 16729. 0.001427 0.01602 0.7427	1150. 18914. 0.001615 0.01813 0.7310	2887. 0.000246 0.00277 0.0174	143. 2322. 0.000198 0.00222 0.0387
CY/S CY/S CZ/S CPM/S CYM/S	0.13452 00037 0.00126 0.00103 0.00092	0.14606 00024 0.00083 0.00117 0.00022	0.15688 0.00041 0.00159 0.00133 0.00043	00372 00005 0.00023 00004 0.00031	0.00551 0.00021 0.00003 0.00003 0.0023
Cocce	0.011985 000033 0.000112 0.000092 0.000082	0.013014 000022 0.000074 0.000104 0.000103	0.013978 0.000037 0.000142 0.000118 0.000038	000332 000005 0.000020 000003 0.000027	0.000491 0.000012 0.000012 0.000003 0.000026
THRUST SIDE NORMAL PITCH YAW TORQUE	36927. -102. 346. 1074. 959.	40043. -66. 226. 1223. 231.	42964. 113. 436. 1383. 445.	-1020. -14. 62. -39. 320. 2883.	1511. -58. 37. 36. 2329.
NE CC NE CC NE CC NE CC	38038. -1001. -754. 2869. 4609.	41238. -1013. -1294. 5128. 4096.	44301. -1258. -1187. 4702. 5251. 17958.	-997. 28. -18. -34. 356.	1560. 156. 82. -182. -228. 2188.
WIND PSIM HUM, & TEMP PRESS RHO	1.1 118. 44. 13.2 101.5	1.0 124. 44. 13.4 101.5	1.3 118. 44. 13.5 101.5	0.7 341. 89. 13.4 101.5	0.7 354. 89. 13.4 101.5
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POWER TORQUE,C CQ/C CQ/S,C FM	1099. 17769. 0.001523 0.01709 0.7581	178. 2900. 0.000248 0.00278 0.0184	154. 2492. 0.06.0213 0.00259 0.0112	2203. 0.000188 0.00211 0.1995	165. 2662. 0.000228 0.00255 0.4128
CT/S CZ/S CPM/S CYM/S	0.15633 00170 0.00125 00003 0.00207	00389 00089 0.00014 00006 0.00059	0.00252 00099 00005 0.00005 0.00068	0.01595 00135 0.00025 0.00079 0.00213	0.02942 00138 0.00033 00010 0.00086
e a a south	0.013929 000152 0.000111 000003 0.000185	000347 000080 0.000012 000006 0.000053	0.000225 000088 000005 0.000002 0.000060	0.001421 000120 0.000022 0.000008 0.000071	0.002621 000123 0.000030 0000077 0.000077
THRUST SIDE NORMAL PITCH YAW TORQUE	42664. -465. 340. -34. 2154.	-1065. -245. 38. -66. 615.	-272- -15- 706- 2496-	4366. -369. 67. 92. 2222.	8052. -378. 91. -108. 898. 2690.
TIC SFIC NFIC PMIC PMIC	42239. -335. -374. -62. 3845.	-1070. 165. -169. 220. -912.	706. 246. -27. 96. -1132. 2541.	4252. 175. 98. -577. -1061.	7920. 181. 55. -685. -403.
WIND PSIW HUM, 4 TEMP PRESS RHO	356. 356. 89. 13.7 101.5	0.6 357. 89. 13.7 101.5	0.7 344. 89. 13.7 101.5	0.7 24. 89. 13.8 101.5	1.0 51. 89. 13.8 101.5
RUN RPH VTIP MTIP COLL	15 586.5 234.0 0.6893	15 16 587.4 234.4 0.6903	15 587.5 234.4 0.6903	15 18 287.5 234.4 0.6903	15 19 234.4 0.6903 -2.5

SPND FB SPND CB FB .3R CB .3R P LINK	5564. -3093. -2911. 92.	-6751. -1032. -2094. -2870. 240.	-4609. -1515. -954. -2760. 373.	-2386. -2152. 198. -2606. 513.	-1122. -2549. 873. -2525. 582.
POWER TORQUE, C CQ/C CQ/S, C FM	214. 3463. 0.000296 0.0332 0.5488	296. 4775. 0.000408 0.00458 0.6521	405. 6469. 0.000553 0.00620 0.7267	542. 8659. 0.000740 0.00831 0.7578	632. 10146. 0.000868 0.00974 0.7661
CT/S CY/S CZ/S CPN/S CYN/S	0.04220 0.00076 0.00013 00022 0.00076	0.05868 00133 0.00060 00006 0.00108	0.07784 00219 0.00072 0.00149 0.00631	0.09728 0.00268 0.00059 0.00160 0.00845	0.10868 00197 0.00012 0.00153 0.00153
CONTRACT	0.003750 000067 0.000012 000019 0.000068	0.005228 000118 0.000054 000005 0.000096	0.006935 000196 0.000064 0.000040 0.000133	0.008667 000238 0.000052 0.000036 0.000142	0.009683 0.000115 0.000022 0.000137
THRUST SIDE NORMAL PITCH YAW	11541. -207. 36. -224. 793.	16772. -363. 165. -61. 1130.	21311. -601. 196. 470. 1558. 6578.	26604. -732. 161. 417. 1666.	29711. -537. 34. 252. 1597.
SF, CC NF, CC NF, CC VM, CC	11396. 60. 80. -881. 366.	15896. 116. 243. -1206. 411.	21110. -373. 5. -209. 2503. 6383.	26317. -302. -515. 1800. 2275. 8379.	29415. -372. -372. 1271. 2836.
WIND PSIM HUM, & TEMP PRESS RHO	0.4 268. 89. 13.9 101.5	0.2 2. 89. 13.9 101.5	1.0 36. 89. 13.9 101.5	0.9 29. 89. 14.1 101.5	0.9 41. 89. 14.1 101.5
RUN POINT RPH VIIP MTIP	587.4 20 234.4 0.6900 -0.5	15 21 287.9 234.6 0.6906	15 22 23 234.8 234.5 0.6904	15 23 587.7 234.5 0.6901 5.5	15 24 587.6 234.4 0.6899 6.5

SPND FB SPND CB FB .3R CB .3R P LINK CT**3/2	36. -2942. 1499. -2407. 655.	1295. -3382. 2179. -2295. 714.	2608. -3892. 2907. -2160. 782.	3868. -4440. 3631. -2053. 826.	5030. -5082. 4292. -1979. P66.
POWER TORQUE,C CQ/S CQ/S,C FM FM	713. 11460. 0.000980 0.01100 0.7737	810. 13056. 0.001117 0.01254 0.7765	921. 14879. 0.01274 0.01430 0.7745	1042. 16851. 0.001444 0.01620 0.7644	1175. 18996. 0.001628 0.01827 0.7446
CT/S CY/S CZ/S CPM/S CYM/S	0.11854 00267 0.00041 0.00143	0.12936 00263 0.00050 0.00140 0.01264	0.14072 00228 0.00028 0.00080 0.00064	0.15159 00248 0.00070 0.00050 0.00057	0.16154 00279 0.00033 0.00075 0.00154
CAN CO	0.010562 000238 0.000037 0.000058 0.000127	0.011526 000236 0.000045 0.000069 0.001127	0.012538 000203 0.000025 0.000071 0.000057	0.013507 000221 0.000062 0.000045 0.000051	0.014393 000248 0.000067 0.000137 0.001640
THRUST SIDE NOPMAL PITCH YAW TORQUE	32404. -730. 113. 679. 1487.	35350. -724. 137. 809. 1454.	38435. -624. 73. 834. 667.	41381. -676. 191. 520. 593.	44074. -766. 99. 776. 1503.
SF, LC NF, LC PM, LC VM, LC	32069. -467. -706. 2967. 2738.	35002. -574. -790. 3377. 3625.	38014. -929. -1072. 4428. 3384. 14219.	40972. -847. -1120. 4007. 3135. 16158.	43550. -1005. -940. 3940. 4564.
WIND PSIM HUM, & TEMP PRESS RHO	1.0 47. 89. 14.1 101.5	• 40 • • 6	1.2 69. 89. 14.1 101.5	3.8 62. 39. 14.1 101.5	0.8 55. 69. 14.1 101.5

15 27 27 234 0.6896 9.5

15 28 287.1 234.2 0.6894 15 29 29 587.0 234.2 0.5893

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RUN POINT PPM VIIP MIIP

SPND CE SPND CE FB . 3F CB . 3K P LINK	5922. -5725. 4884. -1950. 892.	6725. -6416. 5449. -1947. 954.	-13543. -767. -5737. -3028. -451.	-13660. -657. -5744. -3018. -427.	-11523. -451. -4698. -2865. -246.
POWER TORQUE, C CQ, C CQ, S, C FM	1308. 21125. 0.001812 0.02033 0.7185	1455. 23558. 0.002022 0.02273 0.6863	187. 3054. 0.00261 0.00293 0.0130	169. 2760. 0.000236 0.00265 0.0217	141. 2287. 0.000195 0.00219 0.1699
CT/S CZ/S CZ/S CVR/S CQ/S	0.16949 00276 C.00070 00023 U.00138	0.17672 00224 0.00059 0.00053 0.00093	00318 00140 00055 0.00029 0.00042	00418 00120 0.00040 0.00041 0.00041	0.01462 0.00089 0.00033 0.00031
C C C C C C C C C C C C C C C C C C C	0.015101 000246 0.000062 000020 0.000123	0.015746 000200 0.000052 0.000047 0.000083	000283 000125 000049 0.000026 0.00028	000373 000107 0.000036 0.000042 0.000036	0.001303 0.000080 0.000035 0.000030 0.000028
THRUST SIDE NORMAL PITCH YAW TORQUE	46214. -752. 190. -237. 1439. 21290.	48143. -610. 160. 545. 964.	-870. -384. -150. 306. 443.	-1145. -329. 110. 487. 427. 2745.	4004. -245. 108. 346. 223.
T, LC NF, LC PM, LC YM, LC	45796. -861. -975. 2513. 4053.	47671. -1370. -1097. 4114. 5637.	-964. -159. -57. 578. -441.	-1081. -12. -133. 576. -728.	3916. -127. -612. 2031.
MIND PSIW HUM, & TEMP PRESS RHD	0.8 47. 89. 14.1 101.5	1.2 66. 89. 14.2 101.5	1.4 73. 89. 14.1 101.5	1.6 79. 89. 14.1 101.5	1.7 69. 89. 14.1 101.5
RUN POINT RPM VTI P WTI P COLE	15 30 234.1 12.5	15 31 234.0 .6887 13.5	15 32 588.0 234.6 6904	15 33 588.0 234.6 .6904	15 34 588.0 234.6 0.6905 -5.0

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SPND FB SPND CB FB • 3R CB • 3R P LINK	4615. -4849. 4062. -1982. 844.	-13515. -806. -5702. -3042. -437.	-11945. -462. -4912. -2874. -300.	-10570. -462. -4199. -2879. -122.	-9332. -588. -3509. -2923. 30.
POWER TORQUE,C CQ,C CQ/S,C FM FM	1131. 18357. 0.001565 0.01756 0.7575	196. 3186. 0.000271 0.00364 0.0085	139. 2245. 0.000191 0.00214 0.1058	154. 2497. 0.000213 0.00239 0.3258	193. 3144. 0.000268 0.00301 0.4629
CT/S CYN/S CYN/S CYN/S CYN/S	0.15832 00223 0.00017 0.00060 0.01755	00246 00108 00059 0.00016 0.00027	0.01049 00136 0.00013 0.00029 0.00061	0.02392 0.00079 0.00004 0.00030 0.0025	0.03517 00139 0.00050 0.00050 0.00039
C C C C C C C C C C C C C C C C C C C	0.014106 000199 0.000015 0.000054 0.001564	000220 000096 000053 0.000014 0.000024	0.000935 000121 0.000011 0.000026 0.000054	0.002131 0.000071 0.000027 0.000022 0.000022	0.003133 000124 0.000044 0.000045 0.000034
THRUST SIDE NORMAL PITCH YAW TORQUE	43428. -612. 46. 629. 315.	-678. -296. -163. 163. 286.	2889. -374. 36. 305. 536.	6556. -217. 11. 313. 263. 2502.	9637. -381. 136. 525. 403.
TALC NFALC PRACC YMACC	42932. -1148. -1354. 5052. 3912.	-828. -211. -9. 367. -236.	2852. -327. -478. 1469. 504.	6388. -181. -520. 2183. 279.	9491. -161. -690. 2993. 385.
WIND PSIW HUM, & TEMP PRESS RHO	1.8 86. 89. 14.4 101.5	1.3 62. 89. 14.4 101.5	1.0 59. 89. 14.3 101.5	1.4 75. 89. 14.3 101.5	1.6 81. 89. 14.3 101.5
RUN POINT RPH VTIP MTIP COLL	15 45 588.9 235.0 0.6912	15 46 589.9 235.4 0.6923	15 47 590.0 235.4 0.6925 -5.5	15 48 588.6 234.8 0.6909	15 288.5 234.8 0.6908

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SPND FB SPND CB FB .3P CB .3R P LINK	-7639. -855. -2585. -2891. 172.	-5681. -1245. -1529. -2796. 308.	-3440. -1803. -357. -2644. 452.	-2188. -2183. 300. -2567. 522.	-980. -2562. 929. -2472. 591.
POWER TORQUE, C CQ/S, C CQ/S, C FM, C	258. 4175. 0.000356 0.00400 0.5985	348. 5640. 0.00482 0.00540 0.6920	470. 7589. 0.000648 0.00727 0.7505	554. 8993. 0.000768 0.00862 0.7626	637. 10331. 0.000883 0.00991 0.7721
CT/S 1 CY/S 1 CZ/S CYW/S CYW/S	0.05058 00091 0.00029 0.00036 0.00022	0.06801 00142 0.00073 0.00058 0.00041	0.08778 00179 0.00027 0.00056 0.00087	0.09907 0.00224 0.00032 0.00050 0.00084	0.10967 00169 0.00035 0.00059 0.00032
P S S S S S S S S S S S S S S S S S S S	0.004507 000081 0.000026 0.000032 0.000019	0.006060 0.000065 0.000052 0.000036	0.007821 -000159 0.000024 0.000078	008827 000200 000044 000045	009772 000151 000052 000028
THRUST SIDE NORMAL PITCH YAU TORQUE	13860. 0 -249 80. 0 374. 0 225. 0	18630. 0 -390 200. 0 604. 0 426. 0	24033. 0 -489. 1 74. 0 584. 0 911. 0	27117. 0 -614 89. 0 517. 0 878. 0	30009. -463 96. 0 613. 0 329. 0
SP/LC NF/LC PR/LC YR/LC	13686. -188. -703. 2938. 599.	18417. -278. -843. 3483. 1105.	23725. -536. -945. 3796. 7209.	26830. -473. -838. 3322. 2054. 8544.	29661. -613. -1099. 4298. 2313.
WIND PSIW HUM, & TEMP PRESS RRD	1.6 78. 89. 14.3 101.5	1.1 82. 89. 14.3 101.5	0.9 70. 89. 14.3 101.5	1.5 82. 89. 14.3 101.5	1.6 81. 89. 14.3 101.5
RUN POINT RPH VTIP MTIP COLL	588.5 234.8 6908 0.5	15 51 51 234.8 -6907	15 52 534.3 234.7 6905	15 588.2 234.7 6904	588.1 234.6 6.5

SPND FB	258.	1430.	2721.	3711.	4856.
SPND CB	-2952.	-3430.	-3923.	-4489.	-5136.
FB .3R	1585.	2247.	2962.	3569.	4293.
CB .3R	-2345.	-2255.	-2119.	-2072.	-2001.
P LINK	650.	724.	785.	836.	884.
POWER	721.	822.	929.	1050.	1187.
TORQUE, C	11736.	13339.	15068.	16810.	19016.
CQ/C	0.001003	0.001141	0.001289	6,001438	0.001627
CQ/S, C	0.01126	0.01280	0.01447	0,01614	0.01826
FM	0.7838	0.7761	0.7771	0,7520	0.7330
CT/S	0.12034	0.13050	0.14174	0.15046	0.16061
CY/S	0.00141	00221	00228	00229	00263
CZ/S	0.00049	0.00048	00003	0.00042	0.00052
CYH/S	0.00059	0.00070	0.00054	0.00053	0.00019
CYH/S	0.01124	0.00044	0.00036	0.00022	0.01853
626448	0.010722	0.011628	0.012629	0.013406	0.014310
	000126	000197	000203	000204	000234
	0.000044	0.000062	000003	0.000038	0.000046
	0.000052	0.0000639	0.000048	0.000048	0.000017
	0.000004	0.001142	0.000032	0.000020	0.000038
THRUST SIDE NORMAL PITCH YAW TORQUE	32919. -385. 134. 610. 47.	35684. -604. 130. 729. 457.	38751. -624. -9. 560. 373.	41126. -62. 116. 556. 230. 17057.	43898. -719. 141. 201. 439.
SETIC SETIC SETIC SETIC SETIC SETIC	32520. -667. -1197. 4461. 2364.	35266. -871. -1233. 4845. 3264.	38314. -932. -1269. 5009. 3308. 14364.	40674. -1013. -1360. 4995. 3414.	43461. -939. -1172. 3918. 3257. 18432.
WIND PSIW HUM, & TEMP PRESS RHO	1.5 89. 89. 14.3 101.5	1.4 82. 89. 14.3 101.5	1.5 81. 89. 14.3 101.5	2.1 60. 89. 14.2 101.5 1.224	1.7 53. 89. 14.1 101.5
POINT RPM VIIP COLL	15 588.0 234.6 0.6902	15 587.9 234.5 0.6900	15 57 587.8 234.5 0.6900	15 587.6 534.4 0.6899	15 587.4 234.4 0.6899

RUN	MISC	T, LC SP, LC		៩៩	CT/S CY/S	$= \omega$	SPND FB
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7	9	369	20	.0000	.0004	00182	879
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6897	101.5	4128.	-97	800000	00009	0.7126	
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04.0	11	620.	216.	0.000024	0.00027	0.00284	-2638.
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16	•	24440.	0	.01063	.1193	8.1	82
		-49	-337	.00014	0016	601	2686
0	89.	-509-	147.	0.000064	0.00072		1267.
•	+	046	20	.0000	.0004	0109	2084
66	•	95	80	.00010	.0011	0.759	433
•	. 22	569	8	.00102	.0114	.794	00
16		83	02	.01256	.1410	0	371
	8	-80	-406	.00017	.0019	1067	346
	8	-742.	144.	0.000062	0.00010	0.001257	2400-
E	15.	82	28	•0000•	.0004	0141	199
98	-	58	69	.00000	.0008	.175	550
•	. 22	869	10	.00128	.0144	. 791	\$

SPND FB SPND Cb FB . 3R CB . 3R P LINK CT**3/2	4265. -4256. 3465. -1602. 662.	-3199. -2134. -287. -2825. 344.	251. -3203. 1550. -2520. 529.	2468. -4079. 2720. -2307. 659.	4813. -5167. 4078. -2061. 763.
POMER TORCUE, C CQ, C CQ, S, C FM, C	768. 13799. 0.001571 0.01763 0.7712	440. 6971. 0.000646 0.0725 0.7153	0.000998 0.01120 0.7629 0.7629	843. 13531. 0.031262 0.01416 0.7463	17345. 0.001611 0.01603 6.7341
CT/S CY/S CZ/S CYM/S CYM/S	0.16072 00057 c.00089 c.00069 c.00075	0.08807 00025 0.00061 0.00037 0.00338	0.12106 00058 0.00049 00001 0.00036	0.14005 00032 0.00037 0.00083 0.00083	0. 6221 0.000112 0.00054 0.00035 0.01473
	0.014320 000050 0.300079 0.000061 0.000068	0.000022 0.000025 0.000033 0.000034	0.010786 000052 0.000000 0.000000 0.0000077	0.312478 363029 0.300633 0.003674 0.000627	0.614453 000100 0.000048 0.000032
TPRUST SIDE NORMAL PITCY YAW TCFQUE	33015. -116. 183. 537. 595.	22238. -64. 155. 354. 369.	30564. -147. 124. -5. 827.	35309. -81. 94. 796. 296.	40854. -283. 136. 340. 18022.
S A Y LC	32762. -882. -717. 2954. 4558.	22051. -430. -554. 2469. 6979.	30315. -470. -522. 1373. 3102.	34977. -1231. -829. 3986. 5568.	46499. -1179. -1133. 4126. 4849.
WIND PSIM HUM, 4 TIMP PRESS FROM	2.2 19. 89. 15.3 101.5	2.7 9. 89. 15.4 101.5	2.0 11. 89. 15.2 161.5	2.6 21. 39. 15.4 101.5	2.4 19. 89. 11.4 101.5
POINT REPRESENT COLL	16 7 510.5 203.7 0.5982	16 8 566.1 225.9 0.6632	16 55.8 225.8 0.6632	10 565.6 225.7 0.6627	16 11 565.4 225.6 0.6624 12.0

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SPND FE	-3154.	215.	2662.	4704.	-2947.
SPND CB	-2359.	-3456.	-4448.	-5598.	-2542.
FR .3R	-107.	1644.	2981.	4186.	129.
CB .3k	-2970.	-2719.	-2467.	-2348.	-3187.
P LINK	389.	580.	709.	792.	446.
POWSR	515.	754.	967.	1212.	618-
TOPQUE, C	7921.	11691.	15093.	18965.	9187-
CO.C	0.000677	0.000999	0.001292	0.001625	0.000693
CO.S.C	0.00759	0.01122	0.01450	0.01824	0.00784
FM.	0.7282	0.7589	0.7526	0.7166	0.7702
CT/S	0.09131	0.1_115	0.14235	0.16030	0.09523
CV/S	0.00008	00067	00049	00072	00035
C2/S	0.00008	0.00050	0.00070	0.00023	0.00017
CPM/S	0.00033	0.00008	0.00051	0.00024	0.00038
C44/S	0.00009	0.00039	0.00053	0.00047	0.00005
E A B E E E E E E E E E E E E E E E E E	0.008135	0.010794	0.012684	0.014283	0.008485
	0.000000	000060	000044	000064	000031
	0.000007	0.0000045	6.000062	0.000021	0.000034
	0.000030	0.000007	0.000045	0.000021	0.000034
	0.000008	0.000035	0.000047	0.001634	0.000005
THPUST	24997.	33140.	38892.	43749.	29304.
SIDE	1.	-184.	-134.	-197.	-109.
NORMAL	22.	137.	191.	63.	52.
PITCH	348.	86.	526.	245.	445.
TERGUE	99.	411.	550.	485.	63.
SFIC SFIC SFIC SFIC SFIC SFIC SFIC SFIC	24747. -707. -767. 3117. 3134.	32828. -714. -1011. 3338. 11508.	38512. -746. -989. 3460. 3953.	43326. -1107. -1074. 3834. 4969. 16646.	28994. -757. -896. 3581. 3673.
SIND	2.6	2.4	2.4	2.5	2.1
PSIN	23.	18.	25.	25.	48.
TEXT	89.	89.	89.	89.	89.
PRESS	15.5	15.5	15.6	15.6	15.7
PHO	101.5	101.5	101.5	101.5	101.5
PUINT RPM VTIP CCLL	16 12 589.6 235.2 0.6906 5.0	16 13 289.3 235.1 0.6903	16 14 589.1 235.0 0.6899	16 15 588.8 234.9 0.6896	16 16 625.2 249.5 0.7322

SPND FB SPND CB FB .3R CB .3R P LINK	501. -3834. 1977. -2944. 648.	2934. -4966. 3348. -2814. 767.	5780. -6537. 4989. -2632. 832. 0.001830	-13267. -1061. -5459. -3201. -450.	-12025. -859. -4830. -3085. -338.
POWER TORQUE,C CQ,C CQ/S,C FM FM	913. 13403. 0.001020 0.01145 0.7718	1177. 17321. 0.001321 0.01482 0.7505	1531. 22836. 0.001744 0.01957 0.7235	169. 2773. 0.000234 0.00263 0.0145	137. 2247. 0.000190 0.00213 0.0641
CT/S CV/S CZ/S CYM/S CYM/S	0.12388 0.00023 00023 0.00050 00012	0.14418 00037 0.00030 00013 0.00031	0.16792 00066 0.00041 0.00023 00037	00320 00016 0.00052 0.00033 0.00022	0.00749 00050 0.00058 0.00032 0.00041
	0.011038 0.000020 000020 0.000044 000011	0.012846 000033 0.000027 000016 0.000028	0.014962 000059 0.000037 0.000030 000033	000285 000015 0.000046 0.000029 0.000029	0.000667 0.000044 0.000052 0.000028 0.000036
THRUST SIDE NORMAL PITCH YAW TORQUE	38053. 70. -71. 582. -142.	44225. -113. 92. -208. 364.	51434. -202. 127. 265. -429.	-885. -45. 143. 343. 234. 277?	2071. -138. 161. 336. 431. 2248.
T, CC SF, CC PM, CC YM, CC	37666. -1090. -1086. 4647. 4625.	43820. -827. -1120. 3255. 4120.	50967. -1525. -1169. 4183. 5353.	-942. -244. -112. 489. 2684.	2029. -116. -348. 1257. 853.
WIND PSIM HUM, & TEMP PRESS RHO	2.5 28. 89. 15.8 101.5	2.3 3. 89. 15.9 101.5	2.9 49. 89. 15.9 101.5	0.3 68. 66. 8.7 102.0 1.257	0.2 69. 66. 8.8 102.0
RUN POINT RPN WTIP COLL	16 17 624.9 249.3 0.7316	16 18 624.6 249.2 0.7311	16 19 624.1 249.0 0.7306	22 4 583.4 232.8 0.6917	22 583.4 232.8 0.6915

SPND FB SPND CB FB .3R CB .3R P LINK	-10639. -770. -4141. -3046. -177.	-9333. -872. -3433. -3062. -23.	-7840. -1093. -2620. -3049. 118.	-5698. -1497. -1492. -2949. 258.	-3438. -1984. -359. -2788. 400.
POWER TORQUE,C CQ,C CQ/S,C FM	135. 2225. 0.000188 0.00211 0.3069	167. 2735. 0.000231 0.00260 0.4917	3660. 0.000309 0.00347 0.6130	313. 5185. 0.000439 0.00493 0.7293	425. 7057. 0.000598 0.00571 0.7905
CT/S CY/S CZ/S CPM/S CYM/S	0.02105 00038 0.00050 0.00039 0.00023	0.03312 0.00045 0.00070 0.00052 0.0014	0.00633 0.00094 0.00055 0.00050 0.00045	0.06571 00092 0.00106 0.00037 0.0059	0.08498 0.00063 0.00087 0.00057 0.00053
CCPA	0.001875 000034 0.000044 0.000035 0.000020	0.002951 000040 0.000063 0.000047 0.000013	0.004128 0.000084 0.000049 0.000045 0.000040	0.005855 0.000082 0.000033 0.000033	0.007572 000056 0.000078 0.000051 0.000047
THRUST SIDE NORMAL PITCH YAW	5822. -105. 137. 412. 241. 2212.	9162. -124. 194. 553. 151.	12814. -261. 153. 527. 477.	18158. -254. 292. 391. 619.	23471. -173. 241. 602. 554.
SF, LC SF, LC NF, LC YM, LC	5642. -39. -342. 1311. 598.	8975. -349. 1406. 404. 2703.	12646. 6. -419. 1809. 641. 3506.	17963. 183. -305. 905. 341. 5041.	23168. 110. -543. 1871. 1050. 6703.
FIND PSIM HUM, # TEMP PRESS	0.4 147. 66. 8.8 102.0	0.5 254. 66. 8.8 102.0	0.5 168. 66. 8.8 102.0	0.4 166. 66. 8.9 102.0	0.6 167. 66. 8.9 102.0 1.256
RUN RPH RTIP WTIP COLL	22 6 583.4 232.8 0.6915	22 7 583.3 232.7 0.6914	22 8 583.3 232.7 0.6914	22 9 583.2 232.7 0.6911	22 10 583.1 232.6 0.6909

SPND FB SPND CB FB .3R CB .3R P LINK	-977. -2660. 902. -2589. 536.	254. -3077. 1553. -2490. 598.	1544. -3533. 2254. -2371. 671.	2681. -3979. 2856. -2271. 737.	3971. -4533. 3550. -2152. 801.
TORQUE,C CQ,C CQ/S,C FM	575. 9520. 0.000807 0.00906 0.8239	664. 10970. 0.000930 0.01044 0.8203	764. 12611. 0.001069 0.01200 0.8137	861. 14190. 0.001204 0.01351 0.8034 0.7988	977. 16071. 0.001365 0.01532 0.7938
CT/S C2/S C2/S CPM/S C4M/S	0.00045 0.00045 0.00080 0.00066 0.00069	0.11746 00057 0.00099 0.00016 0.00093	0.12824 00016 0.00074 0.00040 0.00088	0.13779 00019 0.00078 0.00102 0.011344	0.14880 00052 0.00125 0.00025 0.01526
T A D C C C C C C C C C C C C C C C C C C	0.009528 000040 0.000071 000005 0.000062	0.010466 000050 0.000088 0.000014 0.000083	0.011426 000009 0.000066 0.000036 0.000079	0.012277 0.000017 0.000034 0.000091 0.001197	0.013254 000046 0.000112 0.000022 0.000054
THRUST SIDE NORMAL PITCH YAW TORQUE	29510. -123. 221. -64. 730.	32404. -:56. 272. 168. 977.	36365. -28. 265. 422. 12314.	37977. -515. 215. 356. 1070.	40981. -143. 345. 262. 639. 16011.
T, LC NF, LC NM, LC VM, LC	29164. 309. -459. 853. 761.	32051. 696. 1892.	34972. -132. -426. 1246. 2577.	37542. -31. -476. 1321. 2592. 13567.	40547. -39. -753. 1964. 1830. 15352.
FIND PRIEST PRESS	0.6 152. 66. 9.0 102.0 1.256	0.5 138. 66. 9.0 102.0	137. 66. 9.0 102.0	0.4 138. 66. 9.1 102.0 1.255	0.3 134. 66. 9.1 102.0
RUN RPH VIIP MTIP COLL	22 11 182.9 232.6 6907	22 12 12 532.8 532.5 6906	22 13 13 232.5 -6905	22 14 232.6 232.5 6902 9.0	22 15 232.5 232.4 .6900

SPND FB	4942.	5941.	6992.	7622.	-13260.
SPND CB	-5052.	-5735.	-6435.	-7171.	-1016.
Fb . 3K	4112.	4709.	5340.	5686.	-5414.
CB . 3R	-2073.	-2031.	-2006.	-2018.	-3338.
CT**3/2	848.	882.	911.	939.	-440.
POWER	1092.	1224.	1374.	1521.	172.
TORQUE, C	17939.	20105.	22576.	24885.	2803.
CQ/C	0.001524	0.001709	0.031921	0.002119	0.000237
CQ/S, C	0.01710	0.01918	0.02156	0.02378	0.00266
FM	0.7701	0.7429	0.7145	0.6805	0.0152
CT/S	0.15718	0.16567	0.17445	0.18087	00334
CY/S	0.00084	00066	00090	00004	60073
CZ/S	0.00055	0.00053	0.00125	0.00076	0.00115
CPM/S	0.00053	0.00046	0.00066	0.00020	0.00057
CYR/S	0.00008	0.00009	0.00178	0.00062	0.00048
C C C C C C C C C C C C C C C C C C C	0.014005	0.014761	0.015544	0.016116	000297
	000075	000059	000081	000004	000065
	0.000049	0.000047	0.0000111	0.000018	0.000103
	0.000047	0.000041	0.000070	0.000055	0.000042
	0.000007	0.000008	0.001918	0.002125	0.000042
THRUST SIDE NORMAL PITCH YAN	43271. -232. 152. 555. 17912.	45576. -183. 146. 483. 89.	47955. -248. 343. 693. 820. 22541.	49687. -12. 208. 214. 650. 24967.	-921. -203. 314. 596. 500.
Y P N P Y P C C C C C C C C C C C C C C C C C	42748.	45047.	47440.	49140.	-931.
	-382.	-405.	-734.	-533.	-248.
	-815.	-699.	-782.	-666.	-94.
	2613.	2296.	2469.	1709.	410.
	2276.	2374.	4107.	3655.	1130.
HUPSIN HUPSIN TONES PRESS	0.2 114. 66. 9.1 102.0 1.255	0.1 143. 66. 9.2 102.0	0.1 171. 66. 9.2 102.0 1.254	353. 66. 102.0 1.254	0.6 175. 66. 10.2 102.0 1.251
POINT POINT WATER COLL	22 16 562.3 252.3 0.6899	22 17 582.2 232.3 0.6896	22 18 582.0 232.2 0.6894 13.0	22 19 581.8 232.1 0.6891	23 584.4 233.2 0.6909

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POINT	PSIN	7	ິທ	. 2	;;	u	
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SPND FB SPND CB FB .3R CB .3R P LINK CT**3/2	-3383. -1956. -295. -2901. 389.	-1000. -2636. 908. -2731. 529.	369. -3076. 1636. -2615. 606.	1496. -3496. 2215. -2510. 655.	3006. -4010. 3027. -2360. 732. 0.001415
POWER TORQUE, C CQ, C CQ, S, C FM	428. 7092. 0.000602 0.00676 0.7990	582. 9620. 0.000817 0.00917 0.8141	11233. 0.000954 0.01071 0.8174	763. 12682. 0.001078 0.01210 0.8168	882. 14669. 0.011248 0.01400 0.8152
CT/S CY/S CZ/S CPM/S CYM/S	0.08608 00118 0.00120 0.00095 0.00066	0.10694 00092 0.00096 0.00099 0.00069	0.11872 00091 0.00109 0.00109 0.00082	0.12860 00149 0.00110 0.00091 0.01191	0.14146 00113 0.00078 0.00104 0.0133
555558 555558	0.007670 000105 0.000107 0.000085 0.000058	0.009529 000082 0.000088 0.000088 0.000062	0.010578 000081 0.000097 0.000073 0.000073	0.011458 000133 0.000098 0.000081 0.001062	0.012604 000100 0.000070 0.000119 0.001227
THRUST SIDE NORMAL PITCH YAW TORQUE	23718. -325. 332. 1002. 688.	29452. -254. 263. 1035. 728.	32683. -250. 250. 1138. 858.	35382. -410. 302. 956. 989.	38897. -310. 216. 1085. 1396.
T.LC SF.LC NF.LC PM.LC YM.LC	23445. -188. -606. 2409. 1799. 6638.	29107. -355. -554. 2379. 2602.	32330. -424. -572. 2597. 3004.	34990. -348. -566. 2172. 2590.	38467. -494. -748. 3217. 3910. 13795.
WIND PSIW HUM, & TEMP PRESS RHO	0.8 131. 66. 10.6 102.0	0.7 136. 66. 102.0 1.249	0.9 139. 66. 10.6 10.20	1.0 140. 66. 10.6 1.249	1.1 140. 66. 10.7 102.0 1.249
RUN POINT RPM WTIP MIIP COLL	23 23 233.0 0.6900	23 10 583.8 232.9 0.6899	23 11 583.7 232.9 0.6897	23 12 583.5 232.9 0.6896 8.0	23 583.5 232.8 0.6893

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VIND PSIN HUM, & TEMP PRESS RHO

MTIP

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-792. 3279. 4353. 5677.

1.1 147. 66. 10.7 102.0

583.3 232.7 0.6891 10.0

-522. -840. 3271. 4047.

0.9 151. 66. 10.7 102.0

23 15 583.2 232.7 0.6890

4453.

33

-651. -872. 3194. 3922. 9796.

0.6 145. 66. 10.7 1.249

583.0 232.6 3.6887 12.0

16197.

2060. 3905. 21666.

158, -30,

0.9 154. 66. 10.8 102.0

23 18 584.2 233.1 0.6900

423

402, 294, 607,

17401 -649

-736.

0.8 155. 66. 10.7 102.0

582.9 232.6 9.6885 13.0

RUM		٠	THRUST	t	1	POWER	PND F
POINT	PSIW	SF,LC	SIDE	CA		UE	SPND CB
R O K	ż	J	NORMAL	22	12		B • 3
H	12		\vdash	CPM	•	S	B .3
MTIP	ES	L	YAH	CYM	YH/	X.	CIN
7	Ē.	_	TORQUE	g	700		**3/
	1.0	•	0	.00152	0110	(1)	82
	-	220	190	.00006	.0006	286	-763
84.	99	195	54	.00001	.0001	00019	357
233.1	10.8	685.	46	0.000029	0.00033	0.00218	-2131.
069	02.	97		.00003	.0003	.219	245
-5	4	9	S	.00019	.0021	. 216	300
	1.2	0	7922.	.00256	.0287	59	-9842
	4	14	-	00000	0000	653	-797-
0	99	-328.	69	0.000087	0.00098		734
		090	638.	.00005	.0006	025	n
689	0	33	13.	00000	0000	.416	-37.
9	4	54	3	-00022	.0024	- 407	0.000130
	1.0	11240.	11380.	.00368	0413	02	22
	(1)	-46	-3	.00000	.0000	374	957
Ò	99	-448.	302.	0.000098	0.00110		-2938-
	-	1725.	741.	90000	.0007	032	162
689	;	439.	9	00000	.0000	.562	-
-	.24	3267.	3306.	.00028	.0031	. 551	2
		•		.00512	.0575	4	6629
		7	-22	.00000	0000	580	1258
0	99	-463.	313.	0.000101	0.00114		-1981.
	Ö	9	53	-00001	.0008	043	3101
689	•	011	•	.00000	.0000	0.681	192
•	-24	m		-00038	- 0042	• 666	36
	1.2	20537.	20792.	.00673	0755	-	62
		S	3	.00011	.0012	188	1684
8	99	-586-	522.	0.000169	0.00190		-606-
233.		2	\$	*0000	.0009	059	968
689	;	53		.0000	- 0005	.760	326
•	. 24	2	43	.00051	.0057	.742	55

SPND FB SPND CB FB .3R CB .3R C LINK	-2503. -2197. 114. -2829. 456.	-577. -2760. 1119. -2659. 557.	638. -3191. 1749. -2562. 628.	1835. -3669. 2415. -2455. 669.	3094. -4169. 3098. -2330. 747.
POWER TORQUE,C CQ/C CQ/S,C FM FM	487. 8156. 0.000694 0.00778 0.7977	613. 10237. 0.000871 0.0978 0.8117	701. 11714. 0.000997 0.01119 0.8084	13387. 0.001140 0.01280 0.8633	910. 15178. 0.001294 0.01452 0.7973
CT/S CV/S CZ/S CYH/S CYH/S	0.09338 00055 0.00138 0.00042 0.00016	0.11068 00103 0.00133 0.00080 0.00057	0.12084 0.00079 0.00131 0.00084 0.00076	0.13166 00028 0.00086 0.00084 0.01255	0.14256 00049 0.00067 0.00068 0.00073
CGAGGGG	0.008320 000049 0.000123 0.000038 0.000014	0.009861 000092 0.000119 0.000071 0.000051	0.010767 000070 0.000117 0.000075 0.00008	0.011731 000025 0.000076 0.000044 0.001118	0.012702 000044 0.000059 0.000061 0.000065
THRUST SIDE NORMAL PITCH YAW TORQUE	25681. -151. 380. 444. 169.	30414. -282. 366. 835. 594.	33196. -217. 359. 881. 798.	36153. -77. 235. 883. 511.	39118. -135. 183. 711. 767.
T, EC SF, EC PM, EC YM, EC	25375. 337. -285. -46. -288.	30034. -65. -517. 1567. 1315.	32813. -167. -420. 1352. 2195. 10885.	35724. -225. -499. 1864. 2303.	36675. -107. -475. 1758. 2335.
WIND PSIW HUM, & TEMP PRESS RHO	1.3 153. 66. 10.9 1.247	1.2 149. 66. 11.0 102.0	1.1 161. 66. 11.0 102.0	1.1 164. 66. 11.0 102.0	1.1 161. 66. 11.1 102.0 1.246
POINT RUN ATIP COLL	23 25 583.8 232.9 0.6894 5.0	23 26 583.7 232.9 0.6892	23 23 232 0.6890 7.5	23 283.5 232.8 0.6889	23 29 583.4 232.8 0.6887

報を表している 100mmの 100mm

POINT RPM	POINT FURTH	T.EC SF,EC NF,EC	THPUST SIDE NOPMAL	ដូចជួ	CT/S CV/S CZ/S	POWER TORQUE, C CQ.C	SPND FR SPND CB FB .3R
~ ~~ .	E S	\$ \$ 5	PITC VA	KA C	: =	- iL	B .3
20	Ī	٦	-	g	6		18**
	-	039	495	.01380	.1549	043	525
m	0	138	152	.0000	.0005	17347	4766
m	99	614	S	.00011	.0012	0147	11
232.	11.	409	247	.0000	.0002	.0166	2190
0.6885	102.0	2825	1074.	0.000092	0.00103	0.7876	
•	• 24	297	075	.20145	.0163	. 775	62
	0	10	61	.01450	.1627	165	447
ന	9	65	4	.0000	.0008	9331	5382
83.	99	978	326	.00010	.0011	00164	4451
232.7	11.2	3801.	1324.	0.000113	0.00127	0.01651	~
688	5	4005	644	.0000	.0006	0.758	845
1.	• 24	149	11	.00162	.0182	. 748	0.001746
	1.2	70	80	.00058	• 0065	54	206
m	4	219	104	.00003	.0003	527	-891
584.2	66	-31.	103.	0.000033	0.00038		
233.	~	44	49	.00004	.0004	0024	3179
689	5	31	235	.0000	. 3002	0.046	-350
9	. 24	4	-	.00021	.0024	046	0
	4	078	219	.00169	.0189	35	0853
m	0	195		.0000	.0004	246	-722
584.2	•99	-241.	95	0.000031	0.00035	00019	-4247.
233.	11.	952		-00003	. 0003	0.0021	115
689	;	682	396	.0000	.0003	0.261	-208
7	. 24	75	08	.00018	.0021	. 25	007
	•	559	79	.00281	0315	63	9482
	5	10	-74	.00002	0000	738	801
84.	99	294	~	.00010	.0011	00023	3519
233.1	11.4	1313.	808	0.000069	9.00077	0.00262	-3124.
689	~	37	11	00000	.0001	.465	-55
7	. 24	0	59	.00022	• 0025	• 452	14

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4 (340	. נ ה) 	7 -	CT**3/2		3 (2197	114	29	456	0.000759	577	2700		717	6007	C	16000.	38	3191	4 4	2562	4 C S	0.001117	1025	2660	417	7455	689	0.001271	700	7 7 7	700	3000	7330	٠,)
POWER	3			10 / J	FKC	ā	9	8126	0.000694	.0077	.792	.773		750	10000		0000	711000	*	01	1714	66000	0.0111	0.808	0.7921	6	7000	00114	0.0128	9.803	0.7879	10	5170	00110	677000	0140	0.7824	
7	` `	;;	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	2	S/82 C6/8	6600		C000 •	0.00138	. 0004	.0001	• 0016	.1106	0010	0013			0.00957	•	.1208	.0007	.0013	.0008	.0007	0.01097	1316	0000	0000	œ	.0004	0	.1475	7000	7000			0.01425	
t	Š	22	ָם קר		3	00833			0.000123	.0000	.00001	.00067	98 60 o •	60000	11000	70000	20000	0.000853		01076	.0000	.00011	.0000	.00006	•	01173	.00000	.0000	0.000075	.00004	.00111	.01270	200004	.0000		90000	3	
US	CI	RHA	PITC	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	TORQUE	5)	100	380	*	16	29	14	-28	99	3	9	10023) 	33196.	217	S	81	98	7	S	-77	m	883.	1	~	-	135	63	-	767	95	
1,1	7,7	F, L	H	.	0710	75		יו ייייייייייייייייייייייייייייייייייי	-CB7-		2	563	34	9	517	56	315	9529) 	32813.	16	420	352	19	88€	7	-22	499	1864.	2303	486	67	10	475	50	233%	174	
QNIM	2	天	7	تعا	%	•	~	7	• 0 0 0	•	• •	• 24	_	9	9	-	02.	1.247	1	7	H	99	11:	•	. 24	•	4	99	11.0	7	. 24	1.	-	99	_	102.0	24	
~	2.	2	H		9			2	232.0	0	00	•	23	Ň	83.	232.	689	•		6 7	7	•	232.	689	•		~	83.	232.8	6 BB	•	23	~	83.	32.	0.6887	•	

POWER SPND FB QUE,C SPND CB CQ,C FB .3R Q/S,C CB .3R FM P LINK	1043. 4525. 73474766. 01479 3911. 01660 -2190. 7876 797.	1165. 5447. 93315362. 01649 4451. 01651 -2129. 7586 845.	15412206. 2527891. 00215 -4916. 00241 -3179. .0468 0.000014	13510853. 2246722. 00191 -4247. 0214 -3115. -2619 -208.	1639482. 2738801. 00233 -3515. 00262 -3124. 4655 -55.
CT/S F C2/S TORG CPM/S CQ CVM/S CQ/S	0.15494 17 0.00128 0.00 0.00024 0.00 0.00103 0.00	0.16277 00088 0.00119 0.00127 0.00062 0.01827 0.0	0.00657 0.00038 0.00038 0.00047 0.0022 0.0022	0.01899 0.00041 0.00035 0.00032 0.00038 0.00011	0.03155 -00027 0.00116 0.00077 0.00011
	0.013805 000049 0.000114 0.000021 0.000092	0.014503 000078 0.000106 0.0000113 0.000055	0.000586 0.000034 0.000042 0.000042 0.00020	0.001692 0.000037 0.000031 0.000029 0.000034	0.0002811 0.000103 0.000103 0.000069
C THPUS C SIO C NOPMA C PITC C YA C TORQU	42495. 1152. 1. 352. 247. 1074.	7. 44616. 7241. 3. 3.6. 1324. 644.	1807. 104. 103. 103. 103. 103. 103. 103. 103.	3. 5219. -114. 95. 338. 2. 396.	8666 74- 1- 319- 1111-
ND SF,L		6. 44100 6657 6978 .2 3801 0 4005		2 5078 19. –195 16. –241 0 682	66 8559 66 - 107 66 - 294 1313 10 374
RUN EI RPM HUN VIIP TE COLL RRE	23 17 30 17 83.3 6 32.7 11 6885 102	23 0 31 15 83.1 6 63.7 11 6883 102 11.5 1.2	23 15 32 15 84.2 6 33.1 11 6893 102 -6.5 1.2	23 14 84.2 6 33.1 11 6893 102 -4.5 1.2	23 14 84.1 6 33.1 11 6891 102

D	Z	ب	S	ដ	7	POWE	NO F
PUINT	#ISd	SF, LC	SIDE	ָל נ	CV/S	TORQUE	ON d
2	=	1	T (22	773		2 · 2 ·
	1 E		LIC	MdD (7	<u>ر</u> د	*)
1	E.S.	٠ ا	X	CYM	\ \ \	۵.	ב ב
20	Ī	_	_	g	6		T**3/
		066	142	.00393	.0442	11	8103
	-	146	-116	.0000	.0004	3569	988
84.	99	319	254	.0000	.0009	00000	2777
233.0	11.4	790.	531.	0.000045	0.00051	0.00341	-3127.
589	.70	617	-53	.00000	.0000	3.595	88
9	4.	318	448	.00029	. 6632	0.575	4
	•	81	013	.00552	.0619	94	6112
		•	-192	.000C.	.0007	4968	1337
84.	66	331	355	.00011	. 3012	00042	1718
	-	Ø		00000	002	0.00475	-3045.
689	02.	486	461	.0000	.0004	. 709	225
	1.245	512	ァ	-0000	• 00	. 685	41
		308	576	.00733	. 3822	05	3878
	C	-194	-168	.000¢5	9000	6965	1838
83.	99	676	272	.30008	60000	00058	-545
233.0	-	2461.	851.	0.000072	3.00081	0.00657	-2886.
688	35.	355	243	.0000	.0002	0.785	366
	*	205	624	• 00026	• 0063	0.758	62
	•	716	0.65	.30911	.1023	35	1548
	~	-309	-160	.00005	.0005	196£	2463
83.	99	792	427	.00013	.0015	0000	607
232.9	_	2650.	926	2.000000	0.00092	0.00850	-2682.
688	02.	196	93	.3000.	. 3000	0.825	501
•	4.	199	745	.00074	.0083	ñ. 803	6
	•	396	0773	66600	.1122	615	-434
_	~	-354	170	.00000	9000.	10364	2778
. TO 100	99	650	352	.00	.0012	.9008B	1195
232.5	11.6	2159.	911.	9.000000	0.000A7	0.03992	-2596.
588	•	144	362	.00002	, 0002	0.824	567
•	. 24	444	55	•0008°	9600.	. 799	၁

SPND FE SPND CB FB .3P CB .3R P LINK	1082. -3292. 1986. -2481. 642.	2172. -3682. 2571. -2376. 699.	3148. -4166. 3126. -2285. 748.	4721. -4802. 4005. -2132. 816. 0.001650	5884. -5472. 4683. -2055. 850.
POWFR TORQUE, C CQ/S, C CQ/S, C FW FW	724. 12205. 0.001041 0.01169 0.8260	811. 13674. 0.001167 0.01310 0.8217	910. 15229. 0.001300 0.01460 0.8009	1052. 17696. 0.001505 0.01689 0.7925	1191. 19841. 0.001697 0.01905 0.7677
CT/S CY/S CZ/S CPM/S CYM/S	0.12540 00683 0.00131 0.00108 0.00077	0.13479 00005 0.00119 0.00108 0.00064	0.14313 00063 0.00126 0.00086 0.01427	0.15673 00063 0.00080 0.00100 0.00118	0.16678 00014 0.00056 0.00103 0.01672
CPACC	0.0011173 000074 0.000117 0.0000696 0.000069	0.012010 060004 0.000106 0.000097 0.000057	0.012752 000056 0.000112 0.000077 0.000032	0.013965 000056 0.000072 0.000105 0.001472	0.014860 000C13 0.000050 0.000079 0.000092
THRUST SIDE WORMAL PITCH YAW TORQUE	34373. -229. 359. 1125. 806.	36927. -13. 326. 1131. 664.	39197. -173. 345. 896. 377.	42890. -172. 220. 1043. 1723.	45603. -39. 153. 922. 1072.
T. LC NF, LC NF, LC VM, LC	34011. -334. -545. 2292. 2707.	36437. -781. -748. 2757. 4108.	38759. -368. -697. 2276. 2481. 14102.	42428. -500. -704. 2966. 4095.	45107. -690. -916. 3516. 4965.
WIND PSIW HUM, 4 TEAP PRESS RHG	1.7 155. 66. 11.6 102.0	1.8 159. 66. 11.6 102.0	1.4 159. 66. 11.6 102.0	102.0 102.0 102.0	1.2 158. 66. 11.7 102.0 1.244
POINT FPW ATIP	23 40 583.6 232.8 0.6883	23 232.8 232.8 0.6882	23 42 583.4 232.8 0.6880	23 43 583.2 232.7 0.6878	23 44 583.0 232.6 0.6875

SPND FE SPND CB FB .3R CB .3R P LINK	6510. -6049. 5094. -2063. 876.	-2929. -2352. 100. -3017. 529.	1088. -3773. 2200. -2737. 0.001218	3972. -5023. 3815. -2462. 838.	6254. -6493. 5123. -2383. 889.
POWER TORQUE, C CQ/C CQ/S, C FX	1307. 21799. 0.001866 0.02094 0.7332	609. 9313. 0.000702 0.00788 0.7757	936. 14319. 0.001082 0.01214 0.7959	1224. 15721. 0.031414 0.01587 0.7840	1551. 23930. 0.001814 0.02036 0.7378
CT/S CV/S C2/S CPM/S CYM/S	0.17220 00047 0.00111 0.00072 0.00054	0.09435 00002 0.00062 0.00061 0.00051	0.12801 0.00026 0.00147 0.00058 0.00111	0.15158 0.00078 0.00088 0.00058 0.01589	0.17071 0.00069 0.00101 0.00039 0.00111
######################################	0.015343 000042 0.000099 0.000064 0.000048	0.0000055 0.000055 0.000055 0.000054 0.000045	0.011405 0.000024 0.000131 0.000051 0.001082	0.013505 0.000070 0.000079 0.000060 0.000085	0.015210 0.000062 0.0000035 0.000035
THPUST SIDE NGRMAL PITCH YAW TORQUE	47052. -129. 305. 747. 562.	29261. -5. 192. 722. 598.	39625. 82. 454. 681. 1308.	46932. 243. 273. 801. 1142.	52782. 214. 312. 465. 1313. 23765.
SF, LC NF, LC PM, LC YM, LC	46548. -664. -871. 2846. 3955.	28970. -49. -471. 1960. 2398.	39225. -334. 187. -967. 4003.	46456. 57. -19. 2927. 16013.	52246. -693. 134. -1230. 5705.
WIND PSIW HUM, & TEMP PRESS AHO	1.3 161. 66. 11.8 102.0	0.0 311. 66. 14.4 102.0	0.0 347. 66. 102.0 1.230	0.1 312. 66. 14.3 102.0 1.231	0.7 156. 66. 14.3 102.0
RUN POINT RPM WTIP COLL	23 45 582.9 232.6 0.6873	24 624.1 249.0 0.7325	24 623.8 248.9 0.7318	24 623.5 248.8 0.7319	24 623.1 248.6 0.7313

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SPND FB SPND CB FB . 3K CB . 3F P LINK CT**3/2	-2771. -1670. -224. -2592. 328.	413. -2568. 1425. -2304. 497.	-2682. -1834. -105. -2722. 359.	-2915. -1982. -92. -2946. 400.	221. -2968. 1555. -2693. 577.
POWER TORUCE,C CQ/C CQ/S/C FM	317. 5801. 0.000653 0.00733 0.7321	487. 8928. 0.001005 0.01128 0.7671	365. 6371. 0.000669 0.00751 0.7344	425. 7065. 0.000677 0.00760 0.7226	625. 10505. 0.01130 0.7498
CT/S CV/S CVM/S CVM/S	0.08758 00028 0.00180 0.00039 00009	0.12045 00079 0.00197 0.00052 0.00072	0.09002 0.00089 0.00133 0.00050 0.00019	0.08994 00029 0.00126 0.00000 0.00031	0.11932 00098 0.00162 00009 0.00077
CCCCCC	0.307803 000025 0.030035 000008	0.010732 000070 0.000046 0.000064	0.008021 000079 0.000119 0.000045 0.000017	0.008014 000026 0.000112 0.000000	0.010631 000087 0.000145 000008 0.000069
THRUST SIDE NORMAL PITCH YAW TOPQUE	18203. -58. 374. 313. -72.	25018. -164. 409. 567.	20048. -198. 297. 425. 164.	21942. -71. 308. -1. 284. 7322.	29111. -239. 396. -85. 716.
T, LC SF, LC NF, LC YM, LC YM, LC	18096. -121. 126. -461. 806.	24857. -303. 260. -852. 2023.	19883. -294. 526. -1385. 1290. 5158.	21777. -12. 658. -2587. 807.	28880. -193. 588. -2730. 1765.
E NIND PSIM HUM, W TEAP PRESS RESS	1.3 50. 66. 14.8 102.0	1.4 50. 65. 14.8 102.0	1.8 37. 65. 14.8 102.0	1.7 19. 66. 14.7 102.0	1.7 36. 66. 14.4 102.0
RUN RPW RTIP KTIP COLL	24 10 511.3 204.0	24 11 203.9 .5995	24 13 529.2 211.2 0.6207	24 14 553.8 221.0 .6497 5.0	24 15 553.6 220.9 0.6497

CY/S TORQUE,C SPND FB CY/S TORQUE,C SPND CB CZ/S CQ,C FB 3R CPM/S CQ/S,C CB 3R CYM/S FM,C CT**3/2
PM CYM/
C C C C C C C C C C C C C C C C C C C
NORMAL PITCH YAN
SFIC PRIC PAIC OILG
PSITA HUM, HUM, TEMP PRESS PRESS
POINT RPM WTIP COLL

SPND FE SPND CE FB . 3R CB . 3R P LINK CT**3/2	-13613. -1072. -5575. -3162. -487.	-11909. -793. -4448. -3013. -300.	-10617. -799. -3763. -3624. -132.	-9318. -950. -3276. -3068. 27.	-7663. -1247. -2357. -3060. 167.
POWER TORQUE,C CQ,C CQ,S,C CQ,S,C FM	173. 2917. 0.000248 0.00278 0.0231	137. 2163. 0.000184 0.30206 0.1128	151. 2343. 0.000199 0.00274 0.3180	194. 2999. 0.00255 0.00287 0.4632	265. 4079. 0.000347 0.00390 0.5780
CT/S CV/S CV/S CV/S CV/S	0.00451 0.00006 0.00011 0.00010	0.01089 00002 00045 0.00022 0.00020	0.02321 0.00031 0.00020 0.00035 0.00011	0.03524 00020 0.00028 0.00022 0.00321	0.05035 00058 00020 0.00056 0.00022
T A B B B B B B B B B B B B B B B B B B	0.000402 0.000005 0.000047 0.000010 0.000009	0.000970 0000043 0.000020 0.000020	0.002068 0.000028 0.000018 0.000031 0.000010	0.003140 000008 000018 0.000025 0.000019	0.004486 000061 000018 0.000020 0.000367
THRUST SIDT NCRWAL PITCH YAN TORQUE	-1241. 17. 146. 118. -103. 2961.	2995. -7. -134. 208. 2228.	6377. 86. 56. 368. 113. 2456.	96 °1. -24. -56. 288. 227.	13825. -187. -54. 581. 232. 4315.
SALC SALC SALC SALC	-1245. 0. 22. -233. 35.	2933. -13. 383. -407. 132.	6320. -92. 392. -1015. 657.	9589. -154. -288. 1077.	13705. -287. -51. 1041. 1322.
PSIN PSIN TEXP PRESS	2.4 2. 75. 12.4 101.8	2.7 2.75.12.4 101.8	2.3 2. 75. 12.7 101.8	2.2 2. 2. 75. 12.7 101.8 1.236	2.3 2. 75. 12.7 101.8 1.236
RUN RPM VIIP MTIP COLL	25 10 586.4 234.0 0.6906	25 11 536.4 234.0 0.6906	25 12 586.3 233.9 0.6903	25 13 586.3 233.9 0.6902	25 14 586.2 233.9 0.6961

SPND FB SPND CB FB .3R CB .3R P LINK	-5865. -1672. -1393. -3005. 295.	-3290. -2318. -154. -2824. 9.000710	-2318. -2697. 352. -2795. 500.	-2190. -2687. 415. -2765. 497.	-665. -3070. 1115. -2594. 556.
POWER TOPQUE,C CQ/C CQ/S,C FM FM	360. 5387. 0.000459 0.00515 0.6428	504. 7699. 0.000656 0.0136 0.7172	584. 8982. 0.000766 0.00859 0.7103	562. 9038. 0.009771 0.00865 0.7236	669. 10409. 0.000888 0.00996 0.7587
CT/S CY/S CZ/S CPM/S CYM/S	0.06636 0.00004 00015 0.00037 0.00040	0.08934 00043 00002 0.00031 0.00045	0.09800 0.00004 00007 C.00065 0.00074	0.09899 00042 00028 0.00075 0.00053	0.11208 00048 00062 0.00075 0.01044
E A S A A S C C C C C C C C C C C C C C C	0.005913 0.000003 000014 0.000033 0.000036	0.007950 0000039 0000028 0.000028	0.008731 0.000004 000006 0.000058 0.000066	0.0008820 0.000038 0.000067 0.000047	0.009986 000043 000055 0.000039 0.000930
THRUST SIDE NORMAL PITCH YAW TORQUE	18218- 11- -42- 391- 418-	24512. -119. -6. 325. 469.	26881. 12. -20. 679. 773.	27153. -116. -77. 787. 551.	30733. -131. -170. 785. 462.
NETC NETC NETC NETC NECC NECC	18075. -173. 336. -698. 1696. 5595.	24309. -166. 261. -634. 1602.	26653. -243. 447. -889. 2469. 5095.	26912. -419. 287. -41. 2541.	30484. -580. 264. 202. 2909.
WIND PSIW HUM, & TEMP PRESS RHO	3.5 2. 75. 12.7 101.8	2.9 2. 75. 12.7 101.8	2.8 2. 75. 12.7 101.8	2.3 2. 75. 12.7 101.8	2.2 2. 75. 12.7 101.8
RUN POINT RPN VTIP COLL	25 15 586.1 233.8 0.6900	25 16 585.9 233.8 0.6898	25 17 585.9 233.8 0.6897	25 18 585.9 233.8 0.6897	25 19 585.8 233.7 3.6896

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	- L	7	ITC	CPM	DW/	S	8	
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3.6		ć		1			•	
7 6	F • 7	ָ כ	7	.01053	.1181	40	5	
7	~	719	263	00000	.0007	1507	141	
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233.7	12.7	1878.	781.	0.000067	0.00075	0 01100	1000	
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• (17.	182	23	90000.	• 0006	0.0130	300	
		3667	872	.00000	.0008	0.751	000	
•	• 23	13453.	14089.	0.001202	0.01350	0.7762	**************************************	
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25	3.3	954	289	.01245	707	Ç	i.	
	2.	-75	-234			706	0007	
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	101.8	3139	222	.00001	.0002	.733	751	
70.	• 23	972	15691.	0.001340	0.01504	0.7785	0.001390	
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25	3°3	41553.	666	-01362	.1529	105	048	
~	2.	128	-178.	000058	- 00065	16982	5121	
87.		562	240	10000	DOOR	00144	2410	
34.	3	567	823	70000	7000	17100	9 6	
	101.8	905	713	90000	7000	70100	2007	
11	2	0 0	- 6		0000	• / 34	S S S	
•	• 43	188	2/6	.00153	.0171	.777	0.001591	
	2.9	38	795	101421	1505	, נ	Č	
	~	200	127	17110	00000	1771	83.7	
2	1 4	7	7	70000	* 000a	18891	5649	
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10407 1000	13.4	1407.	640.	0.000055	0.00061	0180	2189	
))	-	247	-72	00000	.0000	.710	832	
7	• 23	96		.00168	.018	0.7446	0.001695	

SPND FB SPND CB FB .3R CB .3R P LINK	-13650. -512. -5411. -3642. -548.	-11883. -198. -4380. -3456. -342.	-10863. -189. -3869. -3497. -173.	-9569. -342. -3117. -3549. -12.	-7717. -611. -2232. -3486. 118.
POWER TORQUE,C CQ,C CQ/S,C FM FM,C	3109. 0.000266 0.00299 0.0176	142. 2250. 0.000193 0.00216 0.1124	152. 2368. 0.000203 0.09227 0.2726	195. 2948. 0.000252 0.00283 0.4189	261. 3969. 0.00340 0.00382 0.5770
CT/S CY/S CZ/S CYM/S CYM/S	00395 0.00070 0.00064 0.00026 00025	0.01118 0.00065 0.00028 0.00031 00033	0.02111 0.00044 0.00045 0.00040 00006	0.03324 0.00067 0.00019 0.00022 00031	0.04998 0.00056 0.00016 0.00023 0.00020
CYPECT	000352 0.000062 0.000057 0.000023 000622	0.000996 0.000058 0.000025 0.000027 000629	0.001881 0.000039 0.000040 0.000035 000006	0.002962 0.000017 0.000017 0.000028 0.00028	0.004453 0.000050 0.000015 0.000021 0.000018
THRUST SIDE NORMAL PITCH YAW	-1080. 192. 174. 269. -257.	3054. 178. 77. 316. -342. 2310.	5767. 121. 123. 412. -64.	9081. 184. 53. 231. -323. 3178.	13648. 152. 45. 244. 211.
SF/LC NF/LC VM/LC VM/LC	-1031. -22. 144. -251. 256. 2945.	3049. 174. 399. -553. 2249.	5724. 61. 409. -950. 63.	9062. 36. 445. -1247. 226.	13592- 268- 724- -1858- 286- 4150-
WIND PSIW HUM, & TEMP PRESS RHO	3.0 2. 75. 14.2 101.8	2.4 2. 75. 14.2 101.8	2.4 2.75. 14.2 101.8	3.3 2. 75. 14.2 101.8	2.8 2. 75. 14.2 101.8
POINT RPM VTIP MTIP COLL	26 586.5 234.0 0.6895	26 7 234.0 0.6886	26 8 8 234.0 0.6885	26 9 586.4 234.0 0.6884	26 10 586.3 233.9 0.6883

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CNO	יינ מינ	ייי הייי		в. 3	LIN	CT**3/		5587	1025	7 .	7 I	3375	250	0.000484		3533	25	9	3 6		380	9	אורנ		0/61	544	62	455	0.000828	ò	-800	327	1246	026	573	0.000987		50	810	1201	7071	707	584
S A	, Li		2	S	TE.	EM C	•	59	461	77000		7900	. 682	0.7310		4.00 V	442	60063	0.0071	71.22	7 T L .	. 103	7	יולי מיילי מיילי	K 600	0074	.0083	.738	0.7851	S	700	10094	9800	.0097	.765	0.8059	! !	43	1650	20000	0113	77700	0.7013
1	: ;		777	>	\ \ \ \	S/O)	,	.0691	-0005	0000) (5000·	£200·	0	3600	0/00	.0004	.0000	. 0005	0.00045			0880		•		. 0005	.0005	0.6800.0	1117	1000	7000	C000•	. 0005	.0007	0.01022		.1199	.0000	.0002	0000		0.00020
5	2		7 2	٦ ٢	Z. C.	<u> </u>	•	01000.	.00004	00000	000000	70000	.0000	.000050	09600		.0000	00000.	.00005	0.00040	9000		.00881	00001			40000	.00004	00.	.30991			r L	20000	.0000	00.	,	.01068	00000	.00002	00000	20000	0.001030
US	SIDE	T TR	26.40	، ر : 1	~	\supset	000	T C	140.	5	, K	,	3.4	47	9	•	4	2	9	470.	\C		•	58.	10,	•	•	535.	46.	•9	72.	47		? .	815.	22.	•	•	۶.	7.	9	95.	12115. 0
TILC	F,L		, _		٠, ا ا	ت	3		5	9	-2384		ני (2/	23716.)	980	468	1351.	681)		-85	10	160	1017	440	883	30131.	1	8	-2214	7 7 7 7	1994	245	•	47	m O	343	44	40	11610.
GNIA	S		. J) : 	٦.		•	.,	S	14.2		• (77.	3.0		4 L	?	14.	101.8	. 22		2.8	\sim	S	4	; ;	• 6	• 22		2	~	14.0	;	• (77.		• (71	75	14.	01.	1.228
	POINT	RP #	11	ATTO	: 6	נ כ			1 , ₹	9	233.9	KAR) 	•	26	12	7 70	9 0	233 °	688	•		26	~	86.	33.	0.6880		•	26	14	85.	233.8	697	֭֓֞֝֜֜֜֝֜֜֜֝֓֓֓֓֜֜֜֜֜֓֓֓֓֓֓֜֜֜֜֜֓֓֓֓֓֜֜֜֜֡֓֡֓֡֓֡֓	•	76	0 H	٠,	30	233.7	687	•

SPND F8	SPND CB	FB .3R	CB . 3R	P LINK	CT**3/2	1205.	-3232.	2382.	-2874.	650.	0.001236
POWER	TORQUE,C	2,02	C0/8,C	3 :	P.M.C	834.	12959.	0.001112	0.01248	0.7483	0.7856
CT/S	CY/S	C2/S	CPM/S	CYM/S	C0/S	0.12925	00003	0.00050	0.00054	0.00092	0.01310
r)	Ç	23	CPM	CYM	g S	0.011516	000003	0.000044	0.000048	0.000082	0.001168
THRUST	SIDE	NORMAL	PITCH	YAW	TORQUE	35218.	-6-	135.	556.	957.	13604.
T, LC	SFILC	NF, LC	PW, LC	ZYLC	0,10	34907.	-440.	606	-1873.	3500.	13094.
AIND	PSIN	HOM,	TEMP	PRESS	RHO	2.5	2.	75.	14.2	101.8	1.228
ROR	POINT	X A A	VTIP	MTIP	7700	26	16	585.7	233.7	87	0.6

APPENDIX B

ROTOR WAKE DATA

Carrier Company		>	0.1		9.2	6.6	æ. •	5.0	5	-	.5	2.	(I)	4	•5		>	9.7	ď	3.7	٠ •	7	ن •	3.64	17	• 5
OF POOR QUALITY		œ	. 25	0.259	. 22	.24	• 21	.14	.01	.02	.00	.01	00.	00.	00.		٠ ٦	7.4	, 67	60	00	00.	.00	005	.00	00.
	<u>د:</u>	PS	01.75	101.763	01.76	01.76	01.70	01.77	01.76	01.76	61.77	01.78	01.79	01.78	01.78	s	Sd	01.81	01.76	01.80	01.70	01.71	61.71	101.747	01.78	61.18
25 11 0.000970 234.0 2.7 2.7	STATIC PFGB	Fd.	02.00	102.022	61.99	05.00	01.97	01.97	C1.78	01.79	01.77	91.19	01.79	01.19	01.19	CTICNAL PROBE	Ç.	02.05	01.83	01.80	01.69	01.71	01.71	101.738	01.78	01.79
RUN POINT CT VTIP WIND PSIV	PITUT-S	8/8	.20	0.221	.26	•2₿	• 33	• 42	• 62	.72	.30	.02	.07	.17	.22	OIPECTI	x/x	. 2 C	50	. 55	.75	• 30	.85	506-0	9	.10
		>	• 2	2.98	Ç.		4.2	.5		-	•	• 7	4.	'n	•		>	.2	ئى	3.1	~	6.	•	0.36		۲.
		œ	000	0.000	.02	.02	5.	13	• 03	00.	90.	00.	000	00.	000		7	3	.07	00.	.02	00.	00.	C 000° 2	٠ •	30.
	ري دي	S &	01.71	101.716	01.10	01.73	01.73	01.69	c1.76	01.17	01.78	01.78	01.79	01.78	31.79	v	PS	01.71	31.71	51.76	31.76	01.77	01.73	101.790	01.73	01.78
25 10 10 234.0 2.4 101.8	TATIC PROB	r d	01.71	101.721	01.72	01.75	01.72	01.82	ú1.30	01.77	01.78	01.78	01.79	01.79	01.79	AL PPOBS	э. Б.	01.70	61.19	C1.17	01.74	01.77	01.70	161.780	¢1.78	31.78
POINT CT VIIP VIIP PSIE	PITUT-S	R/R	. 26	0.721	• 20	. 28	. 33	77.	• 62	.72	9 8•	• 05	.07	.17	177	DIRECTION	R/R	. 23	. 50	. 65	.7	æ.	. 85	6.905	. 95	• 13

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RUN	25				RUN	25			
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Ţ	33.				TI	33.			
	•					•			
S	7				SI				
S					S S				
	(•				(()	9		
P1 101 -8	-STATIC PRUBE	S N			PITUT-S	STATIC PPUBE	ES		
K/ K	t d	PS	?	>	F/9	L d	S G.	•	>
. 20	02.13	01.84	.28	1.5	.26	32.12	61.86	• 5	0.5
. 22	02.14	01.84	.29	1.9	.22	02.17	01.87	• 29	2.0
0.265	102.167	101.852	0.315	22.58	0.265	102,213	101.891	0.322	22.82
. 26	02.15	01.55	.36	2.0	. 2 ŭ	02.22	01.88	.33	3.3
. 33	02.13	01.85	.28	1.3	.33	02.19	01.88	.31	2.3
. 42	02.07	01.62	.24	6.6	.42	02.15	01.86	• 29	1.8
.62	31.86	01.79	.07	7.0	.62	91.98	01.79	.18	7.2
. 72	01.19	01.77	.01	4.6	.72	01.77	01.77	000	
. 80	01.19	01.78	.01	• 5	.90	01.79	01.79	00.	æ
• 05	01.79	01.78	့		.02	01.19	01.19	8	0
.07	91.19	01.78	00.	6	.07	01.79	01.79	00	
. 17	01.79	01.78	.01	~	.17	01.79	91.73	00.	•
. 22	01.79	01.19	00•	4	• 55	01.79	01.79	8	0
DIRECT	RECTIONAL PROBE	ES			DIRECTIONAL	TONAL PROBE	S		
R/R	t d	PS	3	>•	R/R	PŦ	PS	σ	>
25	02.13	01.87	. 25	4.0	.20	02.17	01.91	. 26	0.5
.50	01.93	01.80	.13		.50	02.07	01.85	.21	8.8
• 65	01.10	01.73	.02	6.4	•65	01.89	01.78	. 10	æ
. 75	01.19	01.79	• 00	٠ .	.75	01.79	01.79	00.	1.6
. 80	01.76	01.75	. 20	. 7	• 80	01.19	01.78	00.	• 9
e U	01.77	01.77	00.	æ	• 8 ·	01.19	01.78	.00	٠,
0.905	101.861	101.791	0.011	4-14	30°0	101.793	101.789	0.005	2.71
50.	01.78	01.78	000	4	95	01.79	01.78	000	ຸນ
2	61.10	01.19	3	•	01.	01.19	01.19	9	• 2

		>	.241 19.7	220 18.8	0.317 22.66	.350 24.1	.363 24.2	.374 24.6	.421 26.1	.332 23.1	.147 15.4	.003 2.1	.009 3.7	.004 2.4	.009 3.8		>	.253 20.2	.352 23.8	.330 23.1	.234 19.4	.027 6.5	.031 7.1	0.007 3.46	.001 1.2
	<u>ನ</u>	PS	61.89	01.90	101.950	01.94	01.94	01.92	01.83	01.86	01.75	01.79	C1.19	01.79	31.79	S	PS	01.05	01.95	01.89	01.82	01.78	01.79	101.795	01.79
25 15 15 233.8 233.8 3.5 101.8	ATIC PROB	PT	02.13	02.17	102.267	02.30	02.31	02.20	02.25	02.13	01.89	01.19	01.7	51.10	01.80	CTIUNAL PRUBE	ργ	02.21	02.30	02.22	02.05	01.81	01.81	101.803	01.79
POINT CT	PITOT-ST	8/8	. 20	22	0.265	. 28	.33	. 42	.62	.72	.80	.02	.07	.17	.22	DIRECTION	R/R	.20	.50	•65	.75	9 R.	.85	0.905	3
		>	2.5		23.69	2.8	3.0	3.5	1 · J	6.0	•			8			>	9.1	1.7	~	1.0	4.1	0	2.42	7
		•	. 28	30	0.347	.32	.32	.33	.29	.15	00.	• 00	00.	9.	00•		œ	. 22	.29	. 23	.07	.01	.00	0.004	00
	es S	PS	01.84	01.0	101.895	01.89	01.90	01.86	01.80	91.18	01.78	01.78	01.78	01.78	01.79	10	PS	01.83	01.69	01.83	01.77	01.78	01.79	101.788	01.7A
N4.00 W • 8	PROBE	1	2.12	7.16	1	2.21	2.22	2.22	2.10	1.94	1.78	1.79	01.79	01.79	1.75	L PROBES	PT	02.11	2.18	2.06	1.95	1.79	1.79	C61.10	01.79
0.00448 233. 21.	I TOT-S TAT IC													_	_	RECTIONAL.		_	~	~	_			~	

PUN	1				RUN	7 + 6				
VTIP WIND PSIW PRESS	233.8 233.8 2.9 2.9 101.8				VTIP VTIP VIND PSIM PRESS	0.008731 233.8 2.8 2. 101.8				
-1011	TOT-STATIC PROBE	ES.			PITOT-S	STATIC PPOBE	හ: යු			
R/R	t a	PS	œ	>	R/R	đ.	න ය.	Œ	>	
. 20	02.13	01.90	.23	9.5	20	02.17	01.94	523	C,	
). 221	102.227	161,931	0.296	21.89	0.221	102,275	101.955	0.320		
. 26	02.30	01.95	.34	3.6	.26	02.35	01.97	37	4.7	
. 28	02.33	01.95	.37	4.5	.28	02.38	01.97	. 40	5.5	
•33	02.39	01.97	.41	5.8	.33	02.41	01.99	. 41	9	
. 42	02.37	01.96	.41	5.8	. 42	02.40	01.97	. 42	6.1	
• 62	02.39	01.86	.53	9.3	.62	02.44	01.88	. 56	0.1	
.72	02.28	01.81	.47	7.6	.72	02.39	01.83	. 55	0.0	
.80	02.03	01.68	.35	3.8	.89	02.06	01.71	. 35	3.9	
.02	01.78	01.78	00.	2.5	.02	01.80	01.79	.01	4.0	
6	61.78	01.78	.00	7	.07	01.80	01.79	00	.2	
17	01.78	01.78	.00	Ş	.17	01.80	01.79	00	6	
22	01.76	01.78	00.	7	•22	01.80	01.79	.00	3.75	
IRECTIONAL	IONAL PROBES	ં			DIRECTI	CTIONAL PROBE	ស			
R/R	F d	PS	Œ	>	R/R	F d	Sa	Œ	>	
2	02.21	01.95	.26	9.0	.20	02.22	01.96	36	, S	
20	02.40	01.98	41	5.9	50	02.44	02.01	4	6.4	
65	02.34	01.91	. 42	6.2	• 65	02.43	01.95	4	8.0	
75	02.20	C1.82	.37	4.6	.75	02.38	01.88	50	8.6	
8	01.85	01.65	.20	.2	.80	02.07	01.65	. 42		
82	01.82	01.16	• 05	9.4	• 85	01.76	01.72	.03	7.9	
8	01.78	01.76	.02	5	•90	01.78	01.78	00.	S	
.956	101.787	101.788	001	1.12	0.956	101.793	101.788	0.005	2-73	
2	01.19	01.78	0	4	•10	01.79	01.78	00	4	

		bs of A	01.945 0.329 23.	01.917 0.347 23.	01.955 0.457 27.1	01.9/2 0.418 20.	01.979 0.431 20.	01.935 0.437 27. 01.876 0.632 31.	01.807 0.654 32.	01.710 0.094 12.	01.776005 2.	01.781002 1.	01.787004 2.	01.787001 1.		A & Sd	01.949 0.255 20.	02.005 0.445 26.	01.907 0.499 28.	01.784 0.390 25.	01.734 0.266 20.	01.670 0.053 9.	01.789008 3.7	01.790 0.001 1.
25 0.009986 233.7 2.2 2.2 2.1	TATIC PROBES	P.	62.274	02.263	102.411 1	02.390	02.410	02.508	02.462	01.804	01.771	01.779	01.783	01.785	CTIONAL PROBES	Łd	02.204	02.450	02.406	02.175	02.000	01.723	101.781	01.792
PRESSE SERVICE	PITOT-S	R/R	.20	.22	0.265	07.	 	7.4	.72	.80	.02	.07	.17	• 22	DIRECTI	9/9	.20	.50	•65	.75	.80	• 85	506.0	.95
		>	2.7	4.0	25.25	4 1	יי מ מיי	200	7.3	1.2	1.2	6	5	7		>	0.8	6.5	C.	9.2	9.1	6.	2.45	4
		G	.32	.35	0.394	φ. Σ	• 40 • 40	1 T	46	.27	900	00.	90.	0		Œ	• 26	. 43	. 48	.52	.22	.01	0.004	80.
	S E	S d	01.95	96.10	101.970	16.10	01.99	01.00	01.80	01.71	01.80	01.77	01.77	01.78	S	ه	01.92	01.99	01.94	01.83	01.61	01.76	101.781	01.79
25 16 16 0.008820 233.8 2.3 2.3	TATIC PROBE	Fd	02.27	02.32	102.363	02.35	92.39	02-41	62.27	01.99	01.80	01.78	01.77	01.79	PROBE	Lo	02.19	02.43	02.42	02,36	01.84	01.17	101.784	01.79
ESS	I TOT-STATIC	R/R	20	22	- 265	82	3	75	2	80	6	0	11	22	DIRECTIONAL	F/R	20	50	55	75	80	S	• 905	2

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	>	4.8	7.1. 7.1.	8.2	6.4	2.95	. 70		>	9.9	2.4	5.9	50	1.58
	œ	.322	. 4 5	493	.822	.005			œ	. 24	.652	.397	8	005
v	ខ្ម	01.958 01.975	01.990 01.997	01.990	01.824	101.774 0 101.776 0	01.783		PS	01.978	02.006	01.165	01.667	101.778 -
25 21 233.6 233.6 1.8 101.8	I di	02.280	02.444 02.451 02.451	02.483	02.646	780	01.790	IONAL PROBES	1 d	02.223	02.658	01.562	01.660	101.777
RUN PUINT CT O VTIP MIND PSIN PRESS	2/8	.20	286	42	.80	1.023	.22	DIRECTIO	R/R	.20	.65	80	9.00	0.956
	>	2.8	. 25	7.4	4.3	3.18 2.05	30		>	80	2.6	2.1	0 4	20.0
	Œ	.26	43.42	46	.72	0.006			œ	. 23	.65	.63	000	0.003
ر د	PS	01.91	01.97 01.98 02.00	01.99	01.85	101.784	01.19		PS	01.93	01.89	01.67	01.66	
25 20 233.7 2.4 2.4 101.8	B	02.27	02.42 02.42 02.43	02.46 02.59	02.57	101.790	01.79	NAL PROBES	P.T	02.16	02.55	02.30	01.56 01.77	101.787
RUNT CT 0.019 VTIP 23 WIND PSIW PRESS 10	R/R	22.	. 28 . 33	. 42	.80	1.023 1.070	.22	DIRECTIONAL	R/R	0.205	.65	80	. 95	95

RUN POINT CT	242					362			
VTIP WIND PSIW PRESS)				VTIP WIND PSIW PRESS	1			
PI TOT-S	ITOT-STATIC PROBE	ES			PITOT-	STATIC PROB	ស		
F/R	10	o. Vi	œ	>	R/R	J.	S	æ	^
20	75 57	01 02	,		C	ננ כט	01.01	7	ر بر
<u>.</u>	17.70	24.43	٠. د د	• • • • • • • • • • • • • • • • • • •	, ,	77.70	10 10		֓֞֜֞֜֜֞֜֜֜֞֜֜֜֜֜֜֜֜֜֜֓֓֓֓֓֜֜֜֜֜֓֓֓֓֓֜֜֜֜֜֓֓֓֡֓֜֜֜֓֡֓֡֓֡֓֡֓֡֓
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97.	C4 • 7 0	20.20	. 40	? ;	07.	05.70	00.20	•	٠. د د
• 33	02.45	02.00	44	6.9	٠ ع	02.51	02.02	4	بر 80 م
. 4%	02.49	01.99	• 20	8 .4	• 42	05.20	02.01	• 48	8.1
• 62	02.72	01.90	.81	6.3	• 62	02.78	01.92	. 85	7.3
.72	02.73	01.81	.91	3.5	.72	02.83	01.85	.98	6.6
• 80	02.18	01.37	.81	6.3	.80	02.57	01.50	.07	1.6
.02	01.78	01.78	• 00		.02	01.77	01.77	80.	æ
.07	01.79	01.78	90.	Ş	.07	01.79	01.78	60.	S.
.17	01.78	01.78	.00	80	.17	01.78	01.79	00.	3
. 22	01.78	01.77	90	0	• 22	01.78	01.78	00.	•
DIRECTIONAL	IONAL PROBE	κi			DIRECTIONAL	IONAL PROBE	۶S		
R/R	PŢ	PS	Œ	>	R/R	P	S	æ	>
20	02.13	01.90	. 23	9.4	. 20	02.14	01.90	. 24	9.9
50	02,55	02.04	.51	8.7	.50	02.62	02.06	. 55	0.1
• 65	02.71	02.01	69.		. 65	02.73	01.92	.81	(1
. 75	02.73	01.91	.82	6.5	.75	02.80	01.90	• 90	8.4
. 80	92.10	01.29	.81	6.3	.80	01.62	01.12	.50	8.5
. 85	01.66	01.66	00.	2.0	.85	01.72	01.70	.02	5.7
0.905	101.732	101.730	0.002	1.85	0.905	101.713	101.677	0.036	7.62
. 95	01.74	01.74	.00	.2	.95	01.75	01.76	00	
• 10	01.78	01.78	00.	6	•10	01.17	01.77	00.	

	Œ	00000000000000000000000000000000000000	œ	0.285 0.775 0.842 0.066 0.010 0.003
, ,	S d	101.915 101.944 101.985 102.001 102.017 101.906 101.124 101.785 101.786	S d	101.952 102.076 102.025 101.910 101.567 101.723 101.793
25 24 -014217 234-1 2-9 26 101-8	PI	102.163 102.298 102.512 102.512 102.529 102.815 101.793 101.793 101.795	P.T.	102.237 102.651 102.800 102.752 102.621 101.614 101.733 101.790
RUN POINT CT O VIIP VIND PSIW PRESS	R/R	0.202 0.221 0.255 0.289 0.334 0.426 0.627 1.023 1.023 1.170 1.220	R/R	0.205 0.507 0.756 0.858 0.958

233.96 223.96 229.27 228.27 238.47 241.54 23.67 33.62 33.62 33.62

21.53 30.55 35.47 36.96 110.36 0.94 0.94 0.96

	>	20000	100 100 100 100 100 100 100 100 100 100	>	21.93 13.20 13.20 2.36 2.88 1.77 1.38
	œ	. 234 . 234 . 263 . 218	00000000000000000000000000000000000000	œ	000000000000000000000000000000000000000
	SS PS	01.74 01.78 01.77	101.791 101.791 101.751 101.751 101.770 101.786	s Ps	101.823 101.756 101.807 101.783 101.784 101.787 101.787
26 .000996 .234.0 .2.4 .101.8	ATIC PROBE	02.02 02.02 02.03	101.938 101.798 101.724 101.775 101.783 101.788	PROBE PT	102.118 101.663 101.311 101.773 101.781 101.781 101.787
MAT TO THE STATE OF THE STATE O	PITOT-ST R/R	222	00.00 00	DIRECTIONAL R/R	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0
	>	2000	10778771 107887871 108878999	>	611.3.1.4.4.6.5.3.0.9.1.6.5.0.0.0.1.1.4.4.6.5.5.0.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1
	œ	10000	00100000000000000000000000000000000000	œ	0.002 0.002 0.003 0.003 0.003
9	S A S	01.71 01.75 01.59	101.646 101.715 101.702 101.770 101.776 101.776	s Ps	101.763 161.784 101.771 101.771 101.779 161.778 101.758
1 2 3 3 7 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	TATIC PROBE	01.82 01.76 01.73 01.73	101.547 101.668 101.747 101.776 101.782 101.783	PRGBE PT	101.789 101.779 101.778 101.78 101.761 101.74 101.759
	P1 T0T-S1 R/R	22,28	0.428 0.428 0.720 1.023 1.170	DIRECTIONAL R/R	0.205 0.507 0.655 0.756 0.858 0.956 1.107

	>	21.48	2.6	3.0	2.5	2.0	e e	9	70	•	•	7		>	4.0	*	3.5	9	8.	8	3.38	<u>.</u>	6
	3	0.283	.31	.32	• 31	• 29	.17	• 02	3	•	000	8		σ	. 25	.21	.11	.00	99	00.	0.007	00.	00.
S	Sd	101.871	01.87	01.87	01.87	01.85	01.79	01.77	R/ - 10	7.10	01-78	01.78	v	PS	01.90	01.85	01.80	01.78	01.78	01.78	101.790	01.79	01.79
26 9 0.002962 234.0 3.3 101.8 STATIC PROBES	P.	102.154	02.19	02.20	02.18	02.15	01.96	01.79	01.10	00.10	01-19	01.79	IONAL PROBE	†d	02.16	02.06	01.91	01.79	01.19	01.79	101.797	01.19	01.79
RUN CT CT VTIP WIND PSIN PRESS	R/9	0.202	.26	.28	•33	• 42	•62	.72	5 c			.22	DIRECTIO	R/R	.20	•50	•65	.75	•80	•85	0.905	•95	•10
	>	20-14	1.8	2.1	1.6	8.4	1.6	`.	4, 6	•	0			>	0	4.8	٣,	4.	7	ထ္	1.47	6	•1
	•	0.249	.29	• 30	.28	• 20	80.	70.	9 6		00	00.		Œ	.24	.13	.01	.01	90	00.	001	90.	90
S	۲۶	101-836	01.85	61.84	01.84	01.82	01.78	01.77	01.18	01-10	01.78	01.78	v	PS	01.88	01.80	01.77	01.75	01.75	01.75	101.780	01.78	01.78
26 8 0.001881 234.0 2.4 2.4 101.8	P.	102.085	02.14	02.15	02.13	02.03	98-10	01.78	01.18	01.78	01.78	01.77	ONAL PRORES	14	02.12	01.94	01.19	01.73	01.76	01.74	101.779	01.78	01.77
POINT CT 0.001 VIIP 23 WIND PSIW PRESS 10	R/R	0.202	. 26	. 28	. 33	. 42	.62	. 72	200		117	. 22	DIRECTIONAL	R/R	. 20	.50	• 65	. 75	- 80	• 85	0.905	• 95	• 10

		>	22.05	4.6	5.2	4.7	4.8	5.8	1.6	2.9		-		5		Λ	1.5	3.4	ŝ	5.6	4.3	7.	5.93	6.	6
		œ	0.298	37	.39	.37	.37	4.	• 28	10	80.	00.	8	00.		œ	• 28	.33	.31	.15	.01	.01	022	.01	. 01
	رم در	PS	101.925	01.94	01.94	01.94	01.92	01.81	01.77	31.75	01.17	01.77	01.77	01.78	S	PS	01.97	01.92	01.84	01.77	01.73	01.75	101.758	01.77	01.77
26 11 0.006162 233.9 2.8 2.8 101.8	STATIC PROBE	₽ d	102,223	02.32	92.33	02.32	02.30	02.22	02.06	01.86	01.76	01.77	01.77	01.78	CTIONAL PROBE	Fd	02.25	02.26	02.16	01.92	01.74	01.73	101.746	01.75	01.76
POINT CT CT CT PSIN PSIN PSIN PSIN PSIN PSIN PSIN PSIN	PITOT-S1	8/R	9.202	20.	.28	•33	.42	.62	.72	.80	.02	.07	1.1	-22	DIRECTIC	R/R	.20	.50	.65	.75	.80	.85	0.905	•95	•10
		>	18.97	3.3	3.9	3.9	3.2	2.3	7.5	6.5	7	3.	Ģ	.2		^	1.0	6.0	9	9.6			2.84	σ.	Ö
		œ	0.221	. EE	.35	.35	.33	• 30	• 18	.02	00.	.00	00.	90		œ	.27	.27	.19	.05	.00	00	005	00.	• 00
	S	PS	101.898	01.91	01.91	01.91	01.89	01.81	01.79	01.77	01.79	01.19	01.79	01.79	ဟ	PS	01.95	01.89	01.82	01.78	01.77	01.78	101.782	01.78	01.78
26 10 10 233.9 2.8 2.8 101.8	ratic Probe	PT	102.119	02.25	02.26	02.26	02.23	02.12	01.98	01.79	01.79	01.78	01.78	01.79	PROBE	j.a £.,	02.22	02.16	02.01	01.84	01.77	01.77	101.777	01.77	01.78
RUN POINT CT VTIP WIND PSIW PESS	PITOT-STATIC	R/R	0.202	12	. 28	.33	. 42	. 62	.72	• 80	. 02	.07	.17	. 22	DIRECTIONAL	R/R	. 20	. 50	. 65	.75	.8	. 85	0.905	. 95	• 10

26 12 0.007800 233.8 3.0					POINT CT VTIP WIND	26 13 0.008819 233.8 2.8			
	<u>u</u>				S S S S S S S S S S S S S S S S S S S		3 1		
-	S.	3		>	R/R	-	ς, ο.	œ	>
02.189 101.934 0.25	01.934 0.25	.25		0	.20	02,25	01.95	30	7.4
02.249 171.948 0.30	11.948 0.30	.30		2.1	.22	02.28	01.95	. 32	3.1
02.357 101.966 0.39	√1.966 0.39	.39		5.2	.26	02.37	01.98	• 39	5.2
02.391 101.971 0.41	01.971 0.41	. 41		6.1	.28	02.40	01.97	.42	6.2
02.392 101.983 0.40	01.983 0.40	. 40		5.8	•33	02.41	01.99	. 42	6.3
02.377 101.958 0.41	01.958 0.41	-41		6.1	• 42	02.40	01.97	• 43	6.5
02.379 101.858 0.52	01.858 0.52	. 52		1.6	•62	02.43	01.87	• 56	e • •
02.266 101.809 0.45	01.809 0.45	. 45		7	.72	02.35	01.81	5.54	æ, 1
101.793 101.787 0.007	01.787 0.00	00		3.36	1.023	101-786	101.781	0.005	7.73
01.796 101.788 0.00	01.788 0.00	00.			.07	01.78	01.78	00	α,
01.795 101.787 0.00	01.787 0.00	90.		K	.17	01.79	01.78	00.	0
01.798 101.790 0.00	01.790 0.00	9		•	•22	01.78	01.78	• 00	9
RECTIONAL PROBES	S				DIRECT	RECTIONAL PROBE	တ		
D Sd Ld	S	Œ		>	R/R	P	PS	œ	>
02.302 102.001 0.301	02.001 0.301	.301	• •	2.1	.20	02.29	02.00	• 28	1.6
02.379 101.972 0.407 2	01.972 0.407 2	.407 2			.50	02.44	02.00	. 43	6.7
02.306 101.900 0.406 2	01.900 0.406 2	.406 2			.65	02.40	01.93	.46	9
01.991 101.723 0.268 2	01.723 0.268 2	.268 2		₩.	.75	02.17	01.77	. 40	5.6
01.762 101.689 0.072 1	01.689 0.072 1	.072 1		8	.80	01.67	01.60	• 06	0.3
01.778 101.737 0.041	01.737 0.041	.041	~	7	• 85	01.76	01.75	.01	5.1
01.760 101.727 0.033	01.727 0.033	•033	-	٣,	.90	01.78	01.77	60.	•
101.790 101.786 0.004	01.786 0.004	•00 •		2.64	0.956	101.786	101.785	0.001	1.20
01.789 101.788 0.00	00-0 88/-10	2		7.	07.	01.19	01.18	9	S.

The state of the s

OF POOR	QUALITY		>	ن • 0	3.1	. 7	6. 6	6.8	7.2	2.5	3.5	7.1	4.9	•	S.	2.02		>	0.4	7.4	.5	6	2.3	7.5	3.87	63	(7)
			Œ	. 24	• 32	40	• 43	. 44	• 45	•65	69.	. 45	.01	00	00	0.003		œ	25	46	57	5	33	.03	600	00 •	00.
		53	Sá	01.14	01.96	101.985	91.98	0.5.01	01.18	01.89	01.89	01.60	01.77	61.77	01.78	01.79	S	Sd	01.97	02.02	01.96	01.85	01.63	01.65	101.772	01.79	c1.78
26 15 0.010684	2.2	STATIC PRUR	PŢ	02.19	02.29	102,390	02.42	02.45	02.44	r2.54	02.49	02.06	01.79	01.78	01.79	01.79	SNAL PROBE	1 .	02.22	02.48	02.53	02.36	01.94	01.09	101,763	91.19	01.19
		PITOT-S	R/R	20	.22	0.265	• 2 8	•33	. 12	.62	.72	.80	.02	.07	.17	• 22	PIRECTION	3/0	200	.50	. 55	.75	.80	. x 5	6.965	.95	10
			>	.2	3.4	25.08	5.4	7.5	7.5	7	1.1	9.0	1.2	÷	C:	9		>	2.7	S. 0	7	5.5	4.4	2.3	3.22	~	.S.
			c	. 25	• 33	0.386	. 4.2	• 46	. 46	• 59	• 59	.54	00.	3.	.00	00		3	.31	.50	.54	.14	.12	00.	900.0	00.	• 00
		જ	PS	ii s	51.95	6.10	91.97	01.93	01.95	01.81	91.16	31.74	01.77	01.78	77.10	31.77	ະາ	PS	32.01	65.19	01.86	91.73	91.75	01.79	101.788	91.19	01.78
26 14 3.009912 233.8	2.	TATIC 940P	£4 Q.		02.29	02.37	72.40	02.44	02.41	02.46	12.36	02.29	31.78	01.17	01.77	01.77	930bE	1 d	32.33	92.49	02.36	01.87	91.37	31.7R	131.794	51.10	01.78
HGN CT CT	PSISON PRESS	P113T-5TA	R/2	0.232	. 22	• 20	97.	.33	4.	• 62	.72	. 33	.02	.07	. 17	• 22	DIRECTIONAL	K/K	.20.	.50	. 65	. 75	E3	. B.	6.905	. 35	.1.

OF POOR QUALITY

	>	22222222222222222222222222222222222222	>	22 28 28 27 27 27 27 37 37 37 37 37 37 37 37 37 37 37 37 37
	œ	00000 00000 00000 00000 00000 00000 0000	O	00000000000000000000000000000000000000
	ನ ೮ ೮	101.958 101.958 101.994 102.004 102.006 101.380 101.612 101.799 101.786	s S S	102.009 102.021 101.525 101.669 101.73 101.792
26 16 16 233.7 2.5 2.5 101.8	STATIC DPPB ot	102.232 102.337 102.454 102.454 102.547 102.554 101.762 101.768 101.766	UNAL PRORE	102.318 101.959 101.959 101.797 101.759 101.759 101.793
CTUNT CTUNT CTIP PSIA	P1 T0T-5	00000000000000000000000000000000000000	0183C11	00000000000000000000000000000000000000

TABLE 1. - ROTOR SYSTEM CHARACTERISTICS

Number of Blades	•		•	•														3
Rotor Radius																. 7	.62	m
Blade Chord			•													0.3	356	п
Rotor Solidity Ratio .																		
Blade Twist		•											-4	2	(1	nonli	nea	ar)
Blade Precone Angle																	2.	5 .
Rotor Airfoils							,						N	A(CA	64-	вегі	ies

TABLE 2. - PERFORMANCE AND LOADS DATA PARAMETERS

Label	Parameter
CB .3R	mean blade chordwise bending moment at .3R, N-m
COLL	blade collective pitch angle at .75 R, deg
CPM	rotor pitching moment coefficient, C_{PM}
CPM/S	rotor pitching moment coefficient over solidity, C_{PM}/σ
CQ	rotor torque coefficient, C_Q
CQ,C	rotor torque coefficient, corrected for wind, CQ, corrected
CQ/S	rotor torque coefficient over solidity, C_Q/σ
CQ/S,C	rotor torque coefficient over solidity, corrected for wind, $C_{Q,corrected}/\sigma$
CT	rotor thrust coefficient, C_T
CT/S	rotor thrust coefficient over solidity, C_T/σ
CT**3/2	$C_T^{3/2}$
CY	rotor side force coefficient, C_Y
CY/S	rotor side force coefficient over solidity, C_Y/σ
CYM	rotor yawing moment coefficient, C_{YM}
CYM/S	rotor yawing moment coefficient over solidity, C_{YM}/σ
CZ	rotor normal force coefficient, C_Z
CZ/S	rotor normal force coefficient over solidity, C_Z/σ
FB .3R	mean blade flapwise bending moment at .3R, N-m
FM	rotor figure of merit, FM
FM,C	rotor figure of merit, corrected for wind, FMcorrected
HUM,%	relative humidity, percent
MTIP	rotor tip Mach number, M _{tip}
NF,LC	rotor normal force measured by load cells, N
NORMAL	rotor normal force, N
P LINK	mean pitch link load, N
PITCH	rotor pitching moment, N-m
PM,LC	rotor pitching moment measured by load cells, N-m
POINT	data point number
POWER	rotor power, kW
PRESS	atmospheric pressure, kPa
PSIW	wind direction relative to rotor axis, ψ_w , deg
Q,LC	rotor torque measured by load cells, N-m
RHO	air density, p, kg/m³
RPM	rotor rotation speed, revs/minute
RUN	run number



TABLE 2. - continued

Label	Parameter
SF,LC	rotor side force measured by load cells, N
SIDE	rotor side force, N
SPND CB	mean blade spindle chordwise bending inoment, N-m
SPND FB	mean blade spindle flapwise bending moment, N-m
T,LC	rotor thrust measured by load cells, N
TEMP	air temperature, deg celsius
THRUST	rotor thrust, N
TORQUE	rotor toique, N-m
TORQUE,C	rotor torque, corrected for wind, N-m
VTIP	rotor tip speed, V_{tip} , m/s
WIND	wind speed, V_w , m/s
YAW	rotor yawing moment, N-m
YM,LC	rotor yawing moment measured by load cells, N-m

TABLE 3. - INDEX OF RUNS

OE POOR QUALITY

RUN Number	POINT NUMBERS	MTIP	CT/S	₩ IND	WAKE Data
14	15 - 27	0.69	0.000 - 0.157	0.5 - 1.7	СИ
15	3 - 15	0.69	-0.004 - 0.156	7.2 - 3.7	0.1
	16 - 31	0.69	-0.004 - 0.177	0.2 - 1.2	n0
	32 - 45	0.69	-0.003 - 0.158	0.8 - 1.8	40
	46 - 60	0.69	-2.002 - 0.169	0.9 - 2.1	מא
16	3 - 7	0.00	-0.004 - 0.161	0.7 - 2.3	NO
	8 - 11	0.66	0.088 - 0.162	2.3 - 2.7	NU
	12 - 15	0.69	0.091 - 0.160	2.4 - 2.6	NO
	16 - 19	1.73	0.095 - 0.168	2.1 - 2.9	HO
22	4 - 19	0.69	-0.003 - C.181	0.1 - 0.6	NO
23	3 - 17	0.69	-0.003 - 0.175	0.3 - 1.1	VO
	18 - 31	0.69	0.002 - 0.163	0.7 - 1.3	NO
	32 - 45	0.69	1.007 - 0.172	1.2 - 1.8	NO
24	3 - 6	0.73	0.094 - 0.171	0.0 - 3.7	ΝO
	10 - 11	0.60	0.088 - 0.120	1.3 - 1.4	טא
	13	0.52	0.096	1.8	ио
	14 - 17	0,65	0.090 - 0.160	1.5 - 1.7	NO
	18 - 20	0.63	0.143 - 0.161	1.8 - 2.0	ИО
25	10 - 24	0.69	-0.005 - 0.160	1.3 - 3.5	YES
26	6 - 16	0.69	-0.004 - 0.129	2.0 - 3.3	YES

TABLE 4. - LOCATION OF WAKE RAKE PRESSURE TAPS

Pitot-Static Probes

r/R	z/R
0.202	0.364
0.221	0.366
0.265	0.371
0.289	0.374
0.334	0.380
0.428	0.391
0.627	0.415
0.720	0.427
0.801	0.437
1.023	0.464
1.070	0.469
1.170	0.482
1.220	0.488

Directional Probes

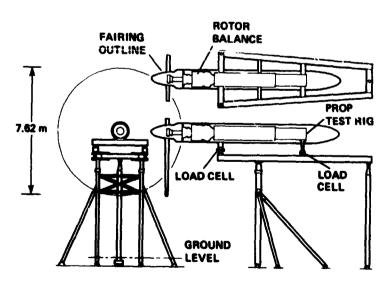
r/R	z/R
0.205	0.292
0.507	0.329
0.655	0.347
0.756	0.359
0.806	0.365
0.858	0.372
0.905	0.377
0.956	0.384
1.107	0.402

TABLE 5. - PRESSURE DATA PARAMETERS

Label	Parameter
СТ	rotor thrust coefficient, C_T
POINT	data point number
PRESS	atmospheric pressure, kPa
PSIW	wind direction relative to rotor axis, ψ_w , deg
PS	wake static pressure, P _S , kPa
PT	wake total pressure, P_T , kPa
Q	wake dynamic pressure, $P_T - P_S$, kPa
\mathbf{R}/\mathbf{R}	pressure tap radial station, r/R
RUN	run number
V	wake velocity, m/s
VTIP	rotor tip speed, V_{tip} , m/s
WIND	wind speed, V_w , m/s

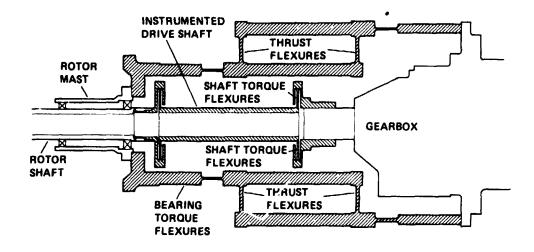
ORIGINAL FACE IS OF POOR QUALITY

1. Outdoor Aerodynamic Research Facility with Prop Test Rig.



2. Prop Test Rig with XV-15 Rotor.

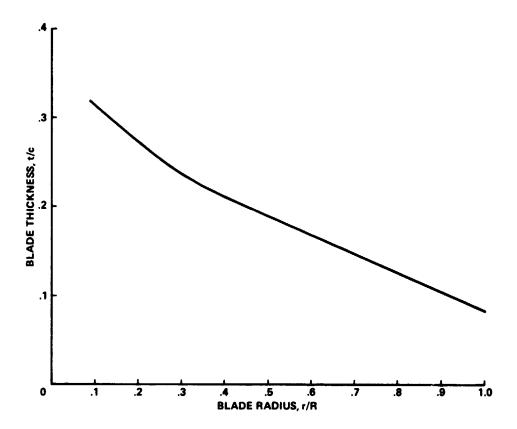
3. Prop Test Rig with XV-15 Rotor.



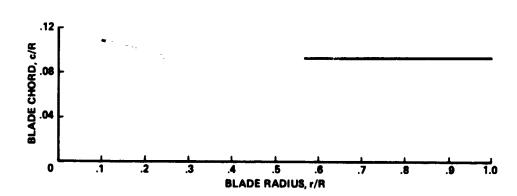
4. Rotor Balance System.



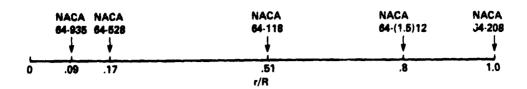
5. Rotor Blade Twist Distribution.



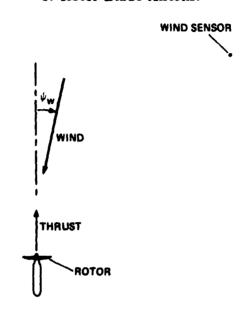
6. Rotor Blade Thickness Distribution.



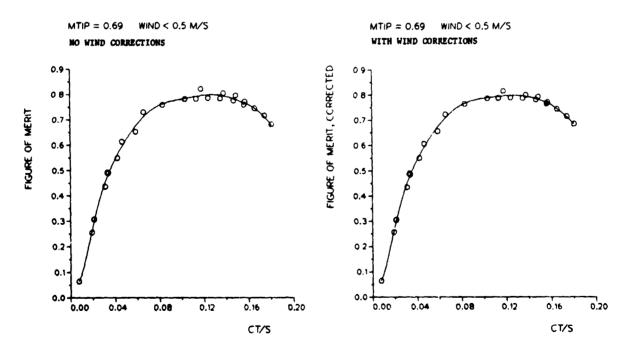
7. Rotor Blade Chord Distribution.



8. Rotor Blade Airfoils.

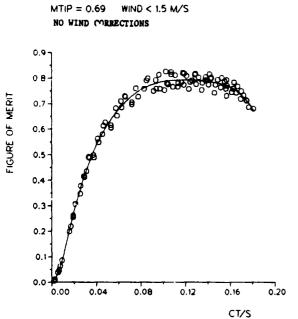


9. Wind Sensor Location.

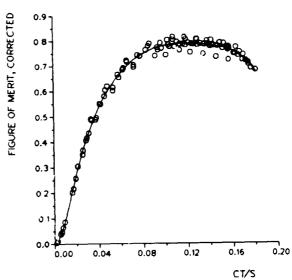


(a) Wind < 0.5 m/s.

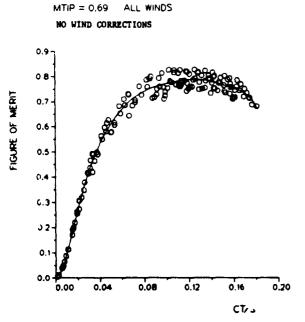
10. Effect of Wind Corrections on Rotor Performance.



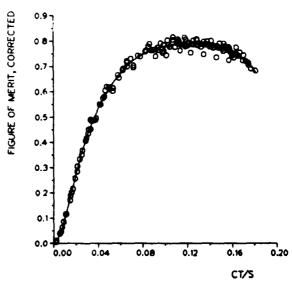
MTIP = 0.69 WIND < 1.5 M/S WITH WIND CORRECTIONS



(b) Wind < 1.5 m/s.

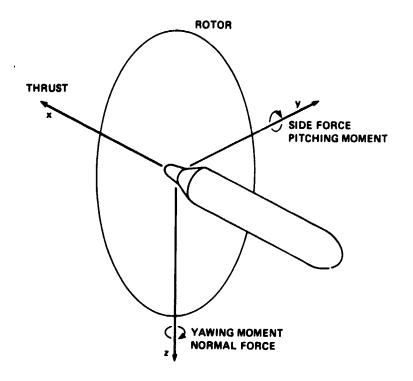


MTIP = 0.69 ALL WINDS WITH WIND CORRECTIONS

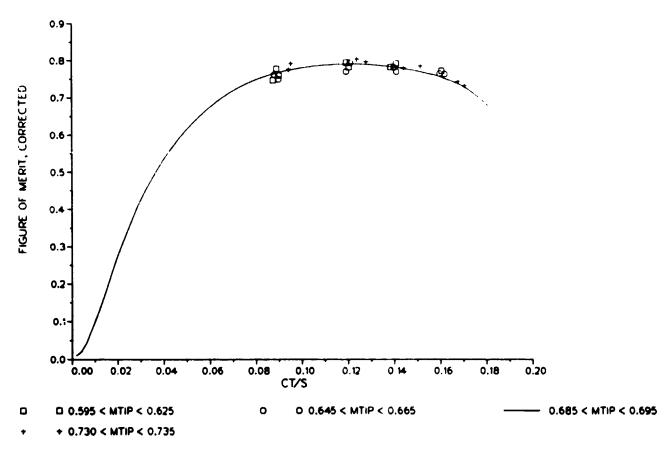


(c) All Winds.

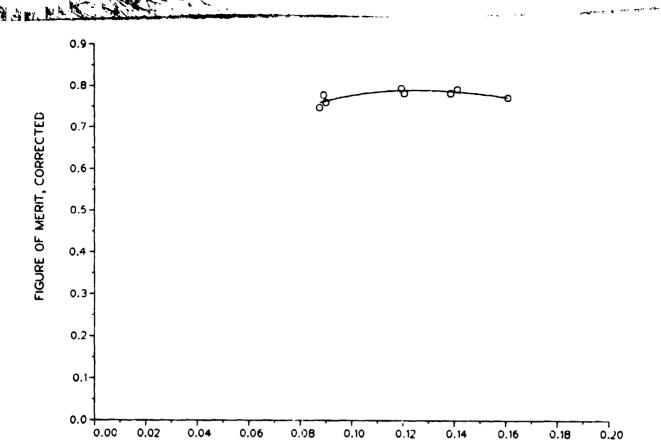
10. Concluded.



11. Rotor Balance Axis System.

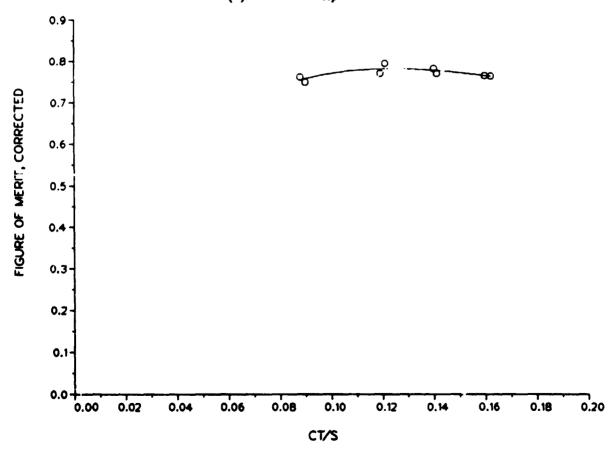


12. Effect of Tip Mach Number on Rotor Performance.



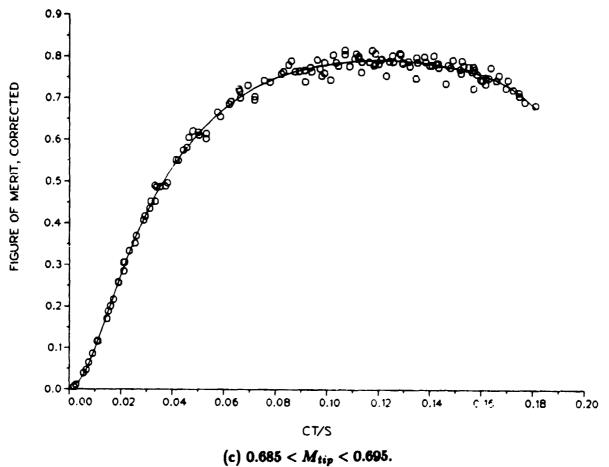
(a) $0.595 < M_{tip} < 0.625$.

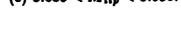
CT/S

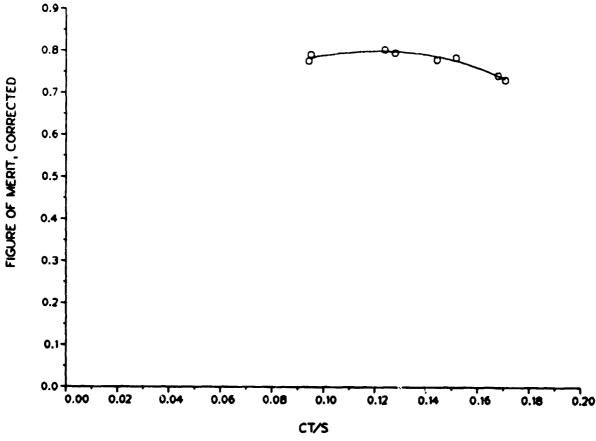


(b) $0.645 < M_{tip} < 0.665$.

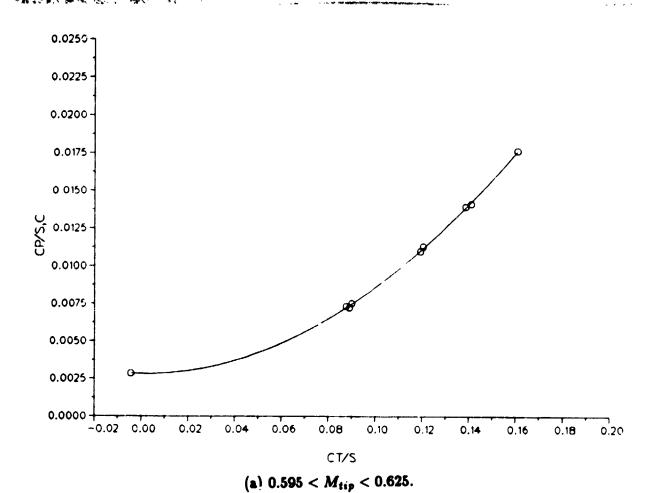
13. Effect of C_T/σ on Rotor Performance.

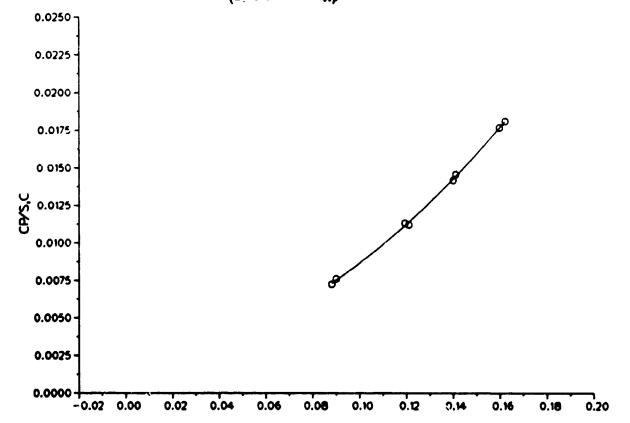






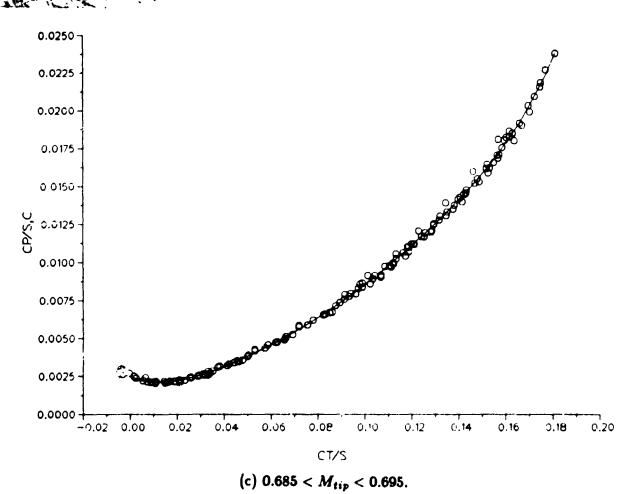
(d) $0.730 < M_{tip} < 0.735$. 13. Concluded.

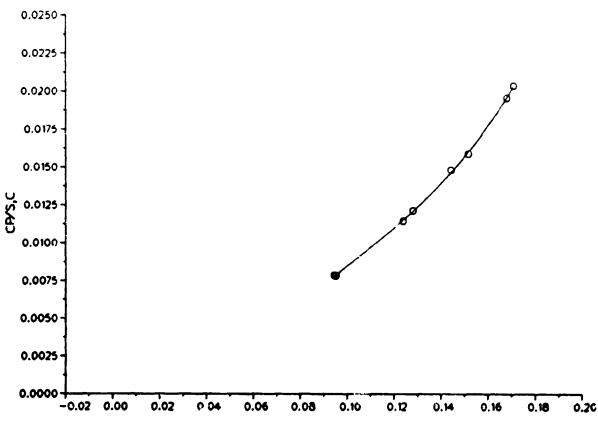




(b) 0.645 < M_{tip} < 0.665.
 14. Effect of C_T/σ on C_{P,corrected}/σ.
 78

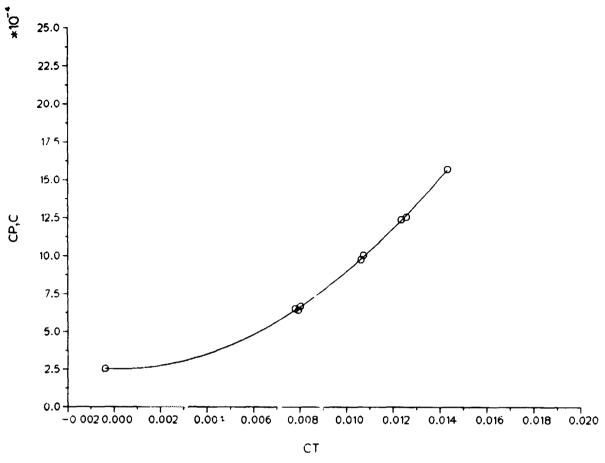
CT/S



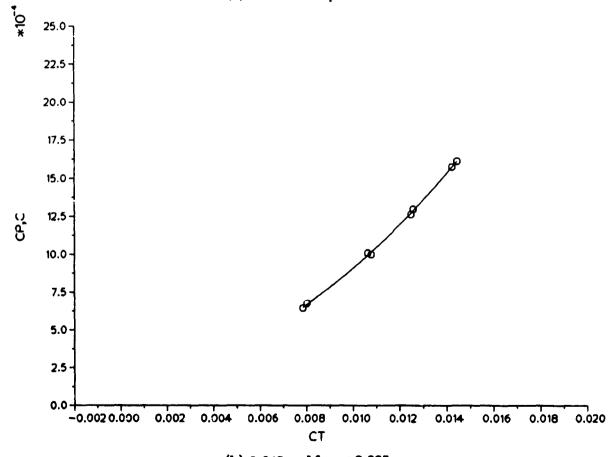


(d) $0.730 < M_{tip} < 0.735$. 14. Concluded. 79

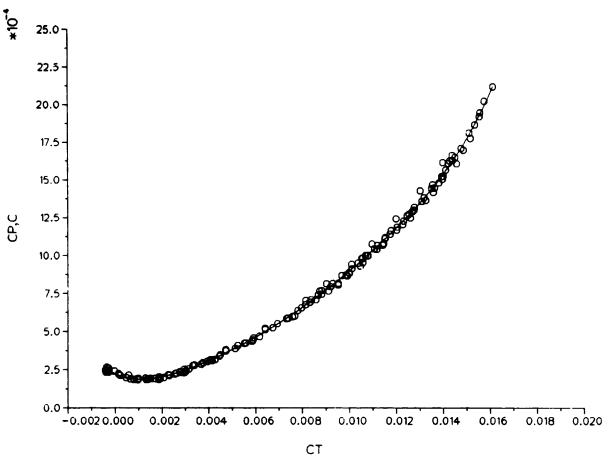
CT/S



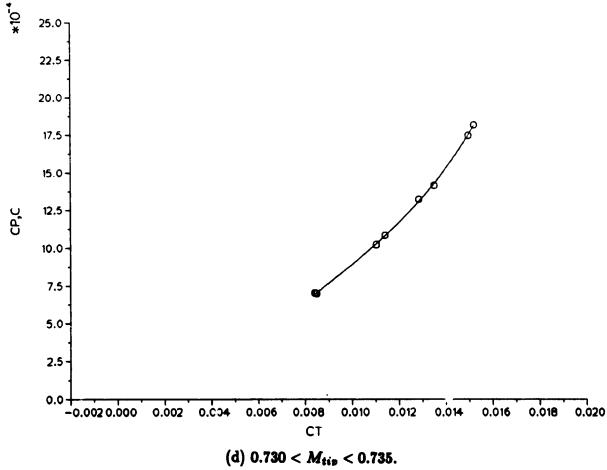
(a) $0.595 < M_{tip} < 0.625$.



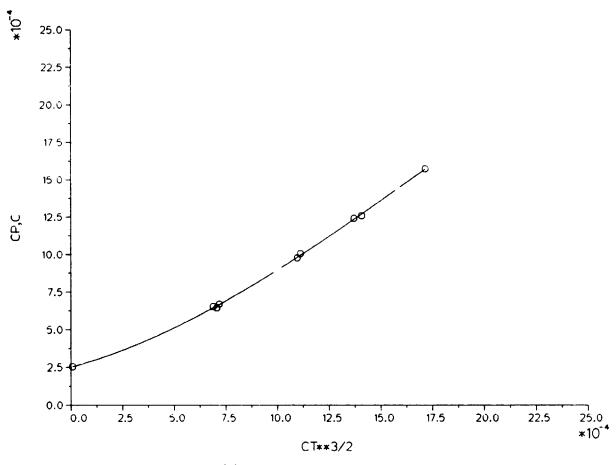
(b) $0.645 < M_{tip} < 0.665$. 15. Effect of C_T on $C_{P,corrected}$.

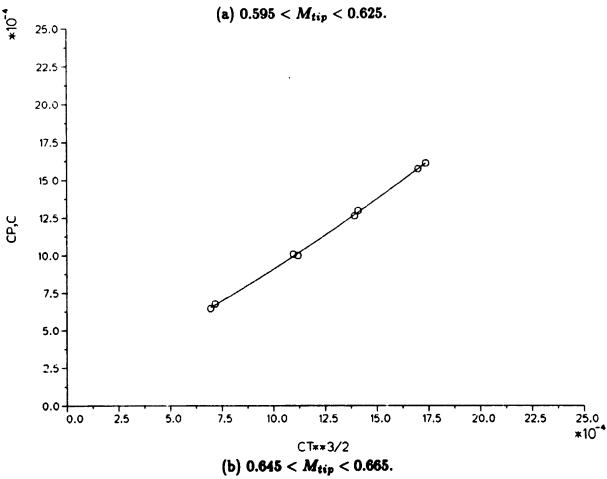




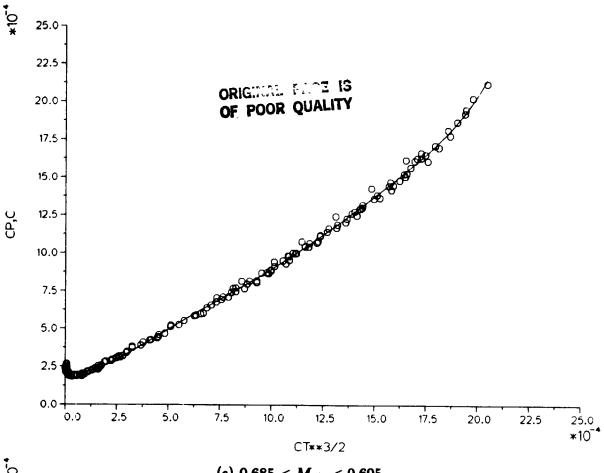


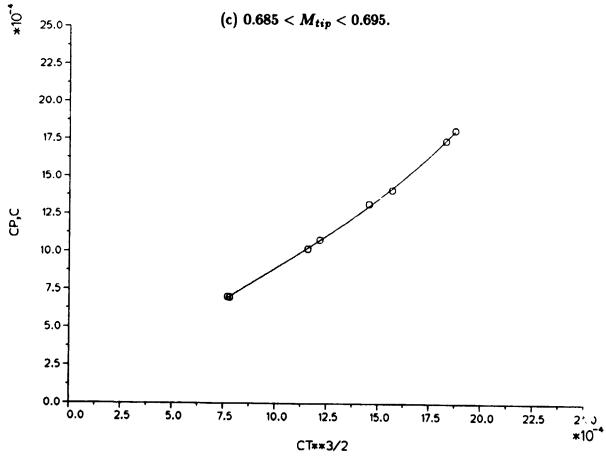
15. Concluded.





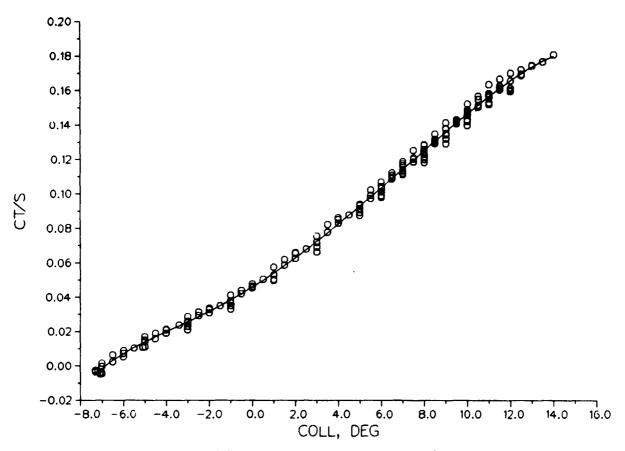
16. Effect of $C_T^{3/2}$ on $C_{P,corrected}$.





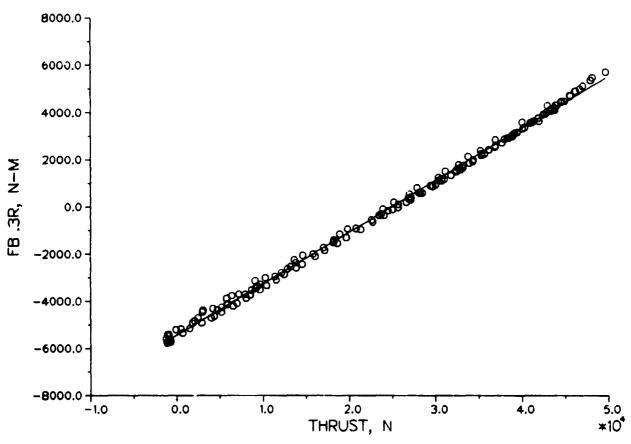
(d) $0.730 < M_{tip} < 0.735$.

16. Concluded.

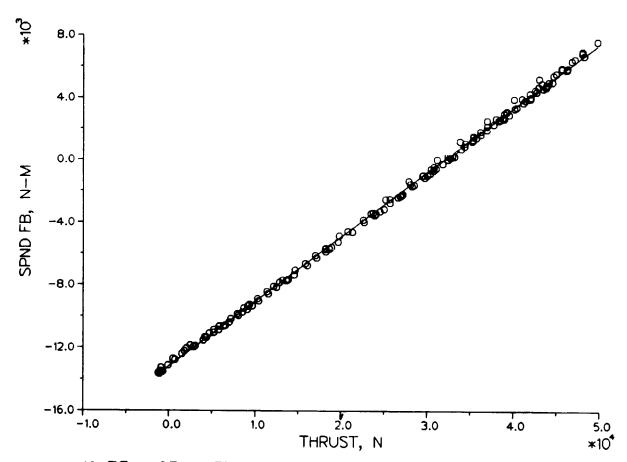


THE LAND COMMENT

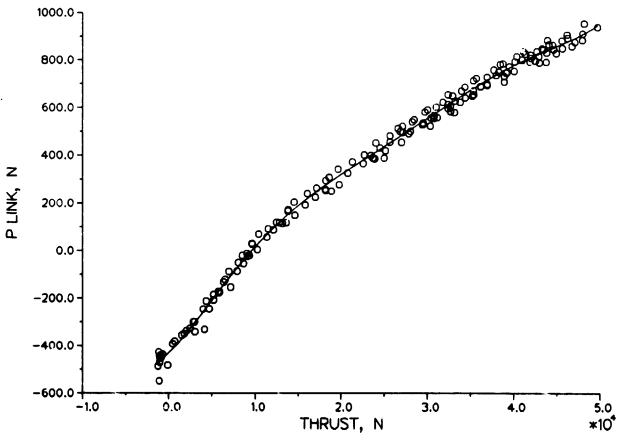
17. Effect of Collective Pitch on C_T/σ .



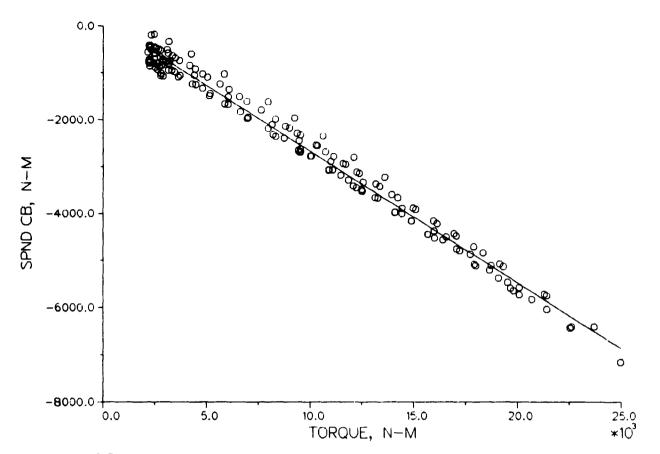
18. Effect of Rotor Thrust on Hub Spindle Flap Bending Moment at 0.06 R.



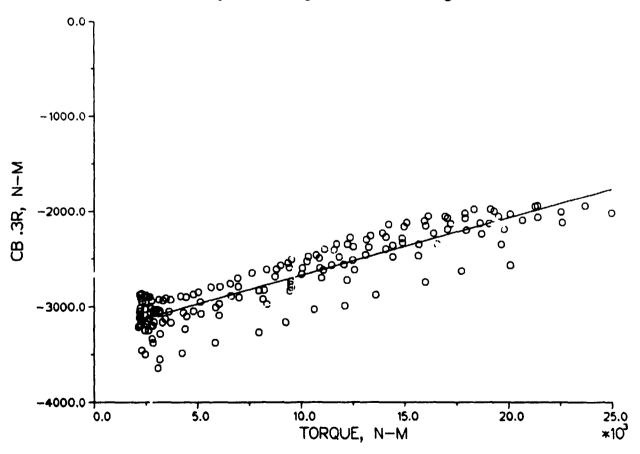
19. Effect of Rotor Thrust on Blade Flap Bending Moment at 0.3 R.



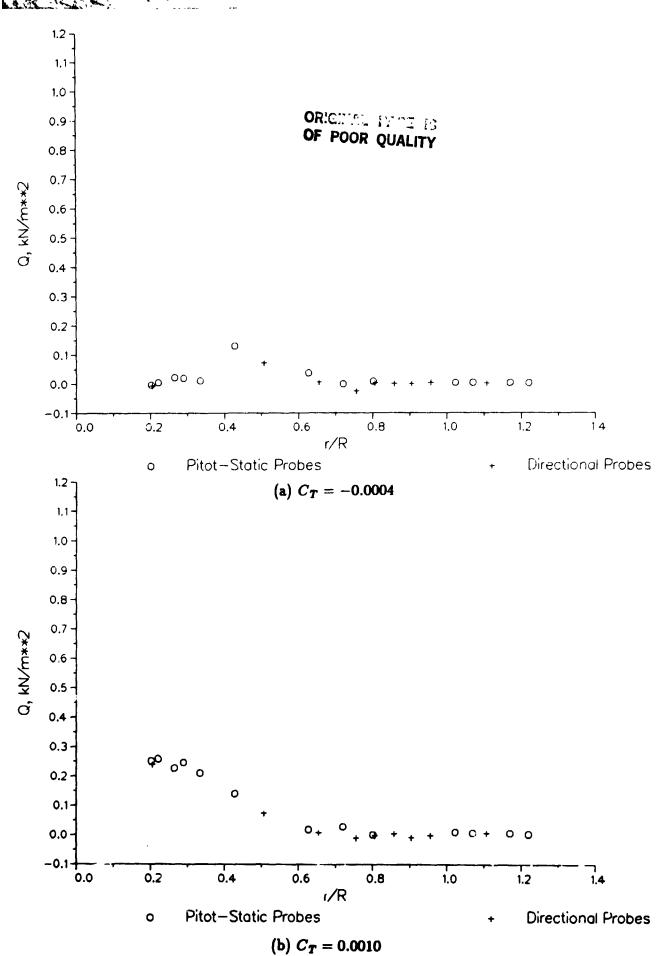
20. Effect of Rotor Thrust on Pitch Link Load.



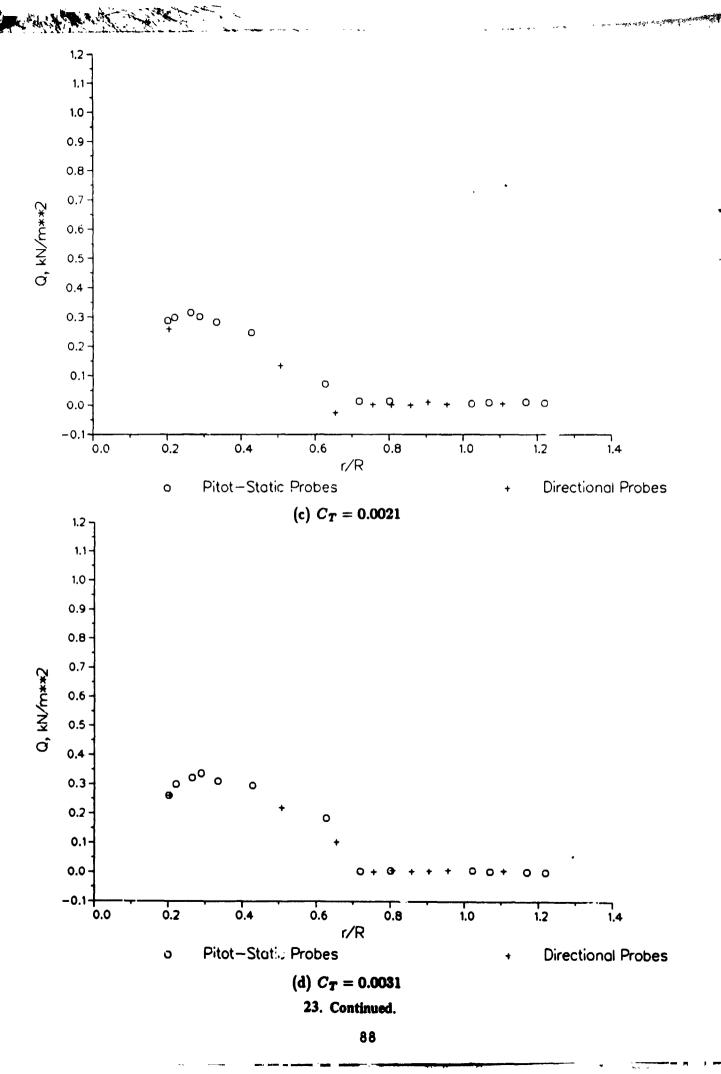
21. Effect of Rotor Torque on Hub Spindle Chord Bending Moment at 0.06 R.

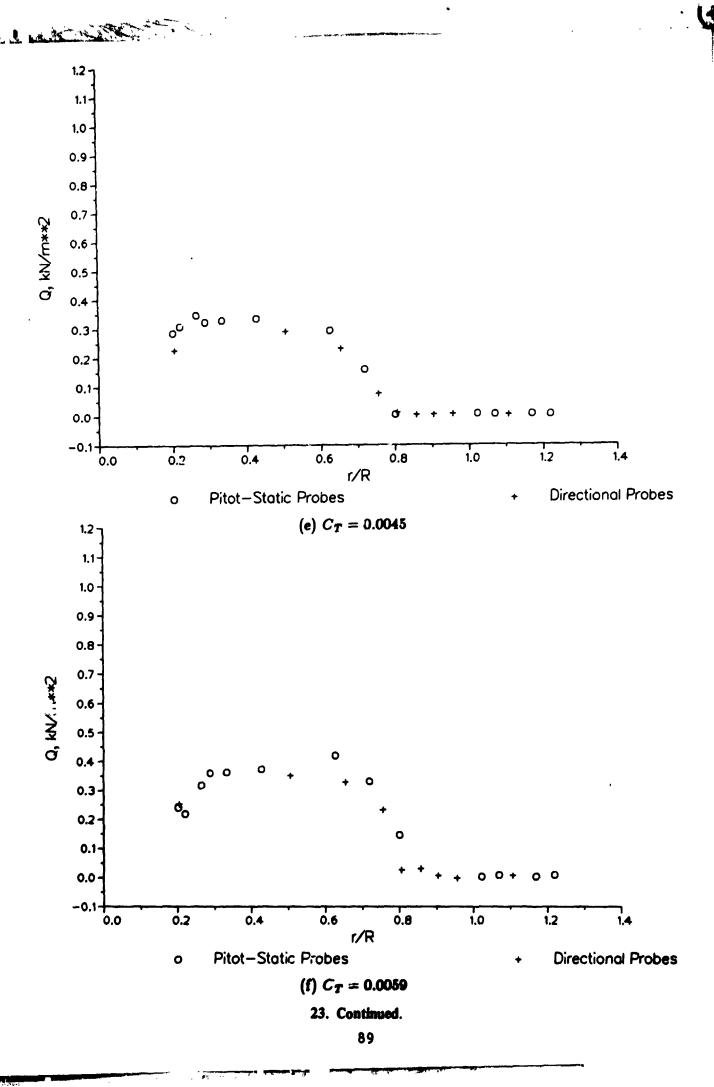


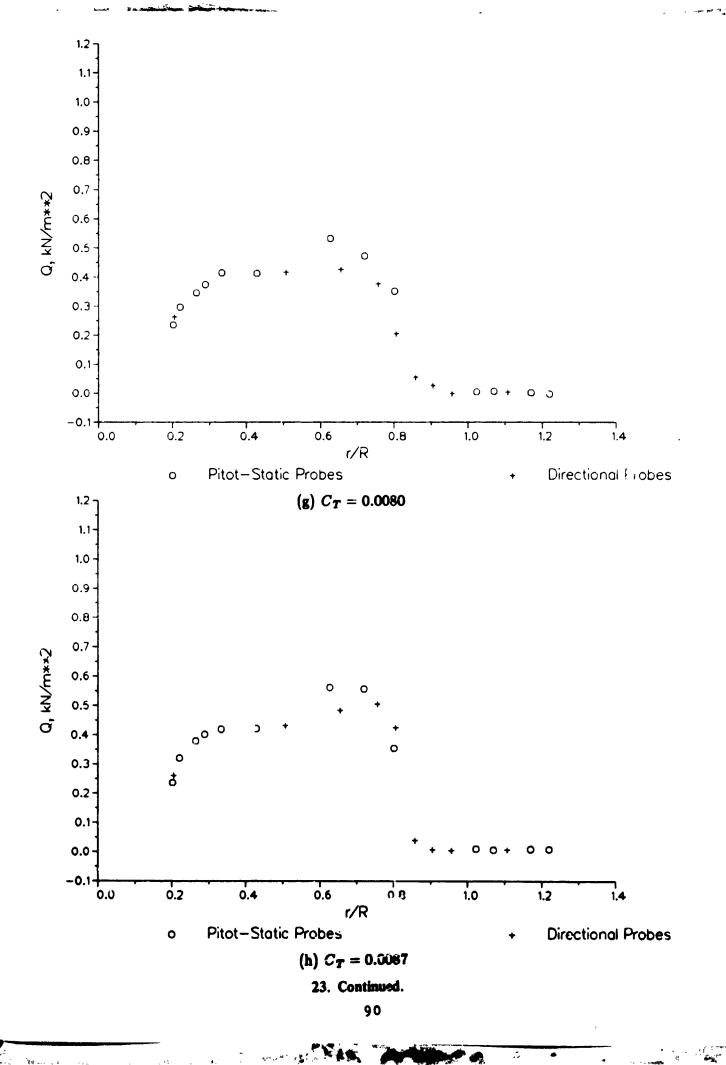
22. Effect of Rotor Torque on Blade Chord Bending Moment at 0.3 R.

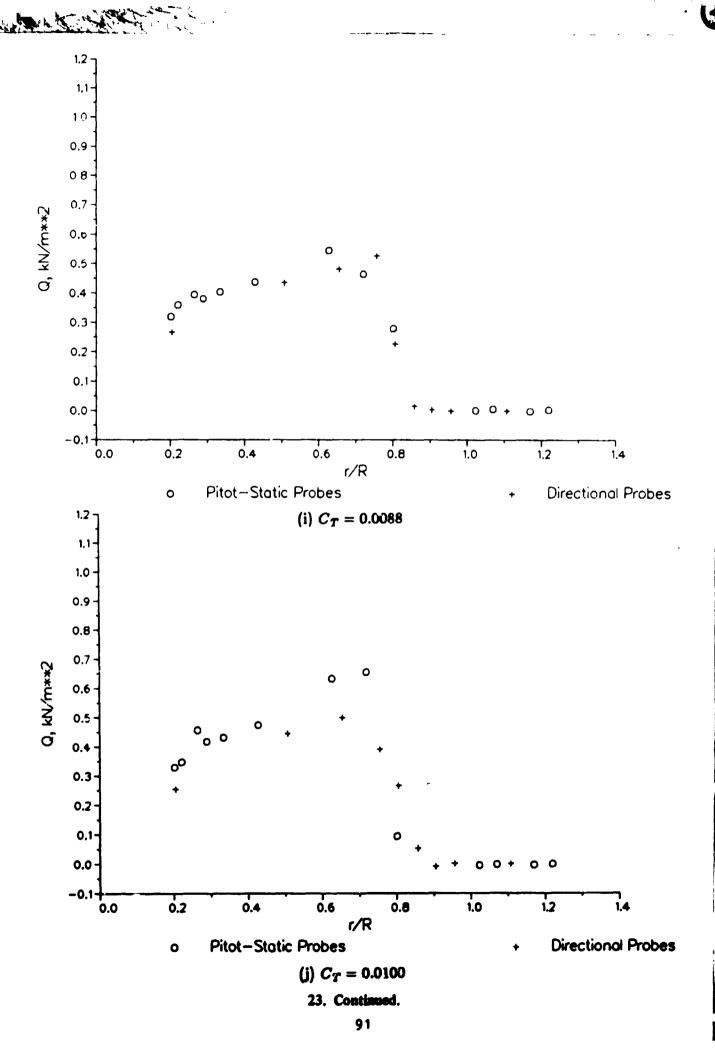


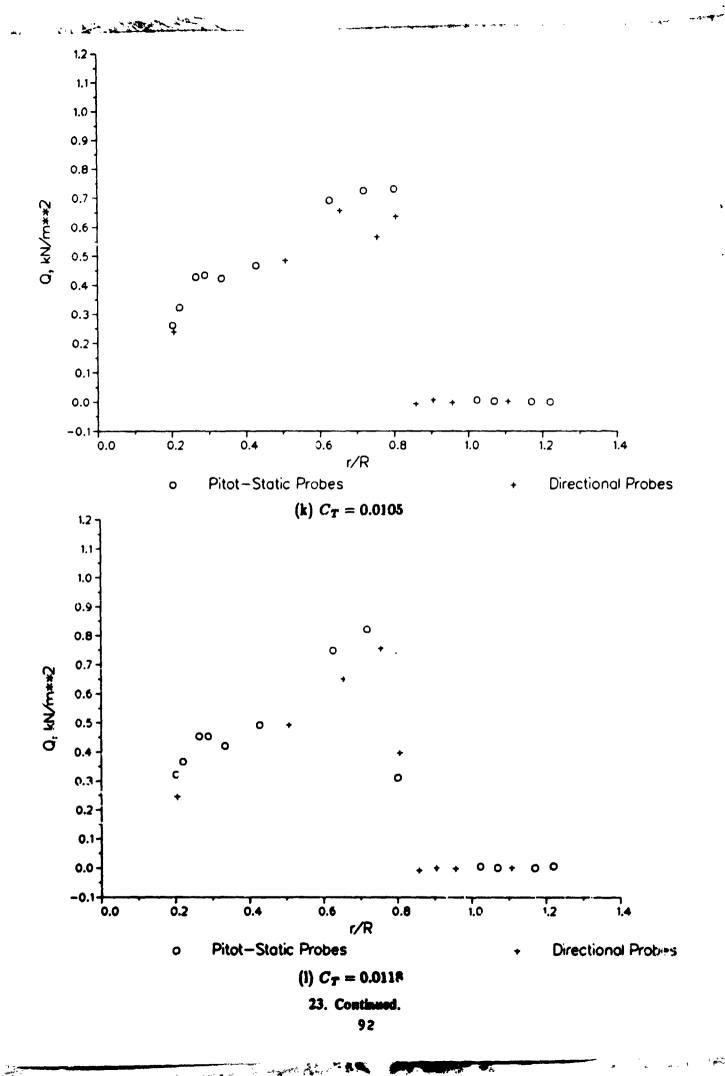
23. Rotor Wake Dynamic Pressure Distribution.

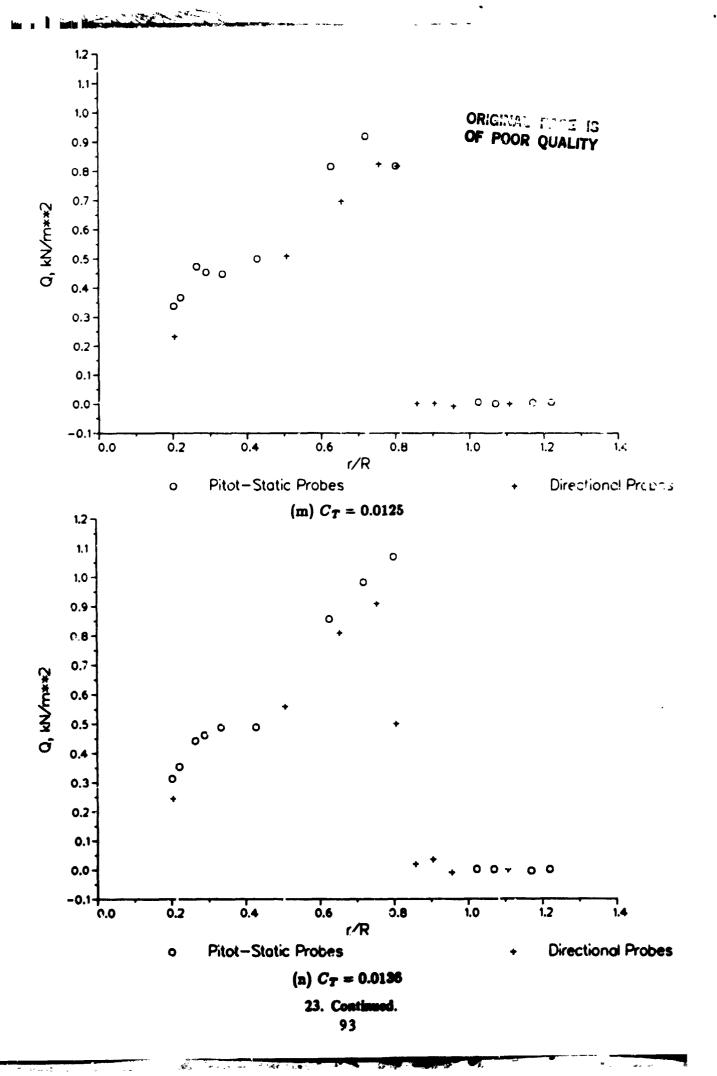


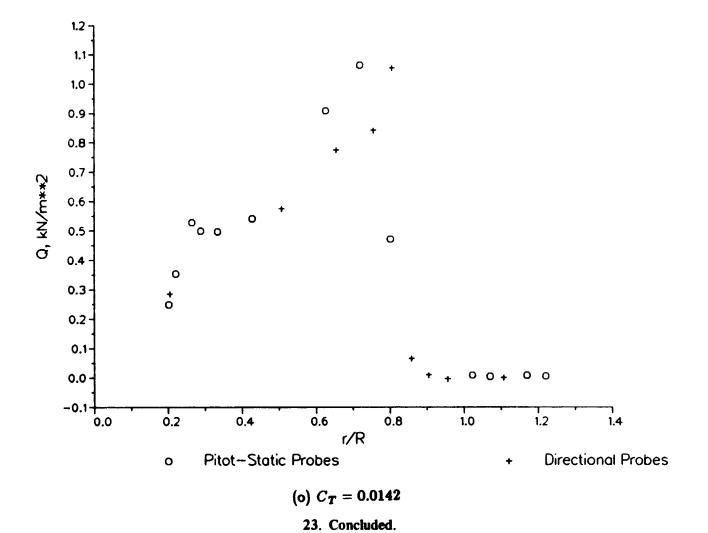












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