

Objective:

The goal of this study is to perform a **linear static analysis** and subsequently conduct a **buckling (stability) analysis** of a curved shell structure using **Code_Aster**. The structure includes **stiffeners** and **ribs**, and the analysis is focused on extracting **critical buckling modes** and understanding structural behavior under a specified moment load.

1. Initialization

DEBUT()

We begin the script using `DEBUT()`, which is required in every Code_Aster command file to initialize the session.

2. Mesh Import

```
mesh = LIRE_MALLAGE(FORMAT='MED', UNITE=20)
```

- **Why:** We import a `.med` mesh file generated earlier (e.g., from Gmsh).
 - **Choice of format:** `'MED'` is the standard mesh format used in Code_Aster.
 - **Unit 20:** Matches the unit used in the `.export` file.
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3. Model Definition

```
model = AFFE_MODELE(...)
```

- **MODELISATION = 'DKT':** We chose `'DKT'` (Discrete Kirchhoff Triangle), a shell element suitable for thin-walled structures and plate bending, especially where curvature and rotational DOFs are important.

- **PHENOMENE = 'MECANIQUE'**: We're dealing with mechanical deformations.
 - **TOUT = 'OUI'**: Applies the model to all mesh elements.
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4. Elemental Properties

elemprop = AFFE_CARA_ELEM(...)

- We defined varying **thicknesses (EPAIS)** for different parts:
 - **Curved surface**: 5 mm (very thin shell).
 - **Stiffeners**: 25 mm (adds longitudinal stiffness).
 - **Ribs**: 50 mm (transverse stiffening).
- **VECTEUR** specifies the **local orientation** of the shell elements for correctly applying thickness and stiffness directions.

Why multiple thicknesses?

To simulate realistic structural behavior. Curved shells alone are weak in buckling; stiffeners and ribs provide directional rigidity.

5. Material Properties

mater = DEFI_MATERIAU(...)

- We defined **isotropic elasticity** with:
 - **E = 70 GPa**: Common for Aluminum alloys.
 - **NU = 0.3**: Poisson's ratio, standard value for metals.

fieldmat = AFFE_MATERIAU(...)

- Applies the material to the entire model.
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6. Boundary Conditions

bc = AFFE_CHAR_MECA(DDL_IMPO=...)

- All degrees of freedom (DX, DY, DZ, DRX, DRY, DRZ) are constrained for the **fixed** group.
 - This creates a **fully clamped edge** or zone, essential to simulate realistic fixed support in buckling problems.
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7. Loading

load = AFFE_CHAR_MECA(FORCE_NODALE=...)

- A **moment (MX = 1.0)** is applied to the group **Node**.
 - This replicates a **unit moment** applied to trigger buckling.
 - The direction (MX) aligns with axis **X**, making it easy to interpret buckling about the X-axis.
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8. Static Analysis

reslin = MECA_STATIQUE(...)

- Solves the **linear static problem** at time **INST=1.0**.
 - **OPTION = 'SIEF_ELGA'**: Stores **stress resultants (forces and moments)** at Gauss points, which are later used to compute geometric stiffness (KG matrix).
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9. Post-Processing: Result Export

IMPR_RESU(FORMAT='MED', RESU=...)

- Exports linear analysis results for visualization in tools like Salome.
 - **UNITE=80** is a standard choice for exporting results.
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10. Extract Stress Field

field = CREA_CHAMP(...)

- **Purpose:** Extract the stress field from the static result (**SIEF_ELGA**) to compute the **geometric stiffness matrix** used in buckling analysis.
 - **Why:** Euler buckling requires initial stress fields to assess how close the structure is to instability.
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11. Stiffness Matrix Assembly

ASSEMBLAGE(...)

- Generates two matrices:
 - **KM:** Linear mechanical stiffness (**RIGI_MECA**).
 - **KG:** Geometric stiffness (**RIGI_GEOM**), derived from pre-stress.
 - **Why needed:** Buckling is an eigenvalue problem that solves:
 $(KM - \lambda KG)\phi = 0$
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12. Buckling Mode Calculation

modes = CALC_MODES(...)

- **TYPE_RESU = 'MODE_FLAMB':** Requests **buckling modes** (not vibration).
 - **OPTION = 'PLUS_PETITE':** Finds **lowest (most critical) eigenvalues** (first to buckle).
 - **NMAX_CHAR_CRIT = 12:** Calculates first 12 buckling modes.
 - **SOLVEUR_MODAL = 'SORENSEN':** A robust eigenvalue solver suitable for large problems.
 - **PREC_SHIFT, SEUIL_CHAR_CRIT:** Precision settings for numerical stability.
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13. Export Buckling Modes

IMPR_RESU(FORMAT='MED', ...)

- Saves **buckling mode shapes** for viewing in Salome or ParaVis.
 - You can visually inspect **how the structure deforms when it buckles**.
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14. End of Analysis

FIN()

Marks the end of the script.

Summary of Choices:

Feature	Why Chosen
<code>MODELISATION= 'DKT'</code>	Shell formulation suited for curved thin structures
Variable <code>EPAIS</code>	Realistic modeling of stiffeners and ribs
Unit moment <code>MX</code>	For normalized buckling analysis
<code>SIEF_ELGA</code> extraction	Required for calculating KG
<code>RIGI_MECA</code> , <code>RIGI_GEOM</code>	Fundamental matrices for buckling problem
<code>SORENSEN</code> solver	Efficient and accurate for modal/buckling problems
Export to <code>.med</code>	Easy integration with Salome for visualization
