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[TITLE]		1

Contents

Notes on Notation	
1. Introduction	1
1.1. Copernicus, Kepler, and Newton	1
1.2. Neptune & Halley's Comet	1
1.3. Einstein and Space-time	1
2. Fundamentals of Orbital Mechanics	2
2.1. Newton's Laws	2
2.2. Kepler's Laws of Planetary Motion	3
3. The <i>n</i> -body Problem	5
4. Conclusion	6
Appendix A. References	

NOTES ON NOTATION

Vectors will be bold with an arrow on top, i.e., $\vec{\mathbf{v}}$ is a vector. All vectors herein will be elements of the vector space \mathbb{R}^3 .

1. Introduction

- $1.1.\ Copernicus,\ Kepler,\ and\ Newton.$
- 1.2. Neptune & Halley's Comet.
- 1.3. Einstein and Space-time.

2. Fundamentals of Orbital Mechanics

First we have to start with some basic definitions.

Definition 2.1 (Distance). The Euclidean distance formula for points $(x_1, y_1, z_1), (x_2, y_2, z_2) \in \mathbb{R}^3$ is

$$d((x_1, y_1, z_1), (x_2, y_2, z_2)) = \sqrt{(x_1 - x_2)^2 + (y_1 - y_2)^2 + (z_1 - z_2)^2}.$$

Note, this also works for points in \mathbb{R}^2 , where $z_1 = z_2 = 0$.

Definition 2.2 (Circle). Let $x, y, r, h, k \in \mathbb{R}$. Then, we say that the set of points (x, y) of equal distance r from a point (h, k) is a **circle**, i.e.,

$$C = \{(x,y) : (x-h)^2 + (y-k)^2 = r^2 \text{ for } x, y, r \in \mathbb{R} \}$$

Definition 2.3 (Ellipse). Let $x, y, a, b, h, k \in \mathbb{R}$. Without loss of generality assume $b \le a$. Then, we say that the set

$$E = \left\{ (x, y) : \frac{(x-h)^2}{a^2} + \frac{(y-k)^2}{b^2} = 1 \right\}$$

is an **ellipse** centered about the point (h,k) whose **major axis** is of length b, and whose **minor axis** is of length a. The **eccentricity** of an ellipse is defined as

$$e = \sqrt{1 - \frac{b^2}{a^2}},$$

this helps us define the foci of an ellipse. The two **foci** of an ellipse are points lying on the major axis, they are $F_1 := (-ae, 0)$ and $F_2 := (ae, 0)$. With these characters we can also define an ellipse thus,

$$E = \{(x, y) : d((x, y), (-ae, 0)) + d((x, y), (ae, 0)) = 2a\}$$

where d is the Euclidean distance.

2.1. Newton's Laws.

Theorem 2.4 (Newton's 1st Law of Motion). Every body persists in its state of being at rest or of moving uniformly straight forward, except insofar as it is compelled to change its state by force impressed.

This describes what we call inertia, or uniform motion. This statement has been refined over time, a modern statement of Newton's 1st Law is along the lines of: "If the sum of force vectors acting on an object is zero, then and only then is the velocity of the object is zero." We can express this mathematically as

$$\sum \vec{\mathbf{F}} = 0 \iff \frac{d\vec{\mathbf{v}}}{dt} = 0.$$

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Theorem 2.5 (Newton's 2nd Law of Motion). The change of momentum of a body is proportional to the impulse impressed on the body, and happens along the straight line on which that impulse is impressed.

Often times this law is stated as "force equals the product of mass and acceleration," however this is more of a consequence of the law, rather than a statement of the law itself. Recall that momentum is defined as $\vec{\rho} = m\vec{\mathbf{v}}$, where m is a constant. Therefore in the language of differential calculus, we can describe the change of momentum of a body thus,

$$\vec{\mathbf{F}} = \frac{d\vec{\boldsymbol{\rho}}}{dt},$$

$$= \frac{d(m\vec{\mathbf{v}})}{dt},$$

$$= m\frac{d\vec{\mathbf{v}}}{dt},$$

$$= m\vec{\mathbf{a}}.$$

This ability to express the force on an object as a derivative is very useful when it comes to differential equations.

Theorem 2.6 (Newton's 3rd Law of Motion). To every action there is always opposed an equal reaction: or the mutual actions of two bodies upon each other are always equal, and directed to contrary parts.

Most people know this law as "every action has an equal and opposite reaction."

Theorem 2.7 (Newton's Law of Universal Gravitation). Every point mass attracts every single other point mass by a force acting along the line intersecting both points. The force is proportional to the product of the two masses and inversely proportional to the square of the distance between them,

$$\vec{\mathbf{F}}_g = \frac{GMm}{r^2}$$

where $\vec{\mathbf{F}}_g$ is the force of gravity; $G = 6.674 \times 10^{-11} \, m^3/kg \, s^2$ is the Gravitational constant; M is the mass of larger object, m the mass of the smaller one, and r is the distance between them.

2.2. Kepler's Laws of Planetary Motion.

Theorem 2.8 (Kepler's 1st Law). The orbit of a planet is an ellipse with the Sun at one of the two foci.

Theorem 2.9 (Kepler's 2nd Law). A line segment joining a planet and the Sun sweeps out equal areas during equal intervals of time.

Theorem 2.10 (Kepler's 3rd Law). The square of the orbital period of a planet is proportional to the cube of the semi-major axis of its orbit.

Proof of all Three Laws. We begin the proof by considering circular motion. Let C be a circle of radius r, and a particle P lying on the boundary of C. Let T be the time required for P to complete a full rotation around the boundary of C starting from a fixed point. Then, the velocity v of P is $v = \frac{2\pi r}{T}$, and the acceleration $a = \frac{2\pi v}{T}$. Solve each of these for T, equate them, and then solve for a.

$$\frac{2\pi r}{v} = \frac{2\pi v}{a},$$

$$ar = v^{2},$$

$$a = \frac{v^{2}}{r}.$$

With a in terms of v, we can invoke Newton's Second Law, which states that the forces acting on P can be written as $\vec{\mathbf{F}}_p = ma$. With $a = v^2/r$, we have

$$\vec{\mathbf{F}}_p = m \frac{v^2}{r}.$$

If we think of v and a as vectors, then $\vec{\mathbf{v}}$ is pointed in the direction of motion, tangent to C, and $\vec{\mathbf{a}}$ is pointed towards the center of C, i.e., $\vec{\mathbf{v}} \perp \vec{\mathbf{a}}$. Therefore we can decompose the forces acting on P into orthogonal components.

Our next goal is to generalize this theory to ellipses, and again decompose the forces acting on a particle travelling along an ellipse into orthogonal components. The ideal coordinate system for this is polar coordinates on the complex plane \mathbb{C}^2 . Let Q be a particle traveling along the boundary of an ellipse that is centered about the origin. In polar coordinates over \mathbb{C}^2 , we can write $Q = re^{i\theta} = \cos(\theta) + i\sin(\theta)$. Thus, the position of Q is a function in two variables, call it $s(r,\theta)$. As with the case of circular motion, we need to find ways of expressing the velocity and accleration of Q. In this case, we do that by taking derivatives of s with respect to time t. We have

$$\frac{ds}{dt} = \frac{dr}{dt}e^{i\theta} + r\frac{d\theta}{dt}e^{i(\theta+\pi/2)};$$

$$\frac{d^2s}{dt^2} = \left(\frac{d^2r}{dt^2} - r\left(\frac{d\theta}{dt}\right)^2\right)e^{i\theta} + \left(2\frac{dr}{dt}\frac{d\theta}{dt} + r\frac{d^2\theta}{dt^2}\right)e^{i(\theta+\pi/2)};$$

3. The n-body Problem

4. Conclusion

APPENDIX A. REFERENCES