Uncertainity

There are several uncertainities in machine design.

The uncertainities arise from several factors including

- Variation in material composition and properties

- Variation in geometric features (e.g. diameter of a bar, thickness of a sheet, etc.)
across a part and across different parts

- Validity of mathematical models

- Intensity of stress concentrations

- Effect of corrosion, wear

Uncertainities need to be accommodated. The paimary methods are deterministic and stochastic method.

The deterministic method establishes a design factor based on absolute uncertainities of a loss-of-function parameter and a maximum-allowable-parameter

The parameter could be load, stress, deflection, etc. Design factor nd is defined as

> loss-of-function parameter maximum allowable parameter

> > Yield strength nd = maximum von-Miner stren

Problem 1

Consider that maximum load on a structure is known with an uncertainty of ±20 percent and load causing failure is known within ± 15 percent. It load causing failur is . 10 kN, determine the design factor and maximum load that can be applied. Derign factor 21-41

801 Worst care

Min. load for failure is 88.5 kN.

Load can be determined to 20%. uncertainty thur, we can apply up tox 1.2 = 8 5 => x=7-1 KN Problem 1

Given that diameter can be fabricated to Smax = 1.5×10²D

to with ±1.7. Of nominal dimension

Support can be centered within ±1.5.1. A

Ob nominal diameter

Weight in known to within 2.1. accuracy

Strength to ±3.5.7.

Peak stress would be at

A

| 5peak | = \frac{1.02}{0.98} \frac{\text{Fnominal}}{\text{Anominal}}

FI M= F8max

F 8max

F 8max

F 8max

+ (1.02 Framinal) × 0.015 Drawinal × 32

Prominal × 0.97

=) $|6peak| = \frac{F}{Anomiral} \times \frac{1.02}{0.98} + \frac{1.02 \times 0.015 \times 8}{0.97}$

= F/Anominal X 1.167

It a weight is dropped from height he maximum load Frax experienced would be

From = Weight (1+ $\sqrt{1+2h\kappa}$) weight

For h=0, which is the case here

F = 2 Mnominar × 9 =

Thus, for max strew not to exceed strength

 $\frac{m_{\text{nominal}}}{A_{\text{nominal}}} \times 2 \times 1.167 < 0.965 (65)_{\text{nominal}}$

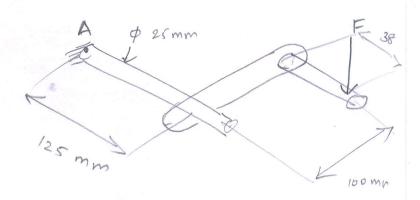
= 6 nominal

Derign factor nd = 2×1.167

≈ 2.42

Problem

Determine maximum normal and shear stress at A



301

F×163

At A

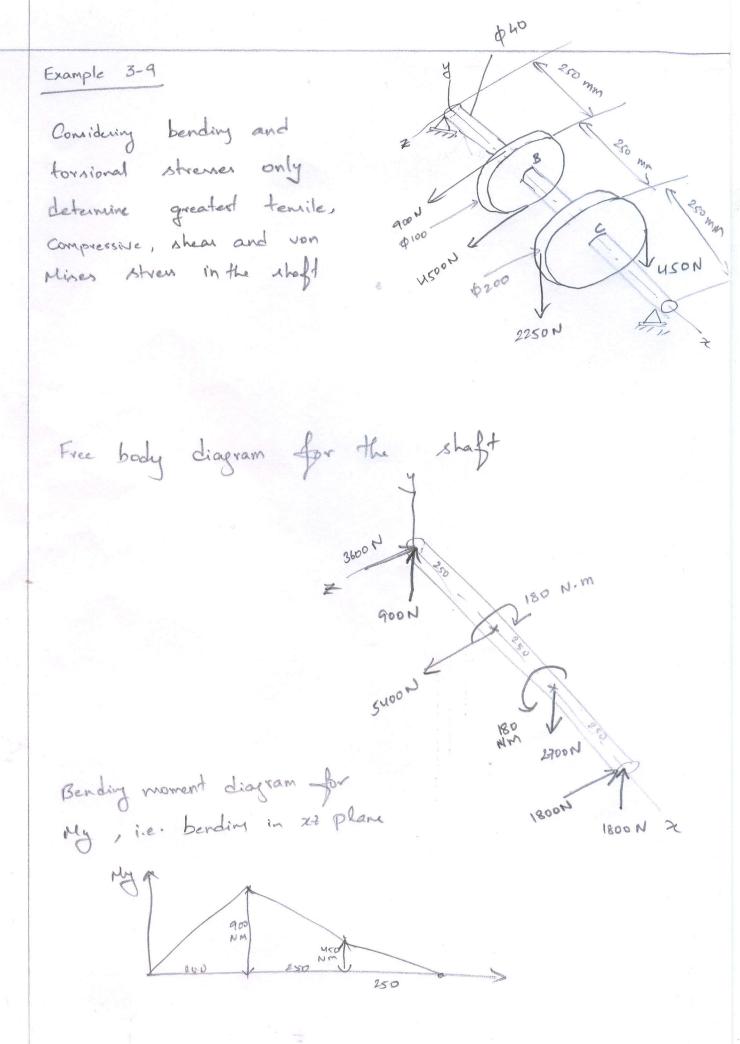
$$\frac{6}{20} = \frac{M_y}{I} = \frac{F \times 163 \times 10 \times 64}{R \times 204} = 269.8 \frac{N}{myd}$$

$$\frac{6}{200} = \frac{Tr}{J} = \frac{F \times 100 \times 10 \times 32}{P \times 20^4} = 82.8 \frac{N}{mpt}$$

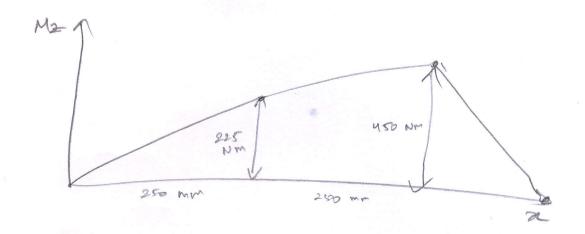
$$\frac{6}{22} = \frac{F}{A} = \frac{F4}{P \times 20^2} = 4.14 \frac{N}{mm^2}$$

$$= \sqrt{\frac{6\pi x}{2}}^2 + 6\pi y^2$$

$$=$$
 $\left(\frac{270}{2}\right)^{\frac{1}{4}} 82.8^{\frac{1}{2}}$



Bending moment diagram for M2, i.e. for bending in my plane



Max temile stren would occur at B

From torsion, max sheen stress occur at in between BC