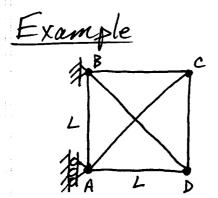


Structural Analysis

- 1) Equilibrium
- 2) Geometry (Kinematics)
- 3) Material behavior (constitutive law)

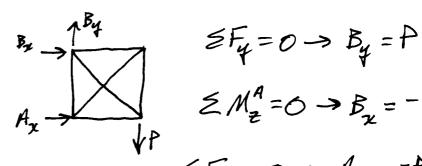


c 6 struts, no joint at center

Determine the forces/stresses in each strut.

E, A are identical for each strut.

Equilibrium:



$$\sum_{VP} \{ \mathcal{M}_{z}^{A} = 0 \rightarrow \mathcal{B}_{z} = -P \}$$

$$\xi F_{\chi} = 0 \rightarrow A_{\chi} = B_{\chi} = A$$

$$-B_{2}=P \xrightarrow{B} BC$$

$$AB$$

 $A_x = P \xrightarrow{AB} AC$

Z equilibrium equations for each joint -> 8 equations However, 3 are redundant b/c we used them for Az, Bz, By. -> 5 equations, 6 unknowns

AB AC AD BC BD CD 至AC+AD A) 2F2: ZFg: AB+ZAC BC + 星的 B) 5Fx: - 等BD ZFy: TAB - 区AC - 豆AC C) 2F2: -BC ZFy: -EBD D) 2F2: -AD 是BD+CD 2Fy:

Show 3 redundant equations.

Eall Fx equations > 0 = 0

Eall Fy equations > 0 = 0

at this point you can't choose to sum all equations because these previous two demonstrations already cover this

We certainly can't go any farther with equilibrium. What else can we relate to these internal forces?

Well we were told that we had a Young's modulus E and cross-sectional area A for each bar, so they must be elastic.

So for each bar we have $\sigma = EE$ where $\sigma = F/A \Rightarrow F = EAE$

$$\begin{vmatrix} AB \\ AC \\ AD \end{vmatrix} = EA \begin{vmatrix} \mathcal{E}_{AB} \\ \mathcal{E}_{AC} \\ \mathcal{E}_{AD} \\ \mathcal{E}_{BC} \\ \mathcal{E}_{BD} \\ \mathcal{E}_{CD} \end{vmatrix}$$

This has not helped us much as far as number of equations us number of unknowns is concerned because we have simply replaced the unknown forces with unknown strains.

Now what?

We still have to consider kinematics.

Consider an arbitrarily oriented stant

$$(\chi_{1}, \chi_{1})$$

$$(\chi_{2}, \chi_{2})$$

$$(\chi_{2}, \chi_{2})$$

$$(\chi_{1}, \chi_{1})$$

$$(\chi_{1}, \chi_{1})$$

What is
$$\varepsilon$$
? $\varepsilon = \frac{L-L_0}{L_0} = \frac{L}{L_0} - 1$

Some geometry:
$$\chi_2 = \chi_1 + L_0 \cos \theta$$

 $\psi_2 = \psi_1 + L_0 \sin \theta$

$$X_{z} = X_{1} + L \cos(\theta + \Delta \theta)$$

$$Y_{z} = Y_{1} + L \sin(\theta + \Delta \theta)$$

Displacements:
$$X_1 = \chi_1 + u_1$$

 $Y_1 = Y_1 + v_1$

$$\chi_2 = \chi_2 + u_2 = \chi_1 + L\cos(\theta + \Delta \theta)$$

$$\chi_1 + L\cos(\theta + u_2) = \chi_1 + u_1 + L\cos(\theta + \Delta \theta)$$

$$Y_{z} = Y_{z} + V_{z} = Y_{1} + L \sin(\theta + \Delta \theta)$$

$$Y_{1} + L_{0} \sin \theta + V_{2} = Y_{1} + U_{1} + L \sin(\theta + \Delta \theta)$$

 $X_2 \cos \theta$: $L_0 \cos^2 \theta + (u_2 - u_1) \cos \theta = L \cos \theta \cos (\theta + a \theta)$

 $\frac{1}{2}$ sind: $L_0 \sin^2\theta + (v_2 - v_1) \sin\theta = L \sin\theta \sin(\theta + a\theta)$

Add these together, divide by to and rearrange

 $\frac{U_z - U_1}{L_0} \cos \theta + \frac{V_z - V_1}{L_0} \sin \theta = \frac{L}{L_0} \left[\cos \theta \cos(\theta + s\theta) + \sin \theta \sin(\theta + s\theta) \right] - 1$

This almost looks like ε .

Note that AD depends on K_1, K_2, V_1, V_2 in a non-linear fashion.

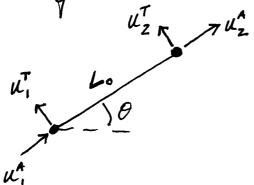
Many structures that are stiff and anchored to prevent rigid body motions have small AD.

 $(02(0+\Delta\theta) = \cos\theta\cos\Delta\theta - \sin\theta\sin\Delta\theta \approx \cos\theta - \Delta\theta\sin\theta$ $\sin(0+\Delta\theta) = \sin\theta\cos\Delta\theta + \cos\theta\sin\Delta\theta \approx \sin\theta + \Delta\theta\cos\theta$ To first order in $\Delta\theta$.

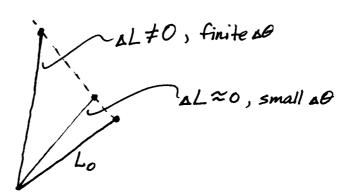
 $\Rightarrow \cos\theta \cos(\theta + a\theta) + \sin\theta \sin(\theta + a\theta)$ $\approx \cos^2\theta - a\theta \sin\theta \cos\theta + \sin^2\theta + a\theta \sin\theta \cos\theta$ $= 1 + O(a\theta^2)$

So to 1st order in all (small rotations) we have $E = \frac{u_z - u_1}{L_0} \cos \theta + \frac{v_z - v_1}{L_0} \sin \theta$

Another approach to <u>linearized</u> kinematics assuming small rotations from the outset. $u_{2}^{T} = \pi u_{2}^{2}$ $u_{1}^{T} = \pi u_{2}^{2}$ $u_{1}^{T} = transverse$ displacement $u_{1}^{T} = transverse$ displacement



Notice that ut and uz either translate the strut (for the UT=UT mode) or rotate the strut (for the ut=-ut mode). Hence the ut's do not cause a strain in the bar. Also note that this is only true for small differences in ut and ut



So, for small rotations of the strut, the change in length of the strut is $\mathcal{U}_{z}^{A} - \mathcal{U}_{i}^{A}$ and the strain is

$$\varepsilon = \frac{\kappa_z^A - \kappa_I^A}{L_o}$$

Next we need the us's in terms of

$$u^{T} \downarrow \downarrow \downarrow u^{A} \qquad u^{A} = u \cos \theta + v \sin \theta$$

$$u^{T} = -u \sin \theta + v \cos \theta$$

$$\mathcal{E} = \frac{\mathcal{U}_z - \mathcal{U}_1}{L_o} \cos \theta + \frac{\mathcal{V}_z - \mathcal{V}_1}{L_o} \sin \theta$$

So we have recovered our previous result. V OK, now let's use this in the analysis of our structure.

$$\mathcal{E}_{AB} = \frac{V_B - V_A}{L}$$

$$\mathcal{E}_{AC} = \frac{\frac{72}{2}(u_c - u_A)}{\sqrt{2}L} + \frac{\frac{72}{2}(v_c - v_A)}{\sqrt{2}L} = \frac{u_c - u_A}{2L} + \frac{V_c - v_A}{2L}$$

$$\mathcal{E}_{AD} = \frac{u_D - u_A}{L}$$

$$\mathcal{E}_{BC} = \frac{u_C - u_B}{L}$$

$$\mathcal{E}_{BC} = \frac{\frac{72}{2}(u_D - u_B)}{\sqrt{2}L} - \frac{\frac{\sqrt{2}}{2}(v_D - v_B)}{\sqrt{2}L} = \frac{u_D - u_B}{2L} - \frac{v_D - v_B}{2L}$$

$$\mathcal{E}_{AD} = \frac{v_C - v_D}{L}$$

$$\begin{vmatrix}
-A_{x} \\
0 \\
-B_{x} \\
-B_{y}
\end{vmatrix} = \begin{bmatrix}
M \end{bmatrix} EA \begin{bmatrix}
N \end{bmatrix} \begin{bmatrix}
W_{B} \\
V_{B}
\end{bmatrix}$$

$$\begin{vmatrix}
0 \\
0 \\
0
\end{vmatrix}$$

$$\begin{vmatrix}
8 \times 6 \\
0
\end{vmatrix}$$

$$\begin{vmatrix}
8 \times 8 \\
- \begin{bmatrix}
K \end{bmatrix}
\end{vmatrix}$$

$$\begin{vmatrix}
V_{C} \\
V_{D}
\end{vmatrix}$$

$$\begin{vmatrix}
V_{D} \\
V_{D}
\end{vmatrix}$$

If we had not previously analyzed the entire structure and solved for A_{χ} , B_{χ} and B_{ψ} , this matrix equation would represent 8 equations for 8 unknown displacements and the 3 unknown reaction forces.

However we also know that some of the displacements at the supports are fixed.

Specifically, up=0, uB=0 and UB=0.

We can now solve the renaining equations for V_A , V_C , V_C , U_D and V_D .

$$U_{A} = -\frac{5+3\sqrt{2}}{4(1+\sqrt{2})^{2}} PL/AE$$

$$U_{C} = \frac{5+3\sqrt{2}}{4(1+\sqrt{2})^{2}} PL/AE$$

$$U_{C} = -\frac{5+3\sqrt{2}}{2(1+\sqrt{2})} PL/AE$$

$$U_{D} = -\frac{7+5\sqrt{2}}{4(1+\sqrt{2})^{2}} PL/AE$$

$$U_{D} = -\frac{27+19\sqrt{2}}{4(1+\sqrt{2})^{2}} PL/AE$$

$$U_{D} = -\frac{27+19\sqrt{2}}{4(1+\sqrt{2})^{2}} PL/AE$$

$$Mm = \begin{pmatrix} 0 & \frac{\sqrt{2}}{2} & 1 & 0 & 0 & 0 \\ 1 & \frac{\sqrt{2}}{2} & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & \frac{\sqrt{2}}{2} & 0 \\ -1 & 0 & 0 & 0 & -\frac{\sqrt{2}}{2} & 0 \\ 0 & -\frac{\sqrt{2}}{2} & 0 & -1 & 0 & 0 \\ 0 & -\frac{\sqrt{2}}{2} & 0 & 0 & 0 & -1 \\ 0 & 0 & -1 & 0 & -\frac{\sqrt{2}}{2} & 0 \\ 0 & 0 & 0 & 0 & \frac{\sqrt{2}}{2} & 1 \end{pmatrix};$$

$$Nn = \begin{pmatrix} 0 & -1 & 0 & 1 & 0 & 0 & 0 & 0 \\ -\frac{1}{2} & -\frac{1}{2} & 0 & 0 & \frac{1}{2} & \frac{1}{2} & 0 & 0 & 0 \\ -1 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & -1 & 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & -\frac{1}{2} & \frac{1}{2} & 0 & 0 & \frac{1}{2} & -\frac{1}{2} \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 & -1 \end{pmatrix};$$

$$= \text{Notice I factored out } \frac{1}{L}.$$

K=-Mm.Nn; E Notice that I changed signs here. I also factored out EA.

$$\begin{pmatrix}
1 + \frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} & 0 & 0 & -\frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & -1 & 0 \\
\frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}} & 0 & -1 & -\frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & 0 & 0 \\
0 & 0 & 1 + \frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & -1 & 0 & -\frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} \\
0 & -1 & -\frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}} & 0 & 0 & \frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} \\
-\frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & -1 & 0 & 1 + \frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} & 0 & 0 \\
-\frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & 0 & 0 & \frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}} & 0 & -1 \\
-1 & 0 & -\frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} & 0 & 0 & 1 + \frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} \\
0 & 0 & \frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & 0 & -1 & -\frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}}
\end{pmatrix}$$

$$\text{Kreduced} = \begin{pmatrix} 1 + \frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} & 0 & 0 \\ -\frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} & 0 & 0 \\ -\frac{1}{2\sqrt{2}} & \frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}} & 0 & -1 \\ 0 & 0 & 0 & 1 + \frac{1}{2\sqrt{2}} & -\frac{1}{2\sqrt{2}} \\ 0 & 0 & -1 & -\frac{1}{2\sqrt{2}} & 1 + \frac{1}{2\sqrt{2}} \end{pmatrix};$$

Freduced =
$$\begin{pmatrix} 0 \\ 0 \\ 0 \\ 0 \\ -P \end{pmatrix};$$

Solve[Kreduced.Ureduced == Freduced, {vA, uC, vC, uD, vD}]

$$\begin{split} & \Big\{ \Big\{ \text{uD} \to -\frac{7 \text{ P} + 5 \sqrt{2} \text{ P}}{4 \left(1 + \sqrt{2} \right)^2}, \text{ vD} \to -\frac{27 \text{ P} + 19 \sqrt{2} \text{ P}}{4 \left(1 + \sqrt{2} \right)^2}, \\ & \text{vA} \to -\frac{5 \text{ P} + 3 \sqrt{2} \text{ P}}{4 \left(1 + \sqrt{2} \right)^2}, \text{ uC} \to -\frac{-5 \text{ P} - 3 \sqrt{2} \text{ P}}{4 \left(1 + \sqrt{2} \right)^2}, \text{ vC} \to -\frac{5 \text{ P} + 3 \sqrt{2} \text{ P}}{2 \left(1 + \sqrt{2} \right)} \Big\} \Big\} \end{split}$$

* Note that I factored out E, H and L.

I replace them in the notes.

P > PL/AE