



**HACETTEPE UNIVERSITY**  
**GEOMATICS ENGINEERING**

**GMT 312 – GLOBAL NAVIGATION AND  
SATELLITE SYSTEMS  
PROJECT**

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### Step – 1 Calculating my epoch and getting input values:

Epoch from student number:  $2+1+6+3+2+7+3+4=28$  and  $28*660=18480$  that means 5 hours and 8 minute.

In this epoch There is 9 GPS satellite in here. I got this data from .rnx file. The satellites are G02-G04-G05-G07-G08-G09-G27-G28-G30

```
> 2020 04 01 05 08 0.0000000 0 25
G02 25231555.823
G04 23922754.021 125714898.64207
G05 22451786.599 117984938.28807
G07 20889500.770 109775070.38608
G08 24359879.080 128012027.28106
G09 21351722.419 112204052.45508
G27 24932824.640 131022877.14016
G28 22653202.271 119043377.06307
G30 20951453.885 110100628.73108
R12 22935223.989 122515840.82307
E02 26364242.870 138545022.91847
```

In my homework I made an 3D Matrix and each layer represent a one satellite. I use my all broadcast values from here and I got those datas from 'brdc0920.20n'

`Matrix(:, :, 1) = [`                      `Matrix(:, :, 2) =`                      `Matrix(:, :, 3) =`

For the pseudoranges I make a 1D new matrix and each column represent a one satellite's C/A code. I got those values from 'MERS00TUR\_R\_20200920000\_01D\_30S\_MO.rnx'.

Workflow:

### Step – 2 Getting satellite's coordinates from broadcast ephemeris values:

I identify elements of data matrices and with for loop calculate the satellite coordinates for the each satellites with emission time epoch. After that I include earth rotation rate to this coordinates and find the final satellite coordinates. (unit of meter)

X	Y	Z
22107531.4391518	-14615147.179383	1654751.96473779
-110551.964201407	26376679.0130249	3000312.89845786
15267103.0778424	-7625701.14851773	20265684.522862
12734692.0096245	10479862.2489481	21198327.4298369
-6712721.21319645	24118326.446445	8497313.33160232
8175674.18802981	21856811.1242014	12618192.6759245
-12197733.1043852	16143116.2646536	16891811.8725527
22451388.0614499	12114094.5470372	-6530685.1079283
21479881.9881033	2559171.5571406	15579310.924969

### Step - 3 Calculating lat lon h values for the MERS station:

With the previous homework's formulas I calculated it. Results are below here. (unit of degree)

Latitude	Longitude
36.5663855333258	34.2558628374097

#### Step - 4 Calculating zenith and azimuth:

Zenith Values for each satellite (unit of degree):

Columns 1 through 5

83.8625846912416	72.2967274772153	56.5003182991529	20.8639519959001	77.8161734432321
------------------	------------------	------------------	------------------	------------------

Columns 6 through 9

39.4002015779332	83.314921723092	63.7394529723072	28.7048700118089
------------------	-----------------	------------------	------------------

Azimuth Values for each satellite (unit of degree):

Columns 1 through 5

-100.703415289875	106.221021077394	-52.7277682343601	11.6242666071019	85.0285973739621
-------------------	------------------	-------------------	------------------	------------------

Columns 6 through 9

95.0948345646251	54.9432440758119	-172.647257613971	-83.7179886032619
------------------	------------------	-------------------	-------------------

#### Step - 5 Calculating Ionosphere effect:

Firstly I calculate the tgps with cal2gpstime.m function and then with the unit of radian lat,lon,zenith,azimuth and alpha beta I calculated the Ionosphere effect with cal\_klob.m function.

Alpha and beta values gathered from brdc0920.20n file's header.

```
alpha=[0.1118D-07 0.7451D-08 -0.5960D-07 -0.5960D-07 ];
beta=[0.9011D+05 0.1638D+05 -0.1966D+06 -0.6554D+05];
```

Ionosphere effect values (unit of meter):

Columns 1 through 5

1.50672089307735	1.56541470165287	1.75893744263038	3.23029943453219	1.52989423988238
------------------	------------------	------------------	------------------	------------------

Columns 6 through 9

2.20198669044199	1.50817405740572	1.65123127547274	2.69811520605192
------------------	------------------	------------------	------------------

#### Step - 6 Adjusting MERS Station's Coordinates:

I made an iteration and it won't stop the condition until gonna be real.

The Condition:

```
if sqrt( (unknown(1,1)^2+unknown(2,1)^2+unknown(3,1)^2) ) < 0.001
```

Unknown is the x parameter from  $x=(AtA)^{-1}(AtL)$  least square formula. Its a 4x1 matrix that shows

[dx dy dz cdt]. In A matrix each row represent a satellite. L is  $R_j$  (calculated range) –  $ro_0$  –  $D_j$  (clock error effect + ionosphere effect + TGD effect. I multiply TGD with c because TGD's unit is second and I made it (meter/second). Now in  $D_j$  element all the elements in meter unit.

If the condition not true it will update the previous values with previous+unknown values.

The latest [dx dy dz cdt] values are:

```
-0.000282635201812315
-0.000429644389831278
-0.000268399551166982
-36427.7605023956
```

Those are coordinate changes which is means adjustment values. I added them to our initial MERS coordinates and found the final MERS coordinates.

```
Xfinal =  
  
4239146.64111737  
  
Yfinal =  
  
2886967.12407036  
  
Zfinal =  
  
3778874.4797316
```

Now I calculate the distance between my XYZ and XYZ in the pdf.

XYZ in the pdf:

X = 4239149.23905405m, Y = 2886968.03012480m, Z = 3778877.17820202m

The Result (unit of meter):

3.85382307355328

Also unknown's 4th column means cdt so I divide it to c and find the receiver clock error:

-0.000121509929720766

Without TDG and Ionosphere effect:

Unknown matrix [dx dy dz cdt]:

```
-0.000282635548942543  
-0.000429644195051262  
-0.00026839522112198  
-36425.5118718322
```

MERS Final XYZ:

```
Xfinal =  
  
4239146.64111736  
  
Yfinal =  
  
2886967.12407036  
  
Zfinal =  
  
3778874.4797316
```

Calculating distance between MERS Final xyz and referance xyz (unit of meter):

3.85382307092052

Receiver clock error:

-0.000121502429096573