

ME 202 - Strength of Materials

Vishal Neeli

1 Review

1.1 Strain Energy

$$u = \frac{1}{2} E \epsilon^2 = \frac{1}{2} \sigma \epsilon = \frac{1}{2} \frac{\sigma^2}{E}$$

1.2 Poisson's Ratio

$\epsilon_{axial} = \frac{\delta}{L}$, δ is the change in longitudinal length
 $\epsilon_{radial} = \frac{\delta'}{r}$, δ' is the change in radius

Poisson's ratio -

$$\nu = - \frac{\epsilon_{radial}}{\epsilon_{axial}}$$
$$0 \leq \nu \leq 0.5$$

Shear stress and strain

$$\tau = G\gamma$$

Relation between G, E and ν -

$$G = \frac{E}{2(1 + \nu)}$$

For a axial load,

$$d\delta = \frac{N(x)dx}{A(x)E(x)}$$

2 Torsion

2.1 Uniform Torsion

- **Angle of twist** : It is the angle by which one end of the rod is displaced wrt the other end of the rod under the effect of some torsion (twisting).
- Consider a small cylindrical section which has length dx , one end has been displaced wrt another by an angle of $d\phi$. Consider a line ab on the circumference along the length of the cylinder which has been changed to ab' . Then shear strain

$$\gamma_{max} = \frac{bb'}{ab} \quad (\text{bb' can be assumed to be a straight line})$$

$$\gamma_{max} = \frac{rd\phi}{dx}$$

- Rate of twist or angle of twist per unit length

$$\theta = \frac{d\phi}{dx}$$

$$\gamma_{max} = r\theta$$

If θ is constant, then

$$\gamma_{max} = \frac{r\phi}{L}$$

This is called as γ_{max} because we are measuring the shear strain at the outer end i.e, with maximum radius and hence, maximum shear strain.

$$\gamma = \rho\theta = \rho \frac{d\phi}{dx}$$

where ρ is the perpendicular distance of the point from the axis (radius) we are considering.

$$\gamma = \frac{\rho}{r} \gamma_{max}$$

2.2 Hooke's Law

Hooke's law for shear stress and shear strain

$$\tau = G\gamma = G\frac{\rho}{r}\gamma_{max} = \frac{\rho}{r}\tau_{max}$$

$$\tau_{max} = G\gamma_{max}$$

2.3 Torsion Formula

- Polar moment of inertia (this is integral over area - double integral)-

$$I_P = \int_A \rho^2 dA$$

where ρ is the distance at which area element dA is located.

- For a circle of radius r ,

$$I_P = \int_A \rho^2 dA = \int_{\theta=0}^{2\pi} \int_{\rho=0}^r \rho^2 (\rho d\theta d\rho) = \frac{\pi r^4}{2}$$

$$I_{Pcircle} = \frac{\pi r^4}{2}$$

- Consider a cross-section of any shape, we are trying to sum all the small torques and equate it to the torque applied on this

$$T = \int_A dM = \int_A \rho \tau dA$$

$$= \int_A \frac{\rho^2}{r} \tau_{max} dA$$

$$= \frac{\tau_{max}}{r} \int_A \rho^2 dA$$

$$T = \frac{\tau_{max}}{r} I_P$$

Or,

$$\tau_{max} = \frac{Tr}{I_P}$$

item The shear stress at distance ρ from the center of the bar with polar moment of inertia I_P is

$$\tau = \frac{T\rho}{I_P}$$

For a rod of circular cross section of radius r ,

$$\tau_{max} = \frac{2T}{\pi r^3} = \frac{16T}{\pi d^3}$$

- Rate of twist

$$\theta = \frac{T}{GI_P}$$

Hence, GI_P is also known as **Torsional Rigidity**

- For a bar in pure torsion ($\theta = \text{const}$), the total angle of twist

$$\phi = \frac{TL}{GI_P}$$

The quantity $\frac{GI_P}{L}$ is also known as **torsional stiffness** of the bar.

- For a thin circular tube, $I_P = 2\pi r^3 t$

2.4 Non-uniform Torsion

For a bar with non-uniform cross-section, tension :

$$\phi = \int_0^L \frac{T(x)dx}{GI_P(x)}$$

3 Hooke's Law

3.1 Stresses in a Plane and Mohr's Circle

Using physical equations or using cauchy's stress tensor with rotation matrix, we get

$$\sigma_{x'} = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta \quad (1)$$

$$\tau_{xy'} = -\frac{\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta \quad (2)$$

$$\sigma_{y'} = \frac{\sigma_x + \sigma_y}{2} - \frac{\sigma_x - \sigma_y}{2} \cos 2\theta - \tau_{xy} \sin 2\theta \quad (3)$$

The above equations can be viewed as parametric equations to a circle.

On squaring and adding, we get Mohr's circle

$$\left(\sigma_{x'} - \frac{\sigma_x + \sigma_y}{2} \right)^2 + \tau_{xy'}^2 = \left(\frac{\sigma_x - \sigma_y}{2} \right)^2 + \tau_{xy}^2$$

$$\text{Center} = \left(\frac{\sigma_x + \sigma_y}{2}, 0 \right) \text{ and Radius} = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2} \right)^2 + \tau_{xy}^2}$$

- This is plotted by taking $\sigma_{x'}$ on x axis and $\tau_{xy'}$ on **negative** y-axis and angle 2θ anticlockwise as positive.
- τ_{xy} is taken as positive if it tends to rotate in the anticlockwise direction, negative otherwise.
- σ_x is taken as positive if it is tensile and negative for compressive.
- A rotation of angle θ in the plane corresponds to a rotation of 2θ on the Mohr's circle.

3.2 Hooke's law for plane stresses

For a point (isotropic material) under planar stresses ($\sigma_{xz} = \sigma_{yz} = \sigma_{zz} = 0$), we have

$$\begin{aligned} \epsilon_x &= \frac{1}{E}(\sigma_x - \nu\sigma_y) \\ \epsilon_y &= \frac{1}{E}(\sigma_y - \nu\sigma_x) \\ \gamma_{xy} &= \frac{\tau_{xy}}{G} \\ \sigma_x &= \frac{E}{1 - \nu^2}(1 + \nu\epsilon_x) \\ \sigma_y &= \frac{E}{1 - \nu^2}(1 + \nu\epsilon_y) \\ \tau_{xy} &= G\gamma_{xy} \end{aligned}$$

3.3 Volume change and Strain-energy density

Let a cuboid of sides a,b and c be under stresses.

$$V_0 = abc$$

$$\begin{aligned} V &= (a + a\epsilon_x)(b + b\epsilon_y)(c + c\epsilon_z) \\ &= abc(1 + \epsilon_x)(1 + \epsilon_y)(1 + \epsilon_z) \\ &= V_0(1 + \epsilon_x + \epsilon_y + \epsilon_z) \end{aligned} \quad \text{ignoring the } \epsilon_x\epsilon_y \text{ terms}$$

Unit volume change e, also known as **dilatation**

$$e = \frac{\Delta V}{V_0} = (1 + \epsilon_x)(1 + \epsilon_y)(1 + \epsilon_z)$$

Strain-energy density in plane stress,

$$\begin{aligned} u &= \frac{1}{2}(\sigma_x\epsilon_x + \sigma_y\epsilon_y + \tau_{xy}\gamma_{xy}) \\ u &= \frac{1}{2E}(\sigma_x^2 + \sigma_y^2 - 2\nu\sigma_x\sigma_y + \frac{\tau_{xy}^2}{2G}) \\ u &= \frac{E}{2(1 - \nu^2)}(\epsilon_x^2 + \epsilon_y^2 + 2\nu\epsilon_x\epsilon_y) + \frac{G\gamma_{xy}^2}{2} \end{aligned}$$

3.4 Hooke's law for triaxial stresses

$$(\sigma_z = \tau_{xz} = \tau_{yz} = 0)$$

$$\epsilon_x = \frac{1}{E}(\sigma_x - \nu\sigma_y - \nu\sigma_z)$$

$$\sigma_x = \frac{E}{(1+\nu)(1-2\nu)}[(1-\nu)\epsilon_x + \nu(\epsilon_y + \epsilon_z)]$$

3.5 Strain Energy in Torsion

$$d\phi = \frac{Tdx}{GJ}$$

$$\text{Energy stored due to torsion, } U = \int_0^L \frac{1}{2} T d\phi$$

$$= \int_0^L \frac{T^2 dx}{2GJ}$$

4 Torsion formula

4.1 Torsion Formula for non-prismatic bars

- For elliptical cross section, maximum shear stress

$$\tau_{max} = \frac{2T}{\pi ab^2}$$

$$\phi = \frac{TL}{GJ_e}$$

$$J_e = \frac{\pi a^3 b^3}{a^2 + b^2}$$

- For triangular cross section

$$\tau_{max} = \frac{T \frac{h}{2}}{J_t}$$

$$\phi = \frac{TL}{GJ_t}$$

$$J_e = \frac{h_t}{15\sqrt{3}}$$

- For rectangular cross section

$$\tau_{max} = \frac{T}{k_1 b t^2}$$

$$\phi = \frac{TL}{(k_2 b t^3)G} = \frac{TL}{J_r G}$$

$$J_r = k_2 b t^3$$

k_1 and k_2 are empirically determined and are dependent on $\frac{b}{t}$

- Thin walled open cross sections We treat flange and web as separate rectangles.

$$J = J_w + 2J_f$$

$$J_f = k_1 b_f t_f^3$$

$$J_w = k_1 (b_w - 2t_f) t_w^3$$

$$\tau_{max} = \frac{2T(\frac{t}{2})}{J}$$

$$\phi = \frac{TL}{GJ}$$

4.2 Thin walled tubes

Consider a small element of length of dx , then shear stresses are τ_a, τ_b, τ_c and τ_d . Then the shear stresses on opposite wall should be equal (to satisfy Newton's second law). Hence,

$$\begin{aligned} F_b &= F_c \\ \tau_b t_b dx &= \tau_c t_c dx \\ \tau_b t_b &= \tau_c t_c \end{aligned}$$

Shear flow f ,

$$\boxed{f = \tau t = \text{const}}$$

Deriving torsion formula for the thin walled tubes-

$$\begin{aligned} T &= \int_0^{L_m} \tau dA \rho \\ &= \int_0^{L_m} f r ds \\ &= 2f A_m \end{aligned}$$

where A_m is the area enclosed by the median line.

$$\boxed{\tau = \frac{T}{2tA_m}}$$

Using shear stress, we get J -

Shear energy density of a solid under pure shear stress is $\frac{\tau^2}{2G}$.

$$\begin{aligned} \text{Total shear energy } U &= \int \int \frac{\tau^2}{2G} t dx ds \\ &= \int \int \frac{f^2}{2Gt} dx ds \\ &= \frac{f^2}{2G} \int_0^L dx \int_0^{L_m} ds \\ &= \frac{T^2 L}{2GA_m^2} \int_0^{L_m} ds \end{aligned}$$

Also, $U = \frac{T^2 L}{2GJ}$. Hence,

$$\boxed{J = \frac{4A_m^2}{\int_0^{L_m} \frac{ds}{t}}}$$

For tube with **constant thickness**,

$$\boxed{J = \frac{4A_m^2 t}{L_m}}$$

For circular tube,

$$J = 2\pi r^3 t$$

For rectangular tube,

$$J = \frac{2b^2 h^2 t_1 t_2}{bt_1 + ht_2}$$

Angle of twist,

$$\begin{aligned} \phi &= \frac{TL}{GJ} \\ &= \frac{TL}{4GA_m^2} \int_0^{L_m} \frac{ds}{t} \end{aligned}$$

$$\boxed{\phi = \frac{TL}{4GA_m^2} \int_0^{L_m} \frac{ds}{t}}$$