

Numerics & Experiments on Low-Speed High-Altitude Wings

Carlo Brunelli, Ir



Royal Military Academy, department of Mechanicarl Engineering, Brussels, Belgium







FRAMEWORK

High-Altitude Pseudo-Satellites

- Semi-persistent coverage of areas
- Chaper than satellite systems
- High response capability

Requirements:

- High altidute cruise: 16-20km
- Low Speed: Mach = 0.1
- Low Reynolds: Re = 250000



HAPS vehicle

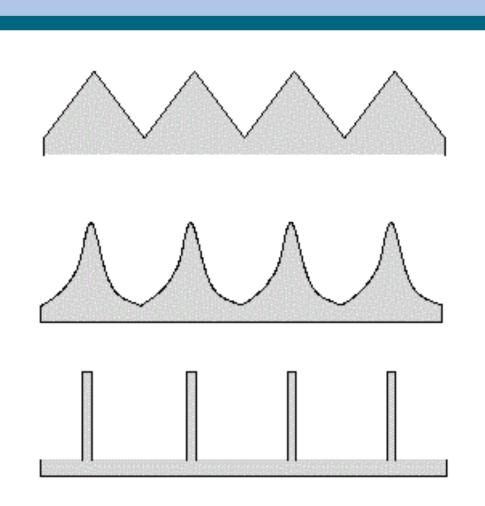
CHALLENGES

Develop numeric method for:

- Wing design
- Aerodynamic performance
- Passive control flow devices
- Laminar Separation Bubble

Use experimental data:

- Validation
- Tuning

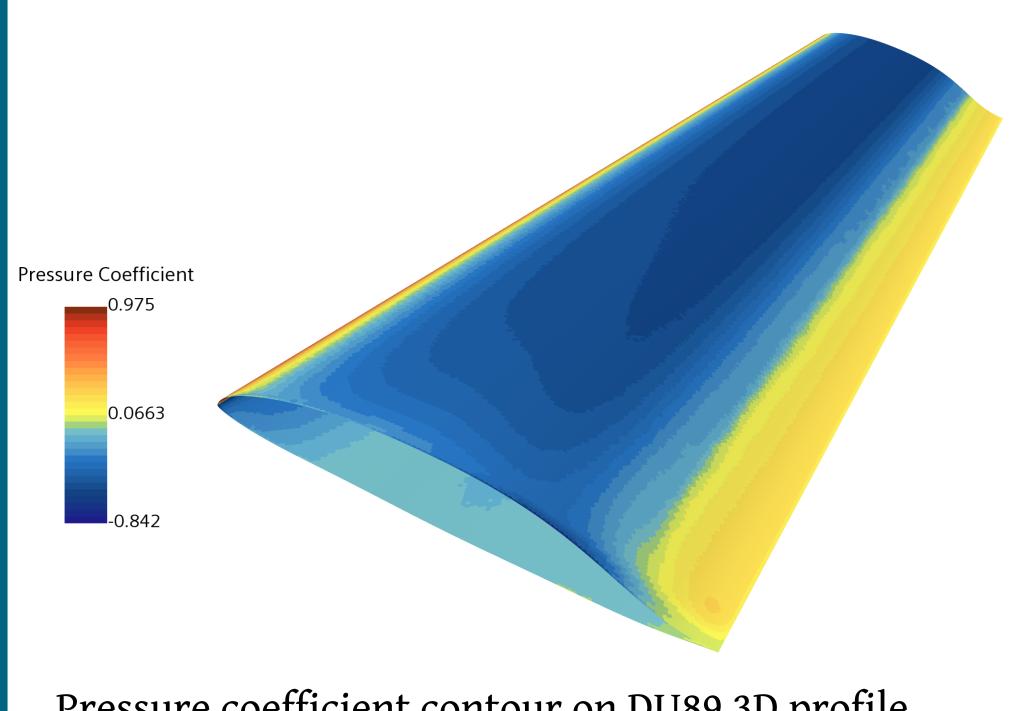


Common types of riblets, passive control flow devices

INITIAL ANALYSIS

CFD simulations in STAR-CCM+

- 2D CFD profile analysis
- 3D CFD profile analysis
- Estimation of physical quantities for setting experimental measurements
- 3D CFD analysis of the wind tunnel

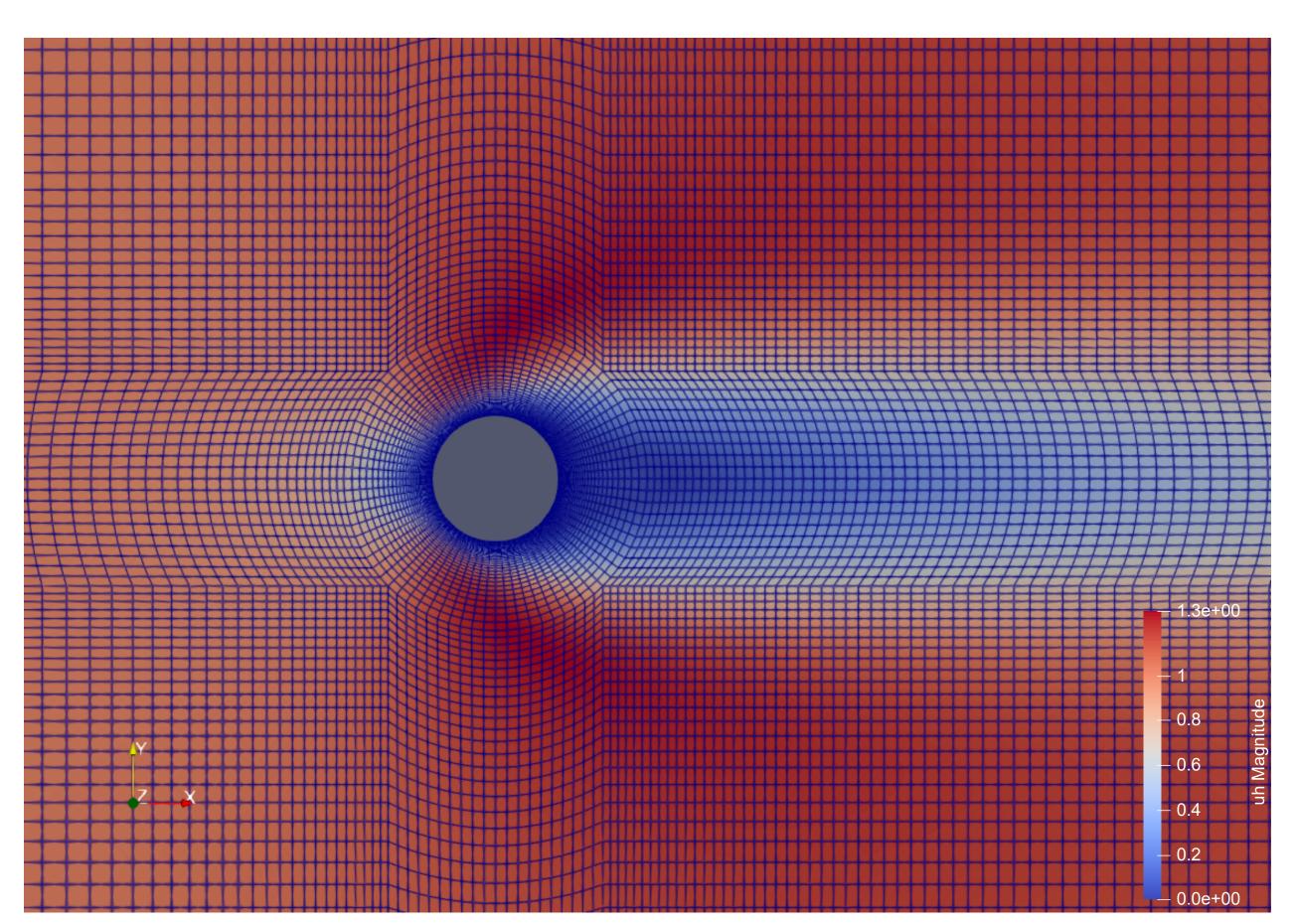


Pressure coefficient contour on DU89 3D profile

VARIATIONAL MULTISCALE METHOD

Implementation of the Variational MultiScale (VMS) framework for solving incompressible Navier Stokes

- Based on stabilized Finite Element Method in the Julia programming language
- Based on FEM → not sensitive to mesh quality
- Variational formulation
- Consistent treatment of all the scales of turbulence
- Equations for small scales have asymptotic behavior near walls
- Unresolved small scales influence only resolved small scales

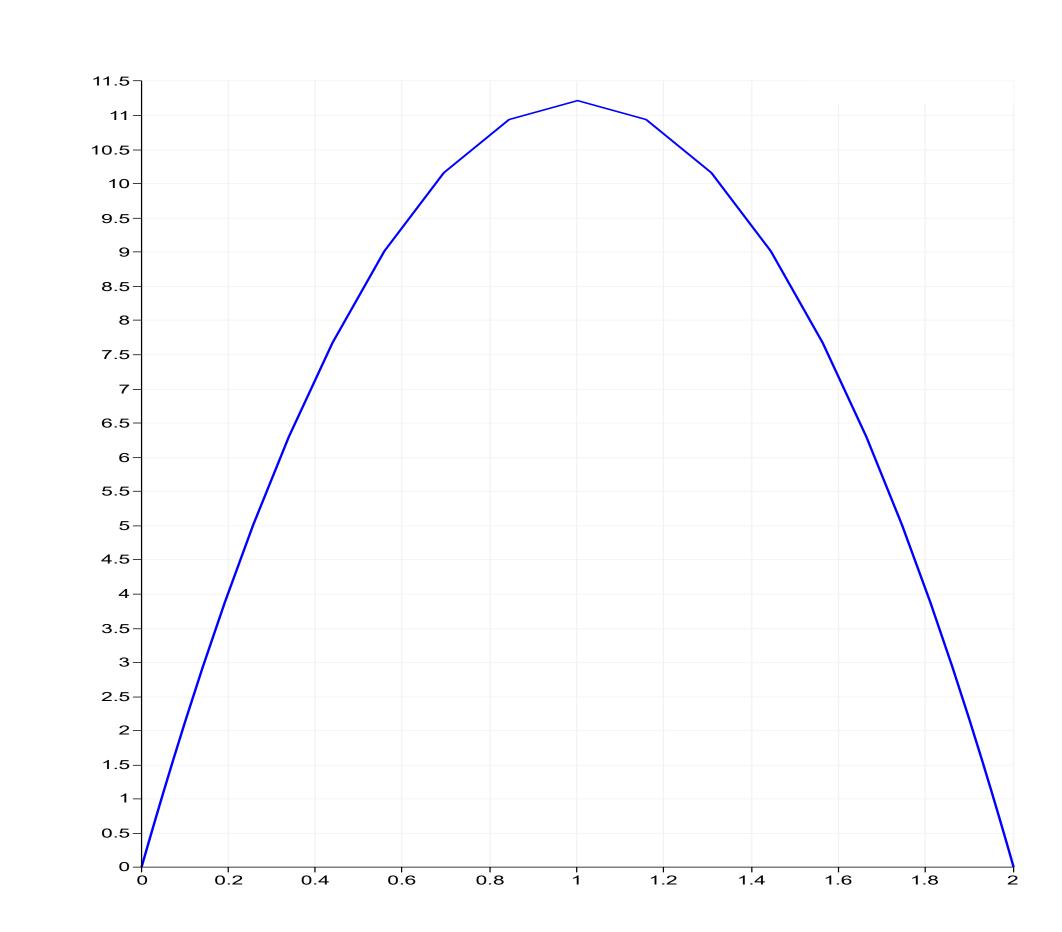


Velocity field of a cylinder and mesh, Re=100

IMPLEMENTATION IN JULIA USING GRIDAP

Using the Julia package Gridap for developing VMS:

- Very expressive API
- Extensible and modular
- High Performance
- Use PETSc library solvers
- Use GMSH for mesh generation



Velocity profile in a laminar channel with periodic bounday conditions

Industrial Partner

