September 6, 2024

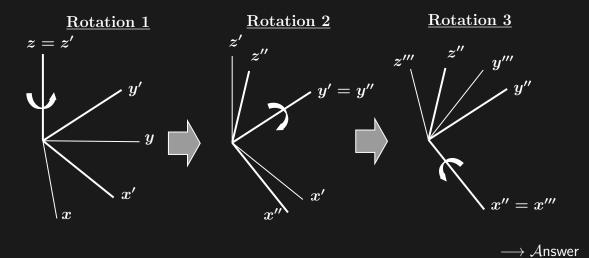
MODULE 2 — Assignment

Problem 1:

Type I Euler angles are also known as the aircraft Euler angles. These are comprised of (1) a rotation ψ about the fixed Z-axis, resulting in a primed axis system; (2) a rotation θ about the y'-axis resulting in a double-primed system; and (3) a rotation ϕ about the x"-axis, resulting in the final xyz body-fixed frame.

- a Neatly sketch this sequence of rotations.
- b Derive the rotation matrix that maps the XYZ frame to the body-fixed xyz frame.

The following sketch shows the rotations:



This rotation matrix is defined by the following three sequential rotations:

$$R_{1} = \begin{bmatrix} \cos(\psi) & \sin(\psi) & 0 \\ -\sin(\psi) & \cos(\psi) & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$R_{2} = \begin{bmatrix} \cos(\theta) & 0 & -\sin(\theta) \\ 0 & 1 & 0 \\ \sin(\theta) & 0 & \cos(\theta) \end{bmatrix}$$

$$R_{3} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos(\phi) & \sin(\phi) \\ 0 & -\sin(\phi) & \cos(\phi) \end{bmatrix}$$

The overall rotation matrix R that maps the XYZ frame to the body-fixed xyz frame is:

$$R = R_1(\psi)R_2(\theta)R_3(\phi)$$

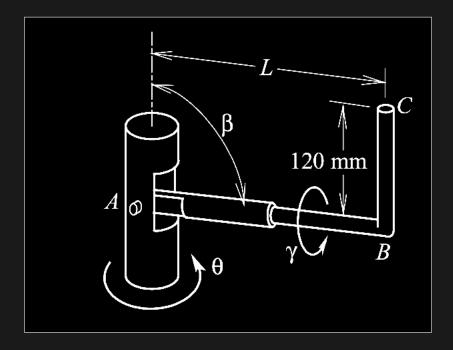
Which becomes:

$$R = \begin{bmatrix} \cos\left(\psi\right)\cos\left(\theta\right) & \cos\left(\phi\right)\sin\left(\psi\right) + \cos\left(\psi\right)\sin\left(\phi\right)\sin\left(\theta\right) & \sin\left(\phi\right)\sin\left(\psi\right) - \cos\left(\phi\right)\cos\left(\psi\right)\sin\left(\theta\right) \\ -\cos\left(\theta\right)\sin\left(\psi\right) & \cos\left(\phi\right)\cos\left(\psi\right) - \sin\left(\phi\right)\sin\left(\psi\right)\sin\left(\theta\right) & \cos\left(\psi\right)\sin\left(\phi\right) + \cos\left(\phi\right)\sin\left(\psi\right)\sin\left(\theta\right) \\ \sin\left(\theta\right) & -\cos\left(\theta\right)\sin\left(\phi\right) & \cos\left(\phi\right)\cos\left(\theta\right) & \cos\left(\theta\right) \end{bmatrix}$$

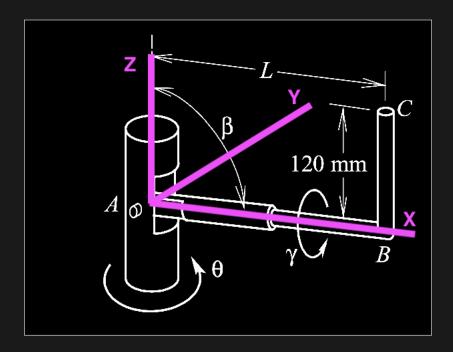
 $\longrightarrow \mathcal{A}$ nswer

Problem 2: EXERCISE 3.18

A hydraulic cylinder allows the length of arm AB to vary, and servomotors control the rotation angles θ about the vertical, β about pin A, and γ about axis AB, with $\gamma=0$ corresponding to bar BC being situated in the vertical plane as shown. In the initial position L=250 mm, $\theta=0$, $\beta=90^{\circ}$, and $\gamma=0$. In the final position, $\theta=\beta=120^{\circ}$, $\gamma=90^{\circ}$, and L=500 mm. Determine the corresponding displacement of end C.



We will use the coordinate frame as defined in the following figure:



Where

$$\begin{bmatrix} X' \\ Y' \\ Z' \end{bmatrix} = R_A \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}$$

The rotation about these axes is

$$R_{A} = \begin{bmatrix} \cos(\beta - 90) & 0 & -\sin(\beta - 90) \\ 0 & 1 & 0 \\ \sin(\beta - 90) & 0 & \cos(\beta - 90) \end{bmatrix} \begin{bmatrix} \cos(\theta) & \sin(\theta) & 0 \\ -\sin(\theta) & \cos(\theta) & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

This gives

$$R_{A_i} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$R_{A_f} = \begin{bmatrix} -\frac{\sqrt{3}}{4} & \frac{3}{4} & -\frac{1}{2} \\ -\frac{\sqrt{3}}{2} & -\frac{1}{2} & 0 \\ -\frac{1}{4} & \frac{\sqrt{3}}{4} & \frac{\sqrt{3}}{2} \end{bmatrix}$$

The displacement of point A to C is defined as

$$\Delta \bar{r}_C = \bar{r}_{C/A_f} - \bar{r}_{C/A_i}$$

Where

$$\bar{r}_{C/A_i} = \begin{bmatrix} L_i \\ -120\sin(\gamma_i) \\ 120\cos(\gamma_i) \end{bmatrix} = \begin{bmatrix} 250 \\ 0 \\ 120 \end{bmatrix} \text{ mm}$$

$$\bar{r}_{C/A_f} = \begin{bmatrix} L_f \\ -120\sin(\gamma_f) \\ 120\cos(\gamma_f) \end{bmatrix} = \begin{bmatrix} 500 \\ 120 \\ 0 \end{bmatrix} \text{ mm}$$

Therefore

$$\Delta \bar{r}_C = \bar{r}_{C/A_f} - \bar{r}_{C/A_i} = \begin{bmatrix} 250 \\ 120 \\ -120 \end{bmatrix}$$
mm

We can compute the displacement of point C in terms of the original axes we defined as

$$\Delta \bar{r}'_C = R_{A_f}{}^T \Delta \bar{r}_C + \left(R_{A_f}{}^T - R_{A_i}{}^T \right) \bar{r}_{C/A_i} = \begin{bmatrix} -570.43 \\ 315.00 \\ -370.00 \end{bmatrix} \text{ mm}$$

So the displacement is:

$$\Deltaar{r}_C' = \left[egin{array}{c} -570.43 \ 315.00 \ -370.00 \end{array}
ight] \; ext{mm}, \;\;\; \left|\Deltaar{r}_C'
ight| = 749.34 \; ext{mm}$$

 $\longrightarrow \mathcal{A}$ nswer

Submitted by Austin Barrilleaux on September 6, 2024.