

**Experiments on Time Reversal Focusing of Acoustic Energy at a Crack  
Location in One Dimension**

by

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## Abstract

Place your abstract here

# Nomenclature

|   |   |
|---|---|
| $\vec{r}_0$                               | Location of source an emitting acoustic pulse   |
| $\hat{R}$                                 | Reflection coefficient at $z = (L + l)$   |
| $\hat{T}$                                 | Transmission coefficient at $z = (L + l)$   |
| $R$                                       | Reflection coefficient at $z = l$   |
| $T$                                       | Transmission coefficient at $z = l$   |
| $\epsilon_{33}^T$                         | PZT permittivity constant for the z direction   |
| $\frac{\partial w_c}{\partial z}(l_c, t)$ | Focused wave response at the crack location due to time reversal focusing                         |
| $\frac{\partial w_{cr}}{\partial z}$      | Strain caused by wave portion transmitting to transducer B  |
| $\frac{\partial w_{ct}}{\partial z}$      | Strain caused by wave portion reflecting to transducer A  |
| $\frac{\partial w}{\partial z}$           | The strain resulting from the stress wave that is played by transducer A during the initial phase |
| $\nabla^2$                                | Laplacian operator  |

|                |  |
|----------------|--|
| $\nu$          | Spatial dissipation factor   |
| $\overline{R}$ | Reflection coefficient at $z = l_c$  |
| $\overline{T}$ | Transmission coefficient at $z = l_c$                                      |
| $\rho^c$       | Density of the crack   |
| $\rho$         | Density of the transducer material   |
| $\rho^*$       | Density of the rod material  |
| $A$            | Cross-sectional area of the rod  |
| $a$            | Transducer cross sectional area  |
| $c^*$          | Velocity of wave propagation through the rod                               |
| $C_{33}^*$     | Elastic constant in the z direction  |
| $C_{33}^c$     | Elastic coefficient of the crack in the z direction                        |
| $C_c$          | Equivalent capacitance for the transducers                                 |
| $c_p$          | The speed of sound through the plate material.                             |
| $C_{33}$       | PZT elastic constant in the z direction                                    |
| $D_3$          | Electric displacement in the z direction                                   |
| $d_{33}$       | Strain constant relating the strain and field intensity in the z direction |

$dz$  Differential element of length for the rod

$E_3$  Electric field intensity in the z direction

$h_i(\vec{r}_0)$  Integral of the transducer surface  $S_i$

$h_i^{ac}(t)$  Function which relates pressure to voltage for the transducers

$l$  Transducer length

$l_T$  Distance from the crack to transducer B

$P$  Axial pressure applied to rod end

$S_{33}$  Strain in the z direction

$T_3$  Stress in the z direction

$V_a^+(i\omega)$  Reversed signal that is played back by Transducer A

$V_a^+(i\omega)$  Voltage produced by the wave at transducer A

$V_b^+(i\omega)$  Reversed signal that is played back by Transducer B

$V_b^+(i\omega)$  Voltage produced by the wave at transducer B

$vec_r$  Position of the acoustic pressure field

$w$  Displacement of the rod

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# Chapter 1

## Introduction

Each day, more than 100 billion meteoroids larger than a micro-gram come speeding through Earth's atmosphere [1]. A large number of debris objects are presently in orbit around the earth (see Figure 1.1). NASA states that there are over 21,000 debris objects larger than a softball, 500,000 particles larger than a marble, and over 100 million smaller pieces that cannot be tracked. Due to the large number of objects traversing the space surrounding earth, it is likely that impact will occur with lightweight, low orbiting space structures and satellites which can result in varying degrees of damage [2]. The most likely impact will come from the smallest pieces as they occur in the largest numbers and are hard to avoid since they are hard to detect. These smaller particles will cause damage in the form of surface cracks and abrasions. Technologies exist such as the Whipple Shield (or meteor bumper) which help to mitigate the effects of impacts by absorbing some of the energy or breaking

larger objects into smaller pieces [3]. However, it is still likely that particles can make it all the way to the underlying structure and cause damage which will build up over time. Access to these structures is difficult, costly, and dangerous. Repair missions will inevitably leave behind more debris which will increase the likelihood of future damage occurring to the structure being repaired as well as other structures in orbit. Materials with the ability to automatically heal damage as it occurs are very desirable for these applications [4].

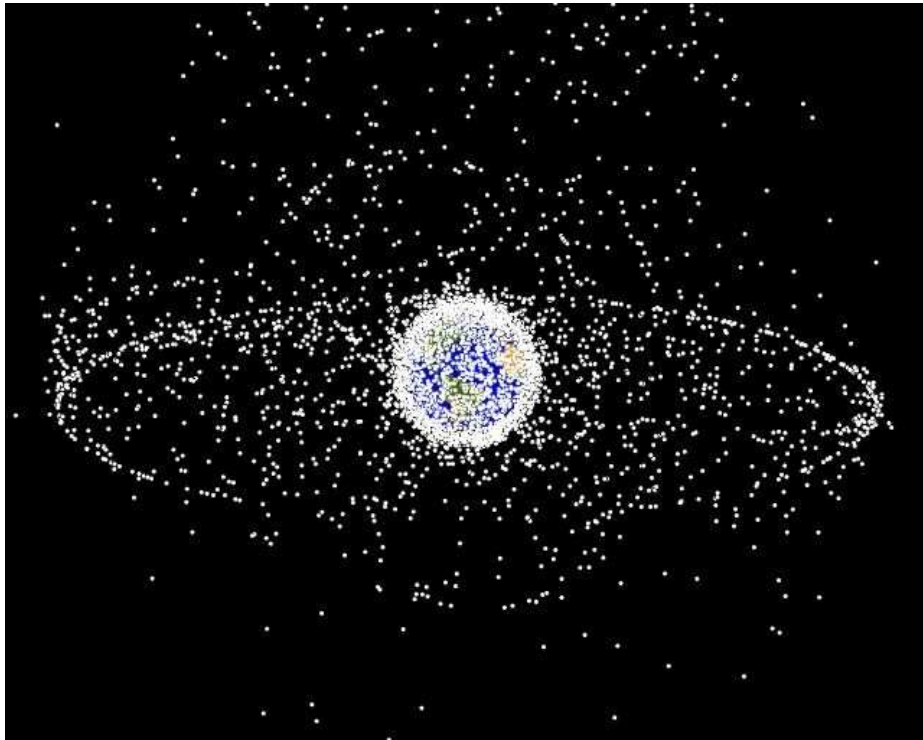


Figure 1.1: Image from NASA which depicts the debris objects that are currently orbiting the Earth. Some examples of debris objects include: broken spacecraft, upper stages of launch vehicles, debris that is intentionally released from missions, debris from collisions, and paint flecks. Much of the debris seems benign due to its size but is made dangerous by its high velocity (up to  $60\text{km/s}$  for smaller pieces).  
imagesource:[<http://orbitaldebris.jsc.nasa.gov/photogallery/beehives/GEO640.jpg>]

## 1.1 Self-Healing

Crack healing in materials owes some of its early studies to Wool and O'Connor who modeled crack healing in polymers and experimented with cantilever beams [5,6]. They found that polymer crack healing occurs in the following five stages: a) surface rearrangement, b) surface approach, c) wetting, d) diffusion, and e) randomization. Sloof, Song, and others have developed materials in which thermal activation caused a mending of cracks [7–11]. Caruso et. al. reported on material strength recovery in thermoplastics by using solvent-based healing agents [12]. Inspired by the idea of biological entities being able to automatically heal wounds, a self-healing material was made by White et. al. in which a catalyst and microcapsules filled with a reactive fluid were distributed throughout a material [13]. When the material was cracked, the capsules released their fluid which polymerized upon contacting the catalyst and effectively fused the faces of the crack together (Figure 1.2). Sottos et. al. have improved this self-healing implementation and found that damaged materials recovered up to 90% of their original strength [14]. The White and Sottos group have also studied crack healing by introducing hardener filled microcapsules into an epoxy that was molded into a double cantilever beam fracture specimen [15]. This concept has been extended to a self-healing coating that contained epoxy filled microcapsules and was tested on cold rolled steel sheets with good results [16]. Manuel et. al. created a matrix containing wires that could apply a force to the material to close a crack and

then the material was heated to weld the crack together [17]. Adhesives with self-healing properties have been created by Jin et. al. and tested on composite laminates where they were found to increase the life of specimens subject to fatigue [18]. Self repairing chemicals have been successfully implemented on composite airplane components and were able to return to 88% of their original strength after impact and shear testing [19]. Recently, there has been a large interest in applying self-healing methods to help prolong the life of concrete which is very prone to cracks [20].

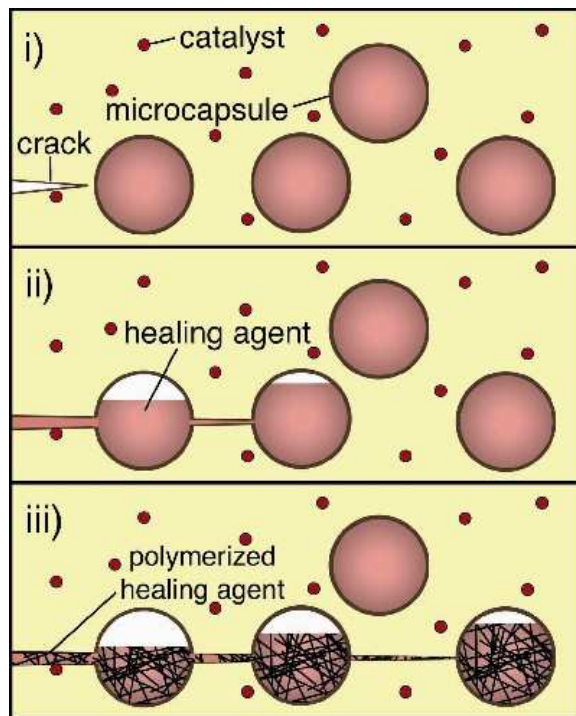


Figure 1.2: Concept drawing of a self-healing material in which micro-capsules filled with a reactive fluid and a catalyst are spread throughout that material. The drawing depicts the stages of damage, fluid release, and polymerization.  
source:[<http://www.howstuffworks.com/self-healing-spacecraft1.htm>]

## 1.2 Accelerating the Self-Healing Process

Typically, the damaging process will continue to occur as the material attempts to mend itself together. It is therefore important that the material heals itself quickly so that the damage process does not dominate and prevent the material from reaching a full mechanical recovery. Studies have been performed on fatigue crack propagation in a self healing material [21]. Sheng, Burattini, and many others reported on increasing the healing rate by improving the materials that are used [22–26]. The use of U.V. light has been researched as a way to accelerate the healing process in metalopolymers and ethyl cellulose based copolymers [27,28]. Wool and O’Connor found in their work that an increase in pressure at the crack interface during the early stages of healing or an increase in temperature at any stage can increase the rate at which the material recovers [5]. Murphy et. al. have studied direct heating as a method to increase healing capabilities [29]. Direct heating to improve healing has also been applied to concrete [30,31]. Applying pressure to a crack location via acoustic energy was studied by Korde et al. both theoretically and experimentally which found that acoustic energy can accelerate the healing process [32–34]. Fettig et. al. have also found that healing is optimized by ensuring good mixing in the early stages which can be brought about by localized pressure [35]. Sarrazin treated cracked nylon dog-bone specimens with ultrasonic probes which increased the temperature within the specimen and caused a fusing of the crack faces [36]. By using ultrasonic waves both

the temperature and pressure at a recovery site could be increased which results in a faster healing time.

## 1.3 Implementation

Two main problems arise when implementing any of the previously described external stimulation methods to accelerate the healing process; i) detecting when damage has actually occurred and ii) applying the stimulation energy only to the damage location. In regards to i), it would be highly inefficient and potentially damaging to continually introduce energy if healing is not taking place. Along similar lines of reasoning, one would want only to apply energy to the actual recovery location for increased efficiency and so that unintended damage to the system does not occur. Acoustic energy is the method chosen here as it offers solutions to both problems.

### 1.3.1 Damage Sensing

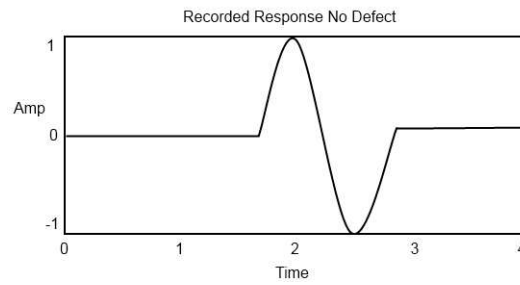
Picture a rod with transducers on both ends that are capable of playing and recording sound. First, one transducer propagates a low energy acoustic stress-waves through the material and the other records the response of the material. If this process is repeated periodically and the response is monitored then a change in the response is observed if there is a change in the medium (e.g., damage has occurred; Figure 1.3). The acoustic energy could then be applied after this change in response has been detected and cease once the system determines that the response has returned to its

initial state within some tolerance level. Imaging using ultrasonics has been around for some time now and is probably most recognized in ultrasounds that are used during pregnancy [37]. Ultrasonic imaging has been widely used in nondestructive testing to find cracks within structures. Harris et. al. used acoustic emission to monitor the fatigue crack growth in both aluminum and steel samples [38]. Detecting cracks in concrete by using ultrasound was reported by Zinin et. al. [39]. Acoustic waves have also been used to detect cracks in very fragile materials such as eggshells [40]. Recently, ultrasound has been used in dentistry to detect microcracks within teeth [41].

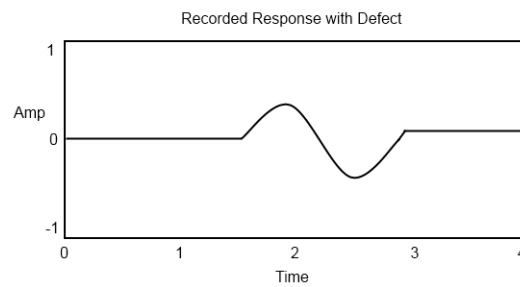
### **1.3.2 Time Reversal Focusing**

Besides the mechanics of focusing the energy at the crack location, the problem arises that the location of the defect is not necessarily known. A method is needed that not only localizes the energy at the healing site but it does so without knowledge of the physical location of that site. Acoustic time reversal signal processing is one such method that possesses the aforementioned qualities. Picture again the cracked medium with transducers on both ends. As before, a stress-wave is played by one transducer. This time, however, both transducers record signals. The stress-wave propagates through the medium and strikes the crack which causes the wave to split into multiple components that transmit to the opposite transducer and reflect back to the original sending transducer where they are recorded. If the transducers re-





(a) Acoustic wave propagating through medium with no defect



(b) Acoustic wave propagating through medium with a crack

Figure 1.3: a) A transducer sends an acoustic wave through a medium and a transducer on the opposite end records the response. This material is seen without a defect and the simulated response is shown; b) Same as a), but a defect is now present in the medium which causes a change in the response recorded on the opposite end and is seen in the simulated recorded response.

amplify and playback these signals in a time reversed fashion then the waves meet at the point where they split (here, the crack) which causes a focusing of their energy at that point. The focused wave splits again into multiple components that are recorded by each transducer and the time reversal process is repeated iteratively with a better focusing being achieved on each iteration until limits due to dissipation are reached (Figure 1.4 illustrates this concept). The time reversal concept can be extended to

multiple dimensions in which the transducers are placed around the boundary. In a self-healing material, the focusing of energy at a crack location would accelerate the healing process at that site.

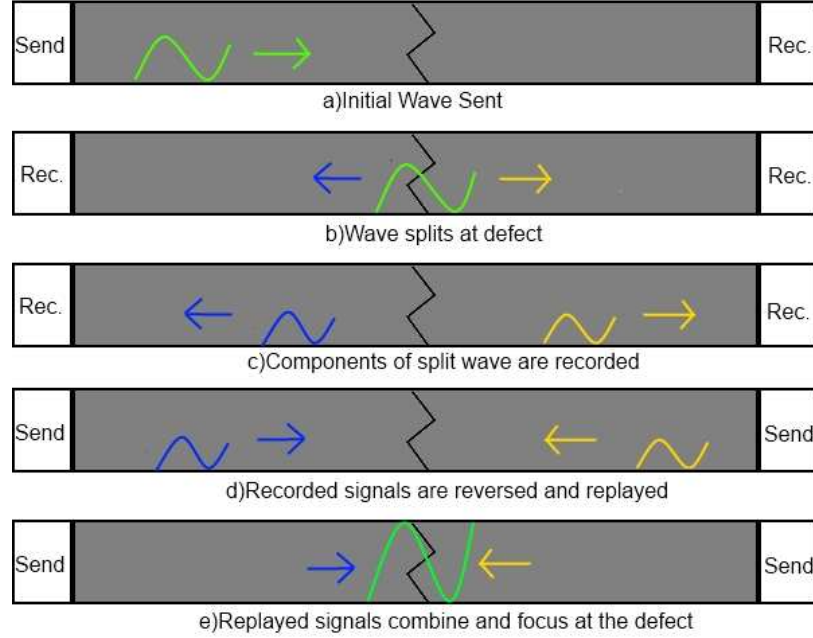


Figure 1.4: Concept drawing of time reversal in one dimension; a) An initial wave is sent out by a transducer; b) The wave travels down the rod until it hits the crack and splits into multiple components; c) The components of the incident wave travel towards each transducer where they are recorded; d) The recorded signals are amplified and replayed in a time reversed fashion; e) The wave components simultaneously reach the crack location where they first split and combine to cause a focusing of their amplitude at that point.

Mathematically, time reversal is the time domain analog of phase conjugation which has been used in adaptive optics to correct for wave front phase aberrations by forming a mirror to produce the conjugate phase aberrations of the incoming wave which results in a corrected wave front [42]. Acoustic time reversal has been studied extensively by Fink et. al. in which he first setup an array of transducers to send and

receive ultrasonic waves through a rubber medium with random thicknesses and a hydrophone embedded as a reflector/receiver. Fink also has shown iterative focusing by experimenting with an array of transducers and two metal wires as reflectors. In the iterative experiments, it was shown that the energy focused on the strongest reflector after successive iterations [43]. Experiments were performed by Derode et. al. in which a transducer played a signal through a liquid containing over 2000 rods and an array of transducers located on the opposite side of the rods played the captured signals back and it was found that the replayed signals converged on the original source [44]. Time reversal has been applied to acoustic communications in both the ocean and air to achieve better signal to noise ratio by focusing the signal at the desired receiver [45–47]. Great promise has also been shown with applying time reversal to biomedical application such as the focusing of ultrasonic waves to destroy kidney stones or hyperthermia brain treatment [48]. The concept of time reversal is not only applicable to acoustics but can also be used with electromagnetic waves [49]. Liu et. al. have shown experimentally that bit-rate-errors in radio communications can be reduced by applying time reversal [50]. MIT has effectively created a microwave cannon by using the time reversal process to focus microwave energy [51].

## 1.4 Objectives

The overall goal, as devised by Dr. Korde, is to use time reversal acoustic focusing to accelerate the crack recovery process of a self-healing polymer by locally increasing the temperature and pressure. The work presented here looks at a subsection of that

project which is the time reversal focusing at a crack location. There has been much theoretical and experimental work performed on applying time reversal in multiple dimensions. Here, the case of time reversal in one dimension is explored. Fouque et. al. performed calculations and simulations on one dimensional time reversal in random media with an embedded reflector to determine the reflector's location [52]. Korde performed calculations on focusing at a crack with piezoelectric transducers and time reversal processing [53]. To the author's knowledge, time reversal focusing at a defect location in one dimension has not been performed experimentally. In this work experiments are performed on circular steel and nylon rods with piezoelectric transducers to send and receive signals. This could have direct applications to the rods used in deployable space structures that may become cracked over time. Although the analysis is eased in this scenario, there are difficulties introduced in performing the experiments in one dimensional structures of finite length that are not present when performing time reversal in multiple dimensions. For instance, it is required that a transducer both sends a signal and reads a signal in the same iteration such that the ringing of the transducer may interfere with the signal being read.

The specific objectives of this work are to show:

1. Modeling and experimental verification of acoustic time reversal crack focusing in one dimension
2. Experimental verification of acoustic crack detection in one dimension

3. Experimental verification of iterative focusing and convergence of acoustic time reversal crack focusing in one dimension
  - (a) Show that with each iteration, the amplitude seen at the defect increases until a convergence point is reached

# Chapter 2

## Modeling

This chapter begins with the models and derivations for the case of one dimensional time reversal and is based upon work performed by Korde [53]. Following that material will be a brief overview of two dimensional time reversal which is drawn from a number of published sources.

### 2.1 One Dimensional Wave Equation

In this section the one dimensional wave equation for a linear elastic material with uniform density and cross-sectional area will be derived. The derivation is based on knowledge from numerous books such as Kolsky [54] and Brown et. al. [55].

Figure 2.1 gives a basic diagram that is used for the wave equation derivation, with  $P$  being the axial pressure,  $dz$  being the material's differential element of length, and  $A$  being the cross-sectional area of the material. Define the stress in the  $z$  direction as:

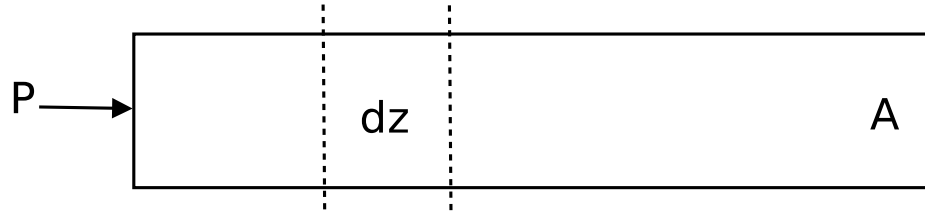


Figure 2.1: Diagram for deriving the one dimensional wave equation given an axial pressure applied to one end of the material with uniform density and cross-sectional area.

$$T_3 = \frac{P}{A} \quad (2.1)$$

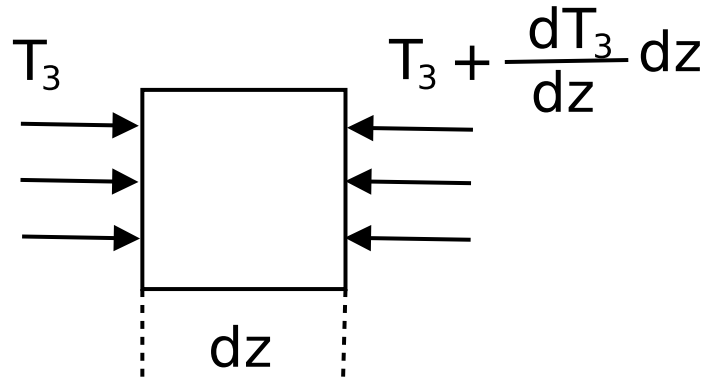


Figure 2.2: Differential element showing the stresses acting on the rod.

Applying Newton's second law and summing the forces,

$$-T_3 A + \left(T_3 + \frac{dT_3}{dz} dz\right) A = A \rho^* dz \frac{d^2 w}{dt^2} \quad (2.2)$$

With  $\rho^*$  being the density of the rod material and  $w$  being the displacement.

Dividing through by the area,

$$-T_3 + T_3 + \frac{dT_3}{dz}dz = \rho^* dz \frac{d^2w}{dt^2} \quad (2.3)$$

$$\frac{dT_3}{dz}dz = \rho^* dz \frac{d^2w}{dt^2} \quad (2.4)$$

Canceling the  $dz$  term,

$$\frac{dT_3}{dz} = \rho^* \frac{d^2w}{dt^2} \quad (2.5)$$

$C_{33}^*$  is the elastic constant in the  $z$  direction and is defined as the ratio of the stress and strain in the  $z$  direction. This is given by,

$$C_{33}^* = \frac{T_3}{S_{33}} \implies T_3 = C_{33}^* S_{33} \quad (2.6)$$

With  $S_{33}$  being the strain in the  $z$  direction. Strain is the ratio of the change in material length to the original length and in this case is given by,

$$S_{33} = \frac{dw}{dz} \quad (2.7)$$

Inserting 2.7 in to 2.6 we see that

$$T_3 = C_{33}^* \frac{dw}{dz} \quad (2.8)$$

Putting 2.8 back in to 2.5 and realizing that  $w$  is a function of both  $t$  and  $z$ ,



$$C_{33}^* \frac{\partial^2 w}{\partial z^2} = \rho^* \frac{\partial^2 w}{\partial t^2} \quad (2.9)$$

Dividing through by  $\rho^*$  and moving the LHS term to the RHS, we arrive at

$$\frac{\partial^2 w}{\partial t^2} - c^{*2} \frac{\partial^2 w}{\partial z^2} = 0 \quad (2.10)$$

With  $c^*$  being the velocity of the wave propagation and is defined as,

$$c^* = \sqrt{\frac{C_{33}^*}{\rho^*}} \quad (2.11)$$

For the fixed-fixed unforced case, if we assume that  $w(z, t)$  is a linear function then the solution can be found by a separation of variables approach. Let

$$w(z, t) = Z(z)T(t) \quad (2.12)$$

Inserting 2.12 into 2.10

$$\frac{\partial^2 ZT}{\partial t^2} - c^{*2} \frac{\partial^2 ZT}{\partial z^2} = 0 \quad (2.13)$$

Dividing through by  $ZT$

$$\frac{1}{c^{*2}T} \frac{\partial^2 T}{\partial t^2} = \frac{1}{Z} \frac{\partial^2 Z}{\partial z^2} \quad (2.14)$$

Since  $Z$  and  $T$  are independent, this can only be true if each term is equal to some constant

$$\frac{1}{c^{*2}T} \frac{\partial^2 T}{\partial t^2} = \frac{1}{Z} \frac{\partial^2 Z}{\partial z^2} = -\lambda^2 \quad (2.15)$$

Then two ordinary differential equations are obtained

$$T''(t) + \lambda^2 c^{*2} T(t) = 0 \quad (2.16)$$

$$Z''(z) + \lambda^2 Z(z) = 0 \quad (2.17)$$

The solution for  $Z(z)$  is

$$Z(z) = A \cos(\lambda z) + B \sin(\lambda z) \quad (2.18)$$

Inserting the boundary conditions  $Z(0) = 0$  and  $Z(L) = 0$  2.18 reduces to

$$Z(z) = B_n \sin\left(\frac{n\pi z}{L}\right) \quad (2.19)$$

With

$$\lambda_n = \frac{n\pi}{L} \quad (2.20)$$

Now letting

$$\lambda_n c^{*2} = -\omega_n^2 \quad (2.21)$$

and inserting 2.21 in to 2.16 and applying a zero disturbance initial condition  $\dot{T}(0) = 0$  the solution for  $T(t)$  is found to be

$$T(t) = C_n \cos(\omega_n t) \quad (2.22)$$

Combining 2.19 and 2.22 and summing to infinity the solution for  $w(z, t)$  is found such that

$$w(z, t) = \sum_{n=1}^{\infty} A_n \cos(\omega_n t) \sin(\lambda_n z) \quad (2.23)$$

Applying the principles of orthogonality

$$A_n = \frac{2}{L} \int_0^L w(z, 0) \sin\left(\frac{\lambda_n z}{L}\right) dz \quad (2.24)$$

## 2.2 Piezoelectric transducers

This section describes the governing equations for  $d_{33}$  piezoelectric transducers that are placed on each end of a rod. The following work is based on derivations by Dr. Korde [53].

The constitutive equations for the transducers are:

$$T_3 = \overline{C}_{33} S_3 + \overline{d}_{33} D_3 \quad (2.25)$$

and

$$D_3 = \overline{\epsilon_{33}^T} E_3 + \frac{d_{33}}{s_{33}^E} S_3 \quad (2.26)$$

Where  $D_3$ ,  $E_3$ , and  $d_{33}$  are the electric displacement, electric field intensity, and the strain constant relating strain and field intensity, respectively, all in the z direction. The other variables are given by the following equations:

$$s_{33}^E = \frac{1}{C_{33}} \quad (2.27)$$

$$\overline{C_{33}} = \frac{1}{s_{33}^E \left( 1 - \frac{d_{33}^2}{s_{33}^E \epsilon_{33}^T} \right)} \quad (2.28)$$

$$\overline{d_{33}} = - \frac{d_{33}/\epsilon_{33}^T}{s_{33}^E \left( 1 - \frac{d_{33}^2}{s_{33}^E \epsilon_{33}^T} \right)} \quad (2.29)$$

$$\overline{\epsilon_{33}^T} = - \frac{\epsilon_{33}^T}{s_{33}^E \left( 1 - \frac{d_{33}^2}{s_{33}^E \epsilon_{33}^T} \right)} \quad (2.30)$$

With  $C_{33}$  and  $\epsilon_{33}^T$  being the PZT elastic constant in the z direction and permittivity constant in the z direction respectively.

The equivalent circuits for the transducers are modeled as a resistor and capacitor in series as shown in Figure 2.3 and with  $C_c$  being the equivalent capacitance and is given by

$$C_c = \overline{\epsilon_{33}} \pi a^2 / l \quad (2.31)$$

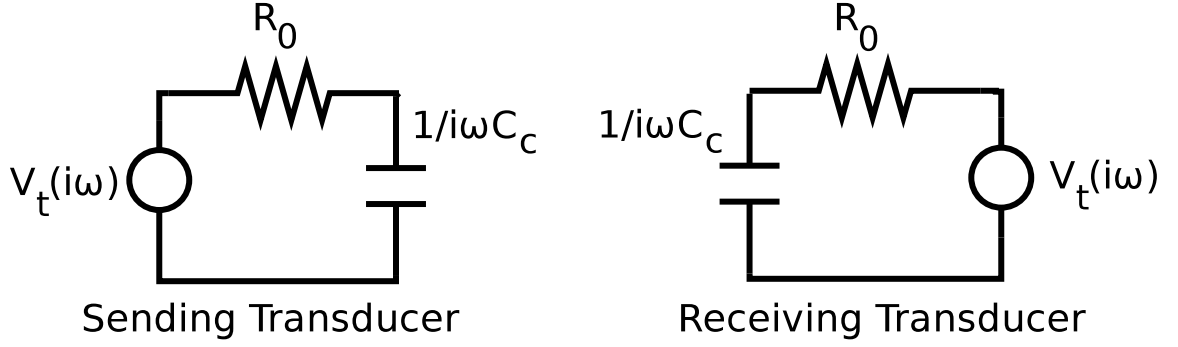


Figure 2.3: Equivalent circuit diagram for the piezoelectric transducers.

Where  $a$  and  $l$  are the transducer area and length respectively.

## 2.3 Rod with Transducers

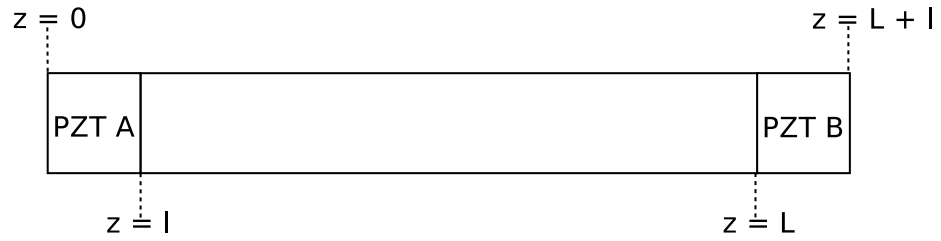


Figure 2.4: Diagram of a rod with  $d_{33}$  piezoelectric transducers placed on each end.

Consider a rod with  $d_{33}$  transducers affixed to each end as pictured in Figure 2.4. We wish to model propagation of a wave through the system with the wave being sent by one of the transducers. The one-dimensional wave results from Section 2.1 can be applied here such that the equation of the motion for the wave propagating through the transducers is the same as through the rod but with a different density and elastic coefficient which will be denoted as  $\rho$  and  $C_{33}$  respectively. The wave velocity through the transducer is then described as

$$c = \sqrt{\frac{C_{33}}{\rho}} \quad (2.32)$$

So for  $0 \leq z \leq l$  and  $L < z \leq (L + l)$  we have

$$\frac{\partial^2 w}{\partial t^2} - c^2 \frac{\partial^2 w}{\partial z^2} = 0 \quad (2.33)$$

For the rod domain,  $l < z \leq L$ , we just apply equation 2.10. Let the stress and deflection across the interfaces be continuous. When a wave encounters either of the interfaces it will experience refraction in which part of the wave will be reflected and part will be transmitted into the next medium. This phenomena can be accounted for by finding the reflection and transmission coefficients for each interface. For  $z = l$  these coefficients are found to be

$$\mathbf{R} = \frac{\overline{C_{33}} + \frac{\overline{d_{33}d_{33}}}{2s_{33}^E} - C_{33}^*}{\overline{C_{33}} + \frac{\overline{d_{33}d_{33}}}{2s_{33}^E} + C_{33}^*} \quad (2.34)$$

$$\mathbf{T} = \frac{2 \left( \overline{C_{33}} + \frac{\overline{d_{33}d_{33}}}{2s_{33}^E} \right)}{\overline{C_{33}} + \frac{\overline{d_{33}d_{33}}}{2s_{33}^E} + C_{33}^*} \quad (2.35)$$

Similarly, for the interface at  $z = (L + l)$

$$\hat{\mathbf{R}} = \frac{\overline{C_{33}} - \frac{\overline{d_{33}d_{33}}}{2s_{33}^E} - C_{33}^*}{\overline{C_{33}} + \frac{\overline{d_{33}d_{33}}}{2s_{33}^E} + C_{33}^*} \quad (2.36)$$

$$\hat{T} = \frac{2C_{33}^*}{\overline{C}_{33} + \frac{d_{33}d_{33}^*}{2s_{33}^E} + C_{33}^*} \quad (2.37)$$

## 2.4 Cracked Rod with Transducers

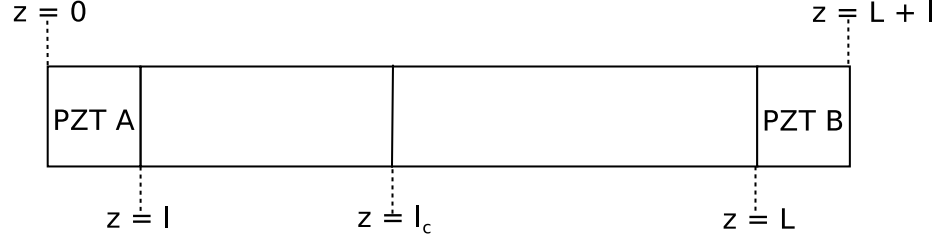


Figure 2.5: Diagram of a rod with  $d_{33}$  piezoelectric transducers placed on each end and a crack located at an arbitrary location within the system, with  $l_c$  being the distance of the crack from the origin and is not known a prior.

In this section a crack located at an arbitrary position is introduced into the system modeled in Section 2.3 and is shown in Figure 2.5. The crack can be modeled as another interface boundary within the system as we saw previously. If a stress wave is played out from one of these transducers, say transducer A, then it will travel through the rod and refracts upon reaching the crack location due to a change in the medium. The crack can then be accounted for as a change in the material density and elastic coefficient from  $\rho^*$  to  $\rho^c$  and from  $C_{33}^*$  to  $C_{33}^c$  respectively. Again, let the stress and strain at the crack interface be continuous. As seen before we will have a reflection coefficient and a transmission coefficient which are given by

$$\overline{R} = \frac{C_{33}^* - C_{33}^c}{C_{33}^* + C_{33}^c} \quad (2.38)$$

$$\bar{T} = \frac{2C_{33}^*}{C_{33}^* + C_{33}^c} \quad (2.39)$$

## 2.5 One Dimensional Time Reversal Focusing at a Crack

Considered here is the time reversal playback of the system modeled thus far. First there is an interrogatory phase in which transducer A plays out a stress wave which will propagate along the rod towards transducer B on the opposite end. When the wave reaches the crack location it will refract causing part of its energy to be transmitted and part to be reflected and is described by the coefficients shown in Section 2.4. The strain caused by the wave reflected portion returning to the transducer A is

$$\frac{\partial w_{cr}}{\partial z} = \bar{R} \frac{\partial w_t}{\partial z} \quad (2.40)$$

For the transmitted portion of the wave, its strain will be

$$\frac{\partial w_{ct}}{\partial z} = \bar{T} \frac{\partial w_t}{\partial z} \quad (2.41)$$

In both 2.40 and 2.41

$$\frac{\partial w_t}{\partial z} = T \frac{\partial w}{\partial z} \quad (2.42)$$



This will create a voltage at transducer A which is found in the frequency domain as

$$V_a^+(i\omega) = -\frac{d_{33}\pi a^2}{2s_{33}^E C_c} \frac{\partial w_{cr}}{\partial z}(i\omega) \quad (2.43)$$

and at transducer B

$$V_b^+(i\omega) = \frac{d_{33}\pi a^2}{2s_{33}^E C_c} \hat{T} \frac{\partial w_{ct}}{\partial z}(i\omega) \quad (2.44)$$

These voltages are recorded by each of the transducers. The reversal of these recorded voltages is just the complex conjugate for each signal such that the replayed voltage by each transducer will be

$$V_a^+ R(i\omega) = V_a^+ * (i\omega) \quad (2.45)$$

and

$$V_b^+ R(i\omega) = V_b^+ * (i\omega) \quad (2.46)$$

In addition to playing back the reversed signals, each transducer must also play back an idle period to account for the wave propagation time and the likely possibility that the crack is located closer to one transducer than the other. This information is inferred from the signal recorded and the appropriate zero period is added to the transducers. If the reversed signals are played back by each transducer with the

necessary idle periods then the waves reach the defect at the same instant where they constructively interfere. The combined strain produced at the defect in the time domain form is found to be

$$\begin{aligned} \frac{\partial w_c}{\partial z}(l_c, t) = & -\frac{2s_{33}^E \mathbf{T}}{d_{33}\pi a^2 R_0} V_0 \delta(t) \left( \frac{\hat{\mathbf{T}} \mathbf{M} \mathbf{T}}{R_0 C_c} \right) \times \\ & \left( \overline{\mathbf{R}}(-2\nu l_c) + \overline{\mathbf{T}} \exp(-2\nu l_T) \times \exp \left[ -\frac{1}{R_0 C_c} \left( t - \frac{3l_c}{c^*} - \frac{2l}{c} \right) \right] \right) \end{aligned} \quad (2.47)$$

Where  $\nu$  is a spatial dissipation factor and  $l_T$  is the distance from the crack to transducer B.

Through the use of Green functions it is found that the amplification of the focusing at the defect will increase if the time reversal is applied in an iterative fashion such that after each transducer replays a time reversed signal it records the new response and plays that record back in a time reversed fashion. This amplification will continue until it becomes limited by material dissipation at which point the algorithm has converged. Because the rod is of finite length there will be natural vibrations produced when the rod is excited. However, it is argued that if a sufficient delay time is introduced between successive iterations then natural vibrations of the rod die down while the amplification at the crack location increases.

## 2.6 Time Reversal Focusing in Two Dimensions

Here a brief overview is given of focusing acoustic energy in two dimensions using the time reversal method and is drawn from Fink [56]. Figure 2.6 depicts a scenario in

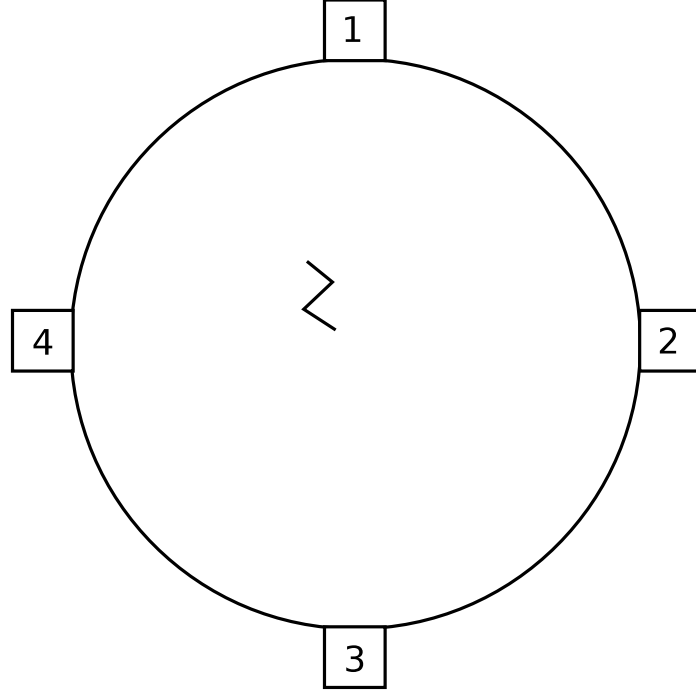


Figure 2.6: Concept diagram of a two dimensional time reversal setup in which transducers are located around the boundary of a circular plate. A crack exists on the plate and is the point at which the transducers will focus acoustic energy.

which transducers are located around the boundary of a plate and a defect is located at an arbitrary position on the plate. The method used to focus acoustic energy at the crack is a natural extension of the time reversal method used for the rod. The equation of motion for the wave propagation is the similar to 2.10 except we now introduce the Laplacian to account for the multiple spatial dimensions which gives

$$\frac{\partial^2 w}{\partial t^2} - c_p^2 \nabla^2 w(\vec{r}, t) = 0 \quad (2.48)$$

Where  $c_p$  is the speed of sound through the plate material and  $\vec{r}$  is the position of the acoustic pressure field.

As before, a transducer sends out an initial pulse. Let the field produced in  $\vec{r}$  at time  $t$  caused by a pulse emitted at time  $t_0$  by a source located at  $\vec{r}_0$  be represented by a Green function

$$G(\vec{r}_0, t_0 | \vec{r}, t) \quad (2.49)$$

Because of the reciprocity theorem,

$$G(\vec{r}_0, t_0 | \vec{r}, t) = G(\vec{r}, t_0 | \vec{r}_0, t) \quad (2.50)$$

Which means that if  $\vec{r}$  replays the pulse it feels due to  $\vec{r}_0$  then  $\vec{r}_0$  will experience that same excitation. Let the integral over the transducer surface  $S_i$  be

$$h_i(\vec{r}_0) = \int_{S_i} G(\vec{r}_0, t_0 | \vec{r}, t) d\vec{r} = \int_{S_i} G(\vec{r}, t_0 | \vec{r}_0, t) d\vec{r} \quad (2.51)$$

The function relating pressure to voltage for the transducers is written as

$$h_i^{ac}(t) \quad (2.52)$$

After the wave sent by a transducer reaches the defect in the plate, it will refract and the crack will act as an acoustic source located at  $\vec{r}_0$ . The voltage recorded by each transducer is seen as the convolution of

$$h_i^{ac}(t) * h_i(\vec{r}_0, t) \quad (2.53)$$

These signals are time reversed for each transducer such that the replayed signals

are

$$h_i^{ac}(T-t) * h_i(\vec{r}_0, T-t) \quad (2.54)$$

The response at the crack due to one transducer is the convolution of the replayed signal and the original pulse emitted from the crack

$$h_i^{ac}(T-t) * h_i(\vec{r}_0, T-t) * h_i(\vec{r}_0, t) * h_i^{ac}(t) \quad (2.55)$$

The contribution due to all of the transducer is simply the sum over  $i$  of the individual contributions

$$\sum_i h_i^{ac}(T-t) * h_i(\vec{r}_0, T-t) * h_i(\vec{r}_0, t) * h_i^{ac}(t) \quad (2.56)$$

## Chapter 3

# Experiments

This chapter describes the experiments that were carried out to investigate the different components of one dimensional time reversal focusing at a defect location which is at an a prior unknown location. The components investigated include: i) detecting that a crack as occurred within the rod, ii) comparing theory to experiment for the interrogatory phase and for the first iteration of time reversal focusing, and iii) iterative focusing to achieve better a response at the defect until convergence is reached. All of the hardware and algorithms used are detailed. Code samples are given in the appendices.

### 3.1 Hardware and Software

The main access point to the system was a desktop pc which ran Windows XP and acted as the host computer. The power house of the entire system was a National Instruments 7853R FPGA card. This FPGA card was of PXI form factor and was

housed in an NI proprietary PXI chassis which was linked to the PC via MXI cable (Figure 3.1). The PC contained a PCI-E card that allowed for the MXI connection. LabVIEW software with the extra real-time and FPGA modules was used to compile and write programs on the Windows PC. Once written and compiled to VHDL code the programs were then downloaded to the FPGA card over the MXI connection. The program execution is such that a host program runs on the desktop PC and communicates with the program running on the remote FPGA card. The time critical processes such as signal acquisition and playback are performed using the FPGA whereas the tasks that could take an arbitrary amount of time were executed on the desktop PC. The FPGA card does not provide a large amount of current and it was necessary to amplify the output signals using op-amp amplifier circuits which were capable of providing up to  $30V_{PP}$  at  $200kHz$  (Figures 3.2 and 3.3). The interfacing of the data acquisition cables to the FPGA card was handled using an NI SCB-60 input/output box which provided screw terminal connections inside of a shielded box (Figure 3.4). The entire hardware hierarchy is shown in Figure 3.5. Finally, all of the data was imported into MATLAB.

## 3.2 Crack Detection Tests

The first topic investigated is if the system is able to sense when a defect has become present within the material. This step is important because it would be inefficient and potentially damaging to attempt to perform acoustic focusing when no defect is present in the system.

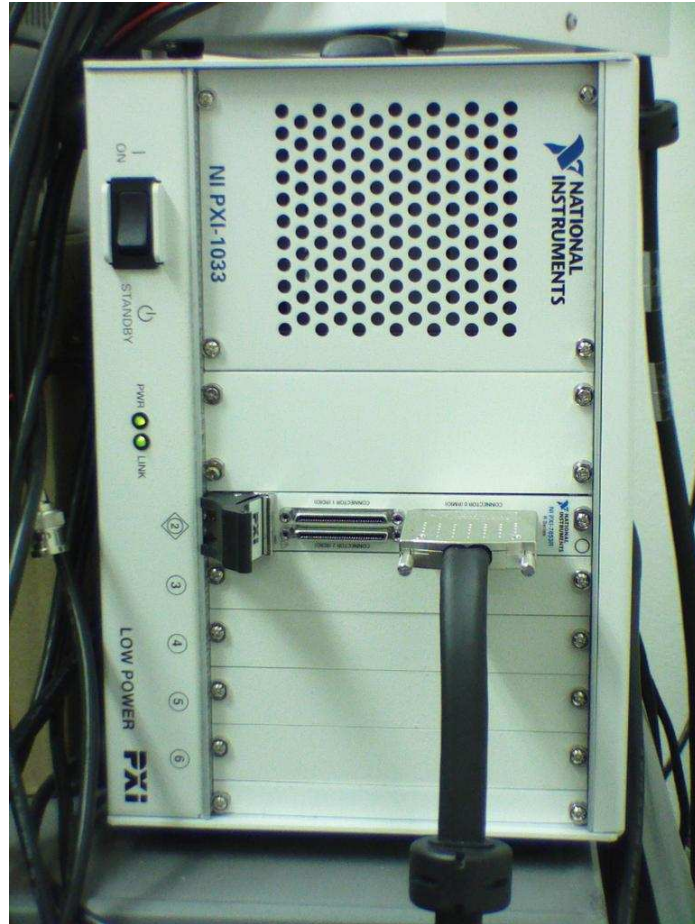


Figure 3.1: National Instruments PXI chassis which contains the 7853R FPGA card. The 7853R is a very powerful card with a large amount of programming flexibility due to the fact that it can be programmed using a graphical language (LabVIEW) instead of needing to code directly in VHDL which can be cumbersome.

The setup used for the crack detection involved the use of both nylon and steel solid circular rods. Each rod was approximately  $12.7 \times 10^{-3}m$  in diameter. The steel rod was tested first and found to have a density of  $7.894 \times 10^3 kg/m^3$ , an elastic constant of  $169.39 \times 10^9 N/m^2$ , and a length of  $579 \times 10^{-3}m$ . The actuators used to send and receive the signals were piezoceramic stack actuators (PZT) with a diameter of  $13 \times 10^{-3}m$  with a piezoelectric strain constant of  $d_{33} = 152 \times 10^{-12} C/N$  and a



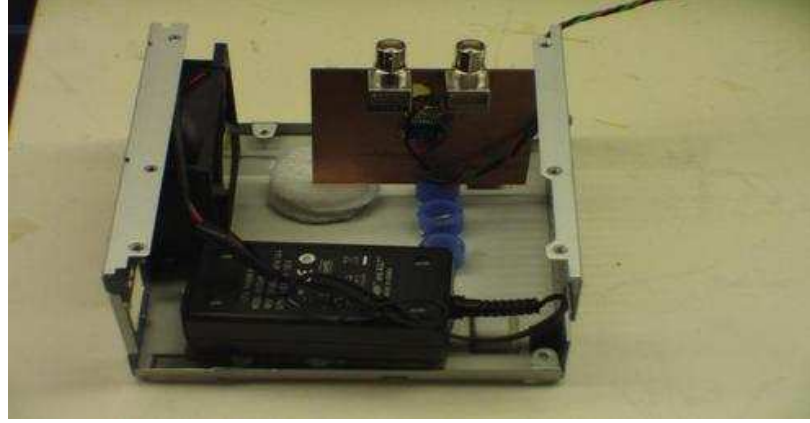


Figure 3.2: Single amplifier circuit being placed into the custom enclosure that was built from a PC power supply casing. Many thanks go to Miles Wickersham and Andy Downs for their help with designing the circuits.



Figure 3.3: Completed amplifier box containing four op-amp amplifier circuits with BNC input and output

relative permittivity of  $\epsilon_{33} = 450\epsilon_0$  with  $\epsilon_0 = 8.854 \times 10^{-12} F/m$ . Two transducers were affixed to each end of the steel rod using accelerometer putty with PZT A being on the left side of the rod and PZT B being on the right side. This system was



Figure 3.4: NI SCB-60 shielded input/output box that is used for interfacing electrical connections to the 7853R FPGA card.

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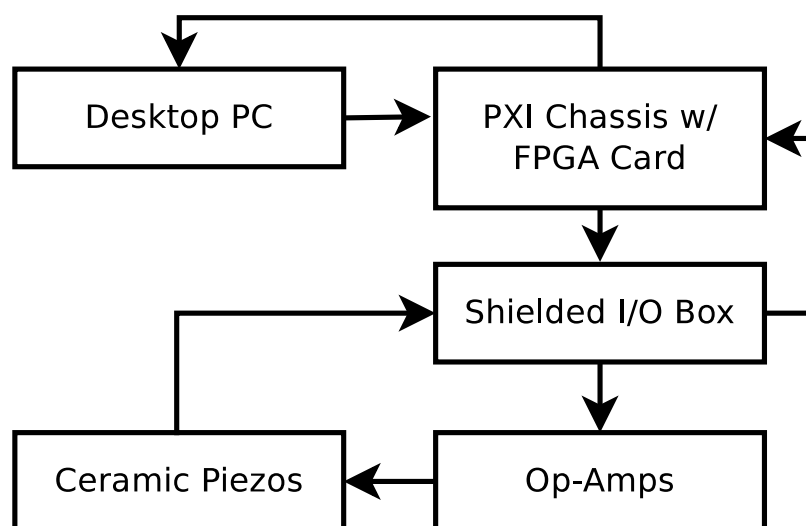
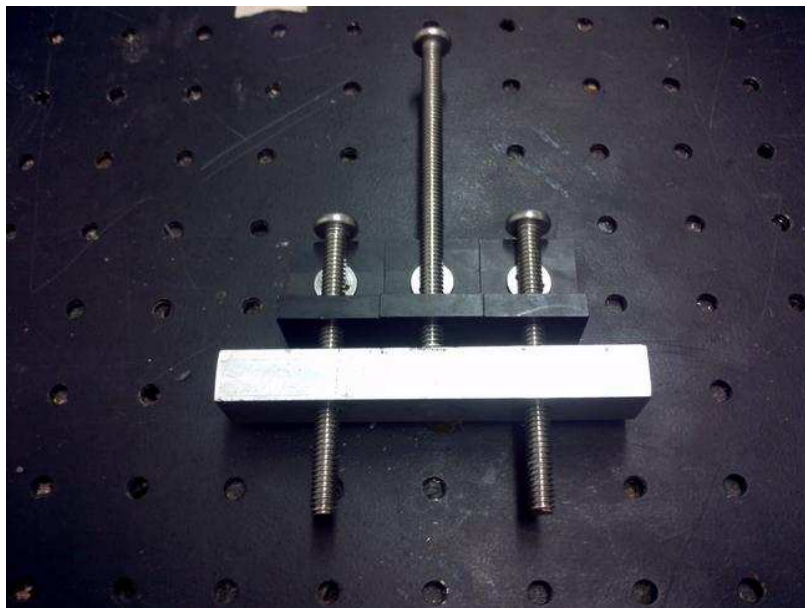


Figure 3.5: Diagram which shows the communication hierarchy among the different hardware pieces that were used for testing. The communication between the PC and FPGA card was performed using a MXI cable. The acquisition cables interfaced to the FPGA through an NI shielded I/O box and cable. Any other connections were made using normal insulated copper wire.

then placed into compression using a custom built mechanism. The compression mechanism was created using common nuts and bolts, optics table hardware, and a few machined aluminum blocks (Figure 3.6). The compression could be controlled by the use of a single bolt and was tightened with a small screw driver to one-quarter turn past being finger tight. The experiments were carried out on an optics table which significantly reduced external vibrations through the use of air damping. The setup is shown in Figure 3.7. Also pictured in the setup is a terminal block that was used for handling the copper wire connections (Figure 3.8). The terminal block also contains a diode circuit consisting of two diodes that is placed between the amplifier output lines and the analog read in lines for each PZT (Figure 3.9). The diode circuit snuffs out noise from the amplifier that was found to be large enough to overshadow some test results and will be discussed more in the results section.

The software used consisted of two main programs which were the host program and the FPGA program. The host program ran on the PC and the FPGA program was downloaded to the FPGA card where it was executed. Communication between the host and FPGA programs was handled with the FPGA's First-In-First-Out memory protocol (FIFO) to which the host program had direct memory access (DMA). The synchronization of the data transactions between the host and FPGA programs is handled using a handshaking protocol implemented by IRQs.

The purpose of this set of experiments was to determine if the occurrence of a crack could be detected within the steel rod. The idea is to send a signal from PZT



(a) Left side compression mechanism



(b) Right side compression mechanism

Figure 3.6: (a) The left side compression mechanism. The middle bolt was used to adjust the compression in the system by using it to move the aluminum block back and forth; (b) Right side compression mechanism. This had two nuts to adjust the aluminum bar's positions so it could accommodate different rod sizes.

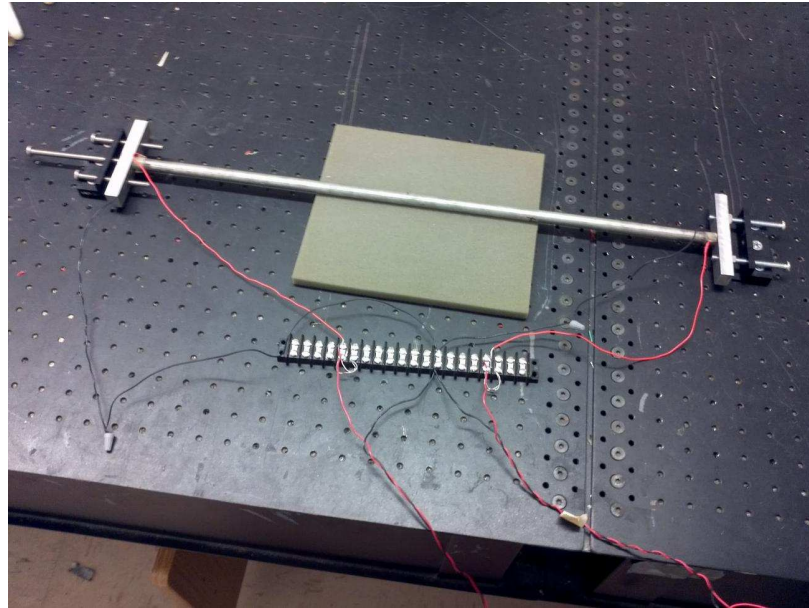


Figure 3.7: Steel rod with PZTs affixed to each end using accelerometer putty. The system is in compression using the custom mechanism and is seen on the optics table. The PZT on the left is referred to as PZT A and the right PZT is referred to as PZT B.

A and record the response at PZT B. This continues and successive responses are compared to the first response and the program attempts to detect a significant change between the responses which indicates that something in the medium has changed and potentially represents the appearance of a crack. For the implementation, the host program first generated an initial multi-tone wave. The wave is composed of individual sinusoidal waves of different frequency that were concatenated together to form the overall multi-tone wave. The center frequency for the multi-tone wave was  $115\text{kHz}$  and had a bandwidth of  $30\text{kHz}$ . The relationship between the number of points needed to create the wave and the sampling frequency of the FPGA card ( $740.741\text{ kHz}$  here) was used to obtain the correct frequency for each wave. A total of



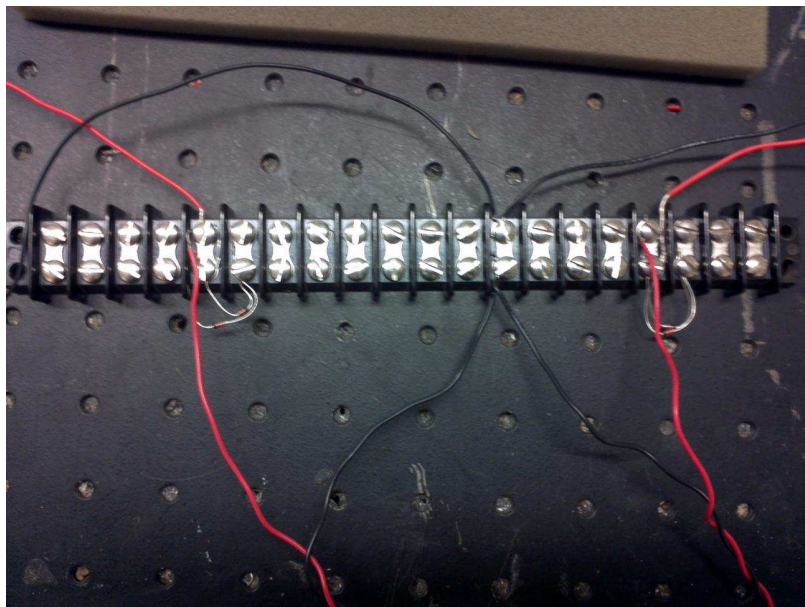


Figure 3.8: Simple terminal block that was used for handling the connections among the various data acquisition wires. Also contained on the terminal block is a diode circuit between the amplifier output and the analog read in line for each PZT.

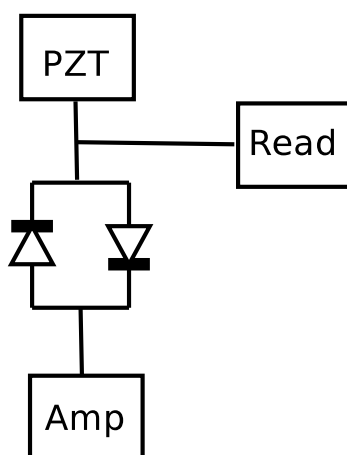


Figure 3.9: Diode circuit that is placed between the amplifier output and analog read in lines for each PZT. This circuit snuffs out the noise produced by the amplifiers at the expense of a slight reduction in voltage output to the PZTs.

7,8, and 9 data points were used by the program to represent a single sinusoidal wave period of the  $130kHz$ ,  $115kHz$ , and  $100kHz$  wave components, respectively. The

$130kHz$  wave was placed in the center of the wave grouping with a  $115kHz$  wave on both sides of it and then a  $100kHz$  wave on the ends of the grouping (Figure 3.10).

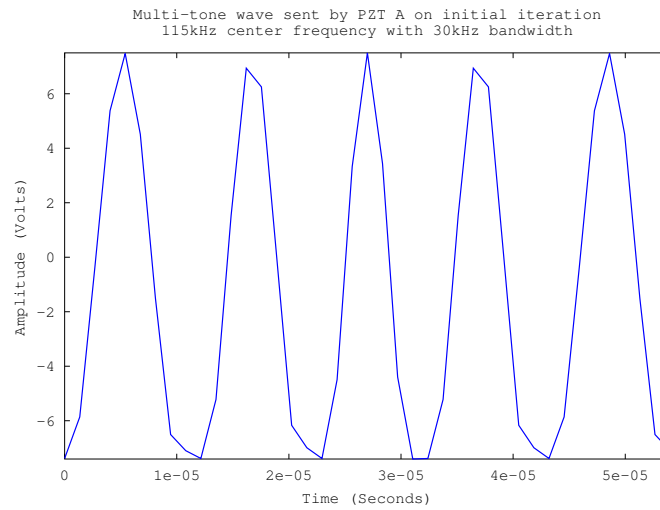


Figure 3.10: Mutli-tone wave that was played out by PZT A. The wave was created by concatenating single period sinusoidal waves of different frequency. The wave in the middle is  $130kHz$ , the waves next to it are  $115kHz$ , and the outside waves are  $100kHz$ . It was found that this multi-tone wave caused a better response in the material versus using a single frequency.

After the wave was generated it was transferred to the FPGA program which then zeroed all of its output values and waited for  $200ms$ . This wait time was chosen based upon experimental results for the time of a signal to decay when propagating through a steel or nylon rod. Once the wait time elapsed, the signal was sent into the rod by PZT A. The wave propagated through the rod and was recorded on the other end by PZT B. This was performed five times total and the response for each iteration was recorded. With the data for the undamaged rod recorded it was time to test a damaged sample. The damaged steel rod was created by taking the undamaged rod that was used and cutting it in to two pieces of arbitrary length (Figure 3.11). A

PZT was placed on one end of each piece and the free ends of the pieces were pressed together. What was formed was essentially the same rod that was previously but with a crack location at an arbitrary position (Figure 3.12). The program was then ran five times on the new setup with a signal being sent by PZT A and recorded at PZT B.

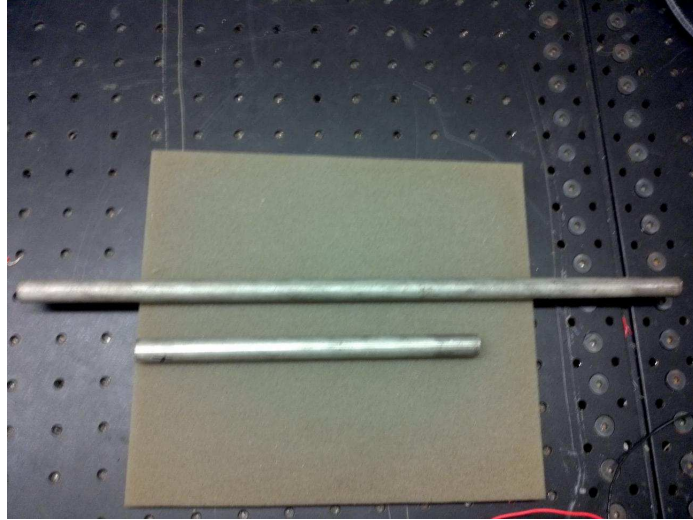
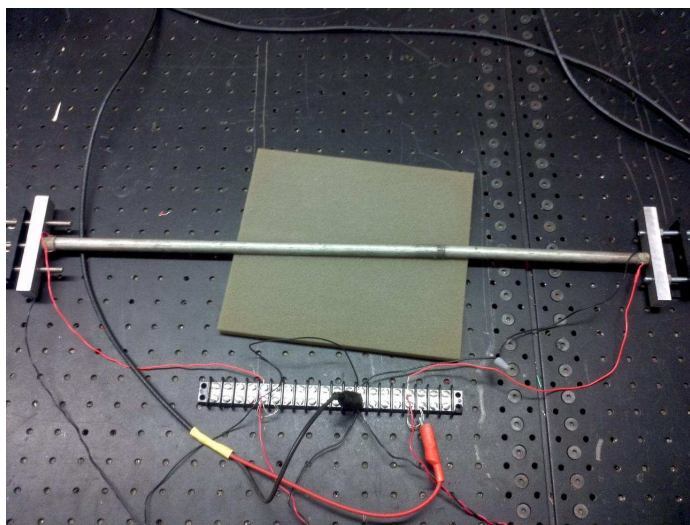


Figure 3.11: Undamaged rod that has been cut in to two pieces of random length. This pieces are then pushed together (end to end) to represent a rod with a crack location.

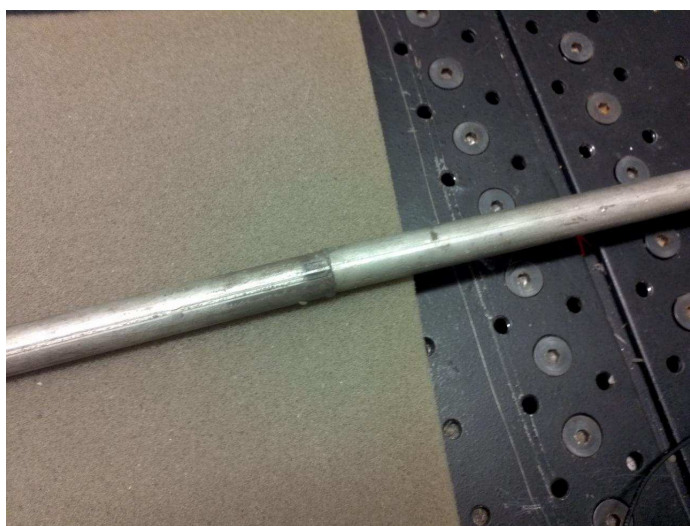
A MATLAB program read in the data from each run and compared the responses using a simple sum of differences squared formula. The five signals for each test were also averaged together to get a better idea of what the signal looks like in the undamaged tests compared to the damaged rod tests.

These exact same tests were then carried out for a nylon rod. The nylon rod was found to have a density of  $0.854 \times 10^3 \text{ kg/m}^3$ , an elastic constant of  $4.68121 \times 10^9$  and a length of  $148 \times 10^{-3} \text{ m}$ . The setup for the nylon rod tests is shown in Figures 3.13





(a) Full view of steel 'damaged rod' tests



(b) Close up of the defect location

Figure 3.12: (a) Full view of the two steel rod pieces that are pressed together to represent a rod with a defect location.; (b) Close up view of the defect location in the steel rod system which should cause a noticeable change in the response at PZT B.

and 3.14.

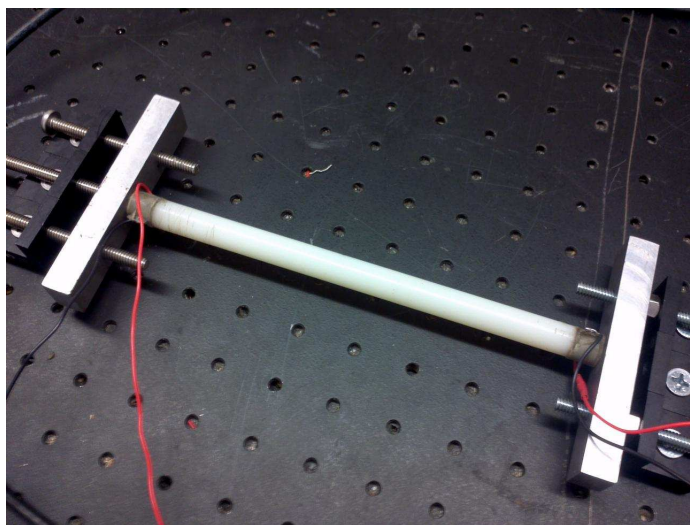
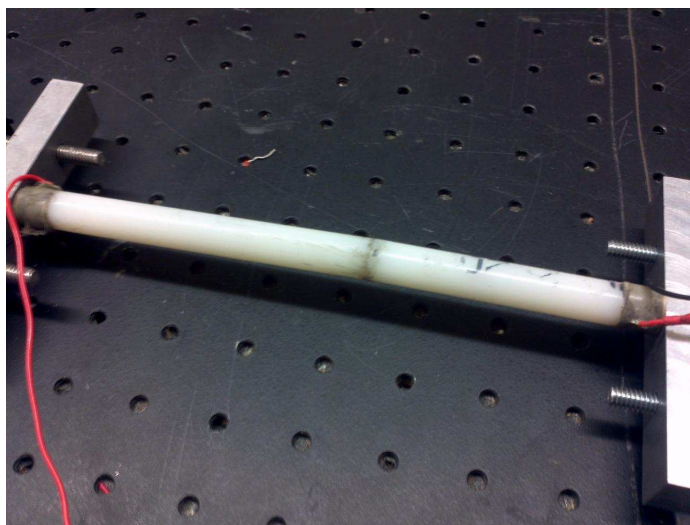


Figure 3.13: Undamaged nylon rod that is setup for testing. A PZT was affixed to each rod end using accelerometer putty and placed into compression using the custom mechanism.

### 3.3 Time Reversal Tests

The next step was to perform acoustic focusing at a crack located at an a priori unknown location. The tests were performed on both steel and nylon rods. The setup was very similar to that described for the crack detection tests. The main difference between these setups is that for the time reversal testing an additional PZT was placed between the two rod sections and will be referred to as the 'defect PZT'. The defect PZT acted as a defect and also provided a way to recorded the response at that location so that the overall effectiveness of the focusing algorithm could be determined. For the steel rod tests, combinations of three rod sections of different length were used which gave a total of three test combinations. The lengths



(a) Full view of nylon 'damaged rod' tests



(b) Close up of the defect location

Figure 3.14: (a) Full view of the two nylon rod pieces that are pressed together to represent a rod with a defect location.; (b) Close up view of the defect location in the nylon rod system which should cause a noticeable change in the response at PZT B.

of the steel rods were  $579.0 \times 10^{-3}m$ ,  $437.0 \times 10^{-3}m$ , and  $428.0 \times 10^{-3}m$ . Similarly for the nylon rods three lengths were used which were  $148.0 \times 10^{-3}m$ ,  $105.0 \times 10^{-3}m$ , and  $75.0 \times 10^{-3}m$ . The setup is shown schematically in Figure 3.15 and pictorially in Figures 3.16 and 3.17.

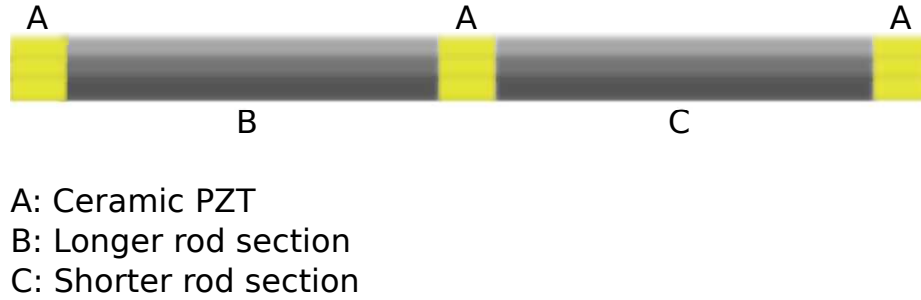
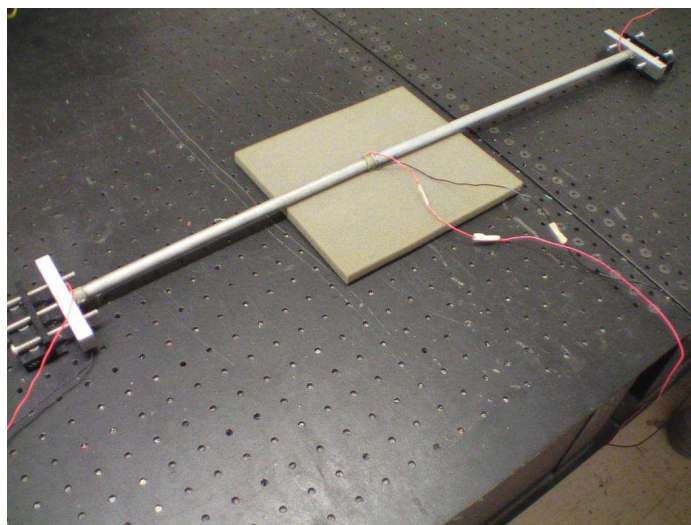


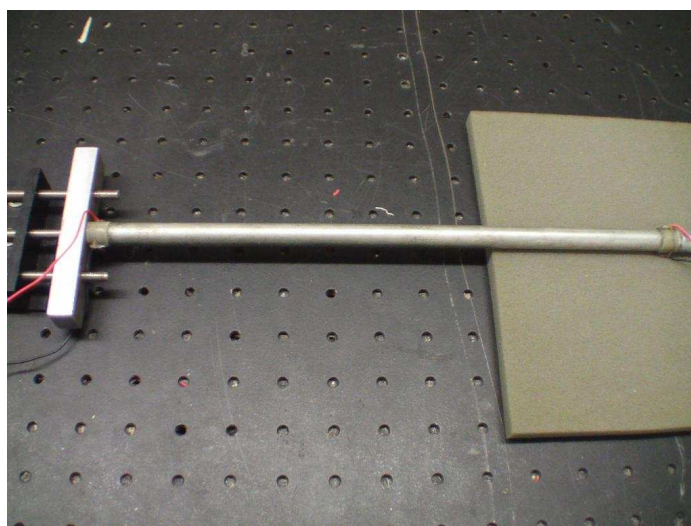
Figure 3.15: Schematic view of the time reversal setup. The PZTs had a length of  $12.0 \times 10^{-3}m$ . The middle PZT is referred to as the defect PZT. The two rod sections used in the tests were of different length and for generality this is just noted as 'longer' and 'shorter' section in the diagram. For the steel rods, the three lengths used were  $579.0 \times 10^{-3}m$ ,  $437.0 \times 10^{-3}m$ , and  $428.0 \times 10^{-3}m$  which gave a total of three combinations ( $579.0 \times 10^{-3}m$  and  $437.0 \times 10^{-3}m$ ,  $579.0 \times 10^{-3}m$  and  $428.0 \times 10^{-3}m$ ),  $437.0 \times 10^{-3}m$  and  $528.0 \times 10^{-3}m$ ). The lengths of the nylon rods used were  $148.0 \times 10^{-3}m$ ,  $105.0 \times 10^{-3}m$ , and  $75.0 \times 10^{-3}m$ . The lengths were chosen at random and were unknown to the program during testing.

The first part of the time reversal algorithm is the interrogatory phase. As with the crack detection tests, the host program generated an initial multi-tone wave that was to be played out by PZT A (although the algorithm was indifferent to which PZT plays the initial signal). Before playing out the wave the FPGA zeroed its output values and waited the specified period ( $200ms$  here). PZT A sent the wave and simultaneously began recording data. Both the defect PZT and PZT B began also began recording data as the wave was being played by PZT A. The signal propagated through the rod and towards PZT B. On its way to PZT B the wave struck the



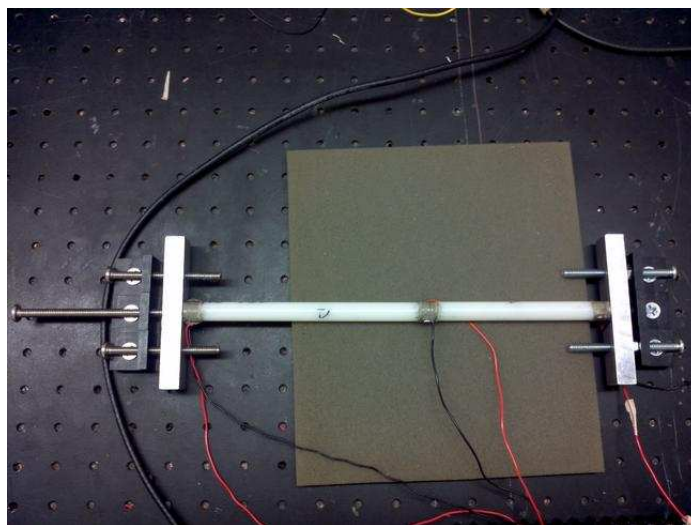


(a) Full view of steel rod time reversal tests

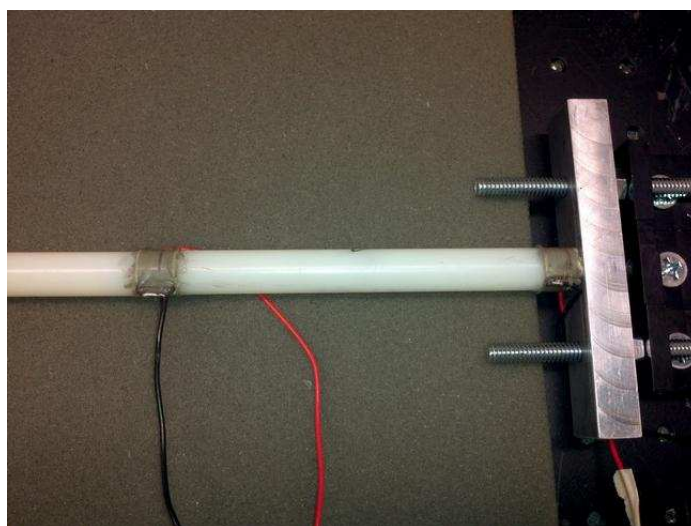


(b) Close up of steel rod time reversal tests

Figure 3.16: (a) One set of steel rods that were used for testing; (b) Closer view of the left side steel rod (shorter rod) section. The PZT on the right in (b) was the defect PZT and the PZT on the left in (b) was PZT B (not shown is PZT A).



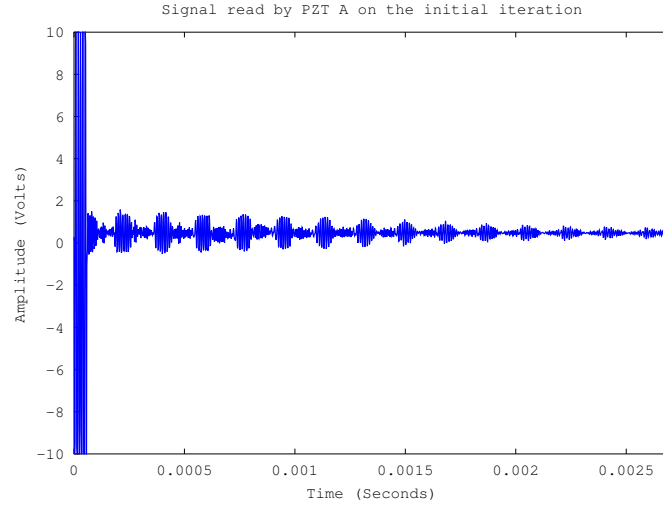
(a) Full view of steel rod time reversal tests



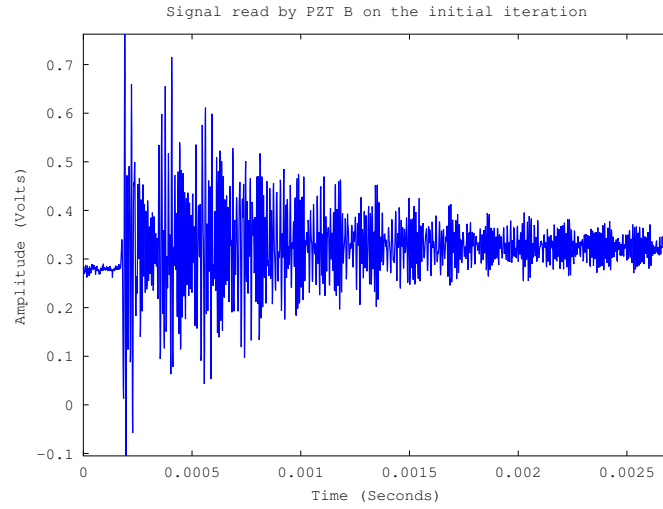
(b) Close up of steel rod time reversal tests

Figure 3.17: (a) One set of nylon rods that were used for testing; (b) Closer view of the right side steel rod (shorter rod) section. The PZT on the left in (b) was the defect PZT and the PZT on the right in (b) was PZT A (not shown is PZT B).

defect PZT. Upon impacting the defect PZT, the wave was split into multiple components part of which were reflected back towards PZT A and part which continued transmission towards PZT B. The reflected component struck PZT A where it was recorded. The transmitted component reached PZT B where it was also recorded. The components again split upon impacting the boundaries which caused more reflection and transmission components and were also recorded by PZT A and PZT B (Figure 3.18). The recording continued until the desired number of samples for the iteration was reached (2000 samples was chosen here based upon experimental observations).



(a) Response recorded at PZT A in the interrogatory phase



(b) Response recorded at PZT B in the interrogatory phase

Figure 3.18: (a) The response recorded at PZT A (sending PZT) during the interrogatory phase. These tests were carried out using steel rods which were  $437 \times 10^{-3}m$  and  $428 \times 10^{-3}m$  in length. It is seen that the initial wave played out by PZT A is included in the signal recorded by PZT A and is of substantially higher amplitude than the rest of the data recorded. This was one big difficulty with performing in one dimension. To make the algorithm work, it was necessary to zero out the section of the signal which contained the original wave and also to bypass ringing left over from the PZT playing the wave; (b) The response recorded at PZT B (opposite end) during the initial iteration.



## Chapter 4

### Results

## Chapter 5

### Recommended Future Work

## Chapter 6

## Conclusion

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