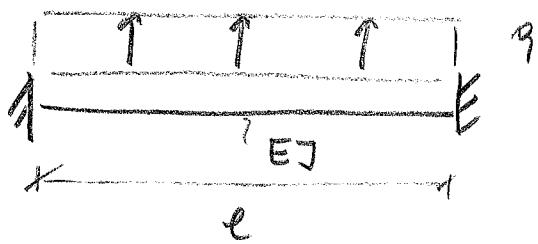


Exercise

Compare the mid-displacement obtained from:



- the exact solution
- the Ritz solution using 1, 2 and 3 trigonometric functions
- the Galerkin solution using 1 trigonometric function

1. Exact solution

$$\begin{cases} EI W_{xxxx} = q \\ W(0) = 0 \\ W_{xx}(0) = 0 \\ W(l) = 0 \\ W_{xx}(l) = 0 \end{cases} \quad \left| \begin{array}{l} \\ \\ \\ \text{essential conditions} \\ \end{array} \right.$$

The solution is given by

$$W = W^H + W^P$$

where:

$$W^P = \frac{1}{24} \frac{q}{EI} x^4$$

and

$$W^H = A_0 + A_1 x + A_2 x^2 + A_3 x^3$$

So:

$$W = A_0 + A_1 x + A_2 x^2 + A_3 x^3 + \frac{1}{24} \frac{q}{EI} x^4$$

Impose the boundary conditions:

$$W(0) = 0 \Rightarrow A_0 = 0$$

$$W_K(0) = 0 \Rightarrow A_1 = 0$$

$$W(l) = 0 \Rightarrow A_2 + A_3 l + \frac{9}{24EI} l^2 = 0$$

$$W_K(l) = 0 \Rightarrow 2A_2 + 3A_3 l + \frac{9}{6EI} l^2 = 0$$

which leads to:

$$A_3 = -\frac{9}{12EI} l ; \quad A_2 = \frac{9}{24EI} l^2$$

and so:

$$W = \frac{9l^4}{24EI} \left[\left(\frac{x}{l}\right)^2 - 2\left(\frac{x}{l}\right)^3 + \left(\frac{x}{l}\right)^4 \right]$$

$$W(l_2) = \frac{9l^4}{384EI}$$

2. Ritz solution - 1 term

The trial function is taken as:

$$\phi_1 = 1 - \cos \frac{2\pi x}{l}$$

and $W = C_1 \phi_1$

Note that the trial function satisfies the 4 essential conditions:

$$W(0) = W_x(0) = W(l) = W_{xx}(l) = 0$$

The stiffness matrix and the load vectors are:

$$K_{11} = \int_0^l \phi_{1xx} EJ \phi_{1xx} dx$$

$$f_1 = \int_0^l \phi_1 q dx$$

In this case

$$k_{11} = \frac{8\pi^4 EJ}{l^3}$$

$$f_1 = q l$$

The unknown amplitude is then:

$$C_1 = f_1/k_{11} = \frac{q l^4}{8\pi^4 EJ}$$

The displacement field is then

$$W = \frac{q l^4}{8\pi^4 EJ} \left(1 - \cos \frac{2\pi x}{l} \right)$$

and

$$W\left(\frac{l}{2}\right) = \frac{q l^4}{4 EJ \pi^4}$$

3. Ritz solution - 2 terms

$$\phi_1 = 1 - \cos \frac{2\pi x}{l} \quad \phi_2 = 1 - \cos \frac{4\pi x}{l}$$

The stiffness matrix reads:

$$k_{11} = \int_0^l \phi_{1/xx} EJ \phi_{1/xx} dx = \frac{8\pi^4 EJ}{l^3} \quad (\text{already available})$$

$$k_{12} = k_{21} = \int_0^l \phi_{1/xx} EJ \phi_{2/xx} dx = 0$$

$$k_{22} = \int_0^l \phi_{2/xx} EJ \phi_{2/xx} dx = \frac{128\pi^4 EJ}{l^3}$$

The load vector is:

$$f_1 = q l$$

$$f_2 = q l$$

The system to be solved is:

$$\begin{bmatrix} k_{11} & 0 \\ 0 & k_{22} \end{bmatrix} \begin{Bmatrix} c_1 \\ c_2 \end{Bmatrix} = \begin{Bmatrix} f_1 \\ f_2 \end{Bmatrix}$$

which leads to:

$$c_1 = \frac{q l^4}{8 E J \pi^4} \quad c_2 = \frac{q l^4}{128 E J \pi^4}$$

The displacement at the midspan is still

$$\omega(l/2) = \frac{9l^4}{4EI\pi^4}$$

which is not surprising as $\phi_2(l/2) = 0$

4. Ritz solution - 3 terms

$$\phi_1 = 1 - \cos \frac{2\pi x}{l}; \quad \phi_2 = 1 - \cos \frac{4\pi x}{l}; \quad \phi_3 = 1 - \cos \frac{6\pi x}{l}$$

Due to the orthogonality of the functions, the only terms to be computed are:

$$k_{33} = \int_0^l \phi_{3/x} EI \phi_{3/x} dx = \frac{648 EI \pi^4}{l^3}$$

$$f_3 = \int_0^l \phi_3 q dx = ql$$

The final system is:

$$\begin{bmatrix} k_{11} & & \\ & k_{22} & \\ & & k_{33} \end{bmatrix} \begin{Bmatrix} c_1 \\ c_2 \\ c_3 \end{Bmatrix} = \begin{Bmatrix} f_1 \\ f_2 \\ f_3 \end{Bmatrix}$$

leading to:

$$c_1 = \frac{ql^4}{8EI\pi^4} \quad c_2 = \frac{ql^4}{128EI\pi^4} \quad c_3 = \frac{ql^4}{648EI\pi^4}$$

The mid-span displacement is then:

$$w(l/2) = \frac{4l}{162} \frac{q l^4}{EI\pi^4}$$

5. Galerkin solution - 1-term

The governing equation reads:

$$EI w_{xxxx} = q, \text{ so:}$$

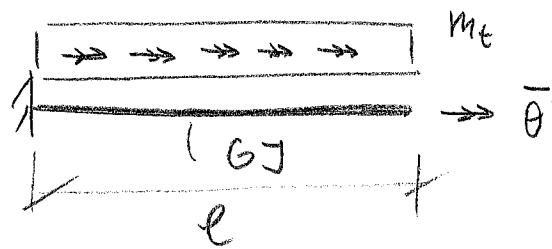
$$\int_0^l \phi_1 EI \phi_1_{xxxx} dx c_1 = \int_0^l \phi_1 q dx$$

which is:

$$\frac{8 EI \pi^4}{l^3} c_1 = q l$$

$$\Rightarrow c_1 = \frac{q l^4}{8 EI \pi^4} \quad (\text{same result obtained with Ritz})$$

Exercise



Obtain the exact displacement field and compare it with the 1-term solvish obtained using Ritz with polynomial functions

Exact solvish

$$\delta \Pi = \int_0^l \delta \theta_{xx} GJ \theta_{xx} dx - \int_0^l \delta \theta m_f dx$$

$$= - \int_0^l \delta \theta GJ \theta_{xx} dx + \delta \theta GJ \theta_x \Big|_0^l - \int_0^l \delta \theta m_f dx$$

$$\begin{cases} GJ \theta_{xx} + m_f = 0 \\ GJ \theta_{xx}(0) = 0 \quad \text{or} \quad \boxed{\delta \theta(0) = 0} \\ GJ \theta_{xx}(l) = 0 \quad \text{or} \quad \boxed{\delta \theta(l) = 0} \end{cases}$$

The problem is then:

$\theta_{xx} = - \frac{m_f}{GJ}$
$\theta(0) = 0$
$\theta(l) = \bar{\theta}$

The solvish is found as:

$$\theta = \theta^H + \theta^P$$

with

$$\theta^k = -\frac{1}{2} \frac{m+}{GJ} x^2$$

and

$$\theta^H = A_0 + A_1 x$$

Imposing the boundary conditions:

$$\theta(0) = 0 \Rightarrow A_0 = 0$$

$$\theta(l) = \bar{\theta} \Rightarrow A_1 l - \frac{1}{2} \frac{m+}{GJ} l^2 = \bar{\theta} \Rightarrow A_1 = \frac{\bar{\theta}}{l} + \frac{1}{2} \frac{m+}{GJ} l$$

The displacement field is then:

$$\theta(x) = \frac{1}{2} \frac{m+}{GJ} l^2 \left(\frac{x}{l} - \frac{x^2}{l^2} \right) + \frac{\bar{\theta}}{l} x$$

Ritz solution - 1 term

In this case the problem is characterized by two essential conditions. However, one of the two conditions is non-homogeneous.

The expansion is then taken as:

$$\theta = \sum_{i=1}^N c_i \phi_i + \phi_0$$

where

$\phi_i \rightarrow$ satisfy the homogeneous part of the essential conditions

$\phi_o \rightarrow$ satisfies the non-homogeneous essential boundary conditions.

More specifically:

$$\begin{aligned}\phi_i(0) &= 0 & \phi_o(0) &= 0 \\ \phi_i(l) &= 0 & \text{and} & \\ & & \phi_o(l) &= \bar{\theta}\end{aligned}$$

The simplest choice for ϕ_o is a linear polynomial

$$\phi_o(x) = \frac{\bar{\theta}}{l} x$$

The functions ϕ_i should be taken as a second order polynomial, so that the two conditions can be imposed:

$$\theta^H = \alpha_0 + \alpha_1 x + \alpha_2 x^2$$

$$\theta(0) = 0 \Rightarrow \alpha_0 = 0$$

$$\theta(l) = 0 \Rightarrow \alpha_1 = -\alpha_2 l$$

$$\Rightarrow \theta^H = \alpha_2 (x^2 - lx)$$

The Ritz functions are then taken as:

$$\boxed{\theta = c_1 (x^2 - lx) + \frac{\bar{\theta}}{l} x}$$

Due to the fact that the exact solution is quadratic, it is clear that the assumed Ritz expansion will lead to the exact solution. This is easily verified:

$$\int_0^l \delta \theta_{1x} G J \theta_{1x} dx - \int_0^l \delta \theta m_+ dx = 0$$

but

$$\theta = c_1 \phi_1 + \phi_0 \quad \text{and} \quad \delta \theta = \delta c_1 \phi_1 \quad \text{so:}$$

$$\delta c_1 \int_0^l \phi_{1xx} G J (\phi_{1xx} c_1 + \phi_{0xx}) dx = \delta c_1 \int_0^l \phi_1 m_+ dx$$

or

$$\underbrace{\delta c_1 \int_0^l \phi_{1xx} G J \phi_{1xx} dx}_{K_{11}} c_1 = \underbrace{\delta c_1 \int_0^l (\phi_1 m_+ - \phi_{1xx} G J \phi_{0xx}) dx}_{f_1}$$

where

$$K_{11} = G J \frac{l^3}{3} \quad \text{and} \quad f_1 = - \frac{m_+ l^3}{6}$$

and so:

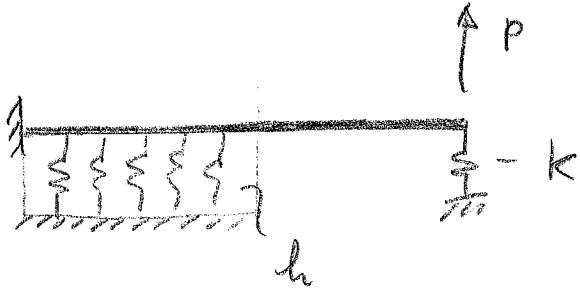
$$c_1 = - \frac{m_+}{2 G J}$$

leading to:

$$\theta = c_1 \phi_1 + \phi_0 = \frac{1}{2} \frac{m_+ l^2}{G J} \left(\frac{x}{l} - \frac{x^2}{l^2} \right) + \bar{\theta} \frac{x}{l}$$

as obtained previously

Exercise



Essential conditions:

$$\begin{cases} w(0) = 0 \\ w_{xx}(0) = 0 \end{cases} \Rightarrow \begin{cases} \phi_i(0) = 0 \\ \phi_{ix}(0) = 0 \end{cases}$$

The trial functions are taken as:

$$\phi_i = x^{i+1} \quad i = 1, \dots, N$$

Considering one single term:

$$\phi_1 = x^2$$

The PRW reads:

$$\int_0^l \delta w_{xx} E J w_{xx} dx + \int_0^{l/2} f w h w dx + \delta w(l) k w(l) \\ = \delta w(l) P$$

Upon substitution of the Ritz expansion it is obtained:

$$\delta c_1 \left[\int_0^l \phi_{1xx} E J \phi_{1xx} dx + \int_0^{l/2} \phi_1 h \phi_1 dx + \dot{\phi}_1(l) K \phi_1(l) \right] c_1 \\ = \delta c_1 \dot{\phi}_1(l) P$$

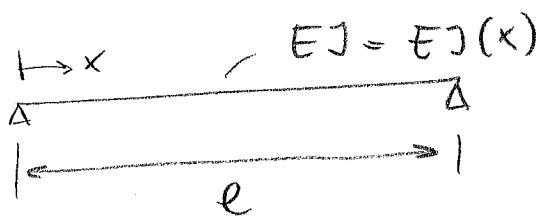
which leads to:

$$\left(4EJl + \frac{hl^3}{160} + l^4 K \right) c_1 = Pl^2$$

from which:

$$c_1 = \frac{160 Pl}{640 EJ + hl^4 + 160 l^3 K}$$

Exercise



Determine the first bending natural frequency using 1 trigonometric function and the method of Ritz

$$EJ = EJ_0 \sin \frac{\pi x}{l}$$

Ritz solution - 1 term

The essential conditions are:

$$w(0) = 0$$

$$w(l) = 0$$

The Ritz approximation is then taken as:

$$w = C_1 \sin \frac{\pi x}{l}$$

To evaluate the free vibrations, the external work is written as:

$$\begin{aligned} \delta W_e &= - \int_V (\delta u \rho \ddot{u} + \delta w \rho \ddot{w}) dV \\ &= - \int_V [(\delta u_0 - z \delta w_{0/x}) \rho (\ddot{u}_0 - z \ddot{w}_{0/x}) + \delta w_0 \rho \ddot{w}_0] dV \\ &= - \int_V [\delta u_0 (\rho \ddot{u}_0 - z \rho \ddot{w}_{0/x}) - \delta w_{0/x} (\rho z \ddot{u}_0 + \rho z^2 \ddot{w}_{0/x}) \\ &\quad + \delta w_0 \rho \ddot{w}_0] dV \\ &= - \int_0^l [\delta u_0 (I_0 \ddot{u}_0 - I_1 \ddot{w}_{0/x}) - \delta w_{0/x} (I_1 \ddot{u}_0 - I_2 \ddot{w}_{0/x}) \\ &\quad + \delta w_0 \rho \ddot{w}_0] dV \end{aligned}$$

The term I_1 is zero; the rotary inertia I_2 is negligible so:

$$\delta W_e = - \int_0^l (f u_0 I_0 \ddot{u}_0 + f w_0 I_0 \ddot{w}_0) dx$$

The first contribution is stabilized with the axial vibrations. The second one contributes to the bending vibrations. So.

$$\delta W_i = \delta W_e \text{ is:}$$

$$\int_0^l f w_{1xx} T \psi_{1xx} dx + \int_0^l f w_0 I_0 \ddot{w}_0 dx = 0$$

$$w(x, t) = c_1 e^{i\omega t} \sin \frac{\pi x}{l} = c_1 \phi_1 e^{i\omega t}$$

So:

$$\underbrace{\left[f c_1 \int_0^l \phi_{1xx} T \psi_{1xx} dx \right]}_{K_{11}} c_1 - \underbrace{\left[f c_1 \omega^2 \int_0^l \phi_1 I_0 \phi_1 dx \right]}_{M_{11}} c_1 e^{i\omega t} = 0$$

The problem is then in the form

$$(K_{11} - \omega^2 M_{11}) c_1 = 0$$

where:

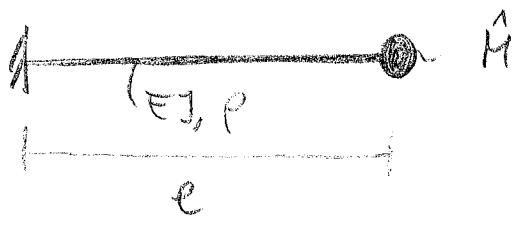
$$K_{11} = \frac{4\pi^3}{3l^3} EI_0 \quad M_{11} = I_0 \frac{l}{2}$$

The first circular frequency is then obtained as:

$$\omega_1 = \sqrt{\frac{K_{11}}{M_{11}}} = \sqrt{\frac{EI_0}{I_0} \frac{8\pi^2}{3L^4}}$$

Exercise

Evaluate the first natural frequency considering a polynomial expansion.



Evaluate the possibility of considering one-dof approximations using static deflections.

Weak form

$$\delta W_i = \int_0^l \delta w_{xx} E J w_{xx} dx$$

$$\delta W_e = - \int_0^l \delta w I_o \dot{w} dx - \delta w(l) \dot{M} \dot{w}(l)$$

Ritz functions

The simplest choice is to consider a polynomial expansion like:

$$w = c_1 x^e \Rightarrow \phi_1 = x^e$$

(the constant and linear contributions are not considered
for as they are not respectful of the boundary conditions
in $x=0$)

$$\begin{cases} w(0) = 0 \\ w_{xx}(0) = 0 \end{cases} \leftarrow \text{essential conditions}$$

Upon substitution in the weak form, it is obtained:

$$\delta C_1 \int_0^l \phi_{1/x} \epsilon J \phi_{1/x} dx C_1 = -\delta C_1 \int_0^l \phi_1 I_0 \phi_1 dx \ddot{C}_1 \\ - \delta C_1 \phi_1(l) \hat{M} \phi_1(l) \ddot{C}_1$$

or

$$\delta C_1 K_{11} C_1 = -\delta C_1 M \ddot{C}_1 + \delta C_1$$

where

$$K_{11} = \int_0^l \phi_{1/x} \epsilon J \phi_{1/x} dx = 4 \epsilon J l$$

$$M_{11} = \int_0^l \phi_1 I_0 \phi_1 dx + \phi_1(l) \hat{M} \phi_1(l) = l^4 \left(\frac{4}{3} I_0 l + \hat{M} \right)$$

The equation of motion is:

$$M_{11} \ddot{C}_1 + K_{11} C_1 = 0$$

and setting $C_1 = \dot{C}_1 e^{i\omega t}$ if it is

$$(-\omega^2 M_{11} + K_{11}) C_1 = 0$$

and so:

$$\omega^2 = \frac{k_{11}}{M_{11}}$$

Shape functions - 2 terms

Expand now up to the 3rd order:

$$W = C_1 x^2 + C_2 x^3 \Rightarrow \phi_1 = x^2; \phi_2 = x^3$$

The expression of the stiffness and mass matrices is still in the form:

$$K_{ij} = \int_0^l \phi_{i,xx} E J \phi_{j,xx} dx$$

$$M_{ij} = \int_0^l \phi_i I_o \phi_j dx + \phi_i(l) \vec{M} \phi_j(l)$$

The terms K_{11} and M_{11} are already available.

The novel contributions to be calculated are:

$$\begin{aligned} K_{12} &= \int_0^l \phi_{1,xx} E J \phi_{2,xx} dx \\ &\approx 6EJ l^2 \end{aligned}$$

$$\begin{aligned} M_{12} &= \int_0^l \phi_1 I_o \phi_2 dx + \phi_1(l) \vec{M} \phi_2(l) \\ &\approx l^5 \left(\frac{1}{6} I_o L + \vec{M} \right) \end{aligned}$$

and

$$\begin{aligned} K_{22} &= \int_0^l \phi_{2,xx} E J \phi_{2,xx} dx \\ &\approx 12EJ l^2 \end{aligned}$$

$$M_{22} = \int_0^l \phi_2 \mathbb{I}_0 \phi_2 dx + \phi_2(l) \hat{H} \phi_2(l)$$

$$= l^6 \left(\frac{1}{4} \mathbb{I}_0 l + \hat{H} \right)$$

Ritz functions - static deflections

The accuracy to dof ratio can be improved if the shape functions are properly selected.

While a polynomial expansion in the form

$\psi_i = c_i x^{i+1}$ with $i=1 \dots N$ certainly guarantees the orthogonality (the trial functions are a complete set of functions), it can be necessary to increase the number of dots in order to achieve the desired level of accuracy.

A good approach for obtaining a good solution with one single dof consists in selecting a static deflected shape. How is it done?

One possibility is to consider a cantilever beam with a concentrated force at the tip. It is easy to understand that this kind of shape would be appropriate when the tip mass M is much greater with respect to the mass associated with the beam itself, i.e.

$$M \gg I_c l$$

A second possibility would be to consider the deflected shape associated with a uniformly distributed load.

This case would represent the effect of distributed inertial forces whenever they are much greater with

respect to the inertial force due to the tip mass; in other words, this choice would be the most natural one whatever

$$I_0 l \gg M$$

Consider now the two cases:

1. $M \gg I_0 l$

The deflected shape is selected by solving the problem:



$$\left\{ \begin{array}{l} EJ W_{xxxx} = 0 \\ W(0) = 0 \\ W_{xx}(0) = 0 \\ W_{xxx}(l) = 0 \\ W_{xxxx}(l) = -\frac{P}{EY} \end{array} \right. \quad \left. \begin{array}{l} \text{essential conditions} \\ \text{natural conditions} \end{array} \right.$$

The solution of the problem is obtained after integrating the equilibrium equation and imposing the boundary conditions. It is obtained as:

$$W = \frac{1}{6EJ} \left(3x^2 l - x^3 \right)$$

The Ritz function is then taken as:

$$w = \varphi_1 \phi_1 \quad \text{with} \quad \phi_1 = 3x^2 l - x^3$$

Note that the static analysis allows to obtain the single dof expansion where both quadratic and cubic terms are accounted for. Their relative weight is indeed given by the solution of the static problem.

The stiffness and the mass matrix are now obtained as:

$$\begin{aligned} k_{11} &= \int_0^l k_{11x} EI \phi_{1xx} dx \\ &= 12 EI l^3 \end{aligned}$$

$$\begin{aligned} M_{11} &= \int_0^l \phi_1 I_0 \phi_1 dx + \phi_1(0) \vec{M} \phi_1(l) \\ &= l^6 \left(\frac{33}{35} I_0 l + 4 \vec{M} \right) \end{aligned}$$

and so:

$$\omega^2 = \frac{k_{11}}{M_{11}}$$

$$2. \quad J_0 l > M$$

The deflected pattern is now given by considering the following problem



$$\left\{ \begin{array}{l} EI w_{xxxx} = q \\ w(0) = 0 \\ w_x(0) = 0 \\ w_{xx}(0) = 0 \\ w_{xxx}(0) = 0 \end{array} \right. \quad \left. \begin{array}{l} \text{+ essential conditions} \\ \text{+ natural conditions} \end{array} \right.$$

The problem is solved as:

$$w = w^H + w^P$$

- particular integral
- homogeneous integral

$$\text{where } w^P = \frac{1}{24} \frac{q}{EI} x^4$$

$$w^H = A_0 + A_1 x + A_2 x^2 + A_3 x^3$$

Impose the boundary conditions it is:

$$w = \frac{1}{24} \frac{q}{EI} \left(x^4 - 4x^3 l + 6x^2 l^2 \right)$$

The single-dot Ritz representation is then

$$W = \sigma_i \phi_i \quad \text{with} \quad \phi_i = x^4 - 4x^3l + 6x^2l^2$$

where ϕ_i is now including the polynomial terms up to the fourth-order. Note that the pre-factors of the fourth-order contribution does not imply better accuracy with respect to the previous choice of the shape (that was including the quadratic and cubic terms only). Indeed the higher order contribution is related to the other one by pre-defined weighting factors (x^4 , x^3 and x^2 are not independent each other). These weighting factors are going to be a proper choice when $I_0 l \gg M$. On the contrary the weighting factors associated with the previous choice are going to be the best choice when $M \gg I_0 l$.

The stiffness and mass contributions are now obtained as:

$$k_{11} = \frac{144}{5} EJ l^5$$

$$M_{11} = \rho^8 \left(\frac{104}{45} I_0 l + 9M \right)$$

and so:

$$\omega_i = k_{11}/M_{11}$$

Comparison of results

The results obtained with the different approaches are now compared in terms of predicted circular frequencies ω . More specifically the comparison is conducted by expressing ω in the nondimensional form:

$$\bar{\omega} = \omega \sqrt{\frac{l^3 M}{EI}}$$

The dimensionlessization is introduced in order to make the results independent on the specific values chosen for M , l and EI .

Consider now two cases: one where $M \gg I_0 l$ and another one where $M \ll I_0 l$

1. Case 1: $M = 10 I_0 l$

(Consider $I_0 = 1$, $l = d$; $EI = 1$; $M = 10$)

	polynomial	tip load	uniform load
$N = 1$	1.9803	1.7120	1.7663
$N = 2$	1.7120		
$N = 3$	1.7120		
$N = 4$	1.7120		

Values of $\bar{\omega}$

estimate of the exact frequency value.

The convergence is from above as it is a ratio between the stiffness and the mass properties of the structure (recall, for the 1 dof case, $w = \sqrt{\frac{k}{m}}$): as the number of dof is increased, the stiffness of the structure gets smaller and, consequently, the value of the frequency reduces.

$$2. \text{ Case 2: } H = \frac{1}{1000} I_0 l$$

(Galerkin: $I_0 = 1$; $l = 1$; $Ef = 1$; $\mu = \frac{1}{1000}$)

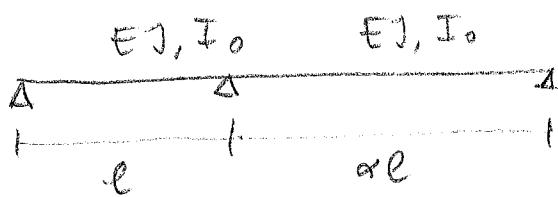
	polynomial	Tip load	Uniform
$N=1$	1.4107	1.1258	1.1141
$N=2$	1.1149		
$N=3$	1.1100		
$N=4$	1.1096		

Values of $\bar{w} \cdot 10$

Comments to the results

1. In the first case the choice of the tip load static deflection turns out to be the most efficient one. The one-dot approximation leads to better results with respect to the uniform load deflection as well as the polynomial one.
2. In the second case, as expected, the uniform load static deflection is the most appropriate one. The one-dot solution is more accurate with respect to the tip-load approximation and the polynomial expansions with $N=1$ and 2 .
3. The convergence of the solution is from above and the Ritz approximation is an upper bound.

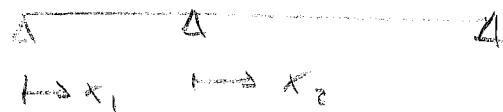
Exercise



Determine the first natural frequency by using a single-dof approximation

Weak form

Introduce two reference systems as illustrated in the figure



$$\begin{aligned} S\dot{W}_i &= \int_0^\Delta fW_{ix}^i EJ W_{ix}^i dx_1 + \int_0^\Delta fW_{xx}^i EJ W_{xx}^i dx_2 \\ S\dot{W}_e &= - \int_0^\Delta \{W^i I_0 \dot{W}^i\} dx_1 - \int_0^\Delta \{W^i I_0 \dot{W}^i\} dx_2 \end{aligned}$$

Ritz functions

The essential conditions of the problem are summarized here below

$$\begin{array}{|c|} \hline \Delta & \Delta & \Delta \\ \hline \left. \begin{array}{l} w^i(l) = w^i(0) = 0 \\ w'_{ix}(l) = w'_{ix}(0) \end{array} \right| \\ \hline \end{array}$$

$w^i(0) = 0$ $w^i(l) = 0$

A possible choice is given by the adoption of sine-type functions in the form

$$w^1 = C_1 \phi_1 \quad \text{with} \quad \phi_1 = \sin \frac{\pi x_1}{l}$$

$$w^2 = C_2 \phi_2 \quad \text{with} \quad \phi_2 = \sin \frac{\pi x_2}{l}$$

The two functions clearly satisfy the essential conditions at the two outer simply-supported conditions:

$$w^1(0) = C_1 \sin(0) = 0 \quad \checkmark$$

$$w^2(l) = C_2 \sin(\pi) = 0 \quad \checkmark$$

The inner condition regarding the transverse deflection is satisfied as well:

$$w^1(l) = w^2(0) = 0 \quad \checkmark$$

The essential condition on the rotation is not satisfied yet, and has to be imposed. In particular:

$$w_{xx}^1 = \frac{\pi}{l} \cos \frac{\pi x_1}{l} C_1$$

$$w_{xx}^2 = \frac{\pi}{l} \cos \frac{\pi x_2}{l} C_2$$

Imposing now:

$$w_{xx}^1(l) = w_{xx}^2(0) \quad , \text{ it is obtained}$$

$$\frac{\pi}{l} \cos(\pi) C_1 = \frac{\pi}{l} \cos(0) C_2$$

and so:

$$-\frac{\pi}{l} C_1 = \frac{\pi}{\alpha l} C_2$$

or

$$C_2 = -\alpha C_1$$

The one-dof representation of the deflected patterns is then:

$$W^1 = C_1 \sin \frac{\pi x_1}{l} \quad \text{and} \quad W^2 = -C_1 \alpha \sin \frac{\pi x_2}{\alpha l}$$

$$\text{or } \phi_1 = \sin \frac{\pi x_1}{l} \quad \text{and} \quad \phi_2 = -\alpha \sin \frac{\pi x_2}{\alpha l}$$

where C_1 is the degree of freedom associated with the expansion of ψ_1 and ψ_2 .

The stiffness matrix is obtained as:

$$k_{11} = \int_0^l \phi_{1xx} EI \phi_{1xx} dx_1 + \int_0^{\alpha l} \phi_{2xx} EI \phi_{2xx} dx_2 \\ = \frac{EI \pi^4 (\alpha+1)}{2 \alpha l^3}$$

whilst the mass matrix reads

$$M_{11} = \int_0^l \phi_1 I_0 \phi_1 dx_1 + \int_0^{\alpha l} \phi_2 I_0 \phi_2 dx_2 \\ = \frac{1}{2} I_0 l (1+\alpha^3)$$

and then:

$$\omega_1^2 = \frac{k_{11}}{M_{11}} = \frac{E J \pi^4}{(q^2 - \alpha + 1) I_0 l^4 \alpha}$$

When $\alpha = 1$ (the two beams have equal mass),
the square of the circular frequency is:

$$\omega_1^2 = \frac{E J \pi^4}{I_0 l^4}$$

Corresponding to the value of ω_1^2 obtained for
one single simply-supported beam