

Q(11)

$$\frac{7}{14} \quad DT = \frac{LHR}{4\pi K} = \frac{280}{4\pi \times 0.1} = 198.9 \quad \checkmark$$

$$\sigma^* = \frac{\alpha E (DT)}{4(1-\nu)} = \frac{8.2 \times 10^{-6} \times 290 \times 100 \times 198.9}{4(1-0.3)} \quad \checkmark$$

Now far with the crack go.

$$\sigma_{BB} = - \sigma^* (1 - 3\eta^2) = \sigma_{tra.}$$

$$\frac{120 \text{ MPa}}{\sigma^*} = -1 + 3\eta^2$$

$$\frac{120}{\sigma^*} + 1 = 3\left(\frac{r}{R}\right)^2 \quad \checkmark$$

We know σ^* from above. $R = 4.5 \text{ mm}$

$r =$ Crack length.

- on your way, but did not complete

max

$$\text{at } \eta = 1$$



$$\sigma_{\theta\theta} = -\sigma^* (1-3.)$$

$$\sigma_{\theta\theta} = 2\sigma^*$$



we know σ^* from previous calculations

$$\sigma_{\theta\theta} = \sigma_{\max} = 2\sigma^*$$

Q(2)

$\sigma = 50 \text{ MPa.}$ fuel rod.

$r_{\text{av}} = 5.4 \text{ mm}$

$t_{\text{clad}} = 1.2 \text{ mm.}$

1/14

a) $\sigma_{\theta} = \frac{PR}{\delta} = \frac{50 \text{ MPa} \times 5.4 \text{ mm}}{1.2 \text{ mm}} = 225 \text{ MPa.}$

$\sigma_z = \frac{PR}{2\delta} = \frac{50 \times 5.4 \text{ mm}}{2 \times 1.2 \text{ mm}} = 112.5 \text{ MPa.}$

$\sigma_r = \frac{-P}{2} = \frac{-50 \text{ MPa.}}{2} = -25 \text{ MPa.}$

b) $r = 5.6 \text{ mm}$

$\sigma_{rr} = \frac{-P \left[\left(\frac{R_o}{r} \right)^2 - 1 \right]}{\left(\frac{R_o}{R_i} \right)^2 - 1} = \frac{-50 \left(\frac{5.6}{5.4} \right)^2 - 1}{\left(\frac{5.6}{5.4} \right)^2 - 1} = -725.98 \text{ MPa.}$

$\sigma_{\theta\theta} = \frac{P \left[\left(\frac{R_o}{r} \right)^2 + 1 \right]}{\left(\frac{R_o}{R_i} \right)^2 - 1} = \frac{50 \left(\frac{5.6}{5.4} \right)^2 + 1}{\left(\frac{5.6}{5.4} \right)^2 - 1} = 725.98 \text{ MPa.}$

$\sigma_{zz} = \frac{P}{\left(\frac{R_o}{R_i} \right)^2 - 1} = \frac{50}{\left(\frac{5.6}{5.4} \right)^2 - 1} = 662.72 \text{ MPa}$

③

ϵ_{\max}

$$E = 180 \text{ GPa.}$$

$$\nu = 0.28$$

max strain is in the hoop direction.

$$\epsilon_{\theta\theta} = \frac{1}{E} (\sigma_{\theta\theta} - \nu(\sigma_{rr} + \sigma_{zz}))$$

$$\epsilon_{\theta\theta} = \frac{1}{180 \times 1000 \text{ MPa}} (726 \text{ MPa} - 0.28(-726 + 663))$$

$$\epsilon_{\theta\theta} = 4.1 \times 10^{-3}$$

$$\epsilon_{\theta\theta} = 0.413 \%$$

but with wrong
stresses

Q(3)

change in gap thickness. \rightarrow thermal expansion.

13/14

$$R_F = 0.52 \text{ cm.}$$

$$t_{\text{gap}} = 0.005 \text{ cm.}$$

$$T_{C0} = 550 \text{ K}$$

$$t_{\text{clad}} = 0.08 \text{ cm.}$$

$$k_{\text{fuel}} = 0.05 \text{ W/cm}\cdot\text{K}$$

$$k_{\text{gap}} = 0.003 \text{ W/cm}\cdot\text{K}$$

$$k_{\text{clad}} = 0.15 \text{ W/cm}\cdot\text{K}$$

$$LHR = 225 \text{ W/cm}$$

$$\alpha_c = 4.5 \times 10^{-6} \text{ K}^{-1}$$

$$\alpha_f = 15 \times 10^{-6} \text{ K}^{-1}$$

$$T_{\text{ref}} = 300 \text{ K}$$

Fuel	gap	clad.
R_F	t_{gap}	t_c
0.52	0.005	0.08
$k = 0.05$	$k = 0.003$	$k = 0.15$
α_f		α_c
LHR		higher conductivity.
		lower temp.

$T_{C0} = 550 \text{ K}$

$$\Delta \delta = R_c \alpha_c (T_{C0} - T_{\text{ref}}) - R_f \alpha_f (T_f - T_{\text{ref}})$$

change in gap

$$T_f = ?$$

$$R_c = 0.52 + 0.005 + 0.04$$

$$R_c = 0.565$$

Fuel.

gap

clad.

$$T_f - T_{Ci} = \frac{LHR}{2\pi R_F h_{\text{gap}}}$$

$$h_{\text{gap}} = \frac{k_{\text{gap}}}{t_{\text{gap}}}$$

$$h_{\text{gap}} = \frac{0.003}{0.005} = 0.6$$

$$T_f = \frac{LHR}{2\pi R_F \cdot 0.6} + T_{Ci}$$

$$T_f = \frac{225}{2\pi \times 0.52 \times 0.6} + 586.7$$

$$T_f = 701.5 \text{ K}$$

$$T_{Ci} - T_{C0} = \frac{LHR \cdot t_c}{2\pi R_F \cdot k_{\text{clad}}}$$

$$T_{Ci} = \frac{LHR \cdot t_c}{2\pi R_F \cdot k_{\text{clad}}} + T_{C0}$$

$$T_{Ci} = \frac{225 \times 0.08}{2\pi \times 0.52 \times 0.15} + 550 \text{ K}$$

$$T_{Ci} = 586.7 \text{ K}$$

$$T_0 = \frac{LHR}{4\pi k} + T_f$$

$$T_0 = \frac{225}{4\pi \times 0.05} + 701.5$$

$$T_0 = 1059.6 \text{ K}$$

$$\Delta S = R_c \alpha_c (T_c - T_{ref}) - R_p \alpha_p (T_p - T_{ref})$$

$$\Delta S = \left[0.565 \times 4.5 \times 10^{-6} \left(\frac{586 + 550}{2} - 300 \right) \right] - \left[0.52 \times 15 \times 10^{-6} \left(\frac{1060 + 1001}{2} - 300 \right) \right]$$

$$\Delta S = \boxed{}$$

almost there

Question (4)

fuel pellet.

8/12

grain size $D = 8 \mu\text{m}$. irradiated.

$$\phi = 2 \times 10^{13} \text{ f/cm}^3 \cdot \text{s}$$

$$D_{\text{diffusion}} = 2 \times 10^{-15} \text{ cm}^2/\text{s}$$

gas atoms/cm³ ? released.

after 2 years.

$$\gamma = 0.3017$$

using Booth model.

$$\tau = \frac{D \times t}{a^2} = \frac{2 \times 10^{-15} \times (2 \times 365 \times 24 \times 60 \times 60)}{(8 \times 10^{-4})^2}$$

$$\tau = 0.1971 > \pi^{-2}$$

$$f = 1 - \frac{6}{\pi^2} e^{-\pi^2 \frac{Dt}{a^2}} \rightarrow \text{this is annealing eqn}$$

→ we want in-pile eqn

$$f = 1 - \frac{6}{\pi^2} e^{-\pi^2 \frac{2 \times 10^{-15} \times (2 \times 365 \times 24 \times 60 \times 60)}{(8 \times 10^{-4})^2}}$$

$$f = 0.913$$

if we multiplied the fraction by the total gas production we will get the # of atoms escaped. ✓

total Gas production. = $\gamma \dot{F} \cdot t$ ✓

$N_t = 0.3017 \times 2 \times 365 \times 60 \times 60 \times \dot{F}$ we provide you \dot{F}

$\dot{F} = N_u^{235} \phi \sigma_F V$ we have the volumetric neutron flux.

$N_t = 0.3017 \times 2 \times 365 \times 60 \times 60 \times 2 \times 10^{13} = 1.68 \times 10^{13}$

gas atoms released

$= f \times N_t = 0.913 \times 1.68 \times 10^{13} = 1.5 \times 10^{13} \text{ atom.}$

Numbers might have some errors. !!!

Q(5)

→ strain hardening.

8/8

~~the increase in the stress~~ the region at which the stress relates to strain through $\sigma = k \epsilon^n$ where dislocations starts to multiply and interact with each other. ✓

→ what causes strain hardening.

- multiplication of dislocations. ✓
 - dislocations will be entangled and will impede the motion of each other. ✓
-

Q(6) stoichiometry of UO₂ impacts.

6/6

- melting temperature. ✓
- thermal conductivity. ✓
- process dependent diffusion ✓
 - grain growth.
 - fission gas release.
 - creep. ✓
- chemical reactions at inner cladding surface.
-

Q(7)

three things all fuel performance codes should be able to do.

→ Fuel → temp profile
volumetric change ✓

6/6

→ Cladding → temp profile. ✓
→ stress.

→ Gap → Gap heat transport.
→ mechanical interaction between fuel and clad. ✓
→ Gap pressure.

Q(8) fission gas release stages.

stage I : { gas atoms produced from fission. ✓
diffusion towards GBs ✓
intragranular bubbles formed. ✓

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Stage II → Gas bubbles nucleate → grow → Interconnect. ✓

Stage III → Gas travel through interconnected bubbles.
to the surface of the fuel. ✓

Q(9)

performance results from high burn up structure.

- instability of crystalline ~~material~~ structure. ✓
- restructuring driven by energy stored. ✓
- Pu production in the periphery and increased fissile density. ✓
- grains subdivide from $10 \mu\text{m}$ \longrightarrow $100 - 200 \text{ nm}$. ✓
- densely porous structures are formed. ✓
- material conductivity degrades due to pores. ✓
- fission gas is retained in the pores. ✓

6/6

Q(10)

0-D defect

- vacancy.
- interstitial.
- substitutional.

✓

4/4

3D defects.

- voids / bubbles.
- grain boundaries.
- precipitates / inclusions.

✓

Q (11) driving force for fuel densification.

⇒ change in free energy from the decrease in surface area of pores and lowering of surface free energy. ✓

* driving force for grain growth.

- reduction of grain boundary energy. ✓

↳ curvature driving force

because it drives the GB to be straight.

- temp gradient.

- elastic energy gradient. ✓

- dislocation energy gradients.

in General ⇒ you are trying to reduce Gibbs free energy.

Q (12) valence state of $UO_2 \rightarrow +4$ ✓

possible valence states of U. ✓

U^{4+} , U^{5+} , U^{6+} ✓

4/4