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Q(1)

$$\Delta T = \frac{\Delta HR}{4\pi k} = \frac{280}{4\pi \times 0.1} = 198.9$$

$$\sigma^* = \frac{\alpha E (\Delta T)}{4(1-\nu)} = \frac{8.2 \times 10^{-6} \times 290 \times 100 \times 198.9}{4(1-0.3)}$$

Now far with the crack go.

$$\sigma_{BB} = - \sigma^* (1 - 3\eta^2) = \sigma_{fract}$$

$$\frac{120 \text{ MPa}}{\sigma^*} = -1 + 3\eta^2$$

$$\frac{120}{\sigma^*} + 1 = 3 \left(\frac{r}{R} \right)^2$$

We know σ^* from above. $R = 4.5 \text{ mm}$

$r = \boxed{}$ Crack length.

max

$$\text{at } \eta = 1$$

$$\sigma_{\theta\theta} = -\sigma^* (1-3.)$$

$$\sigma_{\theta\theta} = 2\sigma^*$$

we know σ^* from previous calculations

$$\sigma_{\theta\theta} = \sigma_{\max} = 2\sigma^*$$

Q(2)

$\sigma = 50 \text{ MPa.}$ fuel rod.

$r_{\text{av}} = 5.4 \text{ mm}$

$t_{\text{clad}} = 1.2 \text{ mm.}$

a) $\sigma_{\theta} = \frac{PR}{\delta} = \frac{50 \text{ MPa} \times 5.4 \text{ mm}}{1.2 \text{ mm}} = 225 \text{ MPa.}$

$\sigma_z = \frac{PR}{2\delta} = \frac{50 \times 5.4 \text{ mm}}{2 \times 1.2 \text{ mm}} = 112.5 \text{ MPa.}$

$\sigma_r = \frac{-P}{2} = \frac{-50 \text{ MPa.}}{2} = -25 \text{ MPa.}$

b) $r = 5.6 \text{ mm}$

$\sigma_{rr} = \frac{-P \left(\left(\frac{R_o}{r} \right)^2 - 1 \right)}{\left(\left(\frac{R_o}{R_i} \right)^2 - 1 \right)} = \frac{-50 \left(\left(\frac{5.6}{5.4} \right)^2 - 1 \right)}{\left(\left(\frac{5.6}{5.4} \right)^2 - 1 \right)} = -725.98 \text{ MPa.}$

$\sigma_{\theta\theta} = \frac{P \left(\left(\frac{R_o}{r} \right)^2 + 1 \right)}{\left(\left(\frac{R_o}{R_i} \right)^2 - 1 \right)} = \frac{50 \left(\left(\frac{5.6}{5.4} \right)^2 + 1 \right)}{\left(\left(\frac{5.6}{5.4} \right)^2 - 1 \right)} = 725.98 \text{ MPa.}$

$\sigma_{zz} = \frac{P}{\left(\left(\frac{R_o}{R_i} \right)^2 - 1 \right)} = \frac{50}{\left(\left(\frac{5.6}{5.4} \right)^2 - 1 \right)} = 662.72 \text{ MPa}$

③

ϵ_{\max}

$$E = 180 \text{ GPa.}$$

$$\nu = 0.28$$

max strain is in the hoop direction.

$$\epsilon_{\theta\theta} = \frac{1}{E} (\sigma_{\theta\theta} - \nu(\sigma_{rr} + \sigma_{zz}))$$

$$\epsilon_{\theta\theta} = \frac{1}{180 \times 1000 \text{ MPa}} (726 \text{ MPa} - 0.28(-726 + 663))$$

$$\epsilon_{\theta\theta} = 4.1 \times 10^{-3}$$

$$\epsilon_{\theta\theta} = 0.413 \%$$

Q(3)

change in gap thickness. \rightarrow thermal expansion.

$$R_F = 0.52 \text{ cm.}$$

$$t_{\text{gap}} = 0.005 \text{ cm.}$$

$$T_{\text{co}} = 550 \text{ K}$$

$$t_{\text{clad}} = 0.08 \text{ cm.}$$

$$k_{\text{fuel}} = 0.05 \text{ W/cm}\cdot\text{K}$$

$$k_{\text{gap}} = 0.003 \text{ W/cm}\cdot\text{K}$$

$$k_{\text{clad}} = 0.15 \text{ W/cm}\cdot\text{K.}$$

$$\text{LHR} = 225 \text{ W/cm}$$

$$\alpha_c = 4.5 \times 10^{-6} \text{ K}^{-1}$$

$$\alpha_f = 15 \times 10^{-6} \text{ K}^{-1}$$

$$T_{\text{ref}} = 300 \text{ K}$$

Fuel	gap	clad.
R_F	t_{gap}	t_c
0.52	0.005	0.08
$k = 0.05$	$k = 0.003$	$k = 0.15$
α_f		\downarrow higher conductivity.
LHR		\downarrow lower temp.

$T_{\text{co}} = 550 \text{ K}$

$$\Delta \delta = R_c \alpha_c (T_c - T_{\text{ref}}) - R_f \alpha_f (T_f - T_{\text{ref}})$$

change in gap

$$T_f = ?$$

$$R_c = 0.52 + 0.005 + 0.04$$

$$R_c = 0.565$$

Fuel.

$$T_{\text{co}} = \frac{\text{LHR}}{4\pi R_F} + T_f$$

$$T_{\text{co}} = \frac{225}{4\pi \times 0.05} + \boxed{701.5 \text{ K}}$$

$$T_{\text{co}} = 1059.6 \text{ K}$$

$$T_f - T_{\text{ci}} = \frac{\text{LHR}}{2\pi R_F h_{\text{gap}}}$$

$$h_{\text{gap}} = \frac{k_{\text{gap}}}{t_{\text{gap}}}$$

$$h_{\text{gap}} = \frac{0.003}{0.005} = 0.6$$

$$T_f = \frac{\text{LHR}}{2\pi R_F \cdot 0.6} + T_{\text{ci}}$$

$$T_f = \frac{225}{2\pi \times 0.52 \times 0.6} + \boxed{586.7 \text{ K}}$$

$$T_f = \boxed{701.5 \text{ K}}$$

$$T_{\text{ci}} - T_{\text{co}} = \frac{\text{LHR} \cdot t_c}{2\pi R_F \cdot k_{\text{clad}}}$$

$$T_{\text{ci}} = \frac{\text{LHR} \cdot t_c}{2\pi R_F \cdot k_{\text{clad}}} + T_{\text{co}}$$

$$T_{\text{ci}} = \frac{225 \times 0.08}{2\pi \times 0.52 \times 0.15} + 550 \text{ K}$$

$$T_{\text{ci}} = \boxed{586.7 \text{ K}}$$

$$\Delta S = R_c \alpha_c (T_c - T_{ref}) - R_p \alpha_p (T_p - T_{ref})$$

$$\Delta S = \left[0.565 \times 4.5 \times 10^{-6} (550 - 300) \right] - \left[0.52 \times 15 \times 10^{-6} (1060 - 300) \right]$$

$$\Delta S = \boxed{}$$

Question (4)

Fuel pellet.

grain size $D = 8 \text{ } \mu\text{m}$. irradiated.

$$\phi = 2 \times 10^{13} \text{ f/cm}^2 \cdot \text{s}$$

$$D_{\text{diffusion}} = 2 \times 10^{-15} \text{ cm}^2/\text{s}$$

gas atoms/cm³ ? released.

after 2 years.

$$\gamma = 0.3017$$

using Booth model.

$$\tau = \frac{D \times t}{a^2} = \frac{2 \times 10^{-15} \times (2 \times 365 \times 24 \times 60 \times 60)}{(8 \times 10^{-4})^2}$$

$$\tau = 0.1971 > \pi^{-2}$$

$$f = 1 - \frac{6}{\pi^2} e^{-\pi^2 \frac{Dt}{a^2}}$$

$$f = 1 - \frac{6}{\pi^2} e^{-\pi^2 \frac{2 \times 10^{-15} \times (2 \times 365 \times 24 \times 60 \times 60)}{(8 \times 10^{-4})^2}}$$

$$f = 0.913$$

if we multiplied the fraction by the total gas production we will get the # of atoms escaped.

$$\text{total Gas production} = Y \dot{F} \cdot t$$

$$N_t = 0.3017 \times 2 \times 365 \times 60 \times 60 \times \dot{F}$$

$$\dot{F} = N_u^{235} \phi \sigma_F V \quad \text{we have the volumetric neutron flux.}$$

$$N_t = 0.3017 \times 2 \times 365 \times 60 \times 60 \times 2 \times 10^{13} = 1.68 \times 10^{13}$$

gas atoms released

$$= f \times N_t = 0.913 \times 1.68 \times 10^{13} = 1.5 \times 10^{13} \text{ atom.}$$

Numbers might have some ~~errors~~ errors. !!!

Q(5)

→ strain hardening.

~~the increase in the stress~~ the region at which the stress relates to strain through $\sigma = k \epsilon^n$ where dislocations starts to multiply and interact with each other.

→ what causes strain hardening.

- multiplication of dislocations.
 - dislocations will be entangled and will impede the motion of each other.
-

Q(6) stoichiometry of UO₂ impacts.

- melting temperature.
- thermal conductivity.
- process dependent diffusion
 - grain growth.
 - fission gas release.
 - creep.
- chemical reactions at inner cladding surface.
-

Q(7)

three things all fuel performance codes should be able to do.

→ Fuel → temp profile
volumetric change

→ Cladding → temp profile.
→ stress.

→ Gap → Gap heat transport.
→ mechanical interaction between fuel and clad.
→ Gap pressure.

Q(8) fission gas release stages.

stage I : { gas atoms produced from fission.
diffusion towards GBs
intragranular bubbles formed.

Stage II → Gas bubbles nucleate → grow → Interconnect.

Stage III → Gas travel through interconnected bubbles.
to the surface of the fuel.

Q(9)

performance results from high burn up structure.

- instability of crystalline material structure.
 - restructuring driven by energy stored.
 - Pu production in the periphery and increased fissile density.
 - grains subdivide from $10 \mu\text{m} \longrightarrow 100-200 \text{ nm}$.
 - densely porous structures are formed.
 - material conductivity degrades due to pores.
 - fission gas is retained in the pores.
-

Q(10)

0-D defect

- vacancy.
- interstitial.
- substitutional.

3D defects.

- voids / bubbles.
- grain boundaries.
- precipitates / inclusions.

Q (11) driving force for fuel densification.

⇒ change in free energy from the decrease in surface area of pores and lowering of surface free energy.

* driving force for grain growth.

- reduction of grain boundary energy.

↳ curvature driving force
because it drives the GB to be straight.

- temp gradient.

- elastic energy gradient.

- dislocation energy gradients.

in General ⇒ you are trying to reduce Gibbs free energy.

Q (12) valence state of $UO_2 \rightarrow +4$

possible valence states of U.

U^{4+} , U^{5+} , U^{6+} .