## 1. )a.) oxide thickness?

$$t^{4}(1) = (C.C)_{e-7} \exp\left(\frac{11949}{Gas}\right)$$
  
 $t^{4} = 133.15 \text{ Jays}$ 

$$\Rightarrow \int_{\infty}^{\infty} \left( N_{\text{m}} \right) = 5.1 \exp\left( \frac{-550}{625} \right)$$

$$\Rightarrow S(Nm) = S^{m} + K_{L}(t-t^{m})$$

$$K_{L} = 7.48e-6 \exp\left(\frac{-12500}{625}\right)$$

$$K_{L} = 1.54e-14$$

$$S = 2.12 + 1.54e-14(400-132)$$

$$C_{H} = 2 + \delta \rho_{2ro_{3}} \left( \frac{M_{H}}{M_{o}} \right)$$

$$\left( X - \frac{\delta}{\rho_{BR}} \right) \rho_{2r}$$

$$\Rightarrow \int_{a}^{a} (Nm) = 5.1 \exp\left(\frac{-250}{625}\right) \qquad \qquad (4 = \frac{500 - \frac{5.12}{1.50}}{500 - \frac{5.12}{1.50}})(0.5) \times 10^{\circ}$$

## 2.) Total fuel vol change?

Givens 
$$\Rightarrow \beta = \frac{ft}{N_0}$$
  
 $\prec_{kh} = \frac{1}{N_0} = \frac{1}$ 

$$T = 1200 \text{ K}$$
 $T = 1200 \text{ K}$ 
 $T = 300 \text{ K}$ 
 $R = 0.0105 \text{ FIMA}$ 

$$D_R = 0.01$$

$$B_0 = 5 \frac{m m^D}{K_0 u} = 0.01$$

$$\beta_{D} = 5 \frac{m D}{K_{9} U} \implies \epsilon_{4L} = \lambda_{4L} \Delta T$$

$$\beta_{UON} = 10.97 \% c \qquad \epsilon_{4L} = \lambda_{4L} \Delta T$$

$$\epsilon_{4L} = \lambda_{4L} \Delta T$$

$$\mathcal{E}_{4h} = 6.0099$$

$$\Rightarrow 6_{b} = D\rho_{0} \left[ \underbrace{P L_{n}(0.01)}_{C_{0}P_{0}} - 1 \right]$$

$$\Rightarrow \epsilon_{b} = \sum_{h=0}^{N} \underbrace{P L_{n}(0.01)}_{C_{0}P_{0}} - 1$$

$$\Rightarrow \epsilon_{b} = \sum_{h=0}^{N} \underbrace{P L_{n}(0.01)}_{C_{0}P_{0}} - 1$$

$$\epsilon_{tot} = 0.0099 - 6.0099 + 0.006$$

$$\epsilon_{tot} = 0.0069$$

$$\epsilon_{tot} = 0.0069$$

$$E_D = 0.01 \left[ exp \frac{0.0105 L_n(0.01)}{1(0.0053)} - 1 \right]$$

TK

$$\Rightarrow 6_{SFP} = 5.577e - 2pP$$

$$= (5.577e - 2)(10.97) (6.0105)$$

$$E_{SFP} = 0.0067$$

$$\Rightarrow E_{GFP} = 1.96e - 38pp(3800 - T) = exp(-0.0163(3800 - T))exp(-178pp)$$

$$= (1.96e - 38)(10.97)(0.0105)(3800 - 1300) = exp(-0.0163(3800 - 1300))exp(-178.)$$

$$exp(-0.0163(3800 - 1300))exp(-178.)$$

$$exp(-0.0163(3800 - 1300))exp(-178.)$$

$$exp(-0.0163(3800 - 1300))exp(-178.)$$

$$exp(-0.0163(3800 - 1300))exp(-178.)$$

## 3.) Ir Creep! >> G = (4.1ex) - (5.3e7) T (Pa) =) É tot = Éss + Éir Givens om = 200MPa G= 2.72e4 Em = 2.68e-58 + 2.19e-10 T = 600 K => Es = Ao (om ) enp (-Q xT) Etat = 2.19e-10 LHR = 150 W/cm t=1.5 yrs Ess = (4e-24) (200 ) Exp (-2.7et) (27) (-2.7et) Exot = Exot t n=5 A = 4e-24 5-1 Ers = 2.68 e-58 5-1 Q= 2.705 7/md E = (5.19e-10)(4.73e7) ⇒ Eir = Co I om Cs I = 3e11 LHR = 4.5e13 ( Exx = 0.0164 = 1.04% 6; = (2.74e-24)(4.5e13) (200) Eir=2.19e-105

- 4.) 1. Soluble Oxiber
  2. Insoluble Oxibes
  3. Metals
  4. Volatiles
  5. Noble Gases
- S.) Microstructure modeling:
  Whilzes material microstructure
  relationships that are functions of
  State variables and fuel conditions
  to determine material properties of
  fuel and cladding
  Has the potential to provide a
  more predictive fuel performance
  capability
  - (.) 1. Low neutron X-section
    2. Good thermal conductivity
    3. Affordable Cost

7.) Zr diffuser via Soret diffusion up the temperature gradient, and Zr also has different solubilities in U phases, leading to distinct zones of Zr content in radial rings

1/1

- 8.) MOX has different

  Neutronics, Fission gas release,
  thermal conductivity, etc.

  Designed to operate at much higher

  LHR
- 9.) 1. Corrosive environment

  3. Susceptible material

  3. Susticient stress

  4. Sufficient time

Process at PCI involves combination of high internal mechanical stress in the cladding and a corrosive environment resulting from volatile fission products accumulating in such-dal gap

- 10.) Can occur during LOCA transients
  Phase-Field modeling being used to
  account for effects of surface tension
  and you habite pressure
- 11.) RIA dependent yron control rod insertion/ejection LOCA - coalant flow reduced or lost altegether
- RIA example-Chernobyl
- 12) 1. Improved Fuel properties
  2 Improved Cladding Properties
  Cladding Coating/liner
  - 13.)1.PCMI a. Cladding Oxidation and Hydrogen Pickup