

Exam 1 - Nuc E 497

Question 1

-0, 30/30

$$K = 12.5 \text{ W/mK}$$

$$\rho_U = 7.5 \text{ g[U] / cm}^3$$

a) The fissile isotope is U-235.

It would be $\pm 0.7 \text{ wt\%}$.

$$N_A = 6.022 \times 10^{23} \text{ atoms/mol}$$

 x : m^g of atoms of U in the fuel

$$M_U \approx 238 \text{ amu}$$

$$b) N_{U235} = x \text{ g Na} \frac{\rho_U}{M_U}$$

$$U_3Si_2 \leadsto N_{U235} = 3 \times g \times N_A \times \frac{\rho_U}{M_U}$$

$$\rho_U = 11.31 \text{ g/cm}^3$$

$$N_{U235} = 3 \times 0.03 \times 6.022 \times 10^{23} \times \frac{11.31}{238}$$

$$N_{U235} = 2.5755 \times 10^{21}$$

Now we calculate g that will lead to a same N_{U235} :

$$U_3Si_5 \leadsto 2.5755 \times 10^{21} = 3 \times g \times 6.022 \times 10^{23} \times \frac{7.5}{238}$$

$$g = 0.0452 \leadsto \boxed{g = 4.52 \%}$$

c) It is most than U_3Si_2 because it has a lower density of fissile atoms and has a lower thermal conductivity.

Question 2

-4, 31/35

$$a) \quad T_{\infty} - T_{cool} = \frac{LHR}{2\pi R_f h_{cool}}$$

$$T_{ci} - T_{\infty} = \frac{LHR t_c}{2\pi R_f k_c}$$

$$T_s - T_{ci} = \frac{LHR}{2\pi R_f h_{gap}}$$

$$LHR = 250 \text{ W/cm}$$

$$R_f = 0.45 \text{ cm}$$

$$T_{cool} = 580 \text{ K}$$

$$Y = 0.05$$

$$h_{cool} = 2.5 \frac{\text{W}}{\text{cm}^2 \text{K}}$$

$$\delta_{gap} = 80 \times 10^{-4} \text{ cm}$$

$$\delta_{clad} = 0.06 \text{ cm}$$

$$h_{gap} = \frac{K_{He}^{1-Y} K_{Xe}^Y}{\delta_{gap}}$$

summing up and plugging the values:

$$\text{Using } K_{gas} = A \times 10^{-6} T^{0.79}$$

$$A = 16 \text{ He}$$

$$A = 0.7 \text{ Xe}$$

with the initial temperature T_{cool} ,

$$K_{He} = 0.0024$$

$$\text{W/cmK}$$

$$K_{Xe} = 1.06 \times 10^{-4} \text{ W/cmK}$$

$$T_s = 985.7 \text{ K}$$

-1, math error, $T_s = 958.2 \text{ K}$

$$b) \quad E = 246.7 \text{ GPa} \quad \nu = 0.25 \quad \alpha = 7.5 \times 10^{-6} \text{ } ^\circ\text{C}^{-1}$$

calculating fuel centerline T_0 :

$$T_0 = \frac{LHR}{4\pi k} + T_s = 1085.2 \text{ K}$$

$$K_{UN} = 0.2 \frac{\text{W}}{\text{cmK}}$$


Using the equations for thermal stress inside the fuel pellet and knowing that the hoop stress will be the larger:

$$\sigma_{\theta\theta}(\eta) = -\sigma^* (1 - 3\eta^2) \quad \therefore \sigma^* = \frac{\alpha E (T_0 - T_s)}{4(1-\nu)} \quad \therefore \eta = \frac{r}{R_s}$$

The largest value is when $\eta = 1$ ($r = R_s$) @ the surface

$$\sigma_{\theta\theta}(R_s) = + \frac{\alpha E \Delta T}{2(1-\nu)} = \boxed{122.7 \text{ MPa}}$$

c) $\alpha_{\text{UO}_2} = 1.2 \times 10^{-5} \text{ } 1/\text{K}$ while $E = 200 \text{ GPa}$

as the hoop stress is proportional to these two, it would be lower since $\alpha_{\text{UO}_2} < \alpha_{\text{UN}}$ and $E_{\text{UO}_2} < E_{\text{UN}}$. 

-3, It is higher, but primarily because k is smaller so the ΔT would be much higher

d)

- a)
- Steady-state
 - axisymmetric
 - T is constant in z
 - K is independent of T

- b)
- Static body
 - Neglect gravity
 - axisymmetric
 - small strains
 - solution does not change w/ z
 - Isotropic material response

Question 3

-4, 31/35

↳ Thin wall

- a)
- Static body
 - neglect gravity
 - axisymmetric
 - small strains
 - solution does not change with z
 - stress is constant through the wall of the cylinder
 - Isotropic material response

b) Thin wall solution:

$$\bar{\sigma}_\theta = \frac{PR}{\delta}$$

$$\bar{\sigma}_z = \frac{PR}{2\delta}$$

$$\bar{\sigma}_r = -\frac{P}{2}$$

$$P = 6 \text{ MPa}$$

$$R_i = 5.6 \text{ mm}$$

$$\delta = 0.6 \text{ mm}$$

$\bar{\sigma}_\theta = 56.0 \text{ MPa}$	$\bar{\sigma}_z = 28.0 \text{ MPa}$	$\bar{\sigma}_r = -3.0 \text{ MPa}$
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c) for thicker walls:

$r = R_i$

$r = R_o$

$$\sigma_{rr}(r) = -P \frac{\left(\frac{R_o}{r}\right)^2 - 1}{\left(\frac{R_o}{R_i}\right)^2 - 1} \quad \text{---} \quad -6 \text{ MPa} \quad \text{---} \quad 0 \text{ MPa}$$

$$\sigma_{\theta\theta}(r) = P \frac{\left(\frac{R_o}{r}\right)^2 + 1}{\left(\frac{R_o}{R_i}\right)^2 - 1} \quad \text{---} \quad 56.16 \text{ MPa} \quad \text{---} \quad 50.16 \text{ MPa}$$

$$\sigma_{zz} = P \frac{1}{\left(\frac{R_o}{R_i}\right)^2 - 1} \quad \text{---} \quad 25.08 \text{ MPa} \quad \text{---} \quad 25.08 \text{ MPa}$$

For this case, the thin wall approximation would work.

The highest hoop stress in the thick wall approximation is @ $r = R_i$ and it is not much higher than the thin wall value (56.16 compared to 56 MPa, 0.3% higher).

The hoop stress is the highest of all three terms and it decreases with r , so thin wall solution is safe to predict failure.

Conservative but hoop stress varies by more than 10% across thickness

d)

$$\epsilon_{rr} = \frac{1}{E} (\sigma_{rr} - \nu (\sigma_{\theta\theta} + \sigma_{zz}))$$

$$E = 70 \text{ GPa}$$

$$\nu = 0.41$$

$$\epsilon_{\theta\theta} = \frac{1}{E} (\sigma_{\theta\theta} - \nu (\sigma_{rr} + \sigma_{zz}))$$

$$\epsilon_{zz} = \frac{1}{E} (\sigma_{zz} - \nu (\sigma_{\theta\theta} + \sigma_{rr}))$$

$$\begin{bmatrix} \epsilon_{rr} \\ \epsilon_{\theta\theta} \\ \epsilon_{zz} \end{bmatrix} = \frac{1}{E} \begin{bmatrix} 1 & -\nu & -\nu \\ -\nu & 1 & -\nu \\ -\nu & -\nu & 1 \end{bmatrix} \begin{bmatrix} \sigma_{rr} \\ \sigma_{\theta\theta} \\ \sigma_{zz} \end{bmatrix}$$

↓
Strain
Tensor

↓
Stress
Tensor

plugging numerical values and using the stresses calculated in b) :

$$\epsilon_{rr} = -5.34 \times 10^{-4}$$

$$\epsilon_{\theta\theta} = 6.53 \times 10^{-4}$$

$$\epsilon_{zz} = 8.95 \times 10^{-5}$$

-4, Didn't write stress and strain in tensor form

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