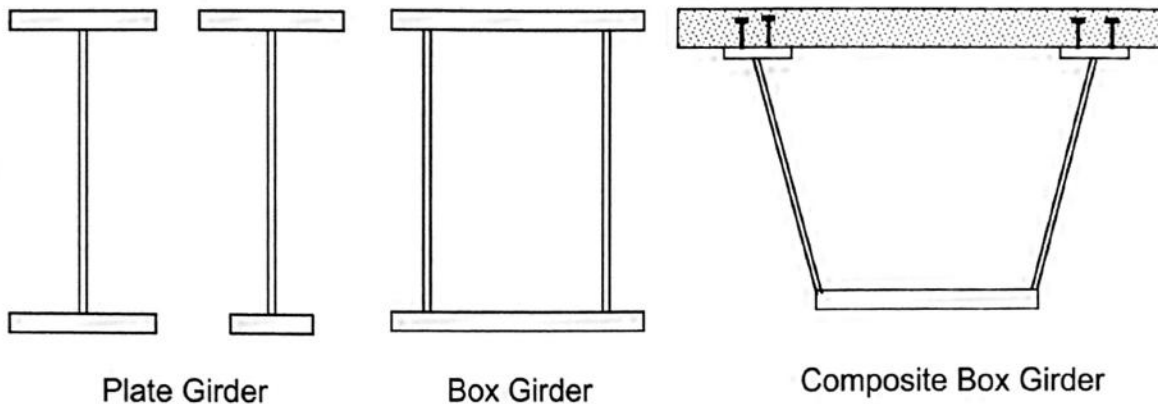


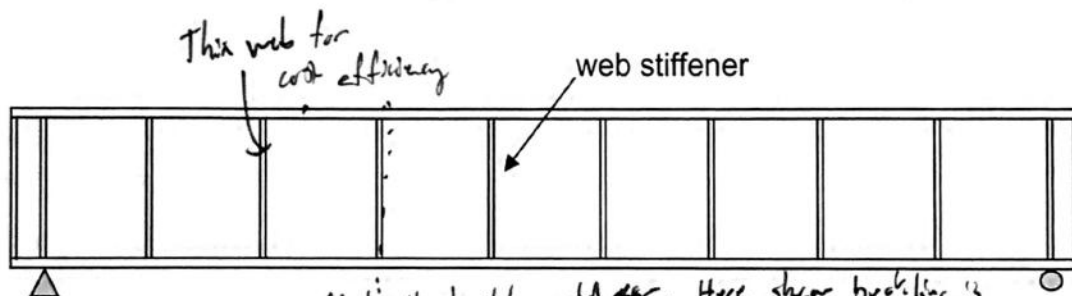
Chapter 5. Plate Girders

Plate girders are custom fabricated beam members utilised to carry large loads over long spans. They are used in buildings and industrial structures for long span floor girders to create large open space. Plate girders are very commonly used in bridge construction, too.



$$M = \frac{wL^2}{8}$$

$$F = M/d$$

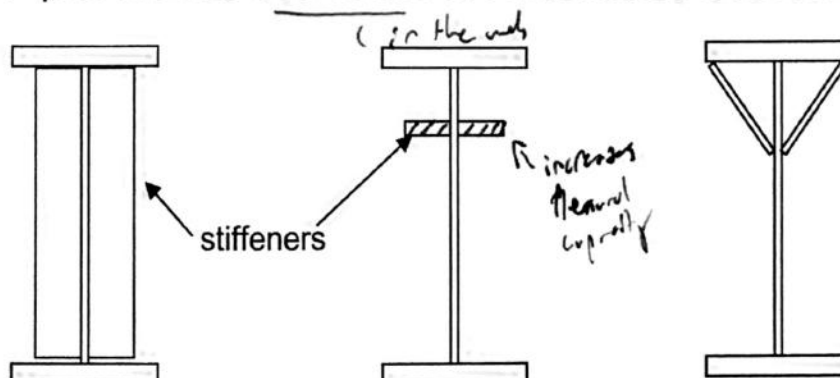


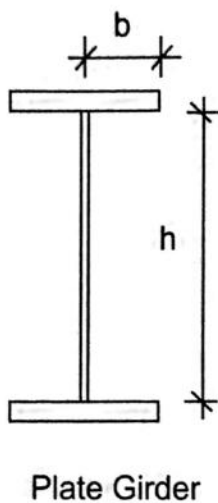
It dim 4, buckling would occur. Here, shear buckling is dominant. Stiffeners are used

A plate girder is generally defined as a built-up flexural member having a slender web. Special attention should be paid on the presence of slender web when designing a plate girder. Web stiffeners are generally required.

Web Stiffeners: functions

1. ♦ prevent buckling due to compression from bending and shear
2. ♦ promote tension field action to increase shear strength
3. ♦ prevent web local failure at concentrated load locations



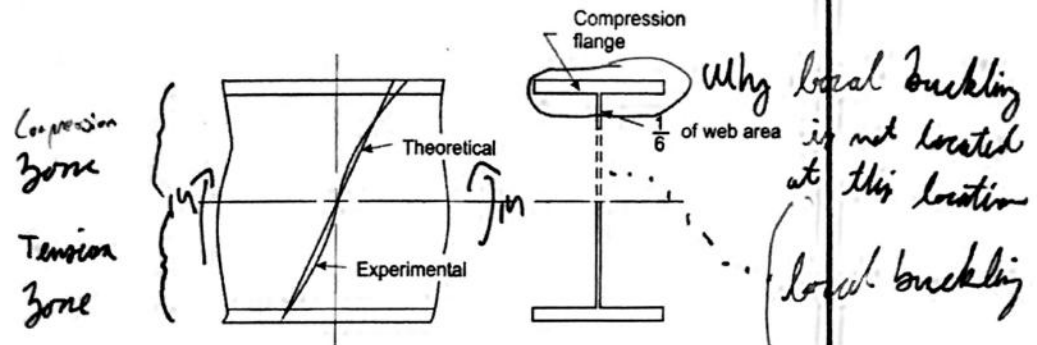


Normally, for a plate girder,

Flange: $b/t \leq 200/\sqrt{F_y}$ (at least Class 3)

Web: $h/w > 1900/\sqrt{F_y}$ (usually Class 4)

It is economical to use thin web. However, it leads to many problems which must be addressed in design.



Effect of Thin Webs on Moment Resistance

- ♦ Web is thin and not perfectly flat, therefore, it tends to buckle or deflect.
- ♦ Part of the longitudinal shortening of the compression zone is associated with the above geometry changes and only the remainder of the shortening is associated with theoretical strain/stress relationship.
- ♦ "Soft web": the web will be less effective than expected and the flange will receive the higher stress.

Moment resistance reduction is applied since local buckling of web decreases the moment resistance. It is a linear reduction that is a function of web slenderness, the ratio of the area of flange to the area of web, and buckling load of the web.

$$M'_r = M_r \left[1.0 - 0.0005 \frac{A_w}{A_f} \left(\frac{h}{w} - \frac{1900}{\sqrt{F_y}} \right) \right]$$

Must be greater than zero

$$M'_r = M_r \left[1.0 - 0.0005 \frac{A_w}{A_f} \left(\frac{h}{w} - \frac{1900}{\sqrt{M_r/\phi S}} \right) \right]$$

2014

A_w = web area; A_f = compression flange area; S = elastic section modulus

M_r ($M_r \leq \phi M_y$) is the factored moment resistance based on Cl. 13.5 or 13.6.

No reduction in moment resistance if

$$h/w \leq 1900/\sqrt{F_y}$$

14.3.4
flange provides stiffness

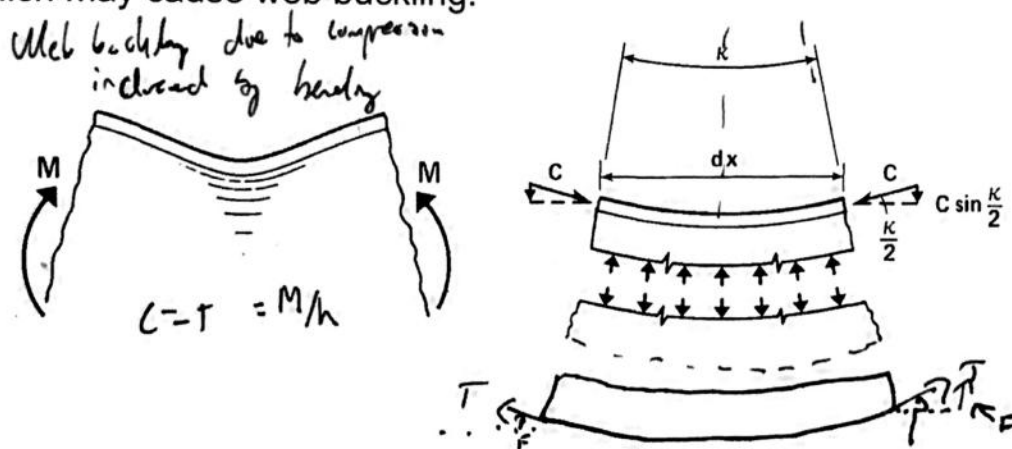
With the presence of axial force C_r in addition to the bending, the resistance moment for a plate girder is

$$M'_r = M_r \left[1.0 - 0.0005 \frac{A_w}{A_f} \left(\frac{h}{w} - \left(1 - 0.65 \frac{C_f}{\phi C_y} \right) \frac{1900}{\sqrt{F_y}} \right) \right]$$

$\frac{C_f}{\phi C_y} \leq 1$
 $C_y = F_y A$

Vertical Buckling of Web

As shown in below, the curved (deformed) flange exerts vertical pressure on which may cause web buckling.



where, κ is the bending curvature. The similar situation exists on tension flange, however, buckling will not occur on tension side.

Applied force of the web F :

The vertical force transmitted to web from compressive flange over the length dx is

$$F = 2 C \sin (\kappa/2) \approx 2 C \kappa/2 = C \kappa = A_f \sigma_f \kappa$$

where,

$$\therefore \kappa = \frac{\epsilon_f dx}{h/2} \quad \therefore F = 2 A_f \sigma_f \epsilon_f dx / h$$

A_f and σ_f is the area and normal stress of compressive flange, respectively.

Resisting force of the web R :

Based on elastic buckling, the vertical resisting force over the length dx is

$$R = \sigma_{cr} A$$

where,

A is the web area in compression over the length dx , $A = w dx$, and σ_{cr} is the elastic buckling stress for web plate:

Plate eqn $\sigma_{cr} = \frac{\pi^2 E}{(1 - \mu^2) \left(\frac{KL}{r} \right)^2} = \frac{k \pi^2 E}{(1 - \mu^2) \left(\frac{L}{r} \right)^2}$; where, $k = \frac{1}{K^2}$ based on conditions of the

$$\therefore r = \sqrt{\frac{I}{A}} = \sqrt{\frac{\frac{1}{12} w^3 dx}{w dx}} = \frac{w}{\sqrt{12}} \quad \text{and} \quad L = h$$

$$\therefore \sigma_{cr} = \frac{k \pi^2 E}{12(1 - \mu^2) \left(\frac{h}{w} \right)^2}$$

Let $k = 1$ (conservative), therefore,

$$R = \sigma_{cr} A = \frac{\pi^2 E}{12(1 - \mu^2) \left(\frac{h}{w} \right)^2} w dx \quad \uparrow A$$

Considering the equilibrium between applied and resisting forces,

$$\frac{2A_f \sigma_f \epsilon_f dx}{h} = \frac{\pi^2 E}{12(1 - \mu^2) \left(\frac{h}{w} \right)^2} w dx$$

$$F = R$$

solve for h/w and note that $A_w = hw$, therefore

$$\frac{h}{w} = \sqrt{\frac{\pi^2 E A_w}{24(1 - \mu^2) A_f \sigma_f \epsilon_f}}$$

By considering,

- ♦ Residual stress:

$$\epsilon_f = (\sigma_y + \sigma_r) / E = \frac{F_y + 15}{8} \cdot \frac{1}{E}$$

- ♦ $A_f / A_w = 0.5$ (conservative)

- ♦ Poisson's ratio $\mu = 0.3$ and Young's modulus $E = 2 \times 10^5$ MPa

then, $h/w = \frac{83.00}{F_y}$ *SI Unit Only*

14.3.1

This is the maximum permissible web slenderness h/w . F_y is associated compression flange steel. This limit may be waived if analysis indicates that buckling of the compression flange into the web will not occur at factored load levels.

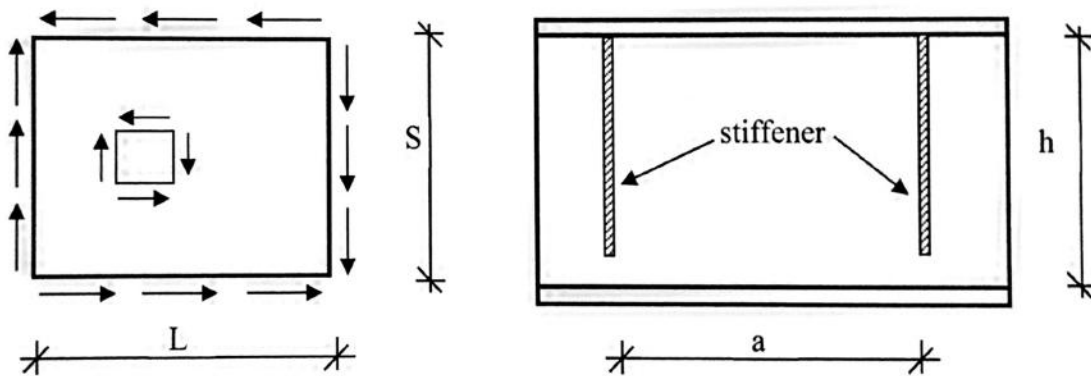
$$F_y = 300 \text{ MPa}, h/w = 276.6; \text{ if } w = 25 \text{ mm}, h = 6915 \text{ mm}$$

$$F_y = 350 \text{ MPa}, h/w = 237.1; \text{ if } w = 25 \text{ mm}, h = 5925 \text{ mm}$$

13.4.1.1

Web Buckling under Shear

Although bending can be presented unaccompanied by shear, a transversely loaded plate girder will always have shear combined with moment. The thin web of the plate girder could buckle under the shear stress.



S = Smaller panel dimension; L = Larger panel dimension

$$\therefore \tau_{cr} = \frac{k\pi^2 E}{12(1-\mu^2)\left(\frac{S}{w}\right)^2}; \quad \text{and } k = 5.34 + \frac{4}{\left(\frac{L}{S}\right)^2}$$

$$\therefore \tau_{cr} = \left(5.34 + \frac{4}{\left(\frac{L}{S}\right)^2} \right) \frac{\pi^2 E}{12(1-\mu^2)\left(\frac{S}{w}\right)^2} = \left(\frac{5.34}{\left(\frac{S}{L}\right)^2} + 4 \right) \frac{\pi^2 E}{12(1-\mu^2)\left(\frac{L}{w}\right)^2}$$

Using plate girder nomenclature, a , h , F_{cre} , etc. The elastic critical plate buckling stress in shear

$$F_{cre} = \frac{k_v \pi^2 E}{12(1 - \mu^2) \left(\frac{h}{w}\right)^2}$$

note: Plate shear buckling coefficient

If h is smaller panel dimension, i.e. $a/h \geq 1$, then

$$k_v = 5.34 + \frac{4}{\left(\frac{a}{h}\right)^2}$$

web:

- hinged edge, $k_v = 5.34$

- fixed edge, $k_v = 8.94$

- fixed edge, if the compression flange of the girder is continuously contact with concrete slab

otherwise, if $a/h < 1$

$$k_v = \frac{5.34}{\left(\frac{a}{h}\right)^2} + 4$$

However, S16 doesn't acknowledge it

Poisson's ratio $\mu = 0.3$ and Young's modulus $E = 2 \times 10^5$ MPa

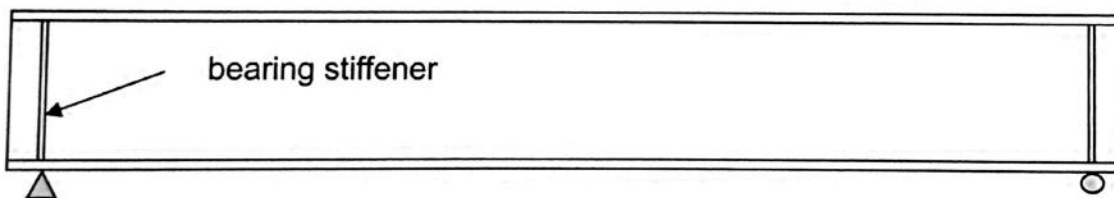
13.4.1.1(b)

$$F_{cre} = \frac{180000 k_v}{\left(\frac{h}{w}\right)^2}$$

SI unit only

F_{cre} is not directly related to F_y. This is similar to the case of column elastic buckling strength

Shear Buckling of Un-stiffened Web



Let $a/h = 20$, then $k_v = 5.34 + 4 / (a/h)^2 = 5.34 + 0.01$

It is almost as the same as $a/h = \infty$, therefore $k_v = 5.34$

Elastic critical buckling stress,

$$F_{cre} = \frac{180000 \times 5.34}{\left(\frac{h}{w}\right)^2} = \frac{961200}{\left(\frac{h}{w}\right)^2}$$

IF AND ($k_v = 83000 / F_y$), $F_y = 300 \text{ MPa}$

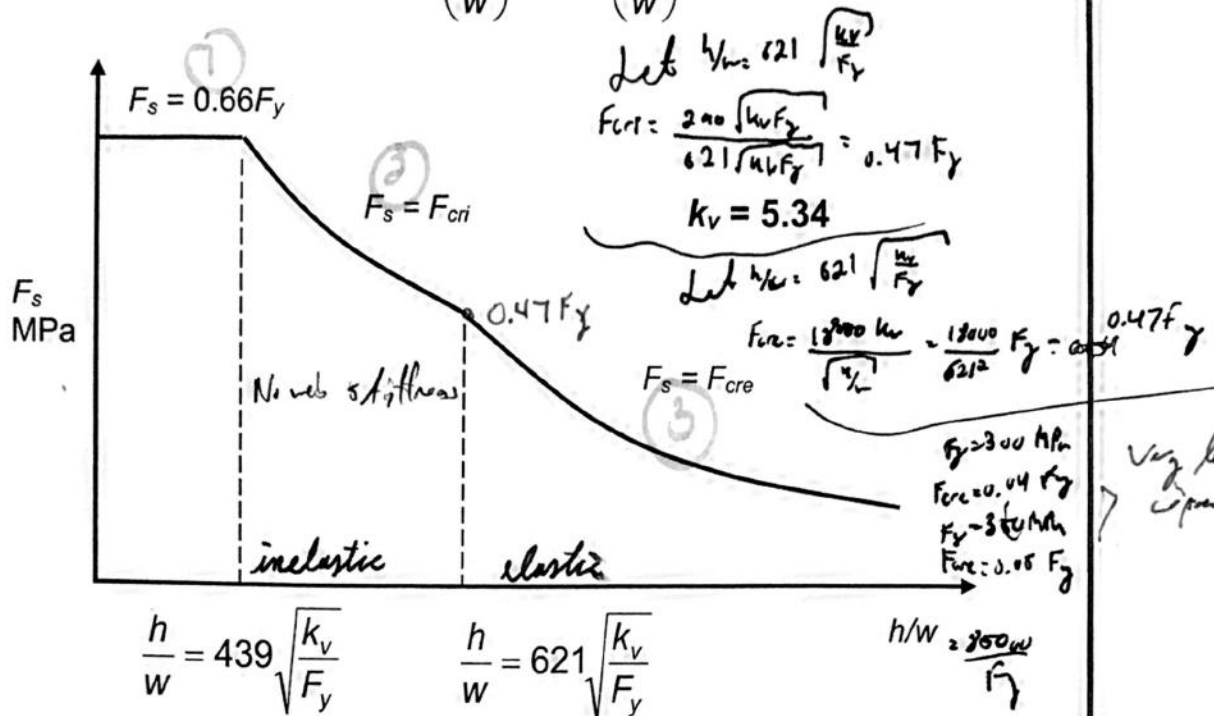
$F_{cre} = 12.1 \text{ MPa}$

IF $F_y = 300 \text{ MPa}$

$F_{cre} = 17 \text{ MPa}$

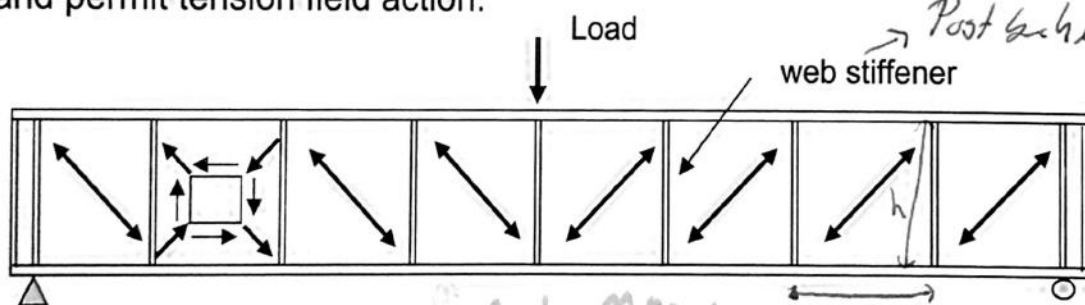
Inelastic critical buckling stress (residual stress, test results)

$$F_{cri} = \frac{290 \sqrt{k_v F_y}}{\left(\frac{h}{w}\right)} = \frac{670 F_y}{\left(\frac{h}{w}\right)}$$

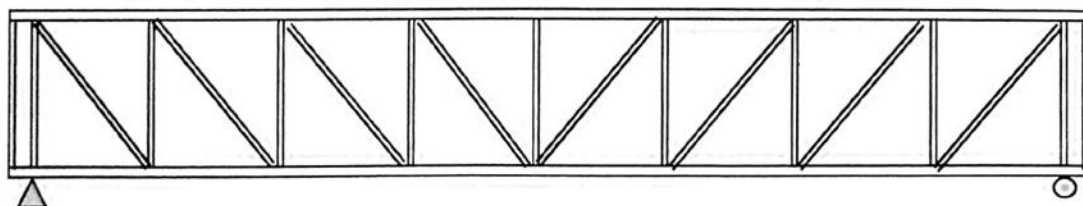


Shear Buckling of Stiffened Web

Tension Field Action: shear capacity > theoretical value? Post buckling
Web stiffeners improve shear strength of web – increase buckling stress and permit tension field action.



Similar to diagonal members of a Pratt Truss



V_b – shear capacity of web contributed by normal beam capacity (capacity of theoretical buckling)

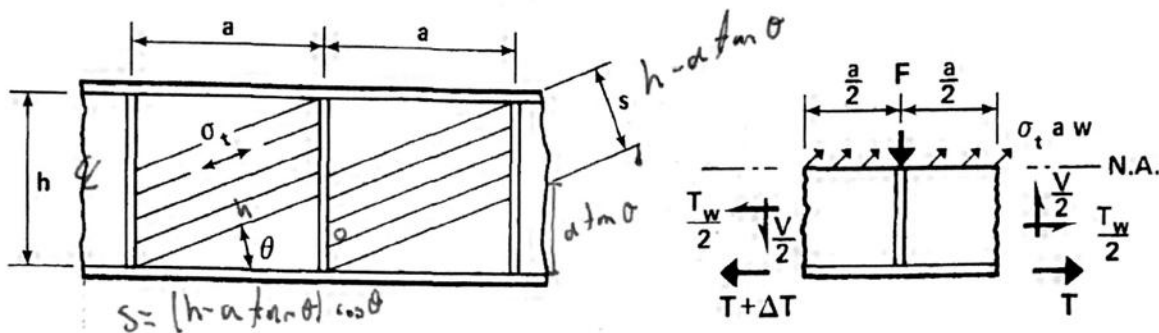
V_t – additional shear capacity of web due to tension field action

V_u – the ultimate shear capacity of web

$$V_u = V_b + V_t$$

$$V_b = \tau_{cr} h w$$

Post buckling strength (only develops after web buckles)
Ans: hw



$$s = h \cos \theta - a \sin \theta$$

$$\therefore V_t = \frac{\sigma_t w \sin \theta s}{\sigma_t \sin \theta (w - s)} = \sigma_t w \sin \theta (h \cos \theta - a \sin \theta)$$

to obtain maximum V_t ,

$$\text{Let } \frac{\partial V_t}{\partial \theta} = 0; \quad \Rightarrow \theta = \tan^{-1} \left(\sqrt{1 + (a/h)^2} - a/h \right)$$

$$V_t = \sigma_t w h \frac{1}{2 \sqrt{1 + (a/h)^2}}$$

Therefore, the ultimate shear strength is

$$V_u = V_b + V_t = \tau_{cr} h w + \sigma_t \frac{w h}{2 \sqrt{1 + (a/h)^2}}$$

By assuming the linear relationship between tension and shear for element subjected to pure shear or pure tension, with $\theta = 45^\circ$ (approximately), the yield condition of combining tension and shear is

$$\frac{\sigma_t}{\sigma_y} = 1 - \frac{\tau_{cr}}{\tau_y} \quad \sigma_t = \left(1 - \frac{\tau_{cr}}{\tau_y} \right) \sigma_y$$

Then, the ultimate shear strength can be rewritten as

$$V_u = \tau_{cr} hw + \frac{\sigma_y wh}{2\sqrt{1+(a/h)^2}} - \frac{\tau_{cr} \sigma_y}{\tau_y} \frac{wh}{2\sqrt{1+(a/h)^2}} \quad (A)$$

The critical plate buckling stress in shear

$$\tau_{cr} = \frac{k_v \pi^2 E}{12(1-\mu^2)(h/w)^2} \quad \frac{180760 k_v}{(h/w)^2}$$

If $a/h \geq 1$, then

$$k_v = 5.34 + \frac{4}{(a/h)^2} \quad \frac{4}{(1)^2} = 180760 SE$$

otherwise, if $a/h < 1$

Substituting following eq (A) $k_v = \frac{5.34}{(a/h)^2} + 4$

$$\mu = 0.3; \quad E = 2 \times 10^5 \text{ MPa}; \quad \tau_y = F_y / \sqrt{3} \text{ and } \sigma_y = F_y$$

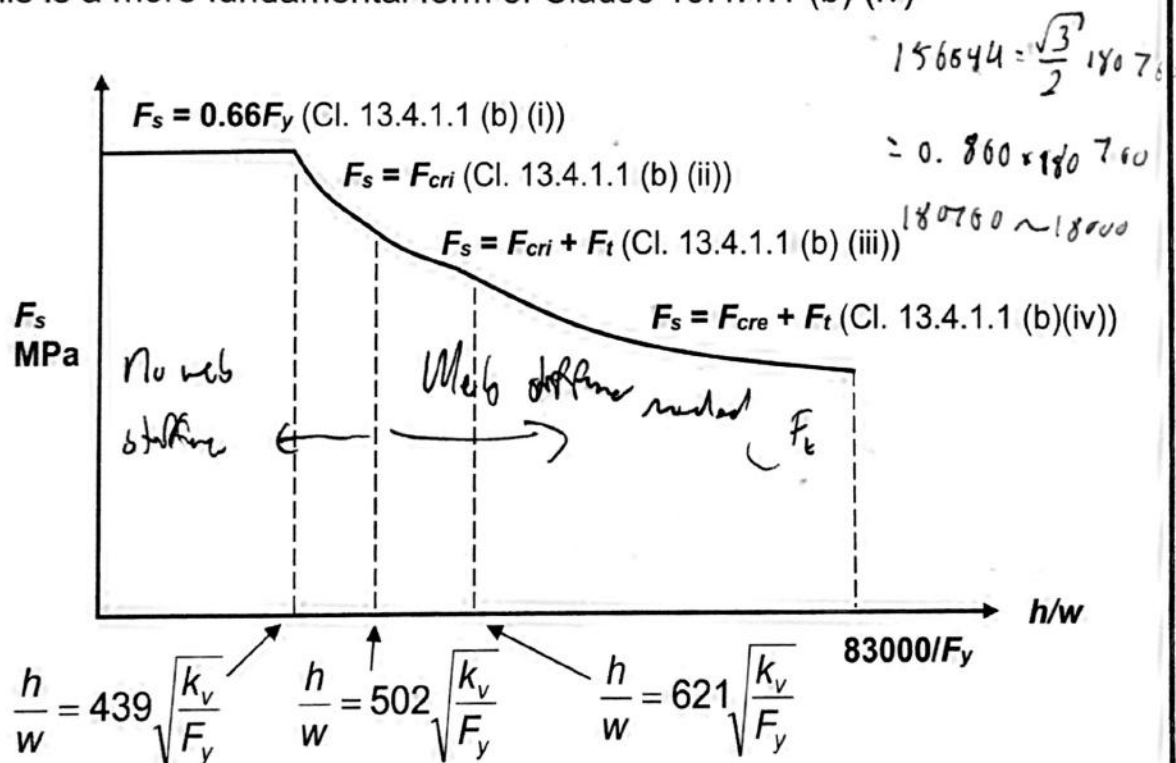
The ultimate shear stress, $F_s = V_u / A_w = V_u / (hw)$

$$F_s = \frac{180760 k_v}{(h/w)^2} + \frac{F_y}{2\sqrt{1+(a/h)^2}} - \frac{156544 k_v}{(h/w)^2 \sqrt{1+(a/h)^2}} \quad (A)$$

$$\approx F_{cre} + k_a(0.5F_y - 0.866F_{cre})$$

Eq (A)

This is a more fundamental form of Clause 13.4.1.1 (b) (iv)



Shear Resistance :**Stiffened Webs of Flexural Members with Two Flanges**

13.4.1

Elastic Analysis

13.4.1.1

The factored shear resistance, V_r , developed by the web of a flexural member shall be taken as

$$V_r = \phi A_w F_s$$

where, $A_w = d \times w$ for rolled shapes and $h \times w$ for girders; F_s is taken as follows:

(a) when $\frac{h}{w} \leq 439 \sqrt{\frac{k_v}{F_y}}$;

$F_s = 0.66 F_y$ *Web shear yielding*

(b) when $439 \sqrt{\frac{k_v}{F_y}} < \frac{h}{w} \leq 502 \sqrt{\frac{k_v}{F_y}}$;

$F_s = F_{cri}$ *Web inelastic shear buckling*

No web stiffener
↑

(c) when $502 \sqrt{\frac{k_v}{F_y}} < \frac{h}{w} \leq 621 \sqrt{\frac{k_v}{F_y}}$;

$F_s = F_{cri} + k_a(0.50 F_y - 0.866 F_{cri})$

Web inelastic shear buckling strength + post inelastic shear buckling strength

↓
w/ web stiffener

(d) when $621 \sqrt{\frac{k_v}{F_y}} < \frac{h}{w}$;

$F_s = F_{cre} + k_a(0.50 F_y - 0.866 F_{cre})$

where, k_v = shear buckling coefficient

(i) when $\frac{a}{h} < 1$;

$k_v = 4 + \frac{5.34}{(a/h)^2}$

Web elastic shear buckling strength + post elastic shear buckling

(ii) when $\frac{a}{h} \geq 1$;

$k_v = 5.34 + \frac{4}{(a/h)^2}$

a/h = aspect ratio, the ratio of the distance between stiffeners to web depth

$$F_{cri} = 290 \frac{\sqrt{F_y k_v}}{(h/w)}$$

k_a = aspect coefficient = $\frac{1}{\sqrt{1 + (a/h)^2}}$

$$F_{cre} = \frac{180000 k_v}{(h/w)^2}$$