- Before continuing, we want to remind you that this a brief introduction to the FVM.
- Let us use the general transport equation as the starting point to explain the FVM,

$$\int_{V_P} \underbrace{\frac{\partial \rho \phi}{\partial t} dV}_{\text{temporal derivative}} + \int_{V_P} \underbrace{\nabla \cdot (\rho \mathbf{u} \phi) dV}_{\text{convective term}} - \int_{V_P} \underbrace{\nabla \cdot (\rho \Gamma_{\phi} \nabla \phi) dV}_{\text{diffusion term}} = \int_{V_P} \underbrace{S_{\phi} (\phi) dV}_{\text{source term}}$$

- We want to solve the general transport equation for the transported quantity $\,\phi\,$ in a given domain, with given boundary conditions BC and initial conditions IC.
- This is a second order equation. For good accuracy, it is necessary that the order of the discretization is equal or higher than the order of the equation that is being discretized.
- By the way, starting from this equation we can write down the Navier-Stokes equations (NSE). So everything we are going to address also applies to the NSE.

Let us use the general transport equation as the starting point to explain the FVM,

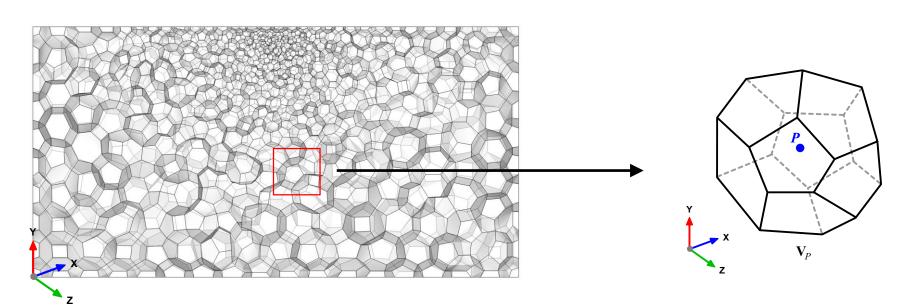
$$\int_{V_P} \underbrace{\frac{\partial \rho \phi}{\partial t} dV}_{\text{temporal derivative}} + \int_{V_P} \underbrace{\nabla \cdot (\rho \mathbf{u} \phi) dV}_{\text{convective term}} - \int_{V_P} \underbrace{\nabla \cdot (\rho \Gamma_{\phi} \nabla \phi) dV}_{\text{diffusion term}} = \int_{V_P} \underbrace{S_{\phi} (\phi) dV}_{\text{source term}}$$

- Hereafter we are going to assume that the discretization practice is at least second order accurate in space and time.
- As consequence of the previous requirement, all dependent variables are assumed to vary linearly around a point P in space and instant t in time,

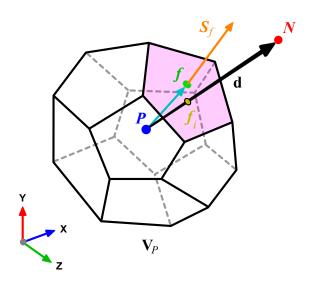
$$\phi(\mathbf{x}) = \phi_P + (\mathbf{x} - \mathbf{x}_P) \cdot (\nabla \phi)_P \quad \text{where} \quad \phi_P = \phi(\mathbf{x}_P)$$
$$\phi(t + \delta t) = \phi^t + \delta t \left(\frac{\partial \phi}{\partial t}\right)^t \quad \text{where} \quad \phi^t = \phi(t)$$

Profile assumptions using Taylor expansions around point *P* (in space) and point *t* (in time)

- Let us divide the solution domain into a finite number of arbitrary control volumes or cells, such as the one illustrated below.
- Inside each control volume the solution is sought.
- The control volumes can be of any shape (e.g., tetrahedrons, hexes, prisms, pyramids, dodecahedrons, and so on).
- The only requirement is that the elements need to be convex and the faces that made up the control volume need to be planar.
- We also know which control volumes are internal and which control volumes lie on the boundaries.



- In the FVM, a lot of overhead goes into the data book-keeping of the domain information.
- We know the following information of every control volume V_P in the domain:
 - The control volume V_P has a volume V and is constructed around point P, which is the centroid of the control volume. Therefore the notation V_P .
 - The vector from the centroid **P** of V_P to the centroid **N** of V_N is named **d**.
 - We also know all neighbors V_N of the control volume V_P .
 - The control volume faces are labeled *f*, which also denotes the face center.
 - The location where the vector **d** intersects a face is f_i .
 - The face area vector S_f point outwards from the control volume, is located at the face centroid, is normal to the face and has a magnitude equal to the area of the face.
 - The vector from the centroid P to the face center f is named Pf.

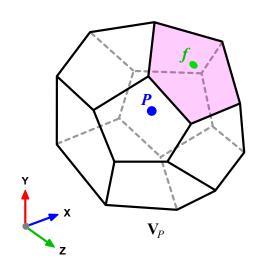


In the control volume illustrated, the centroid P is given by,

$$\int_{V_P} \left(\mathbf{x} - \mathbf{x}_P \right) dV = 0$$

In the same way, the centroid of face f is given by

$$\int_{S_f} (\mathbf{x} - \mathbf{x}_P) dS = 0$$



• Finally, we assume that the values of all variables are computed and stored in the centroid of the control volume V_P and that they are represented by a piecewise constant profile (the mean value),

$$\phi_P = \overline{\phi} = \frac{1}{V_P} \int_{V_P} \phi(\mathbf{x}) dV$$

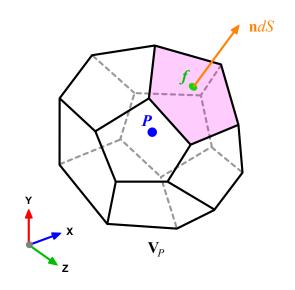
- This is known as the collocated arrangement.
- All the previous approximations are at least second order accurate.

Let us recall the Gauss or Divergence theorem,

$$\int_{V} \nabla \cdot \mathbf{a} dV = \oint_{\partial V} d\mathbf{S} \cdot \mathbf{a}$$

where ∂V_P is a closed surface bounding the control volume V_P and dS represents an infinitesimal surface element with associated normal ${\bf n}$ pointing outwards of the surface ∂V_P , and ${\bf n} dS = d{\bf S}$

- The Gauss or Divergence theorem simply states that the outward flux of a vector field through a closed surface is equal to the volume integral of the divergence over the region inside the surface.
- This theorem is fundamental in the FVM, it is used to convert the volume integrals appearing in the governing equations into surface integrals.



Let us use the Gauss theorem to convert the volume integrals into surface integrals,

$$\int_{V_P} \underbrace{\frac{\partial \rho \phi}{\partial t} dV}_{\text{temporal derivative}} + \int_{V_P} \underbrace{\nabla \cdot (\rho \mathbf{u} \phi) dV}_{\text{convective term}} - \int_{V_P} \underbrace{\nabla \cdot (\rho \Gamma_{\phi} \nabla \phi) dV}_{\text{diffusion term}} = \int_{V_P} \underbrace{S_{\phi} (\phi) dV}_{\text{source term}}$$

$$\int_{V} \nabla \cdot \mathbf{a} dV = \oint_{\partial V} d\mathbf{S} \cdot \mathbf{a}$$

$$\frac{\partial}{\partial t} \int_{V_P} (\rho \phi) \, dV + \oint_{\partial V_P} \underbrace{d\mathbf{S} \cdot (\rho \mathbf{u} \phi)}_{\text{convective flux}} - \oint_{\partial V_P} \underbrace{d\mathbf{S} \cdot (\rho \Gamma_{\phi} \nabla \phi)}_{\text{diffusive flux}} = \int_{V_P} S_{\phi} (\phi) \, dV$$

 At this point the problem reduces to interpolating somehow the cell centered values (known quantities) to the face centers.

 Integrating in space each term of the general transport equation and by using Gauss theorem, yields to the following discrete equations for each term

Convective term:

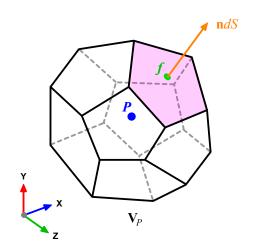
这里指面上的f点,phi_f指在 点处的向量

$$\int_{V_P} \underbrace{\nabla \cdot (\rho \mathbf{u}\phi) \, dV}_{\text{convective term}} = \oint_{\partial V_P} \underbrace{d\mathbf{S} \cdot (\rho \mathbf{u}\phi)}_{\text{convective flux}} = \sum_f \int_f d\mathbf{S} \cdot (\rho \mathbf{u}\phi)_f \approx \sum_f \mathbf{S}_f \cdot (\overline{\rho \mathbf{u}\phi})_f = \sum_f \mathbf{S}_f \cdot (\rho \mathbf{u}\phi)_f$$

By using Gauss theorem we convert volume integrals into surface integrals

where we have approximated the integrant by means of the mid point rule, which is second order accurate

$$\int_{V} \nabla \cdot \mathbf{a} dV = \oint_{\partial V} d\mathbf{S} \cdot \mathbf{a}$$



 Integrating in space each term of the general transport equation and by using Gauss theorem, yields to the following discrete equations for each term

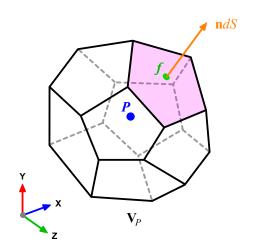
Diffusive term:

$$\int_{V_P} \underbrace{\nabla \cdot (\rho \Gamma_{\phi} \nabla \phi) \, dV}_{\text{diffusion term}} = \oint_{\partial V_P} \underbrace{d\mathbf{S} \cdot (\rho \Gamma_{\phi} \nabla \phi)}_{\text{diffusive flux}} = \sum_{f} \int_{f} d\mathbf{S} \cdot (\rho \Gamma_{\phi} \nabla \phi)_{f} \approx \sum_{f} \mathbf{S}_{f} \cdot (\overline{\rho \Gamma_{\phi} \nabla \phi})_{f} = \sum_{f} \mathbf{S}_{f} \cdot (\rho \Gamma_{\phi} \nabla \phi)_{f}$$

By using Gauss theorem we convert volume integrals into surface integrals

where we have approximated the integrant by means of the mid point rule, which is second order accurate

$$\int_{V} \nabla \cdot \mathbf{a} dV = \oint_{\partial V} d\mathbf{S} \cdot \mathbf{a}$$



 Integrating in space each term of the general transport equation and by using Gauss theorem, yields to the following discrete equations for each term

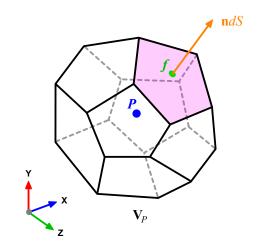
Gradient term:

$$(
abla \phi)_P = rac{1}{V_P} \sum_f ({f S}_f \phi_f)$$
 where ${f S}_f \phi_f$

如果phi 是标量函数,控制体的中心点梯度等于各个面上f点处的值乘以面的法向量如果phi 是向量函数,我就不知道怎么理解了?

where we have approximated the centroid gradients by using the Gauss theorem. This method is second order accurate

$$\int_{V} \nabla \cdot \mathbf{a} dV = \oint_{\partial V} d\mathbf{S} \cdot \mathbf{a}$$



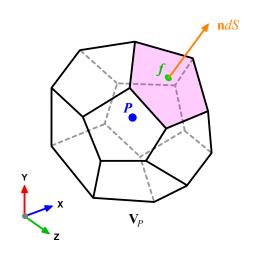
 Integrating in space each term of the general transport equation and by using Gauss theorem, yields to the following discrete equations for each term

Source term:

This approximation is exact if S_{ϕ} is either constant or varies linearly within the control volume; otherwise is second order accurate.

Sc is the constant part of the source term and Sp is the non-linear part

$$\int_{V} \nabla \cdot \mathbf{a} dV = \oint_{\partial V} d\mathbf{S} \cdot \mathbf{a}$$



 Using the previous equations to evaluate the general transport equation over all the control volumes, we obtain the following semi-discrete equation

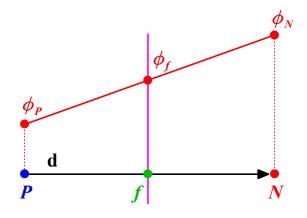
$$\int_{V_P} \underbrace{\frac{\partial \rho \phi}{\partial t} dV}_{\text{temporal derivative}} + \sum_{f} \underbrace{\mathbf{S}_f \cdot (\rho \mathbf{u}\phi)_f}_{\text{convective flux}} - \sum_{f} \underbrace{\mathbf{S}_f \cdot (\rho \Gamma_\phi \nabla \phi)_f}_{\text{difussive flux}} = \underbrace{(S_c V_P + S_p V_P \phi_P)}_{\text{source term}}$$

where $\mathbf{S}\cdot(\rho\mathbf{u}\phi)=F^C$ is the convective flux and $\mathbf{S}\cdot(\rho\Gamma_\phi\nabla\phi)=F^D$ is the diffusive flux.

- And recall that all variables are computed and stored at the centroid of the control volumes.
- The face values appearing in the convective and diffusive fluxes have to be computed by some form of interpolation from the centroid values of the control volumes at both sides of face f.

Interpolation of the convective fluxes

 By looking the figure below, the face values appearing in the convective flux can be computed as follows,



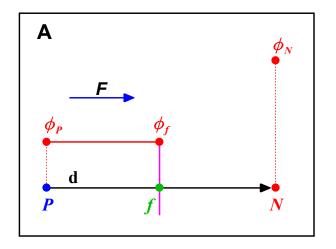
$$\phi_f = f_x \phi_P + (1 - f_x) \phi_N$$

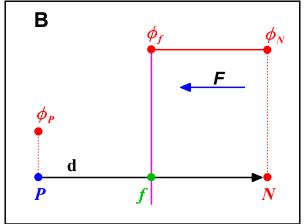
$$f_x = \frac{fN}{PN} = \frac{|\mathbf{x}_f - \mathbf{x}_N|}{|\mathbf{d}|}$$

- This type of interpolation scheme is known as linear interpolation or central differencing and it is second order accurate.
- However, it may generate oscillatory solutions (unbounded solutions).

Interpolation of the convective fluxes

 By looking the figure below, the face values appearing in the convective flux can be computed as follows,



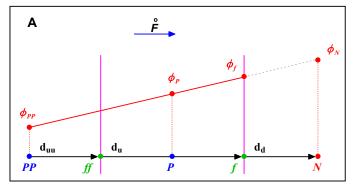


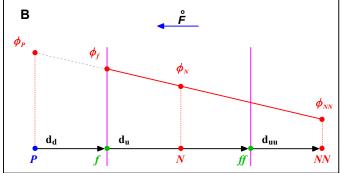
$$\phi_f = \begin{cases} \phi_f = \phi_P & \text{for } \mathring{F} \ge 0, \\ \phi_f = \phi_N & \text{for } \mathring{F} < 0. \end{cases}$$

- This type of interpolation scheme is known as upwind differencing and it is first order accurate.
- This scheme is bounded (non-oscillatory) and diffusive.

Interpolation of the convective fluxes

 By looking the figure below, the face values appearing in the convective flux can be computed as follows,





$$\phi_f = \begin{cases} \phi_P + \frac{1}{2}(\phi_P - \phi_{PP}) = \frac{2}{3}\phi_P - \frac{1}{2}\phi_{PP} & \text{for } \mathring{F} \ge 0, \\ \phi_N + \frac{1}{2}(\phi_N - \phi_{NN}) = \frac{2}{3}\phi_N - \frac{1}{2}\phi_{NN} & \text{for } \mathring{F} < 0. \end{cases}$$

- This type of interpolation scheme is known as second order upwind differencing (SOU), linear upwind differencing (LUD) or Beam-Warming (BW), and it is second order accurate.
- For highly convective flows or in the presence of strong gradients, this scheme is oscillatory (unbounded).

Interpolation of the convective fluxes

To prevent oscillations in the SOU, we add a gradient or slope limiter function $\,\psi(r)$.

$$\phi_f = \begin{cases} \phi_P + \frac{1}{2} \psi_P^-(\phi_P - \phi_{PP}) & \text{for } \mathring{F} \ge 0, \\ \phi_N - \frac{1}{2} \psi_P^+(\phi_{NN} - \phi_N) & \text{for } \mathring{F} < 0. \end{cases}$$

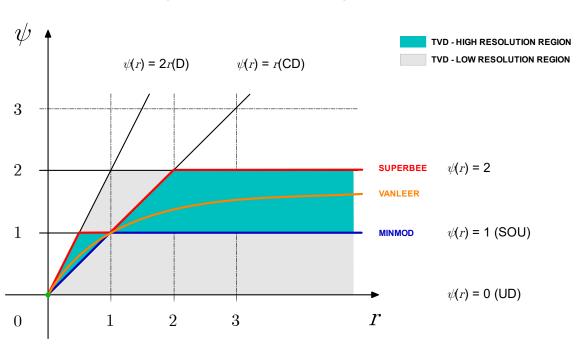
- When the limiter detects strong gradients or changes in slope, it switches locally to low resolution (upwind).
- The concept of the limiter function $\psi(r)$ is based on monitoring the ratio of successive gradients, *e.g.*,

$$r_P^- = \frac{\phi_N - \phi_P}{\phi_P - \phi_{PP}}$$
 and $r_P^+ = \frac{\phi_P - \phi_N}{\phi_N - \phi_{NN}}$

• By adding a well designed limiter function $\psi(r)$, we get a high resolution (second order accurate), and bounded scheme. This is a TVD scheme.

Interpolation of the convective fluxes – TVD schemes

- A TVD scheme, is a second order accurate scheme that does not create new local undershoots and/or overshoots in the solution or amplify existing extremes (high resolution).
- The choice of the limiter function $\psi(r)$ dictates the order of the scheme and its boundedness. High resolution schemes falls in the blue area and low resolution schemes falls in the grey area.
- The drawback of the limiters is that they reduce the accuracy of the scheme **locally** to first order, when $r \le 0$ (sharp gradient, opposite slopes). However, this is justify when it serves to suppress oscillations.
- The various limiters have different switching characteristics and are selected according to the particular problem and solution scheme. No particular limiter has been found to work well for all problems, and a particular choice is usually made on a trial and error basis.
- The Sweby diagram (Sweby, 1984), gives the necessary and sufficient conditions for a scheme to be TVD.



$$\phi_f = \begin{cases} \phi_P + \frac{1}{2} \psi_P^-(\phi_P - \phi_{PP}) & \text{for } \mathring{F} \ge 0, \\ \phi_N - \frac{1}{2} \psi_P^+(\phi_{NN} - \phi_N) & \text{for } \mathring{F} < 0. \end{cases}$$

UD = upwind SOU = second

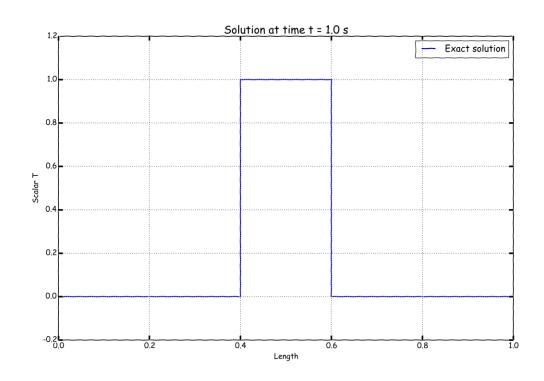
SOU = second order upwind

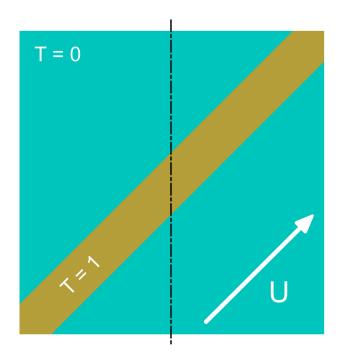
CD = central differencing

D = downwind

Interpolation of the convective fluxes – TVD schemes

- Let us see how the superbee, minmod and vanleer TVD schemes behave in a numerical schemes killer test case:
 - The oblique double step profile in a uniform vector field (pure convection).
- By the way, this problem has an exact solution.

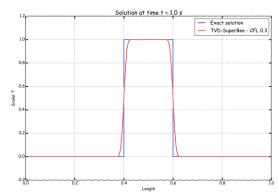


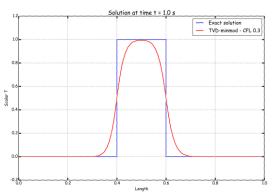


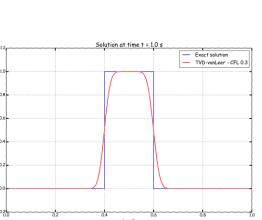
Interpolation of the convective fluxes – Linear and non-linear limiter functions

- Let us see how the superbee, minmod and vanleer TVD schemes behave in a numerical schemes killer test case.
 - The oblique double step profile in a uniform vector field (pure convection).

SuperBee - Compressive Minmod - Diffusive







vanLeer - Smooth

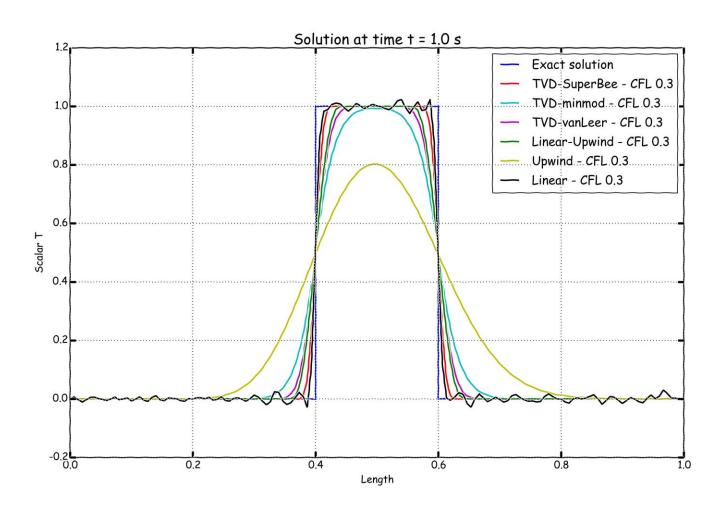
Interpolation of the convective fluxes – Linear and non-linear limiter functions

- Comparison of linear upwind method (2nd order) and upwind method (1st order).
- The upwind method is extremely stable and non-oscillatory. However is highly diffusive.
- On the other side, the linear upwind method is accurate but oscillatory in the presence of strong gradients.



Interpolation of the convective fluxes – Linear and non-linear limiter functions

Let us see how the linear and non-linear limiter functions compare.



Interpolation of the convective fluxes – Unstructured meshes

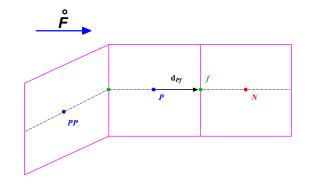
- All higher-order schemes we have seen so far assume line structure (figure A).
- That is, the cell centers PP, P, and N are all aligned.
- In unstructured meshes, it is not straightforward to use the previous schemes, as the cell center
 PP is not aligned with the vector connecting cells P and N (figure B).
- Higher-order schemes for unstructured meshes are an area of active research and new ideas continue to emerge.

A B d_{Pf} f



Interpolation of the convective fluxes – Unstructured meshes

- A simple way around this problem is to redefine higher-order schemes in terms of gradients at the control volume P.
- For example, using the gradient of the cells, we can compute the face values as follows,



Upwind
$$\rightarrow \phi_f = \phi_P$$

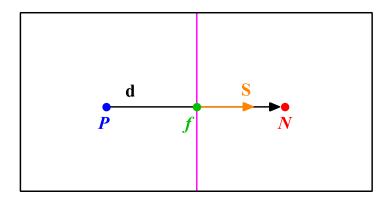
Central difference
$$\rightarrow \phi_f = \phi_P + \nabla \phi_f \cdot \mathbf{d}_{Pf}$$

Second order upwind differencing
$$\rightarrow \phi_f = \phi_P + (2\nabla\phi_P - \nabla\phi_f) \cdot \mathbf{d}_{Pf}$$

- Notice that in this new formulation the cell PP does not appear any more.
- The problem now turns in the accurate evaluation of the gradients at the cell and face centers.
 This can be done using Gauss method as previously explained.

Interpolation of diffusive fluxes in a orthogonal mesh

By looking the figures below, the face values appearing in the diffusive flux in an orthogonal mesh can be computed as follows,



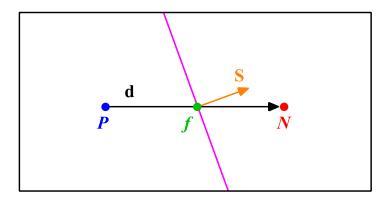
$$\mathbf{S} \cdot (\nabla \phi)_f = |\mathbf{S}| \frac{\phi_N - \phi_P}{|\mathbf{d}|}.$$

ohi 是向量函数,当phi _N与 ohi _P与相减除以它们的距离 d即为梯度

 This is a central difference approximation of the first order derivative. This type of approximation is second order accurate.

Interpolation of diffusive fluxes in a non-orthogonal mesh

By looking the figures below, the face values appearing in the diffusive flux in a non-orthogonal mesh (20°) can be computed as follows,

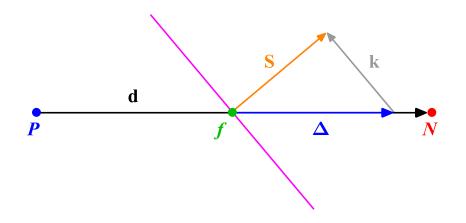


$$\mathbf{S} \cdot (\nabla \phi)_f = \underbrace{|\Delta_{\perp}| \frac{\phi_N - \phi_P}{|\mathbf{d}|}}_{\text{orthogonal contribution}} + \underbrace{\mathbf{k} \cdot (\nabla \phi)_f}_{\text{non-orthogonal contribution}}$$

 This type of approximation is second order accurate but involves a larger truncation error. It also uses a larger numerical stencil, which make it less stable.

Correction of diffusive fluxes in a non-orthogonal mesh

- By looking the figures below, the face values appearing in the diffusive flux in a non-orthogonal mesh (40°) can be computed as follows.
- Using the over-relaxed approach, the diffusive fluxes can be corrected as follow,



Over-relaxed approach

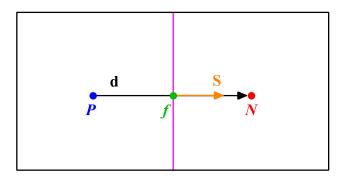
$$\Delta_{\perp} = \frac{\mathbf{d}}{\mathbf{d} \cdot \mathbf{S}} |\mathbf{S}|^2.$$

$$\mathbf{S} = \Delta_{\perp} + \mathbf{k}.$$

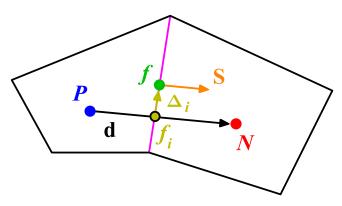
$$\mathbf{S} \cdot (\nabla \phi)_f = \underbrace{|\Delta_{\perp}| \frac{\phi_N - \phi_P}{|\mathbf{d}|}}_{\text{orthogonal contribution}} + \underbrace{\mathbf{k} \cdot (\nabla \phi)_f}_{\text{non-orthogonal contribution}}$$

Mesh induced errors

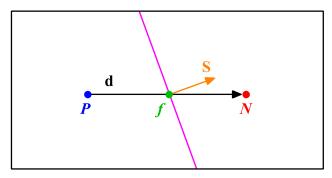
- In order to maintain second order accuracy, and to avoid unboundedness, we need to correct non-orthogonality and skewness errors.
- The ideal case is to have an orthogonal and non skew mesh, but this is the exception rather than the rule.



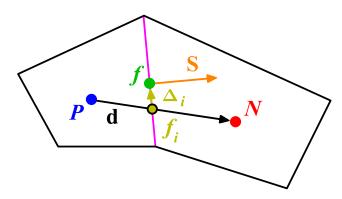
Orthogonal and non skew mesh



Orthogonal and skew mesh



Non-orthogonal and non skew mesh



Non-orthogonal and skew mesh

Temporal discretization

 Using the previous equations to evaluate the general transport equation over all the control volumes, we obtain the following semi-discrete equation,

$$\int_{V_P} \underbrace{\frac{\partial \rho \phi}{\partial t} dV}_{\text{temporal derivative}} + \sum_{f} \underbrace{\mathbf{S}_f \cdot (\rho \mathbf{u} \phi)_f}_{\text{convective flux}} - \sum_{f} \underbrace{\mathbf{S}_f \cdot (\rho \Gamma_{\phi} \nabla \phi)_f}_{\text{difussive flux}} = \underbrace{(S_c V_P + S_p V_P \phi_P)}_{\text{source term}}$$

where $\mathbf{S} \cdot (\rho \mathbf{u} \phi) = F^C$ is the convective flux and $\mathbf{S} \cdot (\rho \Gamma_{\phi} \nabla \phi) = F^D$ is the diffusive flux.

- After spatial discretization, we can proceed with the temporal discretization. By proceeding in this way we are using the Method of Lines (MOL).
- The main advantage of the MOL method, is that it allows us to select numerical approximations of different accuracy for the spatial and temporal terms. Each term can be treated differently to yield to different accuracies.

Temporal discretization

Now, we evaluate in time the semi-discrete general transport equation

$$\int_{t}^{t+\Delta t} \left[\left(\frac{\partial \rho \phi}{\partial t} \right)_{P} V_{P} + \sum_{f} \mathbf{S}_{f} \cdot (\rho \mathbf{u} \phi)_{f} - \sum_{f} \mathbf{S}_{f} \cdot (\rho \Gamma_{\phi} \nabla \phi)_{f} \right] dt$$

$$= \int_{t}^{t+\Delta t} \left(S_{c} V_{P} + S_{p} V_{P} \phi_{P} \right) dt.$$

- At this stage, we can use any time discretization scheme, e.g., Crank-Nicolson, euler implicit, forward euler, backward differencing, adams-bashforth, adams-moulton.
- It should be noted that the order of the temporal discretization of the transient term does not need to be the same as the order of the discretization of the spatial terms.
 Each term can be treated differently to yield different accuracies. As long as the individual terms are at least second order accurate, the overall accuracy will also be second order.

Linear system solution

After spatial and temporal discretization and by using equation

$$\int_{t}^{t+\Delta t} \left[\left(\frac{\partial \rho \phi}{\partial t} \right)_{P} V_{P} + \sum_{f} \mathbf{S}_{f} \cdot (\rho \mathbf{u} \phi)_{f} - \sum_{f} \mathbf{S}_{f} \cdot (\rho \Gamma_{\phi} \nabla \phi)_{f} \right] dt = \int_{t}^{t+\Delta t} (S_{c} V_{P} + S_{p} V_{P} \phi_{P}) dt$$

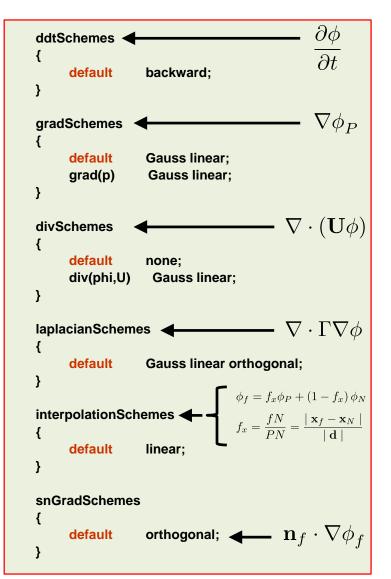
in every control volume V_P of the domain, a system of linear algebraic equations for the transported quantity ϕ is assembled,

This system can be solved by using any iterative or direct method.

So, what does OpenFOAM® do?

- It simply discretize in space and time the governing equations in arbitrary polyhedral control volumes over the whole domain. Assembling in this way a large set of linear discrete algebraic equations (DAE), and then it solves this system of DAE to find the solution of the transported quantities.
- Therefore, we need to give to OpenFOAM® the following information:
 - Discretization of the solution domain or the mesh. This information is contained in the directory constant/polyMesh
 - Boundary conditions and initials conditions. This information is contained in the directory 0
 - Physical properties such as density, gravity, diffusion coefficient, viscosity, etc. This information is contained in the directory constant
 - Physics involve, such as turbulence modeling, mass transfer, source terms, etc. This information is contained in the directories constant and/or system
 - How to discretize in space each term of the governing equations (diffusive, convective, gradient and source terms). This information is set in the <code>system/fvSchemes</code> dictionary.
 - How to discretize in time the obtained semi-discrete governing equations. This information is set in the system/fvSchemes dictionary.
 - How to solve the linear system of discrete algebraic equations (crunch numbers). This information is set in the <code>system/fvSolution</code> dictionary.
 - Set runtime parameters and general instructions on how to run the case (such as time step and maximum CFL number). This information is set in the system/controlDict dictionary.
 - Additionally, we may set sampling and functionObjects for post-processing. This information is contained in the specific dictionaries contained in the directory system/

Where do we set all the discretization schemes in OpenFOAM®?



- The fvSchemes dictionary contains the information related to the discretization schemes for the different terms appearing in the governing equations.
- The discretization schemes can be chosen in a term-by-term basis.
- The keyword ddtSchemes refers to the time discretization.
- The keyword gradSchemes refers to the gradient term discretization.
- The keyword divSchemes refers to the convective terms discretization.
- The keyword laplacianSchemes refers to the Laplacian terms discretization.
- The keyword interpolationSchemes refers to the method used to interpolate values from cell centers to face centers. It is unlikely that you will need to use something different from linear.
- The keyword snGradSchemes refers to the discretization of the surface normal gradients evaluated at the faces.
- Remember, if you want to know the options available for each keyword you can use the banana method.

Time discretization schemes

- There are many time discretization schemes available in OpenFOAM®.
- You will find the source code in the following directory:
 - \$WM_PROJECT_DIR/src/finiteVolume/finiteVolume/ddtSchemes
- These are the time discretization schemes that you will use most of the times:
 - steadyState: for steady state simulations (implicit/explicit).
 - **Euler:** time dependent first order (implicit/explicit), bounded.
 - backward: time dependent second order (implicit), bounded/unbounded.
 - CrankNicolson: time dependent second order (implicit), bounded/unbounded.
- First order methods are bounded and stable, but diffusive.
- Second order methods are accurate, but they might become oscillatory.
- At the end of the day, we always want a second order accurate solution.
- If you keep the CFL less than one when using the Euler method, numerical diffusion is not that much (however, we advise you to do your own benchmarking).

Convective terms discretization schemes

- There are many convective terms discretization schemes available in OpenFOAM® (more than 50 last time we checked).
- You will find the source code in the following directory:
 - \$WM_PROJECT_DIR/src/finiteVolume/interpolation/surfaceInterpolation
- These are the convective discretization schemes that you will use most of the times:
 - upwind: first order accurate.
 - linearUpwind: second order accurate, bounded.
 - linear: second order accurate, unbounded.
 - vanLeer: TVD, second order accurate, bounded.
 - **limitedLinear:** second order accurate, unbounded, but more stable than pure linear. Recommended for LES simulations (kind of similar to the Fromm method).
- First order methods are bounded and stable but diffusive.
- Second order methods are accurate, but they might become oscillatory.
- At the end of the day, we always want a second order accurate solution.

Convective terms discretization schemes

When you use **linearUpwind** for **div(phi,U)**, you need to tell OpenFOAM® how to compute the velocity gradient or **grad(U)**:

Same applies for scalars (e.g. k, epsilon, omega, T)

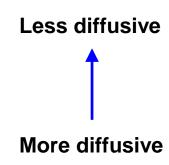
Gradient terms discretization schemes

- There are many gradient discretization schemes available in OpenFOAM®.
- You will find the source code in the following directory:
 - \$WM_PROJECT_DIR/src/finiteVolume/finiteVolume/gradSchemes
- These are the gradient discretization schemes that you will use most of the times:
 - Gauss
 - leastSquares
- To avoid overshoots or undershoots when computing the gradients, you can use gradient limiters.
- Gradient limiters increase the stability of the method but add diffusion due to clipping.
- You will find the source code in the following directory:
 - \$WM_PROJECT_DIR/src/finiteVolume/finiteVolume/gradSchemes/limitedGradSchemes
- These are the gradient limiter schemes available in OpenFOAM®:
 - cellLimited, cellMDLimited, faceLimited, faceMDLimited
- All of the gradient discretization schemes are at least second order accurate.

Gradient terms discretization schemes

These are the gradient limiter schemes available in OpenFOAM®:

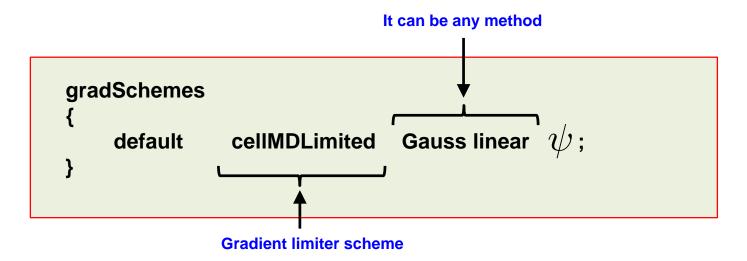
cellMDLimited
cellLimited
faceMDLimited
faceLimited



- Cell limiters will limit cell-to-cell values.
- Face limiters will limit face-to-cell values.
- The multi-directional (dimensional) limiters (cellMDLimited and faceMDLimited), will apply the limiter in each face direction separately.
- The standard limiters (cellLimited and faceLimited), will apply the limiter to all components of the gradient.
- The default method is the Minmod.

Gradient terms discretization schemes

• The gradient limiter implementation in OpenFOAM®, uses a blending factor ψ .



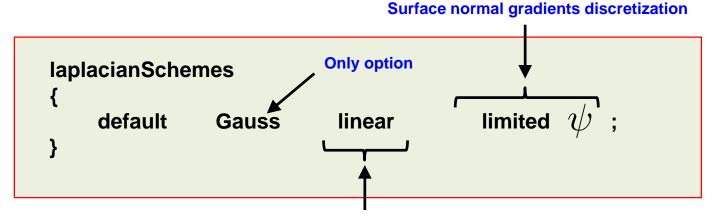
- Setting ψ to 0 is equivalent to turning off the gradient limiter. You gain accuracy but the solution might become unbounded.
- By setting the blending factor equal to 1 the limiter is always on. You gain stability but you give up accuracy (due to gradient clipping).
- If you set the blending factor to 0.5, you get the best of both worlds.
- You can use limiters with all gradient discretization schemes.

Laplacian terms discretization schemes

- There are many Laplacian terms discretization schemes available in OpenFOAM®.
- You will find the source code in the following directory:
 - \$WM_PROJECT_DIR/src/finiteVolume/finiteVolume/snGradSchemes
- These are the Laplacian terms discretization schemes that you will use most of the times:
 - orthogonal: mainly limited for hexahedral meshes with no grading (a perfect mesh).
 Second order accurate, bounded on perfect meshes, without non-orthogonal corrections.
 - **corrected:** for meshes with grading and non-orthogonality. Second order accurate, bounded depending on the quality of the mesh, with non-orthogonal corrections.
 - **limited:** for meshes with grading and non-orthogonality. Second order accurate, bounded depending on the quality of the mesh, with non-orthogonal corrections.
 - uncorrected: usually used on bad quality meshes with grading and non-orthogonality.
 Second order accurate, without non-orthogonal corrections. Stable but more diffusive than the limited and corrected methods.

Laplacian terms discretization schemes

The limited method uses a blending factor ψ .

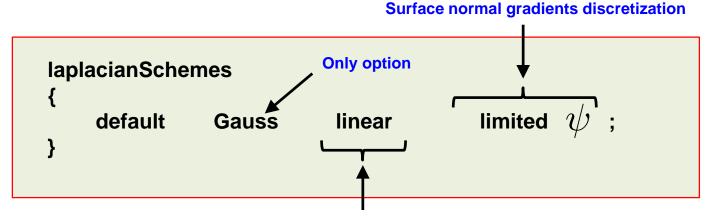


Interpolation method of the diffusion coefficient

- Setting ψ to 1 is equivalent to using the **corrected** method. You gain accuracy, but the solution might become unbounded.
- By setting the blending factor equal to 0 is equivalent to using the uncorrected method. You
 give up accuracy but gain stability.
- If you set the blending factor to 0.5, you get the best of both worlds. In this case, the nonorthogonal contribution does not exceed the orthogonal part. You give up accuracy but gain stability.

Laplacian terms discretization schemes

The limited method uses a blending factor ψ .

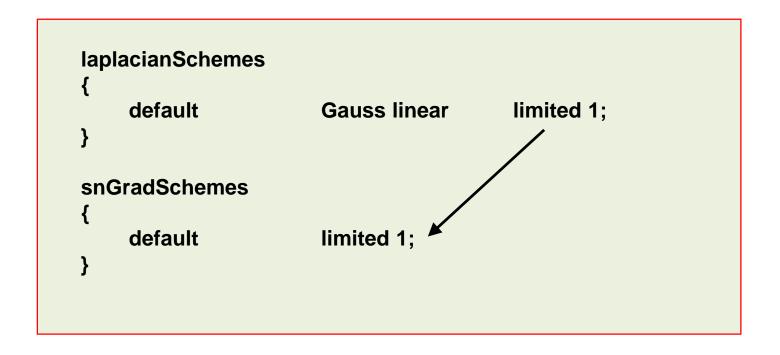


Interpolation method of the diffusion coefficient

- For meshes with non-orthogonality less than 75, you can set the blending factor to 1.
- For meshes with non-orthogonality between 75 and 85, you can set the blending factor to 0.5
- For meshes with non-orthogonality more than 85, it is better to get a better mesh. But if you
 definitely want to use that mesh, you can set the blending factor to 0.333, and increase the
 number of non-orthogonal corrections.
- If you are doing LES or DES simulations, use a blending factor of 1 (this means that you need good meshes).

Laplacian terms discretization schemes

- The surface normal gradients terms usually use the same method as the one chosen for the Laplacian terms.
- For instance, if you are using the limited 1 method for the Laplacian terms, you can use the same method for snGradSchemes:



What method should I use?

```
ddtSchemes
                   CrankNicolson 0:
     default
gradSchemes
     default
                   cellLimited Gauss linear 1:
                   cellLimited Gauss linear 1:
     grad(U)
divSchemes
     default
                           none;
     div(phi,U)
                           Gauss linearUpwindV grad(U);
     div(phi,omega)
                           Gauss linearUpwind default;
     div(phi,k)
                           Gauss linearUpwind default;
     div((nuEff*dev(T(grad(U)))))
                                  Gauss linear;
laplacianSchemes
                   Gauss linear limited 1:
     default
interpolationSchemes
     default
                   linear:
snGradSchemes
                   limited 1;
     default
```

 This setup is recommended for most of the cases.



- It is equivalent to the default method you will find in commercial solvers.
- In overall, this setup is second order accurate and fully bounded.
- According to the quality of your mesh, you will need to change the blending factor of the laplacianSchemes and snGradSchemes keywords.
- To keep temporal diffusion to a minimum, use a CFL number less than 2.
- If during the simulation the turbulence quantities become unbounded, you can safely change the discretization scheme to upwind. After all, turbulence is diffusion.

A very accurate but oscillatory numerics

```
ddtSchemes
                   backward:
     default
gradSchemes
                   Gauss leastSquares;
     default
divSchemes
     default
                           none;
    div(phi,U)
                           Gauss linear;
                           Gauss linear;
     div(phi,omega)
     div(phi,k)
                           Gauss linear:
     div((nuEff*dev(T(grad(U))))) Gauss linear;
laplacianSchemes
                   Gauss linear limited 1;
     default
interpolationSchemes
     default
                   linear;
snGradSchemes
                   limited 1;
     default
```

- If you are looking for more accuracy, you can use this method.
- In overall, this setup is second order accurate but oscillatory.
- Use this setup with LES simulations or laminar flows with no complex physics, and meshes with overall good quality.
- Use this method with good quality meshes.

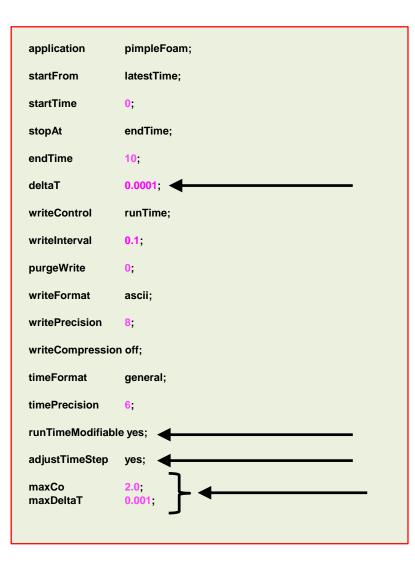
A very stable but too diffusive numerics

```
ddtSchemes
     default
                   Euler:
gradSchemes
     default
                    cellLimited Gauss linear 1:
                   cellLimited Gauss linear 1:
     grad(U)
divSchemes
     default
                           none;
     div(phi,U)
                           Gauss upwind;
     div(phi,omega)
                           Gauss upwind:
     div(phi,k)
                           Gauss upwind;
     div((nuEff*dev(T(grad(U))))) Gauss linear;
laplacianSchemes
     default
                   Gauss linear limited 0.5:
interpolationSchemes
     default
                    linear:
snGradSchemes
                   limited 0.5;
     default
```

- If you are looking for extra stability, you can use this method.
- This setup is very stable but too diffusive.
- This setup is first order in space and time.
- You can use this setup to start the solution in the presence of bad quality meshes or strong discontinuities.
- Remember, you can start using a first order method and then switch to a second order method.
- Start robustly, end with accuracy.

On the CFL number

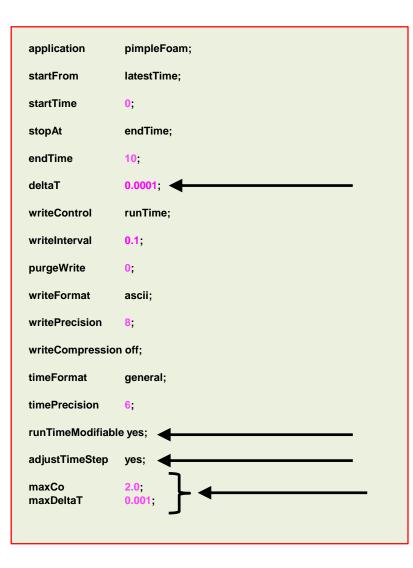
How to control the CFL number



- You can control the CFL number by changing the mesh cell size or changing the time-step size.
- The easiest way is by changing the time-step size.
- If you refine the mesh, and you would like to have the same CFL number as the base mesh, you will need to decrease the time-step size.
- On the other side, if you coarse the mesh and you would like to have the same CFL number as the base mesh, you will need to increase the time-step size.
- The keyword **deltaT** controls the time-step size of the simulation (0.0001 seconds in this generic case).
- If you use a solver that supports adjustable time-step (adjustTimeStep), you can set the maximum CFL number and maximum allowable time-step using the keywords maxCo and maxDeltaT, respectively.

On the CFL number

How to control the CFL number



- The option adjustTimeStep will automatically adjust the time step to achieve the maximum desired courant number (maxCo) or time-step size (maxDeltaT).
- When any of these conditions is reached, the solver will stop scaling the time-step size.
- To use these features, you need to turn-on the option adjustTimeStep.
- Remember, the first time step of the simulation is done using the value defined with the keyword deltaT and then it is automatically scaled (up or down), to achieve the desired maximum values (maxCo and maxDeltaT).
- It is recommended to start the simulation with a low time-step in order to let the solver scale-up the time-step size.
- If you want to change the values on-the-fly, you need to turnon the option **runTimeModifiable**.
- The feature adjustTimeStep is only present in the PIMPLE family solvers, but it can be added to any solver by modifying the source code.

On the CFL number

The output screen

- This is the output screen of a solver supporting the option adjustTimeStep.
- In this case maxCo is equal 2 and maxDeltaT is equal to 0.001.
- Notice that the solver reached the maximum allowable maxDeltaT.

```
Courant number (mean and maximum values)
Courant Number mean: 0.10863988 max: 0.73950028
deltaT = 0.001 ← Current time-step
Time = 30.000289542261612

    Simulation time

                                                             One PIMPLE iteration (outer loop), this is equivalent to PISO
PIMPLE: iteration 1
DILUPBICG: Solving for Ux, Initial residual = 0.003190933, Final residual = 1.0207483e-09, No Iterations 5
DILUPBICG: Solving for Uy, Initial residual = 0.0049140114, Final residual = 8.5790109e-10, No Iterations 5
DILUPBICG: Solving for Uz, Initial residual = 0.010705877, Final residual = 3.5464756e-09, No Iterations 4
GAMG: Solving for p, Initial residual = 0.024334674, Final residual = 0.0005180308, No Iterations 3
GAMG: Solving for p, Initial residual = 0.00051825089, Final residual = 1.6415538e-05, No Iterations 5
time step continuity errors : sum local = 8.768064e-10, global = 9.8389717e-11, cumulative = -2.6474162e-07
GAMG: Solving for p, Initial residual = 0.00087813032, Final residual = 1.6222017e-05, No Iterations 3
GAMG: Solving for p, Initial residual = 1.6217958e-05, Final residual = 6.4475277e-06, No Iterations 1
time step continuity errors : sum local = 3.4456296e-10, qlobal = 2.6009599e-12, cumulative = -2.6473902e-07
ExecutionTime = 33091.06 s ClockTime = 33214 s

    CPU time and wall clock

fieldMinMax domainminandmax output:
   \min(p) = -0.59404715 at location (-0.019\ 0.02082288\ 0.072) on processor 1
   \max(p) = 0.18373302 at location (-0.02083962 -0.003 -0.136) on processor 1
    \min(U) = (0.29583255 - 0.4833922 - 0.0048229716) at location (-0.02259661 - 0.02082288 - 0.072) on processor 0
    \max(U) = (0.59710937 \ 0.32913292 \ 0.020043679) at location (0.11338793 \ -0.03267608 \ 0.12) on processor 3
   min(nut) = 1.6594481e-10 at location (0.009 - 0.02 \ 0.024) on processor 0
   \max(\text{nut}) = 0.00014588174 at location (-0.02083962 0.019 0.072) on processor 1
yPlus yplus output:
   patch square y+: min = 0.44603573, max = 6.3894913, average = 2.6323389
    writing field yPlus
```

After spatial and temporal discretization and by using equation

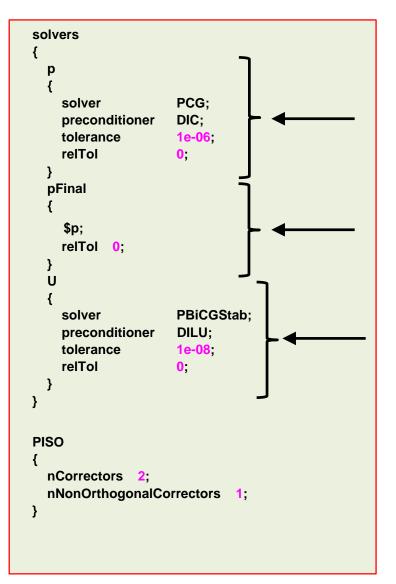
$$\int_{t}^{t+\Delta t} \left[\left(\frac{\partial \rho \phi}{\partial t} \right)_{P} V_{P} + \sum_{f} \mathbf{S}_{f} \cdot (\rho \mathbf{u} \phi)_{f} - \sum_{f} \mathbf{S}_{f} \cdot (\rho \Gamma_{\phi} \nabla \phi)_{f} \right] dt = \int_{t}^{t+\Delta t} (S_{c} V_{P} + S_{p} V_{P} \phi_{P}) dt$$

in every control volume V_P of the domain, a system of linear algebraic equations for the transported quantity ϕ is assembled

This system can be solved by using any iterative or direct method.

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC;
    tolerance
                      1e-06:
    relTol
                     0;
  pFinal
    $p;
    relTol 0:
                     PBiCGStab:
    solver
    preconditioner
                     DILU;
    tolerance
                     1e-08:
    relTol
                     0;
PISO
  nCorrectors 2;
  nNonOrthogonalCorrectors 1;
```

- The equation solvers, tolerances, and algorithms are controlled from the sub-dictionary solvers located in the fvSolution dictionary file.
- In the dictionary file fvSolution, and depending on the solver you are using you will find the additional subdictionaries PISO, PIMPLE, and SIMPLE, which will be described later.
- In this dictionary is where we are telling OpenFOAM® how to crunch numbers.
- The solvers sub-dictionary specifies each linear-solver that is used for each equation being solved.
- If you forget to define a linear-solver, OpenFOAM® will let you know what are you missing.
- The syntax for each entry within the solvers sub-dictionary uses a keyword that is the word relating to the variable being solved in the particular equation and the options related to the linear-solver.



- In this generic case, to solve the pressure (**p**) we are using the **PCG** method with the **DIC** preconditioner, an absolute **tolerance** equal to 1e-06 and a relative tolerance **relTol** equal to 0.
- The entry **pFinal** refers to the final pressure correction (notice that we are using macro syntax), and we are using a relative tolerance **relTol** equal to 0.
- To solve the velocity field (**U**) we are using the **PBiCGStab** method with the **DILU** preconditoner, an absolute **tolerance** equal to 1e-08 and a relative tolerance **relTol** equal to 0.
- The linear solvers will iterative until reaching any of the tolerance values set by the user or reaching a maximum value of iterations (optional entry).
- FYI, solving for the velocity is relative inexpensive, whereas solving for the pressure is expensive.
- The pressure equation is particularly important as it governs mass conservation.
- If you do not solve the equations accurately enough (tolerance), the physics might be wrong.
- Selection of the tolerance is of paramount importance.

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC:
    tolerance
                     1e-06:
    relTol
                     0;
  pFinal
    $p;
    relTol 0:
                     PCG:
    solver
    preconditioner
                     DILU;
    tolerance
                     1e-08:
    relTol
                     0;
PISO
  nCorrectors 2;
  nNonOrthogonalCorrectors 1;
```

- The linear solvers distinguish between symmetric matrices and asymmetric matrices.
- The symmetry of the matrix depends on the structure of the equation being solved.
- Pressure is a symmetric matrix and velocity is an asymmetric matrix.
- If you use the wrong linear solver, OpenFOAM® will complain and will let you know what options are valid.
- In the following error screen, we are using a symmetric solver for an asymmetric matrix,

```
-> FOAM FATAL IO ERROR:
Unknown asymmetric matrix solver PCG

Valid asymmetric matrix solvers are:

4
(
BICCG
GAMG
P
smoothSolver
```

Linear solvers

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC;
    tolerance
                     1e-06:
    relTol
                     0;
  pFinal
    $p;
    relTol 0:
                     PBiCGStab;
    solver
    preconditioner
                     DILU;
    tolerance
                     1e-08:
    relTol
                     0;
PISO
  nCorrectors 2;
  nNonOrthogonalCorrectors 1;
```

- The linear solvers are iterative, *i.e.*, they are based on reducing the equation residual over a succession of solutions.
- The residual is a measure of the error in the solution so that the smaller it is, the more accurate the solution.
- More precisely, the residual is evaluated by substituting the current solution into the equation and taking the magnitude of the difference between the left and right hand sides (L2-norm).

$$\left|\mathbf{A}\phi^{\mathbf{k}} - \mathbf{b}\right| = \left|\mathbf{r}^{\mathbf{k}}\right|$$

 It is also normalized to make it independent of the scale of the problem being analyzed.

Residual =
$$\frac{|\mathbf{r}|}{\text{Normalization factor}} < \text{Tolerance}$$

```
solvers
                      PCG:
    solver
    preconditioner
                      DIC:
    tolerance
                      1e-06:
    relTol
                      0;
  pFinal
    $p;
    relTol
                      PBiCG:
    solver
    preconditioner
                      DILU;
    tolerance
                      1e-08:
    relTol
                      0;
    miniter
    maxIter
PISO
  nCorrectors 2:
  nNonOrthogonalCorrectors 1;
```

- Before solving an equation for a particular field, the initial residual is evaluated based on the current values of the field.
- After each solver iteration the residual is re-evaluated. The solver stops if either of the following conditions are reached:
 - The residual falls below the solver tolerance, tolerance.
 - The ratio of current to initial residuals falls below the solver relative tolerance, relTol.
 - The number of iterations exceeds a maximum number of iterations, **maxiter**.
- The solver tolerance should represent the level at which the residual is small enough that the solution can be deemed sufficiently accurate.
- The keyword maxiter is optional and the default value is 1000.
- The user can also define the minimum number of iterations using the keyword miniter. This keyword is optional and the default value is 0.

- These are the linear solvers available in OpenFOAM®:
 - GAMG
 - PBiCG
 - PBiCGStab
 - PCG
 - smoothSolver
 - diagonalSolver

- → Multigrid solver
- → Newton-Krylov solver
- → Newton-Krylov solver
- → Newton-Krylov solver
- → Smooth solver

- You will find the source code of the linear solvers in the following directory:
 - \$WM PROJECT DIR/src/OpenFOAM/matrices/lduMatrix/solvers

Linear solvers

- When using Newton-Krylov solvers, you need to define preconditoners.
- These are the preconditioners available in OpenFOAM®:

DIC

FDIC

diagonal

DILU

GAMG

noPreconditioner

- You will find the source code in the following directory:
 - \$WM_PROJECT_DIR/src/OpenFOAM/matrices/lduMatrix/preconditioners
- The **smoothSolver** solver requires the specification of a smoother.
- These are the smoothers available in OpenFOAM®:

DIC

DILUGaussSeidel

nonBlockingGaussSeidel

DICGaussSeidel

FDIC

symGaussSeidel

DILU

- GaussSeidel
- You will find the source code in the following directory:
 - \$WM PROJECT DIR/src/OpenFOAM/matrices/lduMatrix/smoothers

- As you can see, when it comes to linear solvers there are many options and combinations available in OpenFOAM®.
- When it comes to choosing the linear solver, there is no written theory.
- It is problem and hardware dependent (type of the mesh, physics involved, processor cache memory, network connectivity, partitioning method, and so on).
- Most of the times using the **GAMG** method (geometric-algebraic multi-grid), is the best choice for symmetric matrices (*e.g.*, pressure).
- The **GAMG** method should converge fast (less than 20 iterations). If it's taking more iterations, try to change the smoother.
- And if it is taking too long or it is unstable, use the PCG solver.
- When running with many cores (more than 1000), using the PCG might be a better choice.

- For asymmetric matrices, the PBiCGStab method with DILU preconditioner is a good choice.
- The **smoothSolver** solver with smoother **GaussSeidel**, also performs very well.
- If the PBiCGStab method with DILU preconditioner mysteriously crashed with an error related to the preconditioner, use the smoothSolver or change the preconditioner.
- But in general the PBiCGStab solver should be faster than the smoothSolver solver.
- Remember, asymmetric matrices are assembled from the velocity (**U**), and the transported quantities (**k**, **omega**, **epsilon**, **T**, and so on).
- Usually, computing the velocity and the transported quantities is inexpensive and fast, so it is a good idea to use a tight tolerance (1e-8) for these fields.
- The diagonal solver is used for back-substitution, for instance, when computing
 density using the equation of state (we know p and T).

- A few comments on the linear solvers residuals (we will talk about monitoring the residuals later on).
 - Residuals are not a direct indication that you are converging to the right solution.
 - The first time-steps the solution might not converge, this is acceptable.
 - Also, you might need to use a smaller time-step during the first iterations to maintain solver stability.
 - If the solution is not converging, try to reduce the time-step size.

```
Time = 50

Courant Number mean: 0.044365026 max: 0.16800273
smoothSolver: Solving for Ux, Initial residual = 1.0907508e-09, Final residual = 1.0907508e-09, No Iterations 0 smoothSolver: Solving for Uy, Initial residual = 1.4677462e-09, Final residual = 1.4677462e-09, No Iterations 0 DICPCG: Solving for p, Initial residual = 1.0020944e-06, Final residual = 1.0746895e-07, No Iterations 1 time step continuity errors: sum local = 4.0107145e-11, global = -5.0601748e-20, cumulative = 2.637831e-18 ExecutionTime = 4.47 s ClockTime = 5 s

fieldMinMax minmaxdomain output:
    min(p) = -0.37208345 at location (0.025 0.975 0.5)
    max(p) = 0.77640927 at location (0.975 0.975 0.5)
    min(U) = (0.00028445255 -0.00028138799 0) at location (0.025 0.025 0.5)
    max(U) = (0.00028445255 -0.00028138799 0) at location (0.025 0.025 0.5)

Residuals
```

Linear solvers

- So how do we set the tolerances?
- The pressure equation is particularly important, so we should resolve it accurately.
 Solving the pressure equation is the expensive part of the whole iterative process.
- For the pressure equation you can start the simulation with a tolerance equal to 1e-6
 and relTol equal to 0.01. After a while you change these values to 1e-6 and 0.0,
 respectively.
- If the solver is too slow, you can change the convergence criterion to 1e-4 and relTol
 equal to 0.05. You usually will do this during the first iterations.

Loose tolerance

```
p
{
    solver PCG;
    preconditioner DIC;
    tolerance 1e-6;
    relTol 0.01;
}
```

Tight tolerance

```
p
{
    solver PCG;
    preconditioner DIC;
    tolerance 1e-6;
    relTol 0.0;
}
```

Linear solvers

- For the velocity field (U) and the transported quantities (asymmetric matrices), you
 can use the following criterion.
- Solving for these variables is relative inexpensive, so you can start right away with a tight tolerance

Loose tolerance

```
U
{
    solver PBiCGStab;
    preconditioner DILU;
    tolerance 1e-8;
    relTol 0.01;
}
```

Tight tolerance

```
U
{
    solver PBiCGStab;
    preconditioner DILU;
    tolerance 1e-8;
    relTol 0.0;
}
```

- It is also a good idea to set the minimum number of iterations (miniter) to 3.
- If your solver is doing too many iterations, you can set the maximum number of iterations (maxIter).
- But be careful, if the solver reach the maximum number of iterations it will stop, we are talking about unconverged iterations.
- Setting the maximum number of iterations is specially useful during the first timesteps where the linear solver takes longer to converge.
- You can set miniter and maxiter in all symmetric and asymmetric linear solvers.

Linear solvers

- When you use the PISO or PIMPLE method, you also have the option to set the tolerance for the final pressure corrector step (pFinal).
- By proceeding in this way, you can put all the computational effort only in the last corrector step (**pFinal**).
- For all the intermediate corrector steps, you can use a more relaxed convergence criterion.
- For example, you can use the following solver and tolerance criterion for all the intermediate corrector steps (p), then in the final corrector step (pFinal) you tight the solver tolerance.

Loose tolerance for p

Tight tolerance for pFinal

- When you use the **PISO** or **PIMPLE** method, you also have the option to set the tolerance for the final pressure corrector step (**pFinal**).
- By proceeding in this way, you can put all the computational effort only in the last corrector step (**pFinal** in this case).
- For all the intermediate corrector steps (**p**), you can use a more relaxed convergence criterion.
- If you proceed in this way, it is recommended to do at least 2 corrector steps (nCorrectors).

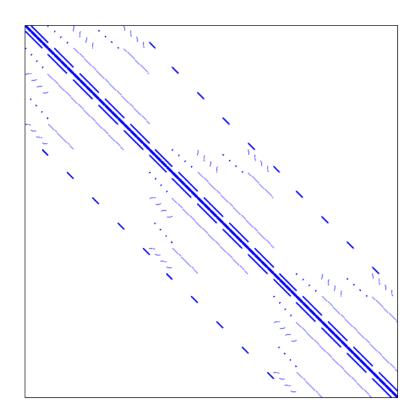
```
Courant Number mean: 0.10556573 max: 0.65793603
deltaT = 0.00097959184
Time = 10
PIMPLE: iteration 1
DILUPBICG: Solving for Ux, Initial residual = 0.0024649332, Final residual = 2.3403547e-09, No Iterations 4
DILUPBICG: Solving for Uy, Initial residual = 0.0044355904, Final residual = 1.8966277e-09, No Iterations 4
DILUPBiCG: Solving for Uz, Initial residual = 0.010100894, Final residual = 1.4724403e-09, No Iterations 4
      Solving for p, Initial residual = 0.018497918, Final residual = 0.00058090899, No Iterations 3
GAMG: Solving for p, Initial residual = 0.00058090857, Final residual = 2.5748489e-05, No Iterations 5
time step continuity errors : sum local = 1.2367812e-09, qlobal = 2.8865505e-11, cumulative = 1.057806e-08
GAMG: Solving for p, Initial residual = 0.00076032002, Final residual = 2.3965621e-05, No Iterations 3
GAMG: Solving for p, Initial residual = 2.3961044e-05, Final residual = 6.3151172e-06, No Iterations 2
time step continuity errors : sum local = 3.0345314e-10, global = -3.0075104e-12, cumulative = 1.0575052e-08
DILUPBiCG: Solving for omega, Initial residual = 0.00073937735, Final residual = 1.2839908e-10, No Iterations 4
DILUPBICG: Solving for k, Initial residual = 0.0018291502, Final residual = 8.5494234e-09, No Iterations 3
ExecutionTime = 29544.18 s ClockTime = 29600 s
```

- As we are solving a sparse matrix, the more diagonal the matrix is, the best the convergence rate will be.
- So it is highly advisable to use the utility renumberMesh before running the simulation.
 - \$> renumberMesh -overwrite

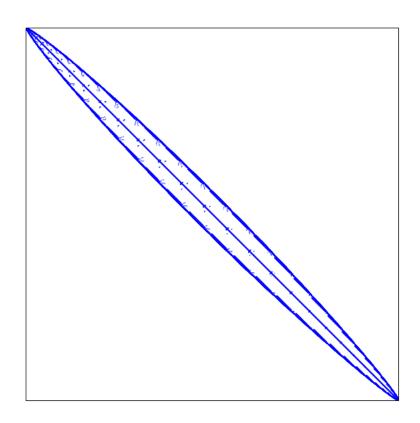
- The utility renumberMesh can dramatically increase the speed of the linear solvers, specially during the firsts iterations.
- You will find the source code and the master dictionary in the following directory:
 - \$WM_PROJECT_DIR/applications/utilities/mesh/manipulation/renumberMesh

Linear solvers

 The idea behind reordering is to make the matrix more diagonally dominant, therefore, speeding up the iterative solver.



Matrix structure plot before reordering



Matrix structure plot after reordering

Note:

This is the actual pressure matrix from an OpenFOAM® model case

On the multigrid solvers

- The development of multigrid solvers (GAMG in OpenFOAM®), together with the development of high resolution TVD schemes and parallel computing, are among the most remarkable achievements of the history of CFD.
- Most of the time using the GAMG linear solver is fine. However, if you see that the linear solver is taking too long to converge or is converging in more than 100 iterations, it is better to use the PCG linear solver.
- Particularly, we have found that the **GAMG** linear solver in OpenFOAM® does not perform very well when you scale your computations to more than 500 processors.
- Also, we have found that for some multiphase cases the PCG method outperforms the GAMG.
- But again, this is problem and hardware dependent.
- As you can see, you need to always monitor your simulations (stick to the screen for a while). Otherwise, you might end-up using a solver that is performing poorly. And this translate in increased computational time and costs.

On the multigrid solvers tolerances

 If you go for the GAMG linear solver for symmetric matrices (e.g., pressure), the following tolerances are acceptable for most of the cases.

Loose tolerance for p

```
solver
                            GAMG:
   tolerance
                            1e-6:
   relTol
                            0.01:
                            GaussSeidel;
   smoother
   nPreSweeps
   nPostSweeps
   cacheAgglomeration
                            on;
                            faceAreaPair;
   agglomerator
   nCellsInCoarsestLevel
                            100;
   mergeLevels
   minlter
}
```

Tight tolerance for pFinal

```
pFinal
   solver
                            GAMG:
   tolerance
                            1e-6:
   relTol
                            GaussSeidel;
   smoother
   nPreSweeps
   nPostSweeps
   cacheAgglomeration
                            on:
   agglomerator
                            faceAreaPair;
   nCellsInCoarsestLevel
                            100;
   mergeLevels
   minlter
```

NOTE:

The GAMG parameters are not optimized, that is up to you. Most of the times is safe to use the default parameters.

Linear solvers tolerances – Steady simulations

- The previous tolerances are fine for unsteady solver.
- For extremely coupled problems you might need to have tighter tolerances.
- You can use the same tolerances for steady solvers. However, it is acceptable to use a looser criterion.
- You can also set the convergence controls based on residuals of fields. The controls are specified in the **residualControls** sub-dictionary of the dictionary file fvSolution.

Exercises

- Choose any tutorial or a case of your own and do a benchmarking of the linear solvers.
- Using your benchmarking case, find the optimal parameters for the GAMG solver.
- Use different linear solvers for p and pFinal (symmetric matrices). Do you see any advantage?
- Do a benchmarking of the different reordering methods available
 - (Hint: look for the dictionary renumberMeshDict)
- Compare the performance of the asymmetric solvers PBiCG, PBiCGStab, and smoothSolver. Do you see any significant difference between both solvers?
- Is it possible to switch between segregated and coupled linear solvers on-the-fly?
- In what files are located the controls of the SIMPLE, PISO, and PIMPLE methods?

(Hint: for example, using grep look for the keyword nCorrectors in the directory src/finiteVolume)

Pressure-Velocity coupling in OpenFOAM®

 To solve the Navier-Stokes equations we need to use a solution approach able to deal with the nonlinearities of the governing equations and with the coupled set of equations.

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0,$$

$$\frac{\partial (\rho \mathbf{u})}{\partial t} + \nabla \cdot (\rho \mathbf{u} \mathbf{u}) = -\nabla p + \nabla \cdot \tau,$$

$$\frac{\partial (\rho e_t)}{\partial t} + \nabla \cdot (\rho e_t \mathbf{u}) = \nabla \cdot q - \nabla \cdot (p \mathbf{u}) + \tau : \nabla \mathbf{u},$$

$$+$$

Additional equations deriving from models, such as, volume fraction, chemical reactions, turbulence modeling, combustion, multi-species, etc.

Pressure-Velocity coupling in OpenFOAM®

- Many numerical methods exist to solve the Navier-Stokes equations, just to name a few:
 - Pressure-correction methods (Predictor-Corrector type).
 - SIMPLE, SIMPLEC, SIMPLER, PISO.
 - Projection methods.
 - Fractional step (operator splitting), MAC, SOLA.
 - Density-based methods and preconditioned solvers.
 - Riemann solvers, ROE, HLLC, AUSM+, ENO, WENO.
 - Artificial compressibility methods.
 - Artificial viscosity methods.
- The most widely used approaches for solving the NSE are:
 - Pressure-based approach (predictor-corrector).
 - Density-based approach.
- Historically speaking, the pressure-based approach was developed for low-speed incompressible flows, while the density-based approach was mainly developed for high-speed compressible flows.
- However, both methods have been extended and reformulated to solve and operate for a wide range of flow conditions beyond their original intent.

- Pressure-based methods are the default option in most of CFD solvers (OpenFOAM® included).
- Pressure-based methods are intrinsically implicit.
- Two pressure-based solution methods are generally available, namely:
 - Segregated method.
 - Coupled method.
- In the pressure-based approach the velocity field is obtained from the momentum equations (there is some mathematical manipulation involved).
- In the segregated algorithm, the individual governing equations for the primitive variables are solved one after another.
- The coupled approach solves the continuity, momentum, and energy equation simultaneously, that is, coupled together.
- The segregated algorithm is memory-efficient, since the discretized equations need only be stored in the memory one at a time.
- However, the solution convergence is relatively slow (in comparison to coupled solvers) as the
 equations are solved one at a time.
- In OpenFOAM®, you will find segregated pressure-based solvers.
- But coupled pressure-based solvers are under active development.

- In OpenFOAM®, you will find segregated pressure-based solvers.
- The following methods are available:
 - SIMPLE (Semi-Implicit Method for Pressure-Linked Equations)
 - SIMPLEC (SIMPLE Corrected/Consistent)
 - PISO (Pressure Implicit with Splitting Operators)
- Additionally, you will find something called PIMPLE, which is an hybrid between SIMPLE and PISO. This method is formulated for very large timesteps and pseudo-transient simulations.
- You will find the solvers in the following directory:
 - \$WM_PROJECT_DIR/applications/solvers

- In OpenFOAM®, the PISO and PIMPLE methods are formulated for unsteady simulations.
- Whereas, the SIMPLE and SIMPLEC methods are formulated for steady simulations.
- If conserving time is not a priority, you can use the PIMPLE method for pseudo transient simulations.
- The pseudo transient PIMPLE method is more stable than the SIMPLE method, but it has a higher computational cost.
- Depending on the method and solver you are using, you will need to define a specific sub-dictionary in the dictionary file fvSolution.
- For instance, if you are using the PISO method, you will need to specify the PISO sub-dictionary.
- And depending on the method, each sub-dictionary will have different entries.

On the origins of the methods

SIMPLE

• S. V. Patankar and D. B. Spalding, "A calculation procedure for heat, mass and momentum transfer in three-dimensional parabolic flows", Int. J. Heat Mass Transfer, 15, 1787-1806 (1972).

SIMPLE-C

• J. P. Van Doormaal and G. D. Raithby, "Enhancements of the SIMPLE method for predicting incompressible fluid flows", Numer. Heat Transfer, 7, 147-163 (1984).

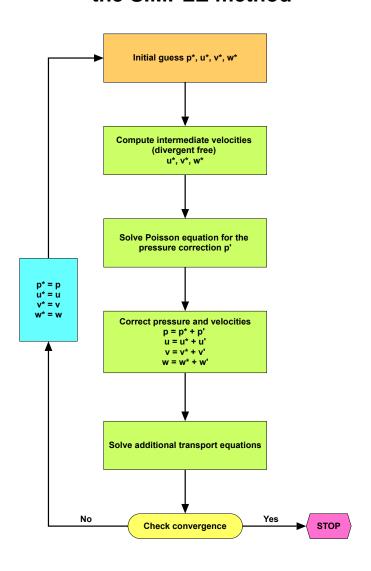
PISO

R. I. Issa, "Solution of the implicitly discretized fluid flow equations by operator-splitting", J. Comput. Phys., 62, 40-65 (1985).

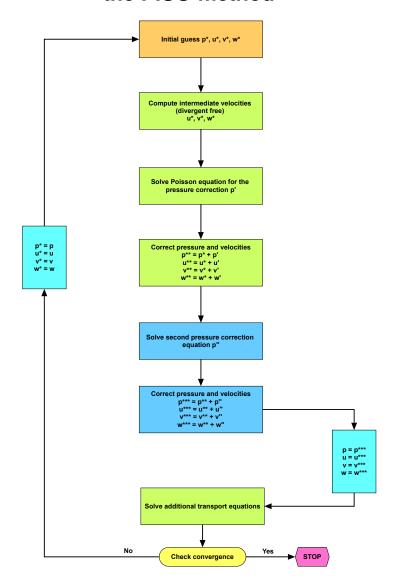
PIMPLE

- Unknown origins.
- Useful reference:
 - I. E. Barton, "Comparison of SIMPLE and PISO-type algorithms for transient flows, Int. J. Numerical methods in fluids, 26,459-483 (1998).

Pressure-velocity coupling using the SIMPLE method

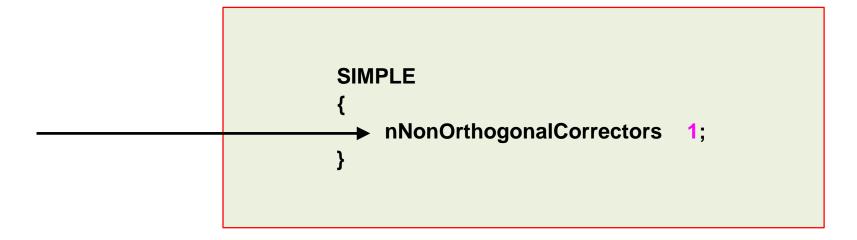


Pressure-velocity coupling using the PISO method



The SIMPLE sub-dictionary

- This sub-dictionary is located in the dictionary file fvSolution.
- It controls the options related to the **SIMPLE** pressure-velocity coupling method.
- The **SIMPLE** method only makes 1 correction.
- An additional correction to account for mesh non-orthogonality is available when using the SIMPLE method. The number of non-orthogonal correctors is specified by the nNonOrthogonalCorrectors keyword.
- The number of non-orthogonal correctors is chosen according to the mesh quality. For orthogonal meshes you can use 0, whereas, for non-orthogonal meshes it is recommended to do at least 1 correction.
- However, it is strongly recommended to do are least 1 non-orthogonal correction.



The SIMPLE sub-dictionary

- You can use the optional keyword consistent to enable or disable the SIMPLEC method.
- · This option is disable by default.
- In the **SIMPLEC** method, the cost per iteration is marginally higher but the convergence rate is better so the number of iterations can be reduce.
- The **SIMPLEC** method relaxes the pressure in a consistent manner and additional relaxation of the pressure is not generally necessary.
- In addition, convergence of the **p-U** system is better and still is reliable with less aggressive relaxation of the momentum equation.

The SIMPLE sub-dictionary

- Typical under-relaxation factors for the SIMPLE and SIMPLEC methods.
- Remember the under-relaxation factors are problem dependent.

SIMPLE

```
relaxationFactors
    fields
                   0.3;
         p
    equations
                   0.7;
                   0.7;
         omega
                   0.7:
```

SIMPLEC

```
relaxationFactors
    equations
         U
                   0.9;
                   0.9:
         omega
                   0.9;
```

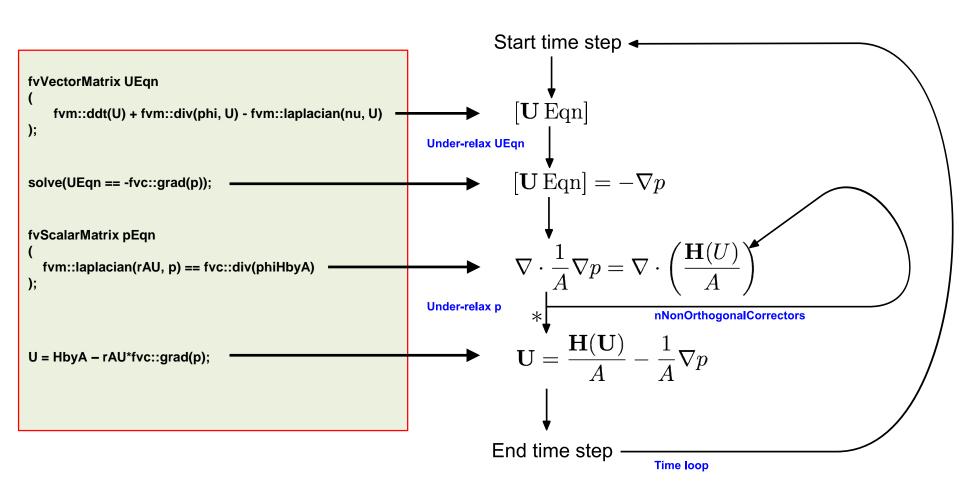
The SIMPLE sub-dictionary

- If you are planning to use the SIMPLEC method, we recommend you to use under-relaxation factors are little bit more loose that the commonly recommended values.
- If during the simulation you still have some stability problems, try to reduce all the values to 0.5.
- Remember the under-relaxation factors are problem dependent.
- It is also recommended to start the simulation with low values (about 0.3), and then increase the values slowly up to 0.7.

SIMPLEC

```
relaxationFactors
    fields
                    0.7;
    equations
                    0.7:
                    0.7:
                    0.7:
                    0.7:
          omega
```

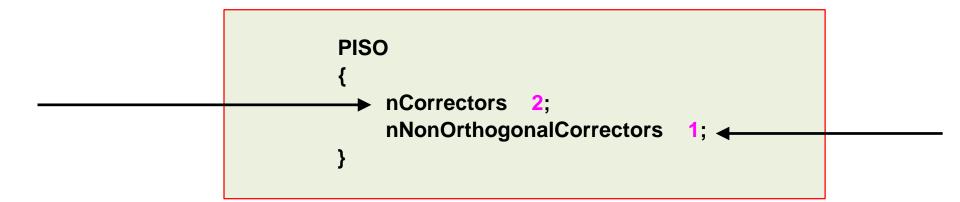
The SIMPLE loop in OpenFOAM®



*
$$\phi = \mathbf{S}_f \cdot [(\mathbf{H}/A)_f - (1/A)_f (\nabla p)_f]$$

The PISO sub-dictionary

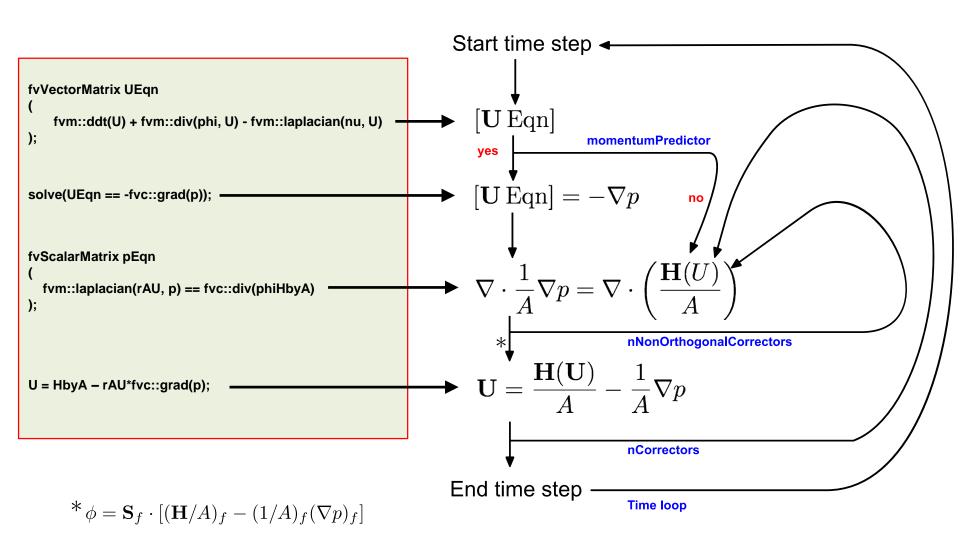
- This sub-dictionary is located in the dictionary file fvSolution.
- It controls the options related to the **PISO** pressure-velocity coupling method.
- The **PISO** method requires at least one correction (**nCorrectors**). To improve accuracy and stability you can increase the number of corrections.
- For good accuracy and stability, it is recommended to use 2 nCorrectors.
- An additional correction to account for mesh non-orthogonality is available when using the PISO method. The number of non-orthogonal correctors is specified by the nNonOrthogonalCorrectors keyword.
- The number of non-orthogonal correctors is chosen according to the mesh quality. For orthogonal meshes you can use 0, whereas, for non-orthogonal meshes it is recommended to do at least 1 correction.



The PISO sub-dictionary

- You can use the optional keyword momentumPredictor to enable or disable the momentum predictor step.
- The momentum predictor helps in stabilizing the solution as we are computing better approximations for the velocity.
- It is clear that this will add an extra computational cost, which is negligible.
- In most of the solvers, this option is enabled by default.
- It is recommended to use this option for highly convective flows (high Reynolds number). If you are working with low Reynolds flow or creeping flows it is recommended to turn it off.
- Remember, when you enable the option **momentumPredictor**, you will need to define the linear solvers for the variables **.*Final** (we are using regex notation).

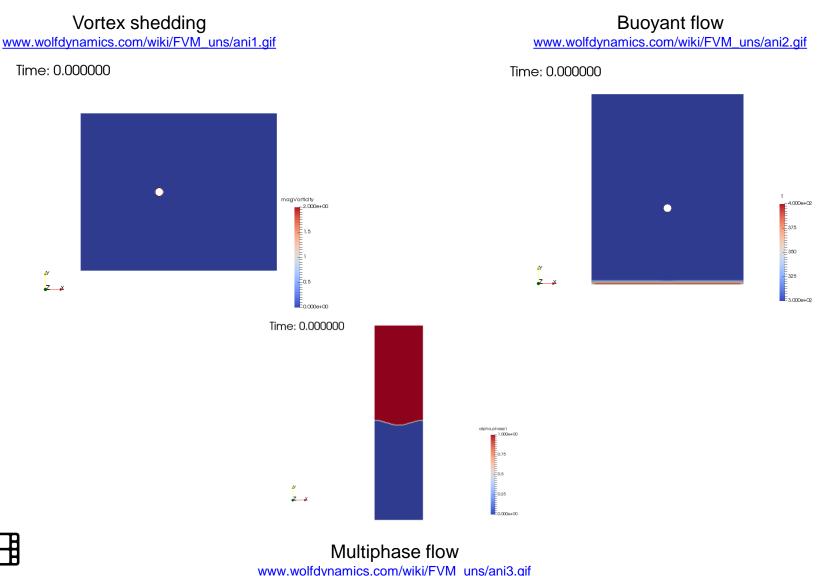
The PISO loop in OpenFOAM®



The PIMPLE sub-dictionary

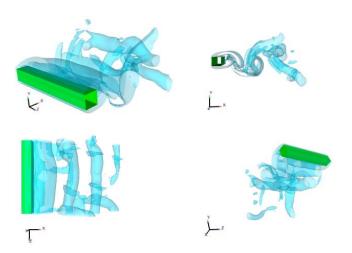
- This sub-dictionary is located in the dictionary file fvSolution.
- It controls the options related to the **PIMPLE** pressure-velocity coupling method.
- The PIMPLE method works very similar to the PISO method. In fact, setting the keyword nOuterCorrectors to 1 is equivalent to running using the PISO method.
- The keyword **nOuterCorrectors** controls a loop outside the **PISO** loop.
- To gain more stability, especially when using large time-steps, you can use more outer correctors (**nOuterCorrectors**). Have in mind that this will highly increase the computational cost.
- Also, if you use under-relaxation factors, your solution is not anymore time accurate. You are working in pseudo-transient simulations mode.

A few examples of unsteady applications:



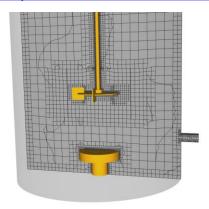
A few examples of unsteady applications:

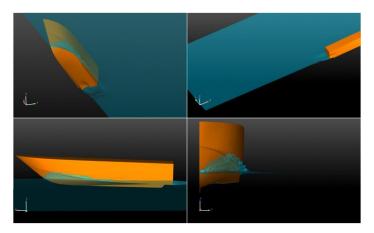
Turbulent flows - SRS www.wolfdynamics.com/wiki/FVM_uns/ani4.gif



Sliding grids – Continuous stirred tank reactor

www.wolfdynamics.com/wiki/FVM_uns/ani5.gif



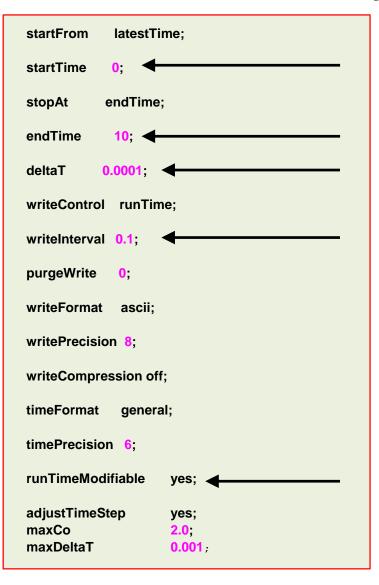




Marine applications - Sea keeping www.wolfdynamics.com/wiki/FVM_uns/ani6.gif

- Nearly all flows in nature and industrial applications are unsteady (also known as transient or time-dependent).
- Unsteadiness is due to:
 - Instabilities.
 - Non-equilibrium initial conditions.
 - Time-dependent boundary conditions.
 - Source terms.
 - Chemical reactions.
 - Moving or deforming bodies.
 - Turbulence.
 - Buoyancy.
 - Convection.
 - Multiple phases

- Select the time step. The time-step must be chosen in such a way that it resolves the time-dependent features and maintains solver stability.
- Select the temporal discretization scheme.
- Set the tolerance (absolute and/or relative) of the linear solvers.
- Monitor the CFL number.
- Monitor the stability and boundedness of the solution.
- Monitor a quantity of interest.
- And of course, you need to save the solution with a given frequency.
- Have in mind that unsteady simulations generate a lot of data.
- End time of the simulation?, it is up to you.
- In the controlDict dictionary you need to set runtime parameters and general instructions on how to run the case (such as time step and maximum CFL number). You also set the saving frequency.
- In the fvSchemes dictionary you need to set the temporal discretization scheme.
- In the fvSolution dictionary you need to set the linear solvers.
- Also, you will need to set the number of corrections of the velocity-pressure coupling method used (e.g. PISO or PIMPLE), this is done in the fvSolution dictionary.
- Additionally, you may set **functionObjects** in the *controlDict* dictionary. The **functionObjects** are used to do sampling, probing and co-processing while the simulation is running.



- The controlDict dictionary contains runtime simulation controls, such as, start time, end time, time step, saving frequency and so on. Most of the entries are self-explanatory.
- This generic case starts from time 0 (**startTime**), and it will run up to 10 seconds (**endTime**).
- It will write the solution every 0.1 seconds (**writeInterval**) of simulation time (**runTime**).
- The time step of the simulation is 0.0001 seconds (**deltaT**).
- It will keep all the solution directories (purgeWrite).
- It will save the solution in ascii format (writeFormat) with a precision of 8 digits (writePrecision).
- And as the option **runTimeModifiable** is on (**yes**), we can modify all these entries while we are running the simulation.

```
startFrom
             latestTime;
startTime
            0;
stopAt
           endTime;
endTime
            10;
           0.0001;
deltaT
writeControl runTime;
writeInterval 0.1;
purgeWrite
             0;
writeFormat ascii:
writePrecision 8:
writeCompression off;
timeFormat
             general;
timePrecision 6;
runTimeModifiable
                     yes;
adjustTimeStep
                     yes;
maxCo
maxDeltaT
                     0.001
```

- In this generic case, the solver supports adjustable time-step (adjustTimeStep).
- The option adjustTimeStep will automatically adjust the time step to achieve the maximum desired courant number (maxCo) or time-step size (maxDeltaT).
- When any of these conditions is reached, the solver will stop scaling the time-step size.
- Remember, the first time step of the simulation is done using the value defined with the keyword deltaT and then it is automatically scaled (up or down), to achieve the desired maximum values (maxCo and maxDeltaT).
- It is recommended to start the simulation with a low time-step in order to let the solver scale-up the time-step size.
- The feature adjustTimeStep is only present in the PIMPLE family solvers, but it can be added to any solver by modifying the source code.
- If you are planning to use large time steps (CFL much higher than 1), it is recommended to do at least 3 correctors steps (nCorrectors) in PISO/PIMPLE loop.

```
startFrom
             latestTime;
startTime
            0;
stopAt
           endTime;
endTime
            10:
           0.0001;
deltaT
writeControl adjustableRunTime;
writeInterval 0.1;
purgeWrite
             0;
writeFormat ascii:
writePrecision 8:
writeCompression off;
timeFormat
             general;
timePrecision 6;
runTimeModifiable
                     yes;
adjustTimeStep
                     yes;
maxCo
maxDeltaT
                     0.001
```

- A word of caution about adjustable time-step (adjustTimeStep).
- This option will automatically adjust the time step to achieve the maximum desired courant number (maxCo) or time-step size (maxDeltaT).
- If the maxDeltaT condition is not reached, the solver will adapt
 the time-step to achieve the target maxCo, and as the time-step
 is not fixed this might introduce spurious oscillations in the
 solution.
- It is recommended to use this option at the beginning of the simulation and as soon as the solution stabilizes try fixed the time-step.
- Also, try to avoid using adjustable time step together with the option adjustableRunTime.
- The option adjustableRunTime will adjust the time-step to save the solution at the precise write intervals, and this might introduce numerical oscillations due to the fact that the time-step is changing.
- Also, the fact that you are using an adaptive time-step can have a negative effect when doing signal analysis.

```
\partial \phi
ddtSchemes <
                                             \partial t
                 backward;
     default
gradSchemes
                 Gauss linear:
     default
     grad(p)
                  Gauss linear:
divSchemes
     default
                  none;
     div(phi,U)
                  Gauss linear:
laplacianSchemes
                 Gauss linear orthogonal;
     default
interpolationSchemes
     default
                 linear;
snGradSchemes
                 orthogonal;
     default
```

- The fvSchemes dictionary contains the information related to time discretization and spatial discretization schemes.
- In this generic case we are using the backward method for time discretization (ddtSchemes).
- This scheme is second order accurate but oscillatory.
- The parameters can be changed on-the-fly.

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC:
    tolerance
                     1e-06:
    relTol
                     0;
  pFinal
    $p;
    relTol 0:
  "U.*"
                     smoothSolver;
    solver
                     symGaussSeidel;
    smoother
                     1e-08:
    tolerance
    relTol
                     0;
PIMPLE
  nOuterCorrectors 1;
  nCorrectors 2:
  nNonOrthogonalCorrectors 1;
```

- The fvSolution dictionary contains the instructions of how to solve each discretized linear equation system.
- As for the controlDict and fvSchemes dictionaries, the parameters can be changed on-the-fly.
- In this generic case, to solve the pressure (**p**) we are using the **PCG** method with the **DIC** precondtioner, an absolute **tolerance** equal to 1e-06 and a relative tolerance **relTol** equal to 0.
- The entry pFinal refers to the final pressure correction (notice that we are using macro syntax), and we are using a relative tolerance relTol equal to 0.
- To solve U and UFinal (U.* using regex), we are using the smoothSolver method with an absolute tolerance equal to 1e-08 and a relative tolerance relTol equal to 0.
- The solvers will iterative until reaching any of the tolerance values set by the user or reaching a maximum value of iterations (optional entry).
- FYI, solving for the velocity is relative inexpensive, whereas solving for the pressure is expensive.

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC;
    tolerance
                      1e-06:
    relTol
  pFinal
    $p;
    relTol 0:
  "U.*"
                     smoothSolver;
    solver
                     symGaussSeidel;
    smoother
    tolerance
                      1e-08:
    relTol
PIMPLE
  nOuterCorrectors 1;
  nCorrectors 2:
  nNonOrthogonalCorrectors 1;
```

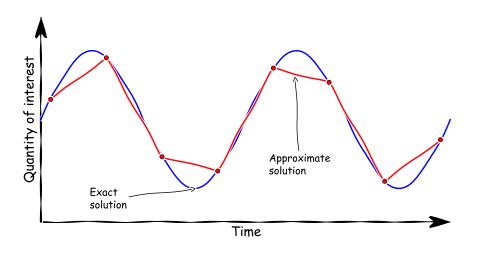
- The fvSolution dictionary also contains the PIMPLE and PISO sub-dictionaries.
- The **PIMPLE** sub-dictionary contains entries related to the pressure-velocity coupling method (the **PIMPLE** method).
- Setting the keyword nOuterCorrectors to 1 is equivalent to running using the PISO method.
- Remember, you need to do at least one PISO loop (nCorrectors).
- To gain more stability, especially when using large time-steps, you can use more outer correctors (**nOuterCorrectors**).
- Adding corrections increase the computational cost (nOuterCorrectors and nCorrectors).
- In this generic case, we are using 1 outer correctors (nOuterCorrectors), 2 inner correctors or PISO correctors (nCorrectors), and 1 correction due to non-orthogonality (nNonOrthogonalCorrectors).
- If you are using large time steps (CFL much higher than 1), it is recommended to do at least 3 correctors steps (nCorrectors) in PISO/PIMPLE loop.

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC;
    tolerance
                      1e-06:
    relTol
                     0;
  pFinal
    $p;
    relTol 0:
  U
                     smoothSolver;
    solver
                     symGaussSeidel;
    smoother
                     1e-08:
    tolerance
    relTol
                      0;
PISO
  nCorrectors 2;
  nNonOrthogonalCorrectors 1;
```

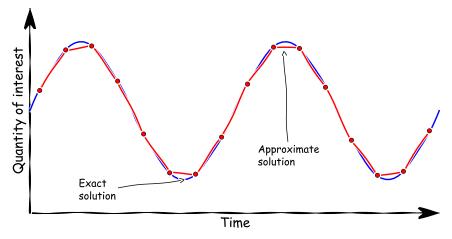
- If you use the **PISO** method for pressure-velocity coupling, you will need to define the **PISO** sub-dictionary.
- In this generic case we are doing two PISO corrections and one orthogonal correction.
- You need to do at least one PISO loop (nCorrectors).
- If you are using large time steps (CFL much higher than 1), it is recommended to do at least 3 correctors steps (nCorrectors) in PISO/PIMPLE loop.

How to choose the time-step in unsteady simulations and monitor the solution

Remember, when running unsteady simulations the time-step must be chosen in such a way that it resolves the time-dependent features and maintains solver stability.



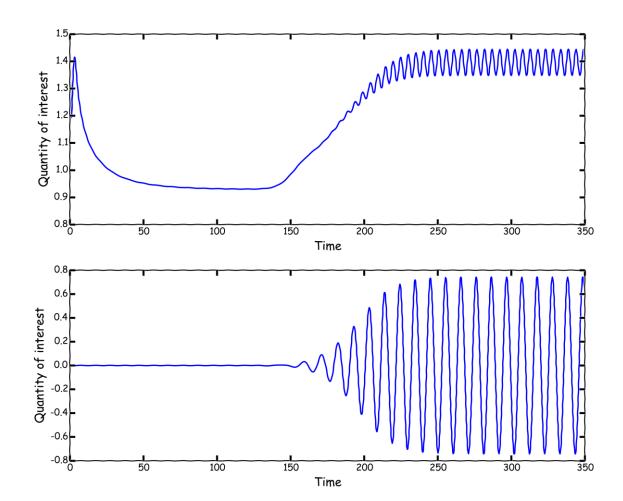
When you use large time steps you do not resolve well the physics



By using a smaller time step you resolve better the physics and you gain stability

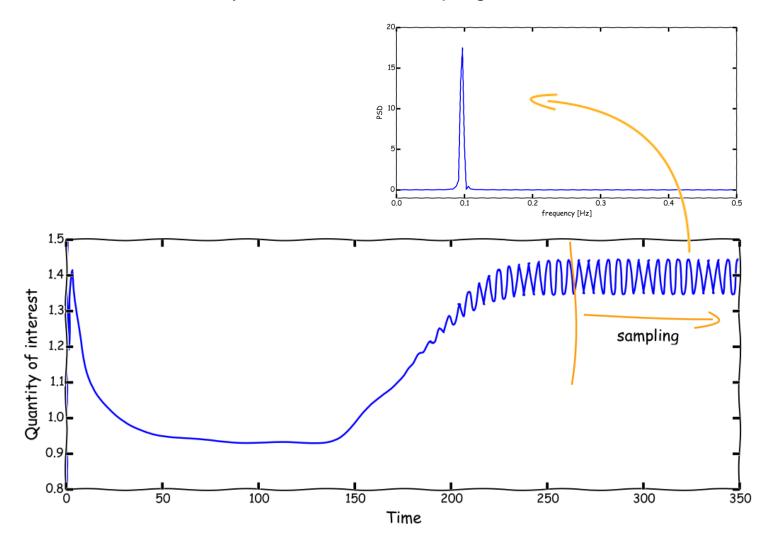
Monitoring unsteady simulations

- When running unsteady simulations, it is highly advisable to monitor a quantity of interest.
- The quantity of interest can fluctuate in time, this is an indication of unsteadiness.



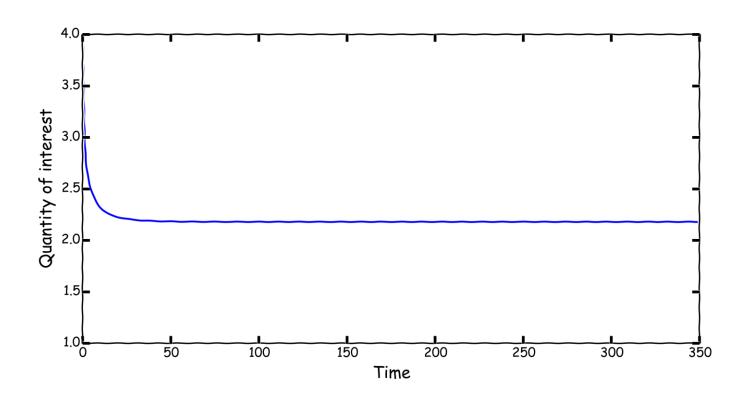
Sampling unsteady simulations

Remember to choose wisely where to do the sampling.



I am running an unsteady simulations and the QOI does not change

- When you run unsteady simulations, flow variables can stop changing with time. When this happens, we say we have arrived to a steady state.
- Remember this is the exception rather than the rule.
- If you use a steady solver, you will arrive to the same solution (maybe not), in much less iterations.



What about steady simulations?

- First of all, steady simulations are a big simplification of reality.
- Steady simulations is a trick used by CFDers to get fast outcomes with results that might be even more questionable.
- As mentioned before, most of the flows you will encounter are unsteady.
- In steady simulations we made two assumptions:
 - We ignore unsteady fluctuations. That is, we neglect the temporal derivative in the governing equations.
 - We perform time averaging when dealing with stationary turbulence (RANS modeling)
- The advantage of steady simulations is that they require low computational resources, give fast outputs, and are easier to post-process and analyze.
- In OpenFOAM® is possible to run steady simulation.
- To do so, you need to use the appropriate solver and use the right discretization scheme.
- As you are not solving the temporal derivative, you do not need to set the time step. However, you need to tell OpenFOAM® how many iterations you would like to run.
- You can also set the residual controls (residualControl), in the fvSolution dictionary file. You set the residualControl in the SIMPLE sub-dictionary.
- If you do not set the residual controls, OpenFOAM® will run until reaching the maximum number of iterations (endTime).

What about steady simulations?

- You also need to set the under-relaxation factors.
- The under-relaxation factors control the change of the variable ϕ .

$$\phi_P^n = \phi_P^{n-1} + \alpha(\phi_P^{n^*} - \phi_P^{n-1})$$

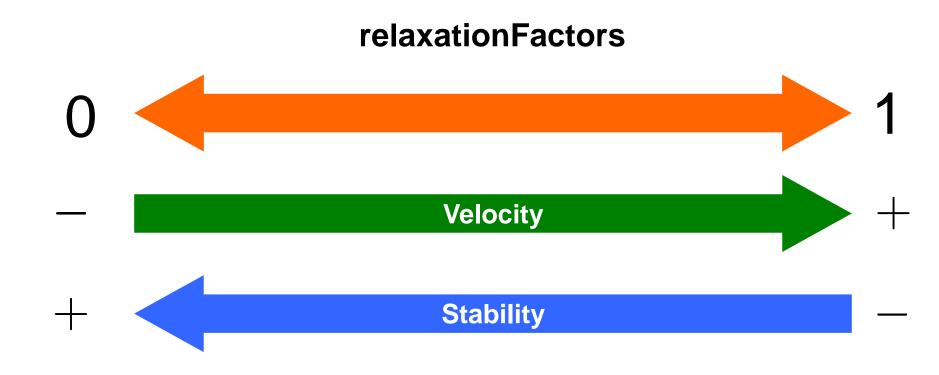
- If lpha < 1 we are using under-relaxation.
- Under-relaxation is a feature typical of steady solvers using the SIMPLE method.
- If you do not set the under-relaxation factors, OpenFOAM® will use the default hard-wired values (1.0 for all field variables or no under-relaxation).
- These are the under-relaxation factors commonly used with SIMPLE (industry standard),

р	0.3;
U	0.7;
k	0.7;
omega	0.7;
epsilon	0.7;

- According to the physics involved you will need to add more under-relaxation factors.
- Finding the right under-relaxation factors involved experience and a lot of trial and error.

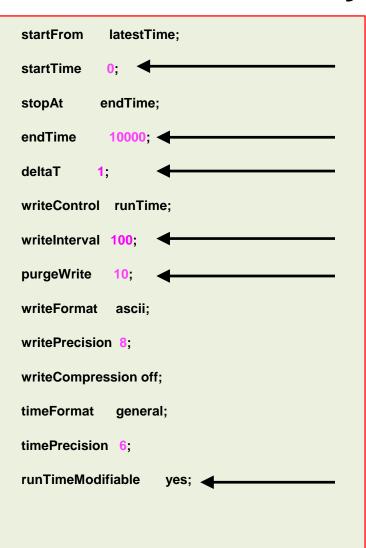
What about steady simulations?

The under-relaxation factors are bounded between 0 and 1.



 Selecting the under-relaxation factors it is kind of equivalent to selecting the right time step.

- In the controlDict dictionary you need to set runtime parameters and general instructions on how to run the case (such as the number of iterations to run). You also set the saving frequency.
- In the fvSchemes dictionary you need to set the temporal discretization scheme, for steady simulations it must be steadyState.
- In the fvSolution dictionary you need to set the linear solvers, underrelaxation factors and residual controls.
- Also, you will need to set the number of corrections of the velocity-pressure coupling method used (e.g. SIMPLE), this is done in the fvSolution dictionary.
- Additionally, you may set functionObjects in the controlDict dictionary.
 The functionObjects are used to do sampling, probing and co-processing while the simulation is running.



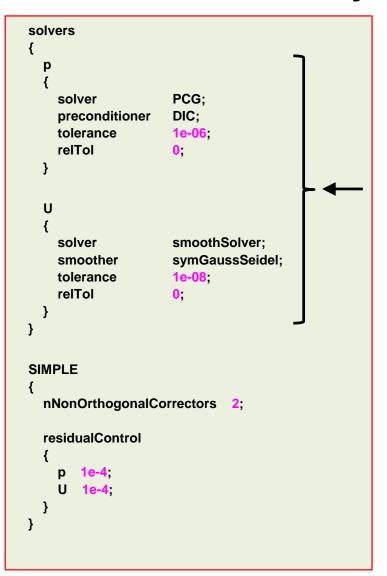
- The controlDict dictionary contains runtime simulation controls, such as, start time, end time, time step, saving frequency and so on. Most of the entries are self-explanatory.
- As we are doing a steady simulation, let us talk about iterations instead of time (seconds).
- This generic case starts from iteration 0 (**startTime**), and it will run up to 10000 iterations (**endTime**).
- It will write the solution every 100 iterations (**writeInterval**) of simulation time (**runTime**).
- It will advance the solution one iteration at a time (deltaT).
- It will keep the last 10 saved solutions (purgeWrite).
- It will save the solution in ascii format (writeFormat) with a precision of 8 digits (writePrecision).
- And as the option runTimeModifiable is on (true), we can
 modify all these entries while we are running the simulation.

```
ddtSchemes
     default
gradSchemes
                Gauss linear:
     default
     grad(p)
                 Gauss linear:
divSchemes
     default
                none:
                bounded Gauss linear:
     div(phi,U)
laplacianSchemes
                Gauss linear orthogonal;
     default
interpolationSchemes
     default
                linear;
snGradSchemes
     default
                orthogonal;
```

- The fvSchemes dictionary contains the information related to time discretization and spatial discretization schemes.
- In this generic case and as we are interested in using a steady solver, we are using the **steadyState** method for time discretization (**ddtSchemes**).
- It is not a good idea to switch between steady and unsteady schemes on-the-fly.
- For steady state cases, the bounded form can be applied to the divSchemes, in this case, div(phi,U) bounded Gauss linear.
- This adds a linearized, implicit source contribution to the transport equation of the form,

$$\frac{\partial \mathbf{u}}{\partial t} + \nabla \cdot (\mathbf{u}\mathbf{u}) - (\nabla \cdot \mathbf{u})\mathbf{u} = \nabla \cdot (\Gamma \nabla \mathbf{u}) + S$$

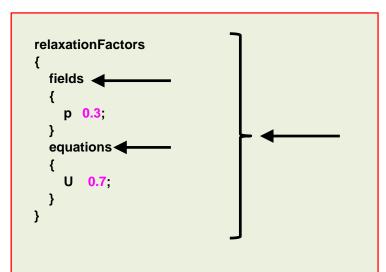
- This term removes a component proportional to the continuity error. This acts as a convergence aid to tend towards a bounded solution as the calculation proceeds.
- At convergence, this term becomes zero and does not contribute to the final solution.



- The fvSolution dictionary contains the instructions of how to solve each discretized linear equation system.
- As for the controlDict and fvSchemes dictionaries, the parameters can be changed on-the-fly.
- In this generic case, to solve the pressure (p) we are using the PCG method with the DIC preconditioner, an absolute tolerance equal to 1e-06 and a relative tolerance relTol equal to 0.
- To solve U we are using the smoothSolver method with an absolute tolerance equal to 1e-08 and a relative tolerance relTol equal to 0.
- The solvers will iterative until reaching any of the tolerance values set by the user or reaching a maximum value of iterations (optional entry).

```
solvers
                     PCG:
    solver
    preconditioner
                     DIC:
    tolerance
                     1e-06:
    relTol
                     smoothSolver:
    solver
                     symGaussSeidel:
    smoother
                     1e-08:
    tolerance
    relTol
                     0:
SIMPLE
 nNonOrthogonalCorrectors 2;
 residualControl
      1e-4:
      1e-4:
```

- The *fvSolution* dictionary also contains the **SIMPLE** subdictionary.
- The **SIMPLE** sub-dictionary contains entries related to the pressure-velocity coupling method (the **SIMPLE** method).
- Increasing the number of nNonOrthogonalCorrectors
 corrections will add more stability but at a higher computational
 cost.
- Remember, nNonOrthogonalCorrectors is used to improve the gradient computation due to mesh quality.
- In this generic case, we are doing 2 corrections due to nonorthogonality (nNonOrthogonalCorrectors).
- The SIMPLE sub-dictionary also contains convergence controls based on residuals of fields. The controls are specified in the residualControls sub-dictionary.
- The user needs to specify a tolerance for one or more solved fields and when the residual for every field falls below the corresponding residual, the simulation terminates.
- If you do not set the **residualControls**, the solver will iterate until reaching the maximum number of iterations set in the <code>controlDict</code> dictionary.



- The *fvSolution* dictionary also contains the **relaxationFactors** sub-dictionary.
- The relaxationFactors sub-dictionary which controls underrelaxation, is a technique used for improving stability when using steady solvers.
- Under-relaxation works by limiting the amount which a variable changes from one iteration to the next, either by modifying the solution matrix and source (equations keyword) prior to solving for a field or by modifying the field directly (fields keyword).
- An optimum choice of under-relaxation factors is one that is small enough to ensure stable computation but large enough to move the iterative process forward quickly.
- These are the under-relaxation factors commonly used (SIMPLE),

Steady simulations vs. Unsteady simulations

- Steady simulations require less computational power than unsteady simulations.
- They are also much faster than unsteady simulations.
- But sometimes they do not converge to the right solution.
- They are easier to post-process and analyze (you just need to take a look at the last saved solution).
- You can use the solution of an unconverged steady simulation as initial conditions for an unsteady simulation.
- Remember, steady simulations are not time accurate, therefore we can not use them to compute temporal statistics or compute the shedding frequency

