

Chapter 3

Harmonically Excited Vibration

(3.1) (a) $\delta = \frac{W}{k} = \frac{50}{4000} = 0.0125 \text{ m}$

(b) $\delta_{st} = \frac{F_0}{k} = \frac{60}{4000} = 0.015 \text{ m}$

(c) $\omega_n = \sqrt{\frac{k}{m}} = \left(\frac{4000 \times 9.81}{50} \right)^{1/2} = 28.0143 \text{ rad/sec}$

$\omega = 6 \text{ Hz} = 37.6992 \text{ rad/sec}$

$$X = \delta_{st} \left| \frac{1}{1 - \left(\frac{\omega}{\omega_n} \right)^2} \right| = 0.015 \left| \frac{1}{1 - \left(\frac{37.6992}{28.0143} \right)^2} \right| = 0.0185 \text{ m}$$

(3.2) $T_b = \frac{2\pi}{\omega_n - \omega} = \frac{2\pi}{2\pi(40.0 - 39.8)} = 5 \text{ sec}$

$k = 4000 \text{ N/m}, m = 10 \text{ kg}, F(t) = 400 \cos 10t \text{ N}$

(3.3) $F_0 = 400 \text{ N}, \omega = 10 \text{ rad/s}$

$\omega_n = \sqrt{\frac{k}{m}} = 20 \text{ rad/s}, \frac{\omega}{\omega_n} = \frac{10}{20} = 0.5 < 1$

Response is given by Eq. (3.9):

$$x(t) = \left(x_0 - \frac{F_0}{k - m\omega^2} \right) \cos \omega_n t + \left(\frac{\dot{x}_0}{\omega_n} \right) \sin \omega_n t + \frac{F_0}{k - m\omega^2} \cos \omega t \quad (\text{E.1})$$

(a) $x_0 = 0.1, \dot{x}_0 = 0 :$

Eg. (E.1) becomes

$$x(t) = \left\{ 0.1 - \frac{400}{4000 - 10(100)} \right\} \cos 20t + \frac{400}{4000 - 10(100)} \cos 10t \\ = -0.033333 \cos 20t + 0.133333 \cos 10t \quad (\text{E.2})$$

(b) $x_0 = 0, \dot{x}_0 = 10 :$

Eg. (E.1) becomes

$$x(t) = \left\{ 0 - \frac{400}{4000 - 10(100)} \right\} \cos 20t + \frac{10}{20} \sin 20t \\ + \left\{ \frac{400}{4000 - 10(100)} \right\} \cos 10t \\ = -0.133333 \cos 20t + 0.5 \sin 20t + 0.133333 \cos 10t \quad (\text{E.3})$$

$$(c) x_0 = 0.1, \dot{x}_0 = 10:$$

Eg. (E.1) becomes

$$\begin{aligned} x(t) &= \left\{ 0.1 - \frac{400}{4000 - 10(100)} \right\} \cos 20t + \frac{10}{20} \sin 20t \\ &\quad + \left\{ \frac{400}{4000 - 10(100)} \right\} \cos 10t \\ &= -0.033333 \cos 20t + 0.5 \sin 20t + 0.133333 \cos 10t \end{aligned} \quad (E.4)$$

$$3.4 \quad k = 4000 \text{ N/m}, m = 10 \text{ kg}, F(t) = 400 \cos 20t \text{ N},$$

$$F_0 = 400 \text{ N}, \omega = 20 \text{ rad/s}$$

$$\omega_n = \sqrt{\frac{k}{m}} = 20 \text{ rad/s}, \frac{\omega}{\omega_n} = \frac{20}{20} = 1$$

Response is given by Eg. (3.15):

$$x(t) = x_0 \cos \omega_n t + \frac{\dot{x}_0}{\omega_n} \sin \omega_n t + \frac{\delta_{st} \omega_n t}{2} \sin \omega_n t \quad (E.1)$$

$$\text{where } \delta_{st} = F_0/k = 400/4000 = 0.1$$

$$(a) x_0 = 0.1, \dot{x}_0 = 0:$$

Eg. (E.1) gives

$$\begin{aligned} x(t) &= 0.1 \cos 20t + \frac{(0.1)(20)t}{2} \sin 20t \\ &= 0.1 \cos 20t + t \sin 20t \end{aligned} \quad (E.2)$$

$$(b) x_0 = 0, \dot{x}_0 = 10:$$

Eg. (E.1) gives

$$\begin{aligned} x(t) &= \frac{10}{20} \sin 20t + \frac{(0.1)(20)t}{2} \sin 20t \\ &= 0.5 \sin 20t + t \sin 20t \end{aligned} \quad (E.3)$$

$$(c) x_0 = 0.1, \dot{x}_0 = 10:$$

Eg. (E.1) gives

$$\begin{aligned} x(t) &= 0.1 \cos 20t + \frac{10}{20} \sin 20t + \frac{(0.1)(20)t}{2} \sin 20t \\ &= 0.1 \cos 20t + 0.5 \sin 20t + t \sin 20t \end{aligned} \quad (E.4)$$

(3.5) $k = 4000 \text{ N/m}$, $m = 10 \text{ kg}$, $F(t) = 400 \cos 20.1t \text{ N}$
 $F_0 = 400 \text{ N}$, $\omega = 20.1 \text{ rad/s}$, $\omega^2 = 404.01 \text{ (rad/s)}^2$
 $\omega_n = \sqrt{\frac{k}{m}} = 20 \text{ rad/s}$

Solution is given by Eq. (3.9):

$$x(t) = \left(x_0 - \frac{F_0}{k-m\omega^2} \right) \cos \omega_n t + \frac{\dot{x}_0}{\omega_n} \sin \omega_n t + \left(\frac{F_0}{k-m\omega^2} \right) \cos \omega t \quad (\text{E.1})$$

(a) $x_0 = 0.1$, $\dot{x}_0 = 0$:

Eq. (E.1) reduces to

$$\begin{aligned} x(t) &= \left\{ 0.1 - \frac{400}{4000 - 10(404.01)} \right\} \cos 20t \\ &\quad + \left\{ \frac{400}{4000 - 10(404.01)} \right\} \cos 20.1t \\ &= 10.075062 \cos 20t - 9.975062 \cos 20.1t \end{aligned} \quad (\text{E.2})$$

(b) $x_0 = 0$, $\dot{x}_0 = 10$:

Eq. (E.1) reduces to

$$\begin{aligned} x(t) &= - \left\{ \frac{400}{4000 - 10(404.01)} \right\} \cos 20t + \frac{10}{20} \sin 20t \\ &\quad + \left\{ \frac{400}{4000 - 10(404.01)} \right\} \cos 20.1t \\ &= 9.975062 \cos 20t + 0.5 \sin 20t \\ &\quad - 9.975062 \cos 20.1t \end{aligned} \quad (\text{E.3})$$

(c) $x_0 = 0.1$, $\dot{x}_0 = 10$:

Eq. (E.1) gives

$$\begin{aligned} x(t) &= \left\{ 0.1 - \frac{400}{4000 - 10(404.01)} \right\} \cos 20t + \frac{10}{20} \sin 20t \\ &\quad + \left\{ \frac{400}{4000 - 10(404.01)} \right\} \cos 20.1t \end{aligned}$$

$$= 10.075062 \cos 20t + 0.5 \sin 20t \\ - 9.975062 \cos 30t \quad (E.4)$$

(3.6) $k = 4000 \text{ N/m}, m = 10 \text{ kg}, F(t) = 400 \cos 30t \text{ N}$
 $F_0 = 400 \text{ N}, \omega = 30 \text{ rad/s}, \omega^2 = 900 \text{ (rad/s)}^2$
 $\omega_n = \sqrt{\frac{k}{m}} = 20 \text{ rad/s}, \frac{\omega}{\omega_n} = 1.5 > 1$

Solution is given by Eq. (3.9):

$$x(t) = \left(x_0 - \frac{F_0}{k-m\omega^2} \right) \cos \omega_n t + \frac{\dot{x}_0}{\omega_n} \sin \omega_n t \\ + \left(\frac{F_0}{k-m\omega^2} \right) \cos \omega t \quad (E.1)$$

(a) $x_0 = 0.1, \dot{x}_0 = 0:$

Eq. (E.1) yields

$$x(t) = \left\{ 0.1 - \frac{400}{4000 - 10(900)} \right\} \cos 20t + \frac{400}{4000 - 10(900)} \cos 30t \\ = 0.18 \cos 20t - 0.08 \cos 30t \quad (E.2)$$

(b) $x_0 = 0, \dot{x}_0 = 10:$

Eg. (E.1) yields:

$$x(t) = - \left(\frac{400}{4000 - 10(900)} \right) \cos 20t + \frac{10}{20} \sin 20t \\ + \left(\frac{400}{4000 - 10(900)} \right) \cos 30t \\ = 0.08 \cos 20t + 0.5 \sin 20t - 0.08 \cos 30t \quad (E.3)$$

(c) $x_0 = 0.1, \dot{x}_0 = 10:$

Eg. (E.1) yields

$$x(t) = \left\{ 0.1 - \frac{400}{4000 - 10(900)} \right\} \cos 20t + \frac{10}{20} \sin 20t \\ + \left\{ \frac{400}{4000 - 10(900)} \right\} \cos 30t \\ = 0.18 \cos 20t + 0.5 \sin 20t - 0.08 \cos 30t \quad (E.4)$$

3.7 $\delta_{st} = \frac{F_0}{k} = \frac{25}{2000} = 0.0125 \text{ m}$
 steady state solution at resonance = $x(t) = \frac{\delta_{st} \cdot \omega_n t}{2} \sin \omega_n t$
 $= 0.00625 \omega_n t \sin \omega_n t \text{ m}$

(a) At end of $\frac{1}{4}$ cycle, $\omega_n t = \frac{\pi}{2}$ and $x(t) = 0.00625 \left(\frac{\pi}{2}\right) \sin \frac{\pi}{2} = 0.009817 \text{ m}$

(b) At end of $2\frac{1}{2}$ cycles, $\omega_n t = 5\pi$ and $x(t) = 0.00625(5\pi) \sin 5\pi = 0$

(c) At end of $5\frac{3}{4}$ cycles, $\omega_n t = 11\frac{1}{2}\pi$ and

$$x(t) = 0.00625 \left(\frac{23}{2}\pi\right) \sin \frac{23}{2}\pi = -0.2258 \text{ m}$$

3.8 $\delta_{st} = \frac{F_0}{k} = \frac{100}{4000} = 0.025 \text{ m}$

$$X = \delta_{st} \cdot \frac{1}{\left|1 - \left(\frac{\omega}{\omega_n}\right)^2\right|}, \quad \left|1 - \left(\frac{\omega}{\omega_n}\right)^2\right| = \frac{\delta_{st}}{X} = \frac{0.025}{20 \times 10^{-3}} = 1.25$$

$$\frac{\omega}{\omega_n} = \sqrt{1.25 + 1} = 1.5$$

$$\omega_n = \omega/1.5 = 5(2\pi)/1.5 = 20.944 \text{ rad/sec}$$

$$m = k/\omega_n^2 = 4000/(20.944)^2 = 9.1189 \text{ kg}$$

3.9 $\omega_n = \sqrt{k/m} = \sqrt{5000/10} = 22.3607 \text{ rad/sec}$

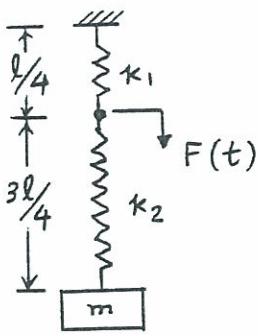
$$\delta_{st} = F_0/k = 250/5000 = 0.05 \text{ m}$$

$$X = \delta_{st} \left\{ \frac{1}{1 - \left(\frac{\omega}{\omega_n}\right)^2} \right\}$$

$$\text{i.e., } \omega = \omega_n \left(1 - \frac{\delta_{st}}{X}\right)^{\frac{1}{2}} = 22.3607 \left[1 - \frac{0.05}{0.10}\right]^{\frac{1}{2}}$$

$$= 15.8114 \text{ rad/sec}$$

3.10



$$k_1 = 4k ; \quad \frac{1}{4k} + \frac{1}{k_2} = \frac{1}{k} , \quad k_2 = \frac{4}{3}k$$

Force transmitted to the mass through k_2 :

$$\tilde{F}(t) = \frac{k_2}{k_1 + k_2} F(t) = \frac{k_1 k_2}{k_1 + k_2} \left(\frac{F_0}{k_1} \right) \cos \omega t$$

$$= k \delta_{st} \cos \omega t \quad \text{where } \delta_{st} = \frac{F_0}{k_1}$$

steady state response of m :

$$x(t) = \frac{\tilde{F}_0}{k \left\{ 1 - \left(\frac{\omega}{\omega_n} \right)^2 \right\}} \cos \omega t$$

$$= \left\{ \frac{\delta_{st}}{1 - \left(\frac{\omega}{\omega_n} \right)^2} \right\} \cos \omega t \quad \text{with } \tilde{F}_0 = k \cdot \delta_{st}$$

3.16

Equivalent stiffness of wing (beam) at location of engine:

$$k = \frac{\text{force}}{\text{deflection}} = \frac{3 E I}{\ell^3} = \frac{3 E (\frac{1}{12} b a^3)}{\ell^3} = \frac{E b a^3}{4 \ell^3}$$

$$\text{Magnitude of unbalanced force: } F = m r \omega^2 = m r \left(\frac{2 \pi N}{60} \right)^2 = \frac{m r \pi^2 N^2}{900}$$

$$\text{Equivalent mass of wing at location of engine: } M = \frac{33}{140} m_w = \frac{33}{140} (a b \ell \rho)$$

$$\text{Equation of motion: } M \ddot{x} + k x = m r \omega^2 \sin \omega t$$

Maximum steady state displacement of wing at location of engine:

$$X = \left| \frac{m r \omega^2}{k - M \omega^2} \right| = \left| \frac{\left(\frac{m r \pi^2 N^2}{900} \right)}{\left\{ \frac{E b a^3}{4 \ell^3} - \frac{33}{140} a b \ell \rho \left(\frac{2 \pi N}{60} \right)^2 \right\}} \right|$$

$$= \left| \frac{m r \ell^3 N^2}{22.7973 E b a^3 - 0.2357 \rho a b \ell^4 N^2} \right|$$

3.17

Rotating unbalanced force, $m r \omega^2$, can be resolved into two components as:

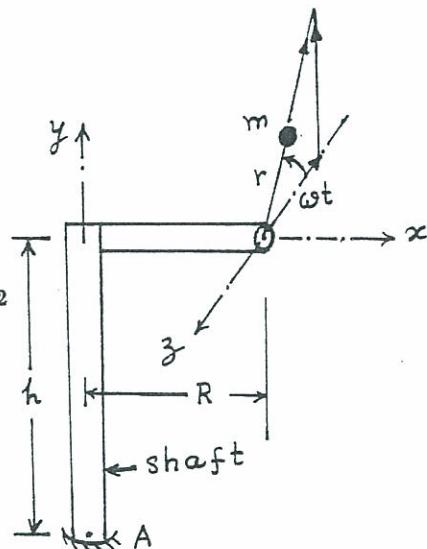
$$F_y = m r \omega^2 \sin \omega t \text{ (parallel y-axis)}$$

$$F_z = m r \omega^2 \cos \omega t \text{ (parallel z-axis)}$$

Maximum bending stress at A:

$$\sigma_b = \frac{1}{I_z} |M_z| \frac{d_o}{2} = \frac{m r \omega^2 R \left(\frac{d_o}{2} \right)}{\frac{\pi}{64} (d_o^4 - d_i^4)}$$

$$= \frac{(0.1) (0.1) (31.416^2) (0.5) \left(\frac{0.1}{2} \right)}{\frac{\pi}{64} (0.1^4 - 0.08^4)} = 8.5124 (10^4) \text{ N/m}^2$$



Maximum torsional stress at A:

$$\sigma_t = \frac{1}{J_y} |M_y| \left(\frac{d_o}{2} \right) = \frac{m r \omega^2 R \left(\frac{d_o}{2} \right)}{\frac{\pi}{32} (d_o^4 - d_i^4)}$$

$$= 4.2562 (10^4) \text{ N/m}^2$$

3.18

Total stiffness with steel specimen:

$$k_{eq} = k_1 + k_2 = 10,217.0296 + 750,000.0 = 760,217.0296 \text{ lb/in}$$

Force in specimen due to magnets (static) due to elongation $X = k_2 X$.

Force in specimen due to a.c. current in magnets (dynamic) due to elongation $X = k_{eq} X - m \omega^2 X$.

$$\text{Ratio of stresses} = \left| \frac{k_2 X}{k_{eq} X - m \omega^2 X} \right| = \frac{1}{2} \quad \text{i.e.,} \quad \left| \frac{k_2}{k_{eq} - m \omega^2} \right| = \frac{1}{2} \cdot sp$$

$$\text{i.e.,} \quad \left| \frac{750,000.0}{760,217.0296 - \left(\frac{40}{386.4} \right) \omega^2} \right| = \frac{1}{2}$$

Squaring both sides of this equation and rearranging gives:

$$107.1225 \omega^4 - 15.7365 (10^8) \omega^2 - 167.207 (10^{14}) = 0$$

$$\text{or } \omega^2 = 0.218378 (10^8) \text{ (positive value)}$$

$$\omega = 4673.0935 \text{ rad/sec} = 743.7442 \text{ Hz}$$

3.19

Equation of motion: $m_{eq} \ddot{x} + k_{eq} x = F(t)$

where $m_{eq} = \text{mass of valve and valve rod plus mass of spring at end} = (20 + (15/3))/386.4 = 0.0647 \text{ lb-sec}^2/\text{in}$.

$k_{eq} = 400 \text{ lb/in}$, $F(t) = A p(t) = A p_0 \sin \omega t = 100 (10) \sin \omega t = 1000 \sin 8t \text{ lb}$.

Response of valve (steady state) = $x_p(t) = X \sin 8t \text{ in}$ where

$$X = \frac{1000}{400 - 0.0647 (8)^2} = 2.5261 \text{ in}$$

3.20

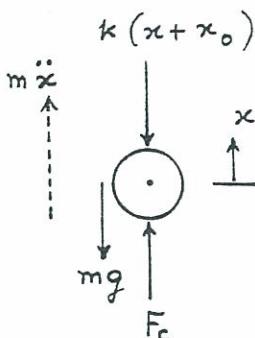
(a) Equation of motion:

$$m_0 \ddot{x} + k(x + x_0) + m_0 g = F_c; \quad \ddot{x} > 0 \quad (1)$$

where F_c = force exerted on the follower by the cam, m_0 = mass of follower plus one third the mass of the spring, and x_0 = initial displacement of the spring.

(b) Force exerted on the follower by the cam:

$$F_c = m_0 \ddot{x} + k(x + x_0) + m_0 g \quad (2)$$

with $x = e \cos \omega t$.

(c) Condition under which follower loses contact with the cam is when F_c is zero and \ddot{x} is negative. Equation (1) can be used to state this condition as:

$$k(x + x_0) + m_0 g \geq |m_0 \ddot{x}| \quad (3)$$

3.21

δ_{st} = static radial displacement of shaft under weight of turbine

δ = radial deflection of shaft during rotation

$\kappa = \frac{48 EI}{l^3}$ = stiffness of centrally loaded simply supported beam

$$m \delta \omega^2 = \kappa(\delta - \delta_{st}) \quad \text{or} \quad m \omega^2 = \kappa - \kappa \cdot \frac{\delta_{st}}{\delta}$$

$$\text{or} \quad \frac{\delta}{\delta_{st}} = \frac{\kappa}{\kappa - m \omega^2} \quad (E_1)$$

$$\text{Critical speed is } \omega_{cri} = \sqrt{\frac{\kappa}{m}} \quad (E_2)$$

If critical speed = $\frac{1}{5}$ th of operating speed,

$$\sqrt{\frac{\kappa}{m}} = \frac{1}{5} \omega \quad (E_3)$$

$$\text{Here } m = 500/386.4 = 1.2940 \text{ lb-s}^2/\text{in}$$

$$\text{and } \omega = 3000 \times 2\pi/60 = 314.16 \text{ rad/sec}$$

For solid shaft (steel) of diameter d and length l ,
Eq. (E₃) gives

$$\frac{48 EI}{m l^3} = \frac{\omega^2}{25} \quad \text{with } E = 30 \times 10^6 \text{ psi and } I = \frac{\pi d^4}{64}$$

i.e., $\frac{l^3}{d^4} = 13836.8 \quad (\text{E4})$

Let $l = 30$ in Eq. (E4) :

$$d = \frac{27000}{13836.8} = 1.9513 \text{ inch and hence } l = 58.5395 \text{ inch.}$$

3.22

$$I = \frac{\pi}{64} (d_o^4 - d_i^4) = \frac{\pi}{64} (4^4 - 3.5^4) = 5.2002 \text{ in}^4$$

$$k = \frac{48 EI}{l^3} = \frac{48(30 \times 10^6)(5.2002)}{(100)^3} = 7488.288 \text{ lb/in}$$

$$m = 500/386.4 = 1.2940 \text{ lb-s}^2/\text{in}$$

$$\omega_n = \sqrt{k/m} = \sqrt{7488.288/1.2940} = 76.0719 \text{ rad/sec}$$

$$\text{Eccentricity} = r = 2 \text{ in, eccentric mass} = m_o = \frac{0.5}{386.4} \text{ lb-s}^2/\text{in}$$

Radial force due to eccentric mass at resonance

$$= F_0 = m_o r \omega^2 = \left(\frac{5}{386.4}\right)(2)(76.0719)^2 = 149.7654 \text{ lb}$$

Let $x(t)$ = radial displacement of turbine.

At resonance, Eq. (3.15) gives, for $x_0 = \dot{x}_0 = 0$,

$$x(t) = \frac{1}{2} \delta_{st} \omega_n t \sin \omega_n t$$

$$\text{where } \delta_{st} = \frac{F_0}{k} = \frac{149.7654}{7488.288} = 0.02 \text{ in}$$

To activate the limit switch, $x(t) = 0.5 \text{ in. and hence}$

$$0.5 = \frac{1}{2} (0.02) (76.0719) t \sin 76.0719 t$$

i.e., $t \sin 76.0719 t = 0.6573 \quad (\text{E1})$

Eg. (E1) is solved by trial and error (assuming values

of $t = 1.0, 0.9, 0.8, 0.7, \text{ etc.}$) as

$$t \approx 0.6760 \text{ sec.}$$

3.23

$$\text{Tip load} = 0.1 \text{ lb, tip mass} = m_o = \frac{0.1}{386.4} = 2.588 \times 10^{-4} \text{ lb-s}^2/\text{in}^4$$

$$I = \frac{1}{12} (0.2) (0.05)^3 = 2.0833 \times 10^{-6} \text{ in}^4$$

$$k = \frac{3EI}{l^3} = \frac{3(30 \times 10^6)(2.0833 \times 10^{-6})}{(10)^3} = 0.1875 \text{ lb/in}$$

$$m = \text{mass of beam} = \frac{0.283}{386.4} (10 \times 0.2 \times 0.05) = 7.324 \times 10^{-5} \text{ lb-s}^2/\text{in}$$

$$\omega_n = \left(\frac{k}{m_0 + 0.23 m} \right)^{\frac{1}{2}} = \left[\frac{0.1875}{(2.588 + 0.7324) 10^{-4}} \right]^{\frac{1}{2}} = 5.6469 \frac{\text{rad}}{\text{sec}}$$

Eq. (3.68) gives

$$\frac{x}{Y} = \left\{ \frac{1 + (2\tau r)^2}{(1-r^2)^2 + (2\tau r)^2} \right\}^{\frac{1}{2}}$$

i.e.,

$$\frac{2.5}{0.05} = 50 = \left\{ \frac{1 + (2 \times 0.01 r)^2}{(1-r^2)^2 + (2 \times 0.01 r)^2} \right\}^{\frac{1}{2}}$$

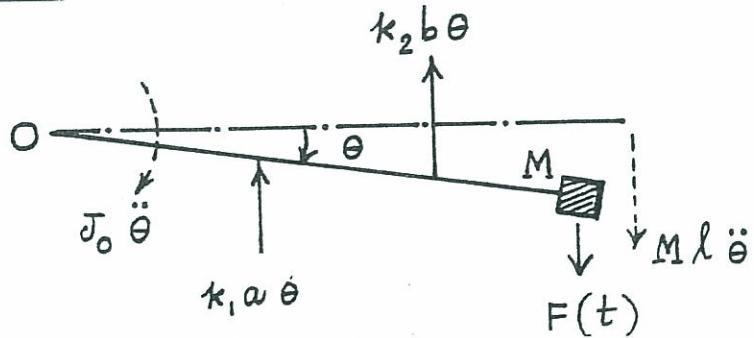
i.e.

$$r^4 - 1.9996 r^2 + 0.9996 = 0$$

$$\text{i.e. } r = \frac{\omega}{\omega_n} = 0.9999$$

$$\therefore \omega = 0.9999 \omega_n = 5.6463 \text{ rad/sec.}$$

3.24



Equation of motion for rotational motion about the hinge O:

$$(J_0 + M\ell^2)\ddot{\theta} + (k_1 a^2 + k_2 b^2)\theta = F(t)\ell = F_0\ell \sin \omega t \quad (1)$$

Steady state response (using Eqs. (3.3) and (3.6)):

$$\theta_p(t) = \Theta \sin \omega t \quad (2)$$

$$\text{where } \Theta = \frac{F_0\ell}{(k_1 a^2 + k_2 b^2) - (J_0 + M\ell^2)\omega^2} \quad (3)$$

$$\text{and } J_0 = \frac{m\ell^2}{12} + m\left(\frac{\ell}{2}\right)^2 = \frac{1}{3}m\ell^2 \quad (4)$$

For given data, $J_0 = \frac{1}{3}(10)(1^2) = 3.3333 \text{ kg-m}^2$, $\omega = \frac{1000(2\pi)}{60} = 104.72 \text{ rad/sec}$, and

$$\Theta = \frac{500 (1)}{5000 (0.25^2 + 0.5^2) - (3.3333 + 50 (1^2)) (104.72^2)} = -8.5718 (10^{-4}) \text{ rad}$$

3.25

Equation of motion for rotation about O:

$$J_0 \ddot{\theta} = -k \frac{\theta \ell}{4} \frac{\ell}{4} - k \frac{\theta 3\ell}{4} \frac{3\ell}{4} + M_0 \cos \omega t$$

i.e., $J_0 \ddot{\theta} + \left(\frac{5}{8} k \ell^2 \right) \theta = M_0 \cos \omega t$

$$\text{where } J_0 = \frac{1}{12} m \ell^2 + m \left(\frac{\ell}{4} \right)^2 = \frac{7}{48} m \ell^2 = \frac{7}{48} (10) (1^2) = 1.4583 \text{ kg-m}^2$$

and $\omega = 1000 \text{ rpm} = 104.72 \text{ rad/sec}$. Steady state solution is:

$$\theta_p(t) = \Theta \cos \omega t$$

where

$$\Theta = \frac{M_0}{\frac{5}{8} k \ell^2 - J_0 \omega^2} = \frac{100}{5000 \left(\frac{5}{8} \right) (1^2) - 1.4583 (104.72^2)} = -0.007772 \text{ rad}$$

3.26

$k = 4000 \text{ N/m}$, $m = 10 \text{ kg}$, $c = 40 \text{ N-s/m}$, $F(t) = 200 \cos 10t$,

$F_0 = 200 \text{ N}$, $\omega = 10 \text{ rad/s}$, $x_0 = 0.1 \text{ m}$, $\dot{x}_0 = 0$

$$\omega_n = \sqrt{\frac{k}{m}} = 20 \text{ rad/s}, \delta_{st} = \frac{F_0}{k} = \frac{200}{4000} = 0.05 \text{ m}$$

$$\zeta = \frac{c}{c_c} = \left(\frac{c}{2 \sqrt{km}} \right) = \left(\frac{40}{2 \sqrt{4000(10)}} \right) = 0.1$$

$$\omega_d = \sqrt{1 - \zeta^2} \omega_n = \sqrt{1 - (0.1)^2} (20) = 19.899749 \text{ rad/s}$$

$$r = \frac{\omega}{\omega_n} = \frac{10}{20} = 0.5$$

$$x = \delta_{st} / \sqrt{(1-r^2)^2 + (2\zeta r)^2} = \frac{0.05}{\sqrt{(1-0.5^2)^2 + (2(0.1)(0.5))^2}} = 0.066082 \text{ m}$$

$$\phi = \tan^{-1} \left(\frac{2\zeta r}{1-r^2} \right) = \tan^{-1} \left(\frac{2 \times 0.1 \times 0.5}{1-0.5^2} \right) = 0.132552 \text{ rad}$$

steady state response, Eq. (3.25):

$$x_p(t) = X \cos(\omega t - \phi)$$

$$= 0.066082 \cos(10t - 0.132552) \text{ m}$$

Total response, Eq. (3.35):

$$x(t) = X_0 e^{-\zeta \omega_n t} \cos(\omega_d t - \phi_0) + X \cos(\omega t - \phi) \quad (\text{E.1})$$

Using the initial conditions x_0 and \dot{x}_0 , Eq. (E.1) gives

$$x_0 = X_0 \cos \phi_0 + X \cos \phi \quad (\text{E.2})$$

$$\text{or } X_0 \cos \phi_0 = x_0 - X \cos \phi \quad (\text{E.3})$$

$$\dot{x}_0 = -\zeta \omega_n X_0 \cos \phi_0 + \omega_d X_0 \sin \phi_0 + \omega X \sin \phi \quad (\text{E.4})$$

$$\text{or } X_0 \sin \phi_0 = \frac{1}{\omega_d} \{ \dot{x}_0 + \zeta \omega_n X_0 \cos \phi_0 - \omega X \sin \phi \} \quad (\text{E.5})$$

For known values, Eqs. (E.3) and (E.5) yield

$$X_0 \cos \phi_0 = 0.034498, \quad X_0 \sin \phi_0 = -0.000922$$

Hence

$$X_0 = \{ (X_0 \cos \phi_0)^2 + (X_0 \sin \phi_0)^2 \}^{\frac{1}{2}} = 0.034510$$

$$\phi_0 = \tan^{-1} \left(\frac{X_0 \sin \phi_0}{X_0 \cos \phi_0} \right) = -0.026710$$

Thus the total response, Eq. (E.1), will be

$$x(t) = 0.034510 e^{-2t} \cos(19.899749 t + 0.026710) \\ + 0.066082 \cos(10t - 0.132552) \text{ m} \quad (\text{E.6})$$

3.27

$$k = 4000 \text{ N/m}, \quad m = 10 \text{ kg}, \quad c = 40 \text{ N-s/m}, \quad F(t) = 200 \cos 10t,$$

$$F_0 = 200 \text{ N}, \quad \omega = 10 \text{ rad/s}, \quad x_0 = 0, \quad \dot{x}_0 = 10 \text{ m/s}$$

From solution of Problem 3.26,

$$\zeta = 0.1, \quad \omega_d = 19.899749 \text{ rad/s}, \quad r = 0.5, \quad X = 0.066082 \text{ m}, \\ \phi = 0.132552 \text{ rad}$$

$$x_p(t) = 0.066082 \cos(10t - 0.132552) \text{ m}$$

$$X_0 \cos \phi_0 = x_0 - X \cos \phi = -0.065502$$

$$X_0 \sin \phi_0 = \frac{1}{\omega_d} \{ \dot{x}_0 + \zeta \omega_n X_0 \cos \phi_0 - \omega X \sin \phi \} = 0.491547$$

$$X_0 = \{ (X_0 \cos \phi_0)^2 + (X_0 \sin \phi_0)^2 \}^{\frac{1}{2}} = 0.495892$$

$$\phi_0 = \tan^{-1} \left(\frac{X_0 \sin \phi_0}{X_0 \cos \phi_0} \right) = -1.438320$$

Thus the total response, Eq. (3.35), is given by

$$x(t) = 0.495892 e^{-2t} \cos(19.899749 t + 1.438320) \\ + 0.066082 \cos(10t - 0.132552) \text{ m}$$

3.28

$$k = 4000 \text{ N/m}, m = 10 \text{ kg}, c = 40 \text{ N-s/m}, F(t) = 200 \cos 20t$$

$$F_0 = 200 \text{ N}, \omega = 20 \text{ rad/s}, x_0 = 0.1 \text{ m}, \dot{x}_0 = 0$$

$$\omega_n = \sqrt{\frac{k}{m}} = 20 \text{ rad/s}, \delta_{st} = \frac{F_0}{k} = \frac{200}{4000} = 0.05$$

$$\zeta = \frac{c}{c_c} = \frac{c}{2\sqrt{km}} = \frac{40}{2\sqrt{4000}(10)} = 0.1$$

$$\omega_d = \sqrt{1 - \zeta^2} \omega_n = \sqrt{1 - 0.1^2} (20) = 19.899749 \text{ rad/s}$$

$$r = \frac{\omega}{\omega_n} = 1$$

$$X = \frac{\delta_{st}}{\{(1-r^2)^2 + (2\zeta r)^2\}^{1/2}} = \frac{0.05}{\{(1-1^2)^2 + (2*0.1*1)^2\}^{1/2}} = 0.25$$

$$\phi = \tan^{-1}\left(\frac{2\zeta r}{1-r^2}\right) = \tan^{-1}(\infty) = \frac{\pi}{2}$$

Steady state response, Eq. (3.25):

$$x_p(t) = X \cos(\omega t - \phi) = 0.25 \cos(20t - \frac{\pi}{2}) \text{ m}$$

Total response, Eq. (3.35):

$$x(t) = X_0 e^{-\zeta \omega_n t} \cos(\omega_d t - \phi_0) + X \cos(\omega t - \phi) \quad (\text{E.1})$$

Using the initial conditions x_0 and \dot{x}_0 , Eq. (E.1) gives

$$X_0 \cos \phi_0 = x_0 - X \cos \phi = 0.1 - 0.25 \cos \frac{\pi}{2} = 0.1$$

$$X_0 \sin \phi_0 = \frac{1}{\omega_d} \{ \dot{x}_0 + \zeta \omega_n X_0 \cos \phi_0 - \omega X \sin \phi \} \\ = (0 + 0.1 * 20 * 0.1 - 20 * 0.25 * \sin \frac{\pi}{2}) / 19.899749 \\ = -0.241209$$

$$\text{Hence } X_0 = \{ (X_0 \cos \phi_0)^2 + (X_0 \sin \phi_0)^2 \}^{1/2} = 0.261117$$

$$\phi_0 = \tan^{-1}\left(\frac{X_0 \sin \phi_0}{X_0 \cos \phi_0}\right) = \tan^{-1}\left(\frac{-0.241209}{0.1}\right) = -1.177783$$

Total response :

$$x(t) = 0.261117 e^{-2t} \cos(19.899749 t + 1.777828) \\ + 0.25 \cos(20t - \frac{\pi}{2}) \text{ m}$$

3.29

$$k = 4000 \text{ N/m}, m = 10 \text{ kg}, c = 40 \text{ N-s/m},$$

$$F(t) = 200 \cos 20t \text{ N}, F_0 = 200 \text{ N}, \omega = 20 \text{ rad/s}$$

$$x_0 = 0, \dot{x}_0 = 10 \text{ m/s}$$

From solution of Problem 3.28,

$$\zeta = 0.1, \omega_n = 20 \text{ rad/s}, \omega_d = 19.899749 \text{ rad/s}, r = 1$$

$$X = 0.25, \phi = \frac{\pi}{2}$$

$$x_p(t) = 0.25 \cos(20t - \frac{\pi}{2}) \text{ m}$$

$$X_0 \cos \phi_0 = x_0 - X \cos \phi = 0 - 0 = 0$$

$$X_0 \sin \phi_0 = \frac{1}{\omega_d} \left\{ \dot{x}_0 + \zeta \omega_n X_0 \cos \phi_0 - \omega X \sin \phi \right\}$$

$$= \frac{1}{19.899749} \left\{ 10 + 0.1 * 20 * 0 - 20 * 0.25 * \sin \frac{\pi}{2} \right\}$$

$$= 0.251260$$

$$\text{Hence } X_0 = \left\{ (X_0 \cos \phi_0)^2 + (X_0 \sin \phi_0)^2 \right\}^{\frac{1}{2}} = 0.251260$$

$$\phi_0 = \tan^{-1} \left(\frac{X_0 \sin \phi_0}{X_0 \cos \phi_0} \right) = 1.570793$$

Total response:

$$x(t) = 0.251260 e^{-2t} \cos(19.899749 t - 1.570793) \\ + 0.25 \cos(20t - \frac{\pi}{2}) \text{ m}$$

3.30 $m = \frac{500}{386.4} \text{ lb-sec}^2/\text{in}$, $F(t) = 200 \sin 100 \pi t \text{ lb}$. Let $X_{\max} = 0.05 \text{ in} < 0.1 \text{ in}$ (maximum permissible value). From Eq. (3.33),

$$X_{\max} = \delta_{st} \frac{1}{2 \zeta \sqrt{1 - \zeta^2}} = 0.05 \quad (1)$$

Let $\zeta = 0.01$. Then $\delta_{st} = \frac{F_0}{k_{eq}} = \frac{200}{k_{eq}}$ and Eq. (1) gives

$$k_{eq} = \frac{200}{2(0.01) \sqrt{1 - 0.0001}(0.05)} = 20.0020 (10^4) \text{ lb/in}$$

Since shock mounts are in parallel, stiffness of each mount $= k = \frac{k_{eq}}{3} = 6.6673 (10^4) \text{ lb/in}$.

$$\zeta = \frac{c_{eq}}{c_c} = \frac{c_{eq}}{\sqrt{2 k_{eq} m}}$$

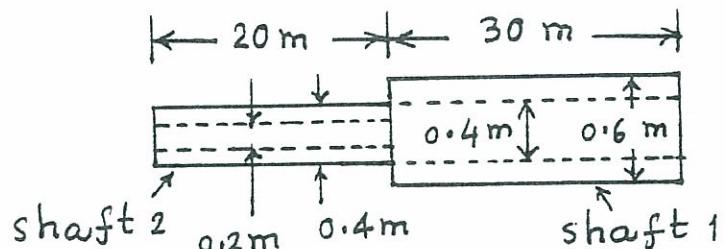
$$\text{or } c_{eq} = \zeta \sqrt{2 k_{eq} m} = 0.01 \sqrt{2 (20.0020 (10^4)) (\frac{500}{386.4})} = 7.1948 \text{ lb-sec/in}$$

$$\text{and hence } c = \frac{c_{eq}}{3} = 2.3983 \text{ lb-sec/in}$$

Equation of motion for torsional system:

3.31 $J_0 \ddot{\theta} + c_t (\dot{\theta} - \dot{\alpha}) + k_t (\theta - \alpha) = 0 \quad (1)$

where θ = angular displacement of shaft and α = angular displacement of base of shaft $= \alpha_0 \sin \omega t$. Steady state response of propeller (Eq. (3.67)):



$$\theta_p(t) = \Theta \sin(\omega t - \phi) \quad (2)$$

$$\text{where } \Theta = \alpha_0 \left\{ \frac{k_t^2 + (c_t \omega)^2}{(k_t - J_0 \omega^2)^2 - (c_t \omega)^2} \right\}^{\frac{1}{2}} \quad (3)$$

$$\text{and } \phi = \tan^{-1} \left\{ \frac{J_0 c_t \omega^3}{k_t (k_t - J_0 \omega^2) + (c_t \omega)^2} \right\} \quad (4)$$

Here $J_0 = 10^4 \text{ kg-m}^2$, $\zeta_t = 0.1$, and $\omega = 314.16 \text{ rad/sec}$. Torsional stiffnesses of shafts:

$$(k_t)_1 = \frac{G_1 J_1}{\ell_1} = \frac{(80 (10^9)) \left(\frac{\pi}{32} (0.6^4 - 0.4^4) \right)}{30} = 27.2272 (10^6) \text{ N-m/rad}$$

$$(k_t)_2 = \frac{G_2 J_2}{\ell_2} = \frac{(80 (10^9)) \left(\frac{\pi}{32} (0.4^4 - 0.2^4) \right)}{20} = 9.4248 (10^6) \text{ N-m/rad}$$

Series springs give:

$$k_t = \frac{(k_t)_1 (k_t)_2}{(k_t)_1 + (k_t)_2} = \frac{(27.2272 (10^6)) (9.4248 (10^6))}{27.2272 (10^6) + 9.4248 (10^6)} = 7.0013 (10^6) \text{ N-m/rad}$$

$$c_t = \zeta (2 \sqrt{J_0 k_t}) = 0.1 (2) \sqrt{(10^4) (7.0013 (10^6))} = 52,919.8624 \text{ N-m-s/rad}$$

From Eq. (3),

$$\Theta = 0.05 \left[\frac{(7.0013 (10^6))^2 + \left\{ 5.2920 (10^4) (314.16^2) \right\}^2}{\left\{ 7.0013 (10^6) - (10^4) (314.16^2) \right\}^2 + \left\{ 5.2920 (10^4) (314.16) \right\}^2} \right]^{\frac{1}{2}} \\ = 9.2028 (10^{-4}) \text{ rad}$$

$$\phi = \tan^{-1} \left\{ \frac{(10^4) (5.2920 (10^4)) (314.16^3)}{7.0013 (10^6) \left[7.0013 (10^6) - (10^4) (314.16^2) \right] + (5.2920 (10^4) (314.16))^2} \right\} \\ = \tan^{-1} (59.3664) = 89.0350^\circ = 1.5540 \text{ rad}$$

3.32 $X = \frac{\delta_{st}}{\{(1-r^2)^2 + (2\tau r)^2\}^{1/2}}$

For maximum X , $\frac{dX}{dr} = -\delta_{st} \cdot \frac{1}{2} \frac{1}{\{(1-r^2)^2 + (2\tau r)^2\}^{3/2}} \cdot \{2(1-r^2)(-2r) + 2(2\tau r)(2\tau)\} = 0$

i.e., $-4r(1-r^2) + 8r\tau^2 = 0$
i.e., $r = \sqrt{1-2\tau^2}$

$X \Big|_{at r=\sqrt{1-2\tau^2}} = \frac{\delta_{st}}{\left[1 - (1-2\tau^2)\right]^2 + (2\tau \sqrt{1-2\tau^2})^2}^{1/2} = \frac{\delta_{st}}{2\tau \sqrt{1-\tau^2}}$

$\therefore \left(\frac{X}{\delta_{st}}\right)_{max} = \frac{1}{2\tau \sqrt{1-\tau^2}}$

3.33 Under a d.c. current (I) through the coil, core rotates by angle θ . Torque developed due to I balances the restoring torque of spring: $aI = k_t \theta$ where a is a constant and k_t is the torsional spring constant. Under an a.c. current $I(t)$, torque developed is $T(t) = aI(t)$ and the equation of motion is:

$$J_0 \ddot{\theta} + c_t \dot{\theta} + k_t \theta = T(t) = a I(t) = a I_0 \cos \omega t \quad (1)$$

Steady state angular displacement of core:

$$\theta_p(t) = \Theta \cos(\omega t - \phi).sp \quad (2)$$

$$\text{where } \Theta = \frac{a I_0}{\left\{(k_t - J_0 \omega^2)^2 + (c_t \omega)^2\right\}^{1/2}} = \frac{\left(\frac{a I_0}{k_t}\right)}{\left[\left\{1 - \left(\frac{\omega}{\omega_n}\right)^2\right\}^2 + \left(2 \zeta \frac{\omega}{\omega_n}\right)^2\right]^{1/2}} \quad (3)$$

When $\omega = 0$ (d.c. current) and $I_0 = 1$ ampere, Eq. (1) gives

$$\Theta_{dc} = \left(\frac{a}{k_t}\right) = 1 \text{ (reading corresponding } \Theta_{dc})$$

and hence $a = k_t = 62.5$.

When $\omega = 50$ Hz = 314.16 rad/sec and $I_0 = 5$ amperes, Eq. (3) gives:

$$\Theta_{ac} = \frac{\left(\frac{a(5)}{k_t}\right)}{\left[\left(1 - \left(\frac{314.16}{250}\right)^2\right)^2 + \left(2(1) \left(\frac{314.16}{250}\right)\right)^2\right]^{1/2}} = 1.9386 \text{ amperes}$$

where $J_0 = 0.001 \text{ N-m}^2$, $k_t = 62.5 \text{ N-m/rad}$, $c_t = 0.5 \text{ N-m-s/rad}$, and
 $\omega_n = \sqrt{\frac{k_t}{J_0}} = \sqrt{\frac{62.5}{0.001}} = 250 \text{ rad/s}$. The steady state value of current indicated by ammeter = 1.9386 amperes (this shows that the ammeter is not accurate).

3.34 Eq. (3.34): $\frac{X_{res}}{\delta_{st}} = \frac{X}{\delta_{st}} \Big|_{\omega=\omega_n} = \frac{1}{2\zeta}$

i.e., $\delta_{st} = 2\zeta \left(\frac{20}{1000} \right) = 0.04 \text{ rad}$ (E₁)

Eq. (3.30): $\frac{X}{\delta_{st}} = \frac{1}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}}$; $r = 0.75 = \frac{\omega}{\omega_n}$

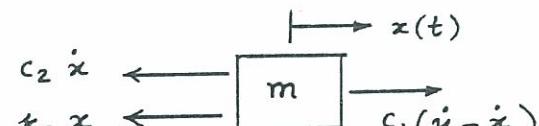
i.e., $\frac{0.01}{\delta_{st}} = \frac{1}{\sqrt{(1-0.75^2)^2 + (2\zeta \times 0.75)^2}}$ (E₂)

Eqs. (E₁) and (E₂) give

$$\frac{0.01}{0.04 \text{ rad}} = \frac{1}{\sqrt{0.1914 + 2.25 \zeta^2}}$$

i.e., $0.1914 + 2.25 \zeta^2 = 16 \zeta^2$

i.e., $\zeta = 0.1180$

3.35 (a) Equation of motion of mass: $m\ddot{x} = c_1(\dot{y} - \dot{x}) - c_2\dot{x} - k_2x$ 

i.e., $m\ddot{x} + (c_1 + c_2)\dot{x} + k_2x = c_1\dot{y} = -c_1\omega Y \sin \omega t$

(b) $x_p(t) = \frac{-(c_1\omega Y/k_2)}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}} \sin(\omega t - \phi)$

where $r = \omega/\omega_n$, $\zeta = (c_1 + c_2)\omega/(2r k_2)$ and $\phi = \tan^{-1}\left(\frac{2\zeta r}{1-r^2}\right)$.

(c) steady-state force transmitted to point P:

$$= k_2 x_p + c_2 \dot{x}_p$$

$$= \frac{-(c_1\omega Y)}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}} \left\{ \sin(\omega t - \phi) + \frac{c_2\omega}{k_2} \cos(\omega t - \phi) \right\}$$

3.40

$$\text{Eq. (3.34) gives } \left(\frac{X}{\delta_{st}} \right)_{\omega=\omega_n} = \frac{1}{2\zeta}$$

$$\text{If } X = \frac{1}{\sqrt{2}} X_{\max} = \frac{1}{\sqrt{2}} X \Big|_{\omega=\omega_n}, \text{ Eq. (3.30) gives}$$

$$\frac{1}{\sqrt{2}} \cdot \frac{1}{2\zeta} = \frac{1}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}}$$

Squaring and rearranging

$$8\zeta^2 = (1-r^2)^2 + 4\zeta^2 r^2 = 1 - 2r^2 + r^4 + 4r^2\zeta^2$$

$$r^4 + r^2(4\zeta^2 - 2) + (1 - 8\zeta^2) = 0$$

$$r^2 = 1 - 2\zeta^2 \pm 2\zeta \sqrt{1 + \zeta^2}$$

Neglecting terms involving ζ^2 ,

$$r^2 = \frac{\omega^2}{\omega_n^2} = 1 \pm 2\zeta$$

Let $\omega = \omega_1$ when $r^2 = 1 - 2\zeta$ and $\omega = \omega_2$ when $r^2 = 1 + 2\zeta$

$$\frac{\omega_2^2 - \omega_1^2}{\omega_n^2} = \frac{(\omega_2 + \omega_1)(\omega_2 - \omega_1)}{\left(\frac{\omega_2 + \omega_1}{2}\right)^2} = 4\zeta$$

$$\therefore \frac{\omega_2 - \omega_1}{\omega_2 + \omega_1} = \zeta$$

3.41

$$k_t = \frac{\pi G}{32 l} d^4 = \frac{\pi (79.3 \times 10^9)}{32 (1)} \left(\frac{4}{100}\right)^4 = 19930.31 \text{ N-m/rad}$$

$$\omega_n = \sqrt{\frac{k_t}{J_0}} = \sqrt{\frac{19930.31}{10}} = 44.6434 \text{ rad/sec}$$

$$\theta_{st} = M_{to}/k_t = 1000/19930.31 = 0.0502 \text{ rad}$$

$$\zeta_t = \frac{c_t}{2 J_0 \omega_n} = \frac{300}{2(10)(44.6434)} = 0.336$$

(a) Eq. (3.30), when written for a torsional system, gives

$$\frac{\Theta}{\theta_{st}} = \frac{1}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}}$$

$$\text{i.e., } \frac{(2/57.2956)}{0.0502} = \frac{1}{\sqrt{(1-r^2)^2 + (2 \times 0.336 r)^2}}$$

$$\text{i.e., } r^4 - 1.05484 r^2 - 1.0679 = 0$$

$$\text{i.e., } r^2 = 2.0655, -0.5171$$

$$\therefore \omega = r \omega_n = \sqrt{2.0655} (44.6434) = 64.16 \text{ rad/sec}$$

(b) Maximum torque transmitted to the support:

$$\begin{aligned}
 M_t(t) &= k_t \theta(t) + c_t \dot{\theta}(t) \\
 &= k_t \oplus \cos(\omega t - \phi) - c_t \oplus \omega \sin(\omega t - \phi) \\
 (M_t)_{\max} &= \sqrt{(k_t \oplus)^2 + (c_t \oplus \omega)^2} \\
 &= \sqrt{\left\{19930.31 \left(\frac{2}{57.2956}\right)\right\}^2 + \left\{300 \left(\frac{2}{57.2956}\right)(64.16)\right\}^2} \\
 &= 967.2 \text{ N-m}
 \end{aligned}$$

3.42

Complete solution is $x(t) = X_0 e^{-\zeta \omega_n t} \cos(\omega_d t + \phi_0) + x \cos(\omega t - \phi)$

$$\omega = 2\pi(3.5) = 21.9912 \text{ rad/sec}, \quad \omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{2500}{10}} = 15.8114 \text{ rad/sec}$$

$$\delta_{st} = \frac{F_0}{k} = \frac{180}{2500} = 0.072 \text{ m}$$

$$\zeta = \frac{c}{2m\omega_n} = \frac{45}{2(10)(15.8114)} = 0.1423, \quad r = \frac{\omega}{\omega_n} = \frac{21.9912}{15.8114} = 1.3908$$

$$\zeta \omega_n = 2.25, \quad \omega_d = \sqrt{1 - \zeta^2} \omega_n = 15.6505$$

$$x = \frac{\delta_{st}}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}} = \frac{0.072}{[(1-1.3908^2)^2 + (2 \times 0.1423 \times 1.3908)^2]^{1/2}}$$

$$= 0.07095 \text{ m}$$

$$\phi = \tan^{-1} \left(\frac{2\zeta r}{1-r^2} \right) = \tan^{-1} \left(\frac{0.3958}{-0.9343} \right) = -22.9591^\circ$$

$$x(t) = X_0 e^{-2.25t} \cos(15.6505t + \phi_0) + 0.07095 \cos(21.9912t + 22.9591^\circ)$$

$$\begin{aligned}
 \dot{x}(t) &= -2.25 X_0 e^{-2.25t} \cos(15.6505t + \phi_0) - 15.6505 X_0 e^{-2.25t} \sin(15.6505t + \phi_0) \\
 &\quad - 21.9912 (0.07095) \sin(21.9912t + 22.9591^\circ)
 \end{aligned}$$

$$x(0) = 0.015 = X_0 \cos \phi_0 + 0.07095 \cos 22.9591^\circ$$

$$X_0 \cos \phi_0 = -0.05033 \quad (\text{E}_1)$$

$$\dot{x}(0) = 5 = -2.25 X_0 \cos \phi_0 - 15.6505 X_0 \sin \phi_0 - 1.5603 \sin 22.9591^\circ$$

$$X_0 \sin \phi_0 = \frac{-0.6086 - 2.25 X_0 \cos \phi_0 - 5}{15.6505} = -0.3511 \quad (\text{E}_2)$$

Eqs. (E₁) and (E₂) give

$$X_0 = \left\{ (-0.05033)^2 + (-0.3511)^2 \right\}^{1/2} = 0.3547$$

$$\phi_0 = \tan^{-1} \left(\frac{0.3511}{0.05033} \right) = \tan^{-1}(6.9760) = 81.8423^\circ$$

3.43

(a) Eq. (3.38) gives $\frac{1}{2\zeta} \approx \left(\frac{\omega}{\omega_n}\right)_{\max} = \frac{0.2}{0.1} = 2$
 $\therefore \zeta = 0.25$

(b) Eqs. (3.42) yield

$$\left(\frac{\omega_1}{\omega_n}\right)^2 \approx 1 - 2\zeta = 0.5 , \quad \omega_1 = \omega_n \sqrt{0.5} = (5 \times 2\pi) \sqrt{0.5} \\ = 22.2145 \text{ rad/sec}$$

$$\left(\frac{\omega_2}{\omega_n}\right)^2 \approx 1 + 2\zeta = 1.5 , \quad \omega_2 = \omega_n \sqrt{1.5} = (5 \times 2\pi) \sqrt{1.5} \\ = 38.4766 \text{ rad/sec}$$

3.44

Amplitude of vibration under base excitation:

$$X = Y \left\{ \frac{\sqrt{k^2 + (c\omega)^2}}{\left[(k - m\omega^2)^2 + (c\omega)^2 \right]^{\frac{1}{2}}} \right\} \\ = \frac{(0.2) \sqrt{k^2 + c^2 (157.08)^2}}{\left[(k - 2000 (157.08)^2)^2 + c^2 (157.08)^2 \right]^{\frac{1}{2}}} = 0.1 \text{ m} \quad (1)$$

Let $k = 5 (10^6)$ N/m. Then Eq. (1) gives:

$$\frac{\sqrt{25 (10^{12}) + 2.4674 (10^4) c^2}}{\sqrt{1966.7717 (10^{12}) + 2.4674 (10^4) c^2}} = 0.5$$

i.e., $1.85055 (10^4) c^2 = 466.6929 (10^{12})$ i.e., $c = 158805.0 \text{ N-s/m}$

3.45

$$\frac{X}{Y} = \left[\frac{k^2 + c^2 \omega^2}{(k - m\omega^2)^2 + c^2 \omega^2} \right]^{\frac{1}{2}} .sp$$

$$\text{or } \frac{10^{-6}}{Y} = \left[\frac{(10^6) + (10^3 (200\pi))^2}{\left\{ 10^6 - \left(\frac{5000}{9.81} \right) (200\pi)^2 \right\}^2 + \left\{ (10^3) (200\pi) \right\}^2} \right]^{\frac{1}{2}}$$

or $Y = 169.5294 (10^{-6}) \text{ m}$

Equation of motion:

$$I_0 \ddot{\theta} + \left(k \frac{\ell}{4} \theta \right) \frac{\ell}{4} + \left(c \frac{\ell}{4} \dot{\theta} \right) \frac{\ell}{4} + \left(k \frac{3\ell}{4} \theta \right) \frac{3\ell}{4} = M_0 \cos \omega t$$

$$\text{or } I_0 \ddot{\theta} + c \frac{\ell^2}{16} \dot{\theta} + \frac{5}{8} k \ell^2 \theta = M_0 \cos \omega t$$

$$\text{where } I_0 = \frac{m \ell^2}{12} + m \left(\frac{\ell}{4}\right)^2 = \frac{7}{48} m \ell^2 = \frac{7}{48} (10) (1^2) = 1.4583 \text{ kg-m}^2$$

$$\frac{c \ell^2}{16} = \frac{(1000)(1^2)}{16} = 62.5 \text{ N-m-s/rad}$$

$$\frac{5}{8} k \ell^2 = \frac{5}{8} (5000) (1^2) = 3125.0 \text{ N-m/rad}$$

$$\omega = \frac{1000 (2\pi)}{60} = 104.72 \text{ rad/sec}$$

Equation of motion becomes:

$$1.4583 \ddot{\theta} + 62.5 \dot{\theta} + 3125.0 \theta = 100 \cos 104.72 t$$

Steady state response is given by Eq. (3.28):

$$\theta_p(t) = \Theta \cos(\omega t - \phi) = \Theta \cos(104.72 t - \phi) \cdot \text{sp}$$

$$\text{where } \Theta = \frac{100}{\sqrt{\left[3125.0 - 1.4583 (104.72^2)\right]^2 + \left[62.5 (104.72)\right]^2}} = 0.006927 \text{ rad}$$

$$\text{and } \phi = \tan^{-1} \left(\frac{62.5 (104.72)}{3125.0 - 1.4583 (104.72^2)} \right) = -0.4705 \text{ rad} = -26.9606^\circ$$

3.47 $m = 100 \text{ kg}$, $F_0 = 100 \text{ N}$, $X_{\max} = 0.005 \text{ m}$ at $\omega = 300 \text{ rpm} = 31.416 \text{ rad/sec}$.
Equations (3.33) and (3.34) yield:

$$\omega = \omega_n \sqrt{1 - 2 \zeta^2} = \sqrt{\frac{k}{m}} \sqrt{1 - 2 \zeta^2} = 31.416$$

$$\text{or } k (1 - 2 \zeta^2) = (100) (31.416^2) = 98,696.5056 \quad (1)$$

$$\text{and } X_{\max} = \delta_{st} \frac{1}{2 \zeta \sqrt{1 - \zeta^2}} = \frac{F_0}{k} \frac{1}{2 \zeta \sqrt{1 - \zeta^2}} = 0.005$$

$$\text{or } k \zeta \sqrt{1 - \zeta^2} = \frac{F_0}{2 (0.005)} = 10,000.0 \quad (2)$$

Divide Eq. (1) by (2):

$$\frac{1 - 2 \zeta^2}{\zeta \sqrt{1 - \zeta^2}} = 9.8696 \quad (3)$$

Squaring Eq. (3) and rearranging leads to:

$$101.4090 \zeta^4 - 101.4090 \zeta^2 + 1 = 0 \quad \text{or} \quad \zeta = 0.0998, 0.9950$$

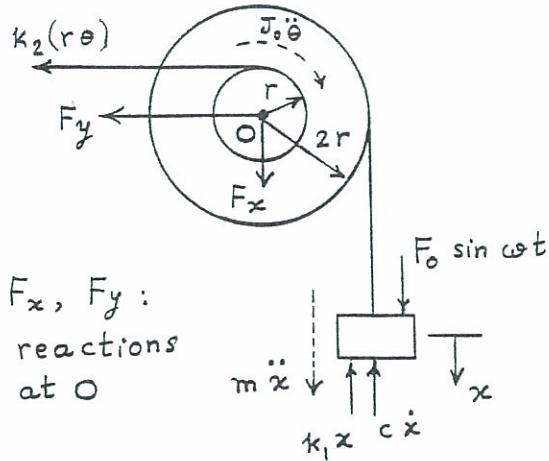
Using $\zeta = 0.0998$ in Eq. (1), we obtain

$$k = \frac{98696.5056}{1 - 2 (0.0998^2)} = 100,702.4994 \text{ N/m}$$

Since $\zeta = \frac{c}{2 m \omega_n}$, we find

$$c = 2 m \omega_n \zeta = 2 (100) \sqrt{\frac{100702.4944}{1000}} (0.0998) = 633.4038 \text{ N-s/m}$$

3.48



Equation of motion for rotation of pulley about O:

$$-k_2(\theta r)r - J_0 \ddot{\theta} - k_1 x(2r) - c \dot{x}(2r) + F_0 \sin \omega t(2r) - m \ddot{x}(2r) = 0 \quad (1)$$

where $\theta = x/(2r)$. Equation (1) can be rearranged as:

$$\left(\frac{J_0}{2r} + 2mr \right) \ddot{x} + 2cr \dot{x} + \left(2k_1 r + \frac{1}{2}k_2 r \right) x = 2r F_0 \sin \omega t \quad (2)$$

For given data, Eq. (2) becomes

$$11\ddot{x} + 50\dot{x} + 112.5x = 5 \sin 20t \quad (3)$$

Steady state response is given by Eq. (3.25):

$$x_p(t) = X \cos(\omega t - \phi)$$

where $X = \frac{5}{\sqrt{\left[\left\{ 112.5 - 11(20^2) \right\}^2 + \left\{ 50(20) \right\}^2 \right]^{1/2}}} = 0.001136 \text{ m}$

and $\phi = \tan^{-1} \left(\frac{50(20)}{112.5 - 11(20^2)} \right) = -0.2291 \text{ rad} = -13.1287^\circ$

3.49 (a)

$$\begin{aligned} I_0 \ddot{\theta} + \left(k \theta \frac{3\ell}{4} \right) \frac{3\ell}{4} + (c \ell \dot{\theta}) \ell &= \frac{\ell}{2} F_0 \sin \omega t \\ \text{or } I_0 \ddot{\theta} + c \ell^2 \dot{\theta} + \frac{9}{16} k \ell^2 \theta &= \frac{F_0 \ell}{2} \sin \omega t \end{aligned}$$

Magnitude of steady state response:

$$\Theta_a = \left(\frac{F_0 \ell}{2} \right) / \left[\left\{ \frac{9}{16} k \ell^2 - I_0 \omega^2 \right\}^2 + (c \ell^2 \omega)^2 \right]^{1/2} \quad (1)$$

(b)

$$\sum M_0 = 0 \text{ (about hinge):}$$
$$I_0 \ddot{\theta} + (k \ell \theta) \ell + \left(c \frac{3\ell}{4} \dot{\theta} \right) \frac{3\ell}{4} = \frac{\ell}{2} F_0 \sin \omega t$$
$$\text{or } I_0 \ddot{\theta} + \frac{9}{16} c \ell^2 \dot{\theta} + k \ell^2 \theta = \frac{F_0 \ell}{2} \sin \omega t$$

Magnitude of steady state response:

$$\Theta_b = \left(\frac{F_0 \ell}{2} \right) / \left[\left\{ k \ell^2 - I_0 \omega^2 \right\}^2 + \left\{ \frac{9}{16} c \ell^2 \omega \right\}^2 \right]^{\frac{1}{2}} \quad (2)$$

Usually, c is small compared to k . If the term containing c is negligible, Θ_a will be smaller than Θ_b . Hence arrangement (a) is desirable.

3.52

$$\ddot{y}(t) = \ddot{x}_g(t) = A \cos \omega t ; \quad \dot{y}(t) = \frac{A}{\omega} \sin \omega t + B_1$$

$$y(t) = -\frac{A}{\omega^2} \cos \omega t + B_1 t + B_2$$

Assuming $y(0) = \dot{y}(0) = 0$, we get

$$y(t) = -\frac{A}{\omega^2} \cos \omega t$$

Equation of motion:

$$m \ddot{x} + k(x - y) = 0$$

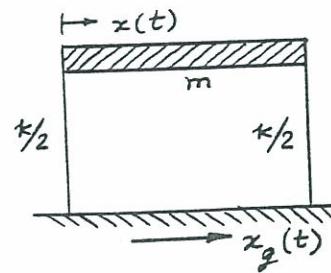
$$\text{i.e., } m \ddot{z} + k z = -m \ddot{y} = -m \ddot{x}_g(t) = -mA \cos \omega t$$

$$\text{where } z = x - y$$

Solution is:

$$z(t) = \frac{-mA \cos \omega t}{k - m \omega^2}$$

$$\therefore x(t) = z(t) + y(t) = -\left(\frac{m}{k - m \omega^2} + \frac{1}{\omega^2}\right) A \cos \omega t$$



From solution of problem 3.52,

3.53

$$\begin{aligned} x(t) &= \left| \frac{-mA}{k - m \omega^2} \right| \sin \omega t - \frac{A}{\omega^2} \sin \omega t \\ &= \left| \frac{-2000 \left(\frac{100}{1000} \right)}{0.1 \times 10^6 - 2000 (25)^2} \right| \sin 25t - \left(\frac{100}{1000} \right) \frac{1}{(25)^2} \sin 25t \end{aligned}$$

For maximum $x(t)$,

$$x(t) = \left(\frac{-200}{1.15 \times 10^6} - \frac{1}{6250} \right) \sin 25t = -3.3391 \times 10^{-4} \sin 25t \text{ m}$$

\therefore Maximum horizontal displacement of floor = 0.3339 mm

3.54

$$m(\ddot{x} - \ddot{y}) + k(x - y) = -m\ddot{y} = -m\ddot{x}_g \quad (E_1)$$

Here $y(t) = x_g(t) = X_g \cos \omega t$, and Eq. (E₁) becomes

$$m\ddot{z} + kz = m\omega^2 X_g \cos \omega t \quad \text{with } z = x - y$$

Solution is:

$$z(t) = \frac{m\omega^2 X_g \cos \omega t}{k - m\omega^2} = \frac{X_g r^2 \cos \omega t}{1 - r^2}$$

$$\text{with } \omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{0.5 \times 10^6}{2000}} = 15.8114 \text{ rad/sec}$$

$$\text{and } r = \omega/\omega_n = 200/15.8114 = 12.6491$$

$$z(t) = \left(\frac{15}{1000} \right) \left\{ \frac{12.6491^2}{1 - 12.6491^2} \right\} \cos 200t = -0.01509 \cos 200t \text{ m}$$

$$x(t) = y(t) + z(t) = \{0.015 \cos 200t - [0.01509] \cos 200t\} \text{ m}$$

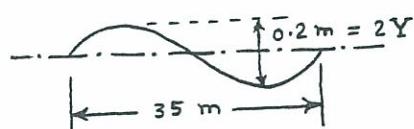
$$\therefore \text{Amplitude of vibration of floor} = 0.03009 \text{ m} = 30.09 \text{ mm.}$$

3.55

Time taken by car to travel one cycle

(35 m) is

$$\tau = \frac{35 \times 3600}{60 \times 1000} = 2.1 \text{ sec}$$



$$\text{Excitation frequency} = \omega = \frac{2\pi}{\tau} = 2.992 \text{ rad/sec}$$

$$\omega_n = 2\pi(2) = 12.5664 \text{ rad/sec}, \quad r = \frac{\omega}{\omega_n} = 0.2381, \quad \zeta = 0.15$$

Amplitude of vibration of car is given by Eq. (3.68):

$$\frac{X}{Y} = \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2} \quad (E_1)$$

$$X = 0.1 \left\{ \frac{1 + (2 \times 0.15 \times 0.2381)^2}{(1 - 0.2381^2)^2 + (2 \times 0.15 \times 0.2381)^2} \right\}^{1/2}$$

$$= 0.105977 \text{ m}$$

The most unfavorable speed corresponds to the maximum of

$\frac{X}{Y}$ in Eq. (E₁). For maximum of $\frac{X}{Y}$ with respect to r ,

$$\frac{d}{dr} \left[\frac{1 + 4\zeta^2 r^2}{1 + r^4 - 2r^2 + 4\zeta^2 r^2} \right] = 0$$

$$\text{i.e., } \frac{(1 + r^4 - 2r^2 + 4\zeta^2 r^2)(8\zeta^2 r) - (1 + 4\zeta^2 r^2)(4r^3 - 4r + 8\zeta^2 r)}{(1 + r^4 - 2r^2 + 4\zeta^2 r^2)^2} = 0$$

$$\text{i.e., } -4r(2\zeta^2 r^4 + r^2 - 1) = 0$$

$$\text{i.e., } r = 0 \quad \text{or} \quad r^2 = \frac{-1 \pm \sqrt{1 + 8\zeta^2}}{4\zeta^2}$$

Feasible value of $r^2 = \frac{-1 + \sqrt{1 + 8(0.15)^2}}{4(0.15)^2} = 0.9586$

$$r = \frac{\omega}{\omega_n} = 0.9791$$

$$\omega = 0.9791 (12.5664) = 12.3035 \text{ rad/sec} = \frac{2\pi}{T}$$

where $\tau = \frac{35 \times 3600}{s \times 1000}$ and s = speed of car in km/hr.

$$\therefore s = \frac{12.3035 \times 35 \times 3.6}{2\pi} = 246.7279 \text{ km/hr.}$$

3.56 Equations (3.73) and (3.68) give

$$F_T = m\omega^2 X = m\omega^2 Y \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$$

$$\frac{F_T}{kY} = \frac{m\omega^2}{k} \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$$

$$= r^2 \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$$

3.57 Eq. (3.75): $m\ddot{z} + c\dot{z} + k z = -m\ddot{y} = m\omega^2 Y \cos \omega t$

steady-state solution is:

$$z(t) = \frac{m\omega^2 Y \cos(\omega t - \phi_1)}{(k - m\omega^2)^2 + (c\omega)^2} = Z \cos(\omega t - \phi_1)$$

$$\text{where } \phi_1 = \tan^{-1}\left(\frac{c\omega}{k - m\omega^2}\right)$$

$$\text{Damping force} = c \frac{dz}{dt} = -c\omega Z \sin(\omega t - \phi_1)$$

Energy absorbed per cycle by the damper (E):

$$E = \int_{t=0}^{2\pi/\omega} c \frac{dz}{dt} \cdot dz = \int_0^{2\pi/\omega} \{-c\omega Z \sin(\omega t - \phi_1)\} \{-\omega Z \sin(\omega t - \phi_1)\} dt$$

$$= c\omega^2 Z^2 \int_0^{2\pi/\omega} \sin^2(\omega t - \phi_1) dt = \pi c\omega Z^2$$

$$\text{Since } Z = m\omega^2 Y / \sqrt{(k - m\omega^2)^2 + (c\omega)^2},$$

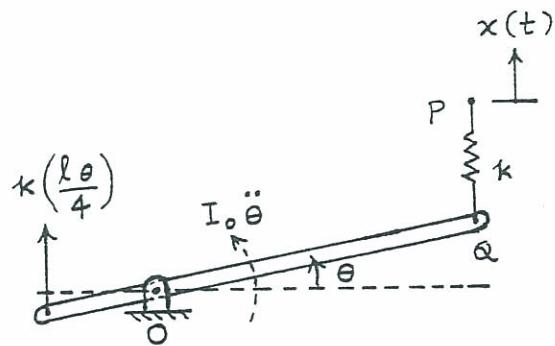
$$E = \left\{ \frac{\pi c\omega (m^2\omega^4 Y)}{(k - m\omega^2)^2 + c^2\omega^2} \right\}$$

$$\text{For maximum power, } \frac{dE}{dc} = 0$$

$$\text{i.e., } \frac{\{(k - m\omega^2)^2 + c^2\omega^2\}(\pi\omega^5 m^2 Y) - \pi c \omega^5 m^2 Y (2c\omega^2)}{\{(k - m\omega^2)^2 + c^2\omega^2\}^2} = 0$$

$$\text{i.e., } c = \left(\frac{k - m\omega^2}{\omega} \right).$$

3.58



Linear displacement of point Q due to $\theta = \frac{3\ell}{4}\theta$ and net compression of spring PQ =

$\frac{3}{4}\ell\theta - x(t)$. Equation of motion:

$$I_0 \ddot{\theta} = -\frac{k\ell\theta}{4} \frac{\ell}{4} - k \left(\frac{3\ell\theta}{4} - x(t) \right) \frac{3\ell}{4} \quad (1)$$

$$\text{where } I_0 = \frac{1}{12} m \ell^2 + m \left(\frac{\ell}{4} \right)^2 = \frac{7}{48} m \ell^2 = \frac{7}{48} (10) (1^2) = 1.4583 \text{ kg-m}^2$$

Hence Eq. (1) can be rewritten as

$$I_0 \ddot{\theta} + \left(\frac{5}{8} k \ell^2 \right) \theta = \left(\frac{3}{4} k \ell x_0 \right) \sin \omega t \quad (2)$$

Steady state angular displacement of the bar is given by Eq. (3.6):

$$\Theta = \left(\frac{3}{4} k \ell x_0 \right) / \left(\frac{5}{8} k \ell^2 - I_0 \omega^2 \right) \quad (3)$$

$$= \left(\frac{3}{4} (1000) (1) (0.01) \right) / \left(\frac{5}{8} (1000) (1^2) - 1.4583 (10^2) \right) = 0.01565 \text{ rad}$$

$$\text{and hence } \theta(t) = \Theta \sin \omega t = 0.01565 \sin 10 t \text{ rad}$$

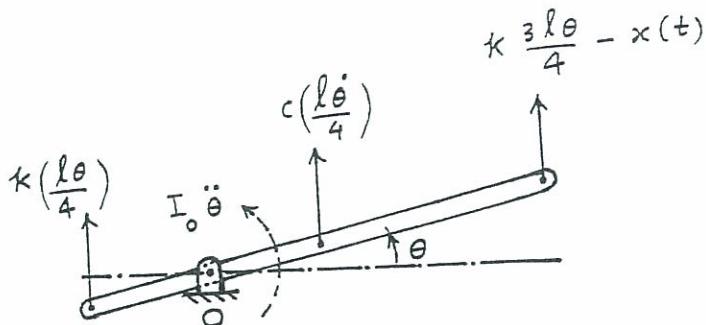
3.59

Equation of motion:

$$I_0 \ddot{\theta} = -k \frac{\ell\theta}{4} \left(\frac{\ell}{4} \right) - c \frac{\ell}{4} \dot{\theta} \left(\frac{\ell}{4} \right) - k \left(\frac{3\ell}{4} \theta - x(t) \right) \frac{3\ell}{4}$$

$$\text{i.e., } I_0 \ddot{\theta} + \frac{1}{16} c \ell^2 \dot{\theta} + \frac{5}{8} k \ell^2 = \frac{3}{4} k \ell x(t) = \frac{3}{4} k \ell x_0 \sin \omega t \quad (1)$$

$$\text{where } I_0 = \frac{1}{12} m \ell^2 + m \left(\frac{\ell}{4} \right)^2 = \frac{7}{48} m \ell^2 = \frac{7}{48} (10) (1^2) = 1.4583 \text{ kg-m}^2 \quad (2)$$



Using given data, Eq. (1) can be expressed as

$$1.4583 \ddot{\theta} + \frac{1}{16} (500) (1^2) \dot{\theta} + \frac{5}{8} (1000) (1^2) \theta = \frac{3}{4} (1000) (1) (0.01) \sin 10 t \\ \text{i.e., } 1.4583 \ddot{\theta} + 31.25 \dot{\theta} + 625.0 \theta = 7.5 \sin 10 t \quad (3)$$

Steady state angular displacement of the bar is given by Eq. (3.28) with:

$$\Theta = \frac{7.5}{\left[\left[625.0 - 1.4583 (10^2) \right]^2 + 31.25^2 (10^2) \right]^{\frac{1}{2}}} = 0.01311 \text{ rad} \\ \phi = \tan^{-1} \left(\frac{31.25 (10)}{625.0 - 1.4583 (10^2)} \right) = 0.5779 \text{ rad} \\ \therefore \theta(t) = \Theta \sin (\omega t - \phi) = 0.01311 \sin (10 t - 0.5779) \text{ rad}$$

3.60

Displacement transmissibility (T):

$$T = \frac{X}{Y} = \left[\frac{1 + (2 \zeta r)^2}{(1 - r^2)^2 + (2 \zeta r)^2} \right]^{\frac{1}{2}}$$

For maximum of T,

$$\frac{dT}{dr} = \frac{1}{2} \left[\frac{1 + 4 \zeta^2 r^2}{(1 + r^4 - 2 r^2) + 4 \zeta^2 r^2} \right]^{-\frac{1}{2}} \\ \frac{[(1 - r^2)^2 + (2 \zeta r)^2] (8 \zeta^2 r) - (1 + 4 \zeta^2 r^2) [4 r^3 - 4 r + 8 \zeta^2 r]}{[(1 - r^2)^2 + (2 \zeta r)^2]^2} = 0$$

This equation can be simplified to obtain:

$$(2 \zeta^2) r^4 + r^2 - 1 = 0 \\ \text{Solution: } r^2 = \frac{-1 \pm \sqrt{1 + 8 \zeta^2}}{4 \zeta^2} \\ \text{or } r = r_m = \frac{1}{2 \zeta} \sqrt{\sqrt{1 + 8 \zeta^2} - 1}$$

3.61

Empty

$$m = \frac{1000}{32.2} = 31.0559 \frac{\text{lb-sec}^2}{\text{ft}}$$

At speed $v = 55$ mph, and
wave length 12 ft,

$$\omega = 2\pi f = 2\pi \left(\frac{v (1760 * 3)}{3600} \right) \frac{1}{12}$$

$$= 42.2371 \text{ rad/sec}$$

$$k = 30000 \text{ lb/ft}$$

$$\zeta = 0.2$$

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{30000}{31.0559}}$$

$$= 31.0805 \text{ rad/sec}$$

$$r = \frac{\omega}{\omega_n} = \frac{42.2371}{31.0805}$$

$$= 1.3590$$

$$(2\zeta r)^2 = (2 * 0.2 * 1.3590)^2$$

$$= 0.2955$$

$$(1-r^2)^2 = (1 - 1.3590^2)^2$$

$$= 0.7170$$

$$\frac{X}{Y} = \left\{ \frac{1 + (2\zeta r)^2}{(1-r^2)^2 + (2\zeta r)^2} \right\}^{\frac{1}{2}}$$

$$= \left\{ \frac{1 + 0.2955}{0.7170 + 0.2955} \right\}^{\frac{1}{2}}$$

$$= 1.1311$$

Amplitude of vibration of automobile is magnified by a factor of 1.1311

Fully Loaded

$$m = \frac{3000}{32.2} = 93.1677 \frac{\text{lb-sec}^2}{\text{ft}}$$

$$\omega = 42.2371 \text{ rad/sec}$$

$$k = 30000 \text{ lb/ft}$$

$$\zeta = 0.2$$

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{30000}{93.1677}}$$

$$= 17.9444 \text{ rad/sec}$$

$$r = \frac{\omega}{\omega_n} = \frac{42.2371}{17.9444}$$

$$= 2.3538$$

$$(2\zeta r)^2 = (2 * 0.2 * 2.3538)^2$$

$$= 0.8864$$

$$(1-r^2)^2 = (1 - 2.3538^2)^2$$

$$= 20.6141$$

$$\frac{X}{Y} = \left\{ \frac{1 + (2\zeta r)^2}{(1-r^2)^2 + (2\zeta r)^2} \right\}^{\frac{1}{2}}$$

$$= \left\{ \frac{1 + 0.8864}{20.6141 + 0.8864} \right\}^{\frac{1}{2}}$$

$$= 0.2962$$

Amplitude of vibration of automobile is diminished by a factor of 0.2962

3.63

$$\text{Equation of motion: } M \ddot{x} + c \dot{x} + k x = m e \omega^2 \sin \omega t$$

where $\omega = \frac{3000 (2\pi)}{60} = 314.16 \text{ rad/sec}$, $M = 100 \text{ kg}$, $c = 2000 \text{ N-s/m}$, $k = 10^6 \text{ N/m}$, $m = 0.1 \text{ kg}$ and $e = r = 0.1 \text{ m}$. Steady state response is:

$$x_p(t) = X \sin(\omega t - \phi)$$

where $X = \frac{m e \omega^2}{[(k - M \omega^2)^2 + (c \omega)^2]^{\frac{1}{2}}}$

$$= \frac{0.1 (0.1) (314.16^2)}{\left[(10^6 - 100 (314.16^2))^2 + (2000 (314.16))^2 \right]^{\frac{1}{2}}} = 110.9960 (10^{-6}) \text{ m}$$

and $\phi = \tan^{-1} \left(\frac{c \omega}{k - M \omega^2} \right) = \tan^{-1} \left(\frac{2000 (314.16)}{10^6 - 100 (314.16^2)} \right) = -0.07072 \text{ rad} = -4.0520^\circ$

3.64

k = spring constant of cantilever beam

$$= \frac{3EI}{l^3} = \frac{3(2.5 \times 10^6)}{4^3} \\ = 0.1172 \times 10^6 \text{ N/m}$$

$$\omega_n = \sqrt{\frac{k}{m_1 + 0.25 m_b}} = \sqrt{\frac{0.1172 \times 10^6}{20 + 0.25(240)}} = 38.2753 \text{ rad/sec}$$

$$\omega = 2\pi (1500)/60 = 157.08 \text{ rad/sec}$$

$$r = \omega/\omega_n = 157.08/38.2753 = 4.1040, \quad r^2 = 16.8428$$

Forced response is given by Eq. (3.79) :

$$x_p(t) = X \sin(\omega t - \phi)$$

where

$$X = \frac{m e}{m_1} \cdot \frac{r^2}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}}$$

$$= \frac{(0.5)(0.15)}{20} \cdot \frac{16.8428}{\sqrt{(1-16.8428)^2 + (2 \times 0.15 \times 4.1040)^2}}$$

$$= 3.9747 \times 10^{-3} \text{ m} = 3.9747 \text{ mm}$$

$$3.65 \quad \delta_{st} = \frac{45}{1000} m = \frac{Mg}{k} = \frac{380 \times 9.81}{k}$$

i.e., $k = 82,840 \text{ N/m}$

$$\omega_n = \sqrt{\frac{k}{M}} = \sqrt{\frac{82,840}{380}} = 14.7648 \text{ rad/sec} ; \quad \omega = \frac{2\pi(1750)}{60} \\ = 183.26 \text{ rad/sec}$$

$$r = \frac{\omega}{\omega_n} = \frac{183.26}{14.7648} = 12.412 ; \quad r^2 = 154.0566$$

(i) Amplitude of vibration

$$x = \frac{me}{M} \frac{r^2}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}} = \frac{0.15}{380} \frac{154.0566}{\sqrt{(154.0566)^2 + 0}} \\ = 3.9732 \times 10^{-4} \text{ m}$$

(ii) Force transmitted to ground

$$= kx = (82840) (3.9732 \times 10^{-4}) = 32.9140 \text{ N}$$

$$3.66 \quad I = \frac{1}{12} (0.5) (0.1)^3 = 0.4167 \times 10^{-4} \text{ m}^4$$

$$k = \frac{192EI}{l^3} = \frac{192 (2.07 \times 10^{11}) (0.4167 \times 10^{-4})}{(5)^3} = 1.3248 \times 10^7 \text{ N/m}$$

$$(a) \omega_n = \sqrt{\frac{k}{M}} = \sqrt{\frac{13.248 \times 10^6}{75}} = 420.2856 \text{ rad/sec}$$

$$\omega = 2\pi(1200)/60 = 125.664 \text{ rad/sec}$$

$$r = \omega/\omega_n = 125.664/420.2856 = 0.299 , \quad r^2 = 0.0894$$

Amplitude of steady-state vibration is given by Eq. (3.30)
with $\zeta = 0$:

$$x = \frac{\delta_{st}}{|r^2 - 1|} = \frac{F_0}{k|r^2 - 1|} = \frac{5000}{(1.3248 \times 10^7)(0.9106)} \\ = 0.4145 \times 10^{-3} \text{ m}$$

(b) Using the effective mass due to self weight of beam
(for a cantilever) to be valid here also,

$$\omega_n = \sqrt{\frac{k}{M + 0.2357 \text{ m}}}$$

where M = mass of motor = 75 kg, and

$$m = \text{mass of beam} = (5 \times 0.5 \times 0.1) \left(\frac{76.5 \times 10^3}{9.81} \right) = 1949.5313 \text{ kg}$$

$$\omega_n = \sqrt{\frac{13.248 \times 10^6}{75 + (1949.5313)(0.2357)}} = 157.4339 \text{ rad/sec}$$

$$r = \omega/\omega_n = 125.664/157.4339 = 0.7982 , \quad r^2 = 0.6371$$

$$X = \frac{\delta_{st}}{|r^2 - 1|} = \frac{F_o}{k |r^2 - 1|} = \frac{5000}{(1.3248 \times 10^7)(0.3629)}$$

$$= 1.0400 \times 10^{-3} \text{ m}$$

Let width = 0.5 m and thickness = t m.

$$3.67 \quad I = \frac{1}{12}(0.5)t^3 = \frac{t^3}{24} \text{ m}^4$$

$$k = \frac{3EI}{l^3} = \frac{3(2.07 \times 10^{11})(t^3/24)}{(5)^3} = 2.07 \times 10^8 t^3 \text{ N/m}$$

$$\omega_n = \sqrt{\frac{k}{M + 0.2357m}}$$

$$\text{where } m = \text{mass of beam} = (5 \times 0.5 \times t) \left(\frac{76.5 \times 10^3}{9.81} \right) = 19495.41 t \text{ kg}$$

$$\omega_n = \sqrt{\frac{2.07 \times 10^8 t^3}{75 + 0.2357(19495.41t)}}$$

$$r = \frac{\omega}{\omega_n} = 125.664 \sqrt{\frac{75 + 4595.0688t}{2.07 \times 10^8 t^3}}$$

$$X = \frac{\delta_{st}}{|r^2 - 1|} = \frac{F_o}{k |r^2 - 1|}$$

$$\text{i.e., } 0.5 = \frac{5000}{(2.07 \times 10^8 t^3) \left\{ (125.664)^2 \left[\frac{75 + 4595.0688t}{2.07 \times 10^8 t^3} \right] - 1 \right\}}$$

$$\text{i.e., } 1.3108 \times 10^4 t^3 - 4595.069t - 74.367 = 0$$

By trial and error, the value of t is found as

$t \approx 0.6 \text{ m}$.
Since this is too large, we can start with a new width such as 1.0 m.

$$3.68 \quad m = (600/9.81) \text{ N}, \quad \omega = 2\pi(1000)/60 = 104.72 \text{ rad/sec}$$

$$k = 6(6000) = 36,000 \text{ N/m}$$

$$\omega_n = \sqrt{k/m} = \sqrt{36000 / \left(\frac{600}{9.81} \right)} = 24.2611 \text{ rad/sec}$$

$$r = \omega/\omega_n = 104.72/24.2611 = 4.3164, \quad r^2 = 18.6311$$

$$X = \frac{F_o}{k |r^2 - 1|} = \frac{m_0 e \omega^2}{k |r^2 - 1|} \quad \text{where } m_0 = \text{unbalanced mass}$$

and $e = \text{eccentricity}$

$$\text{i.e., } 2.5 \times 10^{-3} = \frac{m_o e (104.72)^2}{36000 |17.6311|}$$

$$\text{i.e., } m_o e = 0.1447 \text{ kg-m}$$

$$\therefore \text{Unbalance} = W_o e = m_o g e = 0.1447 (9.81) = 1.4195 \text{ N-m}$$

$$m = \frac{1000}{386.4} = 2.588 \frac{\text{lb-s}^2}{\text{in}} , \quad \omega = \frac{2\pi(1500)}{60} = 157.08 \frac{\text{rad}}{\text{s}}$$

3.69

Possible isolators are: (i) $k = 45000 \text{ lb/in}$, $\zeta = 0$

(ii) $k = 90000 \text{ lb/in}$, $\zeta = 0$

(iii) $k = 45000 \text{ lb/in}$, $\zeta = 0.15$

(iv) $k = 90000 \text{ lb/in}$, $\zeta = 0.15$

We will compare the force transmissibilities of these isolators.

$$\text{Force transmissibility} = T_r = \sqrt{\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2}}$$

$$(i) \omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{45000}{2.588}} = 131.8634 \text{ rad/sec}$$

$$r = \omega/\omega_n = 157.08/131.8634 = 1.1912 , \quad r^2 = 1.4190$$

$$T_r = \frac{1}{|1-r^2|} = \frac{1}{0.419} = 2.3866$$

$$(ii) \omega_n = \sqrt{\frac{90000}{2.588}} = 186.4829 \text{ rad/sec}$$

$$r = \omega/\omega_n = 157.08/186.4829 = 0.8423 , \quad r^2 = 0.7095$$

$$T_r = \frac{1}{|1-r^2|} = \frac{1}{0.2905} = 3.4423$$

$$(iii) \omega_n = \sqrt{\frac{45000}{2.588}} = 131.8634 \text{ rad/sec}$$

$$r = 1.1912 , \quad r^2 = 1.4190 , \quad \zeta = 0.15$$

$$T_r = \sqrt{\frac{1 + (2 \times 1.1912 \times 0.15)^2}{(1 - 1.4190)^2 + (2 \times 1.1912 \times 0.15)^2}} = 1.9282$$

$$(iv) \omega_n = 186.4829 \text{ rad/sec} , \quad r = 0.8423 , \quad r^2 = 0.7095 , \quad \zeta = 0.15$$

$$T_r = \sqrt{\frac{1 + (2 \times 0.8423 \times 0.15)^2}{(1 - 0.7095)^2 + (2 \times 0.8423 \times 0.15)^2}} = 2.6789$$

\therefore Isolation (iii) is best.

3.70

Eq. (3.82):

$$\frac{Mx}{me} = \frac{r^2}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}}$$

When $r = 1$,

$$\frac{Mx}{me} = \frac{1}{2\zeta} \quad \text{or} \quad \frac{M}{me} = \frac{1}{2\zeta x} = \frac{1}{2\zeta(0.55)} = \frac{1}{1.15} \quad (E_1)$$

When $r = \text{large}$,

$$\frac{Mx}{me} \approx 1 \quad \text{or} \quad \frac{M}{me} \approx \frac{1}{x} = \frac{1}{0.15} \quad (E_2)$$

Combining (E₁) and (E₂), we obtain

$$\frac{M}{me} = \frac{1}{0.15} = \frac{1}{1.15}$$

$$\therefore \zeta = 0.1364$$

For each spring,

$$3.71 \quad k = \frac{Gd^4}{64nR^3} = \frac{(11.5385 \times 10^6)(0.25)^4}{64(8)(1.5)^3} = 26.083 \text{ lb/in}$$

$$\text{Total } k = 4(26.083) = 104.332 \text{ lb/in}$$

$$\omega = \frac{2\pi(1800)}{60} = 188.496 \text{ rad/sec}$$

$$m = 100/386.4 \text{ lb-s}^2/\text{in}, \quad M = 750/386.4 \text{ lb-s}^2/\text{in}, \quad \zeta = 0$$

$$\omega_n = \sqrt{\frac{k}{M}} = \sqrt{\frac{104.332}{(750/386.4)}} = 7.3316 \text{ rad/sec}$$

$$r = 188.496/7.3316 = 25.7102, \quad r^2 = 661.0144$$

$$x = \frac{me}{M} \frac{r^2}{\sqrt{(1-r^2)^2 + (2\zeta r)^2}} = \frac{100(0.01)}{750} \left(\frac{661.0144}{660.0144} \right)$$

$$= 1.3354 \times 10^{-3} \text{ in.}$$

$$\omega = 2\pi(1500)/60 = 157.08 \text{ rad/sec}$$

$\downarrow 0.2'' \text{ dia.}$

1500 rpm

3.72

(a) Force due to eccentricity of rotor

$$= me\omega^2 = \left(\frac{30}{386.4}\right)(0.01)(157.08)^2 = 19.1569 \text{ lb.}$$

(b) H.P. = (Force)(eccentricity)(angular velocity)

$$= (19.1569)\left(\frac{0.01}{12}\right)\left(\frac{157.08}{550}\right) = 0.004559 \text{ hp.}$$

3.74

$$\omega_{n, \text{fan}} = \sqrt{\frac{k}{m_{\text{fan}}}}$$

$$= \sqrt{\frac{200}{59/386.4}}$$

$$= 39.3141 \text{ rad/sec}$$

$$\omega = \frac{2\pi(750)}{60} = 78.54 \text{ rad/sec}$$

$$(J_p)_{\text{plate+fan}} = \frac{1}{3} \left(\frac{100}{386.4} \right) (40)^2 + \left(\frac{50}{386.4} \right) (5)^2 = 141.2612 \text{ lb-in-sec}^2$$

$$F_o = me \omega^2 = \left(\frac{50}{386.4} \right) (0.1) (78.54)^2 = 79.8205 \text{ lb}$$

Point R is subjected to the force, $F(t) = F_o \cos \omega t = 79.8205 \cos 78.54 t$
Assume that S is not moving.

Then R is displaced by :

$$x(t) = \frac{F_o \cos \omega t}{|k - m\omega^2|} = \frac{F_o \cos \omega t}{k |1 - (\frac{\omega}{\omega_n})^2|} = \frac{79.8205 \cos \omega t}{200 |1 - (\frac{78.54}{39.3141})^2|}$$

$$= 0.1334 \cos 78.54 t \text{ inch}$$

Let θ = angular displacement of plate PQ.

Displacement of S = 5θ inch

Extension of spring RS = $(5\theta - 0.1334 \cos 78.54 t)$ inch

Restoring moment of spring force about P

$$= 200 [5\theta - 0.1334 \cos 78.54 t] 5 \text{ lb-in}$$

Velocity of Q = $40\dot{\theta}$ inch/sec

Damping force at Q = $40\dot{\theta}(1) = 40\dot{\theta}$ lb

Moment of damping force about P = $40\dot{\theta}(40) = 1600\dot{\theta}$ lb-in

Equation of motion of plate PQ :

$$J_p \ddot{\theta} + 1600 \dot{\theta} + 1000 (5\theta - 0.1334 \cos 78.54 t) = 0$$

$$\text{i.e., } 141.2612 \ddot{\theta} + 1600 \dot{\theta} + 5000\theta = 133.4 \cos 78.54 t \quad (E_1)$$

Comparing (E₁) with Eq. (3.24), the solution of (E₁) can be expressed as $\theta_p(t) = \Theta \cos(\omega t - \phi)$

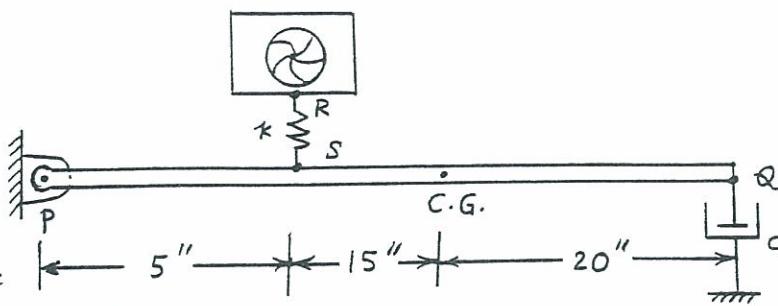
where, from Egs. (3.30) and (3.31), we get

$$\Theta = \frac{(133.4/5000)}{\sqrt{(1-174.2751)^2 + (2 \times 0.9519 \times 13.2013)^2}} = 1.5238 \times 10^{-4} \text{ rad}$$

$$\text{and } \phi = \tan^{-1}(-25.1326/173.2751) = -8.2529^\circ$$

Steady state motion of Q = $\theta_p(40)$

$$= 0.006095 \cos(78.54 t + 8.2529^\circ) \text{ inch}$$



Displacement of $S = \Theta(5)$ inch

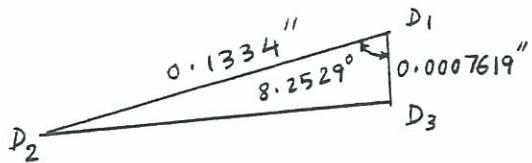
$$= (1.5238 \times 10^{-4})(5) \text{ inch}$$

$$= 0.0007619 \text{ inch}$$

$D_2 D_3$ = maximum deformation of
Spring $\approx 0.1334''$

Max. force transmitted to point $S = k(D_2 D_3)$

$$= 200(0.1334) = 26.68 \text{ lb}$$



3.78

$$\begin{aligned}
 I &= \int_0^{2\pi/\omega} \sin \omega t \cdot \cos(\omega t - \phi) dt \\
 &= \int_0^{2\pi/\omega} \sin \omega t [\cos \omega t \cdot \cos \phi + \sin \omega t \cdot \sin \phi] dt \\
 &= \int_0^{2\pi/\omega} \left\{ \cos \phi (\sin \omega t \cdot \cos \omega t) + \sin \phi (\sin^2 \omega t) \right\} dt \\
 &= \int_0^{2\pi/\omega} \left\{ \cos \phi \left(\frac{\sin 2\omega t}{2} \right) + \sin \phi \left(\frac{1 - \cos 2\omega t}{2} \right) \right\} dt \\
 &= \left. \frac{\cos \phi}{2} \left(-\frac{\cos 2\omega t}{2\omega} \right) \right|_0^{2\pi/\omega} + \left. \frac{\sin \phi}{2} \left(t - \frac{\sin 2\omega t}{2} \right) \right|_0^{2\pi/\omega} \\
 &= \frac{\pi}{\omega} \sin \phi
 \end{aligned}$$

$$\Delta W' = \omega F_0 X \cdot I = \omega F_0 X \sin \phi$$

3.79

Let $x(t)$ = displacement of mass m

New length of each spring, $k_1 = (l^2 + x^2)^{1/2}$

New tension in each spring $k_1 = T = (\sqrt{l^2 + x^2} - l) k_1 + T_0$

Horizontal component of new tension in each spring k_1

$$= T x / \sqrt{l^2 + x^2}$$

Vertical component of new tension in each spring $k_1 = \frac{T \cdot l}{\sqrt{l^2 + x^2}}$

Total friction force = $\mu mg + \frac{2Tl}{\sqrt{l^2 + x^2}}$

when mass moves to right:

Equation of motion of mass m :

$$m \ddot{x} + k_2 x + \frac{2Tl}{\sqrt{l^2 + x^2}} - \mu \left[mg + \frac{2Tl}{\sqrt{l^2 + x^2}} \right] = f_0 A \sin \omega t$$

where A = area of piston.

$$\text{i.e., } m \ddot{x} + x \left(k_2 + 2 \frac{T_0}{l} \right) = \mu mg + 2\mu T_0 + f_0 A \sin \omega t$$

Similarly, when the mass moves to left:

$$m \ddot{x} + x \left(k_2 + 2 \frac{T_0}{l} \right) = -\mu mg - 2\mu T_0 + f_0 A \sin \omega t$$

3.82

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{2100}{2}} = 32.403703 \text{ rad/sec}$$

$$N = \text{vertical force} = mg = 2(9.81) = 19.62 \text{ N}$$

$$\frac{\omega}{\omega_n} = \frac{2.5173268 \times 2\pi}{32.403703} = 0.4881191$$

$$X = \frac{F_0}{k} \left[\frac{1 - \left(\frac{4\mu N}{\pi F_0} \right)^2}{\left(1 - \frac{\omega^2}{\omega_n^2} \right)^2} \right]^{1/2}$$

$$\text{i.e., } 0.075 = \frac{120}{2100} \left[\frac{1 - \left\{ \frac{4\mu (19.62)}{\pi (120)} \right\}^2}{\left(1 - 0.4881191^2 \right)} \right]^{1/2}$$

$$\text{i.e., } 1.3125 = \left(\frac{1 - 0.04334 \mu^2}{0.5802473} \right)^{1/2}$$

$$\text{i.e., } 0.9995666 = 1 - 0.04334 \mu^2$$

$$\text{i.e., } \mu = 0.1$$

3.83

$$(a) k = \frac{W}{\delta_{st}} = \frac{5000}{0.05} = 10^5 \text{ N/m}$$

when $\omega = \omega_n$, Eq. (3.102) gives

$$X = \frac{F_0}{k \beta} \Rightarrow 0.1 = \frac{1000}{(10^5) \beta} \Rightarrow \beta = 0.1$$

$$(b) \Delta W = \pi C_{eq} \omega X^2 = \pi \beta k X^2 \quad \text{where } C_{eq} = \frac{\beta k}{\omega} \text{ from Eq. (3.100)}$$

$$\Delta W = \pi (0.1) (10^5) (0.1)^2 = 314.16 \text{ Joules/cycle}$$

(c) Steady state amplitude at one-quarter of resonant frequency :

$$\frac{\omega}{\omega_n} = 0.25$$

$$X = \frac{F_0}{k \left[\left\{ 1 - \left(\frac{\omega}{\omega_n} \right)^2 \right\}^2 + \beta^2 \right]^{1/2}} = \frac{1000}{10^5 \left[\left\{ 1 - 0.25^2 \right\}^2 + (0.1)^2 \right]^{1/2}}$$

$$= 0.01061 \text{ m}$$

(d) steady state amplitude at thrice the resonant frequency:

$$\frac{\omega}{\omega_n} = 3$$

$$X = \frac{1000}{10^5 \left[(1 - 3^2)^2 + (0.1)^2 \right]^{1/2}} = 0.00125 \text{ m}$$

3.84

$$\Delta W = \pi \beta \times X^r$$

$$\left. \begin{array}{l} 3.8 = \pi \beta (60000) (0.04)^r \\ 9.5 = \pi \beta (60000) (0.06)^r \end{array} \right\} \Rightarrow \begin{array}{l} \beta (0.04)^r = 0.0000202 \\ \beta (0.06)^r = 0.0000504 \end{array}$$

Taking logarithms,

$$\ln \beta + r \ln (0.04) = \ln (0.0000202)$$

$$\ln \beta + r \ln (0.06) = \ln (0.0000504)$$

$$\text{i.e., } \ln \beta - 3.218876 r = -10.809828 \quad \dots \quad (E_1)$$

$$\ln \beta - 2.813411 r = -9.895511 \quad \dots \quad (E_2)$$

$$\text{subtracting } (E_1) \text{ from } (E_2), \quad 0.405465 r = 0.914309$$

$$r = 2.254964$$

$$\text{From } (E_1), \quad \ln \beta = -10.809828 + 3.218876 (2.254964) = -3.551378$$

$$\beta = 0.028685$$

$$\text{Work done} = W = \int F dx = \int F \dot{x} dt$$

3.85 If $F(t) = F_0 \cos \omega t$ and $x(t) = X \cos(\omega t - \phi)$, work done in one cycle

$$\begin{aligned} &= W = - \int_0^{2\pi/\omega} F_0 \cos \omega t \cdot \omega X \sin(\omega t - \phi) dt \\ &= - \frac{F_0 \omega X \cos \phi}{2} \left(-\frac{1}{2\omega} \cos 2\omega t \right)_0^{2\pi/\omega} + \frac{F_0 \omega X \sin \phi}{2} \left(t + \frac{1}{2\omega} \sin 2\omega t \right)_0^{2\pi/\omega} \\ &= F_0 \pi X \sin \phi \end{aligned}$$

$$\text{Given data: } F_0 = 5 \text{ lb}, \omega = 3\pi \frac{\text{rad}}{\text{sec}}, \tau = \frac{2}{3} \text{ sec}, \phi = \frac{\pi}{3}, X = 0.5''$$

$$W = F_0 \pi X \sin \phi = 5\pi (0.5) \sin \frac{\pi}{3} = 6.8018 \text{ lb-in}$$

(i) In one second, it will complete $1\frac{1}{2}$ cycles.

$$W \Big|_{1 \text{ second}} = 1.5 W = 10.2027 \text{ lb-in.}$$

(ii) In four seconds, it will complete 6 cycles.

$$W \Big|_{4 \text{ seconds}} = 6 W = 40.8108 \text{ lb-in.}$$

3.86

$$\text{Damping force} = F = C(\dot{x})^n$$

Energy dissipated per quarter cycle during harmonic motion $x(t) = X \sin \omega t$ is

$$\frac{\Delta W}{4} = \int_0^{\pi/2\omega} C(\dot{x})^n dx = \int_0^{\pi/2\omega} C(\omega X \cos \omega t)^n dx$$

$$\text{But } dx = \dot{x} dt = \omega X \cos \omega t \cdot dt$$

$$\begin{aligned}\Delta W &= 4c \omega^{n+1} X^{n+1} \int_0^{\pi/2\omega} \cos^{n+1} \omega t \cdot dt \\ &= 4c \omega^{n+1} X^{n+1} \left\{ \frac{1}{(n+1)\omega} \cos^n \omega t \cdot \sin \omega t \Big|_0^{\pi/2\omega} + \frac{n}{n+1} \int_0^{\pi/2\omega} \cos^{n-1} \omega t \cdot dt \right\} \\ &= 4c \omega^{n+1} X^{n+1} \left(\frac{n}{n+1} \right) \int_0^{\pi/2\omega} \cos^{n-1} \omega t \cdot dt\end{aligned}$$

Equating this expression to $\pi c_{eq} \omega X^2$, we obtain

$$c_{eq} = \frac{4c \omega^n X^{n-1}}{\pi} \left(\frac{n}{n+1} \right) \int_0^{\pi/2\omega} \cos^{n-1} \omega t \cdot dt \equiv c \omega^n X^{n-1} \alpha_n$$

$$\text{where } \alpha_n = \frac{4}{\pi} \left(\frac{n}{n+1} \right) \int_0^{\pi/2\omega} \cos^{n-1} \omega t \cdot dt \quad \dots \quad (E_1)$$

For example, for $n=2$, (E₁) becomes

$$\alpha_2 = \frac{4}{\pi} \left(\frac{2}{3} \right) \int_0^{\pi/2\omega} \cos \omega t \cdot dt = \frac{8}{3\pi} \left(\frac{\sin \omega t}{\omega} \right)_0^{\pi/2\omega} = \frac{8}{3\pi\omega}$$

$$\text{and hence } c_{eq} = \frac{8c\omega X}{3\pi}$$

which can be seen to be same as the expression found in Example 3.7.

For few other values of n , α_n can be found as follows:

n	1	2	3	4
α_n	$\frac{1}{\omega}$	$\frac{8}{3\pi\omega}$	$\frac{3}{4\omega}$	$\frac{32}{15\pi\omega}$

The amplitude can be found as

$$X = \frac{F_0}{\sqrt{(k-m\omega^2)^2 + c_{eq}^2 \omega^2}} = \frac{F_0}{\sqrt{k^2(1-r^2)^2 + c_{eq}^2 \omega^2}}$$

$$= \frac{F_0}{\sqrt{k^2(1-r^2)^2 + c^2 \omega^{2(n+1)} X^{2(n-1)} \alpha_n^2}}$$

Energy dissipated per cycle for viscous damping = $\pi c \omega X^2$

Energy dissipated per cycle for Coulomb damping = $4\mu N X$

Equivalent viscous damping constant (c_{eq}) is given by

$$\pi c_{eq} \omega X^2 = \pi c \omega X^2 + 4\mu N X$$

$$c_{eq} = \left(c + \frac{4\mu N}{\pi \omega X} \right)$$

3.87

Amplitude X is given by

$$X = \frac{F_0}{\sqrt{(k - m\omega^2)^2 + c_{eq}^2\omega^2}} = \frac{F_0}{\sqrt{k^2(1-r^2)^2 + c_{eq}^2\omega^2}}$$

substituting for c_{eq} , squaring and rearranging,

$$X^2 \{ k^2(1-r^2)^2 + c^2\omega^2 \} + X \left(\frac{8\mu N c \omega}{\pi} \right) + \left(\frac{16\mu^2 N^2}{\pi^2} - F_0^2 \right) = 0$$

3.88

(a) Equation of motion $m\ddot{x} \pm \mu N + c(\dot{x})^3 + kx = F_0 \cos \omega t$
thus the system has combined Coulomb and velocity-cubed damping.

$$\text{For Coulomb damping, } c_{eq1} = \frac{4\mu N}{\pi \omega X} \quad (E_1)$$

For velocity-cubed damping, the equivalent viscous damping coefficient can be obtained from the solution of problem 3.68: (E_2)

$$c_{eq2} = c \omega^3 X^2 \alpha_3$$

Where

$$\alpha_3 = \frac{4}{\pi} \left(\frac{3}{4} \right) \int_0^{\pi/2\omega} \cos^2 \omega t \, dt = \frac{3}{4\omega} \quad (E_3)$$

$$\therefore c_{eq2} = \frac{3}{4} c \omega^2 X^2 \quad (E_4)$$

$$\text{and } c_{eq} = c_{eq1} + c_{eq2} = \frac{4\mu N}{\pi \omega X} + \frac{3}{4} c \omega^2 X^2 \quad (E_5)$$

(b) Steady state amplitude under harmonic force:

$$X = \frac{F_0}{\sqrt{k^2(1-r^2)^2 + c_{eq}^2\omega^2}} = \frac{F_0}{\sqrt{k^2(1-r^2)^2 + \left\{ \frac{4\mu N}{\pi \omega X} + \frac{3}{4} c \omega^2 X^2 \right\}^2 \omega^2}} \quad (E_6)$$

(c) Amplitude ratio:

$$\begin{aligned} \frac{X}{\delta_{st}} &= \frac{X}{(F_0/k)} = \frac{1}{\sqrt{(1-r^2)^2 + \left(\frac{c_{eq}^2 \omega^2}{k^2} \right)}} \\ &= \frac{1}{\sqrt{(1-r^2)^2 + \left\{ \frac{4\mu N}{\pi X k} + \frac{3}{4} c \omega^3 X^2 \right\}^2}} \end{aligned} \quad (E_7)$$

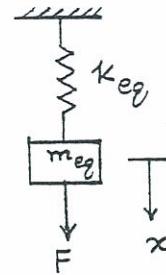
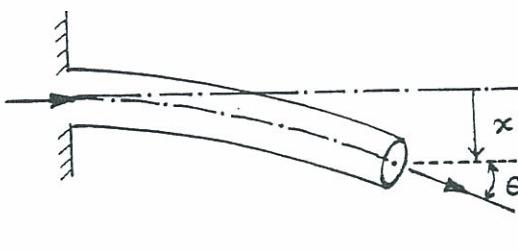
At resonance, $r=1$ and Eq. (E7) reduces to

$$\left. \frac{X}{\delta_{st}} \right|_{\text{resonance}} = \frac{1}{\left\{ \frac{4\mu N}{\pi X k} + \frac{3}{4k} c \omega^3 X^2 \right\}} \quad (E_8)$$

3.89

Model the pipe as a single degree of freedom system with m_{eq} = equivalent mass at end

$$= \frac{33}{140} m \quad (m = \text{mass of pipe; see Problem 2.46}) \text{ and } k_{eq} = \frac{3EI}{\ell^3}. \quad \text{Slope of pipe at end:}$$



$$\theta = \frac{F \ell^2}{2 E I} = \frac{F \ell^3}{3 E I} \left(\frac{3}{2 \ell} \right) = \frac{3 x}{2 \ell}$$

where x = end deflection of the cantilever pipe under a transverse load F . Force induced due to fluid velocity v is $\rho A v^2$. Force acting on the single degree of freedom system (in vertical direction):

$$F = \rho A v^2 \sin \theta \approx \rho A v^2 \theta = \rho A v^2 \frac{3 x}{2 \ell}$$

$$\text{Equation of motion: } m_{eq} \ddot{x} + k_{eq} x = F$$

$$\text{or } \frac{33}{140} m \ddot{x} + \left(\frac{3 E I}{\ell^3} - \frac{3 \rho A v^2}{2 \ell} \right) x = 0$$

$$\text{Instability occurs when } \frac{3 E I}{\ell^3} - \frac{3 \rho A v^2}{2 \ell} < 0 \quad \text{or} \quad v > \sqrt{\frac{2 E I}{\rho A \ell^2}}$$

3.90

Assume Reynolds number (R_e) greater than 1000. Strouhal number (St) for vortex shedding is taken as: $St = \frac{f d}{V} = 0.21$ where f = frequency of vortex shedding, d = diameter of cylinder and V = velocity of fluid (air). At 50 mph speed,

$$V = \frac{50 (1760) (36)}{3600} = 880 \text{ in/sec} \quad \text{and} \quad f = \frac{0.21 V}{d} = \frac{184.8}{d} \text{ Hz} \quad (d \text{ in inches})$$

For the three sections of the antenna, the vortex frequencies are:

$$f_1 = \frac{184.8}{0.3} = 616.0 \text{ Hz}$$

$$f_2 = \frac{184.8}{0.2} = 924.0 \text{ Hz}$$

$$f_3 = \frac{184.8}{0.1} = 1848.0 \text{ Hz}$$

At 75 mph speed,

$$V = \frac{75 (1760) (36)}{3600} = 1320 \text{ in/sec} \quad \text{and} \quad f = \frac{0.21 V}{d} = \frac{277.2}{d} \text{ Hz} \quad (d \text{ in inches})$$

For the three sections of the antenna, the frequencies are:

$$f_1 = \frac{277.2}{0.3} = 924.0 \text{ Hz}$$

$$f_2 = \frac{277.2}{0.2} = 1386.0 \text{ Hz}$$

$$f_3 = \frac{277.2}{0.1} = 2772.0 \text{ Hz}$$

Since the natural frequencies are much smaller, no instability occurs.

3.91

- (a) Equivalent mass of single d.o.f. system = $m_{eq} = M + \frac{33}{140} m$ where m = mass of cylindrical part of the sign post:

$$m = \frac{\pi}{4} (D^2 - d^2) h \rho = \frac{\pi}{4} (0.25^2 - 0.2^2) (10) \left(\frac{76500}{9.81} \right) = 1378.0527 \text{ kg}$$

$$\therefore m_{eq} = 200 + \frac{33}{140} (1378.0527) = 524.8267 \text{ kg}$$

$$I = \frac{\pi}{64} (D^4 - d^4) = \frac{\pi}{4} (0.25^4 - 0.2^4) = 113.208 (10^{-6}) \text{ m}^4$$

Equivalent stiffness of the system:

$$k_{eq} = \frac{3 EI}{h^3} = \frac{3 (207 (10^9)) (113.208 (10^{-6}))}{10^3} = 70,302.168 \text{ N/m}$$

Natural frequency of transverse vibration of sign post:

$$\omega_1 = \left(\frac{k_{eq}}{m_{eq}} \right)^{\frac{1}{2}} = \left(\frac{70302.168}{524.8267} \right)^{\frac{1}{2}} = 11.5738 \text{ rad/sec} = 1.8420 \text{ Hz}$$

- (b) Wind velocity corresponding to maximum vibration of sign post (V) is given by:

$$St = 0.21 = \frac{f_1 D}{V} \quad \text{or} \quad V = \frac{f_1 D}{0.21} = \frac{(1.8420)(0.25)}{0.21} = 2.1929 \text{ m/s}$$

- (c) Maximum force acting on the system due to wind velocity:

$$F(t) = F_0 \sin \omega t = \frac{1}{2} c \rho V^2 A \sin \omega t = \frac{1}{2} (1) (1.2215) (2.19299^2) (8.0) \sin \omega t \text{ N}$$

$$= 23.4958 \sin \omega t \text{ N}$$

where $c = 1$ for a cylinder, ρ = density of air = 1.2215 kg/m^3 , A = projected area of cylindrical part = $(0.8)(10) = 8.0 \text{ m}^2$, and ω = frequency of wind force. Equation of motion:

$$m_{eq} \ddot{x} + c_{eq} \dot{x} + k_{eq} x = F(t)$$

and the maximum steady state displacement of the sign post occurs when $\omega = \omega_1$ and is given by Eq. (3.34):

$$X = \frac{\delta_{st}}{2 \zeta} = \frac{F_0}{k_{eq} (2) \zeta} = \frac{23.4958}{2 (0.1) (70302.168)} = 0.001671 \text{ m}$$

3.92

(a) Equation of motion

or

$$m \ddot{x} + c \dot{x} + kx = F_0 x$$

$$m \ddot{x} + c \dot{x} + (k - F_0) x = 0 \quad (E_1)$$

Assuming the solution $x(t) = C e^{st}$ (E₂)where C is a constant, Eq. (E₁) gives the auxiliary equation

$$\delta^2 + \frac{c}{m} \delta + \left(\frac{k - F_0}{m}\right) = 0 \quad (E_3)$$

Roots of (E₃) are

$$\delta_{1,2} = -\frac{c}{2m} \pm \sqrt{\left(\frac{c}{2m}\right)^2 - \left(\frac{k - F_0}{m}\right)} \quad (E_4)$$

First consider the case of positive stiffness ($k > F_0$). For this case, following possibilities exist.1. If $\left(\frac{c}{2m}\right)^2 > \left(\frac{k - F_0}{m}\right)$:Both s_1 and s_2 will be real and negative and hence

$$x(t) = C_1 e^{s_1 t} + C_2 e^{s_2 t} \quad (E_5)$$

will be stable.

2. If $\left(\frac{c}{2m}\right)^2 = \left(\frac{k - F_0}{m}\right)$:Both s_1 and s_2 will be identical, real and negative.

Solution

$$x(t) = (C_1 + C_2 t) e^{s_1 t} \quad (E_6)$$

will be stable since $e^{s_1 t} \rightarrow 0$ as $t \rightarrow \infty$.3. If $\left(\frac{c}{2m}\right)^2 < \left(\frac{k - F_0}{m}\right)$:Here s_1 and s_2 will be complex conjugates and solution will be

$$x(t) = C e^{-\left(\frac{c}{2m}\right)t} \sin\left(\sqrt{\left\{-\left(\frac{c}{2m}\right)^2 + \left(\frac{k - F_0}{m}\right)\right\}} t + \phi\right) \quad (E_7)$$

This represents a converging oscillatory motion and hence the system will be stable.

Next consider the case of negative stiffness ($k < F_0$). Here

$$\delta_{1,2} = -\frac{c}{2m} \pm \sqrt{\left(\frac{c}{2m}\right)^2 + \left(\frac{F_0 - k}{m}\right)} \quad (E_8)$$

Thus s_1 will be positive and s_2 will be negative, and the solution becomes

$$x(t) = C_1 e^{+s_1 t} + C_2 e^{-|s_2|t} \quad (E_9)$$

This solution can be seen to diverge as $t \rightarrow \infty$.(b) Equation of motion $m \ddot{x} + c \dot{x} + kx = F_0 \dot{x}$

$$\text{or } \ddot{x} + \left(\frac{c - F_0}{m}\right) \dot{x} + \frac{k}{m} x = 0 \quad (E_{10})$$

Assuming $x(t) = C e^{st}$ the auxiliary equation becomes

$$s^2 + \left(\frac{c - F_0}{m}\right)s + \frac{k}{m} = 0 \quad (E_{11})$$

and hence

$$s_{1,2} = -\left(\frac{c - F_0}{2m}\right) \pm \sqrt{\left(\frac{c - F_0}{2m}\right)^2 - \frac{k}{m}} \quad (E_{12})$$

First consider the case of positive damping ($c > F_0$) in (E₁₀). For this case, it can be seen that the system will be stable for all possible values of $\left\{ \left(\frac{c - F_0}{2m}\right)^2 - \frac{k}{m} \right\}$.

Next, consider the case of negative damping ($c < F_0$). Depending on the sign of the quantity under the radical in Eq. (E₁₂), we will have three types of solution.

1. $\left(\frac{c - F_0}{2m}\right)^2 > \frac{k}{m}$. Here both s_1 and s_2 are real and positive and hence $x(t) = C_1 e^{s_1 t} + C_2 e^{s_2 t}$ (E₁₃)

This denotes a diverging nonoscillatory motion; so the system is unstable.

2. $\left(\frac{c - F_0}{2m}\right)^2 = \frac{k}{m}$. Here s_1 and s_2 are identical and are real and positive. Hence $x(t) = (C_1 + C_2 t) e^{s_1 t}$ (E₁₄)

This represents a diverging nonoscillatory solution; so the system will be unstable.

3. $\left(\frac{c - F_0}{2m}\right)^2 < \frac{k}{m}$. Here s_1 and s_2 are complex conjugates and hence

$$s_{1,2} = \left(\frac{F_0 - c}{2m}\right) \pm i \sqrt{\frac{k}{m} - \left(\frac{c - F_0}{2m}\right)^2} \quad (E_{15})$$

The solution becomes

$$x(t) = X e^{\left(\frac{F_0 - c}{2m}\right)t} \sin \left(\sqrt{\frac{k}{m} - \left(\frac{c - F_0}{2m}\right)^2} t + \phi \right) \quad (E_{16})$$

Since the exponent is positive, Eq.(E₁₆) denotes a diverging oscillatory motion and hence the system is unstable.

Thus the condition for dynamic stability of the system can be stated as

$$F_0 \leq c \quad (E_{17})$$

$$(c) \text{ Equation of motion} \quad m\ddot{x} + c\dot{x} + kx = F_0 \ddot{x}$$

or $(m - F_0)\ddot{x} + c\dot{x} + kx = 0$ (E_{18})

With the solution $x(t) = C e^{st}$ (E_{19})

the auxiliary equation will be

$$s^2 + \left(\frac{c}{m - F_0}\right)s + \left(\frac{k}{m - F_0}\right) = 0 \quad (E_{20})$$

The roots are

$$s_{1,2} = -\frac{c}{2(m - F_0)} \pm \sqrt{\left\{\frac{c}{2(m - F_0)}\right\}^2 - \left(\frac{k}{m - F_0}\right)} \quad (E_{21})$$

First consider the case of positive mass ($m > F_0$) in (E_{18}) . In this case, the system will be stable for all values of

$$\left[\left\{\frac{c}{2(m - F_0)}\right\}^2 - \left(\frac{k}{m - F_0}\right) \right].$$

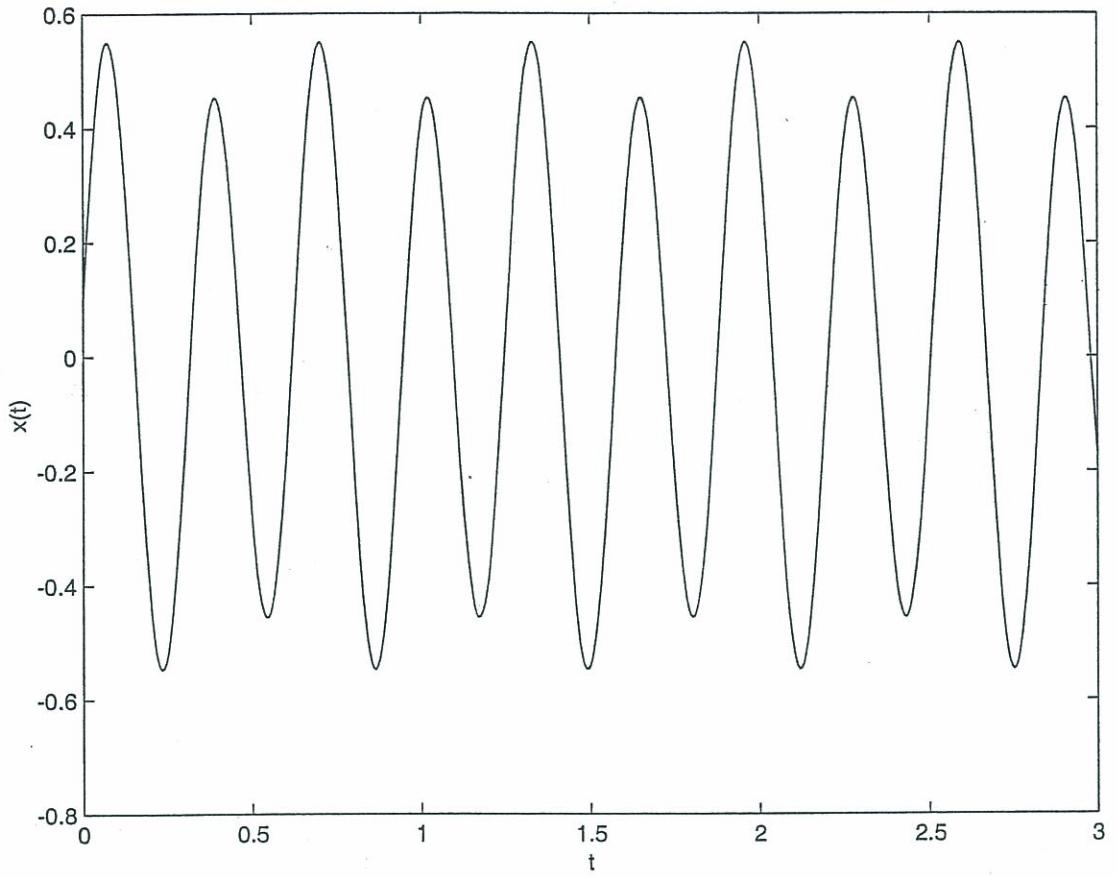
Next consider the case of negative mass ($m < F_0$) in (E_{18}) . For this case s_1 and s_2 can be expressed as

$$s_{1,2} = \frac{c}{2(F_0 - m)} \pm \sqrt{\left\{\frac{c}{2(F_0 - m)}\right\}^2 + \left(\frac{k}{F_0 - m}\right)} \quad (E_{22})$$

This shows that s_1 will be positive and s_2 will be negative; thus the solution will be divergent.

3.103

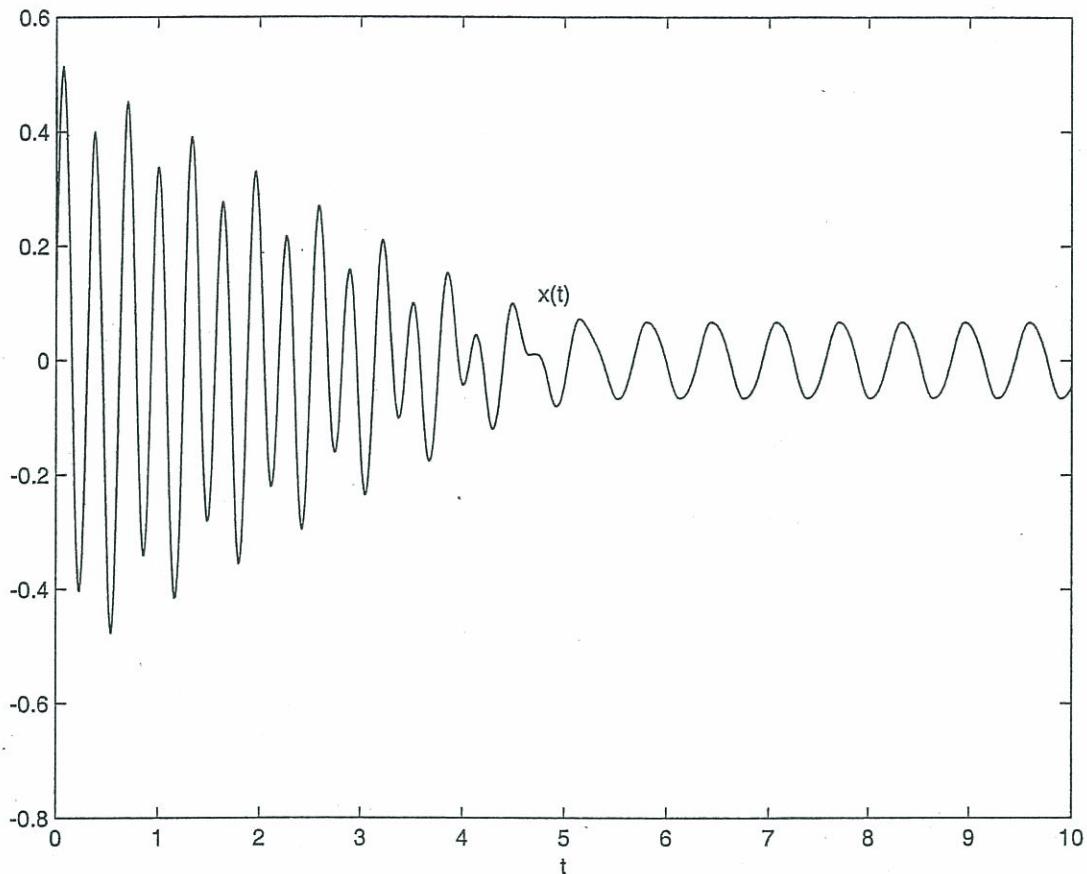
```
% Ex3_103.m
k = 4000;
m = 10;
w = 10;
F0 = 200;
wn = sqrt(k/m);
x0 = 0.1;
x0_dot = 10;
f_0 = F0/m;
for i = 1: 501
    t(i) = 3 * (i-1)/500;
    x(i) = x0_dot*sin(wn*t(i))/wn + (x0 - f_0/(wn^2-w^2))*cos(wn*t(i))...
        + f_0/(wn^2-w^2)*cos(w*t(i));
end
plot(t,x);
xlabel('t');
ylabel('x(t)');
```



3.104

```
% Ex3_104.m
% This program will use the function dfunc3_104.m, they should
% be in the same folder
tspan = [0: 0.01: 10];
x0 = [0.1; 10];
[t,x] = ode23('dfunc3_104', tspan, x0);
disp('      t          x(t)        xd(t)');
disp([t x]);
plot(t,x(:,1));
xlabel('t');
gtext('x(t)');

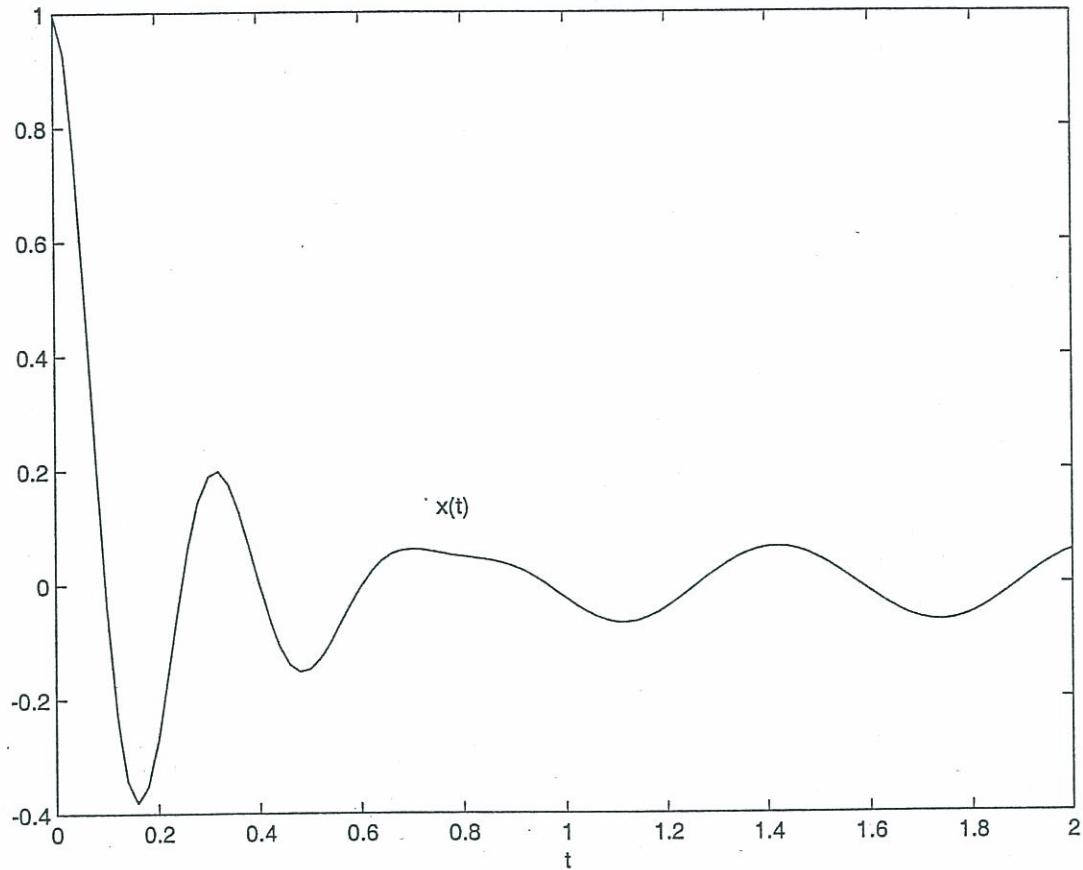
% dfunc3_104.m
function f = dfunc3_104(t,x)
F0 = 200;
w = 10;
u = 0.3;
m = 10;
k = 4000;
f = zeros(2,1);
f(1) = x(2);
f(2) = (F0/m)*sin(w*t) - 9.81*u*sign(x(2)) - (k/m)*x(1);
```



3.105

```
% Ex3_105.m
% This program will use the function dfunc3_105.m, they should
% be in the same folder
tspan = [0: 0.02: 2];
x0 = [1; 0];
[t,x] = ode23('dfunc3_105', tspan, x0);
disp(['      t      x(t)      xd(t)']);
disp([t x]);
plot(t,x(:,1));
xlabel('t');
gtext('x(t)');

% dfunc3_105.m
function f = dfunc3_105(t,x)
m = 100;
k = 40000;
zeta = 0.25;
Y = 0.05;
w = 10;
c = 2 * zeta * sqrt(k*m);
f = zeros(2,1);
f(1) = x(2);
f(2) = k*Y*sin(w*t)/m + c*w*Y*cos(w*t)/m - c*x(2)/m - (k/m)*x(1);
```



>> Ex3_105

t	x(t)	xd(t)
0	1.0000	0
0.0200	0.9272	-6.9328
0.0400	0.7381	-11.5448
0.0600	0.4823	-13.6094
0.0800	0.2096	-13.3072
0.1000	-0.0372	-11.1218
0.1200	-0.2270	-7.7195
<hr/>		
.		
.		
.		
1.8000	-0.0523	0.3904
1.8200	-0.0434	0.4869
1.8400	-0.0329	0.5637
1.8600	-0.0210	0.6179
1.8800	-0.0083	0.6473
1.9000	0.0047	0.6510
1.9200	0.0176	0.6286
1.9400	0.0297	0.5811
1.9600	0.0406	0.5103
1.9800	0.0499	0.4194
2.0000	0.0572	0.3120

3.106

```
%=====
% Ex3_106.m (Program3.m)
% Main program which calls HARESP
%
%=====
% Run "Ex3_106.m" in MATLAB command window. Ex3_106.m and haresp.m should
% be in the same folder, and set the path to this folder
% following seven lines contain problem-dependent data
xm=10.0;
xk=1000;
zeta=0.1;
xc=2*zeta*sqrt(xk*xm);
f0=100.0;
om=20.0;
n=20;
ic=1;
% end of problem-dependent data
[t,x,xd,xdd,xamp,xphi]=haresp(xm,xc,xk,f0,om,ic,n);
% following lines output the results
fprintf('Steady state response of an undamped\n');
fprintf('Single degree of freedom system under harmonic force\n\n');
fprintf('Given data\n');
fprintf('xm = %10.8e\n', xm);
fprintf('xc = %10.8e\n', xc);
fprintf('xk = %10.8e\n', xk);
fprintf('f0 = %10.8e\n', f0);
fprintf('om = %10.8e\n', om);
fprintf('ic = %1.0f\n', ic);
fprintf('n = %2.0f\n\n', n);
fprintf('Response: \n\n');
fprintf('    i           x(i)           xd(i)           xdd(i)\n');
fprintf('    \n\n');
for i=1:n
    fprintf('%2.0f    %10.8e    %10.8e    %10.8e\n', i, x(i), ...
            xd(i), xdd(i));
end
subplot(311);
plot(t,x);
ylabel('x(t)');
gtext('x(t)');
subplot(312);
plot(t,xd);
ylabel('xd(t)');
gtext('xd(t)');
subplot(313);
plot(t,xdd);
ylabel('xdd(t)');
gtext('xdd(t)');
xlabel('t');
```

```

%=====
%
%function haresp.m
%
%=====
function [t,x,xd,xdd,xamp,xphi]=haresp(xm,xc,xk,f0,om,ic,n);
omn=sqrt(xk/xm);
xai=xc/(2.0*xm*omn);
dst=f0/xk;
r=om/omn;
xamp=dst/sqrt((1.0-r^2)^2+(2.0*xai*r)^2);
xphi=atan(2.0*xai*r/(1.0-r^2));
delt=2.0*3.1416/(om*n);
time=0.0;
if ic~=0
    for i=1:n
        time=time+delt;
        t(i) = time;
        x(i)=xamp*cos(om*time-xphi);
        xd(i)=-xamp*om*sin(om*time-xphi);
        xdd(i)=-xamp*om^2*cos(om*time-xphi);
    end
else
    for i=1:n
        time=time+delt;
        t(i) = time;
        x(i)=xamp*sin(om*time-xphi);
        xd(i)=xamp*om*cos(om*time-xphi);
        xdd(i)=-xamp*om^2*sin(om*time-xphi);
    end
end
>> Ex3_106
Steady state response of an undamped
Single degree of freedom system under harmonic force

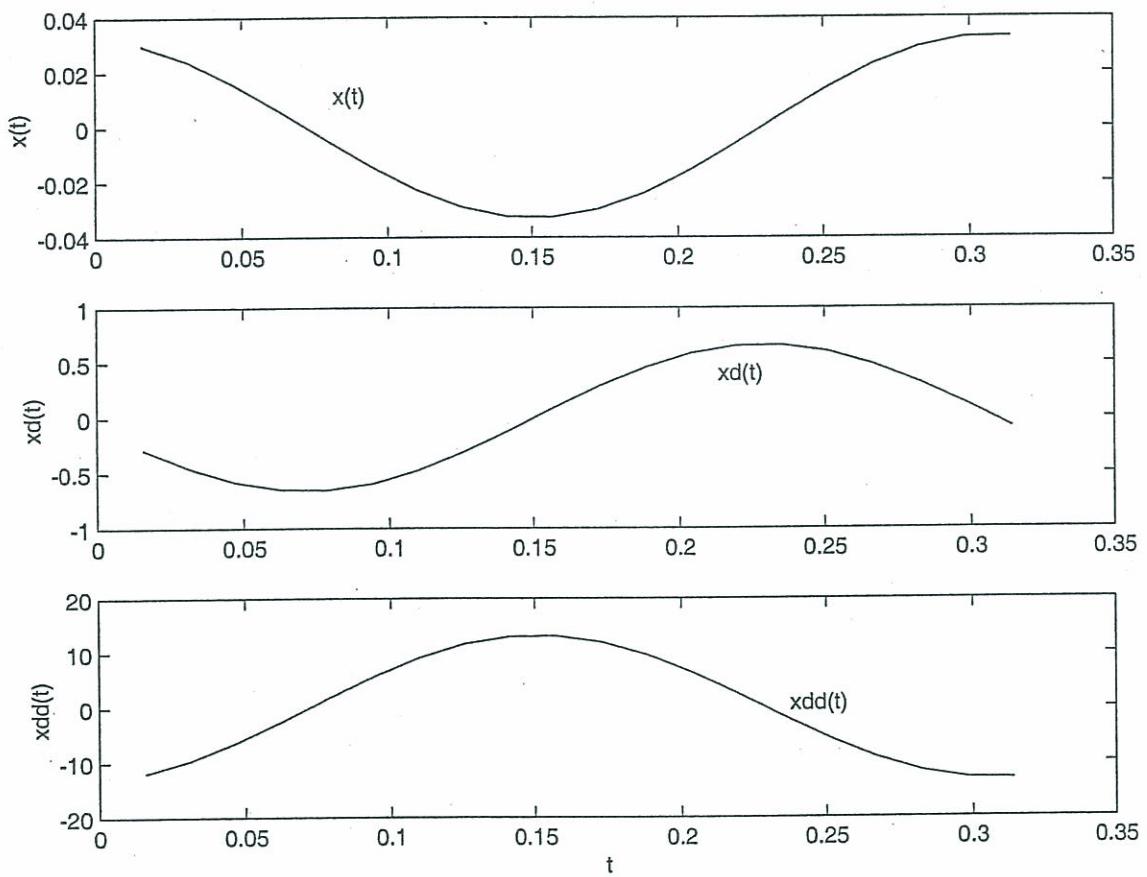
```

Given data
 $xm = 1.0000000e+001$
 $xc = 2.0000000e+001$
 $xk = 1.0000000e+003$
 $f0 = 1.0000000e+002$
 $om = 2.0000000e+001$
 $ic = 1$
 $n = 20$

Response:

i	x(i)	xd(i)	xdd(i)
1	2.97987095e-002	-2.85475021e-001	-1.19194838e+001
2	2.39294085e-002	-4.55669383e-001	-9.57176339e+000
3	1.57177193e-002	-5.81259445e-001	-6.28708774e+000
4	5.96746320e-003	-6.49951518e-001	-2.38698528e+000
5	-4.36693253e-003	-6.55021513e-001	1.74677301e+000
6	-1.42738605e-002	-5.95973141e-001	5.70954420e+000

7	-2.27835571e-002	-4.78586496e-001	9.11342282e+000
8	-2.90630300e-002	-3.14352252e-001	1.16252120e+001
9	-3.24975978e-002	-1.19346877e-001	1.29990391e+001
10	-3.27510596e-002	8.73410566e-002	1.31004238e+001
11	-2.97986046e-002	2.85479399e-001	1.19194419e+001
12	-2.39292411e-002	4.55672899e-001	9.57169644e+000
13	-1.57175058e-002	5.81261754e-001	6.28700233e+000
14	-5.96722446e-003	6.49952395e-001	2.38688978e+000
15	4.36717313e-003	6.55020872e-001	-1.74686925e+000
16	1.42740794e-002	5.95971044e-001	-5.70963176e+000
17	2.27837329e-002	4.78583148e-001	-9.11349314e+000
18	2.90631454e-002	3.14347982e-001	-1.16252582e+001
19	3.24976417e-002	1.19342102e-001	-1.29990567e+001
20	3.27510275e-002	-8.73458687e-002	-1.31004110e+001



3.107

```
%Ex3_107.m
Y= 0.05;
zeta = 0.1;
wn = 8.164966;
w = 2.90889;
r = w/wn;
x = Y * sqrt( (1 + (2 * zeta * r)^2)/((1 - r^2)^2 + (2 * zeta * r)^2) );
disp(' w wn x');
disp([w wn x]);
w = 14.54445;
r = w/wn;
x = Y * sqrt( (1 + (2 * zeta * r)^2)/((1 - r^2)^2 + (2 * zeta * r)^2) );
disp([w wn x]);
w = 29.08890;
r = w/wn;
x = Y * sqrt( (1 + (2 * zeta * r)^2)/((1 - r^2)^2 + (2 * zeta * r)^2) );
disp([w wn x]);
wn = 6.324555;
w = 2.90889;
r = w/wn;
x = Y * sqrt( (1 + (2 * zeta * r)^2)/((1 - r^2)^2 + (2 * zeta * r)^2) );
disp([w wn x]);
w = 14.54445;
r = w/wn;
x = Y * sqrt( (1 + (2 * zeta * r)^2)/((1 - r^2)^2 + (2 * zeta * r)^2) );
disp([w wn x]);
w = 29.08890;
r = w/wn;
x = Y * sqrt( (1 + (2 * zeta * r)^2)/((1 - r^2)^2 + (2 * zeta * r)^2) );
disp([w wn x]);
```

>> Ex3_107

w	wn	x
2.908890000000000	8.164966000000000	0.05722376420338
14.544450000000000	8.164966000000000	0.02410324256879
29.088900000000000	8.164966000000000	0.00524102723160
2.908890000000000	6.324555000000000	0.06325355032007
14.544450000000000	6.324555000000000	0.01275990975243
29.088900000000000	6.324555000000000	0.00336736169683

3.108

$$\text{Equation of motion is } m\ddot{x} + c\dot{x} + kx = c\dot{y} + ky$$

$$\text{when } y(t) = Y \sin \omega t, \quad x_p(t) = X \cos(\omega t - \phi_1 - \phi_2)$$

Complete solution can be expressed as

$$x(t) = x_0 e^{-\zeta \omega_n t} \cos(\omega_d t + \phi_0) + X \cos(\omega t - \phi_1 - \phi_2) \quad (\text{E.1})$$

$$\text{with } X = Y \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2},$$

$$\phi_1 = \tan^{-1} \left(\frac{2\zeta r}{1 - r^2} \right), \quad \phi_2 = \tan^{-1} \left(\frac{1}{2\zeta r} \right), \quad r = \frac{\omega}{\omega_n}.$$

If the initial conditions are known $x(t=0) = x_0$ and $\dot{x}(t=0) = \dot{x}_0$,

$$x_0 = x_0 \cos \phi_0 + X \cos(\phi_1 + \phi_2)$$

$$\text{and } \dot{x}_0 = -\zeta \omega_n x_0 \cos \phi_0 - \omega_d x_0 \sin \phi_0 - \omega X \sin(-\phi_1 - \phi_2)$$

Hence

$$x_0 = \left[\{x_0 - X \cos(\phi_1 + \phi_2)\}^2 + \left\{ \frac{-\dot{x}_0 - \zeta \omega_n x_0 + \zeta \omega_n X \cos(\phi_1 + \phi_2) + \omega X \sin(\phi_1 + \phi_2)}{\omega_d} \right\}^2 \right]^{1/2}$$

$$\phi_0 = \tan^{-1} \left[\frac{-\dot{x}_0 - \zeta \omega_n x_0 + \zeta \omega_n X \cos(\phi_1 + \phi_2) + \omega X \sin(\phi_1 + \phi_2)}{\omega_d \{x_0 - X \cos(\phi_1 + \phi_2)\}} \right]$$

If necessary, the velocity $\dot{x}(t)$ and acceleration $\ddot{x}(t)$ can be found from Eq. (E.1). The computer program and output are given below.

```
C =====
C
C SOLUTION OF PROBLEM 3.108
C MAIN PROGRAM WHICH CALLS BASEX
C RESPONSE OF A SINGLE D.O.F. SYSTEM SUBJECTED TO BASE EXCITATION,
C Y(T)=Y*SIN(OM*T)
C =====
```

```

C FOLLOWING 3 LINES CONTAIN PROBLEM-DEPENDENT DATA
      DIMENSION X(20),XD(20),XDD(20)
      DATA XM,XC,XK,Y,OM,N/2.0,10.0,100.0,0.1,25.0,20/
      DATA X0,XDO/0.01,5.0/
C END OF PROBLEM-DEPENDENT DATA
      CALL BASEX (XM,XC,XK,Y,OM,N,X,XD,XDD,X0,XDO)
      PRINT 100
100  FORMAT (//,33H TOTAL RESPONSE OF AN UNDERDAMPED,,,
2 52H SINGLE D.O.F. SYSTEM UNDER HARMONIC BASE EXCITATION)
      PRINT 200, XM,XC,XK,Y,OM,N
200  FORMAT (//,12H GIVEN DATA:,5H XM =,E15.8,,5H XC =,E15.8,,,
2 5H XK =,E15.8,,5H Y =,E15.8,,5H OM =,E15.8,,5H N =,12)
      PRINT 300,X0,XDO
300  FORMAT (/,20H INITIAL CONDITIONS:,6H X0 =,E15.8,,6H XDO =,
2 E15.8)
      PRINT 400
400  FORMAT (//,10H RESPONSE:,//,5H I ,3X,5H X(I),12X,6H XD(I),
2 11X,7H XDD(I),/)
      DO 500 I=1,N
500  PRINT 600, I,X(I),XD(I),XDD(I)
600  FORMAT (I4,2X,E15.8,2X,E15.8,2X,E15.8)
      STOP
      END
C =====
C
C SUBROUTINE BASEX
C =====
C
      SUBROUTINE BASEX (XM,XC,XK,Y,OM,N,X,XD,XDD,X0,XDO)
      DIMENSION X(N),XD(N),XDD(N)
      OMN=SQRT(XK/XM)
      XAI=XC/(2.0*XM*OMN)
      OMD=OMN*SQRT(1.0-XAI**2)
      R=OM/OMN
      DELT=2.0*3.1416/(OMD*REAL(N))
      XAMP=Y*SQRT(1.0+(2.0*XAI*R)**2/((1.0-R**2)**2+(2.0*XAI*R)**2))
      PHI1=ATAN(2.0*XAI*R/(1.0-R*R))
      PHI2=ATAN(1.0/(2.0*XAI*R))
      XCC=XC
      TIME=0.0
      DO 10 I=1,N
      TIME=TIME+DELT
      XC=X0-XAMP*COS(PHI1+PHI2)
      XS=(-XDO-XAI*OMN*XC+OM*XM*XAMP*SIN(PHI1+PHI2))/OMD
      XZ=SQRT(XC**2+XS**2)
      PZ=ATAN(XS/XC)
      EX=EXP(-XAI*OMN*TIME)
      CS=COS(OMD*TIME+PZ)
      SI=SIN(OMD*TIME+PZ)
      CS12=COS(OM*TIME-PHI1-PHI2)
      SI12=SIN(OM*TIME-PHI1-PHI2)
      X(I)=XZ+EX*CS+XAMP*CS12
      XD(I)=-XAI*OMN*XZ*EX*CS-OMD*XZ*EX*SI-OM*XAMP*SI12
10    XDD(I)=XZ*EX*CS*((XAI*OMN)**2-OMD**2)+2.0*XAI*OMN*OMD*XZ*EX*SI
2 -(OM**2)*XAMP*CS12

```

```
XC=XCC  
RETURN  
END
```

TOTAL RESPONSE OF AN UNDERDAMPED
SINGLE D.O.F. SYSTEM UNDER HARMONIC BASE EXCITATION

GIVEN DATA:

XM = 0.20000000E+01
XC = 0.10000000E+02
XK = 0.10000000E+03
Y = 0.10000000E+00
DM = 0.25000000E+02
N = 20

INITIAL CONDITIONS:

X0 = 0.9999998E-02
XDO = 0.50000000E+01

RESPONSE:

I	X(I)	XD(I)	XDD(I)
1	-0.50330855E-01	-0.57580528E+01	-0.10305269E+02
2	-0.30772516E+00	-0.44899216E+01	0.62558521E+02
3	-0.43412885E+00	-0.65008521E+00	0.85064461E+02
4	-0.38465756E+00	0.22720275E+01	0.28117821E+02
5	-0.27407524E+00	0.18295681E+01	-0.40405914E+02
6	-0.24066399E+00	-0.41657627E+00	-0.39765869E+02
7	-0.28432682E+00	-0.90879965E+00	0.23411983E+02
8	-0.27970907E+00	0.14345939E+01	0.64174423E+02
9	-0.14624044E+00	0.38781688E+01	0.26183165E+02
10	0.39102390E-01	0.33305802E+01	-0.47345394E+02
11	0.12840362E+00	0.26650119E+00	-0.67513680E+02
12	0.80883794E-01	-0.18088942E+01	-0.10970811E+02
13	0.96553117E-02	-0.68976790E+00	0.50778744E+02
14	0.38462169E-01	0.18209940E+01	0.40712753E+02
15	0.14735566E+00	0.22107751E+01	-0.27525837E+02
16	0.19996722E+00	-0.31552225E+00	-0.66972618E+02
17	0.11764607E+00	-0.28227291E+01	-0.26468327E+02
18	-0.18114097E-01	-0.23140993E+01	0.45068512E+02
19	-0.63696094E-01	0.51316518E+00	0.59701672E+02
20	0.10478826E-01	0.21295214E+01	0.43720150E+00

3.111

Unbalanced force in vertical direction = $m e \omega^2 \sin \omega t$ (E₁)

Unbalanced force in horizontal direction = 0

Let M = total mass of the shaker

Equation of motion is $M \ddot{x} + c \dot{x} + kx = m e \omega^2 \sin \omega t$ (E₂)

steady state solution of (E₂) is

$$x(t) = X \sin(\omega t - \phi) \quad (\text{E}_3)$$

where

$$X = \frac{m e r^2}{M [(1-r^2)^2 + (2\zeta r)^2]^{1/2}} \quad (E_4)$$

and $\phi = \tan^{-1} \left(\frac{2\zeta r}{1-r^2} \right) \quad (E_5)$

where $r = \frac{\omega}{\omega_n} = \omega \sqrt{\frac{M}{k}} \quad (E_6)$

Frequency range: 20 Hz to 30 Hz
i.e., 125.664 rad/sec to 188.496 rad/sec

$$\therefore 125.664 \text{ rad/sec} \leq \omega \leq 188.496 \text{ rad/sec} \quad (E_7)$$

$0.1'' \leq X \leq 0.2''$ where X is given by (E₄).

Mean power output over a time period τ is given by

$$P = \frac{1}{\tau} \int_0^\tau F(\tau) \frac{dx}{dt}(\tau) d\tau \quad (E_8)$$

where $\tau = \frac{2\pi}{\omega}$,

$$F(\tau) = m e \omega^2 \sin \omega t, \text{ and}$$

$$\frac{dx}{dt} = \omega X \cos(\omega t - \phi)$$

$$P \geq 1 \text{ hp} \quad (E_9)$$

$$\frac{M}{m} \geq 50 \quad (E_{10})$$

Procedure:

Find ω , e , M , m , k and c
so as to satisfy the requirements stated in
(E₇), (E₈), (E₁₀) and (E₁₁).

3.112 $m = \frac{10^5}{386.4} = 258.7992 \text{ lb-s}^2/\text{in}$, $l = 600''$, $E = 30 \times 10^6 \text{ psi}$

$$k = \frac{3EI}{l^3} = \frac{3(30 \times 10^6)}{(600)^3} \cdot \frac{\pi}{64} (D^4 - d^4) = 0.020453 (D^4 - d^4) \frac{\text{lb}}{\text{in}}$$

$$\omega = 2\pi(15) = 94.248 \text{ rad/sec}; \zeta = 0.15$$

Ground acceleration $\ddot{y}(t) = 193.2 \sin 94.248 t \text{ in/s}^2 \quad (E_{11})$

Equation of motion:

$$m\ddot{z} + c\dot{z} + kz = -m\ddot{y} = -50000 \sin 94.248t \quad (E_2)$$

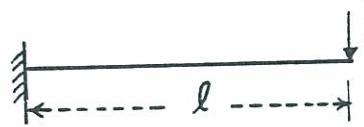
where $z = x - y$ = relative displacement.

$$\text{Let } z(t) = Z \sin(\omega t - \phi) = Z \sin(94.248 t - \phi) \quad (E_3)$$

$$\text{with } Z = \frac{-50000}{\sqrt{(k-m\omega^2)^2 + c^2\omega^2}} \quad (E_4)$$

$$\text{and } \phi = \tan^{-1}\left(\frac{c\omega}{k-m\omega^2}\right) \quad (E_5)$$

$$y_{\max} = \frac{Fl^3}{3EI} \Rightarrow \frac{3y_{\max}}{l^2} = \frac{Fl}{EI}$$



Max. bending moment = $M = Fl$

$$\text{Max. bending stress} = \frac{M(D/2)}{EI} = \frac{Fl}{EI} \cdot \frac{D}{2} = \frac{3D}{2l^2} y_{\max}$$

If maximum relative displacement, $y_{\max} = Z$, is known, max. bending stress (σ_1) is given by

$$\sigma_1 = \frac{3D}{2l^2} \cdot Z$$

Direct compressive stress (σ_2) due to weight of water tank is

$$\sigma_2 = \frac{mg}{\frac{\pi}{4}(D^2-d^2)} = \frac{4 \times 10^5}{\pi(D^2-d^2)}$$

$$\text{Total stress} = \sigma_1 + \sigma_2 \leq 30000 \text{ psi}$$

$$\text{i.e., } \frac{3D}{2l^2} Z + \frac{4 \times 10^5}{\pi(D^2-d^2)} \leq 30000 \quad (E_6)$$

Natural frequency of water tank is

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{0.020453(D^4-d^4)}{258.7992}} \geq 30\pi \frac{\text{rad}}{\text{sec}}$$

$$\text{i.e., } D^4 - d^4 \geq 1.1240 \times 10^8 \quad (E_7)$$

Weight of steel column is

$$\begin{aligned} W_s &= \frac{\pi}{4}(D^2-d^2)l \text{ Pweight} = \frac{\pi}{4}(D^2-d^2)(600)(0.283) \\ &= 133.3609(D^2-d^2) \text{ lb.} \end{aligned} \quad (E_8)$$

Problem: Find $\{d\}$ to minimize W_s subject to restrictions of (E_6) and (E_7) . Also use the conditions :
 $D \geq d$, $D \geq 0$ and $d \geq 0$.

Procedure: Plot the inequalities (E_6) and (E_7) in the $D-d$ space and draw contours of W_s and identify the minimum weight design.

