
GNC Homework 1

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1a

We set up the axes that y is due north and the x-y plane is tangent to the earth surface. Turning 90 degrees in x-y plane will result in **almost** East heading. We ignore the curvature of the earth since the distance between Worcester and Boston is relatively small.

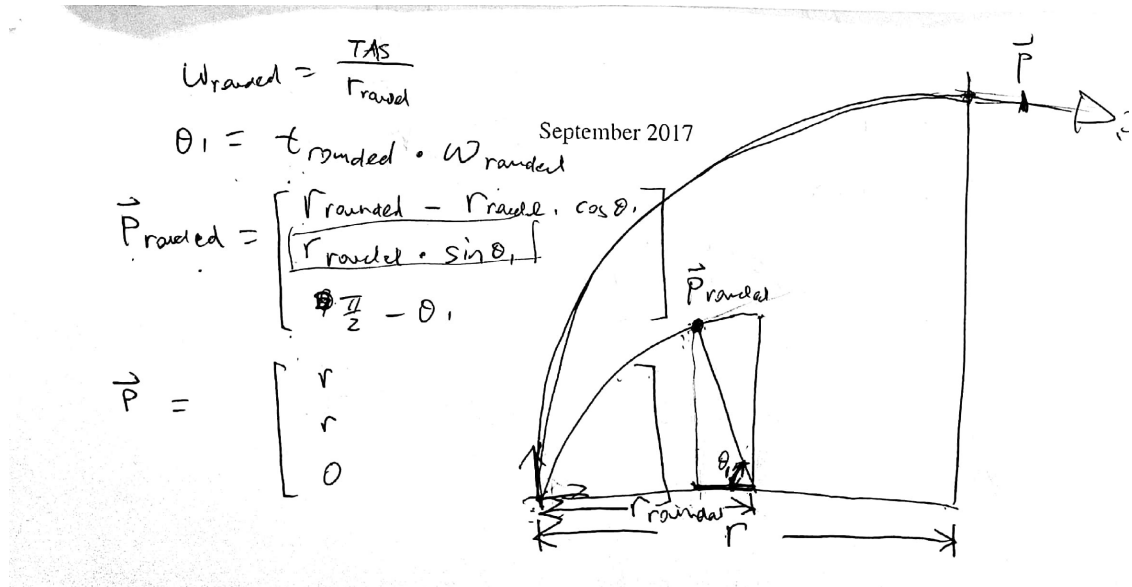
```
TAS = 70; %m/s
g = 9.8039; %gravitational constant at Boston (m/s^2)
bank = 20*pi/180; %radians
syms r;
% Angle of bank formula
% http://www.luizmonteiro.com/Article_Bank_Angle_for_Std_Rate_01.aspx
turn_radius = vpasolve(bank-atan(((TAS^2)/r)/g),r,1000); % m
time = vpa(((90*pi/180)*turn_radius)/TAS, 6) % s
```

time =

30.8144

1b

ignoring the curvature of the earth



```

g_rounded = 10; %(m/s^2)
turn_radius_rounded = vpasolve(bank-atan(((TAS^2)/r)/
g_rounded),r,1000); % m
time_rounded = round(((90*pi/180)*turn_radius_rounded)/TAS); % s

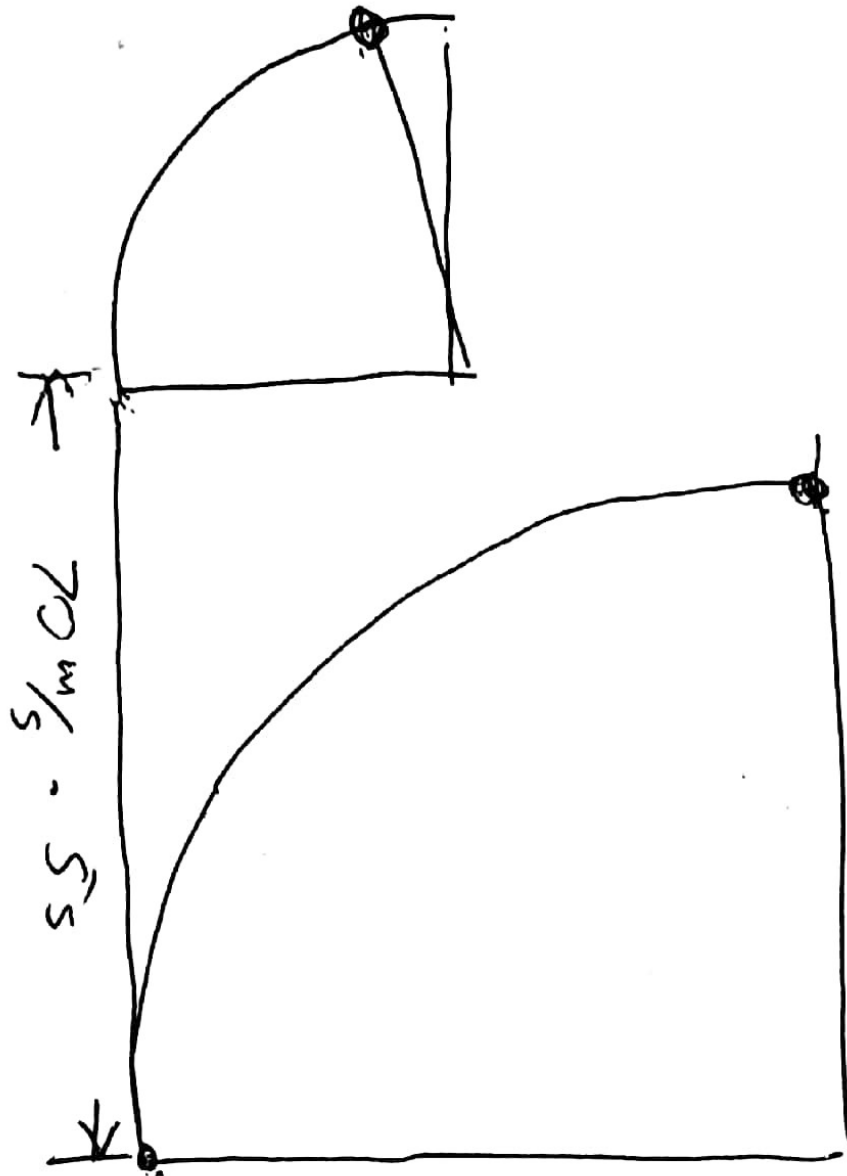
omega_rounded = TAS/turn_radius_rounded; % rad/s
theta_rounded = time_rounded*omega_rounded; % rad
P_rounded = [turn_radius_rounded - turn_radius_rounded *
cos(theta_rounded); % x in m
turn_radius_rounded * sin(theta_rounded); % y in m
pi/2 - theta_rounded]; % heading in rad wrt x axis
P = [turn_radius; turn_radius; 0]; % [m; m; rad]
error = P_rounded - P; % error in [x, y, heading]. unit in [m; m; rad]
vpa(error, 6)

ans =

-41.6345
-27.0086
0.0109239

```

1c



```

P_rounded = [turn_radius_rounded - turn_radius_rounded *
    cos(theta_rounded); % x in m
    turn_radius_rounded * sin(theta_rounded) + TAS*5; % y in m
    pi/2 - theta_rounded]; % heading in rad wrt x axis
P = [turn_radius; turn_radius; 0]; % [m; m; rad]
error = P_rounded - P; % error in [x, y, heading]. unit in [m; m; rad]
vpa(error, 6)

```

ans =

-41.6345

322.991

0.0109239

2a

$$\|P - P_i\| + \epsilon_c = p_i \quad (2.1)$$

$$\|P - P_i\| = \sqrt{(P_x - P_{xi})^2 + (P_y - P_{yi})^2 + (P_z - P_{zi})^2} \quad (2.2)$$

$$\left. \begin{aligned} \|P - P_1\| + \epsilon_c - p_1 &= 0 \\ \|P - P_2\| + \epsilon_c - p_2 &= 0 \\ \|P - P_3\| + \epsilon_c - p_3 &= 0 \\ \|P - P_4\| + \epsilon_c - p_4 &= 0 \end{aligned} \right\} \quad (2.3)$$

$$\left. \begin{aligned} f_1(P, \epsilon_c) &= \|P - P_1\| + \epsilon_c - p_1 = 0 \\ f_2(P, \epsilon_c) &= \|P - P_2\| + \epsilon_c - p_2 = 0 \\ f_3(P, \epsilon_c) &= \|P - P_3\| + \epsilon_c - p_3 = 0 \\ f_4(P, \epsilon_c) &= \|P - P_4\| + \epsilon_c - p_4 = 0 \end{aligned} \right\} \quad (2.4)$$

$$\frac{\partial f_1}{\partial \epsilon_c} = \frac{\partial}{\partial \epsilon_c} (\|P - P_1\| + \epsilon_c - p_1) = 1$$

$$\frac{\partial f_2}{\partial \epsilon_c} = \frac{\partial}{\partial \epsilon_c} (\|P - P_2\| + \epsilon_c - p_2) = 1$$

$$\frac{\partial f_3}{\partial \epsilon_c} = \frac{\partial}{\partial \epsilon_c} (\|P - P_3\| + \epsilon_c - p_3) = 1$$

$$\frac{\partial f_4}{\partial \epsilon_c} = \frac{\partial}{\partial \epsilon_c} (\|P - P_4\| + \epsilon_c - p_4) = 1$$

$$\left. \begin{aligned} f_1(P, \epsilon_c) + \begin{bmatrix} \frac{(P_x^0 - P_{x1})}{\|P^0 - P_1\|} & \frac{(P_y^0 - P_{y1})}{\|P^0 - P_1\|} & \frac{(P_z^0 - P_{z1})}{\|P^0 - P_1\|} & 1 \end{bmatrix} dP &= 0 \\ f_2(P, \epsilon_c) + \begin{bmatrix} \frac{(P_x^0 - P_{x2})}{\|P^0 - P_2\|} & \frac{(P_y^0 - P_{y2})}{\|P^0 - P_2\|} & \frac{(P_z^0 - P_{z2})}{\|P^0 - P_2\|} & 1 \end{bmatrix} dP &= 0 \\ f_3(P, \epsilon_c) + \begin{bmatrix} \frac{(P_x^0 - P_{x3})}{\|P^0 - P_3\|} & \frac{(P_y^0 - P_{y3})}{\|P^0 - P_3\|} & \frac{(P_z^0 - P_{z3})}{\|P^0 - P_3\|} & 1 \end{bmatrix} dP &= 0 \\ f_4(P, \epsilon_c) + \begin{bmatrix} \frac{(P_x^0 - P_{x4})}{\|P^0 - P_4\|} & \frac{(P_y^0 - P_{y4})}{\|P^0 - P_4\|} & \frac{(P_z^0 - P_{z4})}{\|P^0 - P_4\|} & 1 \end{bmatrix} dP &= 0 \end{aligned} \right\} \quad (2.5)$$

* ϵ_c is included in dP

$$\begin{bmatrix} \frac{(P_x^0 - P_{x1})}{\|P^0 - P_1\|} & \frac{(P_y^0 - P_{y1})}{\|P^0 - P_1\|} & \frac{(P_z^0 - P_{z1})}{\|P^0 - P_1\|} & 1 \\ \frac{(P_x^0 - P_{x2})}{\|P^0 - P_2\|} & \frac{(P_y^0 - P_{y2})}{\|P^0 - P_2\|} & \frac{(P_z^0 - P_{z2})}{\|P^0 - P_2\|} & 1 \\ \frac{(P_x^0 - P_{x3})}{\|P^0 - P_3\|} & \frac{(P_y^0 - P_{y3})}{\|P^0 - P_3\|} & \frac{(P_z^0 - P_{z3})}{\|P^0 - P_3\|} & 1 \\ \frac{(P_x^0 - P_{x4})}{\|P^0 - P_4\|} & \frac{(P_y^0 - P_{y4})}{\|P^0 - P_4\|} & \frac{(P_z^0 - P_{z4})}{\|P^0 - P_4\|} & 1 \end{bmatrix} dP = \begin{bmatrix} p_1 - \epsilon_c - \|P^0 - P_1\| \\ p_2 - \epsilon_c - \|P^0 - P_2\| \\ p_3 - \epsilon_c - \|P^0 - P_3\| \\ p_4 - \epsilon_c - \|P^0 - P_4\| \end{bmatrix} \quad \bullet \text{ final eqns}$$

Iterative code:

- 1) Choose initial guess $\mathbf{p}_r^0, \epsilon_c$.
- 2) Choose ϵ_1, ϵ_2 , and $n_{iter-max}$.
- 3) set $n := 0$.
- 4) while $(n < n_{iter-max})$ do
- 5) set $\mathbf{1}_i = [(\mathbf{p}_r^n - \mathbf{p}_i) / \|\mathbf{p}_r^n - \mathbf{p}_i\|, 1]$ for $i = 1, 2, 3, 4$
- 6) solve the linear equation

$$\begin{bmatrix} \mathbf{p}_1 - \epsilon_c - \|\mathbf{p}_r^n - \mathbf{p}_1\| \\ \mathbf{p}_2 - \epsilon_c - \|\mathbf{p}_r^n - \mathbf{p}_2\| \\ \mathbf{p}_3 - \epsilon_c - \|\mathbf{p}_r^n - \mathbf{p}_3\| \\ \mathbf{p}_4 - \epsilon_c - \|\mathbf{p}_r^n - \mathbf{p}_4\| \end{bmatrix} = \begin{bmatrix} \mathbf{1}_1^T \\ \mathbf{1}_2^T \\ \mathbf{1}_3^T \\ \mathbf{1}_4^T \end{bmatrix} \Delta \mathbf{p}$$
 to obtain $\Delta \mathbf{p}$
- 7) set $\mathbf{p}_r^{n+1} := \mathbf{p}_r^n + \Delta \mathbf{p} (1:3, :)$
- 8) set $\epsilon_c^{n+1} = \epsilon_c^n + \Delta \mathbf{p} (4, :)$
- 9) set $n := n + 1$
- 10) if $\|\Delta \mathbf{p}\| < \epsilon_1$ or $\sqrt{\sum_{i=1}^4 (\|\mathbf{p}_r^n - \mathbf{p}_i\| + \epsilon_c - r_i)^2} < \epsilon_2$ then
- 11) break
- 12) Return $\mathbf{p} = \mathbf{p}_r^n$
- 13) Return $\epsilon_c = \epsilon_c^n$

2b

```
function [receiver_position, epsilon_c] =
    trilat_clockbias(satellite_positions, measured_ranges, initial_guess,
        e1, e2, max_iter)
%TRILAT_CLOCKBIAS Trilateration for radio ranging

big_fat_one = @(p_r_guess, p_sat) (p_r_guess - p_sat) / norm(p_r_guess
    - p_sat);
receiver_position = initial_guess;
epsilon_c = 0;
```

```
for iter = 1:max_iter
    A = cell2mat(cellfun(@(p_sat) [big_fat_one(receiver_position,
p_sat); 1]', num2cell(satellite_positions, 1), 'UniformOutput',
false)'));
    b = measured_ranges - ones(size(measured_ranges,1),1)*epsilon_c
- cellfun(@norm, num2cell(bsxfun(@minus, receiver_position,
satellite_positions), 1))';
    d_p = A \ b;
    receiver_position = receiver_position + d_p(1:end-1, :);
    epsilon_c = epsilon_c + d_p(end, :);
    if norm(d_p) < e1 || norm(b) < e2
        return
    end
end
warning('reached maximum iteration')
b
end

sat_positions = ...
    [90 -70 95;
    -90 -5 60;
    -75 -20 85;
    60 80 -70;
    80 90 100;
    35 -45 55;
    -80 65 20]'; % km

rho = ...
    [139.533;
    118.995;
    120.099;
    126.191;
    120.315;
    79.1232;
    110.215]; % km

[receiver_position, epsilon_c] = trilat_clockbias(sat_positions, rho,
zeros(3,1), 1e-6, 1e-6, 1e3) % all in km

c = 299792458; %m/s

clock_bias = epsilon_c * 1e3 /c % sec

receiver_position =

    17.0000
    23.0000
    29.0000

epsilon_c =
```

4.1298

clock_bias =

1.3775e-05

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