Dynamics of Structures

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An Introduction to Dynamics of Structures

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Outline

Part 1

Introduction

Characteristics of a Dynamical Problem

Formulation of a Dynamical Problem

Formulation of the equations of motion

Part 2

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Dynamics of

Definitions

Let's start with some definitions

Dynamic adj., constantly changing

Dynamics noun, the branch of Mechanics concerned with

the motion of bodies under the action of forces

Dynamic Loading a Loading that varies over time

Dynamic Response the Response (deflections and stresses) of a

system to a dynamic loading; Dynamic

Responses vary over time

Dynamics of Structures all of the above, applied to a structural

system, i.e., a system designed to stay in equilibrium

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Dynamics of

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Introduction

a Dynamical Problem

Formulation of a Dynamical

the equations of motion

Dynamics of Structures

Our aim is to determine the stresses and deflections that a *dynamic loading* induces in a *structure* that remains in the neighborhood of a point of equilibrium.

Methods of *dynamic analysis* are extensions of the methods of standard static analysis, or to say it better, static analysis is indeed a special case of *dynamic analysis*.

If we restrict ourselves to analysis of linear systems, it is so convenient to use the principle of superposition to study the combined effects of static and dynamic loadings that different methods, of different character, are applied to these different loadings.

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical

Formulation of the equations of motion

Types of Dynamic Analysis

Taking into account linear systems only, we must consider two different definitions of the loading to define two types of dynamic analysis

Deterministic Analysis applies when the time variation of the loading is fully known and we can determine the complete time variation of all the required response quantities

Non-deterministic Analysis applies when the time variation of the loading is essentially random and is known *only* in terms of some *statistics*In a non-deterministic or *stochastic* analysis the

structural response can be known only in terms of statistics of the response quantities

Our focus will be on deterministic analysis

Dynamics of

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Introduction

Characteristics of a Dynamical

Formulation of a Dynamical

Formulation of the equations of motion

Types of Dynamic Loading

Dealing with deterministic dynamic loadings we will study, in order of complexity,

Harmonic Loadings a force is modulated by a harmonic function of time, characterized by a frequency ω and a phase φ :

$$p(t) = p_0 \sin(\omega t - \varphi)$$

Periodic Loadings a periodic loading repeats itself with a fixed period T:

$$p(t) = p_0 f(t)$$
 with $f(t) \equiv f(t + T)$

Non Periodic Loadings here we see two sub-cases,

the loading can be described in terms of an analytical function, e.g.,

$$p(t) = p_o \exp(\alpha t)$$

the loading is measured experimentally, hence it is known only in a discrete set of instants; in this case, we say that we know the loading time-history.

Dynamics of Structures

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Characteristics of a Dynamical Problem

Formulation of a Dynamical

Formulation of the equations of motion

Characteristics of a Dynamical Problem

A dynamical problem is essentially characterized by the relevance of *inertial forces*, arising from the accelerated motion of structural or serviced masses.

A dynamic analysis is *required* only when the inertial forces represent a significant portion of the total load.

On the other hand, if the loads and the deflections are varying *slowly*, a static analysis will provide an acceptable approximation.

We will define slowly

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical Problem

Formulation of the equations of motion

Formulation of a Dynamical Problem

In a structural system the inertial forces depend on the time derivatives of displacements while the elastic forces, equilibrating the inertial ones, depend on the spatial derivatives of the displacements. ... the natural statement of the problem is hence in terms of *partial differential equations*.

In many cases it is however possible to simplify the formulation of the structural dynamic problem to *ordinary* differential equations.

Dynamics of

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Introduction

Characteristics of a Dynamical Problem

Formulation of a Dynamical Problem

Formulation of the equations of motion

Lumped Masses

In many structural problems we can say that the masses are concentrated in a discrete set of *lumped masses* (e.g., in a multi-storey building most of the masses is concentrated at the level of the storeys'floors).

Under this assumption, the analytical problem is greatly simplified:

- 1. the inertial forces are applied only at the lumped masses,
- 2. the only deflections that influence the inertial forces are the deflections of the lumped masses,
- 3. using methods of static analysis we can determine those deflections,

thus consenting the formulation of the problem in terms of a set of ordinary differential equations, one for each relevant component of the inertial forces.

Dynamics of

Giacomo Boffi

Introduction

Characteristics of a Dynamical Problem

Formulation of a Dynamical Problem

Formulation of the equations of

Dynamic Degrees of Freedom

The *dynamic degrees of freedom* (DDOF) in a discretized system are the displacements components of the lumped masses associated with the relevant components of the inertial forces.

If a lumped mass can be regarded as a *point* mass then at most 3 translational DDOFs will suffice to represent the associated inertial force.

On the contrary, if a lumped mass has a discrete volume its inertial force depends also on its rotations (inertial couples) and we need at most 6 DDOFs to represent the mass deflections and the inertial force.

Of course, a continuous system has an infinite number of degrees of freedom.

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical Problem

Formulation of the equations of motion

Generalized Displacements

The lumped mass procedure that we have outlined is effective if a large proportion of the total mass is concentrated in a few points (e.g., in a multi-storey building one can consider a lumped mass for each storey).

When the masses are distributed we can simplify our problem expressing the deflections in terms of a linear combination of assigned functions of position, the coefficients of the linear combination being the *generalized coordinates* (e..g., the deflections of a rectilinear beam can be expressed in terms of a trigonometric series).

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical Problem

Formulation of the equations of motion

Generalized Displacements, cont.

To fully describe a displacement field, we need to combine an infinity of linearly independent *base functions*, but in practice a good approximation can be achieved using just a small number of functions and degrees of freedom.

Even if the method of generalized coordinates has its beauty, we must recognize that for each different problem we have to derive an ad hoc formulation, with an evident loss of generality.

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical Problem

Formulation of a Dynamical Problem

Formulation of the equations of

Finite Element Method

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical Problem

Formulation of the equations of motion

Finite Element Method

digital computer.

- ▶ In the FEM, the structure is subdivided in a number of non-overlapping pieces, called the *finite elements*, delimited by common *nodes*.
- ► The FEM uses *piecewise approximations* (i.e., local to each element) to the field of displacements.

The finite elements method (FEM) combines aspects of lumped

reliable method of analysis, that can be easily programmed on a

mass and generalized coordinates methods, providing a simple and

- ▶ In each *element* the displacement field is derived from the displacements of the *nodes* that surround each particular element, using *interpolating functions*.
- ➤ The displacement, deformation and stress fields in each element, as well as the inertial forces, can thus be expressed in terms of the unknown *nodal displacements*.
- ► The *nodal displacements* are the dynamical DOFs of the FEM model

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical Problem

Formulation of the equations of motion

Finite Element Method

Some of the most prominent advantages of the FEM method are

- 1. The desired level of approximation can be achieved by further subdividing the structure.
- 2. The resulting equations are only loosely coupled, leading to an easier computer solution.
- For a particular type of finite element (e.g., beam, solid, etc)
 the procedure to derive the displacement field and the element
 characteristics does not depend on the particular geometry of
 the elements, and can easily be implemented in a computer
 program.

Dynamics of

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Characteristics of a Dynamical Problem

Formulation of a Dynamical Problem

Formulation of the equations of

Writing the equation of motion

In a deterministic dynamic analysis, given a prescribed load, we want to evaluate the displacements in each instant of time.

In many cases a limited number of DDOFs gives a sufficient accuracy; further, the dynamic problem can be reduced to the determination of the time-histories of some selected component of the response.

The mathematical expressions, ordinary or partial differential equations, that we are going to write express the *dynamic* equilibrium of the structural system and are known as the *Equations* of *Motion* (EOM).

The solution of the EOM gives the requested displacements. The formulation of the EOM is the most important, often the most difficult part of a dynamic analysis.

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of Dynamical Problem

Formulation of the equations of motion

Writing the EOM, cont.

We have a choice of techniques to help us in writing the EOM, namely:

- ▶ the D'Alembert Principle,
- ▶ the Principle of Virtual Displacements,
- ▶ the Variational Approach.

Dynamics of

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Introduction

Characteristics of a Dynamical Problem

Formulation of Dynamical

Formulation of the equations of motion

D'Alembert principle

By Newton's II law of motion, for any particle the rate of change of momentum is equal to the external force,

$$\vec{p}(t) = \frac{d}{dt} (m \frac{d\vec{u}}{dt}),$$

where $\vec{u}(t)$ is the particle displacement.

In structural dynamics, we may regard the mass as a constant, and thus write

$$\vec{p}(t) = m\ddot{\vec{u}},$$

where each operation of differentiation with respect to time is denoted with a dot.

If we write

$$\vec{p}(t) - m\ddot{\vec{u}} = 0$$

and interpret the term $-m\ddot{\vec{u}}$ as an *Inertial Force* that contrasts the acceleration of the particle, we have an equation of equilibrium for the particle.

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical Problem

Formulation of a Dynamical

Formulation of the equations of motion

D'Alembert principle, cont.

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical

Formulation of the equations of motion

The concept that a mass develops an inertial force opposing its acceleration is known as the D'Alembert principle, and using this principle we can write the EOM as a simple equation of equilibrium. The term $\vec{p}(t)$ must comprise each different force acting on the particle, including the reactions of kinematic or elastic constraints, internal forces and external, autonomous forces. In many simple problems, D'Alembert principle is the most direct

and convenient method for the formulation of the EOM.

Principle of virtual displacements

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical Problem

Formulation of a Dynamical Problem

Formulation of the equations of motion

In a reasonably complex dynamic system, with e.g. articulated rigid bodies and external/internal constraints, the direct formulation of the *EOM* using D'Alembert principle may result difficult. In these cases, application of the *Principle of Virtual Displacements* is very convenient, because the reactive forces do not enter the equations of motion, that are directly written in terms of the motions compatible with the restraints/constraints of the system.

Principle of Virtual Displacements, cont.

For example, considering an assemblage of rigid bodies, the *pvd* states that necessary and sufficient condition for equilibrium is that, for every *virtual displacement* (i.e., any infinitesimal displacement compatible with the restraints) the total work done by all the external forces is zero.

For an assemblage of rigid bodies, writing the EOM requires

- 1. to identify all the external forces, comprising the inertial forces, and to express their values in terms of the *ddof*;
- 2. to compute the work done by these forces for different virtual displacements, one for each *ddof*;
- 3. to equate to zero all these work expressions.

The *pvd* is particularly convenient also because we have only scalar equations, even if the forces and displacements are of vectorial nature.

Dynamics of Structures

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Characteristics of a Dynamical Problem

Formulation of a Dynamical

Formulation of the equations of motion

Variational approach

Variational approaches do not consider directly the forces acting on the dynamic system, but are concerned with the variations of kinetic and potential energy and lead, as well as the *pvd*, to a set of scalar equations.

For example, the equation of motion of a generical system can be derived in terms of the Lagrangian function, $\mathcal{L}=T-V$ where T and V are, respectively, the kinetic and the potential energy of the system expressed in terms of a vector \vec{q} of indipendent coordinates

$$\frac{d}{dt}\left(\frac{\partial \mathcal{L}}{\partial \dot{q}_i}\right) = \frac{\partial \mathcal{L}}{\partial q_i}, \qquad i = 1, \dots, N.$$

The method to be used in a particular problem is mainly a matter of convenience (and, to some extent, of personal taste).

Dynamics of Structures

Giacomo Boffi

Introduction

Characteristics of a Dynamical

Formulation of a Dynamical Problem

Formulation of the equations of motion

1 DOF System

Structural dynamics is all about the motion of a system in the neighborhood of a point of equilibrium.

We'll start by studying the most simple of systems, a single degree of freedom system, without external forces, subjected to a perturbation of the equilibrium.

If our system has a *constant* mass m and it's subjected to a generical, non-linear, internal force $F = F(y, \dot{y})$, where y is the displacement and \dot{y} the velocity of the particle, the equation of motion is

$$\ddot{y} = \frac{1}{m}F(y,\dot{y}) = f(y,\dot{y}).$$

It is difficult to integrate the above equation in the general case, but it's easy when the motion occurs in a small neighborhood of the equilibrium position.

Dynamics of

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

1 DOF System, cont.

In a position of equilibrium, $y_{eq.}$, the velocity and the acceleration are zero, and hence $f(y_{eq.}, 0) = 0$.

The force can be linearized in a neighborhood of $y_{eq.}$, 0:

$$f(y,\dot{y}) = f(y_{\text{eq.}},0) + \frac{\partial f}{\partial y}(y - y_{\text{eq.}}) + \frac{\partial f}{\partial \dot{y}}(\dot{y} - 0) + O(y,\dot{y}).$$

Assuming that $O(y, \dot{y})$ is small in a neighborhood of $y_{eq.}$, we can write the equation of motion

$$\ddot{x} + a\dot{x} + bx = 0$$

where
$$x=y-y_{\rm eq.}$$
, $a=-\left.\frac{\partial f}{\partial \dot{y}}\right|_{\dot{y}=0}$ and $b=-\left.\frac{\partial f}{\partial y}\right|_{y=y_{\rm eq}}$.

In an infinitesimal neighborhood of $y_{eq.}$, the equation of motion can be studied in terms of a linear, constant coefficients differential equation of second order.

Dynamics of

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1 DOF System

Free vibrations of a SDOF system

1 DOF System, cont.

A linear constant coefficient differential equation has the homogeneous integral $x = A \exp(st)$, that substituted in the equation of motion gives

$$s^2 + as + b = 0$$

whose solutions are

$$s_{1,2} = -rac{a}{2} \mp \sqrt{rac{a^2}{4} - b}.$$

The general integral is hence

$$x(t) = A_1 \exp(s_1 t) + A_2 \exp(s_2 t).$$

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

1 DOF System, cont.

Given that for a free vibration problem A_1 , A_2 are given by the initial conditions, the nature of the solution depends on the sign of the real part of s_1 , s_2 , because $s_i = r_i + iq_i$ and

$$\exp(s_i t) = \exp(i q_i t) \exp(r_i t).$$

If one of the $r_i > 0$, the response grows infinitely over time, even for an infinitesimal perturbation of the equilibrium, so that in this case we have an *unstable equilibrium*.

If both $r_i < 0$, the response decrease over time, so we have a *stable* equilibrium.

Finally, if both $r_i = 0$ the roots s_i are purely imaginary and the response is harmonic with constant amplitude.

Dynamics of

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

1 DOF System, cont.

The roots being

$$s_{1,2} = -rac{a}{2} \mp \sqrt{rac{a^2}{4} - b},$$

- ▶ if a > 0 and b > 0 both roots are negative or complex conjugate with negative real part, the system is asymptotically stable,
- ightharpoonup if a=0 and b>0, the roots are purely imaginary, the equilibrium is indifferent, the oscillations are harmonic,
- ▶ if a < 0 or b < 0 at least one of the roots has a positive real part, and the system is unstable.

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Giacomo Boffi

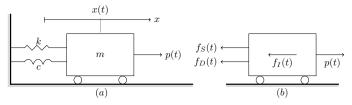
1 DOF System

Free vibrations of a SDOF system

The famous box car

In a single degree of freedom (sdof) system each property, m, a and b, can be conveniently represented in a single physical element

- ▶ The entire mass, m, is concentrated in a rigid block, its position completely described by the coordinate x(t).
- ▶ The energy-loss (the $a\dot{x}$ term) is represented by a massless damper, its damping constant being c.
- ► The elastic resistance to displacement (bx) is provided by a massless spring of stiffness k
- For completeness we consider also an external loading, the time-varying force p(t).



Dynamics of Structures

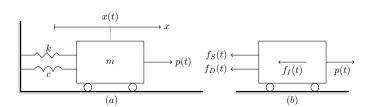
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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Equation of motion of the basic dynamic system



The equation of motion can be written using the D'Alembert Principle, expressing the equilibrium of all the forces acting on the mass *including* the inertial force.

The forces are the external force, p(t), positive in the direction of motion and the resisting forces, i.e., the inertial force $f_{\rm I}(t)$, the damping force $f_{\rm D}(t)$ and the elastic force, $f_{\rm S}(t)$, that are opposite to the direction of the acceleration, velocity and displacement.

The equation of motion, merely expressing the equilibrium of these forces, writing the resisting forces and the external force across the equal sign

$$f_{\rm I}(t) + f_{\rm D}(t) + f_{\rm S}(t) = p(t)$$

Dynamics of

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

EOM of the basic dynamic system, cont.

According to D'Alembert principle, the inertial force is the product of the mass and acceleration

$$f_{\rm I}(t) = m\ddot{x}(t).$$

Assuming a viscous damping mechanism, the damping force is the product of the damping constant c and the velocity,

$$f_{\rm D}(t) = c \dot{x}(t).$$

Finally, the elastic force is the product of the elastic stiffness k and the displacement,

$$f_{\rm S}(t) = k x(t).$$

Dynamics of

Giacomo Boffi

1 DOF System

Free vibrations of

EOM of the basic dynamic system, cont.

Dynamics of Structures

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1 DOF System

Free vibrations of a SDOF system Free vibrations of

Dynamics of

Giacomo Boffi 1 DOF System Free vibrations o a SDOF system

Giacomo Boffi 1 DOF System

Dynamics of

a SDOF system

The differential equation of dynamic equilibrium

$$f_{\rm I}(t) + f_{\rm D}(t) + f_{\rm S}(t) = m \ddot{x}(t) + c \dot{x}(t) + k x(t) = p(t).$$

The resisting forces in the EoM

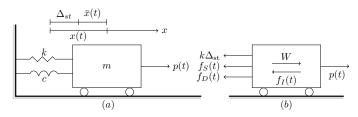
$$f_{\rm I}(t) + f_{\rm D}(t) + f_{\rm S}(t) = p(t)$$

are proportional to the deflection x(t) or one of its time derivatives, $\dot{x}(t), \ddot{x}(t).$

The equation of motion is a linear differential equation of the second order, with constant coefficients.

The resisting forces are, by convention, positive when opposite to the direction of motion, i.e., resisting the motion.

Influence of static forces



Considering the presence of a constant force W, the EOM is

$$m\ddot{x}(t) + c\dot{x}(t) + kx(t) = p(t) + W.$$

Expressing the displacement as the sum of a constant, static displacement and a dynamic displacement,

$$x(t) = \Delta_{st} + \bar{x}(t),$$

and substituting in the EOM we have

$$m\ddot{x}(t) + c\dot{x}(t) + k\Delta_{st} + k\bar{x}(t) = p(t) + W.$$

Influence of static forces, cont.

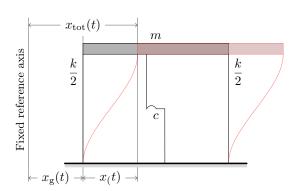
Recognizing that $k \Delta_{st} = W$ (so that the two terms, on opposite sides of the equal sign, cancel each other), that $\dot{x} \equiv \dot{\bar{x}}$ and that $\ddot{x} \equiv \ddot{\bar{x}}$ the *EOM* can be written as

$$m\ddot{\bar{x}}(t)+c\dot{\bar{x}}(t)+k\bar{x}(t)=p(t).$$

The equation of motion expressed with reference to the static equilibrium position is not affected by static forces. For this reasons, all displacements in further discussions will be referenced from the equilibrium position and denoted, for simplicity, with x(t).

Note that the total displacements, stresses. etc. are influenced by the static forces, and must be computed using the superposition of effects.

Influence of support motion



Displacements, deformations and stresses in a structure are induced also by a motion of its support.

Important examples of support motion are the motion of a building foundation due to earthquake and the motion of the base of a piece of equipment due to vibrations of the building in which it is housed.

Dynamics of Structures

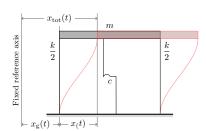
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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Influence of support motion, cont.



Considering a support motion $x_{\rm g}(t)$, defined with respect to a inertial frame of reference, the total displacement is

$$x_{\rm tot}(t) = x_{\rm g}(t) + x(t)$$

and the total acceleration is

$$\ddot{x}_{\rm tot}(t) = \ddot{x}_{\rm g}(t) + \ddot{x}(t).$$

While the elastic and damping forces are still proportional to *relative* displacements and velocities, the inertial force is proportional to the total acceleration,

$$f_{\mathrm{I}}(t) = -m\ddot{x}_{\mathrm{tot}}(t) = m\ddot{x}_{\mathrm{g}}(t) + m\ddot{x}(t).$$

Writing the EOM for a null external load, p(t) = 0, is hence

$$m\ddot{x}_{\mathrm{tot}}(t) + c\dot{x}(t) + kx(t) = 0,$$
 or,
 $m\ddot{x}(t) + c\dot{x}(t) + kx(t) = -m\ddot{x}_{\mathrm{g}}(t) \equiv p_{\mathrm{eff}}(t).$

Support motion is sufficient to excite a dynamic system: $p_{\text{eff}}(t) = -m \ddot{x}_{\text{g}}(t)$.

Dynamics of

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of

Free Vibrations

The equation of motion,

$$m\ddot{x}(t) + c\dot{x}(t) + kx(t) = p(t)$$

is a linear differential equation of the second order, with constant coefficients.

Its solution can be expressed in terms of a superposition of a particular solution, depending on p(t), and a free vibration solution, that is the solution of the so called homogeneous problem, where p(t) = 0.

In the following, we will study the solution of the homogeneous problem, the so-called *homogeneous* or *complementary* solution, that is the *free vibrations* of the SDOF after a perturbation of the position of equilibrium.

Dynamics of

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of an undamped system

An undamped system, where c=0 and no energy dissipation takes place, is just an ideal notion, as it would be a realization of *motus* perpetuum. Nevertheless, it is an useful idealization. In this case, the homogeneous equation of motion is

$$m\ddot{x}(t) + kx(t) = 0$$

which solution is of the form $\exp st$; substituting this solution in the above equation we have

$$(k+s^2m)\exp st=0$$

noting that $\exp st \neq 0$, we finally have

$$(k+s^2m)=0 \Rightarrow s=\pm\sqrt{-\frac{k}{m}}$$

As m and k are positive quantities, s must be purely imaginary.

Dynamics of Structures

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Undamped Free Vibrations

Introducing the *natural* circular frequency ω_n

$$\omega_{\mathsf{n}}^2 = \frac{k}{m},$$

the solution of the algebraic equation in s is

$$s = \pm \sqrt{-\frac{k}{m}} = \pm \sqrt{-1} \sqrt{\frac{k}{m}} = \pm i \sqrt{\omega_n^2} = \pm i \omega_n$$

where $i=\sqrt{-1}$ and the general integral of the homogeneous equation is

$$x(t) = G_1 \exp(i\omega_n t) + G_2 \exp(-i\omega_n t).$$

The solution has an imaginary part?

Dynamics of

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Undamped Free Vibrations

The solution is derived from the general integral imposing the (real) initial conditions

$$x(0) = x_0, \quad \dot{x}(0) = \dot{x}_0$$

Evaluating x(t) for t = 0 and substituting in (39), we have

$$\begin{cases} G_1 + G_2 = x_0 \\ i\omega_n G_1 - i\omega_n G_2 = \dot{x}_0 \end{cases}$$

Solving the linear system we have

$$G_1 = \frac{ix_0 + \dot{x}_0/\omega_n}{2i}, \qquad G_2 = \frac{ix_0 - \dot{x}_0/\omega_n}{2i},$$

substituting these values in the general solution and collecting x_0 and \dot{x}_0 , we finally find

$$x(t) = \frac{\exp(i\omega_n t) + \exp(-i\omega_n t)}{2} x_0 + \frac{\exp(i\omega_n t) - \exp(-i\omega_n t)}{2i} \frac{\dot{x}_0}{\omega_n}$$

Dynamics of

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Undamped Free Vibrations

Dynamics of Structures

Giacomo Boffi

Free vibrations of a SDOF system

Free vibrations of

 $x(t) = \frac{\exp(i\omega_n t) + \exp(-i\omega_n t)}{2} x_0 + \frac{\exp(i\omega_n t) - \exp(-i\omega_n t)}{2i} \frac{\dot{x}_0}{\omega_n}$

Using the Euler formulas relating the imaginary argument exponentials and the trigonometric functions, can be rewritten in terms of the elementary trigonometric functions

$$x(t) = x_0 \cos(\omega_n t) + (\dot{x}_o/\omega_n) \sin(\omega_n t).$$

Considering that for every conceivable initial conditions we can use the above representation, it is indifferent, and perfectly equivalent, to represent the general integral either in the form of exponentials of imaginary argument or as a linear combination of sine and cosine of circular frequency ω_n

Undamped Free Vibrations

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Dynamics of

Free vibrations of a SDOF system

expanding the product and evidencing the imaginary part of the response we have

 $x(t) = (A + iB)(\cos \omega_n t + i \sin \omega_n t) + (C - iD) \times (\cos \omega_n t - i \sin \omega_n t)$

$$\mathcal{I}(x) = i \left(A \sin \omega_{n} t + B \cos \omega_{n} t - C \sin \omega_{n} t - D \cos \omega_{n} t \right).$$

Imposing that $\mathcal{I}(x) = 0$, i.e., that the response is real, we have

Otherwise, using the identity $\exp(\pm i\omega_n t) = \cos \omega_n t \pm i \sin \omega_n t$

$$(A-C)\sin \omega_n t + (B-D)\cos \omega_n t = 0 \rightarrow C=A, D=B.$$

Substituting in x(t) eventually we have

$$x(t) = 2A\cos(\omega_n t) - 2B\sin(\omega_n t).$$

Free vibrations of

Undamped Free Vibrations

Dynamics of

Giacomo Boffi

a SDOF system

Free vibrations of

Our preferred representation of the general integral of undamped free vibrations is

$$x(t) = A\cos(\omega_{\rm n}t) + B\sin(\omega_{\rm n}t)$$

For the usual initial conditions, we have already seen that

$$A=x_0, \qquad B=rac{\dot{x}_0}{\omega_n}.$$

Undamped Free Vibrations

Dynamics of Structures

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Sometimes we prefer to write x(t) as a single harmonic, introducing a *phase difference* ϕ so that the amplitude of the motion, C, is put in evidence:

$$x(t) = C\cos(\omega_n t - \varphi) = C(\cos\omega_n t \cos\varphi + \sin\omega_n t \sin\varphi)$$

= $A\cos\omega_n t + B\sin\omega_n t$

From $A=C\cos\varphi$ and $B=C\sin\varphi$ we have $\tan\varphi=B/A$, from $A^2+B^2=C^2(\cos^2\varphi+\sin^2\varphi)$ we have $C=\sqrt{A^2+B^2}$ and eventually

$$x(t) = C\cos(\omega_{
m n} t - arphi), \quad {
m with} egin{cases} C = \sqrt{A^2 + B^2} \ arphi = {
m arctan}(B/A) \end{cases}$$

Undamped Free Vibrations

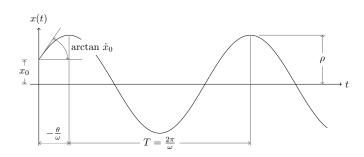
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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system



It is worth noting that the coefficients A, B and C have the dimension of a length, the coefficient ω_n has the dimension of the reciprocal of time and that the coefficient φ is an angle, or in other terms is adimensional.

Behavior of Damped Systems

Dynamics of Structures

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system Under-critically damped SDOF

damped SDOF
Critically damped
SDOF
Over-critically damped
SDOF

The viscous damping modifies the response of a *sdof* system introducing a decay in the amplitude of the response. Depending on the amount of damping, the response can be oscillatory or not. The amount of damping that separates the two behaviors is denoted as *critical damping*.

The solution of the EOM

The equation of motion for a free vibrating damped system is

$$m\ddot{x}(t) + c\dot{x}(t) + kx(t) = 0,$$

substituting the solution exp st in the preceding equation and simplifying, we have that the parameter s must satisfy the equation

$$m s^2 + c s + k = 0$$

or, after dividing both members by m,

$$s^2 + \frac{c}{m}s + \omega_n^2 = 0$$

whose solutions are

$$s = -rac{c}{2m} \mp \sqrt{\left(rac{c}{2m}
ight)^2 - \omega_{
m n}^2} = \omega_{
m n} \left(-rac{c}{2m\omega_{
m n}} \mp \sqrt{\left(rac{c}{2m\omega_{
m n}}
ight)^2 - 1}
ight).$$

Dynamics of Structures

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Free vibrations of a SDOF system

Free vibrations of a damped system

Critically damped SDOF

Over-critically damped

Critical Damping

The behavior of the solution of the free vibration problem depends of course on the sign of the radicand $\Delta = \left(\frac{c}{2m\omega_{\rm n}}\right)^2 - 1$:

 $\Delta < 0$ the roots s are complex conjugate,

 $\Delta = 0$ the roots are identical, double root,

 $\Delta > 0$ the roots are real.

The value of c that make the radicand equal to zero is known as the critical damping,

 $c_{\rm cr} = 2m\omega_{\rm n} = 2\sqrt{mk}$.

Dynamics of

Giacomo Boffi

Free vibrations of a SDOF system

Free vibrations of a damped system

Under-critically damped SDOF Critically damped

Over-critically damped

Critical Damping

A single degree of freedom system is denoted as critically damped, under-critically damped or over-critically damped depending on the value of the damping coefficient with respect to the critical damping.

Typical building structures are undercritically damped.

Dynamics of

Giacomo Boffi

a SDOF system

Free vibrations of a damped system

Under-critically damped SDOF Critically damped SDOF

Over-critically damped SDOF

Damping Ratio

If we introduce the ratio of the damping to the critical damping, or critical damping ratio ζ ,

$$\zeta = \frac{c}{c_{\rm cr}} = \frac{c}{2m\omega_{\rm n}}, \quad c = \zeta c_{\rm cr} = 2\zeta\omega_{\rm n} m$$

the equation of free vibrations can be rewritten as

$$\ddot{x}(t) + 2\zeta\omega_{n}\dot{x}(t) + \omega_{n}^{2}x(t) = 0$$

and the roots $s_{1,2}$ can be rewritten as

$$s = -\zeta \omega_{\mathsf{n}} \mp \omega_{\mathsf{n}} \sqrt{\zeta^2 - 1}.$$

Dynamics of Structures

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Under-critically damped SDOF Critically damped SDOF

Over-critically damped

Free Vibrations of Under-critically Damped Systems

We start studying the free vibration response of under-critically damped SDOF, as this is the most important case in structural dynamics.

The determinant being negative, the roots $s_{1,2}$ are

$$s = -\zeta \omega_{\rm n} \mp \omega_{\rm n} \sqrt{-1} \sqrt{1 - \zeta^2} = -\zeta \omega_{\rm n} \mp i \omega_{\rm D}$$

with the position that

$$\omega_{\mathrm{D}} = \omega_{\mathsf{n}} \sqrt{1 - \zeta^2}.$$

is the damped frequency; the general integral of the equation of motion is, collecting the terms in $\exp(-\zeta\omega_n t)$

$$x(t) = \exp(-\zeta \omega_{\rm D} t) [G_1 \exp(-i\omega_{\rm D} t) + G_2 \exp(+i\omega_{\rm D} t)]$$

Dynamics of

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of

Under-critically

Critically da

Over-critically dampe

Initial Conditions

By imposing the initial conditions, $u(0) = u_0$, $\dot{u}(0) = v_0$, after a bit of algebra we can write the equation of motion for the given initial conditions, namely

$$x(t) = \exp(-\zeta \omega_{\rm n} t) \left[\frac{\exp(i\omega_{\rm D} t) + \exp(-i\omega_{\rm D} t)}{2} u_0 + \frac{\exp(i\omega_{\rm D} t) - \exp(-i\omega_{\rm D} t)}{2i} \frac{v_0 + \zeta \omega_{\rm n} u_0}{\omega_{\rm D}} \right].$$

Using the Euler formulas, we finally have the preferred format of the general integral:

$$x(t) = \exp(-\zeta \omega_{\rm n} t) \left[A \cos(\omega_{\rm D} t) + B \sin(\omega_{\rm D} t) \right]$$

with

$$A = u_0, \qquad B = \frac{v_0 + \zeta \omega_n \ u_0}{\omega_D}.$$

Dynamics of

Giacomo Boffi

1 DOF System

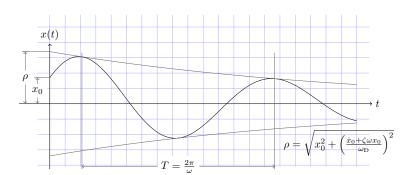
Free vibrations of

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Critically damped SDOF Over-critically damped

The Damped Free Response



Dynamics of Structures

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system

Under-critically

damped SDOF
Critically damped
SDOF
Over-critically dampe

Critically damped SDOF

In this case, $\zeta=1$ and $s_{1,2}=-\omega_{\rm n}$, so that the general integral must be written in the form

$$x(t) = \exp(-\omega_n t)(A + Bt).$$

The solution for given initial condition is

$$x(t) = \exp(-\omega_n t)(u_0 + (v_0 + \omega_n u_0)t),$$

note that, if $v_0 = 0$, the solution asymptotically approaches zero without crossing the zero axis.

Dynamics of

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Free vibrations of a SDOF system

Free vibrations of a damped system

Under-critically damped SDOF Critically damped SDOF

Over-critically dampe

Over-critically damped SDOF

In this case, $\zeta > 1$ and

$$s = -\zeta \omega_{\mathsf{n}} \mp \omega_{\mathsf{n}} \sqrt{\zeta^2 - 1} = -\zeta \omega_{\mathsf{n}} \mp \hat{\omega}$$

where

$$\hat{\omega} = \omega_{\mathsf{n}} \sqrt{\zeta^2 - 1}$$

and, after some rearrangement, the general integral for the over-damped SDOF can be written

$$x(t) = \exp(-\zeta \omega_n t) (A \cosh(\hat{\omega}t) + B \sinh(\hat{\omega}t))$$

Note that:

- ▶ as $\zeta \omega_{\rm n} > \hat{\omega}$, for increasing t the general integral goes to zero, and that
- as for increasing ζ we have that $\hat{\omega} \to \zeta \omega_n$, the velocity with which the response approaches zero slows down for increasing ζ .

Dynamics of Structures

Giacomo Boffi

1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system Under-critically damped SDOF Critically damped

Over-critically damped

Measuring damping

Measuring damping

Dynamics of Structures

Giacomo Boffi

1 DOF System

Free vibrations of

Free vibrations of a damped system Under-critically damped SDOF Critically damped SDOF

Over-critically damped

The real dissipative behavior of a structural system is complex and very difficult to assess.

For convenience, it is customary to express the real dissipative behavior in terms of an *equivalent viscous damping*.

In practice, we *measure* the response of a SDOF structural system under controlled testing conditions and find the value of the viscous damping (or damping ratio) for which our simplified model best matches the measurements.

For example, we could require that, under free vibrations, the real structure and the simplified model exhibit the same decay of the vibration amplitude.

Logarithmic Decrement

Consider a SDOF system in free vibration and two positive peaks, u_n and u_{n+m} , occurring at times $t_n = n(2\pi/\omega_D)$ and $t_{n+m} = (n+m)(2\pi/\omega_D)$.

The ratio of these peaks is

$$\frac{u_n}{u_{n+m}} = \frac{\exp(-\zeta\omega_n n 2\pi/\omega_D)}{\exp(-\zeta\omega_n (n+m) 2\pi/\omega_D)} = \exp(2m\pi\zeta\omega_n/\omega_D)$$

Substituting $\omega_{\rm D}=\omega_{\rm n}\sqrt{1-\zeta^2}$ and taking the logarithm of both members we obtain

$$\ln(\frac{u_n}{u_{n+m}}) = \delta = 2m\pi \frac{\zeta}{\sqrt{1-\zeta^2}}$$

where we have introduced δ , the logarithmic decrement; solving for ζ , we finally get

$$\zeta = \delta \left((2m\pi)^2 + \delta^2 \right)^{-\frac{1}{2}}.$$

Dynamics of

Giacomo Boffi

1 DOF System

a SDOF system

a damped system
Under-critically
damped SDOF

Critically damped

Over-critically damped SDOF

Measuring damping

Recursive Formula for ζ

The equation of the logarithmic decrement,

$$\delta = 2m\pi \frac{\zeta}{\sqrt{1-\zeta^2}}$$

can be formally solved for ζ :

$$\zeta = \delta \, 2m\pi \, \sqrt{1-\zeta^2}$$

obtaining an equation that can be interpreted as generating a sequence

$$\zeta_{n+1} = \delta 2m\pi \sqrt{1 - \zeta_n^2}.$$

Starting our sequence of successive approximations with $\zeta_0=0$ we obtain $\zeta_1=\delta\,2m\pi$ and usually the following iterate $\zeta_2=\delta\,2m\pi\,\sqrt{1-\zeta_1^2}$ has converged to the true value with a number of digits that exceeds the experimental accuracy.

While the recursive formula is useful in itself, it is also useful as a first example of finding better approximations of a system's parameter using an iterative procedure.

Dynamics of

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1 DOF System

Free vibrations of a SDOF system

Free vibrations of a damped system Under-critically damped SDOF Critically damped SDOF Over-critically damped SDOF

Measuring damping