

INVERTED PENDULUM

Bolorkhuu Dariimaa

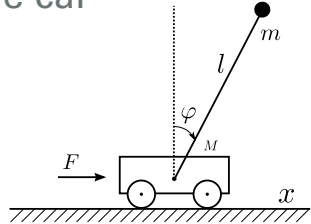
Dresden, 27. November 2014

outline

- introduction
- modelling
- control unit design
- mechanical buildup and electrical components
- assembly
- result
- conspectus and forecast

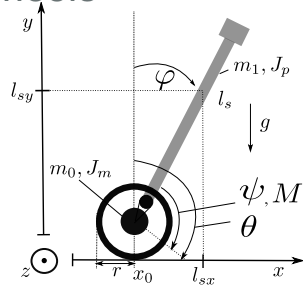
inverted pendulum on the car

- underactuated system
- 2 generalized coordinates (φ, x)
- DOF: 2 (position , angle of the pendulum)



inverted pendulum on wheels

- underactuated system
- 3 coordinates (ψ, φ, θ)
- 1 holonomic constraint
 $\theta = \psi + \varphi$
- DOF: 2 (position of the wheel, angle of the pendulum)



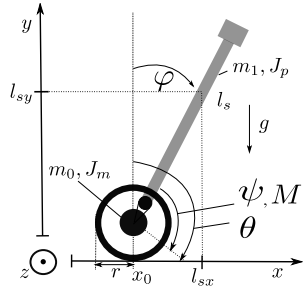
outline

- introduction
- **modelling**
- control unit design
- mechanical buildup and electrical components
- assembly
- result
- conspectus and forecast

derivation of equations of motion

- generalized coordinates
 $[\psi, \varphi]^T =: [\mathbf{q}_1, \mathbf{q}_2]^T =: \mathbf{q}$
- centre of gravity
 $l_{sx} = r(\mathbf{q}_1 + \mathbf{q}_2) + l_s \sin(\mathbf{q}_2)$
 $l_{sy} = l_s \cos(\mathbf{q}_2)$

- φ : absolut angle of the pendulum
- ψ : relativ angle between wheel and pendulum
- θ : rollangle of the wheel



kinetic energy

$$T(q, \dot{q}) = m_0 r^2 \frac{(\dot{q}_1 + \dot{q}_2)^2}{2} + J_m \frac{(\dot{q}_1 + \dot{q}_2)^2}{2} + J_p \frac{\dot{q}_2^2}{2} + \frac{m_1}{2} \left[\left(\frac{d}{dt} l_{sx} \right)^2 + \left(\frac{d}{dt} l_{sy} \right)^2 \right]$$

potential energy

$$U(q) = m_1 g l_{sy}$$

Lagrangian function

$$L = T - U$$

Lagrangian equation

$$\frac{d}{dt} \left(\frac{dL}{d\dot{q}_i} \right) - \frac{dL}{dq_i} = \tau_i, \quad i = 1, 2$$

matrix notation

$$\begin{bmatrix} M_{11}(q) & M_{12}(q) \\ M_{21}(q) & M_{22}(q) \end{bmatrix} \begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \end{bmatrix} + \begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} + \begin{bmatrix} K_1(q) \\ K_2(q) \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix}$$

matrix notation

$$\begin{bmatrix} M_{11}(q) & M_{12}(q) \\ M_{21}(q) & M_{22}(q) \end{bmatrix} \begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \end{bmatrix} + \begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} + \begin{bmatrix} K_1(q) \\ K_2(q) \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix}$$

- mass-matrix

$$\begin{bmatrix} J^* & J^* + l_s m_1 r \cos(q_2) \\ J^* + l_s m_1 r \cos(q_2) & J^* + J_p + m_1 l_s^2 + 2l_s m_1 r \cos(q_2) \end{bmatrix}$$

$$\text{für } J^* = J_m + (m_0 + m_1)r^2$$

matrix notation

$$\begin{bmatrix} \mathbf{M}_{11}(\mathbf{q}) & \mathbf{M}_{12}(\mathbf{q}) \\ \mathbf{M}_{21}(\mathbf{q}) & \mathbf{M}_{22}(\mathbf{q}) \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{q}}_1 \\ \ddot{\mathbf{q}}_2 \end{bmatrix} + \begin{bmatrix} \mathbf{C}_1(\mathbf{q}, \dot{\mathbf{q}}) \\ \mathbf{C}_2(\mathbf{q}, \dot{\mathbf{q}}) \end{bmatrix} + \begin{bmatrix} \mathbf{K}_1(\mathbf{q}) \\ \mathbf{K}_2(\mathbf{q}) \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix}$$

- mass-matrix

$$\begin{bmatrix} J^* & J^* + I_s m_1 r \cos(q_2) \\ J^* + I_s m_1 r \cos(q_2) & J^* + J_p + m_1^2 I_s^2 + 2I_s m_1 r \cos(q_2) \end{bmatrix}$$

für $J^* = J_m + (m_0 + m_1)r^2$

- centrifugal-/coriolis force

$$\begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} = \begin{bmatrix} -I_s m_1 \dot{q}_2^2 r \sin(q_2) \\ -I_s m_1 \dot{q}_2^2 r \sin(q_2) \end{bmatrix}$$

matrix notation

$$\begin{bmatrix} M_{11}(q) & M_{12}(q) \\ M_{21}(q) & M_{22}(q) \end{bmatrix} \begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \end{bmatrix} + \begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} + \begin{bmatrix} K_1(q) \\ K_2(q) \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix}$$

- mass-matrix

$$\begin{bmatrix} J^* & J^* + l_s m_1 r \cos(q_2) \\ J^* + l_s m_1 r \cos(q_2) & J^* + J_p + m_1 l_s^2 + 2l_s m_1 r \cos(q_2) \end{bmatrix}$$

für $J^* = J_m + (m_0 + m_1)r^2$

- centrifugal-/coriolis force

$$\begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} = \begin{bmatrix} -l_s m_1 \dot{q}_2^2 r \sin(q_2) \\ -l_s m_1 \dot{q}_2^2 r \sin(q_2) \end{bmatrix}$$

- link torque via gravitation

$$\begin{bmatrix} K_1(q) \\ K_2(q) \end{bmatrix} = \begin{bmatrix} 0 \\ -g l_s m_1 \sin(q_2) \end{bmatrix}$$

matrix notation

$$\begin{bmatrix} M_{11}(q) & M_{12}(q) \\ M_{21}(q) & M_{22}(q) \end{bmatrix} \begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \end{bmatrix} + \begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} + \begin{bmatrix} K_1(q) \\ K_2(q) \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix}$$

- mass-matrix

$$\begin{bmatrix} J^* & J^* + l_s m_1 r \cos(q_2) \\ J^* + l_s m_1 r \cos(q_2) & J^* + J_p + m_1 l_s^2 + 2l_s m_1 r \cos(q_2) \end{bmatrix}$$

für $J^* = J_m + (m_0 + m_1)r^2$

- centrifugal-/coriolis force

$$\begin{bmatrix} C_1(q, \dot{q}) \\ C_2(q, \dot{q}) \end{bmatrix} = \begin{bmatrix} -l_s m_1 \dot{q}_2^2 r \sin(q_2) \\ -l_s m_1 \dot{q}_2^2 r \sin(q_2) \end{bmatrix}$$

- link torque via gravitation

$$\begin{bmatrix} K_1(q) \\ K_2(q) \end{bmatrix} = \begin{bmatrix} 0 \\ -g l_s m_1 \sin(q_2) \end{bmatrix}$$

- actuation torque

$$\begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix} = \begin{bmatrix} M \\ 0 \end{bmatrix}$$

partial linearization

new virtual input

$$a = \ddot{q}_1$$

inside feedback

$$\tau_1 = \left[M_{11}(q) - M_{12}(q)M_{22}^{-1}(q)M_{21}(q) \right] a - M_{12}(q)M_{22}^{-1}(q) (C_2(q, \dot{q}) + K_2(q, \dot{q})) \\ + C_1(q, \dot{q}) + K_1(q).$$

partial linearized system

$$\ddot{q}_1 = a$$

$$\ddot{q}_2 = -M_{22}^{-1}(q) (C_2(q, \dot{q}) + K_2(q, \dot{q}) + M_{21}(q) a).$$

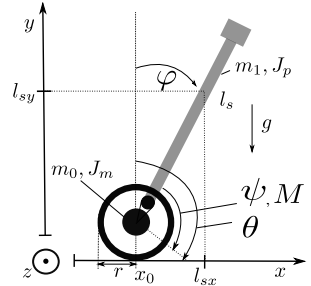
partial linearization

- state vector \mathbb{R}^4
 $\mathbf{x} := [x_1 \ x_2 \ x_3 \ x_4]^T := [\psi \ \varphi \ \dot{\psi} \ \dot{\varphi}]^T$
- the input affine system
 with new input $u = a$

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \mathbf{g}(\mathbf{x})u$$

alternatively

$$\begin{aligned} \dot{x}_1 &= x_3 \\ \dot{x}_2 &= x_4 \\ \dot{x}_3 &= a \\ \dot{x}_4 &= - \mathbf{M}_{22}^{-1}(\mathbf{x}) (\mathbf{C}_2(\mathbf{x}, \dot{\mathbf{x}}) + \mathbf{K}_2(\mathbf{x}, \dot{\mathbf{x}}) + \mathbf{M}_{21}(\mathbf{x}) a) \end{aligned}$$



linearization on the equilibrium point

equilibrium point

$$\mathbf{x}_0 = [x_{1,0} \ x_{2,0} \ x_{3,0} \ x_{4,0}]^T = [0 \ 0 \ 0 \ 0]^T$$

on the state form

$$\begin{bmatrix} \dot{\tilde{x}}_1 \\ \dot{\tilde{x}}_2 \\ \dot{\tilde{x}}_3 \\ \dot{\tilde{x}}_4 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 \\ 0 & a_{42} & 0 & 0 \end{bmatrix} \begin{bmatrix} \tilde{x}_1 \\ \tilde{x}_2 \\ \tilde{x}_3 \\ \tilde{x}_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \\ b_4 \end{bmatrix} \tilde{u}$$

$$a_{42} = \frac{g l_s m_1}{J_p + J_m + l_s^2 m_1 + 2 l_s m_1 r + (m_0 + m_1) r^2}$$

$$b_4 = - \frac{J_m + l_s m_1 r + (m_0 + m_1) r^2}{J_p + J_m + l_s^2 m_1 + 2 l_s m_1 r + (m_0 + m_1) r^2}$$

on the general form

$$\dot{\tilde{\mathbf{x}}}(t) = \mathbf{A} \tilde{\mathbf{x}}(t) + \mathbf{b} \tilde{u}, \quad \tilde{\mathbf{x}}(t_0) = \mathbf{x}_0$$

outline

- introduction
- modelling
- **control unit design**
- mechanical buildup and electrical components
- assembly
- result
- conspectus and forecast

control unit design

kalmanian controllable matrix S

$$S = [b, Ab, A^2b, A^3b] = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & b_4 & 0 & a_{42}b_4 \\ 1 & 0 & 0 & 0 \\ b_4 & 0 & a_{42}b_4 & 0 \end{bmatrix}.$$

determinant

$$\det(S) = -a_{42}^2 \cdot b_4^2, \quad \Rightarrow \text{Rg}(S) = 4.$$

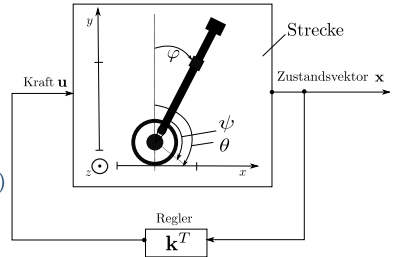
pole of the open plant

$$\det(sI - A) = 0$$

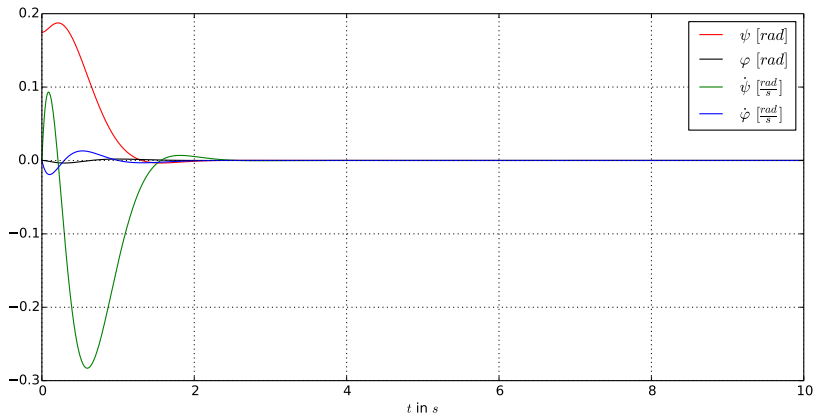
$$s_{1,2}^0 = 0, \quad s_{3,4}^0 = \pm \sqrt{a_{42}}.$$

system is unstable, controller necessary

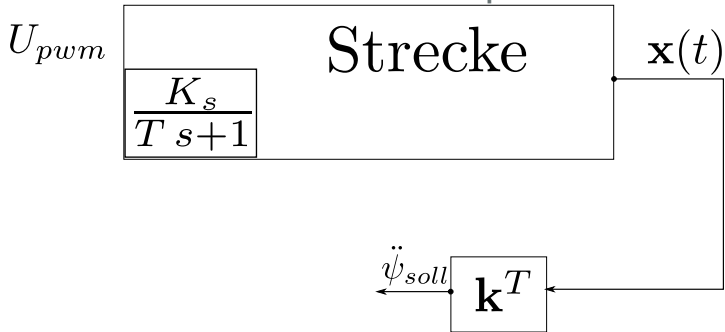
- desired pole
 $s_1 = s_2 = -5, \quad s_{3,4} = -3 \pm 3j$
- characteristic polynomial
 $CLCP = (s + 5)^2(s + 3 - 3j)(s + 3 + 3j)$
- characteristic polynomial
of the controlled system
 $CLCP = \det(sI - A - b(k_1, k_2, k_3, k_4))$
- detection of gain of the controller
via coefficient comparison



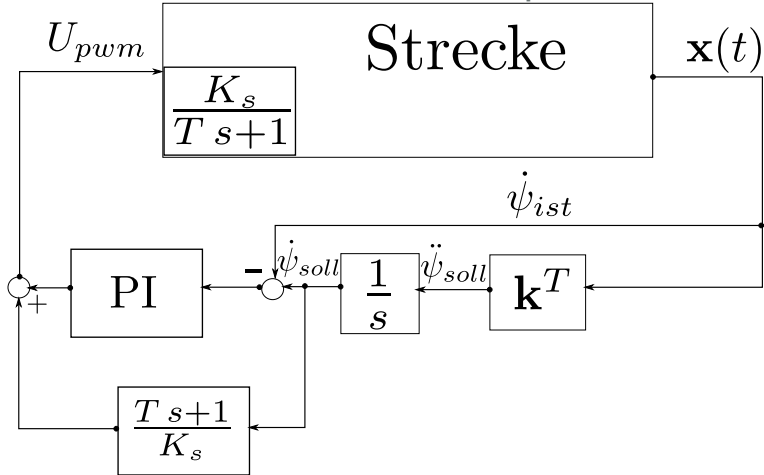
simulation



structure of the control loop



structure of the control loop





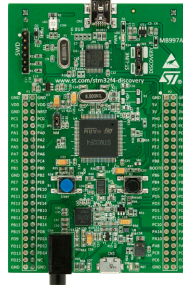
outline

- introduction
- modelling
- control unit design
- **mechanical buildup and electrical components**
- assembly
- result
- conspectus and forecast

used microcontroller STM32F407

good characteristics

- systemclock 168MHz ARM-Core
- 1MB Flash, 512kByte Ram
- floating point unit
(software, hardware)
- many pins
- opensource
 - compiler
 - linker
 - flasher



detriments

- big electricity requirement
by way of comparison to AVR-microcontroller

imagesource: st.com

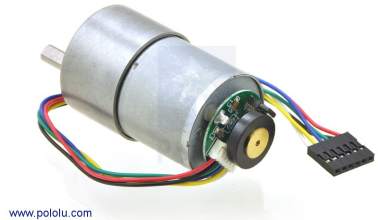
used motors

good characteristics

- operating voltage 12V
- torque 0.7767Nm
- built-in 64 encoder
- gear with 30:1 ratio
- max. 5A current at adherence of motor shaft

detriments

- different gain at the direction



imagesource: pololu.com

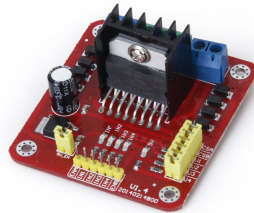
motordriver L298N H-Bridge

good characteristics

- operating voltage 15V
- max. current ca. 1A
- little thermally reliable
- for 2 motors

detriments

- none shunt-resistor

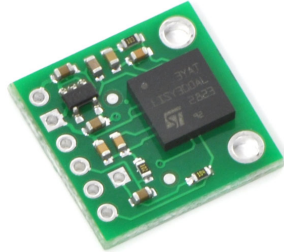


imagesource: amazon.de user: Generic

used gyroskop LISY300AL

good characteristics

- input voltage 5V
- internal 3.3V voltage controller
- very sensitive
- sensitivity $3.3\text{mV}/^\circ/\text{s}$



imagesource: pololu.com

XBEE for wireless communication

good characteristics

- operating voltage 3.3V
- serial communication via USART
- 1600m getatable according to datasl
- data throughput 1Mbps

detriments

- need 2 piece

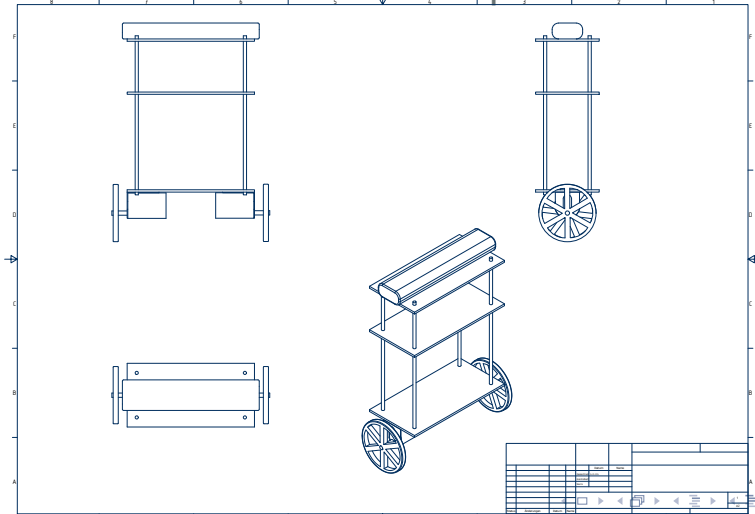


imagesource:
de.rs-online.com

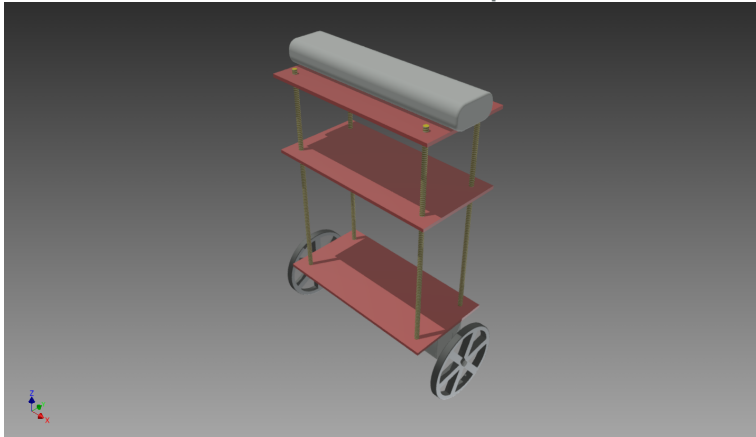




mechanical construction plan



mechanical construction plan





outline

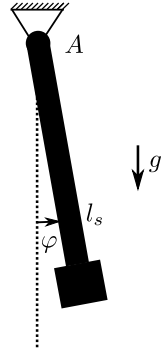
- introduction
- modelling
- control unit design
- mechanical buildup and electrical components
- **assembly**
- result
- conspectus and forecast

parameter identification: moment of inertia J_p

vibration test

- difficult to measure
- $J_p = J_A - m_1 l_s^2$

$$\ddot{\varphi} + \underbrace{\frac{l_s m_1 g}{J_A}}_{\omega_0^2} \sin(\varphi) = 0 \text{ für } \sin(\varphi) \approx \varphi$$
- $J_p = J_A - m_1 l_s^2 = \frac{l_s m_1 g T^2}{4\pi^2} - m_1 l_s^2$
- measured $T \approx 0.9\text{s}$ $J_p \approx 0.0091\text{kgm}^2$



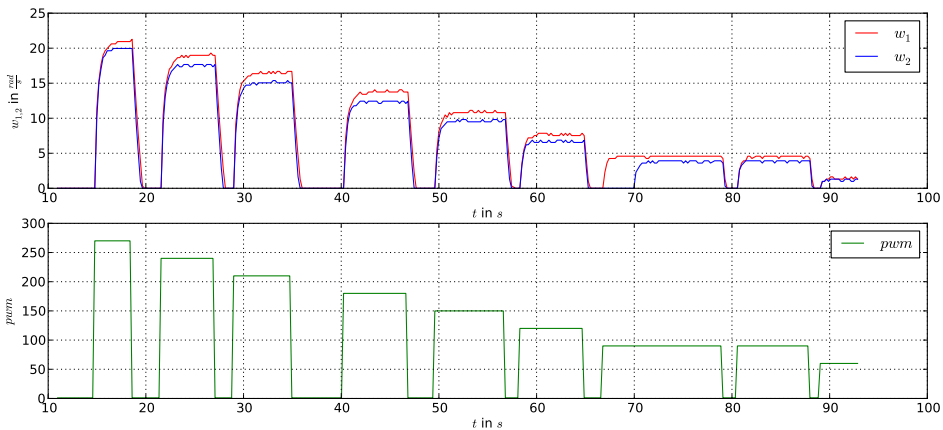
moment of inertia J_m of the wheel



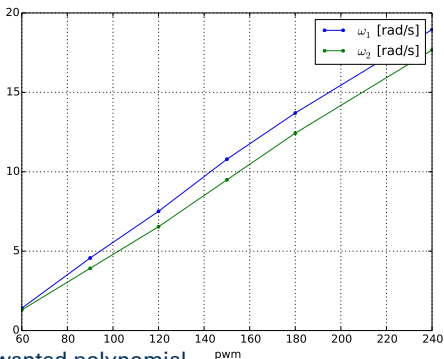
consider as fine solid cylinder

- $J_m = 2 \frac{mr^2}{2} = mr^2 = 6.48 \cdot 10^{-5} \text{kgm}^2$

imagesource: pololu.com



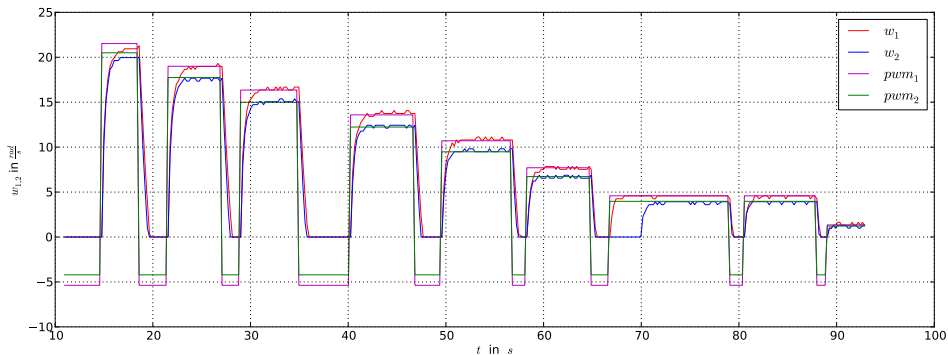
method of least squares



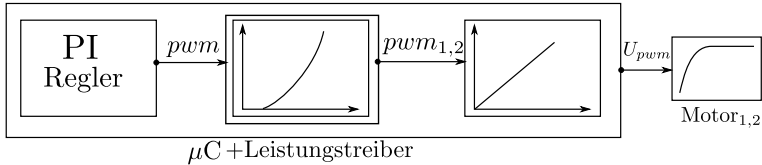
samples

wanted polynomial

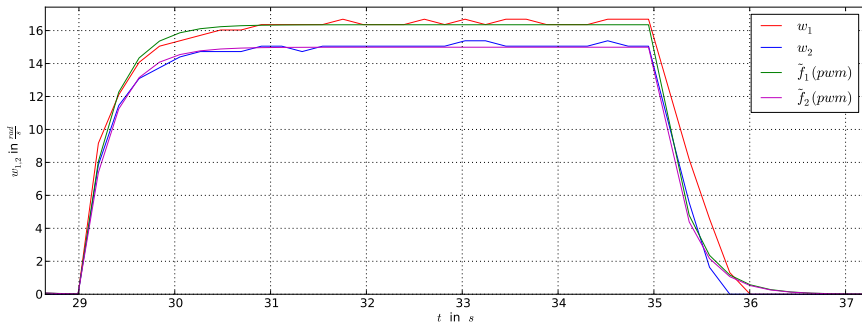
$$\begin{aligned} \text{pwm}_1 &= 9.80666 \cdot 10^{-2} \cdot \text{pwm} - 4.241 \\ \text{pwm}_2 &= 9.18355 \cdot 10^{-2} \cdot \text{pwm} - 4.295. \end{aligned}$$



concept of the PWM-conversion



approximated funktion $T = 0.3s$

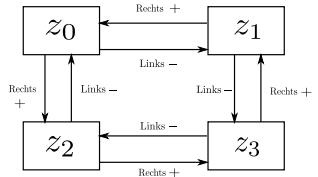
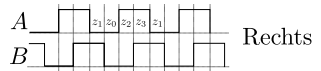


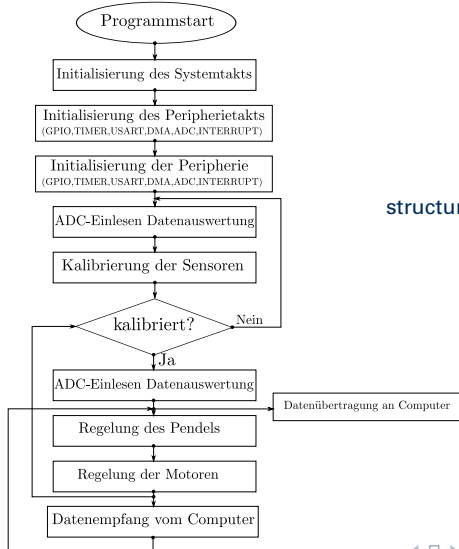
analysis of the encoder

- $\pm 90^\circ$ phase-delayed signal
- allocation of the suites
 $A \in \{0, 1\}, \quad B \in \{0, 1\}$

state	A	B	code
z_0	0	0	0
z_1	0	1	1
z_2	1	0	2
z_3	1	1	3

- right: $z_1 \rightarrow z_0 \rightarrow z_2 \rightarrow z_3 \rightarrow z_0 \dots$
left: $z_2 \rightarrow z_0 \rightarrow z_1 \rightarrow z_3 \rightarrow z_2 \dots$



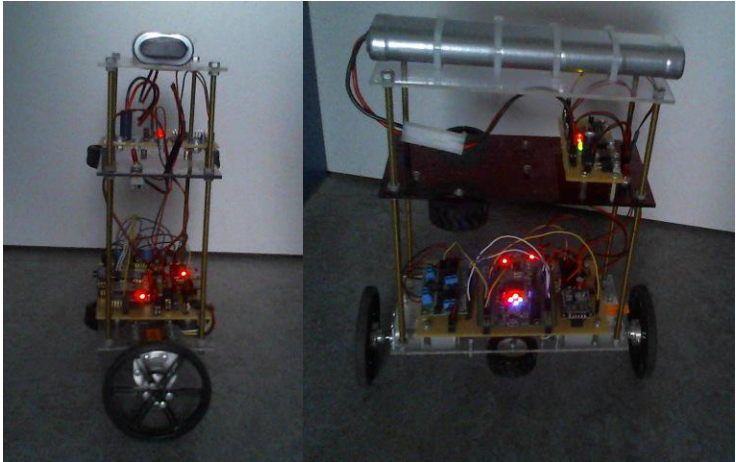


structure chart of the software

outline

- introduction
- modelling
- control unit design
- mechanical buildup and electrical components
- assembly
- **result**
- conspectus and forecast

result



outline

- introduction
- modelling
- control unit design
- mechanical buildup and electrical components
- assembly
- result
- conspectus and forecast

conspectus and forecast

conspectus:

the state controller is very sensitive towards parameter imprecision. using an I-Controller extension the stabilisation works better. the friction of link-room is neglectable. the modelling of friction is necessary.

forecast:

using LQR controller the stabilisation gets better. the optimization is required for the inside PI-control loop.



thanks for your attention!