SCHOOL of ELECTRICAL ENGINEERING & COMPUTING FACULTY of ENGINEERING & BUILT ENVIRONMENT The UNIVERSITY of NEWCASTLE

Comp3320/6370 Computer Graphics

LECTURE 12: Quaternions

Based on chapter 4 of the book "Real-Time Rendering" by Akenine-Möller et al., 4th edition 2018.

Definition

A quaternion is a 4-tuple expresses as

$$\mathbf{\hat{q}} = (\mathbf{q}_{v}, q_{w})$$

$$= iq_{x} + jq_{y} + kq_{z} + q_{w}$$

$$= \mathbf{q}_{v} + q_{w}$$

$$\mathbf{i}^{2} = j^{2} = k^{2} = -1$$

$$ij = -ji = k$$

$$jk = -kj = i$$

$$ki = -ik = j$$

Multiplication:

$$\mathbf{\hat{q}} \, \mathbf{\hat{r}} = (\mathbf{q}_v \times \mathbf{r}_v + q_w \mathbf{r}_v + r_w \mathbf{q}_v, \ q_w r_w - \mathbf{q}_v \cdot \mathbf{r}_v) \qquad \mathbf{\hat{q}} \, \mathbf{\hat{r}} \neq \mathbf{\hat{r}} \, \mathbf{\hat{q}}$$

$$s \, \mathbf{\hat{q}} = \mathbf{\hat{q}} \, s = (s \, \mathbf{q}_v, s \, q_w)$$

Addition:

$$\mathbf{\hat{q}} + \mathbf{\hat{r}} = (\mathbf{q}_v + \mathbf{r}_v, \ q_w + r_w)$$

Operations

• Conjugate: $\hat{\mathbf{q}} * = (-\mathbf{q}_v, q_w)$

• Norm:
$$n(\mathbf{\hat{q}}) = \mathbf{\hat{q}} \mathbf{\hat{q}}^* = \mathbf{\hat{q}} \mathbf{\hat{q}} = \mathbf{q}_v \cdot \mathbf{q}_v + q_w^2$$

 $= q_x^2 + q_y^2 + q_z^2 + q_w^2$
 $= ||\mathbf{\hat{q}}||^2$

• Identity:
$$\hat{\mathbf{i}} = (\mathbf{0}, 1)$$
 $\hat{\mathbf{i}} \hat{\mathbf{q}} = (\mathbf{0} \times \mathbf{q}_v + 1 \mathbf{q}_v + q_w \mathbf{0}, 1 q_w - \mathbf{0} \cdot \mathbf{q}_v)$

$$= (\mathbf{0} + \mathbf{q}_v + \mathbf{0}, q_w - \mathbf{0})$$

$$= \mathbf{q}_v + q_w$$

• Inverse:
$$\hat{\mathbf{q}}^{-1} = \frac{1}{n(\hat{\mathbf{q}})} \hat{\mathbf{q}} *$$

$$n(\mathbf{\hat{q}}) = \mathbf{\hat{q}} \, \mathbf{\hat{q}} * \qquad 1 = \frac{\mathbf{\hat{q}} \, \mathbf{\hat{q}} *}{n(\mathbf{\hat{q}})} \qquad \mathbf{\hat{q}}^{-1} = \frac{(\mathbf{\hat{q}}^{-1} \, \mathbf{\hat{q}}) \, \mathbf{\hat{q}} *}{n(\mathbf{\hat{q}})}$$

Rules

• Conjugate rules:

$$(\hat{\mathbf{q}} *) * = \hat{\mathbf{q}}$$

 $(\hat{\mathbf{q}} + \hat{\mathbf{r}}) * = \hat{\mathbf{q}} * + \hat{\mathbf{r}} *$
 $(\hat{\mathbf{q}} \hat{\mathbf{r}}) * = \hat{\mathbf{r}} * \hat{\mathbf{q}} *$

• Norm rules:

$$n(\mathbf{\hat{q}} *) = n(\mathbf{\hat{q}})$$

$$n(\mathbf{\hat{q}} \mathbf{\hat{r}}) = n(\mathbf{\hat{q}})n(\mathbf{\hat{r}})$$

Linearity:

$$\mathbf{\hat{p}}(\mathbf{\hat{q}}+\mathbf{\hat{r}}) = \mathbf{\hat{p}}\mathbf{\hat{q}}+\mathbf{\hat{p}}\mathbf{\hat{r}}$$
$$(\mathbf{\hat{q}}+\mathbf{\hat{r}})\mathbf{\hat{p}} = \mathbf{\hat{q}}\mathbf{\hat{p}}+\mathbf{\hat{r}}\mathbf{\hat{p}}$$

Associativity:

$$\hat{\mathbf{p}}(\hat{\mathbf{q}}\hat{\mathbf{r}}) = (\hat{\mathbf{p}}\hat{\mathbf{q}})\hat{\mathbf{r}}$$

Alternate Form

• Unit quaternions, $n(\hat{\mathbf{q}}) = 1$, can be written as

$$\mathbf{\hat{q}} = \sin(\phi)\mathbf{u}_q + \cos(\phi)$$

for some unit 3D vector \mathbf{u}_a

• Note that for $\hat{\mathbf{q}} = \sin(\phi)\mathbf{u}_a + \cos(\phi) = (\mathbf{q}_v, q_w)$

$$\mathbf{q}_{v} = \sin(\phi)\mathbf{u}_{q} \qquad q_{w} = \cos(\phi)$$

• Another form is $\hat{\mathbf{q}} = e^{\phi \mathbf{u}_q}$

$$\hat{\mathbf{a}} = e^{\phi \, \mathbf{u}_q}$$

which gives rise to

$$\begin{aligned} \log\left(\mathbf{\hat{q}}\right) &= \log(e^{\phi \mathbf{u}_q}) = \phi \mathbf{u}_q \\ \mathbf{\hat{q}}^{[t]} &= (\sin(\phi)\mathbf{u}_q + \cos(\phi))^t = e^{\phi t \mathbf{u}_q} = \sin(\phi t)\mathbf{u}_q + \cos(\phi t) \end{aligned}$$

Rotations

- Given a point $\mathbf{p} = (p_x \ p_y \ p_z \ p_w)^T$ create a quaternion as $\mathbf{\hat{p}} = ip_x + jp_y + kp_z + p_w$
- Given a unit quaternion $\hat{\mathbf{q}} = (\sin(\phi)\mathbf{u}_q, \cos(\phi))$

$$\hat{\mathbf{q}} \hat{\mathbf{p}} \hat{\mathbf{q}}^{-1}$$

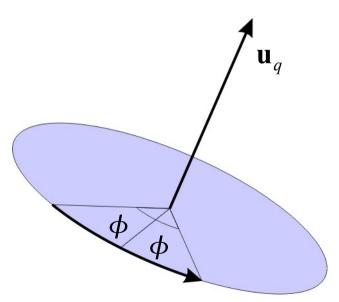
rotates the point \mathbf{p} about the axis \mathbf{u}_q by an angle of 2ϕ

• Since $n(\hat{\mathbf{q}}) = 1$

$$\hat{\mathbf{q}}\,\hat{\mathbf{p}}\,\hat{\mathbf{q}}^{-1} = \hat{\mathbf{q}}\,\hat{\mathbf{p}}\,\hat{\mathbf{q}}^*$$

 Given another unit quaternion r̂ we can first apply q̂ then r̂ by

$$\mathbf{\hat{r}}(\mathbf{\hat{q}}\,\mathbf{\hat{p}}\,\mathbf{\hat{q}}\,*)\mathbf{\hat{r}}\,* = (\mathbf{\hat{r}}\,\mathbf{\hat{q}})\mathbf{\hat{p}}(\mathbf{\hat{r}}\,\mathbf{\hat{q}})\,* = \mathbf{\hat{c}}\,\mathbf{\hat{p}}\,\mathbf{\hat{c}}\,*$$
$$\mathbf{\hat{c}} = \mathbf{\hat{r}}\,\mathbf{\hat{q}}$$



Matrix Conversion

- Matrix-Vector multiplication is more efficient than q p q *
- We can convert a unit quaternion q̂ to a matrix M^q with

$$\mathbf{M}^{q} = \begin{pmatrix} 1 - 2(q_{y}^{2} + q_{z}^{2}) & 2(q_{x}q_{y} - q_{w}q_{z}) & 2(q_{x}q_{z} + q_{w}q_{y}) & 0 \\ 2(q_{x}q_{y} + q_{w}q_{z}) & 1 - 2(q_{x}^{2} + q_{z}^{2}) & 2(q_{y}q_{z} - q_{w}q_{x}) & 0 \\ 2(q_{x}q_{z} - q_{w}q_{y}) & 2(q_{y}q_{z} + q_{w}q_{x}) & 1 - 2(q_{x}^{2} + q_{y}^{2}) & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

Converting from the matrix back to a quaternion

$$m_{21}^{q} - m_{12}^{q} = 4 q_{w} q_{x}$$

$$m_{02}^{q} - m_{20}^{q} = 4 q_{w} q_{y}$$

$$q_{w} = \frac{1}{2} \sqrt{\text{tr}(\mathbf{M}^{q})}$$

$$q_{x} = \frac{m_{21}^{q} - m_{12}^{q}}{4 q_{w}}$$

$$m_{10}^{q} - m_{01}^{q} = 4 q_{w} q_{z}$$

$$\text{tr}(\mathbf{M}^{q}) = 4 q_{w}^{2}$$

$$q_{y} = \frac{m_{02}^{q} - m_{20}^{q}}{4 q_{w}}$$

$$q_{z} = \frac{m_{10}^{q} - m_{01}^{q}}{4 q_{w}}$$

Matrix Conversion

• What if $q_w = 0$?

$$\mathbf{M}^{q} = \begin{pmatrix} 1 - 2(q_{y}^{2} + q_{z}^{2}) & 2(q_{x}q_{y} - q_{w}q_{z}) & 2(q_{x}q_{z} + q_{w}q_{y}) & 0 \\ 2(q_{x}q_{y} + q_{w}q_{z}) & 1 - 2(q_{x}^{2} + q_{z}^{2}) & 2(q_{y}q_{z} - q_{w}q_{x}) & 0 \\ 2(q_{x}q_{z} - q_{w}q_{y}) & 2(q_{y}q_{z} + q_{w}q_{x}) & 1 - 2(q_{x}^{2} + q_{y}^{2}) & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

$$= \begin{pmatrix} 2q_x^2 - 1 & 2q_xq_y & 2q_xq_z & 0 \\ 2q_xq_y & 2q_y^2 - 1 & 2q_yq_z & 0 \\ 2q_xq_z & 2q_yq_z & 2q_z^2 - 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

$$q_x = \sqrt{\frac{m_{00}^q - 1}{2}}$$
 $q_y = \sqrt{\frac{m_{11}^q - 1}{2}}$ $q_z = \sqrt{\frac{m_{22}^q - 1}{2}}$

Spherical Linear Interpolation

• Given two unit quaternions $\hat{\mathbf{q}}$ and $\hat{\mathbf{r}}$, we can interpolate between the two with

$$\hat{\mathbf{s}}(\hat{\mathbf{q}},\hat{\mathbf{r}},t) = (\hat{\mathbf{r}}\,\hat{\mathbf{q}}^{-1})^t\hat{\mathbf{q}}$$

for $t \in [0,1]$

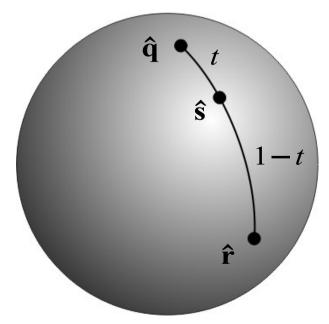
 Easier implemented in software as

$$\mathbf{\hat{s}}(\mathbf{\hat{q}},\mathbf{\hat{r}},t) = \frac{\sin(\phi(1-t))}{\sin(\phi)}\mathbf{\hat{q}} + \frac{\sin(\phi t)}{\sin(\phi)}\mathbf{\hat{r}}$$

$$\cos(\phi) = q_x r_x + q_y r_y + q_z r_z + q_w r_w$$

$$\hat{\mathbf{s}}(\hat{\mathbf{q}}, \hat{\mathbf{r}}, t) = \text{slerp}(\hat{\mathbf{q}}, \hat{\mathbf{r}}, t)$$

If the length of the arc from $\hat{\mathbf{q}}$ to $\hat{\mathbf{r}}$ is l then the length of the arc from $\hat{\mathbf{q}}$ to $\hat{\mathbf{s}}$ is lt and $\hat{\mathbf{s}}$ to $\hat{\mathbf{r}}$ is l(1-t)

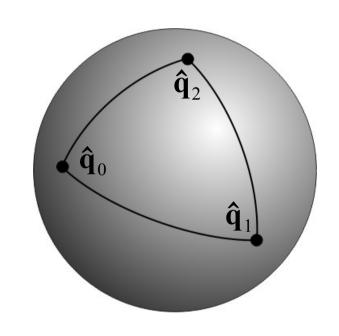


Slerping Through Many Quaternions

 Suppose we have many orientations, and we want to move from one to the other

$$\hat{\mathbf{q}}_0$$
, $\hat{\mathbf{q}}_1$, ..., $\hat{\mathbf{q}}_{n-1}$

 If we interpolate between successive quaternions we observe sudden jerks as the direction changes between interpolations.



 To make the interpolations smooth splines are used.

$$\operatorname{squad}(\mathbf{\hat{q}}_i, \mathbf{\hat{q}}_{i+1}, \mathbf{\hat{a}}_i, \mathbf{\hat{a}}_{i+1}, t) = \operatorname{slerp}(\operatorname{slerp}(\mathbf{\hat{q}}_i, \mathbf{\hat{q}}_{i+1}, t), \operatorname{slerp}(\mathbf{\hat{a}}_i, \mathbf{\hat{a}}_{i+1}, t), 2t(1-t))$$

$$\mathbf{\hat{a}}_i = \mathbf{\hat{b}}_i = \mathbf{\hat{q}}_i \exp\left(-\frac{\log(\mathbf{\hat{q}}_i^{-1}\mathbf{\hat{q}}_{i-1}) + \log(\mathbf{\hat{q}}_i^{-1}\mathbf{\hat{q}}_{i+1})}{4}\right)$$

Rotation from s to t

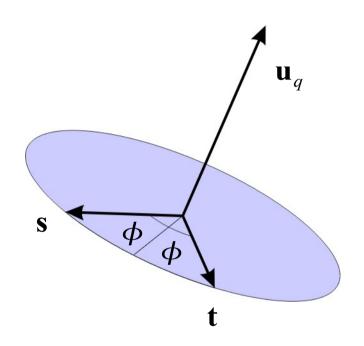
- How can we rotate from direction s to direction t?
- Normalize s and t
- Rotates through 2ϕ radians,
- about the vector u_q

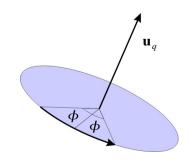
$$\mathbf{u}_q = \frac{\mathbf{s} \times \mathbf{t}}{\|\mathbf{s} \times \mathbf{t}\|}$$

$$\mathbf{s} \cdot \mathbf{t} = \cos(2\phi)$$
$$||\mathbf{s} \times \mathbf{t}|| = \sin(2\phi)$$

$$\mathbf{\hat{q}} = (\sin(\phi)\mathbf{u}_q, \cos(\phi))$$

$$\hat{\mathbf{q}} = \left(\frac{1}{\sqrt{2(1+\mathbf{s}\cdot\mathbf{t})}}(\mathbf{s}\times\mathbf{t}), \frac{\sqrt{2(1+\mathbf{s}\cdot\mathbf{t})}}{2}\right)$$





Rotation from s to t

 Often it is more convenient to represent the rotation from s to t in matrix form

$$\mathbf{R}(\mathbf{s}, \mathbf{t}) = \begin{pmatrix} a + h v_x^2 & h v_x v_y - v_z & h v_x v_z + v_y & 0 \\ h v_x v_y + v_z & a + h v_y^2 & h v_y v_z - v_x & 0 \\ h v_x v_z - v_y & h v_y v_z + v_x & a + h v_z^2 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$$

where

$$\mathbf{v} = \mathbf{s} \times \mathbf{t}$$
 $a = \mathbf{s} \cdot \mathbf{t}$ $h = \frac{1 - a}{\mathbf{v} \cdot \mathbf{v}}$