Effect of morphology on the large-amplitude flapping dynamics of an inverted flag in a uniform flow

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The stability of a cantilevered elastic sheet in a uniform flow has been studied extensively due to its importance in engineering and its prevalence in natural structures. Varying the flow speed can give rise to a range of dynamics including limit cycle behaviour and chaotic motion of the cantilevered sheet. Recently, the 'inverted flag' configuration – a cantilevered elastic sheet aligned with the flow impinging on its free edge – has been observed to produce large-amplitude flapping over a finite band of flow speeds. This flapping phenomenon has been found to be a vortex-induced vibration, and only occurs at sufficiently large Reynolds numbers. In all cases studied, the inverted flag has been formed from a cantilevered sheet of rectangular morphology, i.e. the planform of its elastic sheet is a rectangle. Here, we investigate the effect of the inverted flag's morphology on its resulting stability and dynamics. We choose a trapezoidal planform which is explored using experiment and an analytical theory for the divergence instability of an inverted flag of arbitrary morphology. Strikingly, for this planform we observe that the flow speed range over which flapping occurs scales approximately with the flow speed at which the divergence instability occurs. This provides a means by which to predict and control flapping. In a biological setting, leaves in a wind can also align themselves in an inverted flag configuration. Motivated by this natural occurrence we also study the effect of adding an artificial 'petiole' (a thin elastic stalk that connects the sheet to the clamp) on the inverted flag's dynamics. We find that the petiole serves to partially decouple fluid forces from elastic forces, for which an analytical theory is also derived, in addition to increasing the freedom by which the flapping dynamics can be tuned. These results highlight the intricacies of the flapping instability and account for some of the varied dynamics of leaves in nature.

Key words: flow–structure interactions

1. Introduction

As one of the simplest and most readily found examples of aeroelastic phenomena in nature, the dynamics of leaves excited by an impinging wind has contributed to the understanding of both plant biomechanics as well as fundamental aeroelasticity; see de Langre (2008) for a review. The seminal work of Vogel (1989) introduced the now well-studied phenomena of drag reduction via reconfiguration of flexible bodies. However, most experimental and theoretical studies have focused on leaves oriented in the conventional configuration of a flag fluttering in the wind, i.e. the flow impinges on the restrained end of the leaf with its free edge downstream. Even among leaves in a single tree, their location on the tree and thus orientation with respect to the impinging wind affect their observed dynamics. Thus far, no report has systematically studied the motion of leaves in an inverted orientation, where the leading edge of the leaf is free and the trailing edge is fixed. While a naïve assessment would predict that the leaf would simply bend around the fixed end to return back to the conventional orientation, recent experimental work has shown that the dynamics is much more complex. Figures 1(a) and 1(b) show a single measurement of such an inverted leaf (taken from Sader et al. 2016a) where large-amplitude flapping (limit cycle behaviour) is evident. Detailed studies of the related and canonical 'inverted flag' configuration a cantilevered elastic sheet oriented such that the flow impinges on its free edge with its clamp downstream - provide an explanation for this leaf behaviour and show that orientation of the flag can have a dramatic effect (Kim et al. 2013; Cossé et al. 2014; Ryu et al. 2015; Tang, Liu & Lu 2015; Sader et al. 2016a).

Kim *et al.* (2013) reported the first detailed measurements on the stability and dynamics of the inverted flag, in both air and water. The behaviour of the flag was characterised primarily by two dimensionless groups (Kornecki, Dowell & O'Brien 1976; Shelley & Zhang 2011), a scaled flow speed,

$$\kappa \equiv \frac{\rho U^2 L^3}{D},\tag{1.1}$$

and a mass ratio,

$$\mu \equiv \frac{\rho L}{\rho_s h},\tag{1.2}$$

where ρ is the fluid density, U the impinging flow speed, L is the sheet length (in the direction of flow), the flexural rigidity of the elastic sheet

$$D \equiv \frac{Eh^3}{12(1-\nu^2)},\tag{1.3}$$

where E and ν are its Young's modulus and Poisson's ratio, respectively, h is its thickness and ρ_s is the sheet's density. Note that the dimensionless flow speed, κ , represents a ratio of hydrodynamic forces to elastic restoring forces and can also be interpreted as a dimensionless reciprocal bending stiffness. The mass ratio μ , specifies the relative importance of fluid-to-solid inertia.

A small range of aspect ratios, H/L=1-1.3, where H is the sheet height (perpendicular to the flow) were considered by Kim *et al.* (2013). The inverted flag's dynamics was found to depend predominantly on the dimensionless flow speed, κ , with a variation in the mass ratio, μ , by a factor of $1000 - \text{from } \mu \approx 1$ (air) to 1000 (water) – producing only a moderate change in behaviour. The Reynolds number

528 B. Fan, C. Huertas-Cerdeira, J. Cossé, J. E. Sader and M. Gharib

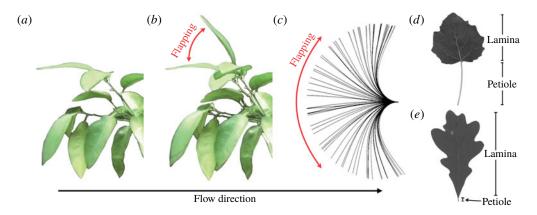


FIGURE 1. (Colour online) (a) Leaves on a branch (image at an initial time), and (b) 0.67 s later with initial position overlaid to highlight flapping. (c) Stroboscopic image of an inverted flag exhibiting large-amplitude flapping instability from an edge-on view. Taken from present study: Flag E (standard rectangular inverted flag) at flow speed $U = 5.43 \text{ m s}^{-1}$; see § 3. (d) Typical white poplar leaf with lobed/heart-shaped base and long petiole, and (e) white oak leaf with acute base and short petiole (from Leafsnap dataset (Kumar et al. 2012), not to scale).

studied was in the range $Re \equiv UL/\nu \approx 10^4 - 10^5$, where ν is the fluid's kinematic viscosity.

It was observed that by increasing the flow speed from zero, the inverted flag's initial stationary state abruptly gave way to large-amplitude periodic flapping (see figure 1c) which eventually ceased (again abruptly) to a deflected stationary state. The value of $\kappa = \kappa_{lower}$ at which large-amplitude flapping began (at low flow speed) was found to be identical in air and water. This was subsequently investigated by Gurugubelli & Jaiman (2015) using a computational method, which provided support for the proposal of Kim et al. (2013) that large-amplitude flapping is initiated by a divergence instability. This was proved mathematically by Sader et al. (2016a) who conducted a linear stability analysis. Sader et al. (2016a) also reported results of an extensive investigation of the underlying physical mechanisms of the inverted flag's dynamics and concluded that large-amplitude flapping is a vortex-induced vibration (VIV). This finding was consistent with previous numerical simulations Ryu et al. (2015) who observed that large-amplitude flapping does not occur for $Re \lesssim 50$; which is strikingly similar to the numerical value for cessation of VIV by an elastic circular cylinder. This aspect has been investigated further by Goza, Colonius & Sader (2018), again using direct numerical simulation, who found that flapping of the inverted flag persists for values of Re below Re \approx 50, but only when the mass ratio is small, i.e. $\mu \ll 1$. This is strikingly similar to the behaviour of an elastic circular cylinder, which is also discussed by Goza et al. (2018).

The dimensionless flow speed, $\kappa = \kappa_{upper}$, at which large-amplitude flapping ceases (at high flow speed) is found to depend on the mass ratio μ , in contrast to the (lower) flow speed at which large-amplitude flapping is initiated. Sader *et al.* (2016a) noted that this observation is consistent with the existence of VIV where changing the mass ratio produces a change in effective damping, and thus the flow speed at which desynchronisation occurs between vortex shedding and vibrational motion of the sheet. Increasing μ reduces the dimensionless flow speed, κ , where large-amplitude flapping ceases, albeit weakly.

A substantial body of work on the inverted flag has now been reported, involving experimental, computational and analytical methods. These include exploration of the inverted flag's energy harvesting potential (Shoele & Mittal 2016; Orrego et al. 2017), which seems particularly attractive given the large oscillation amplitudes that the inverted flag exhibits relative to a conventional flag. The interaction of two inverted flags has been reported and displays an interesting combination of synchronous and asynchronous dynamics depending on the flow speed and relative position, separation and length of the flags (Huang, Wei & Lu 2018; Huertas-Cerdeira, Fan & Gharib 2018; Ryu, Park & Sung 2018). The aspect ratio, H/L, of the inverted flag's elastic sheet has been found to play a critical role and strongly affect the (lower) flow speed at which large-amplitude flapping is initiated (Cossé et al. 2014; Sader et al. 2016a); the (higher) flow speed at which large-amplitude flapping ceases is relatively insensitive to aspect ratio. Indeed, reducing the aspect ratio to values less than unity results in a change in dynamics with elimination of the divergence instability (that initiates large-amplitude flapping) and replacement with a saddle node bifurcation (Sader, Huertas-Cerdeira & Gharib 2016b; Tavallaeinejad, Païdoussis & Legrand 2018) – where multiple linearly stable equilibrium states exist. Investigation of an inverted flag with a splitter plate attached to its trailing edge has also been reported (Gurugubelli & Jaiman 2017). In addition to direct numerical simulations at moderate Reynolds numbers, $Re \sim O(100)$, discussed above, computations using large eddy simulations (LES) at experimentally relevant Reynolds numbers, $Re \sim O(10^5)$ – which produce turbulent flows - have been performed (Gilmanov, Le & Sotiropoulos 2015).

One key aspect that is yet to be studied is the effect of the inverted flag's morphology, i.e. the plan view shape of the flag's elastic sheet, on its resulting stability and dynamics. Such knowledge is particularly relevant to understanding the behaviour of leaves that naturally exhibit a wide array of shapes; see figures 1(d) and 1(e). All reports to date have used the same flag morphology of Kim *et al.* (2013): an elastic sheet of rectangular planform. Here, we explore the effect of changing this shape by varying the geometric parameters of a trapezoidal morphology for the inverted flag; see figure 2(a). This increases the phase space of the resulting dynamics by adding a single shape parameter: the ratio of the free edge height, Δ , to that at its clamp, H; for a rectangular sheet, we have $\Delta/H = 1$. This trapezoidal inverted flag is studied experimentally in the wind tunnel used in Kim *et al.* (2013) and by extending the divergence instability theory reported by Sader *et al.* (2016a) to an arbitrary planform that enables analytical solution. For the latter, we introduce a 'generalised aspect ratio' (Jones 1990; Anderson 2006):

$$AR \equiv \frac{b^2}{S},\tag{1.4}$$

where S is the surface area of the sheet's planform and b is the maximum height of the sheet (perpendicular to the flow).

Most leaves exhibit a short and narrow section – termed a 'petiole' – that is attached to the branch of a tree (or to the tree itself); see figure 1(d). The effect of this morphological feature on the dynamics of leaves with an impinging flow is also yet to be studied. To glean insight into its influence on the dynamics of leaves we experimentally investigate inverted flags that contain such a petiole. This again increases the phase space of the resulting system. Therefore, in this initial study we focus on inverted flags (containing a petiole) with elastic sheets of rectangular planform only.

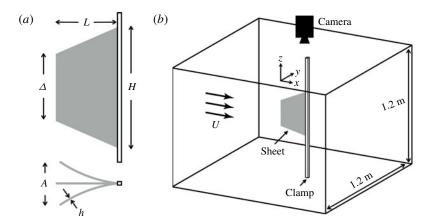


FIGURE 2. (a) Schematic of the trapezoidal inverted flag (without a petiole) studied in § 3 from a side view perspective. (b) Experimental set-up in wind tunnel showing Cartesian coordinate system whose origin is located at the middle of the sheet's leading free edge. (c) Top/camera view showing the straight sheet during flapping.

We begin by summarising details of the experimental apparatus and the measurement procedure in § 2. This is followed in § 3 by measurement data on inverted flags of trapezoidal morphology that do not contain a petiole. A detailed exploration of the underlying physical mechanisms and a discussion of the measurement data are provided. A general theory for the divergence instability of an inverted flag of arbitrary planform is presented in § 4 which includes comparison to the measurements reported in § 3. Finally, the effect of introducing a petiole to an inverted flag of rectangular planform is studied in § 5 together with commensurate theory.

2. Experimental details

Inverted flags of trapezoidal (and rectangular) planforms are constructed from polycarbonate sheets with a Young's modulus E = 2.41 GPa, Poisson's ratio v = 0.38 and density $\rho_s = 1.2 \times 10^3$ kg m⁻³. These flags are studied in an open loop wind tunnel with a square cross section of side length 1.2 m, which is capable of producing a free-stream flow speed, U, between 1.8 and 8.5 m s⁻¹; see figure 2(b) for a schematic of the wind tunnel. Further details on the wind tunnel are reported in Kim et al. (2013), Sader et al. (2016a), Huertas-Cerdeira et al. (2018).

The inverted flag is oriented so that the impinging free stream is parallel to the flag's undeformed polycarbonate sheet. Data are collected by increasing the free-stream speed from rest, and monitoring the inverted flag's dynamics at discrete speeds that are held constant for the measurement. A high-speed camera mounted above the test section captures the sheet motion at 100-600 frames per second for varying flow speeds. Kinematic data of the top edge of the sheet are then extracted. See figure 2(c) for a schematic illustrating the camera viewpoint which shows the sheet thickness, h, and (peak-to-peak) oscillation amplitude, A, of a flapping sheet. We minimise the effects of gravity by clamping the sheet vertically. All deformations reported in this study, unless otherwise noted, are observed to be primarily two-dimensional, i.e. occurring in the xy-plane; see figure 2(b) for the coordinate system. The natural resonant frequency of an inverted flag, f_{res} , is measured by striking the sheet in still air and allowing it to return to its rest position. This provides a natural frequency scale based on the flag's linear response.

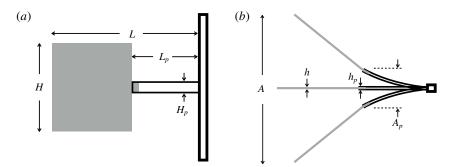


FIGURE 3. Schematic of an inverted flag of rectangular morphology with a petiole. (a) Side view, and (b) top/camera view with straight rest position and flapping dynamics overlaid. The shaded area corresponds to the elastic polycarbonate sheet with a small tab-like overlap region used to join it to the petiole.

2.1. No petioles

We first consider inverted flags without petioles, i.e. the trapezoidal sheet is attached directly to the clamp, formed from a single sheet of polycarbonate. The free stream impinges on the leading free edge of the sheet while the trailing edge of the sheet is clamped between two thin but rigid steel strips. The heights of all trapezoidal sheets are fixed at H=19.05 cm and their lengths are L=15.24 cm (giving H/L=1.25), which results in a Reynolds number of $Re \sim O(10^4)$. We tested polycarbonate sheets with thicknesses of h=0.25 and 0.38 mm, corresponding to mass ratios of $\mu=0.61$ and 0.40, respectively. The trapezoidal shape of the sheets is adjusted in the range $0 \le \Delta/H \le 1.75$ for sheets with $\mu=0.40$, and $0 \le \Delta/H \le 1$ for sheets with $\mu=0.61$. Note that $\Delta/H=0$ corresponds to a triangular morphology.

2.2. Petioles

The effect of including a petiole is studied for sheets of rectangular planform only. The elastic sheet and petiole are both fabricated from polycarbonate sheets. Each petiole is constructed by sandwiching two slender polycarbonate strips and bonding them with cyanoacrylate adhesive. The inverted flag is constructed from a single large rectangular sheet that is joined to the petiole using a small rectangular tab-like overlap region on the sheet; see figure 3(a). The resulting inverted flag is mounted in the wind tunnel by rigidly clamping the downstream end of the petiole to the steel strips (which are described in § 2.1) with the flow impinging on the sheet's free edge. The petiole length L_p , height H_p and thickness h_p are illustrated in figures 3(a) and 3(b). The petiole length, L_p , is varied relative to sheet length, $(L - L_p)$, while fixing total length of the inverted flag, L, height H and the sheet thickness h. The petiole's flexural rigidity is adjusted by changing its thickness, h_p . Parameter values used in this study are given in table 1. By approximating the petiole as a single uniform strip of polycarbonate, the rigidity ratio between the sheet and the petiole is $(h/h_p)^3$; see (1.3).

Also shown in table 1 are measured values for the inverted flags' maximum flapping frequencies (driven by the impinging flow) and their natural resonant frequencies (in the absence of flow). The flapping frequency is determined via the average number of cycles per unit time, because the dynamics is highly periodic, and f_{max} is the maximum frequency of the flapping dynamics as the flow speed is varied (excluding chaotic flapping).

Flag	L	H	h	L_p	H_p	h_p	Petiole	f_{max}	f_{res}
A	17.8	17.8	0.051	6.4	1.3	0.102	Short, high rigidity	2.1	3.3
В	17.8	17.8	0.051	6.4	1.3	0.076	Short, low rigidity	1.5	2.6
C	17.8	17.8	0.051	11.4	1.3	0.102	Long, high rigidity	2.3	4.0
D	17.8	17.8	0.051	11.4	1.3	0.076	Long, low rigidity	1.5	2.7
E	17.8	17.8	0.051	_	_	_	No petiole	2.4	3.5

TABLE 1. Dimensions, petiole description and measured frequencies of rectangular inverted flags with a petiole; Flag E is a rectangular inverted flag with no petiole that is used for reference. These flags are studied in § 5. Here, f_{max} is the measured flapping frequency with an impinging flow and f_{res} is its measured natural frequency in the absence of flow. All length/height/thickness values are reported in cm and frequencies are in Hz.

3. Trapezoidal inverted flag without a petiole

In this section, we report measurements of the stability and dynamics of inverted flags with trapezoidal planforms that do not have petioles. Across all sheet morphologies, we observe the presence of three major behavioural regimes in order of increasing flow speed: a stable zero deflection equilibrium referred to as the 'straight mode', a large-amplitude 'flapping mode' and a stable deflected equilibrium referred to as the 'deflected mode'; see stroboscopic images in figure 4(a) measured over a 2.5 s interval. These three regimes are separated by abrupt transitions at two critical non-dimensional flow speeds: (i) κ_{lower} , where the sheet's zero-deflection equilibrium undergoes a divergence instability, giving way to the flapping mode, and (ii) κ_{upper} , where flapping ceases. The triangular sheet $(\Delta/H = 0)$ requires the highest non-dimensional flow speed to begin and end flapping. As the shape parameter Δ/H is increased, these critical flow speeds decrease, along with the flow speed range over which flapping occurs. Comparing sheets of different μ , we observe slight differences in κ_{lower} (see inset of figure 4a). These differences are small relative to the dependence on Δ/H , and may be due to variations in the intrinsic curvature in the sheets. In the inset of figure 4(a), data for the onset or termination of flapping are included for the range of flow speeds achievable in our experimental apparatus; measurement of both κ_{lower} and κ_{upper} was not possible for some morphologies.

There is very small-scale flutter ($\lesssim 10\%$ of the maximum flapping amplitude) in the sheets at flow speeds less than that required to induce large-amplitude flapping; see stroboscopic images in figure 4(a). This small-scale flutter may arise due to a combination of factors that include the small intrinsic curvature of the sheets, small non-zero angle-of-attack and turbulent fluctuations in the wind tunnel. Indeed, Tavallaeinejad *et al.* (2018) finds that small angles-of-attack/curvature induce this type of small-scale flutter (see figure 10 of that previous study). As a result of this small-scale flutter, there is uncertainty in quantifying κ_{lower} . However, as can be seen from the stroboscopic images in figure 4(a), the onset of flapping remains qualitatively distinct – it occurs within a very narrow band of flow speeds. Therefore, κ_{lower} is measured at the flow speed where the flapping amplitude is half of its maximum value across all flow speeds, consistent with prior work (Kim *et al.* 2013; Sader *et al.* 2016a). As a measure of the uncertainty in κ_{lower} , the error bars in the inset of figure 4(a) indicate the flow speeds at which the flapping amplitude is 25% and 75% of its maximum value.

For the trapezoidal morphologies studied, we observe that the ratio $\kappa_{upper}/\kappa_{lower}$ has an average of 3.8 with a standard deviation of 0.7. Strikingly, this ratio appears

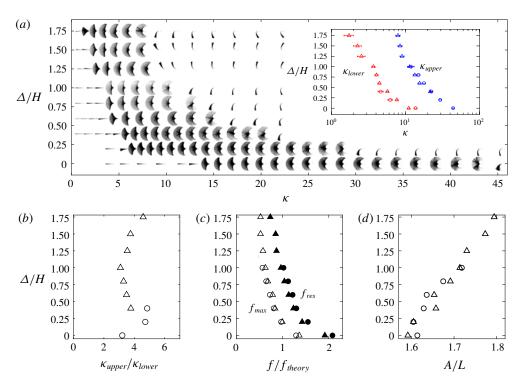


FIGURE 4. (Colour online) Trapezoidal inverted flag behaviour as a function of shape parameter, Δ/H , and the dimensionless flow speed, κ . (a) Stroboscopic images of the sheet motion are taken over 2.5 s and subsequently overlaid. Sheet dynamics with mass ratio $\mu=0.61$ (\bigcirc) is shown for $\Delta/H=0$ and 0.2, while the dynamics at $\mu=0.40$ (\triangle) is given for $\Delta/H>0.2$; this enables all three behavioural regimes to be reported here. Inset gives the dimensionless flow speeds, κ , for the onset (left symbols) and cessation (right symbols) of large-amplitude flapping. (b) Plot of $\kappa_{upper}/\kappa_{lower}$, taken from data in the inset of (a), shows this ratio is approximately constant. (c) Flapping frequencies, f_{max} , and natural resonant frequencies, f_{res} ($\mu=0.61$ (\bigcirc) and $\mu=0.40$ (\triangle)). (d) Maximum amplitude of trapezoidal inverted flags in flow speed range where flapping occurs. Inverted flags with two mass ratios, $\mu=0.40$ (\triangle) and $\mu=0.61$ (\bigcirc), are shown.

to have no clear dependence on Δ/H ; see figure 4(a) (inset) and figure 4(b). This observation that $\kappa_{upper}/\kappa_{lower}\approx 4$ suggests that the flow speed range over which flapping occurs can be estimated solely from knowledge of κ_{lower} for trapezoidal flags – which can in turn be theoretically calculated; see § 4. The validity of this empirical observation for all shapes, however, remains unclear.

The flapping mode can be subdivided into (i) a (limit cycle) steady flapping mode, and (ii) a chaotic flapping mode that arises only at flow speeds close to and just below the flapping–deflected mode transition, κ_{upper} . A steady flapping mode is evident as a smooth gradient in the stroboscopic images of figure 4(a), while aperiodic (chaotic) flapping displays prominent sharp edges. We observe that the range of κ for chaotic flapping is widest for the triangular lamina ($\Delta/H=0$) and becomes narrower as the shape parameter Δ/H increases. For the rectangular sheets ($\Delta/H=1$), our observations are consistent with previous experimental observations made by Kim *et al.* (2013) and Sader *et al.* (2016a), as well as with the theoretical

prediction of $\kappa_{lower} \approx 4.14$ using (2.15) of Sader *et al.* (2016*b*); which is valid for rectangular planforms of arbitrary aspect ratio.

To investigate the effect of the elastic sheet's morphology on the measured (i) flapping frequency, f_{max} , and (ii) natural resonant frequency, f_{res} , and to compare this effect across sheets with different mass ratios μ , we scale all frequencies by the theoretical natural resonant frequency for a cantilevered rectangular sheet in an inviscid fluid (Shen, Chakraborty & Sader 2016),

$$f_{theory} = \frac{C_1^2}{2\pi L^2} \sqrt{\frac{D}{\rho_s h(1+a\mu)}},$$
 (3.1)

where $C_1 = 1.875$ (for the fundamental flexural mode), $(1 + a\mu)$ is the added apparent mass term that accounts for the fluid's inertial load and a is an O(1) dimensionless coefficient that depends on the sheet's aspect ratio, H/L, the geometry of its clamp and, to a lesser degree, the mass ratio, μ . Modelling the rigid steel strips that hold the flag's elastic sheet as a 'line clamp', Shen $et\ al.\ (2016)$ gives a=0.33 for all mass ratios studied. Inserting numerical values for the mass ratios studied, this added apparent mass term in (3.1) shows that the air surrounding the sheet reduces its natural resonant frequency by approximately 10%. Equation (3.1) provides a natural frequency scale for the inverted flag. From figure 4(c), we find that (3.1) well approximates the natural resonant frequency of the rectangular $(\Delta/H=1)$ sheet, i.e. $f_{res}/f_{theory} \approx 1$.

Figure 4(c) shows that (3.1) captures the dominant behaviour of the natural frequencies, f_{res} , of all trapezoidal sheets studied (filled shapes), as their shape is modified. This is not surprising because changing the trapezoidal shape (via Δ/H) alters the mass and stiffness of the sheet in a commensurate manner. As the shape parameter, Δ/H , of the trapezoidal sheet increases, both the flapping and natural resonant frequencies decrease monotonically with the flapping frequency always being less than the natural frequency. The latter observation is consistent with previous measurements on an inverted flag of rectangular morphology (Sader *et al.* 2016*a*); the flapping mode is associated with strong vortex shedding which can significantly modify the fluid load. Strikingly, all measured flapping frequencies scaled by (3.1) are O(1) quantities, as expected for a vortex-induced vibration under moderate fluid loading, $\mu < 1$; see § 2 (Williamson & Govardhan 2004; Sader *et al.* 2016*a*).

All trapezoidal inverted flags begin flapping at a Strouhal number, $St \equiv fA/U$ (following Kim *et al.* (2013)), between 0.15 and 0.22, that decreases with increasing flow speed to a value in the vicinity of 0.05–0.08 before the deflected mode emerges. This is consistent with general observations of vortex-induced vibrations; as noted by Sader *et al.* (2016a) for inverted flags of rectangular planform. Strikingly, the maximum non-dimensional flapping amplitude, A/L, in the flapping regime increased approximately linearly with Δ/H ; see figure 4(d). This may be expected due to slight differences in the vibrating mode shape as a function of flag morphology, which alters the maximum vertical extension (amplitude) during flapping; the difference is approximately 10%. It is possible that differences in the flow dynamics affect the flapping amplitude which represents an interesting avenue for future work.

These measurements have implications for the dynamics of leaves in nature. Based solely on the sheet's morphology and with all else being held constant, our measurements suggest that white poplar leaves (see figure 1d) in an inverted configuration – that are geometrically analogous to $\Delta/H \ll 1$ – would (i) undergo flapping at relatively high flow speeds, and (ii) for a wide range of flow speeds, with

Effect of morphology on an inverted flag

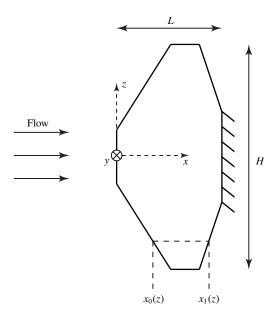


FIGURE 5. Schematic of an inverted flag of arbitrary morphology. Coordinate system, its origin (centre of leading edge) and the geometric parameters used in general theoretical model for a divergence instability in §4 are shown.

(iii) higher flapping frequency and lower amplitude. White oak leaves (see figure 1e) – analogous to $\Delta/H \gg 1$ – on the other hand, would either undergo large-amplitude flapping at relatively low flow speeds and only for a narrow range of speeds, or simply transition directly from the straight mode to the deflected mode and not exhibit any flapping dynamics (Sader *et al.* 2016*b*).

4. Theory for inverted flags of arbitrary morphology

In this section, we present a general theoretical model for the divergence instability, which initiates large-amplitude flapping, of an inverted flag of arbitrary morphology; see figure 5. It is assumed that the flag's elastic sheet is of constant thickness, has uniform material properties and only deforms in the xy-plane; see figure 5. This theory generalises the theoretical framework of Sader $et\ al.\ (2016a)$, which was developed for inverted flags of rectangular planform. It is applied to inverted flags of trapezoidal planform which is then compared to the measurements reported in § 3. Our primary goal here is to provide a simple theoretical framework that facilitates calculation of κ_{lower} and prediction of experimental behaviour. We emphasise that the presented approximate analytical approach – based on a variational analysis – provides a strict upper bound for κ_{lower} (Leissa 1969). We find that this simple approach captures the dominant experimental behaviour with no fitting parameters.

The general theory for arbitrary planforms is first formulated in the asymptotic limit, $AR \to \infty$, which is then corrected for finite generalised aspect ratio, AR. This is achieved using a strip model that utilises the (quasi-steady) analytical solution of Kornecki *et al.* (1976) for a two-dimensional (rectangular) cantilevered plate and Prandtl's lifting line theory (Jones 1990; Anderson 2006), respectively. As such, the derived theory implicitly ignores the nonlinear effects of vortex lift – driven

by vortex shedding along the side edges of the sheet – which manifest themselves with increasing importance as the generalised aspect ratio AR is reduced; see Sader $et\ al.\ (2016b)$. Such nonlinear lift mechanisms are expected to enhance the total lift, and thus reduce the flow speed at which the divergence instability occurs (Sader $et\ al.\ 2016b$). Therefore, neglect of these nonlinear lift mechanisms in the present theory leads to an upper bound on the dimensionless flow speed, κ_{lower} . Finally, we note that this theory ignores all three-dimensional effects, i.e. deformation in the z-direction; this also ensures an upper bound in κ_{lower} is achieved (Leissa 1969). In the range of trapezoidal geometries studied in § 3, all deformations are observed to be two-dimensional. However, the assumed two-dimensional (2-D) deflection may not always hold, e.g. for cases where only a small section of the sheet is attached to the clamp (large Δ/H). Thus, the underlying assumptions of the presented theory must be assessed relative to measurement.

4.1. Strip model for arbitrary planform $(AR \rightarrow \infty)$

In the limit, $AR \to \infty$, as defined in (1.4), the pressure distribution along the cantilevered sheet can be rigorously calculated by dividing the sheet into infinitesimal rectangular strips in the direction of flow (at each spanwise location). This limit corresponds to a sheet that is infinitely wide relative to its length. The dimensionless pressure jump distribution (scaled by $\rho U^2 A_{disp}/L$, where A_{disp} is the magnitude of the sheet's free end static displacement) across each strip is then given by that of an infinitely wide (2-D) rectangular cantilevered sheet (Kornecki *et al.* 1976),

$$p(x, z) = \frac{g(x, z) - g(x_1(z), z)}{\sin \theta(x, z)},$$
(4.1)

where

$$g(x,z) = -\frac{2}{\pi} \int_0^{\pi} \frac{\sin^2 \phi}{\cos \phi - \cos \theta(x,z)} \frac{\partial w(\zeta,z)}{\partial \zeta} d\phi, \qquad (4.2a)$$

$$\zeta = x_0(z) + \frac{x_1(z) - x_0(z)}{2} (1 + \cos \phi), \tag{4.2b}$$

$$\theta(x,z) = \cos^{-1}\left[\frac{2x - x_1(z) - x_0(z)}{x_1(z) - x_0(z)}\right]. \tag{4.2c}$$

Here, w(x, z) is the sheet's dimensionless displacement (scaled by A_{disp}) normal to its plane, x and y are the dimensionless Cartesian coordinates (scaled by the sheet's length, L) parallel and normal to the flow direction, respectively, $x_0(z)$ and $x_1(z)$ are the x-coordinates of the sheet's leading and trailing edge at z, respectively; see figure 5. Furthermore, θ , $\phi \in [0, \pi]$ and at any given z, there is a one-to-one relationship between x and θ , and also ζ and ϕ .

To determine the dimensionless flow speed, κ_{lower} , at which divergence occurs we make use of the small-deflection theory for thin plates (Landau & Lifshitz 1970),

$$\nabla^4 w(x, z) = \kappa \, p(x, z). \tag{4.3}$$

Equation (4.3) is then multiplied by w(x, z) and integrated over the sheet's surface, S, while imposing the sheet's boundary conditions and noting that w(x, z) varies slowly

537

with respect to z (when $H/L \gg 1$), to give

$$\kappa_{lower}^{\infty} = \frac{\int_{S} \left(\frac{\partial^{2} w}{\partial x^{2}}\right)^{2} dS}{\int_{S} p w dS},$$
(4.4)

where the superscript, ∞ , denotes use of the current strip model and is valid for $AR \rightarrow \infty$. Since the pressure, p, is linearly dependent on the displacement, w, it then follows that (4.4) is independent of the displacement magnitude, A_{disp} , as required.

In Sader *et al.* (2016a), the deflection function w is determined using a Rayleigh-Ritz approach where w is expressed as a general power series in x with unknown coefficients; these unknown coefficients are then determined from the stationary point of (4.4). To simplify implementation here, we instead make use of the known deflection function for a rectangular cantilevered sheet ($\Delta/H=1$) under a static point load at its free end,

$$w(x, z) = \frac{1}{2}(2+x)(1-x)^2,$$
(4.5)

which enforces the free end condition at x=0 and the clamp at x=1. This facilitates analytical solution of κ_{lower} for sheets of arbitrary planform by substituting (4.1) and (4.5) into (4.4). Because (4.5) is an approximation to the true deflection function of a sheet of arbitrary morphology, this approach guarantees that the resulting value for κ_{lower}^{∞} is strictly an upper bound (Leissa 1969). This is consistent with the neglect of vortex lift, which leads to an overestimate of κ_{lower} (see above), ensuring the presented theory produces a strict upper bound for κ_{lower} . It is important to note that instead of (4.5), a general power series expansion could be used, as in Sader *et al.* (2016a), at the expense of increased complexity. Clearly, care must be exercised when the expected deflection mode shape deviates significantly from (4.5), which will be discussed in § 4.2.1.

Substituting (4.5) into (4.1) gives the required pressure distribution,

$$p(x,z) = -\frac{3(x_1 - x_0)\sin\theta}{8(1 + \cos\theta)} \left[\frac{8(x_1^2 - 1)}{x_1 - x_0} + 2(x_0 + 3x_1)\cos\theta + (x_1 - x_0)\cos 2\theta \right], (4.6)$$

where x_0 , x_1 and θ are defined in (4.2c) and figure 5, and depend on x and z. Equations (4.4)–(4.6) enable the divergence instability of an inverted cantilever sheet of arbitrary planform to be calculated directly and trivially.

4.2. Trapezoidal planform

The general strip theory in the previous section is now used to calculate κ_{lower}^{∞} for an inverted flag of trapezoidal morphology; this is corrected for finite AR in § 4.3. Making use of (4.5) and (4.6), the integration of (4.4) over the trapezoidal planform then yields

$$\kappa_{lower}^{\infty} = \begin{cases}
\frac{768}{\pi} \frac{3 + \frac{\Delta}{H}}{121 + 404 \frac{\Delta}{H}}, & 0 \leq \frac{\Delta}{H} < 1, \\
\frac{1792}{\pi} \frac{3 + \frac{\Delta}{H}}{186 + 1039 \frac{\Delta}{H}}, & 1 \leq \frac{\Delta}{H} \lesssim O(10),
\end{cases} \tag{4.7}$$

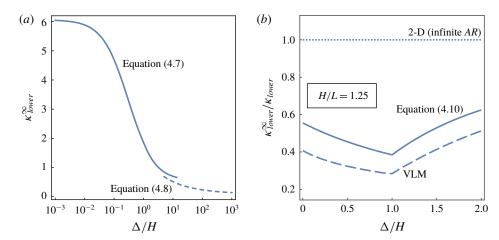


FIGURE 6. (Colour online) Dimensionless flow speeds, κ_{lower}^{∞} and κ_{lower} , for the divergence instability of a trapezoidal inverted flag; superscript ∞ indicates infinite aspect ratio AR. (a) Theory for the divergence instability of a trapezoidal inverted flag of infinite AR. Dimensionless flow speed, κ_{lower}^{∞} , as a function of the shape parameter, Δ/H , for an inverted flag of trapezoidal morphology (no petiole). Equation (4.7) holds for $\Delta/H \lesssim O(10)$, while (4.8) is derived for $\Delta/H \gg 1$. This theory is valid for $H/L \to \infty$, while measurements are performed on sheets of finite H/L = O(1). (b) Finite aspect ratio correction for the dimensionless flow speed of the trapezoidal sheets studied in § 3. Comparison of (4.10), which makes use of the generalised aspect ratio in (4.11), to numerical results obtained with the vortex lattice method (VLM).

where the origin of the O(10) upper limit is explained below in §4.2.1. Note that (4.7) gives $\kappa_{lower}^{\infty} = 1.86$ for an inverted flag of rectangular morphology, i.e. $\Delta/H=1$, whereas the full power series approach of Sader *et al.* (2016a) gives $\kappa_{lower}^{\infty} = 1.85$, which is in excellent agreement.

Figure 6(a) provides numerical results for κ_{lower}^{∞} as a function of the trapezoidal shape parameter, Δ/H , showing that the flow speed at which divergence occurs increases with decreasing Δ/H . For $\Delta/H=0$, corresponding to a triangular sheet whose leading edge is a point, equation (4.7) gives $\kappa_{lower}^{\infty}=6.06$. In the opposite limit, $\Delta/H\to\infty$, which is an inverted flag with a very wide leading edge relative to its clamped end – i.e. a reversed triangle – (4.7) predicts $\kappa_{lower}^{\infty}=0.549$. However, the mode shape in (4.5) obviously does not hold when $\Delta/H\gg 1$, since the flag's rigidity at the clamp will approach zero relative to its leading edge.

In the following subsection, we therefore examine the effect of varying the deflection function (4.5) on the calculated κ_{lower}^{∞} . This is performed for the limiting cases of small and large Δ/H .

4.2.1. Varying the deflection function

 $\Delta/H=0$ corresponds to a triangular sheet whose leading edge is a point. Moreover, (4.5) is derived for a rectangular cantilevered sheet under a static point load at its free end. We therefore vary the mode shape by instead using the corresponding deflection for a triangular cantilever sheet, which gives $w(x, z) = (1 - x)^2$. Calculating the hydrodynamic pressure from (4.1), using this new mode shape, and substituting the result into (4.4), gives $\kappa_{lower}^{\infty} = 64/(3\pi) = 6.79$; this is to be compared to $\kappa_{lower}^{\infty} = 6.06$,

obtained using (4.7). Since the Rayleigh quotient in (4.4) guarantees an upper bound (Leissa 1969), this analysis establishes that $\kappa_{lower}^{\infty} = 6.06$ is more accurate.

The value $\Delta/H \gg 1$ is an inverted flag with a very wide leading edge relative to its (trailing) clamped end, i.e. a reversed triangle. Here, only a very small strip of elastic material joins the sheet to the clamp, relative to the large lift surface. The inverted flag will therefore deflect in a linear (and rigid body) fashion away from the clamp, with a small 'boundary layer' near the clamp that bends elastically. Any non-zero flow speed will deflect the flag in this limit, giving $\kappa_{lower}^{\infty} \to 0$ as $\Delta/H \to \infty$, contrary to the above result for a reversed triangle of $\kappa_{lower}^{\infty} = 0.549$. To account for this expected deformation, the deflection function in the numerator of (4.4) is approximated by that of a trapezoidal plate with a point load at its leading edge, obtained using Euler–Bernoulli beam theory (Landau & Lifshitz 1970). The pressure is evaluated using the expected linearly varying mode shape away from the clamp (rigid body rotation pivoted at the clamp); this mode shape together with the evaluated pressure is used to calculate the denominator of (4.4). This analytical approach captures this asymptotic limit, and gives the following expression:

$$\kappa_{lower}^{\infty} = \frac{24\left(\frac{\Delta}{H} - 1\right)^{3}}{\pi\left(4 + 5\frac{\Delta}{H}\right)\left[2\left(\frac{\Delta}{H}\right)^{2}\log\left(\frac{\Delta}{H}\right) + \frac{\Delta}{H}\left(4 - 3\frac{\Delta}{H}\right) - 1\right]}, \quad \frac{\Delta}{H} \gg 1, \quad (4.8)$$

which is also plotted in figure 6(a). The comparison in figure 6(a) shows that (4.8) provides a natural continuation to very large Δ/H of the expression in (4.7), which is valid only up to moderately large Δ/H . In the asymptotic limit $\Delta/H \to \infty$, equation (4.8) gives

$$\kappa_{lower}^{\infty} \sim \frac{24}{5\pi \left[2 \log \left(\frac{\Delta}{H} \right) - 3 \right]},$$
(4.9)

showing that an infinitesimal flow speed will lead to divergence in this limit, as expected.

4.3. Arbitrary planforms of finite aspect ratio

The above calculations do not account for the three-dimensional geometry of an inverted flag. To include the effect of finite generalised aspect ratio AR, we utilise a similar approach to that reported by Sader *et al.* (2016a). For a flat sheet of arbitrary generalised aspect ratio AR, it is well known that Prandtl's lifting line theory (Jones 1990; Anderson 2006) predicts a reduction in lift slope by factor of approximately 1+2/AR, ignoring the usual (small) Glauert parameter. This reduction in lift is due to the generation of side edge vortices that are always present on wings of finite aspect ratio. Therefore, the dimensionless (squared) flow speed κ_{lower} at which divergence occurs for an inverted flag of arbitrary shape and finite aspect ratio, relative to its infinite AR value, κ_{lower}^{∞} , is enhanced by this same factor, giving the result

$$\kappa_{lower} \approx \kappa_{lower}^{\infty} \left(1 + \frac{2}{AR} \right),$$
(4.10)

where κ_{lower}^{∞} for a trapezoidal morphology is specified by (4.7) and (4.8). Note that for $AR \sim O(1)$, which is typical in the measurements reported in § 3, the value of κ_{lower} is approximately three times larger than κ_{lower}^{∞} . This shows that side edge vortices play an important role in controlling the stability of inverted flags of finite aspect ratio.

To validate (4.10), a comparison of the predictions of (4.10) to those obtained using a VLM (Tornado 2015; Sader *et al.* 2016*a*) is given in figure 6(b). This is performed for sheets with H/L=1.25, which corresponds to those studied experimentally in § 3. The generalised aspect ratio AR (defined in (1.4)) for a trapezoidal sheet is

$$AR \equiv 2 \frac{\max(\Delta, H)^2}{(H + \Delta)L}.$$
(4.11)

Vortex lattice method simulations involve calculating the lifting slope of a rigid trapezoidal sheet as a function of its shape parameter, Δ/H ; they provide a more accurate and rigorous approach as compared to (4.10), if needed. The data in figure 6(b) show that (4.10) provides a good approximation. We note that lifting line theory – which is used to derive (4.10) – makes no distinction between swept and symmetric wings, and is normally applied to symmetric wings only (Jones 1990; Anderson 2006). Even so, equation (4.10) captures the VLM simulations well. Note that the curves in figure 6(b) exhibit cusps at $\Delta/H = 1$. This is consistent with experimental observations; see figures 4(a) and 7. This feature arises from the definition of the generalised aspect ratio for a trapezoid, equation (4.11), and our choice of fixing H and L, and varying Δ/H .

4.4. Comparison of theory and measurement

Figure 7 provides a comparison of the measurement data of § 3 for the divergence instability (at which large-amplitude flapping begins) to the theory of §§ 4.1–4.3 in (4.7), (4.10) and (4.11). Data for two different flag thicknesses, and hence mass ratios $\mu = 0.40$ and 0.61, are reported. This allows assessment of the expected independence of the divergence instability on μ .

Our theory for κ_{lower} exhibits the same qualitative behaviour as measurements and increases in a nonlinear fashion for decreasing shape parameter, Δ/H . The 2-D strip model of (4.7) for infinite AR strongly underestimates the measured data, but when corrected using (4.10) for finite AR yields good agreement. Notably, this finite AR theory slightly overestimates the measurements, which may be due to two independent effects. First, the strip theory of § 4.1 uses an approximate mode shape and thus strictly provides an upper bound on κ_{lower}^{∞} , as discussed. This effect is expected to be small given the good agreement with the exact result for a rectangular planform. Second, the Prandtl lifting line correction in (4.10) implicitly assumes the vortex lines are parallel to the sheet. This is also an approximation and ignores the nonlinear lift generated by the true non-parallel nature of the vortex lines which increases lift (Bollay 1939; Sader et al. 2016b). The omission of this (nonlinear lift) factor leads to (4.10) overestimating the true flow speed at which divergence occurs, as was demonstrated for an inverted flag of rectangular planform by Sader et al. (2016b). The present measurements for a trapezoidal planform are entirely consistent with Sader et al. (2016b), showing that the theoretical model for κ_{lower} in §§ 4.1–4.3 robustly predicts the divergence instability initiating large-amplitude flapping as a function of inverted flag morphology.

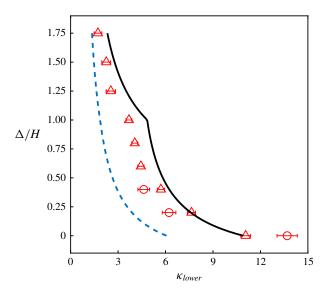


FIGURE 7. (Colour online) Comparison of dimensionless flow speed at divergence, κ_{lower} , obtained experimentally for two mass ratios: $\mu = 0.40$ (\triangle) and $\mu = 0.61$ (\bigcirc) (see § 3), using the theory for infinite aspect ratio in (4.7) (dashed line) and with the finite aspect ratio correction in (4.10) (solid line). Error bars quantify the uncertainty in measuring κ_{lower} for a single sheet, and define the 25% and 75% points discussed in § 3.

5. Rectangular inverted flag with a petiole

Finally, we examine the effect of adding a slender petiole to an inverted flag with a much wider rectangular planform. The dimensions and properties of these flags are detailed in table 1, and a total of 5 different inverted flags are studied; Flags A–E; unlike the other flags, Flag E contains no petiole and is measured for reference only. Measurement of each petiole alone, i.e. without its elastic sheet attached, in the wind tunnel shows that the flow has no perceivable effect on its motion at any obtainable flow speed. This establishes that the petiole merely acts as an elastic support – in essence, an end-loaded cantilever – that only exhibits a bending mode. This property is used in the discussion and theory that follows.

5.1. General observations

A sample of measurements on Flag B is given in figure 8, which shows stroboscopic images and time series of this flag's motion as a function of the impinging flow speed, U. All major behavioural regimes observed previously for inverted flags without a petiole are still found. Namely as the flow speed increases, the flag transitions from a straight (undeflected) mode to a flapping mode (periodic and chaotic), and finally, a deflected mode.

All flags exhibit deformation primarily in the petiole, i.e. the elastic sheet moves predominantly as a rigid (and straight) body. This is in contrast to flags without a petiole, where deformation occurs throughout the entire elastic sheet, cf figures 1(c) and 8. These observations establish that the petiole can provide the primary source of elasticity to the flag when present, behaving as an elastic hinge. Reducing the petiole length, L_p , or increasing the petiole rigidity beyond a critical value results in mechanical failure of the elastic sheet, where large stresses at its junction point

542 B. Fan, C. Huertas-Cerdeira, J. Cossé, J. E. Sader and M. Gharib

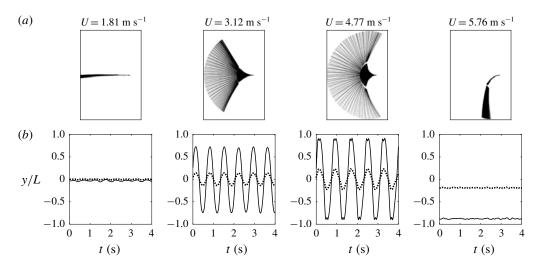


FIGURE 8. Stroboscopic images (a) of Flag B at various flow speeds and corresponding time series (b) of the y-coordinate of the elastic sheet's free end (solid lines) and petiole—sheet junction point (dotted lines).

with the petiole eventually tear the polycarbonate sheet over a few flapping cycles. This observation suggests that damage from flapping in an inverted orientation could impose structural constraints in real leaves. Indeed, it may be a reason why the petiole often extends into the leaf's lamina and gradually tapers off (rather than abruptly ending), while veins into the lamina provide further structural support.

Figure 9(a) gives measurements of (i) the (peak-to-peak) amplitude, A, at free end of the elastic sheet, and (ii) the (peak-to-peak) amplitude, A_p , at the end of the petiole, for each flag as a function of the impinging flow speed, U. The dynamics of the amplitudes, A and A_p , aligns with respect to the flow speed, U. This is consistent with the flag deforming in a monotonically increasing fashion from its clamp to its free end, as seen in figure 8. Flag C, with the longest and most rigid petiole relative to its elastic sheet, requires the highest critical flow speed to initiate and cease flapping. This flag also displays the highest natural resonant frequency and flapping frequency. Interestingly, large-amplitude flapping in Flag E (that does not contain a petiole) is initiated and stops at comparable flow speeds to those of Flag C. At first sight, this appears to coincide with the flapping and natural resonant frequencies of these two flags being similar; see table 1. However, Flag A also contains similar frequencies but displays very different behaviour; see figure 9. This demonstrates that the inverted flag's stability is not determined solely by the presence of a petiole nor by its flapping and natural resonant frequencies, with a more detailed analysis being required.

5.2. Mechanism for the initiation of flapping

The above observations that the petiole (i) is unaffected by the impinging flow, and (ii) provides the primary source of elasticity to the flag, suggests a decoupling of the hydrodynamic and elastic properties of these flags. This in turn enables the development of a simple theoretical model for the divergence instability that makes use of the lift generated by a rigid flat sheet at small angle of attack. It remains to be seen if such a divergence instability initiates large-amplitude flapping, as found

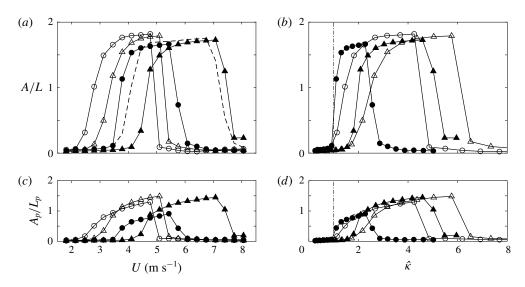


FIGURE 9. Dimensionless (peak-to-peak) flapping amplitude of the elastic sheet's free end (a,b) and at the petiole—sheet junction point (c,d) for Flag A (\bullet), Flag B (\bigcirc), Flag C (\blacktriangle), Flag D (\vartriangle) and Flag E (dashed line). (a,c) Raw data plotted as a function of the dimensional flow velocity, U. (b,d) Same data as in (a,c) plotted versus the rescaled dimensionless flow speed, \hat{k} , defined in (5.5). This rescaling is theoretically predicted to collapse data for the initiation of large-amplitude flapping onto $\hat{k}=1$, which is the divergence flow speed in (5.3); shown as vertical dash-dotted line. Experimentally measured values for the dimensionless flow speeds: $\kappa_{lower}=0.038\pm0.007$, 0.050 ± 0.012 , 0.106 ± 0.015 and 0.124 ± 0.037 for Flags A, B, C and D, respectively. Error bars quantify the uncertainty in measuring κ_{lower} , and define the 25% and 75% points discussed in § 3.

for flags without a petiole. The theory developed in this section enables this issue to be explored.

The dimensional pressure jump across a flat sheet of length, $(L - L_p)$, at angle of attack, α , to an impinging flow of speed, U, follows from (4.1) and is given by

$$p = 2\rho U^2 \alpha \sqrt{\frac{L - L_p}{x} - 1},\tag{5.1}$$

where x is the dimensional coordinate in the flow direction, whose origin is at the leading edge of the rigid sheet; see figure 3(a). This rigid sheet is attached to the petiole, a cantilevered elastic beam of length L_p , whose deflection function, w(x), obeys the Euler–Bernoulli equation (Landau & Lifshitz 1970). It then follows that the force, F, and moment, M, exerted by the flat plate on the end of the petiole are

$$F = \pi \rho U^2 H(L - L_p) w'(L_p), \quad M = -\frac{3\pi}{4} \rho U^2 H(L - L_p)^2 w'(L_p), \quad (5.2a,b)$$

where the rigid sheet's angle of attack, $\alpha \equiv -w'(L_p)$. The small tab-like overlap hinge connecting the sheet and petiole has its own bending stiffness, which is ignored in the present model, i.e. this hinge is assumed to be infinitely stiff.

The Euler-Bernoulli equation for this end-loaded beam is w''''(x) = 0, with boundary conditions, w(L) = w'(L) = 0, $w''(L - L_p) = -M/EI$ and $w'''(L - L_p) = F/EI$, where

544 B. Fan, C. Huertas-Cerdeira, J. Cossé, J. E. Sader and M. Gharib

 $EI = Eh_p^3 H_p/12$. Solving this eigenvalue problem gives the required dimensionless flow speed for which divergence occurs,

$$\kappa_{lower}^{pet} \equiv \frac{\rho U_{lower}^2 L_p^2 \left(L - L_p \right)}{D'} \approx \frac{4}{\pi} \left(1 + \frac{2}{AR} \right) \frac{L_p}{3L - L_p} \frac{H_p}{H},\tag{5.3}$$

where $D' \equiv E h_p^3/12$ is the flexural rigidity of the petiole, U_{lower} is the critical flow speed for divergence and the Prandtl's lifting line correction, (4.10), for the sheet's finite generalised aspect ratio, AR, is included, with

$$AR = \frac{H}{L - L_n}. ag{5.4}$$

We emphasise that (5.3) assumes that the flag's elasticity is isolated to the petiole; see above.

Figure 9(b) presents the same data as in figure 9(a) but now plotted as a function of the rescaled dimensionless flow speed,

$$\hat{\kappa} \equiv \frac{\kappa'}{\kappa_{lower}^{pet}},\tag{5.5}$$

where $\kappa' \equiv \rho U^2 L_p^2 (L - L_p)/D'$. Equation (5.5) implicitly contains the divergence instability's geometric dependence in (5.3) for an inverted flag with a petiole; divergence is theoretically predicted to occur at $\hat{\kappa} \approx 1$. Data for Flag E are omitted from figure 9(b) because this case does not have a petiole.

This rescaled plot shows that (5.3) accurately captures the divergence instability for Flag A – with data collapsing onto $\hat{\kappa} \approx 1$ in the flow speed region where largeamplitude flapping is initiated (at low flow speed). This rescaling also works well for Flag B, although divergence (and the initiation of flapping) occurs at a slightly higher value of $\hat{\kappa}$. The reason for this difference is evident in stroboscopic images of these flags in the flapping regime; see figure 10. While the theory derived in this section assumes the deflection angle of the petiole and elastic sheet are identical at their connection position, this is not satisfied exactly in measurements: Flags A and B displaying different behaviours at this connection point. The thicker (and hence more rigid) the petiole the stronger the discontinuity, which leads to a larger violation of the theory's assumptions for Flag B relative to Flag A; the hinge connecting the sheet and petiole can deform to a greater degree, violating the model's assumption of an infinitely stiff hinge. Even so, this is found to exert a small effect with the derived approximate theory performing well and accurately capturing the divergence instability for both Flags A and B with no fitting parameters. We emphasise that the theory contains a number of approximations - in addition to the rigid hinge assumption and therefore precise agreement between measurement and theory is not expected.

In contrast, the divergence instability and onset of flapping is underestimated by the developed theory for Flags C and D, with divergence occurring at $\hat{\kappa} \approx 2$ (>1). The stroboscopic images in figure 10 also provide some insight into this difference, with strong twisting behaviour in Flags C and D. This is manifest in an apparent discontinuity in the displacement between the petiole and the sheet, which is greatly reduced in Flags A and B. Such behaviour, due to the coupling effects of gravity and twisting of the petiole (the sheet deforms in a rigid fashion with the inverted flag displaying a waving motion as it flaps), is not included in (5.3) and appears to

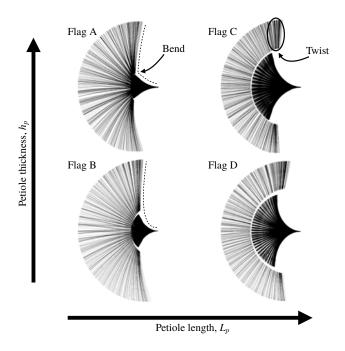


FIGURE 10. Stroboscopic images of Flags A-D in their respective flapping regimes at various flow speeds, which are selected so that the flags exhibit similar flapping amplitudes. Flag A: $U = 4.77 \text{ m s}^{-1}$. Flag B: $U = 4.11 \text{ m s}^{-1}$. Flag C: $U = 6.09 \text{ m s}^{-1}$. Flag D: $U = 4.77 \text{ m s}^{-1}$. Flag A exhibits a notable kink where the petiole connects the elastic sheet, which is smaller than that for Flag B (as indicated by dashed lines). Flags C and D display significant twisting in their elastic sheets, which is manifest by an apparent discontinuity at the connection between the sheet and petiole – the images are taken in the gravity direction.

be causing the divergence instability to occur at twice the theoretical value. Even so, increasing the petiole thickness enhances the value of $\hat{\kappa}$ at which divergence is observed, as for Flags A and B. Similar twisting behaviour has been previously observed for slender inverted flags (Sader *et al.* 2016b). Reducing the petiole length enhances its torsional stiffness and decreases this behaviour; see figure 10. Finally, we note that the observed discrepancies between theory and measurement for Flags C and D may also be related to differences in flow processes, e.g. vortex formation. Flow visualisation remains an avenue for future work.

For Flags A–D, the undeflected mode of the inverted flag abruptly gives way to large-amplitude flapping. This shows that a divergence instability triggers this flapping instability, as observed for inverted flags without a petiole. Finally, we note that the rescaling in (5.5) does not collapse data at higher flow speed where flapping ceases and the deflected mode appears. Thus, the empirical relation of $\kappa_{upper}/\kappa_{lower}\approx 4$ observed for trapezoidal inverted flags with no petiole (§ 3) does not hold for rectangular flags with petioles. Adding a petiole to an inverted flag can therefore enable adjustment of the relative flow speed range over which large-amplitude flapping occurs.

6. Conclusions

Motivated by the form and occurrence of natural leaves, we have studied the effect of morphology on the stability and dynamics of inverted flags. This involved a combination of experiment and theory with good agreement observed throughout. Two approaches were taken to examine this effect: (i) changing the shape of the inverted flag's elastic sheet using a trapezoidal morphology, and (ii) adding a petiole to the widely studied rectangular elastic sheet. This showed that the inverted flag flapping instability, as first reported by Kim *et al.* (2013), is highly robust and persists across a wide range of morphologies.

Trapezoidal inverted flags (without a petiole) were found to exhibit large-amplitude flapping that depends strongly on their morphology, Δ/H . Strikingly, it was observed that the ratio of the flow speed where flapping ceases to that of its onset is $\kappa_{upper}/\kappa_{lower} \approx 4$ over the entire geometric range studied, $0 \leqslant \Delta/H \leqslant 1.75$; it remains to be seen if a similar result holds for other morphologies. A general theoretical model for the onset of a divergence instability, which initiates flapping at low flow speeds, was presented in § 4 for inverted flags of arbitrary morphology. This was applied to the above-mentioned trapezoidal inverted flags, yielding an analytical solution that was found to exhibit good agreement with measurement.

Second, inclusion of a petiole to inverted flags with rectangular morphologies was observed to result in rigid body motion of the flag's elastic sheet, with the petiole providing the requisite elasticity – thus, decoupling hydrodynamic and elastic forces. This motivated development of a simple analytical model for the divergence instability of these flags. Twisting of the sheet was observed for some flags, and linked to an increase in the flow speed at which divergence occurred. Further work may examine the effect of coupling the elastic sheet's morphology to that of a petiole and non-idealities in the dynamics of the flag's motion.

Returning to a biological setting, the white poplar leaf combines a roughly triangular lamina with a long petiole, two features that our study suggests have counteracting effects. The former would increase the critical flow speed for the onset of flapping while the latter would decrease it. Our work therefore motivates wind tunnel experiments on a variety of real tree leaves that could potentially shed light on whether there exists an optimal leaf morphology to resist wind damage in the inverted orientation. Finally, a leaf is not an isolated system in nature but rather part of a larger system of interacting leaves and branches. This study therefore represents only but a first step in understanding such a complex system, which may exhibit new and unexplored phenomena.

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