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# Course Content Overview

01/10/22

1. Measurement & Application of Acoustic Intensity Method
2. Acoustic Filters (Transfer Matrix Method)
3. Sound Transmission Through Barrier Systems
4. Duct Acoustics - higher order Modes
5. Room Acoustics - porous Matls.

# I.0 Acoustic Intensity

Intensity Methods are used to:

- measure sound power of a source
- measure component sound powers
  - rank the noise sources
  - rank & identify paths
- measure transmission loss of barrier systems
- visualize radiation fields
  - impress bass

"Modern" Implementation - signal processing

- fft

Two-Microphone cross spectral procedure

## I.1 Instantaneous Intensity

Intensity = Sound Power / Unit Area (flux)

$$= \frac{(\text{Force})}{\text{Area}} \cdot \text{Velocity} = \text{Pressure} \cdot \text{Velocity}$$

$$\bar{I}_{\text{inst}} = p(t) \bar{u}(t) \quad \begin{matrix} \text{real vector} \\ \text{real sound} \end{matrix} \quad \begin{matrix} \text{particle velocity} \\ \text{pressure} \end{matrix}$$

$$94 \text{ dB} \rightarrow P_{\text{rms}} = 1 \text{ Pa} \dots \frac{P}{u} = \rho v c \Rightarrow u = \frac{P}{\rho v c} = \frac{7.4 \text{ MM}}{\text{s}}$$

## 1.2 Time-Averaged Acoustic Intensity

$$\bar{I} = \frac{1}{T} \int_0^T I_{\text{inst}} dt \quad T = \text{signal period}$$
$$= \frac{1}{T} \int_0^T \rho(t) \bar{u}(t) dt$$

Complex harmonic Signals

$$P(r,t) = P(r) e^{j\omega t} \quad \bar{u}(r,t) = \bar{u}(r) e^{j\omega t}$$

$$I = \operatorname{Re} \{ P \cdot u^* \} \cdot \frac{1}{2}$$

## 1.3 Measurement of Particle Velocity

(i) surface normal velocity

vibrating  $\frac{\uparrow}{\downarrow}$   $\square$  accelerometer / LDV

$$I = \frac{1}{2} \operatorname{Re} \{ P u_n^* \}$$

Cons - Temperature

(Accel) - mass loading

- very local - time consuming

- cross calibration is difficult

- know intensity only at the surface

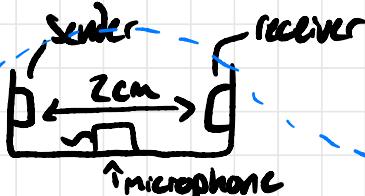
Reminder - Intensity is Time-Averaged

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$$\bar{I} = \frac{1}{2} \operatorname{Re} \{ P_{\text{int}} \}$$

Measurement methods cont.

ii) ultrasonic probe  
(Trondheim)



→ field pressure  
 $\omega$

Sending Transducer emits ultrasonic signal (100 kHz)  
(W<sub>p</sub>)

$$W_c \ll \lambda_p \ll \lambda_s$$

- instantaneous sound speed in the probe region fluctuates as the external sound field moves air back and forth
  - causes the phase of the probe to fluctuate

$$C_x(t) = C_0 + u_x(t)$$

$C_0$  <sup>2</sup> ambient sound speed       $u_x(t)$  <sup>2</sup> particle velocity associated with the external sound field

Receiver at  $c^{-d} k_p d$  <sup>2</sup> phase

$k_p$  <sup>2</sup> probe wave number ...  $k_p = \frac{w_p}{C_0 t} \sim \text{probe free speed}$

$$u(t) = -k_p d = \frac{-w_p d}{C_0 + u_x(t)} = \frac{-w_p d}{C_0 \left(1 + \frac{u_x(t)}{C_0}\right)}$$

$$Q(t) \approx -\frac{w_p}{C_0} d \left(1 - \frac{u_x(t)}{C_0}\right)$$

<sup>2</sup> acoustic Mach number  
cc1

$$Q(t) = Q_0 + Q'(t) \quad Q'(t) = \frac{k_{\text{pd}}}{C_0} U_x(t) \dots k_0 = \frac{\omega p}{C_0}$$

const.

$$U_x(t) = \frac{C_0}{k_{\text{pd}}} Q'(t)$$

if phase can be continuously measured  
particle velocity can be found

to measure intensity, combine with microphone

$$I_x = \frac{1}{2} \operatorname{Re} \{ P \cdot U^* \}$$

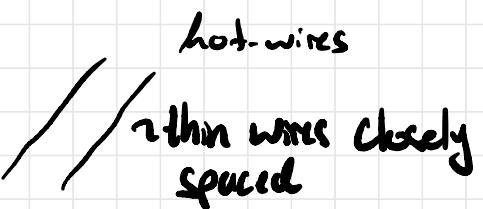
Problems: probe tends to affect the sound field

- sensitive to ambient fluid motion
- calibration is tricky

(iii) microflown TU Wente

MEMS-based...

hot-wires



-measuring temperature fluctuations  
-infer particle velocity

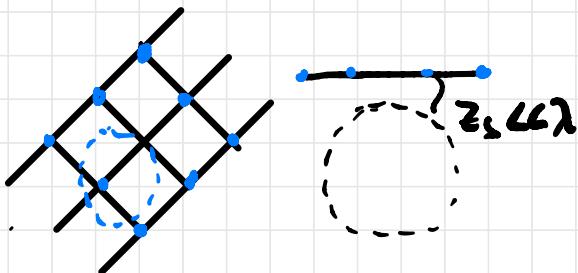
-incorporate a microphone

Problems:

- cost
- calibration

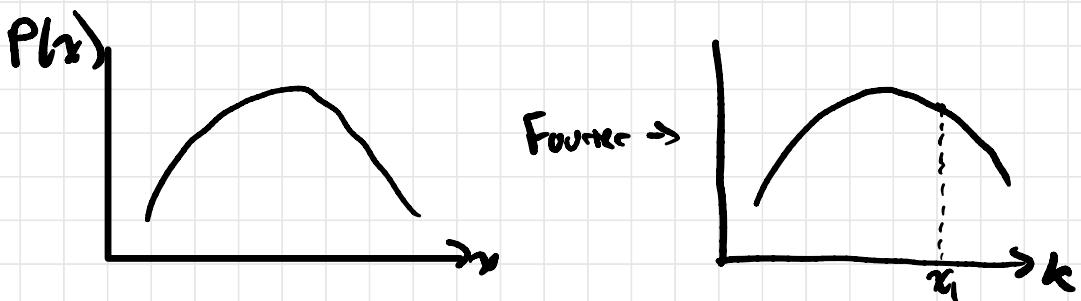
## (iv) Holographic Procedure

### NAR - Nearfield Acoustical Holography

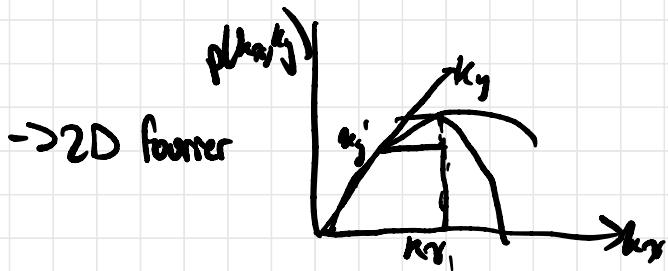
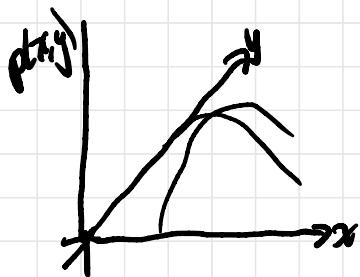


Measurements close to  
Source

- measure spatial distribution of sound pressure  
close to the source



Create wave number spectrum



We are measuring sound, which satisfies the wave equation

$$k^2 = k_x^2 + k_y^2 + k_z^2 \quad k = \frac{\omega}{c}$$

Knowing  $k_x, k_y,$

$$\rightarrow k_z = \sqrt{k^2 - k_x^2 - k_y^2}$$

$$e^{-j(k_x x + k_y y + k_z z)} \sim \text{plane wave}$$

Plane wave decomposition

Sound field is expressed as the sum of an infinite number of plane wave components traveling in all directions

for a freely prop. plane wave  $f_n = f_{\text{SC}}$

can find  $\bar{u}$  by the relation ...

$$\tilde{u}_p(k_x, k_y, k_z) = \rho(k_x, k_y, k_z) / \rho_{\text{SC}}$$

3D particle velocity field can be found in this way

$$P \begin{pmatrix} y \\ x \end{pmatrix} \xrightarrow{\text{FT}} P \begin{pmatrix} k_y \\ k_x \end{pmatrix} \Rightarrow u \begin{pmatrix} k_y \\ k_x \end{pmatrix}$$

→ inverse FT  $u \begin{pmatrix} y \\ x \end{pmatrix}$  at every point can calculate intensity

# Acoustic Intensity

$$I = \frac{1}{2} \operatorname{Re} \{ \rho u^2 \}$$

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## Nearfield Acoustic Holography cont.

Pros - fast

- can project the sound field towards or away from the source

Cons - expensive

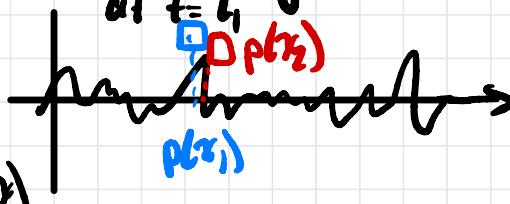
- complicated

## v) Two-Microphone procedure



use linearized momentum equation to relate particle velocity to the pressure gradient

$$\nabla P = -\rho_0 \frac{du}{dt}$$



→  $P(x_1) \quad P(x_1 + \Delta x)$

$$\frac{dP}{dx} = \lim_{\Delta x \rightarrow 0} \frac{P(x_1 + \Delta x) - P(x_1)}{\Delta x}$$

good estimate when  $\Delta x$  is small compared to  $\lambda$   
... if  $\frac{\Delta x}{\lambda} < \frac{1}{10}$  less than a 5% error in the gradient

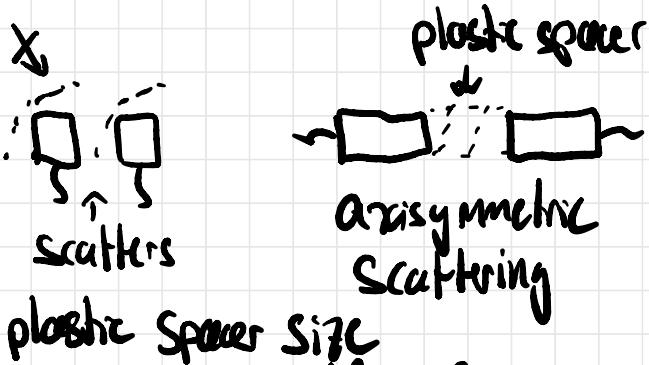
$$\frac{dP}{dx} \approx \frac{P(x_1 + \Delta x) - P(x_1)}{\Delta x}$$

pressure gradient is estimated at the midpoint between microphones.

for 3D gradient



can use 6 mics to measure at "origin"



minimizes directional affects due to scattering from mics + preamps

plastic Spacer size

$\Delta = 12.5 \text{ mm}$  (high frequency,  $f < 5 \text{ kHz}$ )

$\Delta = 50 \text{ mm}$  (low frequency,  $f \geq 1250 \text{ Hz}$ )

Consider the complex harmonic representation

$$p(r, t) = P(\bar{r}, \omega) e^{j\omega t}$$

$$\bar{u}(r, t) = \bar{u}(\bar{r}, \omega) e^{j\omega t}$$

$$-\nabla P = \rho_0 \frac{dU}{dx} \Rightarrow -\nabla P = j\omega \rho_0 \bar{u} \quad \bar{u} = -\frac{\nabla P}{j\omega \rho_0}$$

$$u_x \approx = \frac{P(x+\Delta x) - P(x)}{\omega P_0 (\Delta x)}$$

$$\boxed{P} \times \boxed{P}$$

particle velocity  
at this point

## 1.4 Calculation of Acoustic Intensity

### 1.4.1 Two-Microphone Cross-spectral Method.

$$\bar{I}(\bar{r}, \omega) = \frac{1}{2} \operatorname{Re} \{ P(\bar{r}, \omega) \bar{u}^*(\bar{r}, \omega) \}$$

i) estimate of pressure spectrum

$$\boxed{P_1} \times \boxed{P_2}$$

$$P_x = \frac{P_1 + P_2}{2}$$

ii) estimated particle velocity spectrum

$$u_x = \frac{(P_2 - P_1)}{\omega P_0 \Delta} = \frac{1}{\omega P_0 \Delta} (P_2 - P_1)$$

iii) x-component of intensity

$$I_x = \frac{1}{2} \operatorname{Re} \{ P u^* \} = \frac{1}{2} \operatorname{Re} \left\{ \frac{P_1 + P_2}{2} \left[ \frac{1}{\omega P_0 \Delta} (P_2 - P_1) \right]^* \right\}$$

$$I_x = \frac{1}{2} \operatorname{Re} \left\{ -\frac{1}{2 \omega P_0 \Delta} (P_1 P_2^* + P_1 P_2^* - P_2 P_1^* - P_2 P_1^*) \right\}$$

$$(P_1^* P_2) = (P_1 P_2^*)^* \Rightarrow a+jb \text{ and } a-jb$$

$$a+jb - (a-jb) = j2b$$

$$\frac{1}{2} \operatorname{Re} \left\{ \frac{1}{2 \pi j \Delta} \operatorname{Im} \{ P_1 P_2^* \} \right\} = I_x$$

$$I_x = \frac{1}{2} \operatorname{Im} \{ P_1 P_2^* \} \cdot \frac{1}{\Delta}$$
$$= - \frac{\operatorname{Im} \{ P_1^* P_2 \}}{2 \pi \Delta} \quad \text{Cross-spectrum of the two pressures}$$

use this formula to compute Intensity  
in the direction joining the two microphones

# Acoustic Intensity

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for complex harmonic signals,  $\text{cout}$

$$I_x = \frac{1}{2} \operatorname{Re}\{P u_x^*\}$$

particle velocity estimate

$$u_x = \frac{p(x+\Delta) - p(x)}{2w_{fb}\Delta}$$

$$\begin{array}{ccc} P_1 & P_2 \\ p(x) & p(x+\Delta) \\ \boxed{[x-\Delta, x]} & \mapsto x \end{array}$$

pressure spectrum estimate

$$P(w) = \frac{p(x) + p(x+\Delta)}{2}$$

## Acoustic Intensity Estimate

$$I_x = -\operatorname{Im}\{P_1^* P_2\}$$

$$I_x = \frac{-\operatorname{Im}\{P_1^* P_2\}}{2w_{fb}\Delta}$$

## 1.5 Sources of Error

### 1.5.1 Finite Difference Error

$$\frac{\partial p}{\partial x} = \frac{p(x+\Delta) - p(x)}{\Delta}$$



$$\rightarrow \begin{array}{c} P_1 \\ \boxed{x=0} \end{array} \quad \begin{array}{c} P_2 \\ \boxed{x=\Delta} \end{array} \quad p(x,t) = A e^{j\omega t} e^{j\theta(t)} \\ u_x(x,t) = \frac{A}{P_{fb}} e^{j\omega t} e^{j\theta(t)}$$

$$\text{Exact Intensity: } \frac{1}{2} \operatorname{Re}\{P u_x^*\} = \frac{1}{2} P_c \left\{ \frac{AA^*}{P_{fb}} \right\} = \frac{|A|^2}{2P_{fb}}, \text{ rmsp}$$

Measured intensity:  $P = A e^{-\delta k r} e^{j \omega t}$

$$P_M = \frac{P_1 + P_2}{2} = \frac{A + A e^{-j \Delta}}{2}$$

$$u_m = \frac{j}{w p_0 \Delta} (P_2 - P_1) = \frac{j}{w p_0 \Delta} [A e^{-j \Delta} - A]$$

$$P_M u_m^* = -\frac{j |A|^2}{2 w p_0 \Delta} [e^{j \Delta} - e^{-j \Delta}]$$

$\brace{2 \sin k \Delta}$

$$= \frac{|A|^2}{w p_0 \Delta} \sin k \Delta \quad k = \frac{\omega}{c} \dots \omega = k c$$

$$= \frac{|A|^2}{p_0 c} \cdot \frac{\sin k \Delta}{k \Delta}$$

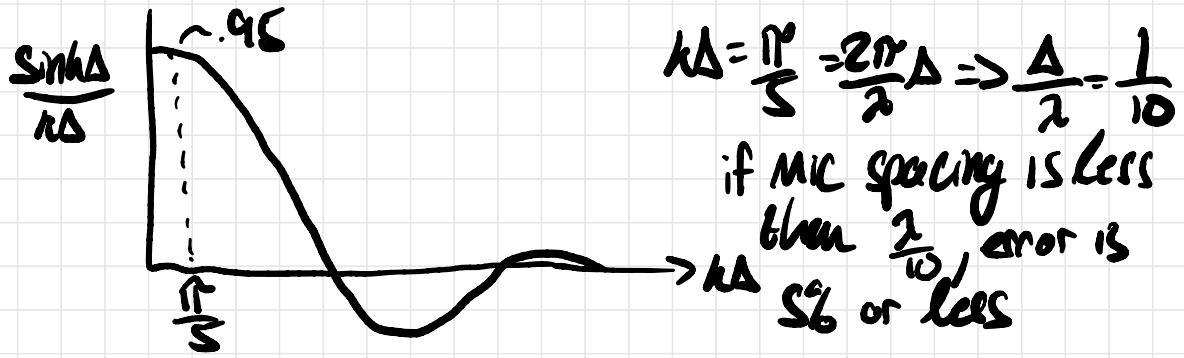
$\brace{-}$  finite difference error  
 $\brace{-}$  exact intensity

$$k \Delta = \frac{2 \pi}{\lambda} \Delta = 2 \pi \left( \frac{\Delta}{\lambda} \right)$$

non-dimensional MIC separation

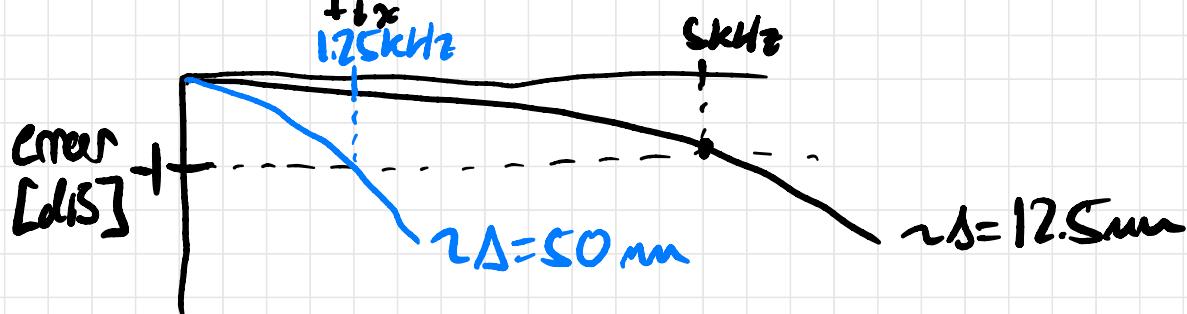
$$\frac{I_{mx}}{I_{bx}} = \frac{\sin k \Delta}{k \Delta}$$

- { finite difference for a plane wave
- { independent of position
- { frequency dependent
- { always less than 1
- { bias error (always one direction)



- magnitude is always less than the true intensity
- error increases with frequency for a fixed  $\Delta$
- establishes a high frequency limit on an accurate measurement.

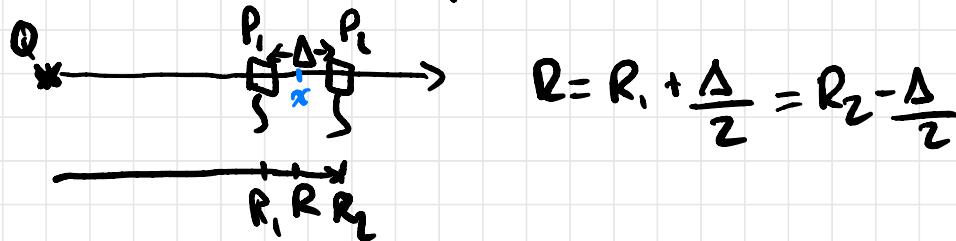
$$\text{Error} = \frac{I_{bx} - I_{nx}}{I_{bx}} = 1 - \frac{\sin k\Delta}{k\Delta}$$



## 1.5.2 Near field Error

-proportional to distance from the source

Monopole in free space



i) exact intensity:  $P(R, t) = \mu_0 \rho_0 c Q \frac{e^{-jkr}}{4\pi R} e^{j\omega t}$

$$U_r(R, t) = \frac{1}{4\pi R} \frac{\partial P}{\partial r} = \frac{Q}{4\pi r} \left[ \frac{1}{R^2} + jk \right] e^{-jkr} e^{j\omega t}$$

<sup>↑</sup>  
near field term

$$I_{tr} = \frac{1}{2} \operatorname{Re} \{ P U_r^* \} = \frac{1}{2} \frac{k^2 \rho_0 c |Q|^2}{(4\pi R)^2} \propto \left(\frac{1}{R}\right)^2$$

ii) measured intensity

$$P_m = \frac{P_1 + P_2}{2} \quad U_{m,r} = \frac{1}{w\rho_0\Delta} (P_2 - P_1)$$

$$I_{m,r} = \frac{1}{2} \operatorname{Re} \{ P_m U_{m,r}^* \} = \frac{1}{2} \frac{k^2 \rho_0 c |Q|^2}{(4\pi)^2 R_1 R_2} \left( \frac{\sin(k\Delta)}{k\Delta} \right)$$

2 finite difference  
error

$$\frac{I_{\text{rec}}}{I_{\text{tr},r}} = \frac{R^2}{(R_1 R_2)} \frac{\sin k\Delta}{k\Delta} \quad R_1 = R - \frac{\Delta}{2} \dots R_2 = R + \frac{\Delta}{2}$$

$$= \frac{1}{1 - \left(\frac{\Delta^2}{4R^2}\right)} \frac{\sin k\Delta}{k\Delta}$$

$\nearrow$  near field error (always  $> 1$ )

in a good measurement

$\frac{\Delta^2}{4R^2} \ll 1 \dots$  can approximate as

$$\frac{I_{\text{rec}}}{I_{\text{tr},r}} = \left(1 + \frac{\Delta^2}{4R^2}\right) \frac{\sin k\Delta}{k\Delta}$$

- depends on distance from the source
  - large errors close to the source
  - not frequency dependent
  - depends on wave front curvature
- depends on source type
- increases with source complexity

$R > 1.1\Delta$  for  $\text{nfe} < 1 \text{ dB}$

for dipole ...  $R > 1.6\Delta$  for  $\text{nfe} < 1 \text{ dB}$

for lat. quadrupole ...  $R > 2.3\Delta$  for  $\text{nfe} < 1 \text{ dB}$

for  $\Delta = 50 \text{ mm}$   $R > 115 \text{ mm}$

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$$\vec{x} \quad \begin{matrix} \square \\ P_1 \end{matrix} \quad \begin{matrix} \square \\ P_2 \end{matrix} \quad I_x = \frac{\text{Im}\{P_1 P_2^*\}}{2\omega \rho_0 \Delta}$$

$$\frac{I_{nx}}{I_{bx}} = \underbrace{\left(1 + \frac{\Delta^2}{4R^2}\right)}_{\text{purely geometric}} \frac{\sin k\Delta}{k\Delta} \quad \boxed{\text{nearfield errors}}$$

## More complex sources

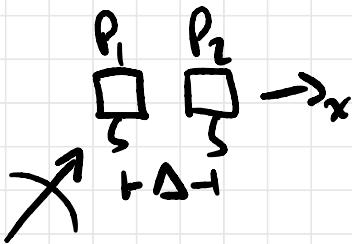
+	-	+	-
-	+	-	+
+	-	+	-
-	+	-	+

vibrating pure

- (much more complicated than quadrupole)

- exercise caution when measuring close to real sources

### 1.5.3 Phase MisMatch Errors



i) Estimate Intensity in the absence of Phase & Amplitude errors

$$P_1 = |P_1| e^{j\phi_1} \quad P_2 = |P_2| e^{j\phi_2}$$

$$\begin{aligned} I_{\text{int},x} &= \frac{\text{Im}\{P_1 P_2^*\}}{2w\rho_0\Delta} = \frac{|P_1||P_2|}{2w\rho_0\Delta} \text{Im}\{e^{j(\phi_1-\phi_2)}\} \\ &= \frac{|P_1||P_2|}{2w\rho_0\Delta} \sin[\delta\phi] \sim \phi_1 - \phi_2 \end{aligned}$$

Intensity is very sensitive to the phase of the sound field.

ii) Estimate intensity in the presence of phase & amplitude errors

$$P_{1c} = a_1 |P_1| e^{j(\phi_1 + \phi_{1c})}$$

$$P_{2c} = a_2 |P_2| e^{j(\phi_2 + \phi_{2c})}$$

$a_1, a_2 \dots$  amplitude error  
 $\phi_{1c}, \phi_{2c} \dots$  phase errors

- MIC characteristics
- Measurement electronics

use these expressions to calculate intensity

$$I_{\text{max}} = \frac{\text{Im}\{\mathbf{P}_1 \mathbf{P}_2^*\}}{2\omega \rho_0 \Delta} = a_1 a_2 |\mathbf{P}_1| |\mathbf{P}_2| \sin(\delta\phi + \delta\alpha)$$

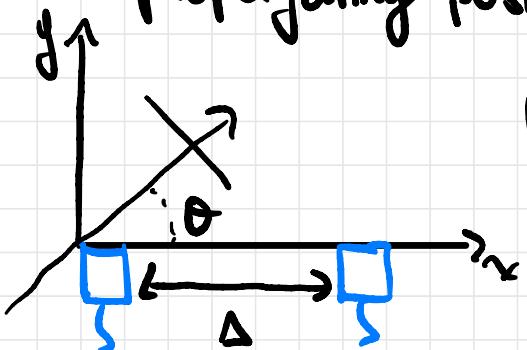
$$\delta\phi = \phi_1 - \phi_2 \quad \delta\alpha = \phi_{1c} - \phi_{2c} \rightarrow \text{phase mismatch}$$

Error  $\frac{I_{\text{max}}}{I_{\text{Intx}}} = a_1 a_2 \frac{\sin(\delta\phi + \delta\alpha)}{\sin(\delta\phi)}$

$a_1, a_2 \rightarrow$  amplitude calibration is easy...

$$a_1 + a_2 \approx 1$$

iii) consider  $\delta\phi$  - plane wave in free space propagating post the MICS



$$\rho \propto e^{-jk_x x} e^{-jk_y y}$$

$$k_x = k \cos\theta$$

$$k_y = k \sin\theta$$

$\phi_1 = 0$  since MICS is at the origin

$$\phi_2 = -k_x \Delta = -k \Delta \cos\theta = -\frac{\omega}{c} \Delta \cos\theta$$

e.g. 100 Hz,  $\Delta = 50 \text{ mm}$ ,  $c = 340 \text{ m/s}$

$$\phi_2 = 0.1 \cos\theta = 6^\circ \cos\theta$$

$\Omega = 0 \quad \Phi_2 = \frac{w}{c} \dots$  largest possible phase diff.

$\Omega = \pi/2 \quad \Phi_2 = 0 \dots$  minimum phase difference

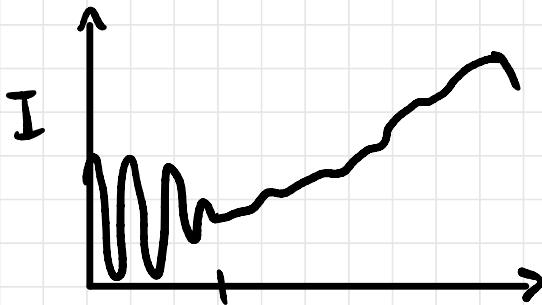
Actual phase difference depends on the nature of the sound field.

#### iv) Error

if  $|\delta\phi + \delta\alpha| \ll 1$  (true in a good measurement)

$$\frac{I_{\text{measured}}}{I_{\text{true}}} = \frac{\delta\phi + \delta\alpha}{\delta\phi} = 1 + \frac{\delta\alpha}{\delta\phi} \underset{\text{assume approx constant}}{\sim}$$

{ continuously decreases as frequency decreases  
+ depends on the sound field}



low frequency region where  $\delta\phi$  is small  
result is dominated by  $\delta\alpha \dots$

Error always becomes large at low frequencies  
or when the actual phase diff is small

Phase mismatch provides low frequency limit to measurements

Lower Limit: Phase Diff

Upper Limit: Finite Difference Error

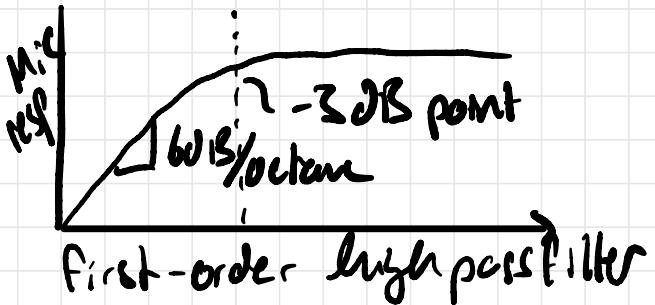
- Space Mics per apert
- Precision Microphones
- Phase Calibration

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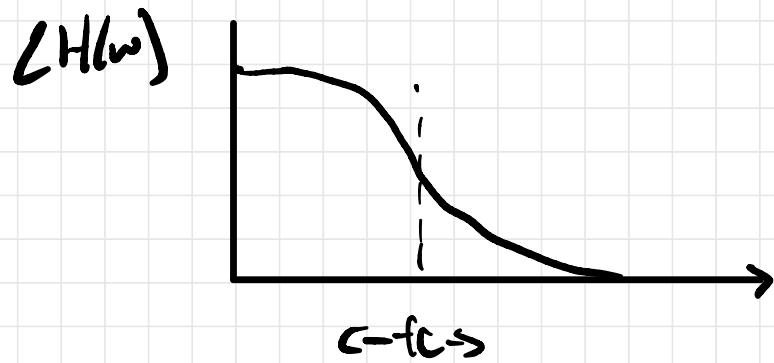
## v) Precision Microphones

~ stainless steel membrane  
conducting back plate

housing  
quartz isolator  
pressure equalization

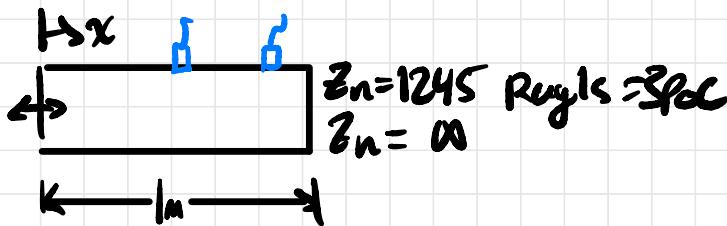


First-order high pass filter



Significant phase difference extending well beyond  $f_c$  because of slow phase change

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PS. 1

$$P = A e^{-j k x} + B e^{j k x}$$

$$U = \frac{A}{Z_n} e^{-j k x} - \frac{B}{Z_n} e^{j k x}$$

BC's at  $x=0$      $U(0) = 0.001 \text{ m/s}$

at  $x=l_m$      $U(l_m) = 0 \text{ (0)}$

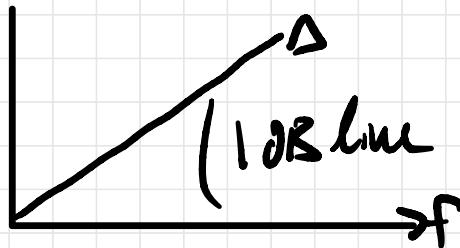
or  $\frac{U(l_m)}{U(0)} = Z_n \text{ (finite impedance)}$

$$I = \frac{1}{2} \operatorname{Re} \{ P U^* \}$$

$$\frac{\operatorname{Im} \{ P_1 P_2^* \}}{2 \omega \rho_0 \Delta} \leftarrow \begin{matrix} \text{Two-mic} \\ \text{intensity...} \end{matrix}$$

$$MSP = \frac{P_p^*}{2} \quad MSU = \frac{U U^*}{2}$$

plot  $10 \log \left( \frac{I_m}{I_t} \right)$   
 finite impedance case  
 only ...



$$\text{True Radial Intensity} = \frac{1}{2} \operatorname{Re}\{P^* u_r\}$$

$$u_r = -\frac{1}{jw\rho_0} \frac{\partial P}{\partial r}$$

- Phase mismatch error - low freq limit to an accurate measurement.

## Calibration Procedure

- can use less expensive microphones

### b) Switching Procedure



Take Geometric Mean of the Cross Spectra  
(Noise Source Must be Stationary)

## Problems

- lack of stationarity

- Twice the number of measurements

### c) Perfectly Matched Mics + Instrumentation

- easy approach

- expensive approach

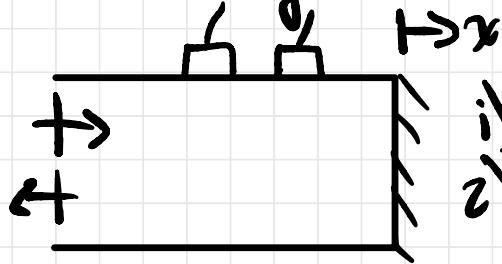
} still necessary to calibrate to ensure an accurate result

FDG: High freq limit

Phase Mismatch: Low freq Limit

NFG: How Close to source

## 1.5.4 Reactivity Error



- i) hard (rigid) termination
  - ii) perfectly absorbing
- $$Z_n = \rho_0 C$$

$$\rho(x) = A e^{-jkx} + B e^{jkx}$$

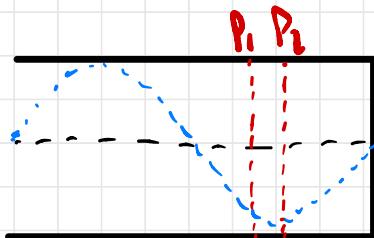
$$u(x) = \frac{A}{\rho_0 C} e^{-jkx} - \frac{B}{\rho_0 C} e^{jkx}$$

i) hard termination

$$u(x=0) = 0 \quad (\text{at termination})$$

so  $B=A \dots$  perfect reflection

$$\begin{aligned} \rho(x) &= A (e^{-jkx} + e^{jkx}) \\ &= 2A \cos kx \end{aligned}$$



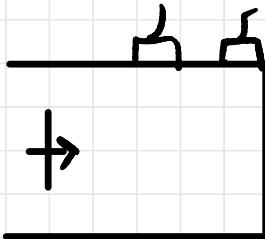
Pure Standing Wave

$$\begin{aligned} I &\propto \sin(\phi_1 - \phi_2) \\ \Rightarrow I &= 0 \end{aligned}$$

Phase difference in this case is either 0 or 180 because of a pure standing wave

ii) Perfectly Absorbing Case (PAC termination)  
no reflection...

$$P(x) = A e^{-kx}$$



$$\delta\phi = kx_1 - kx_2 = k\Delta$$

Phase difference  $\phi_1 - \phi_2 = \delta\phi$   
depends on the environment

$$Im = \frac{Im \{ P_1 P_2^* \}}{2wP_0\Delta} = \frac{|P_1||P_2|}{2wP_0\Delta} \sin(\delta\theta + \delta\alpha)$$

S  
phase mismatch

- Sound field itself affects the measurement accuracy
- difficult to make a good measurement when  $\delta\phi$  is small

Reactivity Index

Reactivity is the degree to which the sound field consists of non-propagating components

Residual Intensity

$$L_I = 10 \log_{10} \left( \frac{I}{I_{ref}} \right) \quad \text{if } I = 0 \Rightarrow L_I = -\infty$$

but  $L_p = \text{something}$  ← hard termination ↑

Absorbing termination

$$L_p = L_I$$

Residual Intensity,  $R_I = L_p - L_I$

for hard termination ...  $R_I = \infty$  bad

for absorbing termination ...  $R_I = 0$  good

Rule of Thumb ...  $R_I > 15 \text{ dB}$

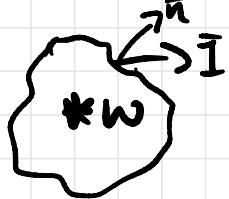
it is difficult to make good measurement

Example of a "Quality Indicator"  
in ASTM/ANSI

Conclusion - Check Quality Indicators to  
ensure a valid and defensible  
measurement.

# 1.6 Applications of Acoustic Intensity Method

## 1.6.1 Sound Power Measurement



$$W = \int_S \vec{I} \cdot \vec{n} dS$$

$\sim$  normal component of intensity

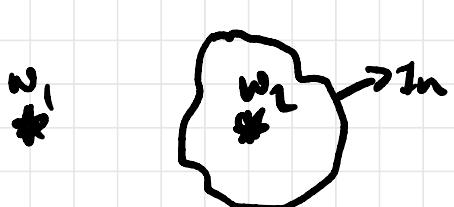
$$= \int_S I_n dS$$

Sound Power,  $W$   
when surface completely encloses the source

if Surface does not enclose

$$W \rightarrow \Rightarrow W = \int_S I_n dS = 0$$

if the source is exterior



$$\int_S I_n dS = W_2$$

$\star w_1 \rightarrow I_n$



$$\int_S I_n dS = w_1$$

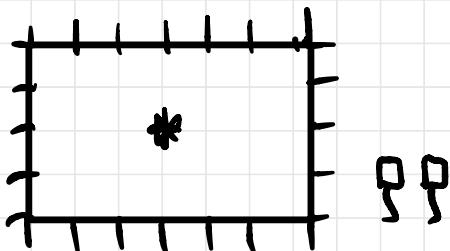
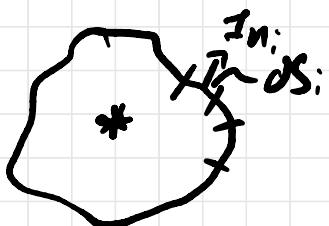
$w_2$

- Can measure sound power even in the presence of reflections
- Can measure sound power even in the presence of other noise sources



i) point wise measurement

$$w = \int_S I_n dS = \sum_{i=1}^n I_{n_i} dS_i$$



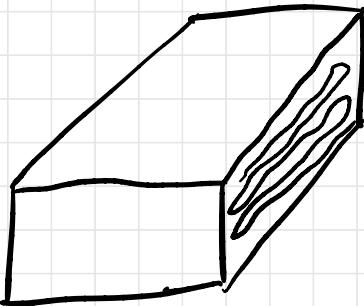
Check accuracy by progressively increasing number of measurements until convergence is achieved

Pro - accuracy

Con - time-consuming

## ii) Scanning Approach

hold probe normal to surface and continually average the cross-spectra



Pro - faster

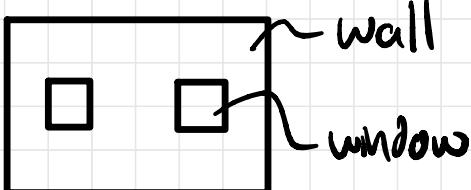
Con - Difficult to quantify accuracy

## 1.6.2 Source Ranking & Identification



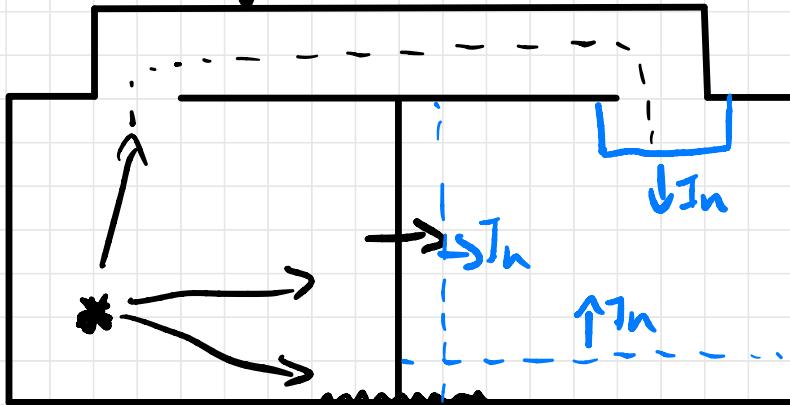
form surfaces that enclose individual sources of a composite source

- component sound
- guide noise control



transportation  
-planes

### 1.6.3 Ranking of Noise Paths

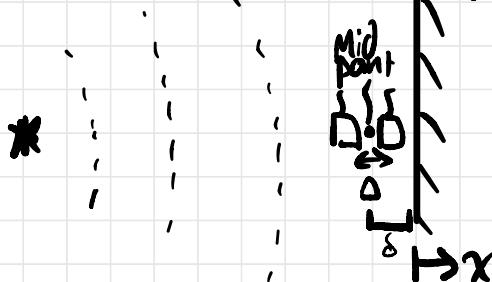


- measure sound power transmitting through various paths...

- to guide noise control application

### 1.6.4 Measurement of Impedance and absorption

$$Z_n = \frac{p(0)}{u(0)}$$



Assume Sound field is planar near the surface

$$Z_n(\delta) = \frac{p(\delta)}{u(\delta)} \quad \delta \text{ is negative ...}$$

$$\rho(x) = A e^{-j k x} + B e^{j k x}$$

$$u(x) = \frac{A}{\rho_0 c} e^{-j k x} - \frac{B}{\rho_0 c} e^{j k x}$$

$$\rho(\delta) = A e^{-j k \delta} + B e^{j k \delta}$$

$u(\delta)$  ...

$$Z_n(\delta) = \frac{(A e^{-j k \delta} + B e^{j k \delta})}{(A e^{-j k \delta} - B e^{j k \delta})} \cdot \rho_0 c \cdot \frac{\frac{1}{A}}{\frac{1}{A}}$$

$$Z_n(\delta) = \frac{e^{-j k \delta} + R e^{j k \delta}}{e^{-j k \delta} - R e^{j k \delta}} \cdot \rho_0 c$$

$$R = \frac{Z_n(0) - \rho_0 c}{Z_n(0) + \rho_0 c}$$

Algebra ...

$$Z_n(0) = \rho_0 c Z_n(\delta) + j \rho_0 c \tan k \delta$$
$$\frac{\rho_0 c + j Z_n(\delta) \tan k \delta}{\rho_0 c + j Z_n(\delta) \tan k \delta}$$

Acoustic impedance transfer formula

Procedure - Measure  $\rho(\delta)$  +  $u(\delta)$

using two-mic probe ... the transfer

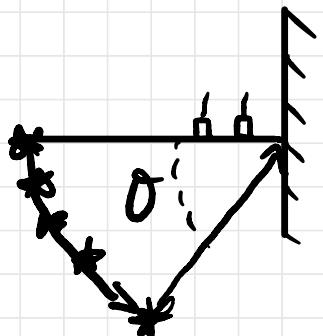
$$Z_n(\delta) \rightarrow Z_n(0)$$

Then find reflection coefficient by

$$R = \frac{Z_n(0) - P_{oc}}{Z_n(0) + P_{oc}} \Rightarrow \alpha = 1 - |R|^2$$

... and find absorption...

Allows in situ measurement

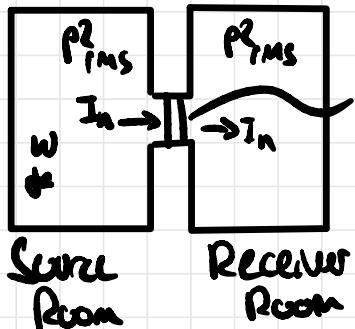


$Z_n(\theta)$

if  $Z_n$  is a function  
of angle of incidence  
(or very angle of source from  
sample)

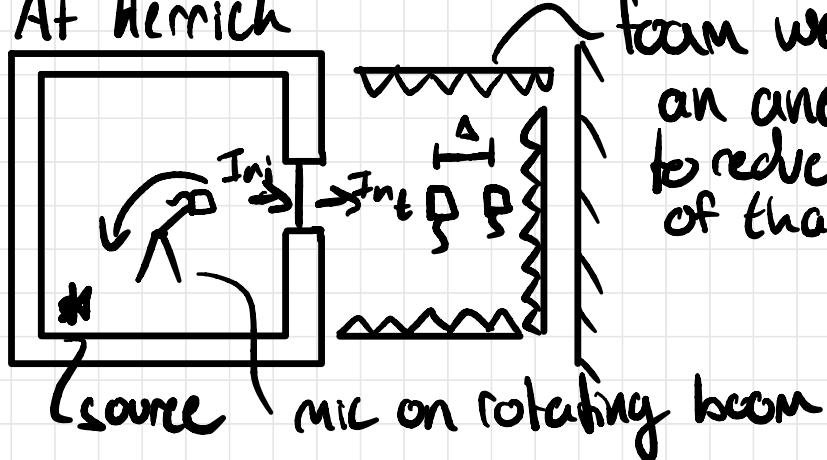
if  $\theta \leq 60^\circ$ , this simple approach works

## 1.6.5 Measurement of transmission loss



barrier under test

At Herrich



foam wedges create  
an anechoic space  
to reduce reactivity  
of that field

(source) mic on rotating beam

Incident Sound Power

$$W_i = \frac{P_{rms}^2}{4\rho c} \cdot Sp \sim \text{panel area}$$

Transmitted Sound Power

25 intensity measurements  
to give transmitted sound

power  $\overbrace{\text{In}_t Sp}$  space averaged transmitted intensity

$$W_t = \overline{I_{n_t}} Sp$$

x	x	x	x	x
x	-	-	-	-
x	-	-	-	-
x	-	-	-	-

5x5  
array

Power Transmission Coefficient

$$T\pi = \frac{\overline{I_{n_t}} Sp}{\frac{P_{rms}^2 Sp}{4\rho c}} \Rightarrow T_L = 10 \log_{10} \left( \frac{1}{T\pi} \right)$$

# Quick Summary -

-Acoustic Intensity commonly means  
Time-Averaged Acoustic Intensity

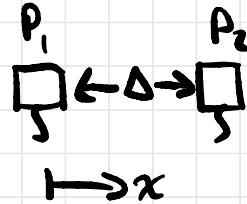
$$I = \frac{1}{2} \operatorname{Re}\{\rho \bar{u}^*\} \Rightarrow I_x = \frac{1}{2} \operatorname{Re}\{\rho u_x^*\}$$

vector quantity

## -Particle Velocity

- direct surface measurements
- ultrasonic procedure
- Holographic procedure
- two-microphone cross-spectral approach
  - finite difference approximation to pressure gradient

$$I_x = \frac{\operatorname{Im}\{P_1 P_2^*\}}{2 \omega \rho_0 \Delta}$$



## -Sources of Error in 2 MIC Method

- finite difference error
- high frequency limit given fixed  $\Delta$

$$\Delta < \frac{1}{10} \lambda$$

- nearfield error
  - limits how close you can be to the source
- Phase Mismatch
  - low frequency limit
  - can be corrected by careful calibration
- Reactivity Error
  - Residual Intensity
$$R_I = L_p - L_I$$

## Applications

- Sound Power