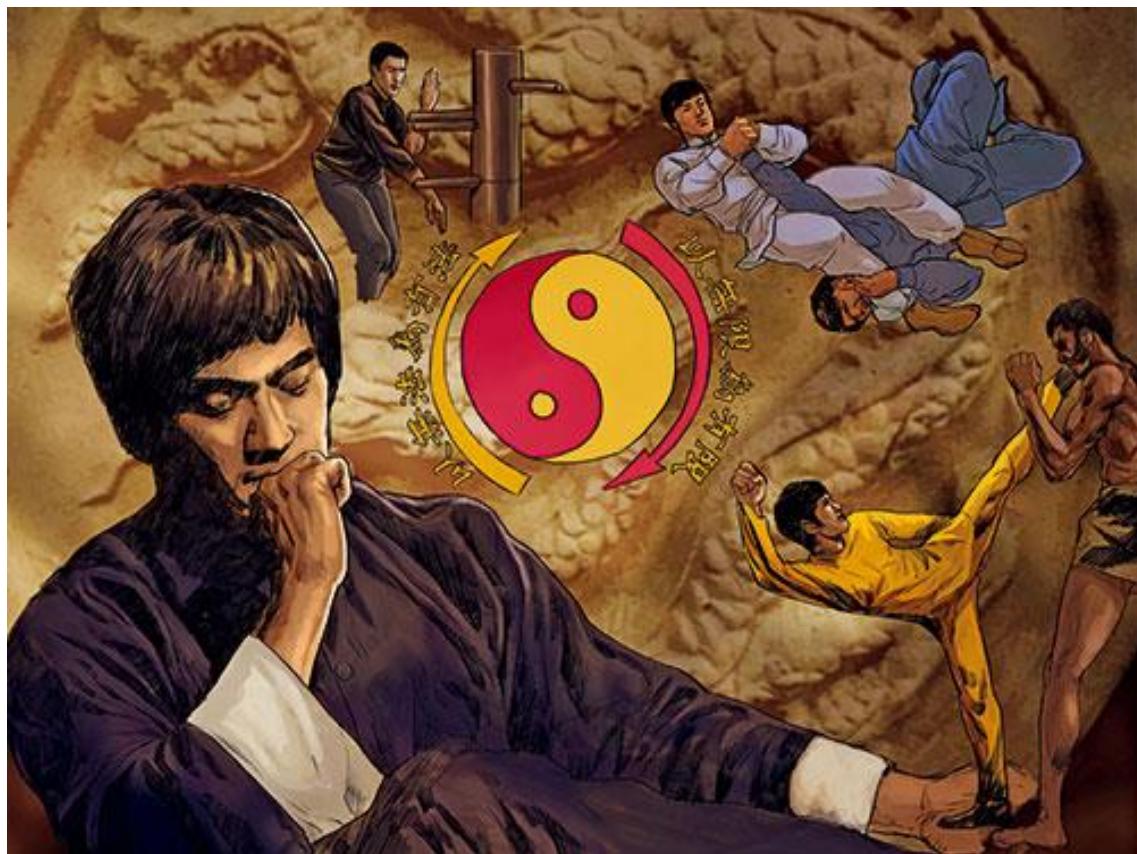


Lecture 25 Final Project



Good at
both Mathematical analysis and Computer programming

**Final Class Examination will be held on
Tuesday (May 9th, 8:00am-11:00am) at
Hearst Mining Building 390**

RR Week Office Hour Schedule

Prof. Li: Tu & Thu. 1:00pm – 3:00pm

GSIs:

Qingsong Tu: MWF 11:00 am – 12:30 pm

Jes Parker: Tu & Thus. 11:00am-12:30 pm

Ben Worsfold Butler: MWF 9:00 am – 10:30 am

Final exam is 25% of overall grade.

It is a competition !

1 A+ (95)

3 A (90)

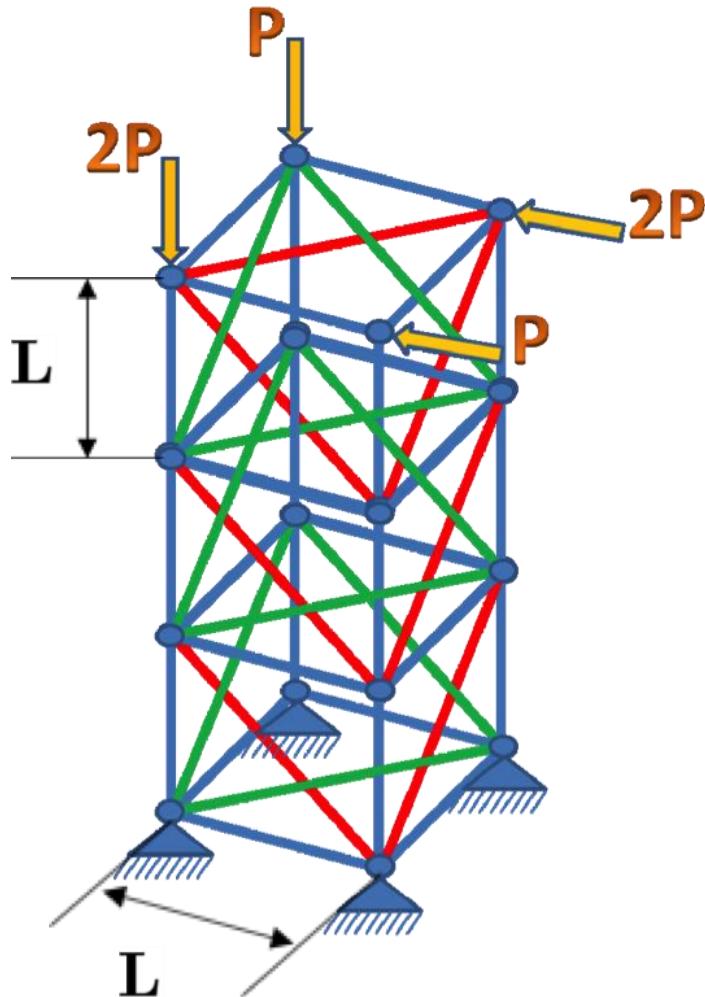
4 A- (85)

4 B+ (80)



It is a research, presentation, and oral examination.

Final Project I. Part I



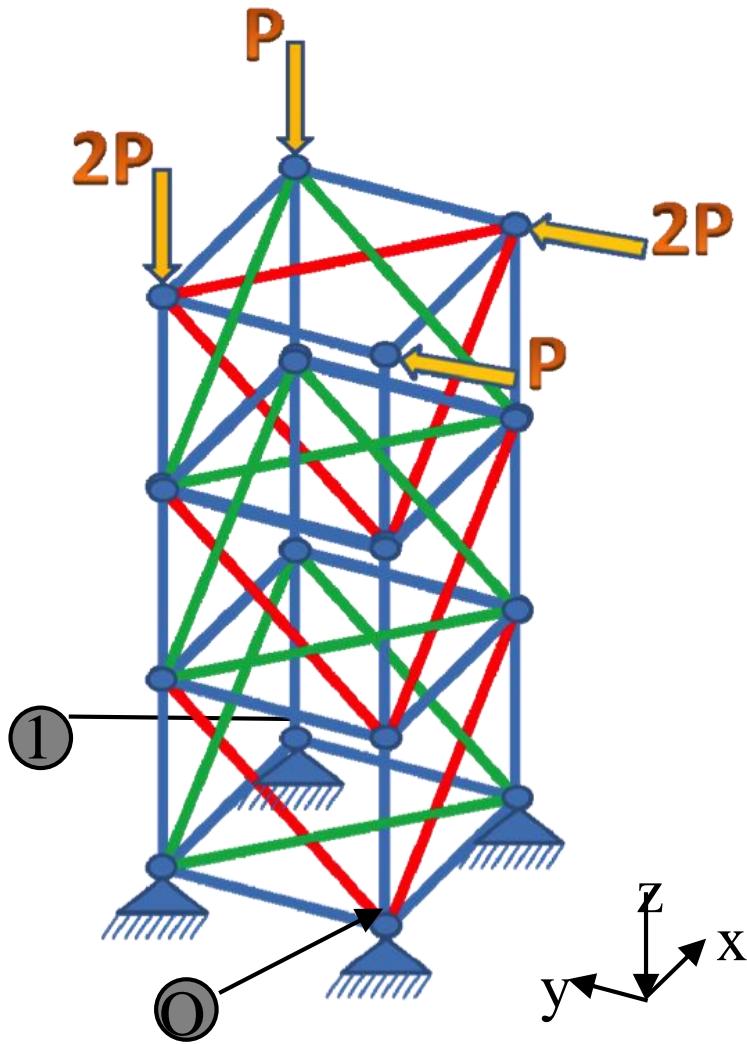
Use matrix analysis to find both the nodal displacements as well as the internal force for the 3D truss in the figure, with the following given values:

$$E = 400 \text{ MPa}$$

$$L = 2.0 \text{ m}$$

$$A = 5 \times 10^{-4} \text{ m}^2$$

$$P = 1000 \text{ N}$$



There are 16 nodes.

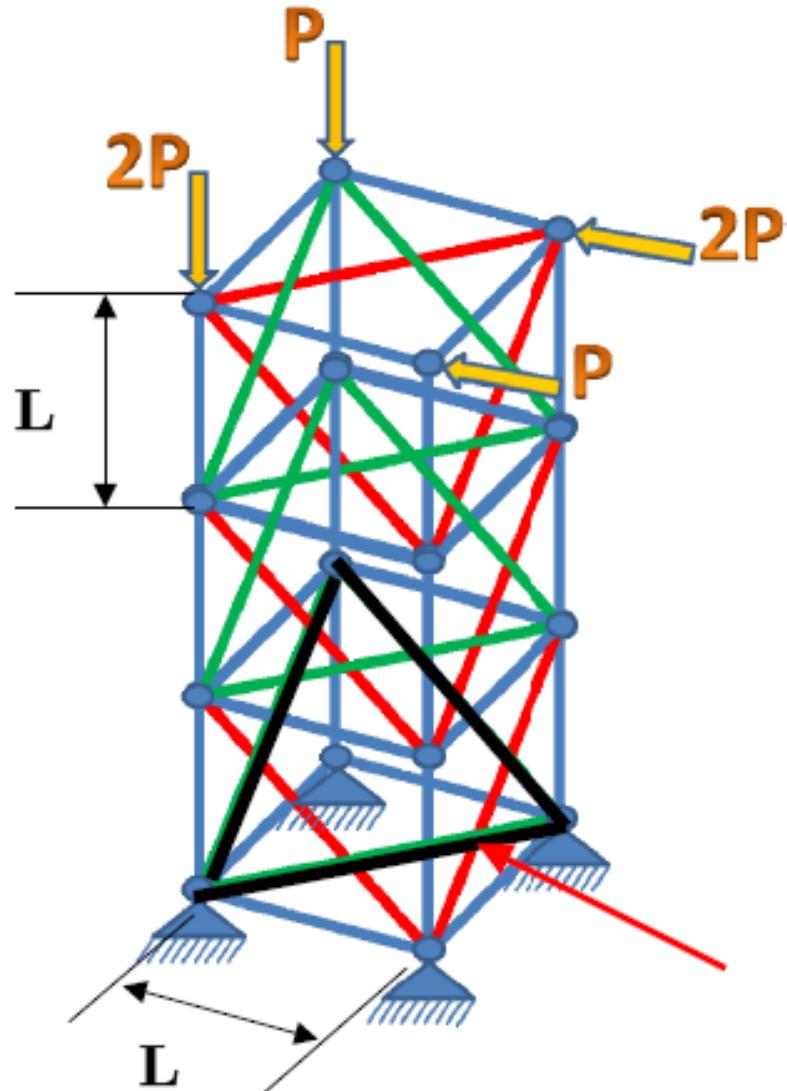
- 1st node begins at the bottom back corner
- 16th node ends at the downward $2P$ force at top

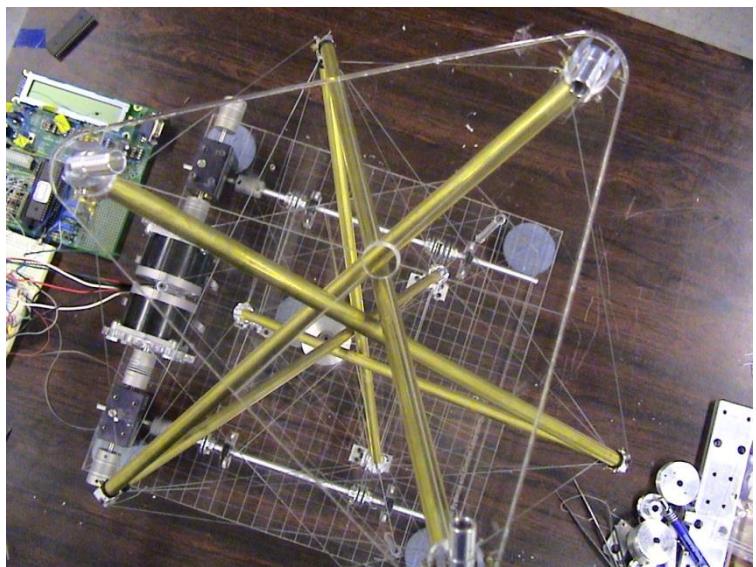
There are 44 members

- 16 horizontal bars 1-16
- 12 vertical bars 17-28
- 9 green bars 29-37
- 7 red bars 38-44

Part II

- Replace lower green triangle (bar 29, 30, 31) with one rope with the same E and A
- Assume rope and nodes are frictionless, so the rope has a uniform tension throughout





Pre-stressed tensegrity structure
as novel meta-materials

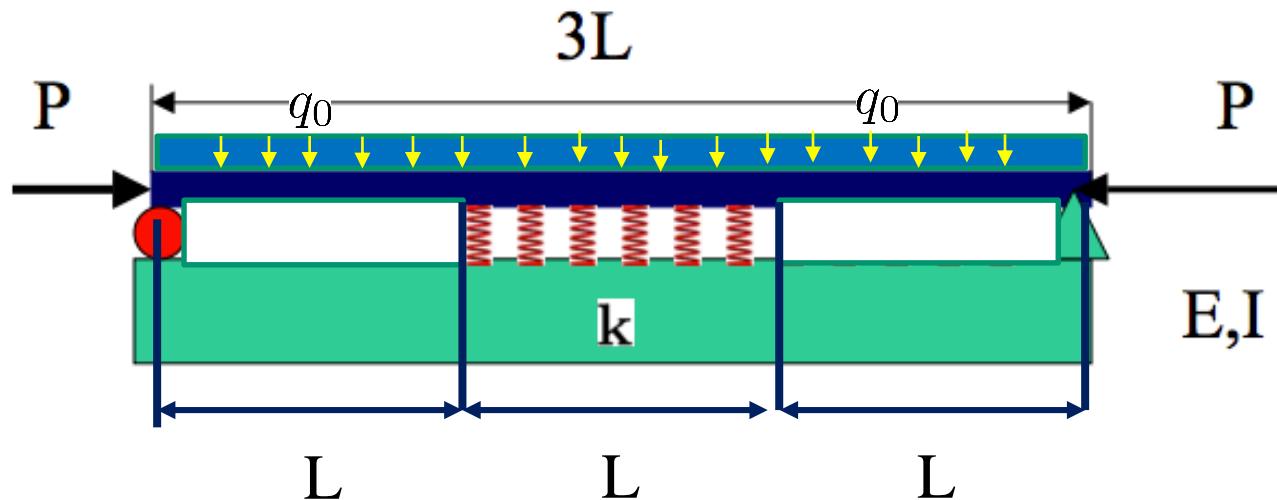
Tensegrity structures with active damping control

Final Project No. II

Buckling of a Beam on a Partial Winkler Foundation

Considering the following structure with $L = 2M$,
 $EI = 100 \text{ NM}^2$, $k = 5000 \text{ N/M}^2$.

The beam is subjected a uniform load of $q_0 = 1000 \text{ N/M}$.
Use the beam element to find the critical load.

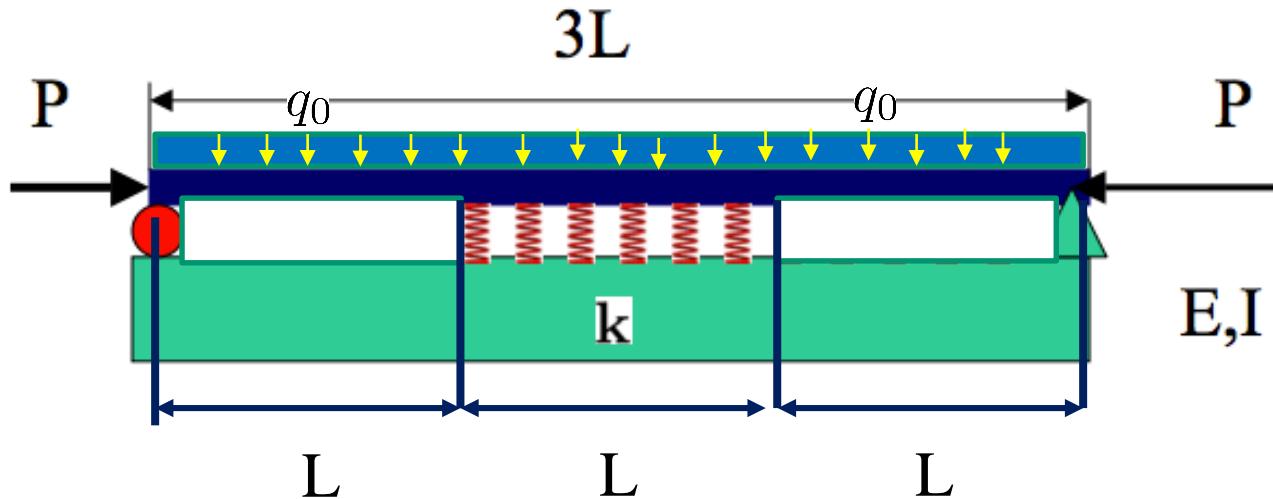


Final Project Topic No. II

Buckling of a Beam on a Partial Winkler Foundation

Extra Points

Find the critical load when $q = 1000 \sin\left(\frac{\pi x}{3L}\right) N/M.$



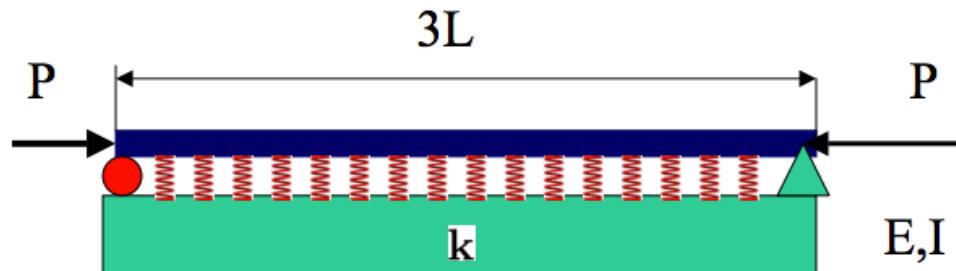
(CE-130N-2013) Matlab Code Extract: solving for Pcrit

modified boundary conditions:

$$idf = [3:2*n-2 2*n];$$

$$idd = [1 2 2*n-1];$$

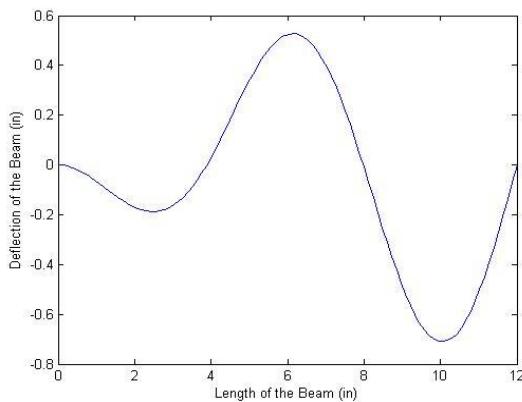
$$c(idd) = [0 0 0];$$



solving for critical buckling load:

$$[V,D] = \text{eig}(Kff,Gff);$$

$$P_{\text{crit}} = D(1,1);$$



To solve the problem, k_{sp} is modified as follows:

$$k_{sp} = \int g_i k g_j dx$$

$$Ksp = k.* [13*dx/35 \quad (11*dx^2)/210 \quad 9*dx/70 \quad -13*dx^2/420; \quad (11*dx^2)/210 \quad dx^3/105 \quad 13*dx^2/420 \quad -dx^3/140; \quad 9*dx/70 \quad 13*dx^2/420 \quad 13*dx/35 \quad -11*dx^2/210; \quad -13*dx^2/420 \quad -dx^3/140 \quad -11*dx^2/210 \quad dx^3/105];$$

$$\mathbf{P}_{\text{crit}} = 8988\mathbf{N}$$

Final Project

Mathematical model:

$$(\blacksquare) \nabla^2 w + (\blacktriangle) = 0 .$$

- Anti-plane problem: $\blacksquare = Gt$ and $\blacktriangle = bt \rightarrow Gt \nabla^2 w + bt = 0.$
- Thin membrane: $\blacksquare = S$ and $\blacktriangle = p \rightarrow S \nabla^2 w + p = 0.$
- St. Venant Torsion: $\blacksquare = 1$ and $\blacktriangle = 0 \rightarrow \nabla^2 \psi = 0.$
- Prantl's membrane analogy: $\blacksquare = 1$ and $\blacktriangle = 2G\theta \rightarrow \nabla^2 \varphi + 2G\theta = 0.$
- Heat conduction problem: $\blacksquare = K$ and $\blacktriangle = Q \rightarrow K \nabla^2 T + Q = 0.$

1. Write a MATLAB function of the form

```
function [p,t,e]=pmesh(pv,hmax,nref)
```

which generates an unstructured triangular mesh of the polygon with vertices `pv`, with edge lengths approximately equal to $h_{\max}/2^{n_{\text{ref}}}$, using a simplified Delaunay refinement algorithm. The outputs are the node points `p` (N -by-2), the triangle indices `t` (T -by-3), and the indices of the boundary points `e`.

- (a) The 2-column matrix `pv` contains the vertices x_i, y_i of the original polygon, with the last point equal to the first (a closed polygon).
- (b) First, create node points along each polygon segment, such that all new segments have lengths $\leq h_{\max}$ (but as close to h_{\max} as possible). Make sure not to duplicate any nodes.
- (c) Triangulate the domain using the `delaunayn` command.
- (d) Remove the triangles outside the domain (see the `inpolygon` command).
- (e) Find the triangle with largest area A . If $A > h_{\max}^2/2$, add the circumcenter of the triangle to the list of node points.
- (f) Retriangulate and remove outside triangles (steps (c)-(d)).
- (g) Repeat steps (e)-(f) until no triangle area $A > h_{\max}^2/2$.
- (h) Refine the mesh uniformly n_{ref} times. In each refinement, add the center of each mesh edge to the list of node points, and retriangulate. Again, make sure not to duplicate any nodes, see e.g. the command `unique(p,'rows')`.

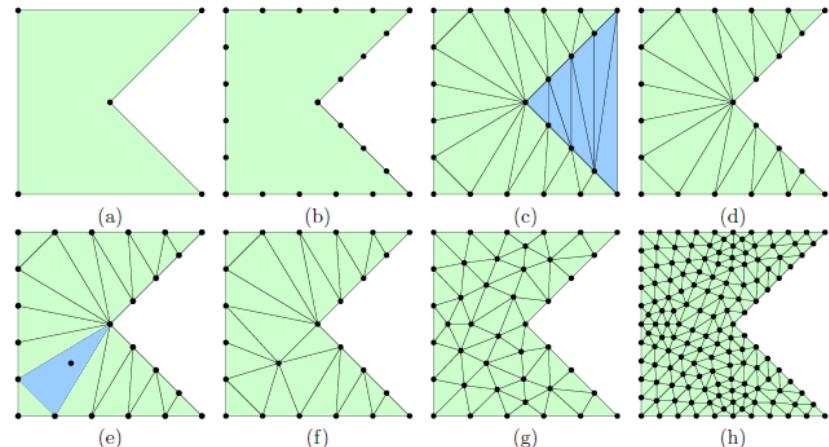
Finally, find the nodes `e` on the boundary using the `boundary_nodes` command. The following commands create the example in the figures. Also make sure that the function works with other inputs, that is, other polygons, h_{\max} , and n_{ref} .

```
pv=[0,0;1,0;.5,.5;1,1;0,1;0,0];
[p,t,e]=pmesh(pv,0.2,1);
tplot(p,t)
```

p is the array for nodal coordinates

t is connectivity array

e is the boundary node array



Name	Search boundary nodes	Date modified	Type	Size
 boundary_nodes.m		11/9/2011 12:28 PM	MATLAB Code	1 KB
 fempoi.m	FEM calculation	11/17/2011 3:46 PM	MATLAB Code	2 KB
 fempoi_test.m	main program	1/24/2017 1:53 PM	MATLAB Code	1 KB
 pmesh.m	generate mesh	11/17/2011 9:07 A...	MATLAB Code	4 KB
 poiconv.m		11/17/2011 9:06 A...	MATLAB Code	1 KB
 poiconv_test.m	Calculate error	11/16/2011 9:11 PM	MATLAB Code	1 KB
 ps6.pdf		4/18/2016 3:39 PM	Adobe Acrobat D...	221 KB
 Ref2-Chapter13.pdf		12/22/2011 5:44 PM	Adobe Acrobat D...	4,817 KB
 tplot.m	plot results	11/15/2011 8:59 PM	MATLAB Code	1 KB

```

clear all; close all; clc;
Square, Dirichlet left/bottom
pv=[0,0;1,0;1,1;0,1;0,0];
[p,t,e]=pmesh(pv,0.2,0);
e=e(p(e,1)==0 | p(e,2)==0);
u=fempoi(p,t,e);
figure; tplot(p,t,u)

```

```

% Circle, all Dirichlet
n=32; phi=2.0*pi*(0:n)/n;
pv=[cos(phi),sin(phi)];
[p,t,e]=pmesh(pv,2*pi/n,0);
u=fempoi(p,t,e);
figure; tplot(p,t,u)

```

```

% Ellipse, all Dirichlet
a=2;
b=1;
n=32; phi=2.0*pi*(0:n)/n;
pv=[a*cos(phi),b*sin(phi)];
[p,t,e]=pmesh(pv,2*pi/n,0);
u=fempoi(p,t,e);
figure; tplot(p,t,u)

```

```

% Triangle, all Dirichlet
a= sqrt(2)/2;
pv=[-a,0;a,0;0,1;-a,0];
[p,t,e]=pmesh(pv,0.2,0);
u=fempoi(p,t,e);
figure; tplot(p,t,u)

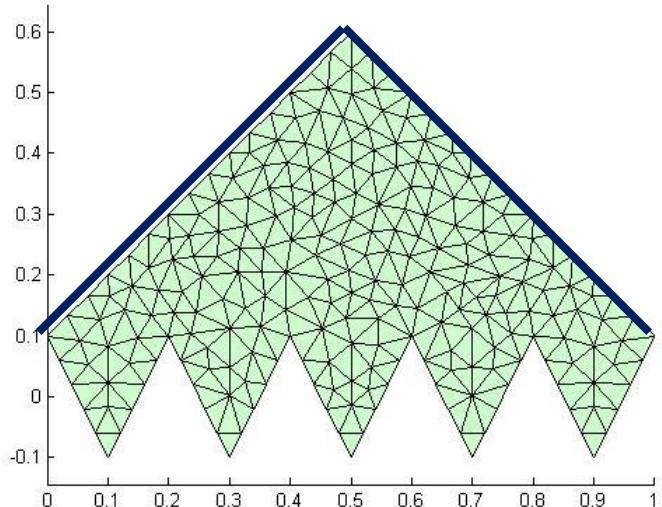
```

```

%Complex polygon geometry, mixed Dirichlet/Neumann
x=(0:.1:1)';
y=.1*cos(10*pi*x);
pv=[x,y; .5,.6;0,0.1];
[p,t,e]=pmesh(pv,0.05,0);
e=e(p(e,2)>=.6-abs(p(e,1)-.5));
u=fempoi(p,t,e);
figure; tplot(p,t,u)

```

fempoi_test.m



**Download Resource Files
from Class website**

Project III. Two-Dimensional Heat Transfer

(100 Points): You are required to solve

$$\frac{\partial}{\partial x} \left(K \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(K \frac{\partial T}{\partial y} \right) + Q = 0 \quad \forall (x, y) \in \Omega; \quad T = T_b \quad \forall (x, y) \in \Gamma_T; \quad \mathbf{n} \cdot \nabla T = q^*, \quad \forall (x, y) \in \Gamma_q$$

where K is the thermal conductivity ($kW/(m \cdot C^\circ)$); Q is the heat source or sink (kW/m^3), and c is the specific heat ($kW \cdot h/kg \cdot C^\circ$).

For steady state problem,

$$\frac{\partial}{\partial x} \left(K \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(K \frac{\partial T}{\partial y} \right) + Q = 0 \quad \forall (x, y) \in \Omega$$

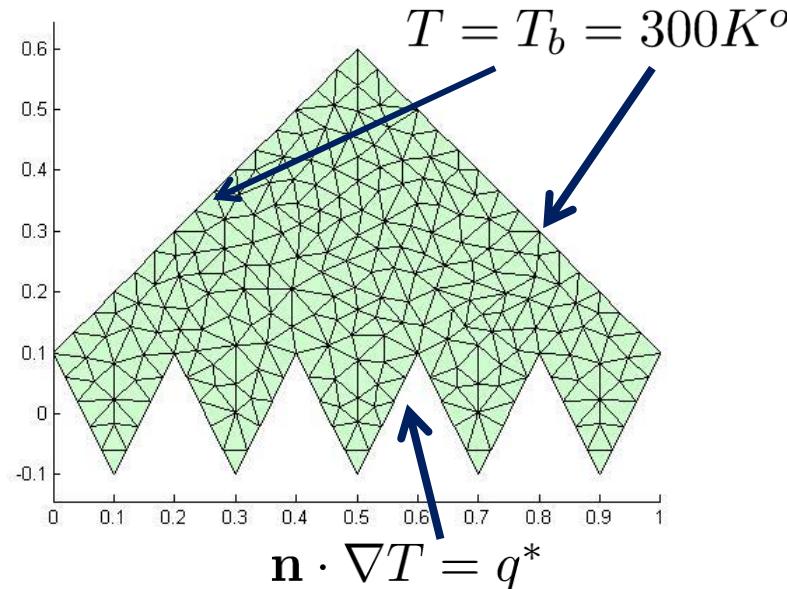
The potential energy is

$$\Pi = \frac{Kt}{2} \int_{\Omega} \left(\left(\frac{\partial T}{\partial x} \right)^2 + \left(\frac{\partial T}{\partial y} \right)^2 \right) d\Omega - t \int_{\Omega} QT d\Omega - \int_{\Gamma_q} q^* T dS$$

The weak formulation is,

$$\delta\Pi = Kt \int_{\Omega} \left(\frac{\partial T}{\partial x} \frac{\partial \delta T}{\partial x} + \frac{\partial T}{\partial y} \frac{\partial \delta T}{\partial y} \right) d\Omega - t \int_{\Omega} Q \delta T d\Omega - \int_{\Gamma_q} q^* \delta T dS = 0$$

In the project, we consider Ω as a circular domain, $K = 1$, $Q = 1$, and $t = 1$; and $q^* = 0.2$.



Extra credits (**5 points**): solving the transient heat transfer problem,

$$\frac{\partial}{\partial x} \left(K \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(K \frac{\partial T}{\partial y} \right) + Q = c\rho \frac{\partial T}{\partial t} \quad \forall (x, y) \in \Omega$$

$\rho = 100$ and $c = 10$.



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Project IV FEM Calculation of Membrane

Use 2D triangle element to simulate membrane deformation (100 Points).

Choose $S = 1$. Solve

$$S \left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) + p(x, y) = 0, \quad \forall (x, y) \in \Omega, \quad \Omega = \left\{ (x, y) \mid \text{an elliptic membrane} \right\}$$

Boundary condition,

$$w(x, y) = 0, \quad \forall (x, y) \in \partial\Omega$$

Choose

$$p(x, y) = q_0, \quad |\mathbf{r}| \leq 1, \quad \text{where } \mathbf{r} = x\mathbf{i} + y\mathbf{j}, \quad \text{and } q_0 = 1.$$

Solve the problem using FEM.

You are required to solve

$$S \left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) + p(x, y) = 0, \quad \forall (x, y) \in \Omega;$$

where S is the constant membrane tension.

The potentail energy is

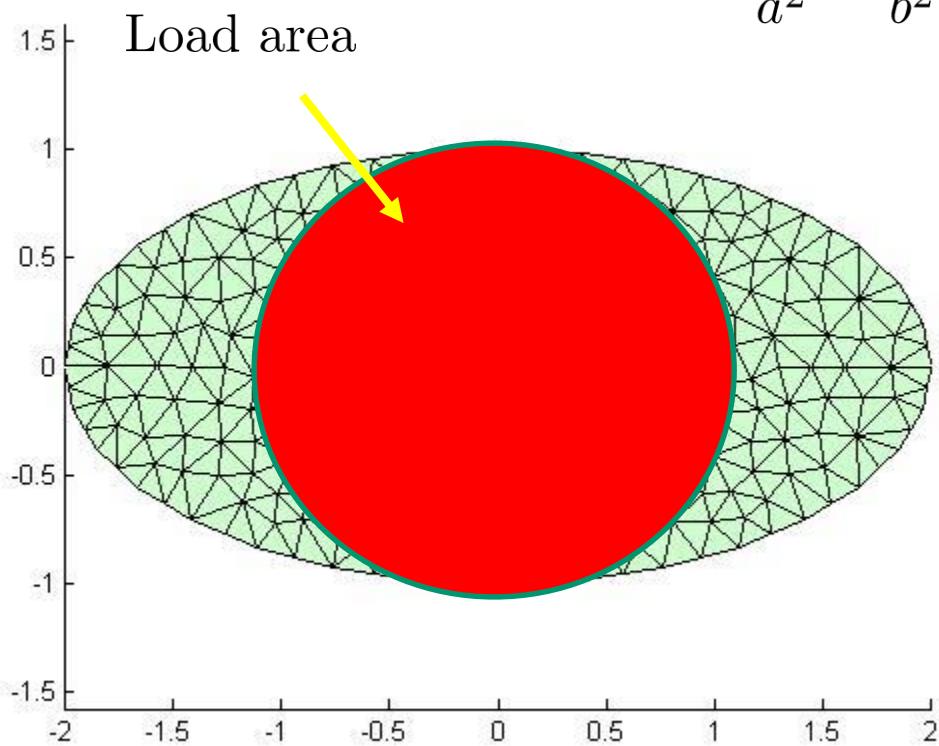
$$\Pi = \frac{S}{2} \int_{\Omega} \left(\left(\frac{\partial w}{\partial x} \right)^2 + \left(\frac{\partial w}{\partial y} \right)^2 \right) d\Omega - \int_{\Omega} q w d\Omega$$

The virtual work principle is

$$S \int_{\Omega} \left(\frac{\partial w}{\partial x} \frac{\partial \delta w}{\partial x} + \frac{\partial w}{\partial y} \frac{\partial \delta w}{\partial y} \right) d\Omega = \int_{\Omega} q \delta w d\Omega$$

In the project, we consider Ω as an equilaterial triangle domain.

$$\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1, \quad a = 2, \quad b = 1$$



A Elliptic Membrane

Extra Points (**5 Points**):

Choose $S = 1$. Solve

$$S \left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) + p(x, y) = \rho \frac{\partial^2 w}{\partial t^2} \quad \forall (x, y) \in \Omega,$$

where $\rho = 10$

$$\Omega = \left\{ (x, y) \mid \text{is an ellipse} \right\}$$

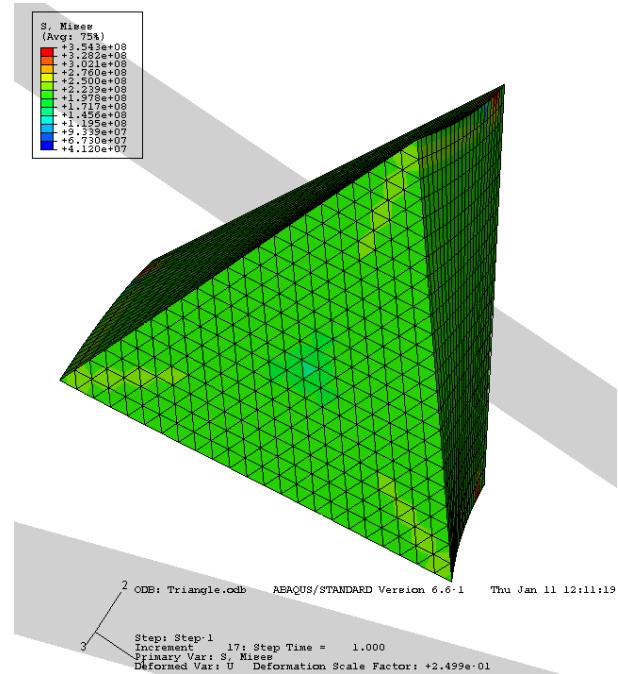
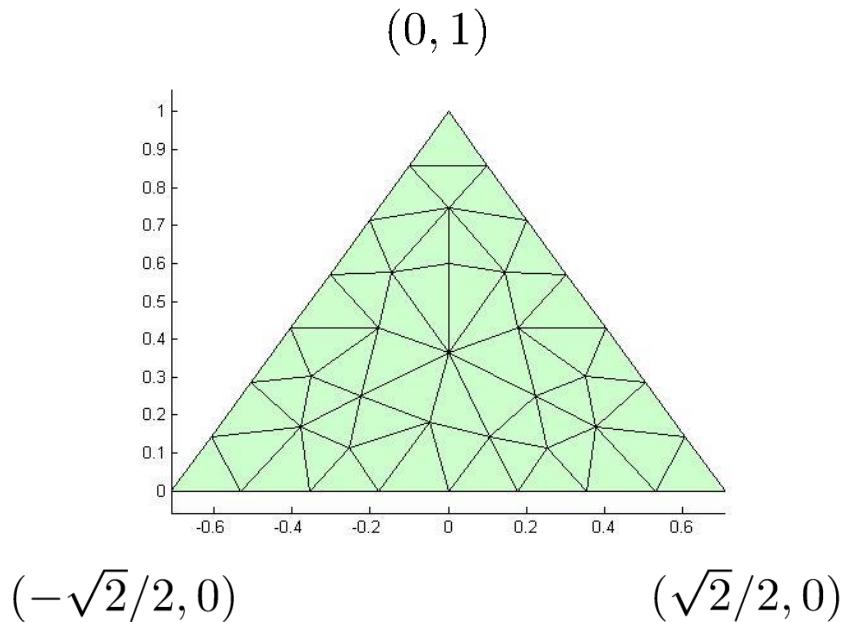
Boundary condition,

$$w(x, y) = 0, \quad \forall (x, y) \in \partial\Omega$$

Solve the problem using FEM.

Project V.

FEM Calculation of Warping (Saint-Venant Torsion)



105 Points: Solving for ψ , σ_{xz} , σ_{yz} and T ; plot for ψ ;

Summary of the St. Venant's Theory of Torsion

$$u = -\theta zy, \quad v = \theta zx, \quad w = \theta \psi$$

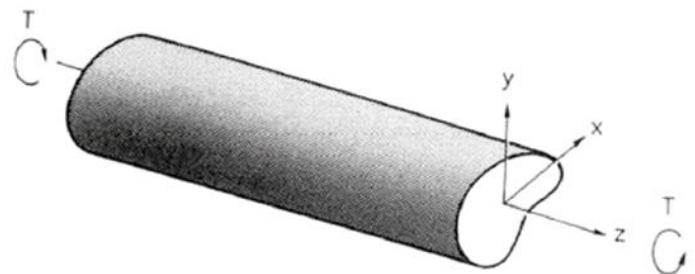
$$\sigma_{xz} = \theta G \left(-y + \frac{\partial \psi}{\partial x} \right),$$

$$\sigma_{yz} = \theta G \left(x + \frac{\partial \psi}{\partial y} \right)$$

$$\frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0, \quad \forall (x, y) \in \Omega$$

$$\frac{\partial \psi}{\partial n} = (y n_x - x n_y), \quad \forall (x, y) \in \partial \Omega$$

$$\theta = 0.2 \text{ rad}, \quad G = 50 \text{ MPa}$$



$$\Pi = \frac{G\theta^2}{2} \int_{\Omega} \left[\left(\frac{\partial \psi}{\partial x} - y \right)^2 + \left(\frac{\partial \psi}{\partial y} + x \right)^2 \right] d\Omega - T\theta$$

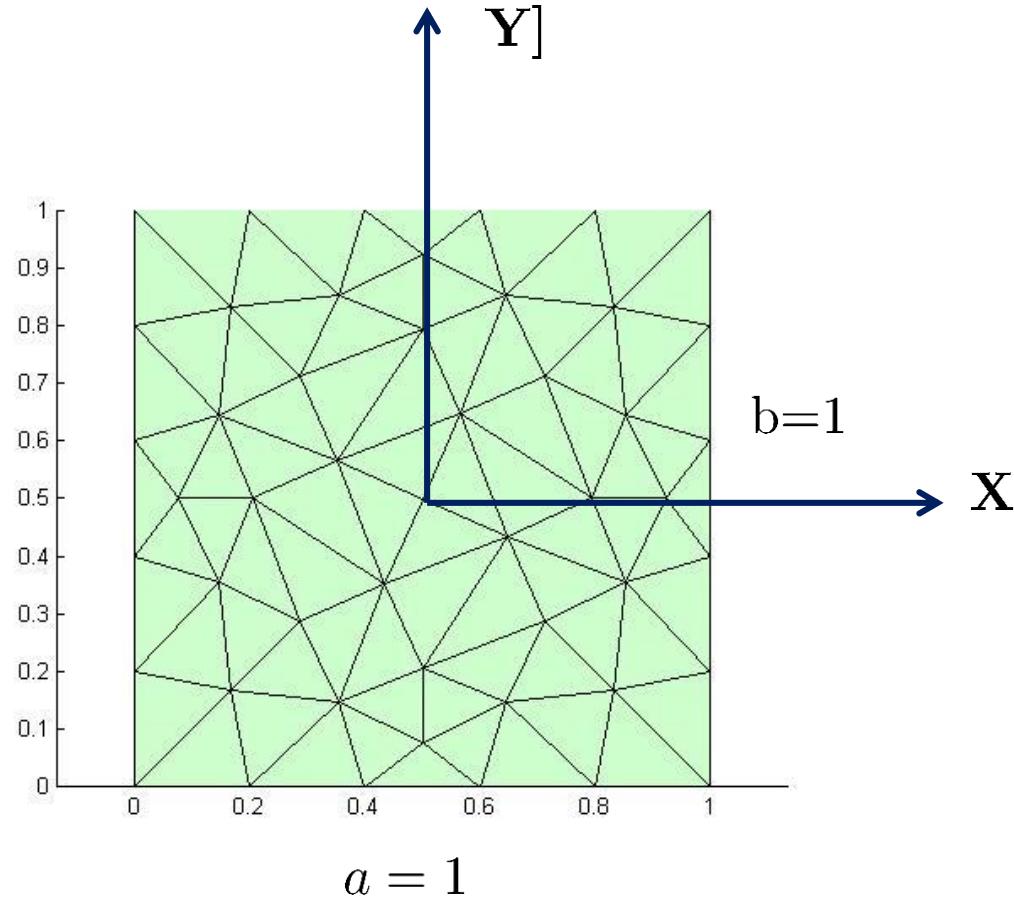
Integration by parts,

$$\begin{aligned} & G\theta^2 \int_{\partial\Omega} \left[\left(\frac{\partial\psi}{\partial x} - y \right) n_x + \left(\frac{\partial\psi}{\partial y} + x \right) n_y \right] \delta\psi dS \\ & - G\theta^2 \int_{\Omega} \left(\frac{\partial^2\psi}{\partial x^2} + \frac{\partial^2\psi}{\partial y^2} \right) \delta\psi d\Omega = 0 . \end{aligned}$$

We recovered the governing equation and the boundary condition.

Given $\theta = 0.2\text{rad}$, we solve for ψ , T , and shear stress distributions.

Project VI Anti-plane Shear Problem



Solve for $w, \gamma_{xz}, \gamma_{yz}, \sigma_{xz}$ and σ_{yz} . (100 Points)

Let $b(x, y) = b_0(1 - 4(x + y))$, the governing equation is

$$\begin{aligned} \frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{yz}}{\partial y} + \frac{\partial \sigma_{zz}}{\partial z} + b(x, y) &= 0 \quad \rightarrow \\ G\left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2}\right) + b(x, y) &= G\nabla^2 w + b(x, y) = 0, \forall (x, y) \in \Omega \\ w &= 0, \quad \forall (x, y) \in \partial\Omega \end{aligned}$$

where $G = 50 MPa$, $t = 1 M$, and $b_0 = 10^5 N/M^3$.

$$\Pi = \frac{1}{2}Gt \int_A \left[\left(\frac{\partial w}{\partial x} \right)^2 + \left(\frac{\partial w}{\partial y} \right)^2 \right] dA - \int_A (bt)w dA$$

Extra Pointe 5: Solve Dynamic Equations:

Let $b(x, y) = b_0(1 - 4(x + y)) \sin \omega t$, with $\omega = 2$.

The governing equation is

$$G\left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2}\right) + b(x, y) = \rho \frac{\partial^2 w}{\partial t^2}, \forall (x, y) \in \Omega$$

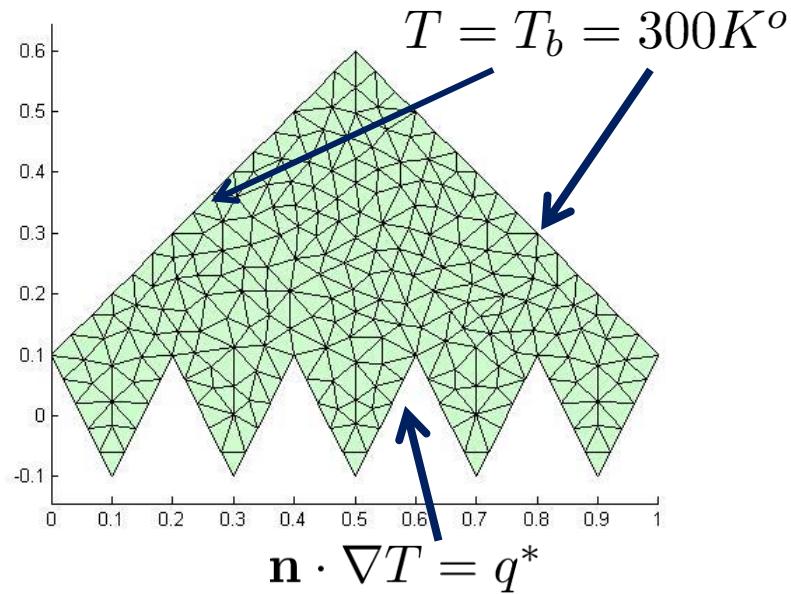
$$w = 0, \forall (x, y) \in \partial\Omega$$

where $G = 50MP_a$ and $b_0 = 10^5 N/M^3$, with $\rho = 7000kg/m^3$.
and proper initial conditions.

The Project III: Weak formulation is,

$$Kt \int_{\Omega} \left(\frac{\partial T}{\partial x} \frac{\partial \delta T}{\partial x} + \frac{\partial T}{\partial y} \frac{\partial \delta T}{\partial y} \right) d\Omega - t \int_{\Omega} Q \delta T d\Omega - \int_{\Gamma_q} q^* \delta T dS = 0$$

In the project, we consider Ω as a circular domain, $K = 1$, $Q = 1$, and $t = 1$; and $q^* = 0.2$.

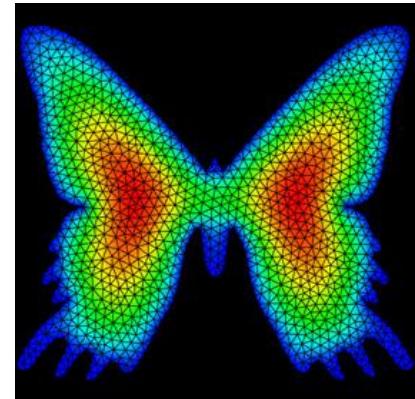
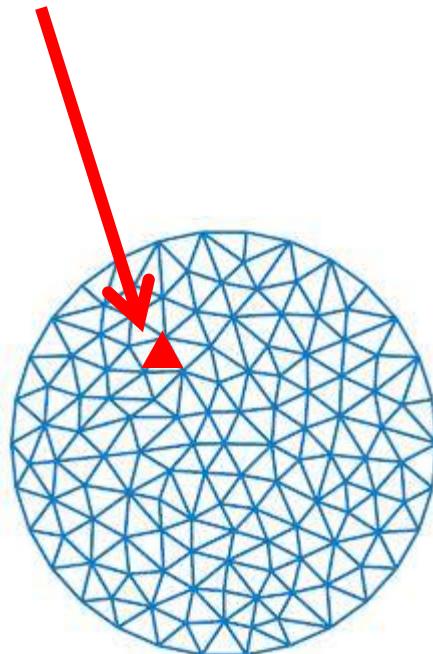


Finite Element Method

$$\delta\Pi = Kt \int_{\Omega} \nabla(\delta T) \cdot (\nabla T) d\Omega - \int_{\Omega} tQ\delta T d\Omega - \int_{\Gamma_q} \bar{q}\delta T dS = 0 .$$



$$\delta\Pi_e = Kt \int_{\Omega_e} \nabla(\delta T) \cdot (\nabla T) d\Omega - \int_{\Omega_e} tQ\delta T d\Omega - \int_{\Gamma_e^q} \bar{q}\delta T dS = 0 .$$



FEM discretization:

$$\Omega = \cup_{e=1}^{nelem} \Omega_e$$

Reproducing Property (Completeness)

Let $T(x, y) = 1, x, y$, and

$$T^h(x, y) = N_1(x, y)T_1 + N_2(x, y)T_2 + N_3(x, y)T_3$$

Then

$$\begin{bmatrix} 1 \\ x \\ y \end{bmatrix} = \begin{bmatrix} 1 & 1 & 1 \\ x_1 & x_2 & x_3 \\ y_1 & y_2 & y_3 \end{bmatrix} \begin{bmatrix} N_1(x, y) \\ N_2(x, y) \\ N_3(x, y) \end{bmatrix}$$

We can solve it,

$$\begin{bmatrix} N_1(x, y) \\ N_2(x, y) \\ N_3(x, y) \end{bmatrix} = \begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix} \begin{bmatrix} 1 \\ x \\ y \end{bmatrix} .$$

$$\text{Inv} \begin{bmatrix} 1 & 1 & 1 \\ x_1 & x_2 & x_3 \\ y_1 & y_2 & y_3 \end{bmatrix} = \begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix}$$

However in Matlab code :
$$\begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix} = \text{Inv} \begin{bmatrix} 1 & x_1 & y_1 \\ 1 & x_2 & y_2 \\ 1 & x_3 & y_3 \end{bmatrix}$$

We then have

$$\begin{bmatrix} N_1(x, y) \\ N_2(x, y) \\ N_3(x, y) \end{bmatrix} = \begin{bmatrix} c_{11} & c_{21} & c_{31} \\ c_{12} & c_{22} & c_{32} \\ c_{13} & c_{23} & c_{33} \end{bmatrix} \begin{bmatrix} 1 \\ x \\ y \end{bmatrix} .$$

$$[\mathbf{g}] := [\nabla T] = \begin{bmatrix} \frac{\partial N_1}{\partial x} & \frac{\partial N_2}{\partial x} & \frac{\partial N_3}{\partial x} \\ \frac{\partial N_1}{\partial y} & \frac{\partial N_2}{\partial y} & \frac{\partial N_3}{\partial y} \end{bmatrix} \begin{bmatrix} T_1 \\ T_2 \\ T_3 \end{bmatrix} .$$

Hence

$$[B] = \begin{bmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} \end{bmatrix} [N_1, N_2, N_3] = \begin{bmatrix} c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix} \rightarrow [\mathbf{B}] = \begin{bmatrix} c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix} .$$

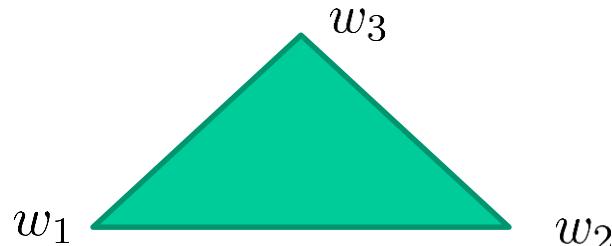
For the antiplane problems,

$$w^h(x, y) = N_1(x, y)w_1 + N_2(x, y)w_2 + N_3(x, y)w_3$$

$$[\gamma] := \begin{bmatrix} \gamma_{xz} \\ \gamma_{yz} \end{bmatrix} = \begin{bmatrix} \frac{\partial w}{\partial x} \\ \frac{\partial w}{\partial y} \end{bmatrix} = \begin{bmatrix} \frac{\partial N_1}{\partial x} & \frac{\partial N_2}{\partial x} & \frac{\partial N_3}{\partial x} \\ \frac{\partial N_1}{\partial y} & \frac{\partial N_2}{\partial y} & \frac{\partial N_3}{\partial y} \end{bmatrix} \begin{bmatrix} w_1 \\ w_2 \\ w_3 \end{bmatrix}.$$

Hence

$$[B] = \begin{bmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} \end{bmatrix} [N_1, N_2, N_3] = \begin{bmatrix} c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix} \rightarrow [\mathbf{B}] = \begin{bmatrix} c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix}.$$



That is why this element is called the Constant Strain Triangle Element (CST).

$$\Big(t\int_{\Omega_e}[B]^T[D][B]d\Omega\Big)[T]-t\int_{\Omega_e}[N]Qd\Omega-\int_{\Gamma_e^q}[N]\bar{q}dS=0~~.$$

$$[K_e]=tK\int_{\Omega_e}\left[\begin{array}{cc}c_{21}&c_{31}\\c_{22}&c_{32}\\c_{23}&c_{33}\end{array}\right]\left[\begin{array}{ccc}c_{21}&c_{22}&c_{23}\\c_{31}&c_{32}&c_{33}\end{array}\right]d\Omega\\=Kt\int_{\Omega_e}\left[\begin{array}{ccc}c_{21}^2+c_{31}^2&c_{21}c_{22}+c_{31}c_{32}&c_{21}c_{23}+c_{31}c_{33}\\c_{22}c_{21}+c_{32}c_{31}&c_{22}^2+c_{32}^2&c_{22}c_{23}+c_{32}c_{33}\\c_{23}c_{21}+c_{33}c_{31}&c_{23}c_{22}+c_{33}c_{32}&c_{23}^2+c_{33}^2\end{array}\right]d\Omega$$

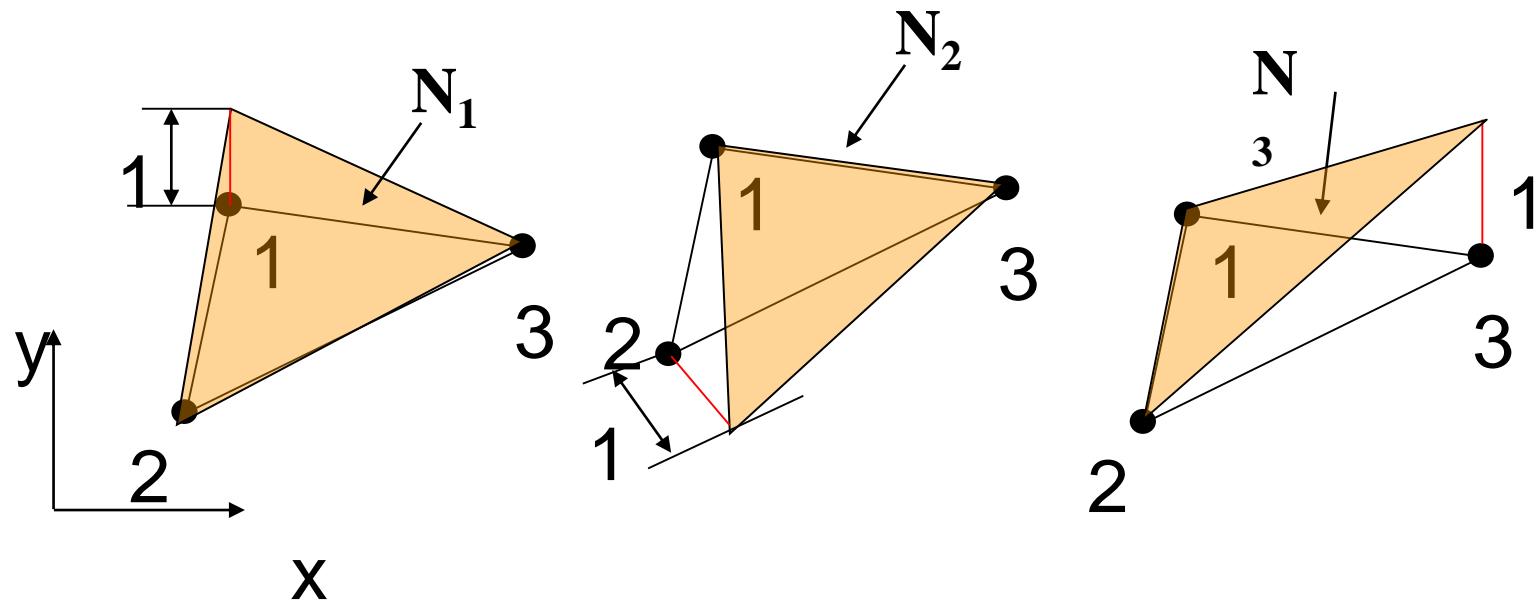
$$K_{ij}=C(2,i)C(2,j)+C(3,i)C(3,j),~~i,j,=1,2,3$$

$$[f_Q]=t\int_{\Omega_e}[N]Qd\Omega,\quad \text{and}\;\;\; [f_q]=\int_{\Gamma_e^q}\bar{q}[N]dS$$

For constant Q, we can calculate

$$[f_Q] = Qt \int_{\Omega_e} [N] d\Omega = \frac{tA_e Q}{3} = \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix} .$$

$$\int_{\Omega_e} \begin{bmatrix} N_1(x, y) \\ N_2(x, y) \\ N_3(x, y) \end{bmatrix} d\Omega = \frac{A_e}{3} \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix}$$



```
function u = fempoi(p,t,e)
```

```
% size of the system
```

```
N = size(p,1); T = size(t,1);
```

```
% matrix allocation
```

```
A = sparse(N,N); b= zeros(N,1);
```

```
Ak = zeros(3,3); bk = zeros(3,1);
```

```
for i = 1:T
```

```
% get the node coordinates of triangle k
```

```
V = [1, p(t(i,1),1), p(t(i,1),2);
```

```
1, p(t(i,2),1), p(t(i,2),2);
```

```
1, p(t(i,3),1), p(t(i,3),2)];
```

fempoi.m

```
C = inv(V)
```

```
area = 0.5*abs(det(V));
```

```
for ik = 1:3
```

```
for jk = 1:3
```

```
Ak(ik,jk) = C(2,ik)*C(2,jk) + C(3,ik)*C(3,jk);
```

```
end
```

```
bk(ik) = area/3;
```

```
end
```

```
% assembly, global stiffness and force vector
```

```
A(t(i,:),t(i,:)) = A(t(i,:),t(i,:)) + Ak*area;
```

```
b(t(i,:)) = b(t(i,:)) + bk;
```

```
end
```

$$\text{Inv} \begin{bmatrix} 1 & x_1 & y_1 \\ 1 & x_2 & y_2 \\ 1 & x_3 & y_3 \end{bmatrix} ?$$

$$= \begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix}$$

$$[K_e] = tK \int_{\Omega_e} \begin{bmatrix} c_{21} & c_{31} \\ c_{22} & c_{32} \\ c_{23} & c_{33} \end{bmatrix} \begin{bmatrix} c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix} d\Omega$$

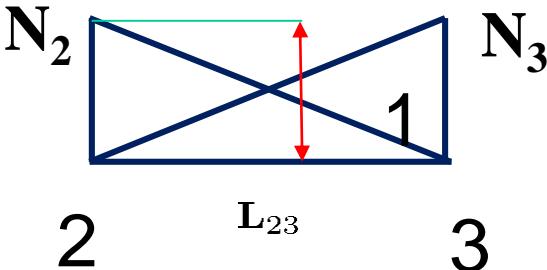
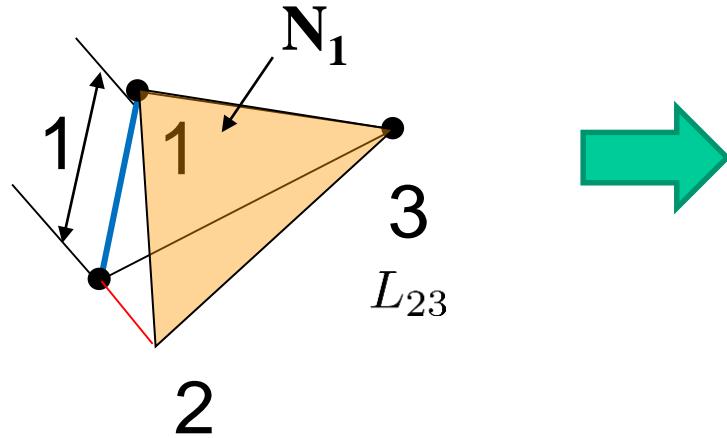
$$K_{ij} = C(2, 1)C(2, j) + C(3, i)C(3, j)$$

$$[f_Q] = \frac{t\Omega_e Q}{3} = \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix} .$$

For constant q^* , we have

$$[f_q] = q^* \int_{\Gamma_e} [N] dS = q^* \int_{\Gamma_e} \begin{bmatrix} N_1 \\ N_2 \\ N_3 \end{bmatrix} dS$$

which can be reduced to



%% Loop over all line element

for i = 1:Nl

Idx = [en(i),en(i+1)]; % The index of line element

% The length of the current line element

len = sqrt((p(Idx(1),1)-p(Idx(2),1))^2 + (p(Idx(1),2)-p(Idx(2),2))^2);

% Assembly with $q^* = 0.2$

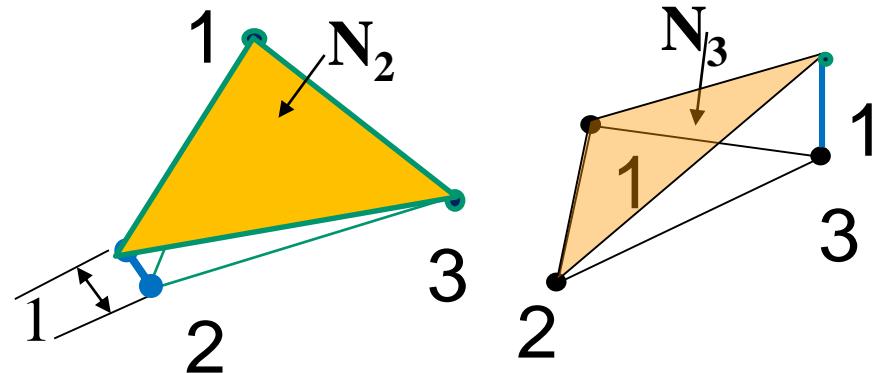
b(Idx) = b(Idx) + 0.2*len/2*[1;1];

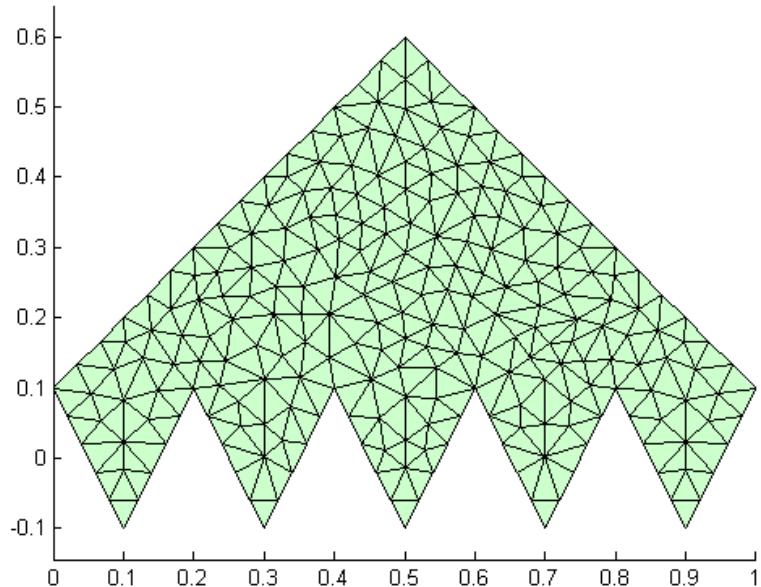
end

$$\frac{q^* L_{12}}{2} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix}$$

$$\frac{q^* L_{23}}{2} \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix}$$

$$\frac{q^* L_{13}}{2} \begin{bmatrix} 1 \\ 0 \\ 1 \end{bmatrix}$$





% Complex polygon geometry,
% mixed Dirichlet/Neumann

```
x=(0:.1:1)';
y=.1*cos(10*pi*x);
pv=[x,y; .5,.6;0,0.1];
[p,t,e]=pmesh(pv,0.05,0);
e=e(p(e,2)>=.6-abs(p(e,1)-.5));
u=fempoi(p,t,e);
figure; tplot(p,t,u)
```

How to find the Neumann Boundary Node Set ?

You only need one line matlab sentence

How to code.

```
% % Define boundary conditions  
% index of Dirichlet boundary  
ed=e(p(e,2)>=.6-abs(p(e,1)-.5));  
% index of Neumann boundary  
en = setdiff(e,ed);  
en = [ed(1);en;ed(2)];
```

$$[f_q] = q^* \int_{\Gamma_e} [N] dS = q^* \int_{\Gamma_e} \begin{bmatrix} N_1 \\ N_2 \\ N_3 \end{bmatrix} dS$$

is your own business !

```
% assembly, global stiffness and force vector
```

```
A(t(i,:),t(i,:)) = A(t(i,:),t(i,:)) + Ak*area;
```

```
b(t(i,:)) = b(t(i,:)) + bk;
```

```
end
```

```
% for i = 1:length(e)
```

```
%   A(e(i),e(i)) = 1.0e30; b(e(i)) = 0;
```

```
% end
```

```
for i = 1:length(e)
```

```
    for j = 1:N
```

```
        A(e(i),j) = 0; A(j,e(i)) = 0;
```

```
    end
```

```
    A(e(i),e(i)) = 1; b(e(i)) = 0;
```

```
end
```

You cannot use this !

```
% solve the system;
```

```
u = A\b;
```

```
end
```

How to find idf and idd
is your own business

$$\begin{bmatrix} \mathbf{K}_{ff} & \mathbf{K}_{fd} \\ \mathbf{K}_{df} & \mathbf{K}_{dd} \end{bmatrix} \begin{bmatrix} \mathbf{T}_f \\ \mathbf{T}_d \end{bmatrix} = \begin{bmatrix} \mathbf{F}_f \\ \mathbf{F}_d \end{bmatrix}$$

$$\mathbf{T}_f = \mathbf{K}_{ff}^{-1} (\mathbf{F}_f - \mathbf{K}_{fd} \mathbf{T}_d)$$

% Perform the solve

Aff = A(idf,idf);

Afd = A(idf,idd);

Adf = A(idd,idf);

Add = A(idd,idd);

Bf = b(idf);

u(idf) = Aff\Bf-Afd*u(idd));

b(idd) = Adf*u(idf)+Add*u(idd)

Project 4:

Choose $S = 1$. Solve

$$S \left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) + p(x, y) = \rho \frac{\partial^2 w}{\partial t^2} \quad \forall (x, y) \in \Omega,$$

$$\Omega = \left\{ (x, y) \mid \text{is an elliptic membrane} \right\}$$

Choose $\rho = 10$.

Boundary condition,

$$w(x, y) = 0, \quad \forall (x, y) \in \partial\Omega$$

Choose

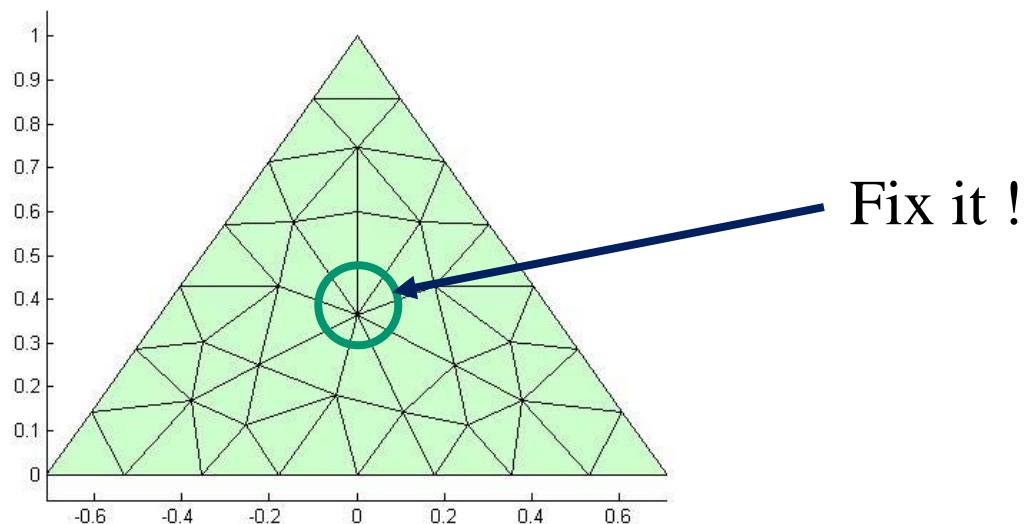
$$p(x, y) = q_0 \mathbf{j} \sin \omega t, \quad \forall |\mathbf{r}| < 1 \text{ where } r = x\mathbf{i} + y\mathbf{j}, \quad \omega = 10 \text{ rad/s, and } q_0 = 1.$$

Project 5:

$$\delta\Pi = G\theta^2 \left\{ \int_{\Omega} \left[\left(\frac{\partial\psi}{\partial x} - y \right) \frac{\partial\delta\psi}{\partial x} + \left(\frac{\partial\psi}{\partial y} + x \right) \frac{\partial\delta\psi}{\partial y} \right] d\Omega \right\} = 0$$

Let $\mathbf{Q} = y\mathbf{i} - x\mathbf{j}$: $\rightarrow \int_{\Omega} \nabla\delta\psi \cdot \nabla\psi d\Omega - \int_{\Omega} \nabla\delta\psi \cdot \mathbf{Q} d\Omega = 0$.

(0, 1)



$(-\sqrt{2}/2, 0)$

$(\sqrt{2}/2, 0)$

Project 6: Solve for $w, \gamma_{xz}, \gamma_{yz}, \sigma_{xz}$ and σ_{yz} .

Let $b(x, y) = b_0(1 - 4(x + y))$, the governing equation is

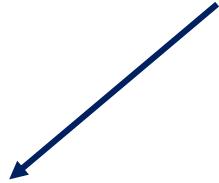
$$\begin{aligned} \frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{yz}}{\partial y} + \cancel{\frac{\partial \sigma_{zz}}{\partial z}} + b(x, y) &= 0 \quad \rightarrow \\ G\left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2}\right) + b(x, y) &= G\nabla^2 w + b(x, y) = 0, \forall (x, y) \in \Omega \\ w &= 0, \quad \forall (x, y) \in \partial\Omega \end{aligned}$$

where $G = 50 MP_a$ and $b_0 = 10^5 N/M^3$.

$$\Pi = \frac{1}{2}Gt \int_A \left[\left(\frac{\partial w}{\partial x} \right)^2 + \left(\frac{\partial w}{\partial y} \right)^2 \right] dA - \int_A (bt)w dA$$

$$\Pi = \frac{1}{2}Gt \int_A \left[\left(\frac{\partial w}{\partial x} \right)^2 + \left(\frac{\partial w}{\partial y} \right)^2 \right] dA - \int_A (bt) w dA$$

How to calculate



$$t \int_{A_e} b(x, y) \delta w(x, y) dA \quad ?$$

$$b(x, y) = b_0(1 - 4(x + y)) .$$

This leads to

$$[f_Q] = t \int_{A_e} b(x, y) \begin{bmatrix} N_1(x, y) \\ N_2(x, y) \\ N_3(x, y) \end{bmatrix} dA$$

Table 10.1 *Exact integrals for a triangle*

m	n	$I = \int_A x^m y^n dx dy$
0	0	$\int dA = A = [x_1(y_2 - y_3) + x_2(y_3 - y_1) + x_3(y_1 - y_2)]/2$
0	1	$\int y dA = A\bar{y} = A(y_1 + y_2 + y_3)/3$
1	0	$\int x dA = A\bar{x} = A(x_1 + x_2 + x_3)/3$
0	2	$\int y^2 dA = A(y_1^2 + y_2^2 + y_3^2 + 9\bar{y}^2)/12$
1	1	$\int xy dA = A(x_1 y_1 + x_2 y_2 + x_3 y_3 + 9\bar{x}\bar{y})/12$
2	0	$\int x^2 dA = A(x_1^2 + x_2^2 + x_3^2 + 9\bar{x}^2)/12$

Possible Questions ?

1. How to derive the weak form if you know the total potential energy ?
2. How to derive the element stiffness matrix ?
3. How to calculate element external force ?
4. How to assemble globale stiffness matrix ?

Project related problems:

5. What is Fourier Law ?
6. What is warping ?
7. What is anti-plane deformation ?
8. How to derive the potential energy for a membrane ?
9. How to derive the governing equation for a beam on the Winkler foundation ?
10. How to derive the A-matrix for a truss structure ?

11. How to take variational derivative ?

Example 7.2

Anti-Plane Shear

We illustrate here the proof that our multidimensional potential energy formulation is equivalent to the partial differential equations given above. We will do this for the anti-plane shear case. Consider a domain A with boundary ∂A , where w is stated to be zero on ∂A . In this case, $\mathcal{S} = \{w(x, y) \mid w = 0 \text{ for } (x, y) \in \partial A\}$ and $\mathcal{V} = \{\delta w(x, y) \mid \delta w = 0 \text{ for } (x, y) \in \partial A\}$. The minimization problem is

$$\min_{w \in \mathcal{S}} \Pi[w(x, y)] = \min_{w \in \mathcal{S}} \left[\int_A \frac{1}{2} G t \nabla w \cdot \nabla w \, dx dy - \int_A b t w \, dx dy \right].$$

**12. What is essential (Dirichelt) boundary condition ?
and what is natural (Neumann) boundary condition ?**

13. Understand and can derive BC for both Saint-Venant torsion and Prandtle stress function problem ?

14. Understand and can derive BC for thermal conduction problem ?

15. How to construct kinematically admissible trial functions;
16. As a special Ritz method, what are the special advantages FEM has ?
17. What is the main difference between the Prandlt stress function method and the Saint-Venant torsion solution ?
18. What are the main differences between method of conservation energy and the minimum potential energy principle ?
19. What is the conservative force ?
20. Given a weak formulation of the assigned project, can you derive the strong form and the natural boundary conditions ?
21. Given a weak formulation of the assigned project, can you derive the FEM discrete algebraic equations ?



Wish you having a great presentation !