

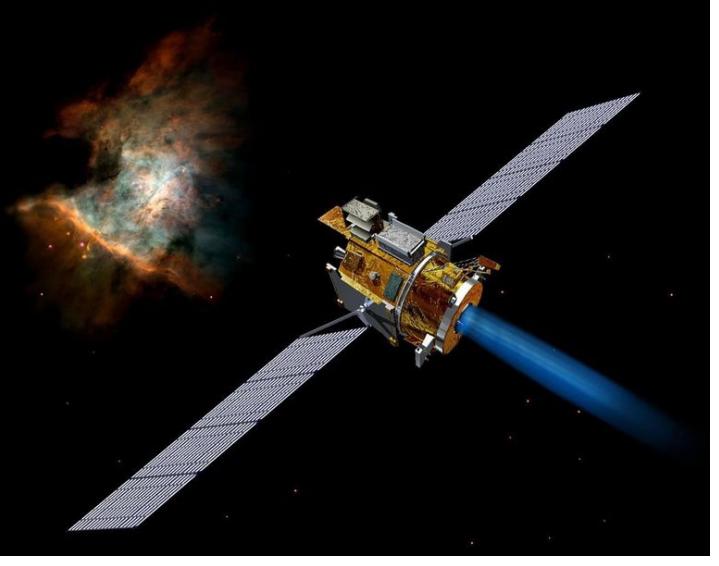
Scaled Inertial Electrostatic Confinement Plasma Source

Matthew Giraldo, Cian Moran, Vahini Patel, Brian Wu, mentored by Rohan Puri

Department of Aerospace Engineering, College of Engineering, University of Illinois at Urbana-Champaign

INTRODUCTION

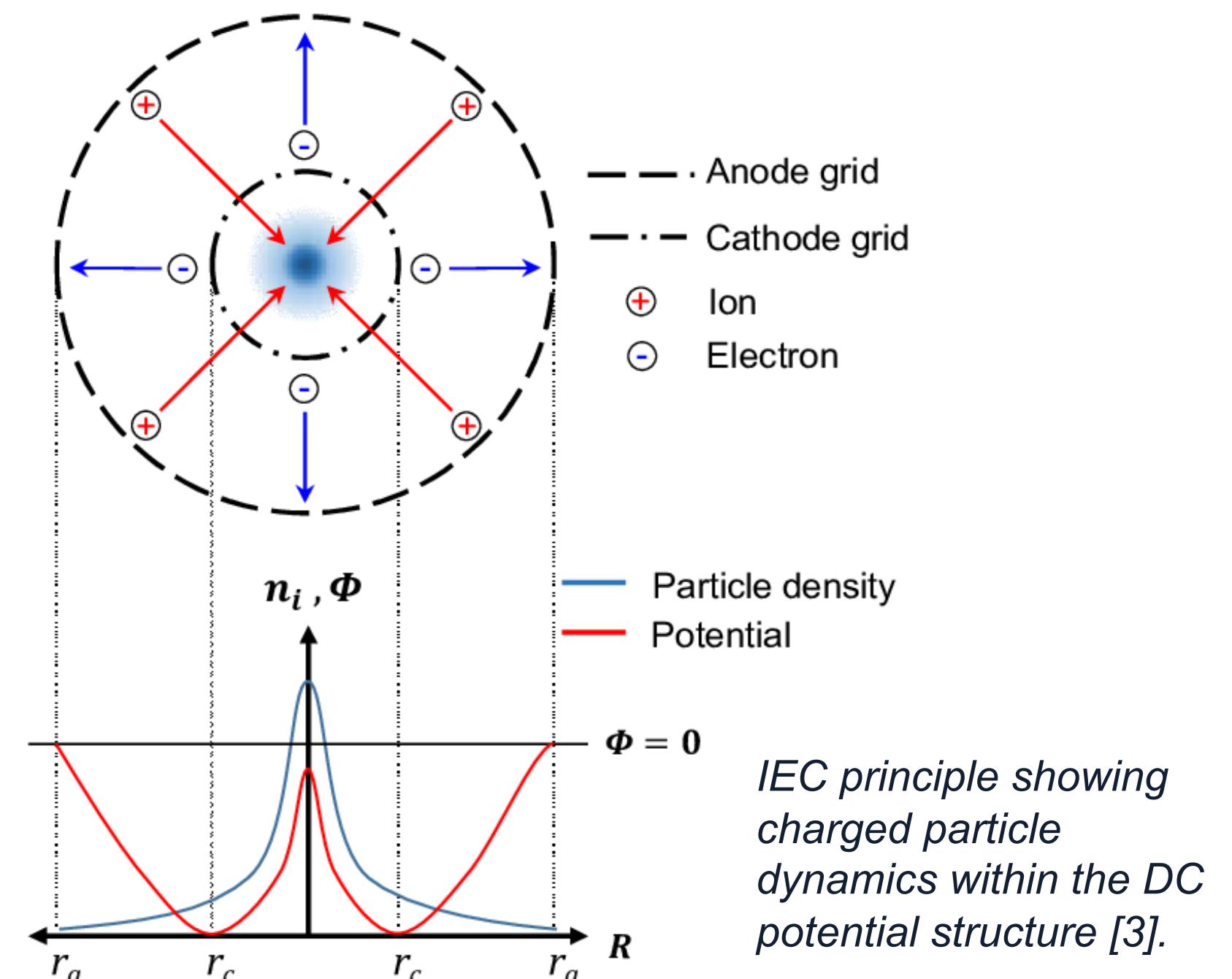
Electric propulsion (EP) systems use electromagnetic (EM) fields to accelerate the propellant to high speeds and generate thrust. [1].



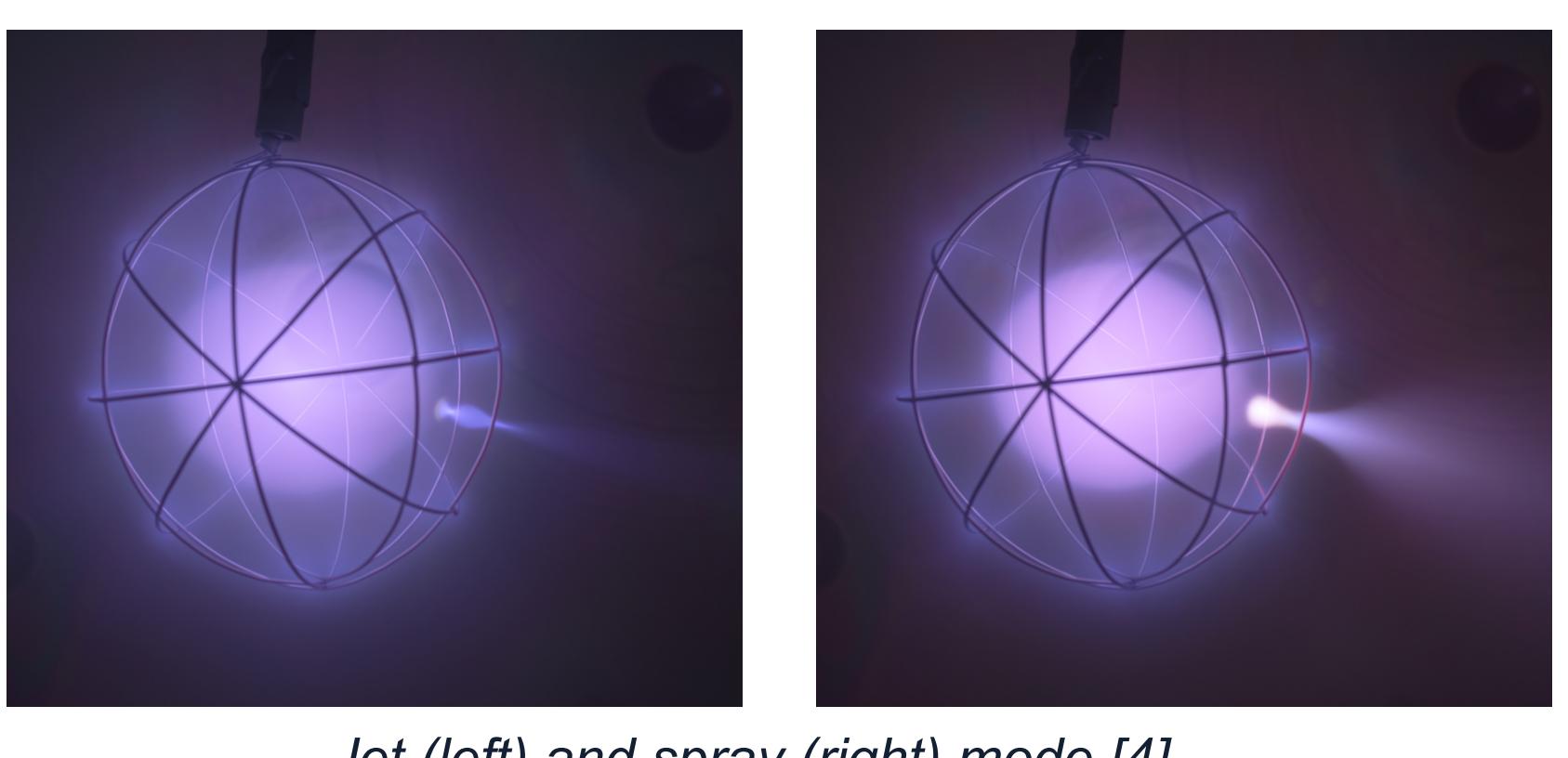
Deep Space 1 (1998)
- first US EP
spacecraft with ion
thrusters [2]

Inertial electrostatic confinement (IEC) systems:

- Use direct current (DC) electric fields to ionize propellant gas
- Sustain plasma by recirculating ions through a transparent cathode grid

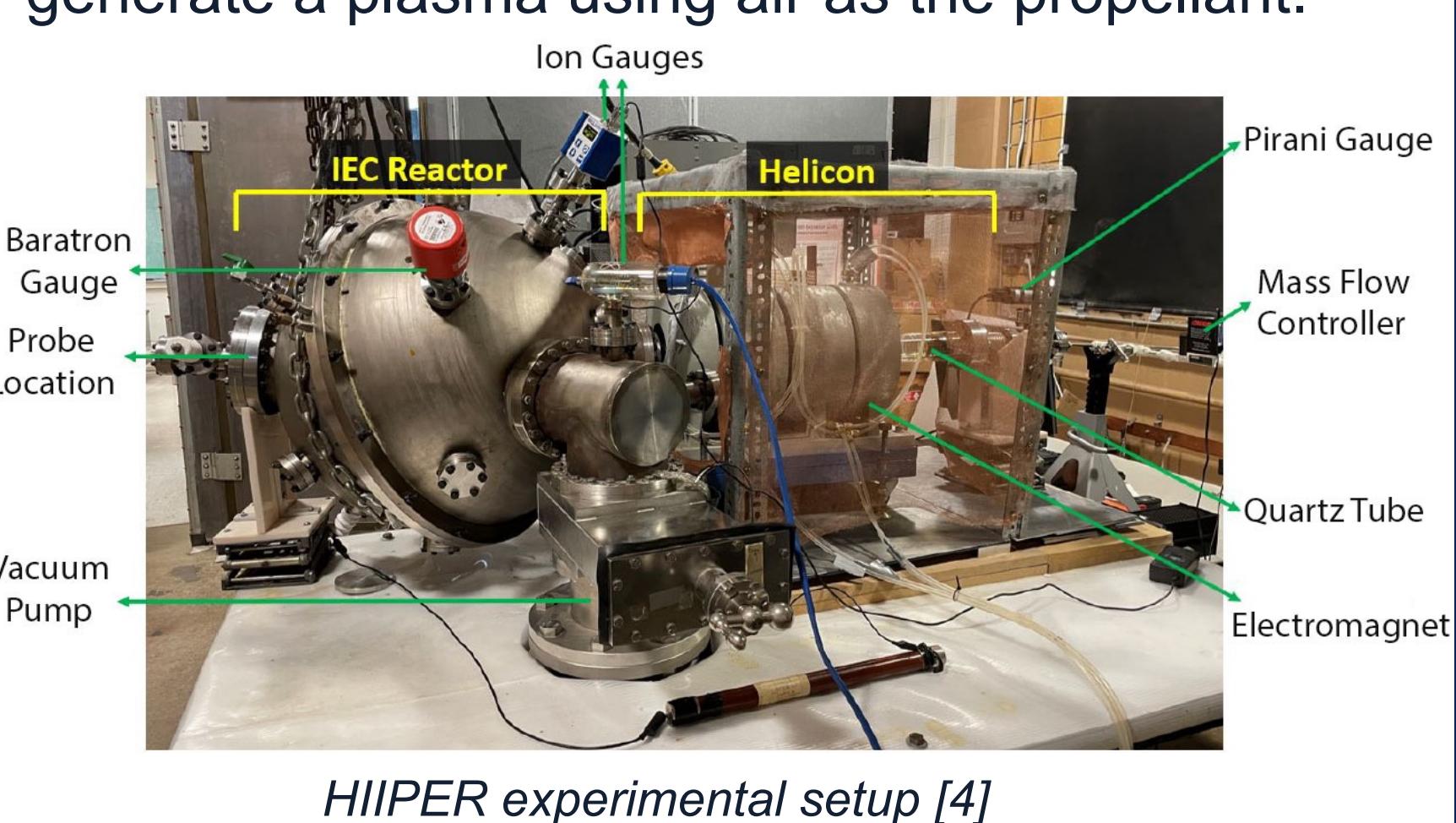


The plasma beam ejects as a collimated jet or diverging spray, capable of producing thrust.



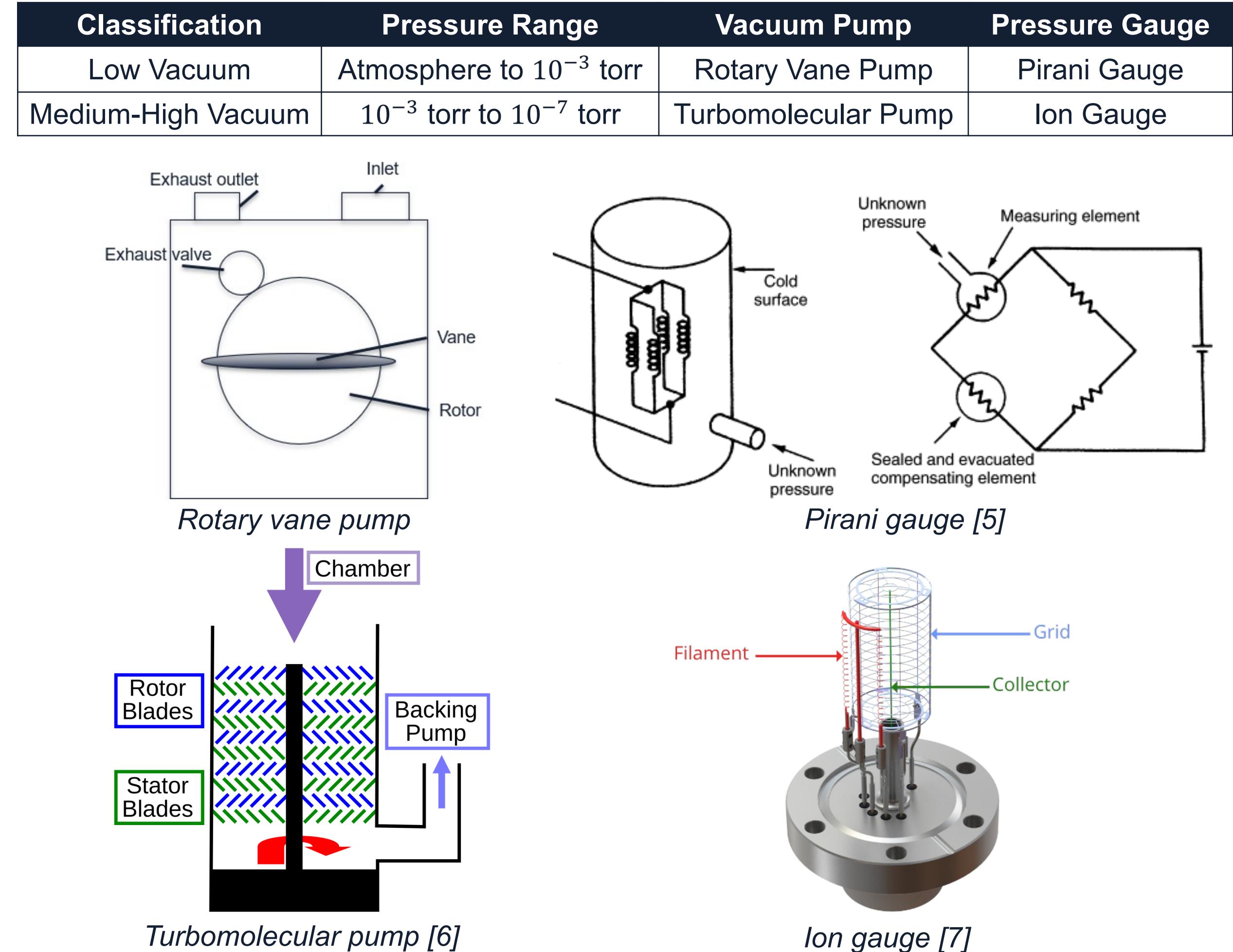
OBJECTIVE

Set up a scaled-down helicon-IEC thruster and generate a plasma using air as the propellant.



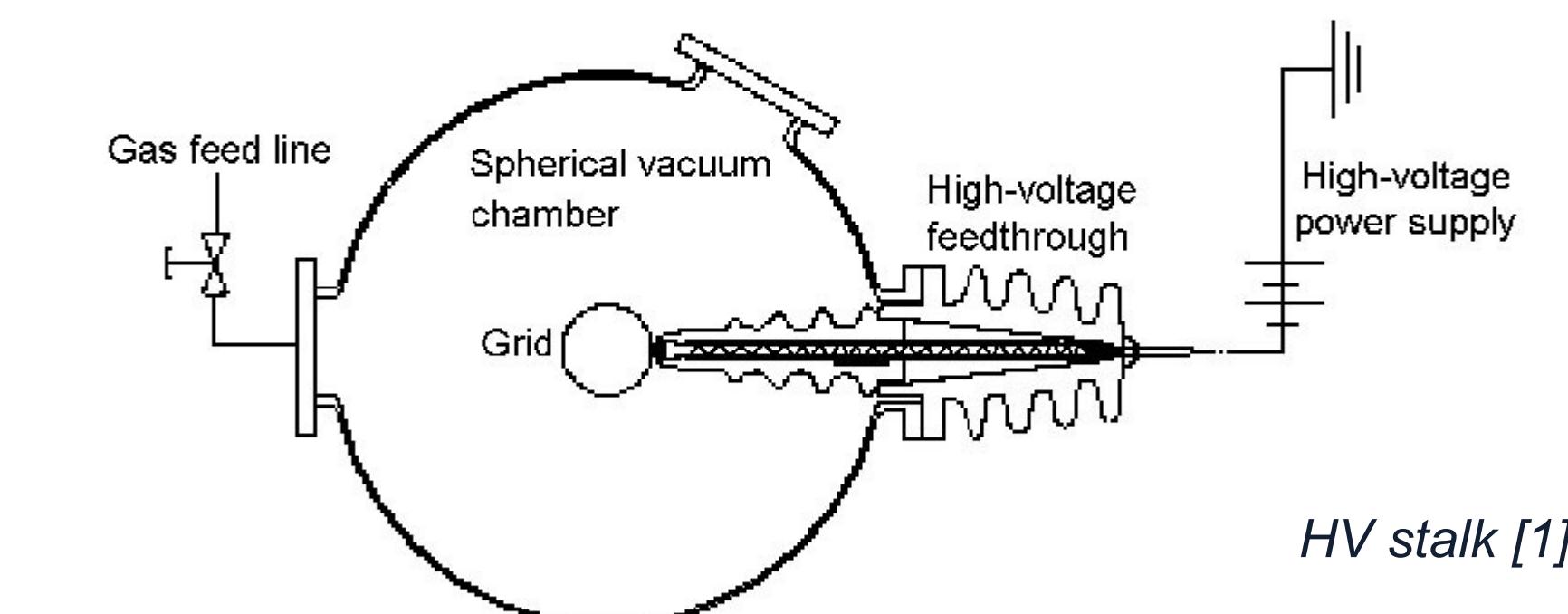
EXPERIMENTAL SETUP

VACUUM COMPONENTS



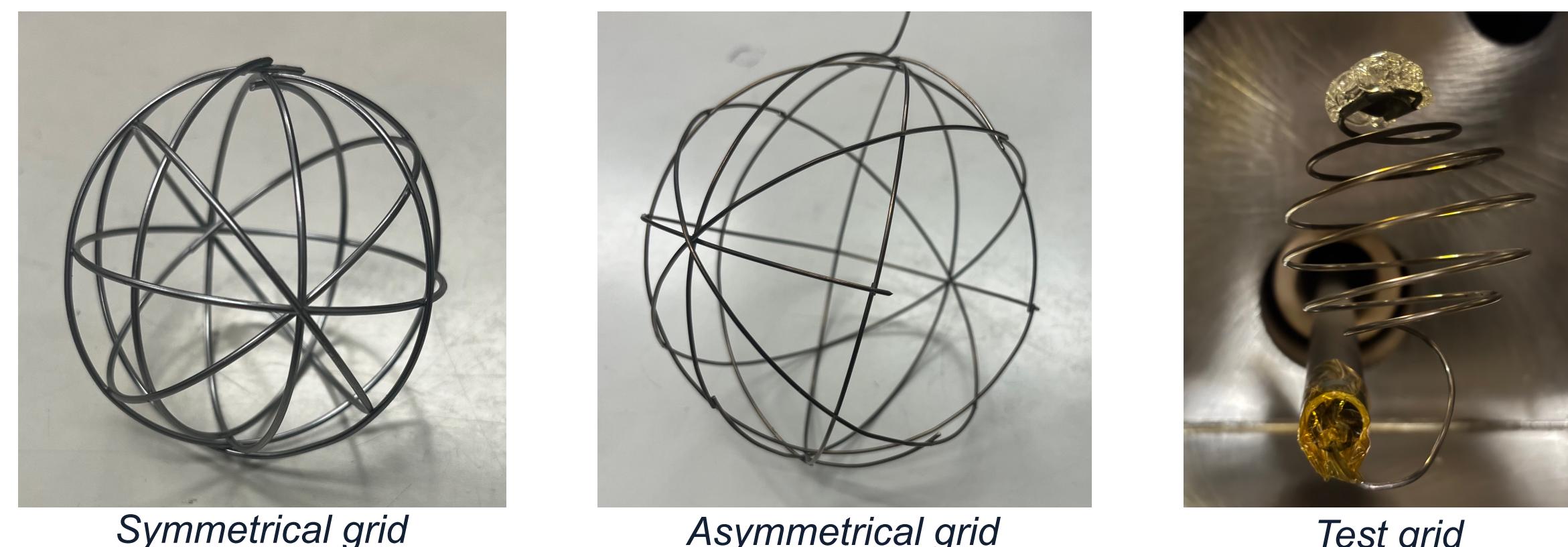
STALK

- Insulates the high-voltage (HV) conductor
- Shields the conductor from the plasma
- Minimizes the electric field stresses

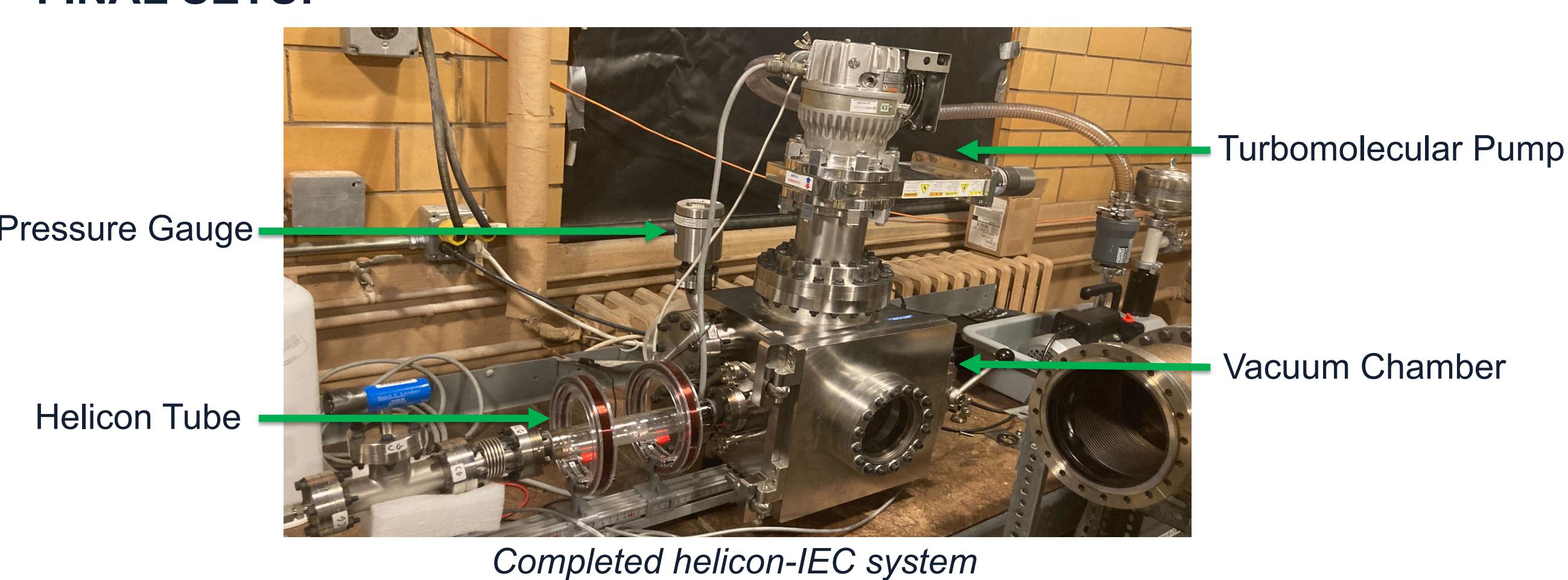


CATHODE GRID

Negative DC bias causes dielectric breakdown of the propellant gas.

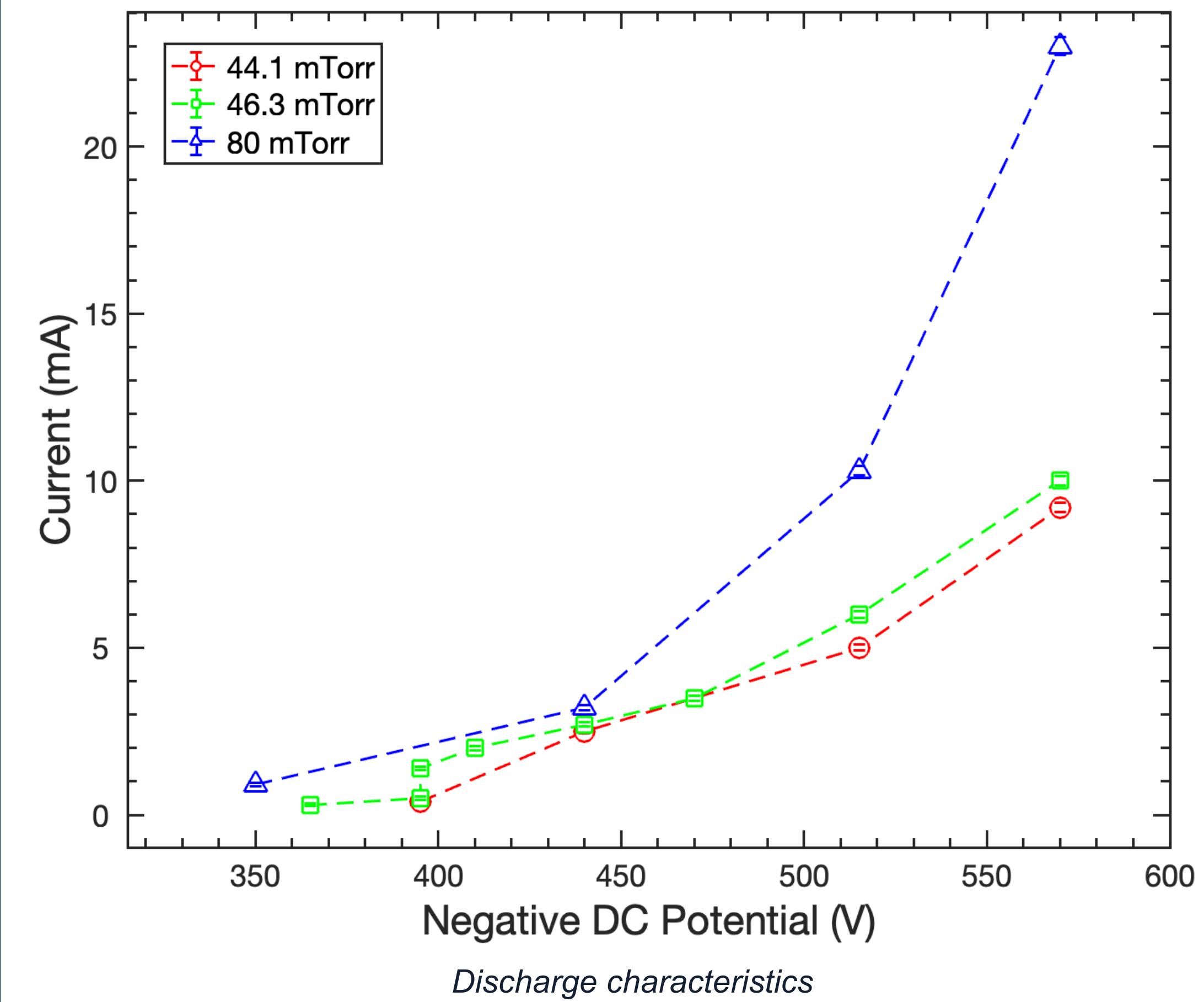


FINAL SETUP



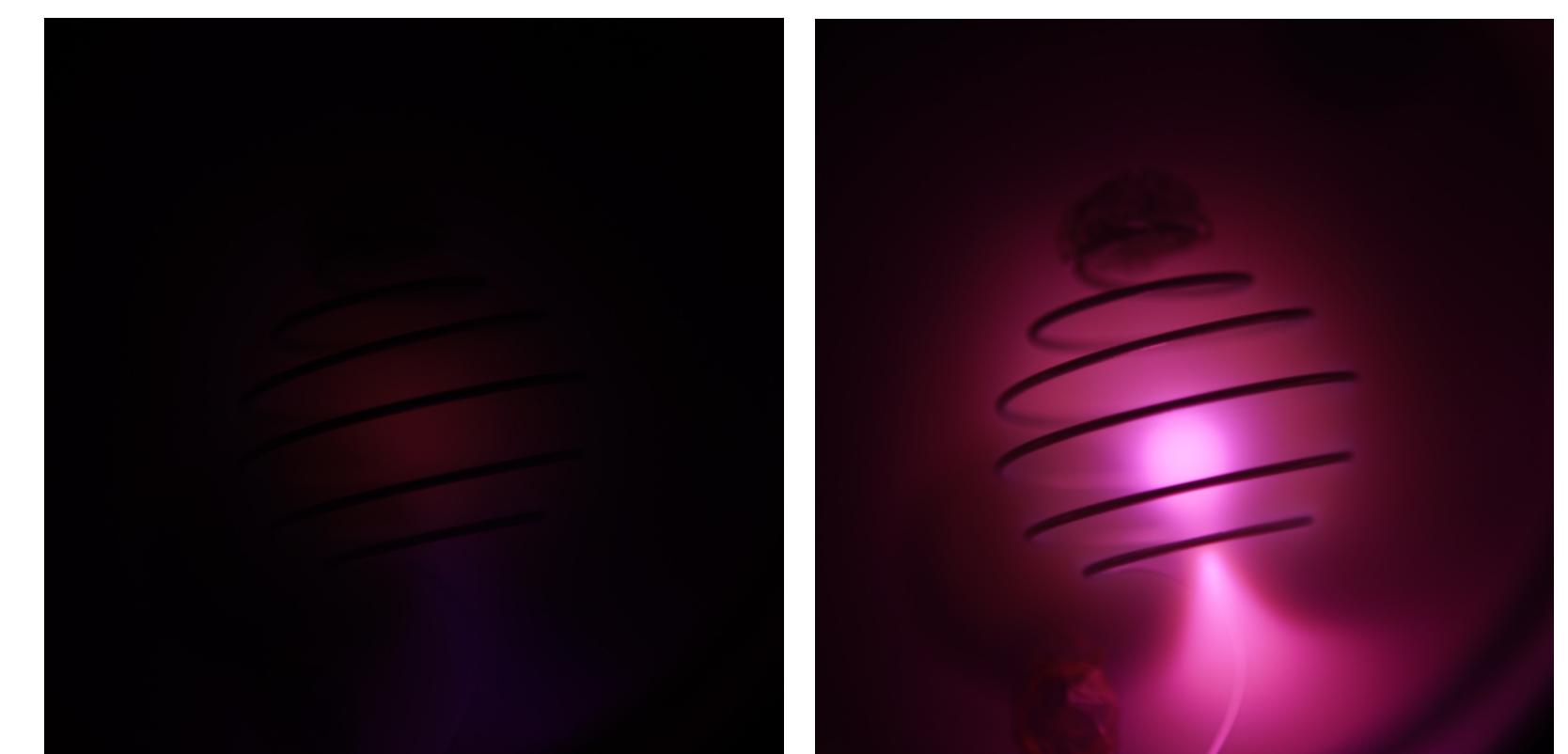
RESULTS

- Grid draws higher current at higher voltages, keeping pressure constant
- For the same voltages, grid draws higher current at higher background gas pressures



RESULTS

- For 46.3 mTorr, an abrupt increase in current is observed at 395 V



$I = 0.5 \text{ mA}$
 $I = 1.4 \text{ mA}$

Plasma discharge at 46.3 mTorr and 395 V

- Background gas pressure measured with lowest mass flow controller (MFC) setting



Pressure reading obtained using convection enhanced Pirani

DISCUSSION

The discharge current is directly proportional to the background gas pressure as well as the magnitude of the applied potential. The light intensity of the plasma increases with drawn current.

A mode transition is observed when the current drawn by the grid increases abruptly, maintaining a fixed voltage at a constant pressure.

FUTURE WORK – Repeat the experiment at lower pressures and with different propellants.

ACKNOWLEDGEMENTS

Special thanks to Rohan Puri for his mentorship and guidance, as well as Dr. George H. Miley and Dr. Erik Ziehm for their support and advice.

REFERENCES

- [1] Miley, G., Momota, H., Linchun, W., Reilly, M., Teigillo, V., Burton, R., Dell, R., Dell, D., and Hargus, W., "IEC Thrusters for Space Probe Applications and Propulsion", AIAA Conference Proceedings, 2009.
- [2] "Deep Space 1", NASA, <https://nssdc.gsfc.nasa.gov/nmc/moon/missions/cassini.html>, October 2024.
- [3] Miley, G., Boebinger, A., Chackack, A., Chan, Y., Ehtesami, M., Harmans, N., Montag, C., Romano, F., Skaten, J., Fasolus, S., Komursaki, K., Schonher, T. (2017). "Research and Development on Electric and Advanced Propulsion at IRS". International Electric Propulsion Conference, 2017.
- [4] Puri, R., Miley, G., Rover, J., Gong, M., and Ziehm, E., "Reassessing Jet Mode of Inertial Electrostatic Confinement Thruster for Space Propulsion Application", AIAA SciTech Forum, 2024.
- [5] Langari, R., and Morris, A.S. "Pirani Gauges". Measurement and Instrumentation, 2012.
- [6] "About turbomolecular pumps", <https://vaccoat.com/blog/learn-more-about-turbomolecular-pumps/>, dated 1 December 2024.
- [7] Taylor, J., "How does an ion gauge work?", Anan Microelectronics Ltd. Available: <https://arunmicro.com/news/how-does-an-ion-gauge-work/>.