



Part(a): Natural Frequencies and Mode Shapes [10]

Weight of roof slab = $(5 \text{ kN/m}^2) (7\text{ m}) (5\text{ m}) = 175 \text{ kN}$

$$\text{Mass, } m = \frac{W}{g} = 1.786 \times 10^4 \text{ kg}$$

Mass moment of inertia about vertical axis passing through O,

$$I_0 = m \cdot \left(\frac{B^2 + D^2}{12} \right) = 1.101 \times 10^5 \text{ kg-m}^2$$

The plan is symmetric about x-axis. Hence, gm in N-S direction will

not cause any motion along X-(E-W) direction.

Motion along X-direction

$$m\ddot{u}_x + 2k_x u_x = 0$$

$$\text{where } k_x = k_A = k_B = 500 \text{ kN/m}$$

$$\Rightarrow \text{Natural frequency } \omega_x = \sqrt{\frac{2k_x}{m}} = \sqrt{\frac{2 \times 500 \times 10^3 \text{ N/m}}{1.101 \times 10^5 \text{ kg}}} = 7.483 \text{ rad/sec}$$

$$\text{" period } T_x = \frac{2\pi}{\omega_x} = 0.84 \text{ sec}$$

(+2)

Motion along Y and θ directions:

$$\text{Mass matrix } \underline{m} = \begin{bmatrix} m & 0 \\ 0 & I_0 \end{bmatrix} = \begin{bmatrix} 1.786 \times 10^4 & 0 \\ 0 & 1.101 \times 10^5 \end{bmatrix} \text{ in kg and m unit}$$

(+1)

Determine stiffness matrix using direct stiffness method

$$u_i = \underline{a}_i \underline{u} \quad \text{where } u_i = \text{displacement of } i\text{th frame}$$

$$\underline{a}_i = \text{transformation matrix}$$

$$\underline{u} = \begin{Bmatrix} u_y \\ \theta \end{Bmatrix}$$

For frame-A,

$$\underline{a}_A = \langle 0 \quad -d/2 \rangle = \langle 0 \quad -2 \rangle$$

$$\text{Similarly, } \underline{a}_B = \langle 0 \quad +d/2 \rangle = \langle 0 \quad 2 \rangle$$

$$\underline{a}_C = \langle 1 \quad -b/2 \rangle = \langle 1 \quad -3 \rangle$$

$$\underline{a}_D = \langle 1 \quad +b/2 \rangle = \langle 1 \quad 3 \rangle$$

∴ Total stiffness matrix

$$\begin{aligned} \underline{k} &= \sum_{i=1}^N \underline{a}_i^T \underline{k}_i \underline{a}_i = \underline{a}_A^T \underline{k}_A \underline{a}_A + \underline{a}_B^T \underline{k}_B \underline{a}_B + \underline{a}_C^T \underline{k}_C \underline{a}_C + \underline{a}_D^T \underline{k}_D \underline{a}_D \\ &= \begin{Bmatrix} 0 \\ -2 \end{Bmatrix} \frac{500}{\times 10^3} \begin{Bmatrix} 0 & -2 \end{Bmatrix} + \begin{Bmatrix} 0 \\ 2 \end{Bmatrix} \frac{500}{\times 10^3} \begin{Bmatrix} 0 & 2 \end{Bmatrix} + \begin{Bmatrix} 1 \\ -3 \end{Bmatrix} \frac{250}{\times 10^3} \begin{Bmatrix} 0 & -3 \end{Bmatrix} + \begin{Bmatrix} 1 \\ 3 \end{Bmatrix} \frac{400}{\times 10^3} \begin{Bmatrix} 0 & 3 \end{Bmatrix} \\ &= \begin{bmatrix} 650 & 450 \\ 450 & 9.85 \times 10^3 \end{bmatrix} \times 10^3 \text{ in N and m unit} \end{aligned}$$

(13)

Perform eigenvalue analysis to get

$$\omega_1 = 5.875 \text{ rad/sec} \quad \text{and} \quad \omega_2 = 9.557 \text{ rad/sec}$$

$$T_1 = 1.07 \text{ sec} \quad \text{and} \quad T_2 = 0.658 \text{ sec}$$

$$\underline{\phi}_1 = \begin{Bmatrix} 1 \\ -0.071 \end{Bmatrix} \quad \underline{\phi}_2 = \begin{Bmatrix} 0.459 \\ 1 \end{Bmatrix} \quad (14)$$

Part (b): Response Spectrum Analysis [20]

Modal masses

$$M_1 = \underline{\phi}_1^T \underline{m} \underline{\phi}_1 = 1.847 \times 10^4 \text{ kg}$$

$$M_2 = \underline{\phi}_2^T \underline{m} \underline{\phi}_2 = 1.139 \times 10^5 \text{ kg}$$

$$L_n^h = \underline{\phi}_n^T \underline{m} \underline{z} \quad \text{where } \underline{z} = \begin{Bmatrix} 1 \\ 0 \end{Bmatrix}$$

$$= \underset{1 \times 2}{\langle \phi_{yn} \phi_{on} \rangle} \underset{2 \times 2}{\begin{bmatrix} m & 0 \\ 0 & I_0 \end{bmatrix}} \underset{2 \times 1}{\begin{Bmatrix} 1 \\ 0 \end{Bmatrix}} = \underset{1}{\langle m \phi_{yn} \quad I_0 \phi_{on} \rangle} \underset{2}{\begin{Bmatrix} 1 \\ 0 \end{Bmatrix}} = \underset{3}{m \phi_{yn}}$$

$$\therefore L_1^n = 1.786 \times 10^7, \quad L_2^n = 1.786 \times 10^7 \times 0.459 = 8.198 \times 10^6$$

(+5)

$$\Gamma_n = \frac{L_n}{M_n} \Rightarrow \Gamma_1 = \frac{1.786 \times 10^7}{1.847 \times 10^7} = 0.967 \quad \& \quad \Gamma_2 = 0.072$$

$$\underline{u}_n = \underline{u}_n^{st} A_n \quad \underline{u}_n^{st} = \frac{\Gamma_n}{\omega_n^2} \underline{\phi}_n \Rightarrow \underline{u}_1^{st} = \begin{Bmatrix} 0.028 \\ -2.084 \times 10^{-3} \end{Bmatrix}$$

$$\underline{u}_2^{st} = \begin{Bmatrix} 3.612 \times 10^{-4} \\ 7.876 \times 10^{-9} \end{Bmatrix}$$

$$\left. \begin{aligned} A_1 &= \frac{1.80}{1.07} \times 0.3g = 0.505g \\ A_2 &= 2.71 \times 0.3g = 0.813g \end{aligned} \right\} \underline{u}_1 = \begin{Bmatrix} 0.139 \\ 0.01 \end{Bmatrix} \begin{matrix} m \\ rad \end{matrix} \text{ Mode-1}$$

$$\underline{u}_2 = \begin{Bmatrix} 2.877 \times 10^{-3} \\ 6.275 \times 10^{-3} \end{Bmatrix} \begin{matrix} m \\ rad \end{matrix} \text{ Mode-2}$$

(+5)

$$\therefore u_y = \sqrt{0.139^2 + (2.877 \times 10^{-3})^2} = 0.139m$$

$$\theta = \sqrt{(0.01)^2 + (6.275 \times 10^{-3})^2} = 0.012 rad$$

(+2)

Use transformation matrices to get u_i , then do $k_i u_i$ to get equivalent lateral force in each mode. Then combine using SRSS.

For frame-A:

$$u_{A1} = \begin{matrix} \underline{a}_A \end{matrix} \begin{Bmatrix} 0 & -2 \end{Bmatrix} \begin{Bmatrix} 0.139 \\ 0.01 \end{Bmatrix} = 0.021m \quad \text{Mode-1}$$

$$u_{A2} = \begin{Bmatrix} 0 & -2 \end{Bmatrix} \begin{Bmatrix} 2.877 \times 10^{-3} \\ 6.275 \times 10^{-3} \end{Bmatrix} = -0.013m$$

$$f_{A1} = k_A u_{A1} = 10.306 \text{ kN} \quad \text{Mode-1}$$

(+8)

$$f_{A2} = k_A u_{A2} = -6.275 \text{ kN} \text{ Mode-1}$$



$$\therefore f_A = \sqrt{(10.306)^2 + (-6.275)^2} = 12.066 \text{ kN} \checkmark$$

Similarly, $f_B = 12.066 \text{ kN}$, $f_C = 42.554 \text{ kN}$, $f_D = 43.918 \text{ kN}$