

Conceptual Data Flow and Components

The basic components required for simulation framework:

- 1. A structural dynamics model that outputs airframe deformation.
 - a. Inputs: Forces and moments acting on the structure
 - b. Outputs: Velocities and accelerations
- 2. Aerodynamic model
 - a. Function of the flight conditions
 - b. Rigid-body attitudes
- 3. EoM block
 - a. Total forces and moments acting on the aircraft
 - b. Aircraft mass properties.
- 4. A gravity model to compute the gravitational forces acting on the aircraft.
- 5. Control surface and flight control systems to simulate controlled flight

Test conditions:

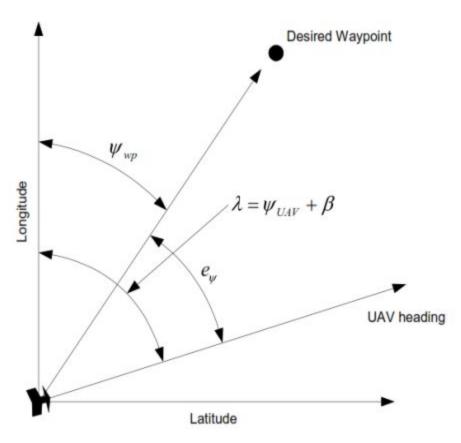
- 1. Parameters tests to assure that the model is free of principle errors.
- 2. Unit tests assess the functionality of an individual software component.
- 3. Simulation provides the infrastructure for operational system integration.
- 4. Flight tests provide a practical proof of concept for validate run-time and
- 5. hardware thresholds.

Equations need to be defined:

- 1. Body-axes force
- 2. Moments
- 3. Kinematics
- 4. Navigations

Inner loop control:

- Lateral Controller, Longitudinal Controller, Altitude Controller
- Roll Rate Controller, Pitch Rate Controller
- Angle Controller, Heading Controller
- Speed Controller, Navigation Controller



waypoint tracking angle