# Lecture 18: Newton-Euler equations of motion, Modelica/Dymola: The Multibody library

- Newton-Euler equations of motion
  - Recap
  - Kinetic energy
  - Example
- Software
  - Dymola and the Modelica.Mechanics.Multibody library

Book: 7.3

### Newton-Euler equations of motion

Newton's law (for particle k)

$$m_k \vec{a}_k = \vec{F}^{(r)}$$

- Newton-Euler EoM for rigid bodies:
  - Integrate Newton's law over body, define center of mass
  - Define torque/moment and angular momentum to handle forces that give rotation about center of mass
  - Define inertia dyadic/matrix

$$\vec{F}_{bc} = m\vec{a}_c$$

$$\vec{T}_{bc} = \vec{M}_{b/c} \cdot \vec{\alpha}_{ib} + \vec{\omega}_{ib} \times \left( \vec{M}_{b/c} \cdot \vec{\omega}_{ib} \right) \not\sim_{\vec{i}_1}$$

(Here: Referenced to center of mass)

Implemented in e.g. Dymola (Modelica.Multibody library)

 $\vec{r}_c$ 

#### Traits of Newton-Euler EoM

(and a preview: Lagrange EoM)

#### Newton-Euler EoM:

- Involves working with vectors
  - Lagrange: Algebraic manipulations
- Forces and moments are central
  - Lagrange: Energy and work are central
- All forces in the system must be considered
  - Lagrange: Forces of constraint are implicitly eliminated with the use of generalized coordinates (and generalized forces)

 $\vec{F}_{bc} = m\vec{a}_c$ 

 $\vec{T}_{bc} = \vec{M}_{b/c} \cdot \vec{\alpha}_{ib} + \vec{\omega}_{ib} \times \left( \vec{M}_{b/c} \cdot \vec{\omega}_{ib} \right)$ 

- Somewhat complicated to use by hand, but can be implemented in computer systems
  - Lagrange: Easier to do by hand, not suitable for complex systems
- d'Alembert's principle: Elimination of forces of constraint (Ch. 7.7)
  - Can simplify application of Newton-Euler EoM
    - Kane's EoM (Ch. 7.8, 7.9)
  - Starting point for Lagrange EoM (Ch. 8.2)

#### Inertia matrix

Found for each rigid body by calculating

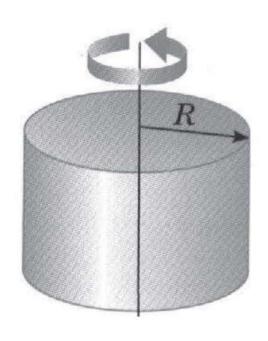
$$M_{b/c}^b = \int_b (\mathbf{r}^b)^\mathsf{T} \mathbf{r}^b I - \mathbf{r}^b (\mathbf{r}^b)^\mathsf{T} dm = \int_b \begin{pmatrix} y^2 + z^2 & -xy & -xz \\ -xy & x^2 + z^2 & -yz \\ -xz & -yz & x^2 + y^2 \end{pmatrix} dm$$

- Constant in body-fixed coordinate system!
- Not constant in inertial coordinate system

$$M_{b/c}^i = R_b^i M_{b/c}^b (R_b^i)^\mathsf{T}$$

- Books and wikipedia have tables for common geometries, otherwise computer programs calculates, or can be calculated/identified based on experiments
- Typically, axis in body-system chosen as body symmetri axis, giving zeros in inertia matrix. If symmetric about all axis, the inertia matrix becomes diagonal.

### Inertia matrix, examples





#### Homogeneous Disk

$$I_{disk} = \frac{1}{4}mr^2 \begin{bmatrix} 1 + \frac{1}{3}\frac{h}{r^2} & 0 & 0\\ 0 & 1 + \frac{1}{3}\frac{h}{r^2} & 0\\ 0 & 0 & \frac{1}{2} \end{bmatrix}$$

$$I_{disk} = \frac{1}{4}mr^{2} \begin{bmatrix} 1 + \frac{1}{3}\frac{h}{r^{2}} & 0 & 0\\ 0 & 1 + \frac{1}{3}\frac{h}{r^{2}} & 0\\ 0 & 0 & \frac{1}{2} \end{bmatrix} \quad I = \begin{bmatrix} 23 & 0 & 2.97\\ 0 & 15.13 & 0\\ 2.97 & 0 & 16.99 \end{bmatrix} kslug - ft^{2}$$

1 slug = 14.6 kg1 ft = 0.304 m

	Kinematics	Kinetics
	Derivatives of position and orientation as function of velocity and angular velocity	Derivatives of velocity and angular velocity as function of applied forces and torques
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www.ntnu.no		TTK4130 Modeling and Simulation

#### **Kinematics** Derivatives of position and

orientation as function of velocity

**Kinetics** 

3D:

Derivatives of velocity and angular

1D:  $m\dot{v} = F$  3D:  $m\dot{\mathbf{v}}_c^i = \mathbf{F}_{bc}^i$ 

Usually convenient to have forces

and velocities in body system:

velocity as function of applied

forces and torques

3D: Depends on parameterization Rotation matrix:

$$\mathbf{\dot{R}}_{b}^{i}=\mathbf{R}_{b}^{i}\left(oldsymbol{\omega}_{ib}^{b}
ight)^{ imes}$$

Euler angles:

gles.
$$\dot{oldsymbol{\phi}} = \mathbf{E}_d^{-1}(oldsymbol{\phi}) oldsymbol{\omega}_{ib}^b$$

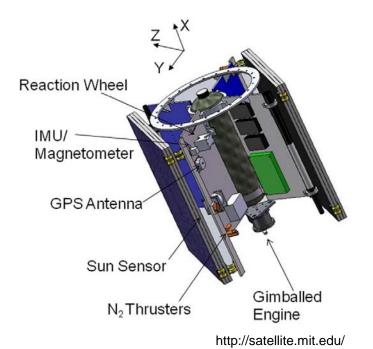
Euler parameters:

rameters. 
$$\dot{\eta}=-rac{1}{2}m{\epsilon}^{ op}m{\omega}_{ib}^{b}$$
  $\dot{m{\epsilon}}=rac{1}{2}\left(\eta \mathbf{I}+m{\epsilon}^{ imes}
ight)m{\omega}_{ib}^{b}$ 

1D: 
$$J\dot{\omega}=T$$
  
3D:  $\mathbf{M}_{b/c}^b\dot{\boldsymbol{\omega}}_{ib}^b+\left(\boldsymbol{\omega}_{ib}^b\right)^{ imes}\mathbf{M}_{b/c}^b\boldsymbol{\omega}_{ib}^b=\mathbf{T}_{bc}^b$ 

### Satellite attitude dynamics





$$\vec{F}_{bc} = m\vec{a}_c$$

$$\vec{T}_{bc} = \vec{M}_{b/c} \cdot \vec{\alpha}_{ib} + \vec{\omega}_{ib} \times \left( \vec{M}_{b/c} \cdot \vec{\omega}_{ib} \right)$$

### Airplane EoM (from book about airplane dynamics)

$$X - mgS_{\theta} = m(\dot{u} + qw - rv)$$

$$Y + mgC_{\theta}S_{\Phi} = m(\dot{v} + ru - pw)$$

$$Z + mgC_{\theta}C_{\Phi} = m(\dot{w} + pv - qu)$$

$$m\left(\mathbf{\dot{v}}_{c}^{b}+\left(\boldsymbol{\omega}_{ib}^{b}\right)^{\times}\mathbf{v}_{c}^{b}\right)=\mathbf{F}_{bc}^{b}$$

$$L = I_{x}\dot{p} - I_{xz}\dot{r} + qr(I_{z} - I_{y}) - I_{xz}pq$$

$$M = I_{y}\dot{q} + rp(I_{x} - I_{z}) + I_{xz}(p^{2} - r^{2})$$

$$N = -I_{xz}\dot{p} + I_{z}\dot{r} + pq(I_{y} - I_{x}) + I_{xz}qr$$

$$\mathbf{M}_{b/c}^b \dot{oldsymbol{\omega}}_{ib}^b + \left(oldsymbol{\omega}_{ib}^b
ight)^ imes \mathbf{M}_{b/c}^b oldsymbol{\omega}_{ib}^b = \mathbf{T}_{bc}^b$$

$$p = \Phi - \dot{\psi}S_{\theta}$$

$$q = \dot{\theta}C_{\Phi} + \dot{\psi}C_{\theta}S_{\Phi}$$

$$r = \dot{\psi}C_{\theta}C_{\Phi} - \dot{\theta}S_{\Phi}$$

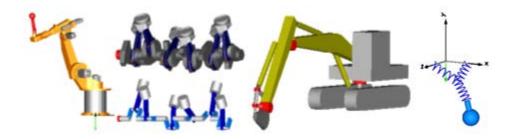
$$\dot{\theta} = qC_{\Phi} - rS_{\Phi} 
\dot{\Phi} = p + qS_{\Phi}T_{\theta} + rC_{\Phi}T_{\theta} 
\dot{\psi} = (qS_{\Phi} + rC_{\Phi})\sec\theta$$

$$oldsymbol{\dot{\phi}} = \mathbf{E}_d^{-1}(oldsymbol{\phi}) oldsymbol{\omega}_{ib}^b$$

Velocity of aircraft in the fixed frame in terms of Euler angles and body velocity components

$$\begin{bmatrix} \frac{dx}{dt} \\ \frac{dy}{dt} \\ \frac{dz}{dt} \end{bmatrix} = \begin{bmatrix} C_{\theta}C_{\psi} & S_{\Phi}S_{\theta}C_{\psi} - C_{\Phi}S_{\psi} & C_{\Phi}S_{\theta}C_{\psi} + S_{\Phi}S_{\psi} \\ C_{\theta}S_{\psi} & S_{\Phi}S_{\theta}S_{\psi} + C_{\Phi}C_{\psi} & C_{\Phi}S_{\theta}S_{\psi} - S_{\Phi}C_{\psi} \\ -S_{\theta} & S_{\Phi}C_{\theta} & C_{\Phi}C_{\theta} \end{bmatrix} \begin{bmatrix} u \\ v \\ w \end{bmatrix}$$

$$\mathbf{\dot{r}}_c^i = \mathbf{v}_c^i = \mathbf{R}_b^i \mathbf{v}_c^b$$



# Modelica Multibody introduction

Adapted from slides by Andreas Heckmann, DLR

### Modelica Multibody: Orientation

Orientation and position of coordinate systems (frames)

```
\vec{i}_3 \vec{v}_{i_1} \vec{b}_3 \vec{b}_2 \vec{b}_1 Body frame
```

```
model ...
import Modelica.Mechanics.MultiBody.Frames;
Frames.Orientation Rib;
Real[3] ui "vector u resolved in frame i";
Real[3] ub "vector u resolved in frame b";
...
equation
...
ui = Frames.resolve1(Rib, ub); // ui = Rib*ub
ub = Frames.resolve2(Rib, ui); // ub = Rib'*ui
```

World frame

- Orientation object  $\mathbf{R}_b^i$ 
  - Describes orientation of system b wrt i (transforms from b to i)
  - Contains:

```
Real T[3, 3] "Transformation matrix from world frame to local frame";
SI.AngularVelocity w[3]

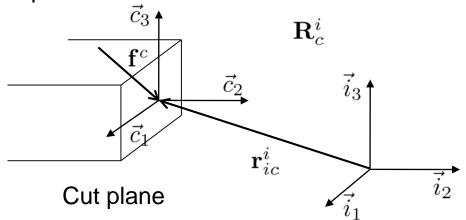
"Absolute angular velocity of local frame, resolved in local frame";
```

- Can be specified using Euler angles or Euler parameters/quaternions
- Many functions to operate on orientation objects

orientationConstraint angular Velocity 1 f angular Velocity 2 f resolve1 f resolve2 f resolveRelative f)resolveDyade1 f )resolveDyade2 f )nullRotation inverseRotation relativeRotation absoluteRotation planarRotationAngle axisRotation f axesRotations (f )axesRotationsAngles f)smallRotation f from nxv (f)from\_nxz (f)from\_T (f)from\_T2 (f)from\_T\_inv · (f ) from\_Q (f)to\_T (f)to\_T\_inv · (f) to\_Q f to\_vector (f)to\_exy (f)axis Quaternions TransformationMatrices Internal

### Modelica Multibody: Connectors I

- Connectors: To connect different rigid bodies
  - Position is resolved in world frame
  - Forces and torques are resolved in local frame

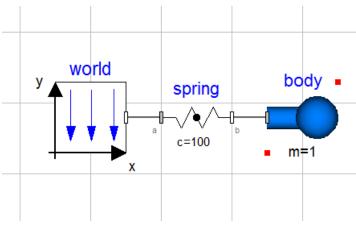


"No flow" variables

```
connector Frame
  "Coordinate system fixed to the component with one cut-force and cut-torque (no icon)"
  import SI = Modelica.SIunits;
SI.Position r_0[3]
    "Position vector from world frame to the connector frame origin, resolved in world frame";
Frames.Orientation R
    "Orientation object to rotate the world frame into the connector frame";
Ilow SI.Force f[3] "Sut-force resolved in connector frame";
flow SI.Torque t[3] "Cut-torque resolved in connector frame";
end Frame;
"Flow" variables
```

### Modelica Multibody: Connectors II

```
model SpringMass
inner Modelica.Mechanics.MultiBody.World world;
Modelica.Mechanics.MultiBody.Parts.Body body(
    m=1,
    r_CM={0,1,0}, // In frame a
    r_0(fixed=true, start={0,0.5,0})); // In world frame
Modelica.Mechanics.MultiBody.Forces.Spring spring(c=100);
equation
    connect(spring.frame_a, world.frame_b);
    connect(spring.frame_b, body.frame_a);
end SpringMass;
```

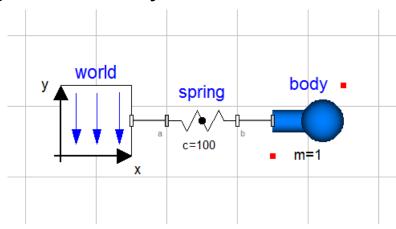


#### Connection rules

- Non-flow variables set equal (that is: frames coincides)
- Flow variables sum to zero (Newton's third law)

#### Modelica.Multibody: Generic body component

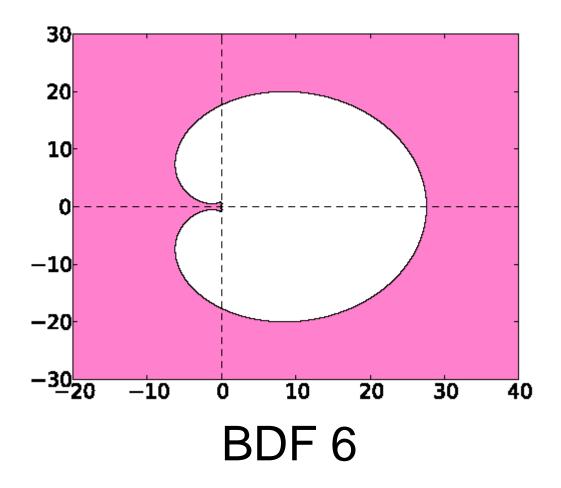
Make SpringMass in Dymola



#### Show

- Parameters (mass,  $r_cm = (0,-0.5,0)$ , Inertia matrix)
- Initial values
- Euler angles

# Stability Region BDF

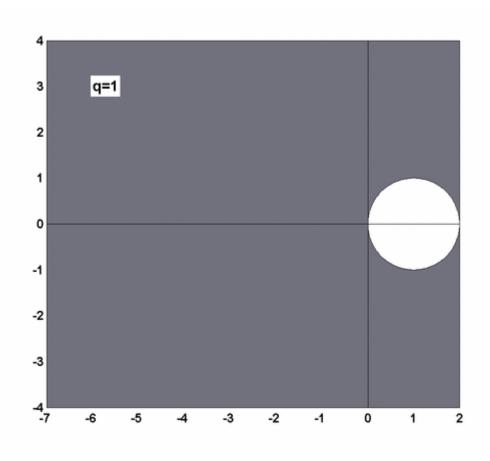


### Padé approximations to es

m	0	1	2	3
0	$\frac{1}{1}$	$\frac{1+s}{1}$	$\frac{1+s+\frac{1}{2}s^2}{1}$	$\frac{1+s+\frac{1}{2}s^2+\frac{1}{6}s^3}{1}$
1	$\frac{1}{1-s}$	$\frac{1+\frac{1}{2}s}{1-\frac{1}{2}s}$	$\frac{1 + \frac{2}{3}s + \frac{1}{6}s^2}{1 - \frac{1}{3}s}$	$\frac{1 + \frac{3}{4}s + \frac{1}{4}s^2 + \frac{1}{24}s^3}{1 - \frac{1}{4}s}$
2	$\frac{1}{1-s+\frac{1}{2}s^2}$	$\frac{1 + \frac{1}{3}s}{1 - \frac{2}{3}s + \frac{1}{6}s^2}$	$\frac{1 + \frac{1}{2}s + \frac{1}{12}s^2}{1 - \frac{1}{2}s + \frac{1}{12}s^2}$	$\frac{1 + \frac{3}{5}s + \frac{3}{20}s^2 + \frac{1}{60}s^3}{1 - \frac{2}{5}s + \frac{1}{20}s^2}$
3	$\frac{1}{1-s+\frac{1}{2}s^2-\frac{1}{6}s^3}$	$\frac{1 + \frac{1}{4}s}{1 - \frac{3}{4}s + \frac{1}{4}s^2 - \frac{1}{24}s^3}$	$\frac{1 + \frac{2}{5}s + \frac{1}{20}s^2}{1 - \frac{3}{5}s + \frac{3}{20}s^2 - \frac{1}{60}s^3}$	$\frac{1 + \frac{1}{2}s + \frac{1}{10}s^2 + \frac{1}{120}s^3}{1 - \frac{1}{2}s + \frac{1}{10}s^2 - \frac{1}{120}s^3}$
		L-stab	le L-stab	le A-st

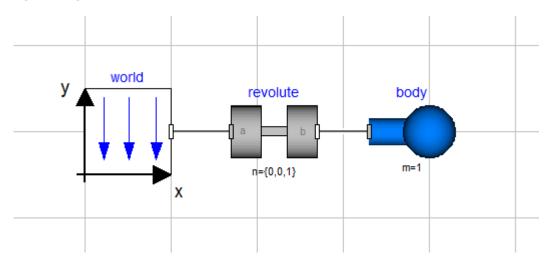
- m = 0: Explicit Runge-Kutta methods with  $p = \sigma$
- m = k: Gauss, Lobatto IIIA/IIIB (incl. implicit mid-point, trapezoidal)
- m = k+1: Radau-methods (incl. implicit Euler)
- m = k+2: Lobatto IIIC

# Stability region for Adams-Moulton



# Modelica.Multibody: Rotations

Make simple pendulum



Show body.frame\_a.R (rotation object)

### Modelica Multibody: Kinematics

- Equations inside the component provide relations between the connector variables on position level
- Example: MultiBody.Parts.FixedTranslation
  - Fixed translation of frame\_b with respect to frame\_a

```
fixedTranslation

a
r={1.1,0.3,2.1}
```

```
model FixedTranslation
   "Fixed translation of frame_b with respect to frame_a"
   ...
equation

frame_b.r_0 = frame_a.r_0 + Frames.resolve1(frame_a.R, r);
frame_b.R = frame_a.R;

/* Force and torque balance */
zeros(3) = frame_a.f + frame_b.f;
zeros(3) = frame_a.t + frame_b.t + cross(r, frame_b.f);
end FixedTranslation;
```

Dymola differentiates these equations twice for (velocity and) accelerations

### Modelica Multibody: Kinetics

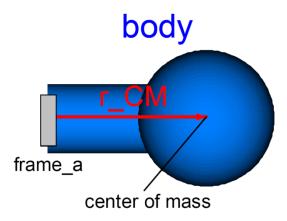
- Newton-Euler equations
  - Accelerations

$$\vec{a}_p = \vec{a}_o + \vec{\alpha}_{ib} \times \vec{r} + \vec{\omega}_{ib} \times (\vec{\omega}_{ib} \times \vec{r}), \ \vec{r} \text{ fixed.}$$

Kinetics

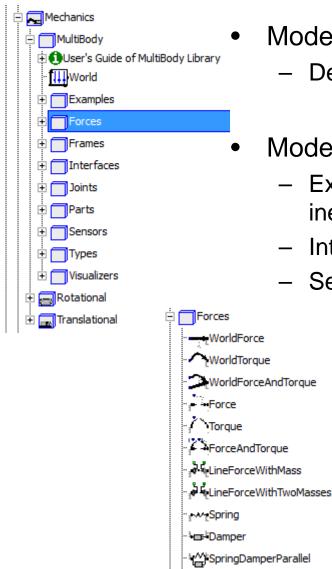
$$\vec{F}_{bc} = m\vec{a}_c$$

$$\vec{T}_{bc} = \vec{M}_{b/c} \cdot \vec{\alpha}_{ib} + \vec{\omega}_{ib} \times (\vec{M}_{b/c} \cdot \vec{r})$$



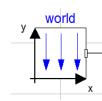
```
model body
  "Rigid body with mass, inertia tensor and one frame connector (12 potential states)"
equation
  // translational kinematic differential equations
 v 0 = der(frame a.r 0);
                                            // r 0, v 0 resolved in world frame
  a a = Frames.resolve2(frame a.R,der(v 0)); // a a resolved in frame a
  // rotational kinematic differential equations
  w_a = Frames.angularVelocity2(frame_a.R);
  z a = der(w a);
  // Newton/Euler equations with respect to center of mass
            a_CM = a_a + cross(z_a, r_CM) + cross(w_a, cross(w_a, r_CM));
            f_CM = m*(a_CM - g_a);
            t CM = I*z a + cross(w a, I*w a);
       frame a.f = f CM
       frame a.t = t CM + cross(r CM, f CM);
end body;
```

### Modelica Multibody: Elementary components I



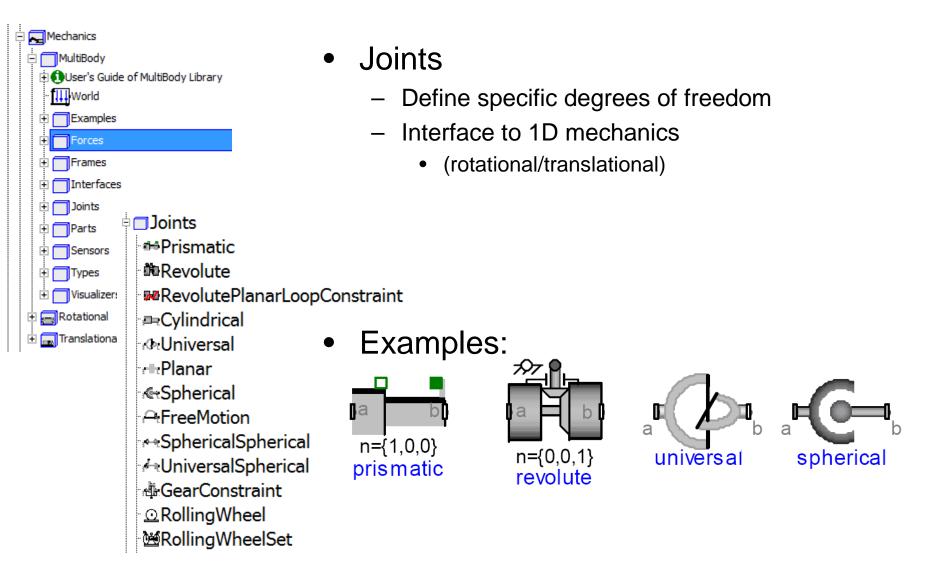
SpringDamperSeries

- Modelica.Mechanics.Multibody.World
  - Defines inertial frame, gravity, animation defaults

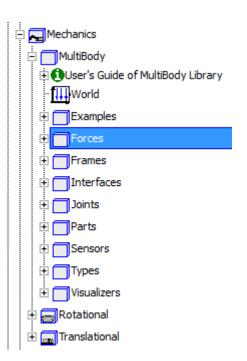


- Modelica.Mechanics.MultiBody.Forces
  - External forces and torques, resolved in body- or inertial frame
  - Interface to Real input functions
  - Several spring/damper configurations

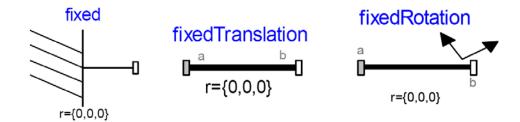
### Modelica Multibody: Elementary components II



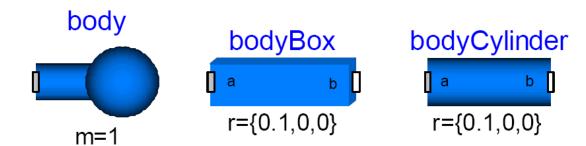
### Modelica Multibody: Elementary components II



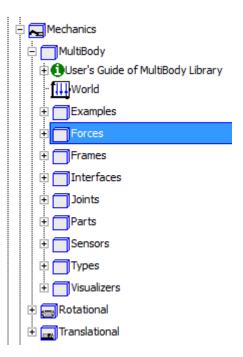
- Modelica.Mechanics.MultiBody.Parts
  - Fixed, Fixed Translation and Fixed Rotation



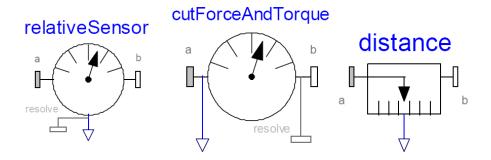
 Rigid bodies with predefined geometric shapes



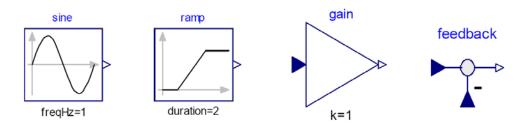
### Modelica Multibody: Elementary components II



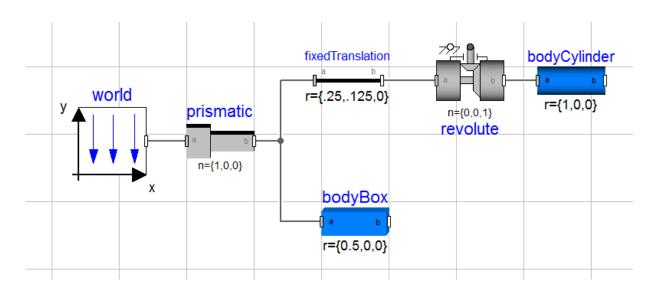
- Modelica.Mechanics.Multibody.Sensors
  - For control and validation purposes



 Modelica.Blocks.Sources + Modelica.Blocks.Math

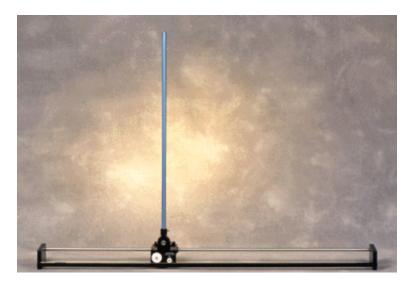


### Example: Inverted pendulum, modeling



- Box: 0.5m x 0.25m x 0.25m
- Cylinder: L = 1m, r = 0.05m





### Example: Inverted pendulum, PD control

