

$$1. \Delta \dot{p} = \frac{1}{I_x} \Delta L_c = \frac{1}{I_x} (-k_1 \Delta p - k_2 \Delta \phi) \quad \text{Lateral}$$

$$\Delta \dot{\phi} = \Delta p$$

$$\dot{y} = F(y, t)$$

$$\dot{y} = \begin{bmatrix} \Delta \dot{p} \\ \Delta \dot{\phi} \end{bmatrix} = \underbrace{\begin{bmatrix} -\frac{k_1}{I_x} & -\frac{k_2}{I_x} \\ 1 & 0 \end{bmatrix}}_A \underbrace{\begin{bmatrix} \Delta p \\ \Delta \phi \end{bmatrix}}_y$$

$$\dot{y} = A \cdot y$$

$$(A - \lambda I) y^* = 0$$

$$\det(A - \lambda I) = 0$$

$$\det(A - \lambda I) = \begin{vmatrix} -\frac{k_1}{I_x} - \lambda & -\frac{k_2}{I_x} \\ 1 & -\lambda \end{vmatrix} = \lambda^2 + \frac{k_1}{I_x} \lambda + \frac{k_2}{I_x} = 0$$

$$\lambda^2 + 2\zeta \omega_n \lambda + \omega_n^2 = 0$$

$$\omega_n = \sqrt{\frac{k_2}{I_x}}$$

$$\zeta = \frac{k_1}{I_x} \cdot \frac{1}{2\sqrt{\frac{k_2}{I_x}}}$$

$$\zeta = \frac{k_1}{2\sqrt{I_x k_2}}$$

$$\lambda = -\zeta \omega_n \pm \sqrt{(\zeta \omega_n)^2 - \omega_n^2}$$

$$\zeta = \frac{1}{2\zeta \omega_n + \sqrt{(\zeta \omega_n)^2 - \omega_n^2}} = \frac{1}{2}$$

$$\lambda_1 = -\zeta \omega_n + \sqrt{\omega_n^2 (\zeta^2 - 1)}$$

$$\lambda_2 = -\zeta \omega_n - \sqrt{\omega_n^2 (\zeta^2 - 1)}$$

$$\omega_n = \frac{2}{3 + \sqrt{3^2 - 1}}$$

$$\lambda_1 = 10 \cdot \lambda_2$$

$$-\zeta \omega_n + \sqrt{\omega_n^2 (\zeta^2 - 1)} = 10(-\zeta \omega_n - \sqrt{\omega_n^2 (\zeta^2 - 1)})$$

$$9\zeta \omega_n + 11\sqrt{\omega_n^2 (\zeta^2 - 1)} = 0$$

$$9\zeta \omega_n = -11\sqrt{\omega_n^2 (\zeta^2 - 1)}$$

$$\frac{81}{121} \zeta^2 \omega_n^2 = \omega_n^2 \zeta^2 + \omega_n^2$$

$$\omega_n = \frac{2}{3 - \sqrt{3^2 - 1}}$$

$$\omega_n = 6.3241$$

$$\frac{81}{121} \zeta^2 = \zeta^2 + 1$$

$$-\zeta^2 \cdot \frac{40}{121} = 1$$

$$\zeta = \sqrt{\frac{121}{40}}$$

$$\zeta = 1.739$$

$$K_2 = \omega_n^2 I_x = (6.324)^2 \cdot 5.8 \times 10^{-5} \quad I_x = 5.8 \cdot 10^{-5} \text{ kg m}^2$$

$$K_2 = 0.00232$$

$$K_1 = \zeta \cdot 2 \sqrt{I_x \cdot K_2} = 1.734 \cdot 2 \sqrt{5.8 \times 10^{-5} \cdot 0.00232}$$

$$K_1 = 0.001276$$

longitudinal

$$\text{Since } 1^2 + \frac{I_x}{I_y} u + \frac{K_y}{I_y}$$

is the same equation as the lateral
we can use the same final derivations

$$\zeta = 1.734$$

$$\omega_n = 6.324$$

$$K_2 = (6.324)^2 \cdot 7.2 \times 10^{-5} = 0.00288$$

$$K_1 = \zeta \cdot 2 \sqrt{I_y \cdot K_2} = 1.734 \cdot 2 \sqrt{7.2 \times 10^{-5} \cdot 0.00288} = 0.001584$$

$$K_3 = 0.001584$$

$$K_4 = 0.00288$$

2. The results make sense since $-K_1 P_1$ will be countering $-K_2 \phi$ so the settling time is longer than just the proportional. Steady hover is not a steady flight condition since it doesn't recover to steady hover conditions.

3. The two models have nearly identical results.

4. Like in the simulations the copter continues to in the x and y directions while maintaining altitude. At about the 6 second mark you can see the command was given to change theta to 5° , and after that point the graph looks similar to problems 2b and 3b.