



- 1) \vec{r} is the vector from O to the cg of the aircraft
 \vec{r}_E is the vector from O to the cg in E frame coordinates
 \vec{r}_E would be represented as $\begin{bmatrix} 100 \\ 0 \\ 0 \end{bmatrix} \text{m}$
 \vec{r}_B is the vector from O to the cg in B frame coordinates
 \vec{r}_B would be represented as $\begin{bmatrix} 0 \\ -100 \cos \phi \\ 100 \sin \phi \end{bmatrix} \text{m}$

- 2) $\vec{v}^E = \frac{d^E}{dt} \vec{r}$ says the inertial velocity^{vector} is equal to the E frame derivative of \vec{r}

\vec{v}_E^E is the inertial velocity in E frame coordinates represented as $\vec{v}_E^E = \begin{bmatrix} 0 & 10 & 0 \end{bmatrix}^T \frac{\text{m}}{\text{s}}$

\vec{v}_B^E is the inertial velocity in B frame coordinates represented as $\vec{v}_B^E = \begin{bmatrix} 10 & 0 & 0 \end{bmatrix}^T \frac{\text{m}}{\text{s}}$

- 3) $\vec{\omega}^{EB}$ is the angular velocity of the B frame as seen from E

$\vec{\omega}_E^{EB}$ is $\vec{\omega}^{EB}$ represented in E frame coordinates as $\begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \frac{\text{rad}}{\text{s}}$

$\vec{\omega}_B^{EB}$ is $\vec{\omega}^{EB}$ represented in B frame coordinates as $\begin{bmatrix} 0 \\ -\sin \phi \\ \cos \phi \end{bmatrix} \frac{\text{rad}}{\text{s}}$

4) $\frac{d^B \vec{r}}{dt}$ is the B frame derivative of \vec{r}

$\left(\frac{d^B \vec{r}}{dt}\right)_E$ is the B frame derivative of \vec{r} in E frame coordinates represented as $[0 \ 0 \ 0]^T \left[\frac{m}{s}\right]$

$\left(\frac{d^B \vec{r}}{dt}\right)_B$ is the B frame derivative of \vec{r} in B frame coordinates represented as $[0 \ 0 \ 0]^T \left[\frac{m}{s}\right]$

5) $\left(\frac{d^E \vec{v}^E}{dt}\right)_E$ is the E frame derivative of the inertial velocity in E frame coordinates represented as $[-10 \ 0]^T \left[\frac{m}{s^2}\right]$

$\left(\frac{d^B \vec{v}^E}{dt}\right)_B$ is the B frame derivative of the inertial velocity in B frame coordinates represented as $[0 \ 0 \ 0]^T \left[\frac{m}{s^2}\right]$

$$6) \frac{d^E \vec{v}^E}{dt} = \frac{d^B \vec{v}^E}{dt} + \vec{\omega}^{EB} \times \vec{v}^E \quad \begin{vmatrix} \hat{N} & \hat{E} & \hat{D} \\ 0 & 0 & 1 \\ 0 & 10 & 0 \end{vmatrix} = -\hat{N} \left[\frac{m}{s^2}\right]$$

$$\frac{d^E \vec{v}^E}{dt} = -\hat{N} \left[\frac{m}{s^2}\right]$$

7) $\vec{f} = m\vec{a}$ which is a vector

$$\vec{f}_E = m \left(\frac{d^E \vec{v}^E}{dt}\right)_E \text{ which is } [-m \ 0 \ 0]^T [N]$$

$$\vec{f}_B = m \left(\frac{d^E \vec{v}^E}{dt}\right)_B \text{ which is } \begin{bmatrix} 0 \\ m \cos \phi \\ m \sin \phi \end{bmatrix} [N]$$

$$8) \vec{v}^E = \vec{\omega} + \vec{v}$$

$$\vec{v}^E = 10\hat{E} \left[\frac{m}{s}\right] \quad \vec{\omega} = 2\hat{N} + 3\hat{E} - 1\hat{D} \left[\frac{m}{s^2}\right]$$

$$\vec{v} = \vec{v}^E - \vec{\omega} = -2\hat{N} + 7\hat{E} + 1\hat{D} \left[\frac{m}{s}\right]$$

$$\boxed{\vec{v} = -2\hat{N} + 7\hat{E} + 1\hat{D} \left[\frac{m}{s}\right]}$$

$$\vec{v}_B = [-2 \quad -2 \cdot (-\cos \phi + \sin \phi) \quad -2 \cdot \sin \phi + \cos \phi]$$

$$\boxed{\vec{v}_B = [2 \quad 2 \cos \phi + \sin \phi \quad 2 \sin \phi + \cos \phi]}$$