

**Alta Avionics, LLC**

**DETAILED PROCEDURES MANUAL**

**(DPM)**

**CRS# JN1R0210**

**1887 South 1800 West**

**Woods Cross, UT 84087**

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# Introduction

This Detailed Procedures Manual has been prepared in accordance with the current Federal Aviation Regulations (FAR's), and the policies of Alta Avionics, LLC.

This manual describes the various procedures practiced by Alta Avionics, LLC in its daily operations in detail.

The general repair, overhaul, or alteration of products will be performed in accordance with the current Federal Aviation Regulations, manufacturer's data, drawings, specifications, and bulletins, or other technical data approved by the administrator.

Each supervisor and inspector working for Alta Avionics, LLC will have access to a current copy of this manual, Alta Avionics, LLC. It will also be available to all repair station personnel. All personnel are required to understand its contents.

This Detailed Procedures Manual requires only Alta Avionics, LLC approval.

This manual describes general policy and procedures that have previously been FAA and Repair Station approved or accepted in the RSM, QCM, Forms, and Training Manual.

# Manual Control

This manual will be maintained by the FAA Coordinator and will provide a hard copy OR electronic Portable Document Format (PDF) for manual access for all personnel. If an electronic manual is requested a PDF version of this manual will be by provided by link for electronic manual access for any personnel.

Any digital version of this manual will be in PDF and archived in an organized manner easily retrievable for historical record and shall be done so in such a manner so as not to interfere with the most current version of this manual. In the event of hardware failure, a hardcopy, or digital file (remote or local) will be used to restore data.

The General Manager will be notified by a department supervisor in the event this manual is not current, and valid for that department’s use, and will identify needed changes using form A-MCR (Manual Change Request). A sample of this form is found in the Forms Manual. The General Manager will have the revisions found necessary, produced in a final form.

The FAA Coordinator will revise manuals as required, and explain the revisions to all employees. An entry into each employee’s training record will be added after each employee has been trained to verify and acknowledge the understating of each revision. Upon approval by an authorized Repair Station representative, the repair station will commence operating within the guidelines of the new revision.

The “List of Effective Pages” will reflect the Approval/Acceptance of the current revision. A file will be maintained, showing on a continuous basis, the disposition of each manual change. Revised areas within the document will be identified by a vertical bar in the margin.

Revisions found “not acceptable” to the FAA/CHDO, which do not conform to applicable regulations, will be addressed by this repair station as a top priority. The identified procedure or action will cease, and acceptable changes implemented immediately. The maintenance/administrative actions that were performed under revisions found “not acceptable” by the FAA/CHDO will be addressed in the following order:

1. Safety of Flight: Aircraft operator to be notified immediately, and advised that aircraft is to remain on the ground until this repair station can correct the problem, or coordinate with another certified repair station to correct the problem.
2. Procedure/Record Keeping: Aircraft operator to be notified immediately, and advised of the problem. The operator will have the option to operate the aircraft until the problem can be corrected.
3. Problems that do not affect aircraft and/or appliances will be dealt with internally and immediately to correct them.

## Additional Fixed Locations

All additional fixed locations under Alta Avionics, LLC will also be supplied with a direct link to this manual on every computer terminal. Each employee will be trained on the procedure to access all the manuals during their initial training. A stored copy will be supplied to each facility in case of computer failure.

# Service Department Repair Procedures

## Work Order Initiation/Repair Preparation Procedures

Alta Avionics, LLC. representative will obtain proper information, aircraft scheduling, expense limits for complaints, and open work order listing complaints.

The Supervisor will acquire knowledge of the new Work order pertaining to the complaints being serviced.

1. Will assign work order to a qualified technician.

The technician will obtain the proper technical information, any expense limits, if they apply, and parts needed for repair.

1. Acquire all needed test equipment and verify the calibration currency of tester to be used.
2. Obtain location of aircraft, and acquire keys, combinations, keycodes, if needed for access.

## In-Aircraft Troubleshooting/Repair Procedures

1. Determine aircraft operating voltage.
2. Perform initial system functional tests for verification of complaints.
3. If problem cannot be verified, consult with supervisor for any further action.
4. If expense limits are going to be exceeded, contact supervisor to acquire authorization from customer to exceed repair limit.
5. If defective articles are found, remove articles and refer to procedures in Section 8 Bench Appliance Repair Procedures.
6. Complete repairs in aircraft; complying with manufacturers, and approved repair procedures. In Progress inspections must be performed when any aircraft re-assembly is required. Complete in-aircraft repairs as required.
7. Upon completion of repairs, perform full tool and test equipment inventory check. If any items are not located, inform supervisor of situation. The items must be located before proceeding.
8. Perform full functional test of repaired equipment, or systems before requesting Return-To-Service test/inspection.
9. Contact appropriate inspector for Return-To-Service test (if deemed necessary), Assess results, and assist as required.
10. Return to Alta Avionics, LLC and complete all necessary paperwork to approve aircraft for return to service.
11. Inform supervisor of completion of repairs, and receive next assignment.

**NOTE:** Samples of forms are found in the Forms Manual.

# Installation Procedures

## Pre-Aircraft Arrival Procedures

1. Open Work Order with reference to customer authorized Installation proposal.
2. Review proposal, and confirm that equipment to be installed is correct (voltage requirements, face color, compatibility, placement, etc.).
3. Verify equipment installation kits are complete, and acquire, or order (if not in stock) any required parts.
4. Gather appropriate installation data; i.e., manufacturers Installation instructions, aircraft wiring diagrams, DER engineering prints and 8110-3, etc., as required.
5. Initiate Installation Inspection Checklist.

## Preliminary Inspection Procedures

1. Lead Technician shall be responsible for Pre-Installation Inspection of aircraft and equipment as received. List any discrepancies noted and inform Customer, as required. Sign Preliminary Inspection block on A-WO. Generate additional travelers, as required, for items on installation Work Order.
2. Acquire Aircraft Registration, current Weight and Balance, and Equipment List. Make copies of each, and return originals to aircraft.

## In-Progress Installation/Inspection Procedures

1. Installation In-Progress Inspections will be continuous during the installation process. The number of In-Progress inspections/signoffs will be determined by the number and complexity of items addressed in the installation Work Order.
2. Begin FAA Form 337 paperwork.
3. Photograph, or sketch existing instrument panels and radio stacks. Verify new equipment placement in panel.
4. Pre-fabricate and test wiring harness on bench as much as possible for new equipment.
5. Open up working areas in aircraft (panels, floor boards, etc.), and inspect affected locations for conflicts in mounting equipment/parts or harness routing.
6. Complete FAA Form 337 and submit to FSDO.
7. Wire new equipment harnesses in aircraft.
8. Install proper circuit protection for new equipment and placard appropriately.
9. Perform continuity checks of all new wiring/interfacing before applying power to aircraft.
10. Perform necessary tests of equipment prior to tie-up.
11. Reassemble working area, and close access panels only after inspector's examination, and successful functional test of all affected systems.
12. Verify all work has been completed per FAA Form 337, and/or STC, and that any necessary placards have been properly applied.

## Final Inspection/Sign Off

1. Update Weight and Balance and Equipment List.
2. Ready aircraft for final inspection by Lead Technician.
3. Complete appropriate documentation, and attach to aircraft records.
4. Sign off by authorized Inspector. Approve aircraft for Return to Service.

\*Samples of forms are found in the Forms Manual.

# Bench Appliance Repair Procedures

## Incoming Equipment/Appliances

1. Hand Carry
   1. Open Work Order, and/or initiate a Work Order A-WO fill the appropriate information in.
   2. An Article Tag (Form AT-AT) will be attached to the appliance. Appliance will then be enclosed in anti-static bag or other appropriate packaging along with Work Order.
   3. Article will then be placed on the repairable shelf to await repair action.
2. Technician Carry-In
   1. Open a Work Order, with the required information filled in.
   2. An Article Tag (Form AT-AT) will be attached to the appliance. Appliance will then be enclosed in antistatic bag, or other appropriate packaging.
   3. Appliance will then be placed on the repairable shelf to await repair action.

## Preliminary Inspection

1. The technician will visually inspect appliance for damage, noting any damage in appropriate section of Work Order, and initialing Preliminary Inspection block.
   1. Initialing preliminary inspection on work order may be accomplished by any means approved in the Alta Avionics, LLC. Repair Station Manual (RSM)
2. The technician will determine if article is repairable at this location based on:
   1. Available technical repair data.
   2. Required test equipment with current calibration status.
      1. If Technician determines that equivalent test equipment will be used, repair technician will document in “Test Equipment Equivalency Log” (see “Test Equipment Equivalency Procedure” in this document).
3. Repairs will be initiated as required, or article will be sent out for repair at an approved facility.

## In-Progress Inspection and Repair Phase

1. Bench In-Progress Inspections will be conducted, and initialed on the Work Order during the repair cycle as required.
2. The technician will note and correct any hidden defects found during troubleshooting; i.e., missing or damaged hardware, incorrect parts, poor workmanship, etc. Defects will be noted on the Work Order in the appropriate area.
3. Appliances requiring parts shall be placed in a bin/container with all associated hardware and sub-assemblies. Container shall be placed in designated Awaiting Parts holding area. The Work Order shall be completed up to that point, placed with the article/appliance, and a Parts Request submitted to the Parts Department to expedite as required.
4. All parts used in the repair, including serialized parts, will be recorded on the Work Order.
5. After repairs have been made, another In-Progress inspection will be conducted and initialed by an authorized Inspector. The article/appliance will then be closed for Final Inspection.

## Final Inspection and Return to Service

1. Once the article/appliance has reached final assembly, it will be subjected to a functional test, and visual inspection.
2. If the final tests were completed satisfactorily, the Final Inspection block will, be initialed by the authorized Inspector.
3. The Work Order will be reviewed for completeness, and signed and dated by the repair technician, and authorized Inspector in the required blocks.
4. An FAA Form 8130-3 will be completed for articles/appliances passing Final Inspection.
5. A copy of the Work Order will be made and attached to the completed FAA Form 8130-3 and function as a repair report. Both forms will be attached to the article.
6. A copy of the FAA Form 8130-3 will be made and attached to the Work Order as a record of the completed FAA Form 8130-3, and scanned into the Work Order Electronic Document Storage (EDS).

## Completion Phase

1. The Appliance will be placed in an anti-static bag, or other appropriate packaging as required (see ESDS DEVICES), with the FAA Form 8130-3, and a copy of the Work Order.
2. Appliance will then be placed into the serviceable articles/equipment locker pending reinstallation, pick-up or shipment.
3. The original Work Order will be checked for completeness and attached to the copy of the FAA Form 8130-3, along with any other appropriate documentation. All documents related to the repair and transactions associated with the work order will be scanned or stored in the Work Order Electronic Document Storage file (EDS).
4. The work area will be cleared and readied for the next repair.
5. Samples of forms are found in the Forms Manual.

# Receiving Procedures

Form A-RPI (Receiving Part Inspection) will be used by the receiving parts inspector to support the part inspection as well as provide a record of received parts. Where appropriate, a copy of the A-RPI form will be attached to the work order record.



Figure 1 Receiving Parts Inspection Form

1. Upon receipt of part/article, the Parts Inspector will examine packaging for damage that might affect the integrity of item received.
   1. Damage (if any) will be noted on packing slip.
   2. Item will be unpacked, noting any visible damage to the item.

**Special attention will be paid to shock indicators.**

* 1. If item is questionable as to condition, it will be tagged and placed in a quarantine bin until an inspector can review its condition.
  2. When a part is determined to be damaged, a claim will be filed, and the part returned to vendor for repair or replacement.

1. Parts received in serviceable/like new condition.
   1. Part numbers and vendor will be verified.
   2. Quantities will be verified.
   3. Serialized articles will be verified and appropriately noted.
2. Parts received for a specific repair will be forwarded to appropriate technician, and entered on the work order in the work order parts section.
3. All documentation will be scanned/copied and stored with the work order.
4. Parts for stock will be forwarded to stock room area, where they will be filed by vendor, P/N, quantity, and lot number and entered into the parts log/database.
5. All incoming parts and materials will be inspected for conformity to purchase orders, and authenticity of new replacement parts. Such parts will be checked for shipping damage, corrosion, rust, or other deterioration. All incoming Parts and Materials orders will be scrutinized to assure that only approved parts are utilized. The procedures detailed herein, AC21- 29B, and any other resources, will be used in making this determination. If any parts are detected, or suspected to be unapproved, the parts will be specially marked and segregated. The parts will be placed in a special holding area designated by the Quality Assurance Manager. An FAA form 8120-11 will be filled out and sent to both the FAA branch that oversees unapproved parts (AVR-20), and to this repair station's assigned inspector at the local FAA Flight Standards District Office (FSDO).

# Stock Control, Segregation, And Identification

1. The system of stock control, segregation, and identification utilized is described herein to enable personnel to determine the adequacy of the stock, the location of parts, the proper identification of parts, and to assure that parts do not deteriorate, or become contaminated with foreign matter prior to use. These Shelf Life items will be so arranged that the items with the greatest shelf life will be issued first. The more recently procured items will be placed in the aft portion of the storage bin. These parts will be dated to ensure that the shelf life is not exceeded.
2. All parts and materials utilized at Alta Avionics, LLC shall be classified as to TYPE as described herein. Following are detailed procedures for disposition of these parts.

## Type "A" Parts and Materials: Piece parts and material (non-serialized)

1. The incoming parts inspector will physically inspect parts for correct type, quantity, condition, and vendor, using the packing slip for verification. This information will be crosschecked against the purchase order.
2. Parts received into the system will be bagged and marked with the following data: Date, purchase order number, quantity, and vendor. Vendor marked bags will be acceptable if all data is present.
3. The parts bag will be placed into an envelope, which will also be marked with date, quantity, and vendor.
4. The envelopes will be placed in an appropriate bin that will then be placed in a location accessible to authorized personnel, and segregated from other parts and materials.
5. No parts bag will contain parts that were received on more than one date. Further shipments of specific parts will be recorded on the envelope, and placed in a separate bag in the envelope. The parts envelope may contain parts bags from more than one date.
6. When a part is removed from the bag, the new quantity will be recorded on the bag under the appropriate date and vendor. This data will also be recorded on the envelope that contained the parts bag and noted in the computer system.
7. Spools containing wire and cable will have a label affixed with the Mil spec #, P.O. # and Lot #. Wire shall be stocked and ordered as required by shop demands.

## Type "B" Parts and Materials:

1. Parts and Materials that have a return to service and/or certification; i.e.: modules, circuit cards. Type "B" Parts and Materials will typically have a serial number.
   1. Incoming inspection procedure for Type "B" Parts and Materials include the same requirements as noted above for Type "A" Parts and Materials. Additionally, the approval for Return to Service, and /or certification information will be included and kept with the part or material. This information will be added to the appropriate work order upon use.

## Type "C" Parts and Materials:

1. Aircraft parts that have not been returned to service, but are repairable. Parts and Materials not intended for use in aircraft.
   1. Type "C" Parts and Materials are segregated from new and serviceable parts in a manner that prevents accidental use.

## Type "D" Parts and Materials:

Expendable parts; window splices, terminals, screws, hardware, etc.

1. Type "D" Parts and Materials will be stocked as free stock in parts bins. New parts only will be stocked.

# Electro Static Discharge Sensitive (ESDS) Devices

## Introduction

Industry has become increasingly aware of the damage electrostatic discharge (ESD) can cause to Metal-Oxide Semiconductor (MOS) devices. Low production yields gave initial early evidence of this. More recently this same evidence has suggested similar ESD sensitivity in other parts; evidence strengthened through use, testing, and failure analysis. The tendency toward greater complexity and increased packaging density has heightened this sensitivity to the point where some state-of-the-art micro technology parts can be destroyed or damaged by static voltages as low as 20 volts.

Microelectronic and semiconductor devices, thick and thin film resistors, chips and hybrid devices, and piezoelectric crystals are all susceptible to common electrostatic voltage levels. All equipment, not having adequate protective circuits, containing these components are ESD sensitive. The human body, all work surfaces, floors (especially if waxed), furniture, personal clothing, clean room garments, packaging materials, and high velocity gas or liquid flow equipment are prime generators of electrostatic voltages. Movements such as sliding, rubbing, or separating of materials can frequently result in electrostatic voltages of 15,000 volts.

Maintenance shops absorb the majority of the expense associated with ESD failure. Latent failures reduce the mean time between repairs (MTBR). To support this maintenance activity, a large inventory of spares must be on hand. Proper ESDS handling will have substantial cost benefits.

ESDS information and procedures is provided in the following paragraphs

11.2 General Information

11.3 Definition of Terms

11.4 Static Safeguarded Work Station

11.5 Repair Tools and Supplies

11.6 Handling Procedures/Precautions

11.7 Transportation and Storage

11.8 Component Level

11.9 Assembly Level

11.10 Additional Precautions

11.11 Anti-static Device Testing

## General Information

The primary objective of all electrostatic prevention methods is to eliminate static charge accumulation. Any subassembly, assembly, or printed circuit board containing ESDS devices is considered electrostatic sensitive and should be handled according to the handling procedures called out in this section.

Top level assemblies or equipment that are fully assembled with all covers and shields in place a properly attached are not normally considered electrostatic sensitive. Follow any packaging or special handling procedures specified for the equipment.

Storage of articles should be in anti-static bags or better with anti-static covers on the article connectors.

## Definition of Terms

Anti-static Materials — Anti-static materials do not charge tribo-electrically and exhibit a surface resistivity between 109 to 1014 Ω per square. These materials are used to replace insulating and static generating materials and also may be used to line static shielding containers.

**Buried Layer Containers** — Containers (bags or tote bins) with a conductive layer placed between insulating or anti-static materials and thus not exposed to the outside. Static shielding is accomplished by this layer.

**Conductive Materials** — Conductive materials exhibit a surface resistivity of less than 10⁵to 10⁹ per square as measured with a surface resistivity meter. Static shielding containers are made from conductive materials.

**Electrostatic Charge** — An electrical charge at rest, caused by the transfer of electrons within a body or from one body to another.

**Electrostatic Discharge (ESD)** — A transfer of electrostatic energy between substances of different electrical potentials. The discharge may occur without direct contact.

**Electrostatic Discharge Sensitive (ESDS) Device** — Electronic devices that are susceptible to damage from electrostatic discharge. These items include all semiconductors, which use MOS, Complimentary Metal-Oxide-Semiconductor (CMOS), P-type Metal-Oxide-Semiconductor (PMOS), N-type Metal-Oxide-Semiconductor (N-MOS) and Gallium Arsenide (GaAs) technology, and other select electrical devices.

**Electrostatic Discharge Sensitive Assemblies** — Any assembly that contains an ESDS device is considered an ESDS assembly. Circuit cards, subassemblies, and modules internal to equipment are also included.

**Foot Strap** — A foot strap is a conductive device that attaches to the foot of an operator to ground the operator to a conductive floor surface. It is not necessary for the foot strap to contact bare skin.

**Static-Dissipative Materials** — Static-dissipative materials exhibit a surface resistivity of 105 to 109 per square as measured with a surface resistivity meter. These materials may be used in place of conductive materials to control the rate of electrical discharge and limit the possibility of sparking.

**Static Charge Generator** — This is a general term for nonconductive or insulating materials (e.g. Adhesive tape, untreated plastic foam, and most plastics). This type of material easily generates and holds a static charge and is a potential hazard to ESDS devices/assemblies.

**Surface Resistivity** — Surface resistivity is a value that indicates the ability of a material to dissipate electrical charges.

**Triboelectric Charge** — Triboelectric charge is a buildup of static charge due to the contact and separation of two materials. Friction or rubbing enhances this effect due to the contact and separation of many parts of the surfaces.

**Wrist Strap** — A wrist strap is usually an elastic band that an operator wears around the wrist. The wrist strap has an electrical connection that is used to connect to a cable. The cable is connected to a conductive work surface pad at ground potential. The wrist strap must contact the bare skin to be effective.

## Static Safeguarded Workstation:

A static safeguarded workstation or static protective work area can be any area so designated for the repair and/or handling of ESDS devices or assemblies.

1. A static safeguarded work station should, as a minimum, have dissipative work surfaces (conductive mats) which are connected to ground. There should be a conductive cable or cord from the work surface that can be connected to a wrist strap. Grounded work surfaces must be kept clean. These surfaces should be cleaned with a Static Control Mat cleaner or spray. Other cleaners may leave a film residue that can reduce the effectiveness of the grounded work surface.
2. Dissipate floor coverings or mats may be applied to all floors in a static safeguarded area. These surfaces should be cleaned with a Static Control Mat cleaner or spray. Other cleaners may leave a film residue that can reduce the effectiveness of the grounded work surface. Do not apply floor wax to any floor covering or mat. The wax acts as an insulator.
3. Ionized air blowers may be used when the repair process requires the use of static charge generators, and other methods of charge dissipation do not work.
4. All electrical equipment and machinery in static safeguarded work areas must be electrically grounded so that the resistance from exposed metallic surfaces to work station ground connections does not exceeded 100 KΩ.
5. Special considerations for test stations:
   1. Electrical power and electrical test signals should be turned off before ESDS devices or assemblies are connected to or disconnected from test connectors.
   2. Power supply voltages should be applied before and removed after test stimuli/signals are applied or removed.
6. Unacceptable practices at or within 2 feet of static safeguarded areas include:
   1. Unpacking of parts or material contained in static generating material when ESDS devices are exposed.
   2. Storing of static generating packaging material within 2 feet of exposed ESDS devices.
   3. Trash cans.
7. Brushes with nonconductive nylon or Materials that are known static generators must be kept at least 2 feet away from static safeguarded work areas. Examples of static generating materials include:
   1. Gloves and smocks made from synthetic materials.
   2. Nonconductive solder removal tools.
   3. Nonconductive plastics such as plastic and Styrofoam cups, plastic work instruction protectors, clear plastic bags, will untreated foam padding packaging material, and tape.
   4. Plastic bristles.

## Repair Tools and Supplies

Special care should be taken when installing or removing ESDS devices to ensure that the

proper tools and supplies are being used. Hand-tools available with handles made of anti-static or static- dissipative materials should be used. Where insulating handles are necessary, separate ground connectors are required, such as on 3-wire soldering irons. Maintenance personnel should always ground the tip of the tool on a conductive table prior to applying it to an ESDS device. Place tools and fixtures on a grounded surface when not in use to help minimize static charge buildup.

## Handling Procedures/Precautions

The following procedures apply for handling ESDS parts or assemblies:

1. Special precautions may be required at test stations to prevent shorting of the assembly under test. Placing the assembly on an insulator is unacceptable.
2. The handling of ESDS devices should be restricted to static safeguarded work areas by personnel wearing either foot straps or wrist straps connected to ground.

**Note:** If wrist straps cannot be used, the work surface must be static safeguarded using an ionized air blower.

1. Anti-static packages should be placed on the grounded surface prior to removal, transfer, or insertion of their contents. This allows dissipation of any accumulated charge.
2. Personnel must be grounded with a wrist or foot strap when handling ESDS devices or assemblies.

**Caution** Use lotion, when necessary due to dry skin, to improve electrical contact between skin and wrist straps. Use only lotion that contains no mineral oil, glycerin, silicone, or lanolin that could contaminate the assembly.

1. Regular inspections should be made to ensure that wrist and foot straps have continuity and that the required series impedance is present. Grounding of the work surface should also be checked. Refer to "Anti-Static Device Testing" for foot, wrist, and work surface testing information.
2. Personnel should minimize contact of ESDS parts or assemblies with their clothing. Synthetic material is an excellent source of static electricity.
3. Paper should not be placed between ESDS anti-static packaging and the static safeguarded work surface, or between ESDS devices and the static safeguarded work surface.
4. Tools and fixtures used should be conductive between the working surface and the gripping point or bases to provide charge neutralization through the operators or stations.
5. Place all ESDS devices or assemblies in anti-static packaging (one item per bag) before removing them from a static-free workstation.
6. Materials that are known static generators must be kept at least 2 feet away from ESDS devices. Examples of static generating materials include:
   1. Gloves and smocks made from synthetic materials. -Nonconductive solder removal tools.
   2. Nonconductive plastics such as plastic and Styrofoam cups, plastic work instruction protectors, clear plastic bags untreated foam padding packaging material, and tape.
   3. Brushes with nonconductive nylon or plastic bristles.
7. Paper notebooks are acceptable if they do not contain plastic sheet protectors. The 2-foot rule does not apply.

## Transportation and Storage

1. Ionized air blowers neutralize charge on all contacted areas but are limited in range. Blowers are especially useful in removal of static charge from insulating surfaces; however, ESDS assemblies should not be placed on these surfaces unless absolutely necessary.
2. If ESDS devices must be transported away from a static safeguarded work area, the ESDS devices must be placed in an anti-static bag. The anti-static bag must cover the ESDS device completely.
3. When moving or storing a complete article, it should be fitted in an anti-static bag or better.

## Component Level

Transportation and storage of ESDS parts at the component level require that all device leads be effectively shorted together. To accomplish this, one or a combination of the following methods should be used.

1. Insert all leads of the device into high-density conductive foam (typically black).
2. Insert devices in an anti-static container.
3. Short all leads together with metal clips or store in grounded metal containers.

**Note**

Conductive bags with paper or other insulating materials bonded to their exterior should not be used unless a conductive path is present from the interior of the bag to an area on the exterior. This path is needed to allow neutralization of internal charges. Foil-lined, nonconductive containers must have foil around the outside and on the bottom for the same reason.

## Assembly Level

Anti-static protection is required for all assemblies containing ESDS parts anytime an assembly is removed from a static safeguarded workstation.

1. The assembly must be entirely enclosed by an anti-static bag. Partial coverage, laying the assembly on the anti-static bag, or wrapping the assembly in the anti-static bag is unacceptable.
2. Whenever anti-static bags are impractical, ionized air may be used.
3. Do not open an anti-static bag without a properly connected wrist/foot strap.
4. All procedures that apply to ESDS devices at the component level also apply to the assembly level.

## Additional Precautions

These additional precautions should be taken to minimize static accumulation and subsequent damage to parts or assemblies.

1. Nylon or synthetic gloves and smocks should not be worn.
2. Plastic cups, paper protectors, and other nonconductive plastics should not be allowed at a static safeguarded workstation.
3. Handle ESDS devices by their cases whenever possible; avoid touching the leads or contacts.
4. Use natural bristle brushes, not synthetic ones.
5. Paper should not be placed between ESDS devices and the static safeguarded work surface.
6. Placing a hand on the static safeguarded work surface provides a ground path equal to the wrist strap as long as the contact is maintained. When the hand is removed from the static safeguarded work surface, the ground path is lost.

## Anti-Static Device Testing

All anti-static equipment should be tested on a regular basis. The following information provides general guidelines that should be maintained as recommended by Alta Avionics, LLC Tests shall be performed in accordance with test equipment manufacturers Operator's Manual.

1. Wrist Strap Cord — Check on a weekly basis.
2. The resistance as checked from the wrist strap metal to the end of the wrist strap cord should be from 470 KΩ to 2MΩ.
3. Work Surface Mat — check on a monthly basis. The resistance as checked across the ground cord terminal and the wrist strap cord terminal should be less than '1MΩ.
4. Work Area — Check on a monthly basis. The resistance from the metal on the wrist strap through the wrist strap cord, work surface mat, and ground cord to the ground cord's termination connector should be from 1 to 10 MΩ.

Table 1 ESDS Work Station Test Equipment

|  |  |  |
| --- | --- | --- |
| **Static Safeguarded Workstation** | **Meter** | **Measurement Parameter** |
| Wrist Strap | 3M company 746 or equivalent model | Pass/Fail Indication is shown |
| Work Surface Mat | 3M Company Model 701 or equivalent | 10/100-V megohmmeter shows a total resistive reading as measured across 2(two)x5lb electrodes |

# Procedures for Insuring Currency of Technical Data.

## Sources for Technical Data

The Repair Station will use, and has approved for use, the following sources for technical data.

1. ATP Microfiche Library.
2. CD ROMs or DVDs provided by manufacturers.
3. Internet sources provided by manufacturers and third-party sources approved by manufacturers such as "Resource One."
4. Limited printed library.
5. Other sources after review and approval of the Chief Inspector.

## Checking currency of printed manuals.

1. Currency of printed manuals will be checked against all possible update sources before work begins. These sources include but are not limited to;
   1. Manufactures Web site
   2. Manufacturers' yearly or current Tech. Pub's Index
   3. Phone/Fax contact with manufacturer.
   4. ATP Microfiche Library (described below)
2. The manual/document used for repair will be identified and recorded on the work order associated with the repair of the aircraft or article.
3. A special class of printed manuals is the "static" class. These manuals are no longer updated by the manufacturers and will be marked as "Static." The last revision date available will be marked on the sticker.

## ATP Microfiche Library

The ATP Microfiche Library is checked for currency by the following method:

1. ATP updates the microfiche monthly and provides a new Pub's Index. The Pub's Index will be used to ensure currency of the data.
2. CD ROMs/DVDs

The CD ROMs and DVDs are checked for currency by the following method:

1. Accessing the manufacturer’s website or Pub's Index. These resources are maintained by the manufacturer or an approved third party.

## Sources for Special Conditions.

1.7 5 1 These sources of data are Alta Avionics, LLC, Vendors (with current audit), or other manufacturers. These will include, but not limited to: Tech Rep's, faxes, and other forms of communication.

## Data Unavailable

If current repair data is not available, the repair will be suspended until data is available, or equipment will be forwarded to an appropriate repair facility.

# Procedures for Test Equipment Equivalency

Equivalency for special tools and equipment is established by comparing the specifications of the recommended and equivalent test equipment. This goes beyond the basic functional criterion used for test equipment. Potentially equivalent test equipment must be proved to be equivalent in function to those recommended by the repair, overhaul and maintenance manual supplied by the appliance manufacture. This may include equivalency in accuracies, tolerances and functional specifications. Alta Avionics, LLC has identified a need to implement a test equipment equivalency during the overhaul/repair bench procedure. Therefore, the following procedure has been implemented to ensure all personnel have a resource for initial and recurrent training.

1. After repairman conducts preliminary inspection (see section 8.2 for Preliminary Inspection details) and determines that test equivalency is required; the following procedure must be followed according to the Test Equipment Equivalency Log.
2. Obtain and copy the current required test equipment section from the maintenance manual.
3. Obtain a blank equivalency data sheet located in Test Equipment Equivalency Log.
4. Using the manufactures required test equipment data, inventory the test equipment to be used. Note any equipment that is not listed on required equipment list on the equivalency data sheet.
5. Verify all non-listed equipment, using specifications of the listed as the standard. List all references used to prove the equivalency of the listed equipment.
6. Attach the manufactures required equipment list to the equivalency data sheet and insert then into the Test Equipment Equivalency Log.
7. Continue with 1.3 Bench Appliance Repair Procedures of appliance according to the set procedure set forth on page 5.

\*\*\* An Example is supplied in the Test Equipment Equivalency Log. This log is located in all Alta Avionics locations in the appliance bench repair area.

# Procedure for Go/No-Go testing

All repair stations controlled under Alta Avionics, LLC are required to follow the Go/No-Go testing procedure in the event avionics items are removed from an aircraft for the intent of troubleshooting an aircraft, aircraft radios and/or systems.

## Go/No-Go Test

1. Technician completes all initial preliminary inspection and during the troubleshooting process determines a Go/No-Go procedure will be required to verify or eliminate suspect equipment
2. Technician removes all suspect equipment from aircraft
3. Technician delivers equipment to service department assistant
4. Service Department Assistant logs in equipment
   1. Attach article tag AT-AT (Article Tag) (see example in Form Manual)
      1. Add all identifying information to AT-AT tag
         1. Reference line will be filled in with Work order number and task
         2. Check appropriate CRS block
            1. **Note: At this time the following sections will not be completed**:
         3. Repair description blocks
         4. Warranty “Thru” date
   2. Technician notes equipment on the aircraft task ‘Major Parts History’ as a removed item
      1. Description, P/N and S/N will be documented
      2. Mark boxes as ‘OFF’ and ‘SV’
   3. Deliver equipment to technician for Go/No-go testing
5. Technician performs Go/No-Go Testing procedure
   1. Item will be bench tested to determine if the item will meet manufactures’ specifications
   2. PASSED – (If equipment meets manufactures specifications)
      1. Complete sticker AT-AT
         1. Check “OPS CHK” box
         2. Label line named “THRU” with N/A
         3. **Note: If equipment meets manufactures specifications it is deemed serviceable and no** **FAA 8130 is required**
      2. Technician reinstalls suspect equipment in aircraft
      3. Technician adds an entry on the Work Order aircraft task ‘Major Parts History’ section showing article removed as being reinstalled in the aircraft as ‘ON’ ‘SV’. (For Work Order reference see the Alta Avionics Forms manual)
         1. Article will be list on “Major Parts History” with Serial number, part number and denoting the article was removed and reinstalled as serviceable
   3. FAILED - (If equipment does not meet manufactures specifications)
      1. Article is returned to Service department Assistant
         1. Assistant will change the status of the line item on the ‘Major Parts History’ section for the removed item from ‘SV’ to ‘RP’ and open a bench task following the Appliance Procedure “Technician Carry-In”

# IFR Certification Procedures – FAR 91.411 & FAR 91.413

## Preliminary Steps

1. Determine test equipment to be used is within calibration period, and the correction chart on the pitot/static instruments is current.
2. Relocate aircraft inside metal hanger with doors closed while static system is under test to avoid traffic avoidance systems interference. If this is not possible, coordinate with the tower for times the test can be done outside. An alternate method would be to use the antenna shields supplied in the ATC-601 test set accessories, or go direct to the transponder antenna connection thru a 34-db. attenuator and complete the encoder check then reconnect the antenna to complete the radiated check.
3. Check for any AD's or bulletins that may apply to the altimeter under test. Check aircraft maintenance manual for special test conditions, such as author Q's, limits, special tolerances, etc. Verify the maximum operating altitude of the altimeter meets or exceeds the operating ceiling of the aircraft under test. Before connecting pitot/static test set to aircraft or instruments, complete barometric scale error test in accordance with FAR Part 43 App. E, table IV. Record the results on Form A-ATI (Altimeter Test/Inspection).
   1. Set barometric scale to 29.92 and record altitude reading of altimeter under test.
   2. Reset scale to values listed in table IV.
   3. Compute difference between actual reading and the 29.92 value.
   4. If altimeter does not exceed tolerance, proceed to next step. If altimeter does exceed tolerance, it must be removed for repair.
4. Inspect Static system to be checked for entrapped moisture, restrictions, or damage.
   1. Cycle pitot/static drains.
   2. If moisture is present, drain and purge system in accordance with manufacturers recommended procedures.
   3. If damage is noted on pitot or static ports, repair as required.
5. Connect aircraft to auxiliary power since the system will be powered up for the duration of the test.
6. Connect pitot/static test set to aircraft static port. Seal off opposite port if applicable, with metal tape. Flag tapped port with flagging material. If test is being completed on a FAR part 121 aircraft, complete log entry stating static ports have been sealed for static tests.
7. Connect pitot adaptor to airspeed pitot and tape off drain holes, if applicable, with electrical tape.
8. Close cross feed on pitot/static test set and apply airspeed, check the aircraft airspeed indicator at approx. 150kts for accuracy (for reference later), and note pitot leak rate.  
     
   **NOTE:** The pitot system leak rate may be found in the aircraft maintenance manual. If a leak rate is not in the manual, 10 kts. per/min. (or less) @150 kts. is acceptable. (AC43.13-1A Chap 12 Sect 4)
9. Open cross feed and perform leak check of static system.
   1. If un-pressurized aircraft, leak test altitude is 1000 ft above field elevation. Leak rate not to exceed 100 fpm (FAR 23/25.1325 par. 2 I).
   2. If pressurized aircraft, leak test altitude is at maximum cabin differential pressure. Leak rate not to exceed 2% of test altitude or 100 fpm, whichever is greater (FAR 23/25.1325 par. 2 ii).  
        
      **NOTE**: A simple way to determine test altitude on pressurized aircraft is to read the altitudes on the cabin pressure controller dial. Find the field elevation on the inner ring of numbers and read the test altitude directly across on the outer ring of numbers. Always refer to the aircraft maintenance manual to confirm the max differential altitude.
10. Set up transponder tester per manufacturer’s instructions. Check for any AD's or bulletins applicable to the transponder under test.

## Test Procedures

1. Set all barometric scales to 29.92 in. hg.
2. Record part numbers and serial numbers of altimeters, blind encoders, air data computers, and transponders in the appropriate spaces on the Altimeter Test/ Inspection Form A-ATI, the ATC Transponder and Mode S Inspection Form A-ATC-TMSI, and the Logbook Entry AL-ATI (Altimeter Test and Inspection).
3. Determine if altimeters are air data driven or pneumatic.
   1. Obtain pitot/static schematics from aircraft maintenance manual if static system is complex.
   2. If working on a pressurized aircraft disconnect cabin pressure altimeter instrument and plug static line for the duration of the test. Remember to reconnect cabin pressure altimeter instrument and leak check to proper test altitude when tests are completed.   
        
      **Note: If cabin pressure altitude instrument is forced below field elevation, it could be damaged.**
   3. If air data driven, connect direct to air data static and pitot inputs to test and certify.
   4. If pneumatic, connect to aircraft static systems and certify as a system.  
        
      **NOTE**: When testing as a system, and the leak rate exceeds 100 FPM at 18,000 Feet, the determination needs to be made as to whether the altimeter meets the CASE LEAK specifications.
4. Determine if Static Defect Correction unit is installed (in most jets). This unit will have a bearing on how the tests are to be performed. A correction curve will need to be obtained from the pilots operating handbook or from the equipment manufacturer, and referred to during the scale error test.
5. Complete altimeter and encoder tests in accordance with the limits called out in FAR Part 43, App E & FAR Part 91.217, and record readings on the Form A-ATI (Altimeter Test Inspection).  
     
   **NOTE:** Scale error tests are accomplished by reading the test altitude on the pitot/static test set altimeter and adjusting the static altitude as read on the test set altimeter plus or minus to match the correction chart. With correction applied, the test altitude will be accurate. Then record the error as read on the aircraft altimeter under test on Form A-ATI (Altimeter Test Inspection).  
     
   **NOTE:** Record field pressure altitude in the test set reference box in the AFTER-EFFECT TEST section as read on the test set altimeter as well as the altimeter under test prior to testing. Use this test set altitude reading as a reference when recording the after-effect test readings. This practice will eliminate errors associated with atmospheric pressure changes during the test, and compensates for “0” hysteresis if using a digital test altimeter.  
   1. Set test unit to each test altitude and record readings
   2. Verify unit under test does not exceed tolerance.
   3. Record mode C readings. The difference between altitude as read on the aircraft altimeter being tested and the mode C reported on the transponder test set will not exceed ± 125 feet on a 95% probability as called out in FAR 91-217.  
        
       **NOTE:** If the encoder maximum altitude is less than the altitude ceiling of aircraft under test, a logbook entry and a panel placard will be completed stating "Mode C altitude encoding limited to xxx feet."
6. Hysteresis Test  
   **Note:** This test must be completed within 15 minutes of reaching maximum test altitude.  
   1. The Hysteresis test is checked at a test altitude equivalent to 50% and 40% of the maximum altitude of the altimeter under test.
   2. Reduce altitude at a rate not to exceed 5,000 fpm, or the maximum rate of VSI, whichever is less. Reduce rate to 3000 fpm when within 3000 ft of the first test point. Hold at 50% point for five (5) minutes and record reading. Reduce altitude to 40% point, hold for five (5) minutes and record reading.
   3. The difference between the up and down readings should not exceed ±75 feet.
7. After Effect Test  
   **NOTE:** This test must be completed within five (5) minutes after Hysteresis test is completed.  
   1. Reduce altitude to field elevation (test set reference).
   2. Compare altimeter reading to original reading before the test was started. Difference should not exceed ±30 feet.
8. Close cross feed on pitot/static test set and apply airspeed. Compare aircraft airspeed indicator reading to the value recorded in preliminary step 8. This check verifies no damage was done to the airspeed indicator during the test.
9. Complete operational check of all pitot/static heaters after hoses, tape, caps, or plugs have been removed. Always remove hoses from pitot/static tester first. This will prevent any possible instrument damage.
10. Complete transponder test and record results on Form A-ATC-TMSI (ATC Transponder and Mode S Inspection). It is advisable, and good practice, to go beyond the requirements of FAR Part 43, Appendix E, and check other parameters of the transponder to insure the unit being tested is in healthy condition.
11. After all checks are completed and found to meet specs, install tag AT-A43 (Tested as PER FAR 43) on altimeters, air data computer, or blind encoder verifying these units have been tested in accordance with FAR Part 91.217 or FAR Part 91.411 as required. Complete all forms including Logbook Entry AL-ATI (Altimeter Test and Inspection). Make a copy of the Logbook Entry AL-ATI (Altimeter Test and Inspection) to be placed in the work order folder, the original to be placed in the aircraft logbook or given to maintenance control of the aircraft. Make of copy of Form A-ATI (Altimeter Test Inspection) and include with the aircraft records or staple or clip to logbook.
12. Record the ID numbers for the test equipment used on Form A-ATI (Altimeter Test Inspection) and Form A-ATC-TMSI (ATC Transponder and Mode S Inspection)

# Audits

## Audit Request

The Chief Inspector is responsible for conducting scheduled audits as well as audits required on an as needed basis. It is best practice to notify the Chief Inspector of a needed Vendor Audit using the AA-AR Form found below.



Instructions for AA-AR (Audit Request)

1. Audit
   1. Indicate whether this is an audit for a NEW or CURRENT Vendor.
   2. Fill in the Name field.
   3. Fill in the Address field.
   4. Fill in the Phone field.
2. Internal
   1. Indicate which general department this Audit is for
3. Reason
   1. Describe the reason the audit is necessary.
4. Requested By.
   1. Indicate who is requesting the Audit.
5. Date
   1. Date that the Audit was requested.

## Calibration Vendor Audit

A Calibration Vendor Audit is an audit of a Vendor that will be doing calibrations on behalf of Alta Avionics, LLC. The purpose of this audit is to ensure the Vendor meets or exceeds the safety and quality standards defined by Alta Avionics, LLC. For this purpose, it is best practice to use the Calibration Vendor Audit Form (AA-CVA) as described below when conducting such an audit.

## Form AA-CVA (Calibration Vendor Audit)











### Instructions for Form AA-CVA (Calibration Vendor Audit)

Note: not all areas of the repair station are subject to all of the audit elements contained in this Audit Checklist.

Where audit elements are not applicable, place an X in the N/A column for that element.

1. Enter Vendor Company name
2. Enter Vendor address
3. Enter Vendor city
4. Enter Vendor State
5. Enter Vendor Zip Code
6. Annotate the particular division of the Vendor Company
7. Enter the Vendor telephone number
8. Enter the Vendor fax number
9. Enter the number of years the Vendor has been in operation at the specific location
10. Enter the number of employees maintained by the Vendor
11. Company Contacts;
    1. Enter the name of the primary point of contact for Vendor Quality Control
    2. Enter the phone number of the Quality Control contact
    3. Enter the name of the primary point of contact for Vendor Inspection
    4. Enter the phone number of the Inspection contact
12. Alta Avionics Audit Details
    1. Enter Vendor Category
    2. Enter an X on the appropriate line for Audit Type
    3. Enter the recommended audit interval in months
    4. Status;
    5. Enter an X on the appropriate line for Acceptance, Conditional Acceptance, or Not Accepted
    6. Circle one of the recommendations of Vendor status (approved vendor list); Add, Delete (remove), Update, Does not Qualify
    7. Enter the date at which the next scheduled audit will take place
    8. Auditor applies signature and date the form and audit are completed
13. Quality Assurance System; these audit element questions are self-explanatory, review and answer all audit element questions
14. Technical Data; these audit element questions are self-explanatory, review and answer all audit element questions
15. Training; these audit element questions are self-explanatory, review and answer all audit element questions
16. Measuring and Test Equipment Calibration (standards); these audit element questions are self-explanatory, review and answer all audit element questions
17. Work Processing; these audit element questions are self-explanatory, review and answer all audit element questions
18. Records; these audit element questions are self-explanatory, review and answer all audit element questions
19. Facilities; these audit element questions are self-explanatory, review and answer all audit element questions
20. Certificates and Reports; these audit element questions are self-explanatory, review and answer all audit element questions
21. Shipping; these audit element questions are self-explanatory, review and answer all audit element questions
22. Shelf Life Program; these audit element questions are self-explanatory, review and answer all audit element questions
23. Electrostatic Discharge Procedures; these audit element questions are self-explanatory, review and answer all audit element questions
24. Drug and Alcohol Program; these audit element questions are self-explanatory, review and answer all audit element questions
25. Repair Station; these audit element questions are self-explanatory, review and answer all audit element questions

**NOTE**: Not all fields always require text entry. Contact your supervisor if there are questions. In cases where data is not required, DO NOT leave blank; enter N/A or mark N/A in the field.

## AA-VA: Vendor Audit

A Vendor Audit is an audit of a Vendor that will be providing ongoing parts and/or services to or on behalf of Alta Avionics, LLC. The purpose of this audit is to ensure the Vendor meets or exceeds the safety and quality standards expected by Alta Avionics, LLC. For this purpose, it is best practice to use the Vendor Audit Form (AA-VA) as described below when conducting such an audit. This form and these instructions should not be used if doing a Vendor Audit on a Vendor that will be doing calibration on behalf of Alta Avionics, LLC. Instead, use refer to the Calibration Vendor Audit form and instructions for auditing calibration vendors.













### Instructions for Form Use: AA-VA: Vendor Audit

Note: not all areas of the repair station or vendor are subject to all of the audit elements contained in this Audit Checklist. Where audit elements are not applicable, place an X in the N/A column for that element or mark N/A in the field.

1. Enter Distributor Company name
2. Enter Distributor address
3. Enter Distributor city
4. Enter Distributor State
5. Enter Distributor Zip Code
6. Annotate the particular division of the Distributor Company
7. Enter the Distributor telephone number
8. Enter the Distributor fax number
9. Enter the number of years the Distributor has been in operation at the specific location
10. Enter the number of employees maintained by the Distributor
11. Company Contacts;
    1. Enter the name of the primary point of contact for Distributor Quality Control
    2. Enter the phone number of the Quality Control contact
    3. Enter the name of the primary point of contact for Distributor Inspection
    4. Enter the phone number of the Inspection contact
    5. Enter the name of the primary point of contact for Distributor Material Control
    6. Enter the phone number of the Material Control contact
12. Alta Avionics Audit Details
    1. Enter Distributor Category
    2. Enter an X on the appropriate line for Audit Type
    3. Enter the recommended audit interval in months
    4. Status;
       1. Enter an X on the appropriate line for Acceptance, Conditional Acceptance, or Not Accepted
    5. Circle one of the recommendations of Distributor status (approved vendor list); Add, Delete (remove), Update, Does not Qualify
    6. Enter the date at which the next scheduled audit will take place
    7. Auditor applies signature and date the form and audit are completed
13. Quality Control System; these audit element questions are self-explanatory, review and answer all audit element questions
14. Measuring and Test Equipment Calibration (standards); these audit element questions are self-explanatory, review and answer all audit element questions
15. Technical Data; these audit element questions are self-explanatory, review and answer all audit element questions
16. Records; these audit element questions are self-explanatory, review and answer all audit
17. element questions
18. Shelf Life Program; these audit element questions are self-explanatory, review and answer all audit element questions
19. Training; these audit element questions are self-explanatory, review and answer all audit
20. element questions
21. Procurement; these audit element questions are self-explanatory, review and answer all audit element questions
22. Material Control; these audit element questions are self-explanatory, review and answer all audit element questions
23. Housing and Facilities; these audit element questions are self-explanatory, review and answer all audit element questions
24. Repair Station; these audit element questions are self-explanatory, review and answer all audit
25. element questions
26. Drug and Alcohol Program; these audit element questions are self-explanatory, review and answer all audit element questions

**NOTE**: Not all fields always require text entry. Contact your supervisor if there are questions. In cases where data is not required, DO NOT leave blank; enter N/A.

# Aircraft Check-in Inspection

## Purpose

The purpose of inspection of an incoming aircraft is to ensure that the aircraft, equipment and condition of the aircraft and equipment is documented. This ensures that there are no questions as to missing equipment, new damage (hanger rash), etc. when the customer picks up the aircraft.

## Procedure

The checklist A-ACI (Aircraft Check-in Inspection) is meant to ensure that the receiver of the aircraft has looked at all the different components of the aircraft and noted its condition, noticeable damage or features, as well as what equipment, headsets, baggage, books, logs, etc. were in the aircraft or given over to the care of the Repair Station.

What to look for

The checklist A-ACI is meant as a reminder to ensure that the receiver has looked at all the components of the aircraft. When looking at each component the Receiver should be looking for (as applicable):

* Damage (dings, dents, bent, warped, holes, torn, worn, ripped, chipped)
* Missing
* Leaks (oil, hydraulic fluid, fuel)
* Loose (screws, bolts, rivets)
* Corrosion (rust, electrolysis, rot, dry rubber)
* Function (lights, controls, vents, knobs, seats, dials, yokes, sticks, etc.)

Form A-ACI is meant as a reminder/checklist to aid the receiver in noting the condition of the aircraft at time of arrival. This form should be signed by both the receiver and owner/operator and kept as part of the work order record

