

Numerical Simulation of Turbulent Combustion Using the Turbulent Flamelet Model

CNF, Atlanta 2019.

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INTRODUCTION

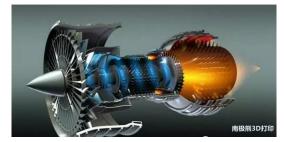
part1.1
part1.2
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Backgound

Introduction

Turbulent combustion is encountered in most practical combustion system such as rocket, ICE, and aircraft engien.





Meaningful to practical systems:

- (a) Improve efficiency, meet demanding standards[1].
- Reduce pollution, environment friendly.

Research aspects

Why is numerical simulation adopted?

- (a) Analytical techniques are difficult to handle.
- (b) Experiments are too expensive to be widely used.

What are the key points in numerical practice?

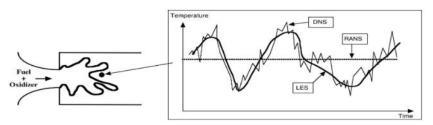
- Interaction between turbulence and combustion.
- Proper flame model.

Numerical simulation of flowfield

Based on CFD, there're 3 possible approaches:

- (a) Direct Numerical Simulation(DNS)-Precise, but costly(Tremendous memory and CPU).
- (b) Large Eddy Simulation(LES)-Compromise between accuracy and computational cost.
- (c) Reynolds Averaged Navier-Stokes(RANS)

 —Inaccurate for combustion phenomenon.



Numerical simulation of flame

Although based on LES, traditional flame models are inadequate:

- Distribution of flame properties are mannually assumed.
- Parameter tuning may be unphysical.

These drawbacks are overcome by the turbulent flamelet model to be introduced:

- Designed especially for LES, with thewer approximation.
- Relations are provided through scaling law, which is based on DNS database[2].

TURBULENT FLAMELET MODEL

art2.2

rt2.3

Theory

Original G.E. in the context of LES:

$$\frac{\partial \bar{\rho} \tilde{Z}}{\partial t} + \nabla \cdot (\bar{\rho} \tilde{Z} \tilde{\vec{u}}) = \nabla \cdot [\bar{\rho} (D + D_T) \nabla \tilde{Z}]$$
 (1)

$$\frac{\partial \bar{\rho} \tilde{Y}_{i}}{\partial t} + \nabla \cdot (\bar{\rho} \tilde{Y}_{i} \tilde{\vec{u}}) = \nabla \cdot [\bar{\rho} (D + D_{T}) \nabla \tilde{Y}_{i}] + \overline{\omega_{i}}$$
 (2)

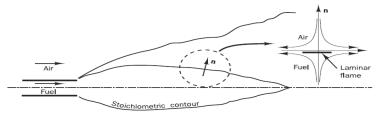
After coordinate transformation(x_1, x_2, x_3, t) \rightarrow (Z, Z_2, Z_3, τ):

$$\begin{split} & \bar{\rho} \frac{\partial \tilde{Y}_{i}}{\partial \tau} + \bar{\rho} \Big(\tilde{\vec{u}} \cdot \nabla_{\perp} \tilde{Y}_{i} + \frac{\partial \tilde{Y}_{i}}{\partial Z_{2}} \frac{\partial Z_{2}}{\partial t} + \frac{\partial \tilde{Y}_{i}}{\partial Z_{3}} \frac{\partial Z_{3}}{\partial t} \Big) = \frac{\bar{\rho} \chi}{2 Le_{T}} \frac{\partial^{2} \tilde{Y}_{i}}{\partial^{2} \tilde{Z}} \\ & + \frac{\partial \tilde{Y}_{i}}{\partial \tilde{Z}} \nabla \cdot \left[\bar{\rho} (\mathcal{D}_{T,i} - \mathcal{D}_{T}) \vec{n} \cdot \frac{\partial \tilde{Z}}{\partial \vec{n}} \right] + \nabla \cdot (\bar{\rho} \mathcal{D}_{T,i} \nabla_{\perp} \tilde{Y}_{i}) + \overline{\omega_{i}} \end{split}$$
(3

Laminar Flamelet assumption

Locally, the characteristic timescale of chemical reaction is much smaller that that of flow($t_c \ll t_f$).

Thus, local flame structure can be described by the difffusion flame under counterflow configuration.

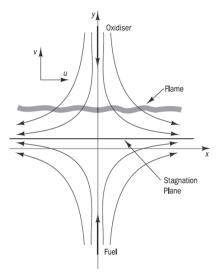


Each micro flamelet can be described by Z and χ :

- Z describes chemical reaction.
- \blacksquare χ indicates turbulence stretching effect.

Thus, a database can be pre-computed for later looking-up.

Turbulent Flamelet



Unlike the laminar flamelet introduced above, G.E. of the counterflow flame is slightly modified by our

turbulet flamelet model from

$$\rho \frac{\mathrm{DY_i}}{\mathrm{Dt}} = \mathcal{D}_i \frac{\partial^2 Y_i}{\partial^2 x} + \omega_i (T, \vec{Y})$$
(4)

to

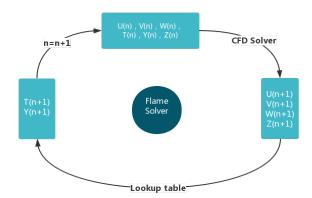
$$\bar{\rho} \frac{D\tilde{Y}_{i}}{Dt} = \mathcal{D}_{i} \frac{\partial^{2} \tilde{Y}_{i}}{\partial^{2} x} + \tilde{\omega}_{i} (\tilde{T}, \tilde{\vec{Y}})$$
(5)

The two equations share similar form, but have totally different meanings.

Solution procedure

Based on the filtered turbulent flamelet database generated in the way descirbed above, the full solution procedure that incorporates a CFD solver can be described as follows:

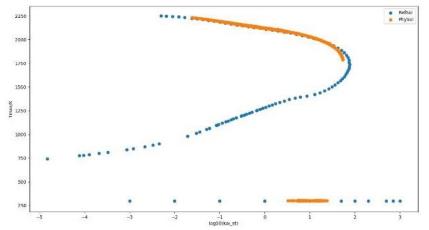
Numerical Results



NUMERICAL RESULTS

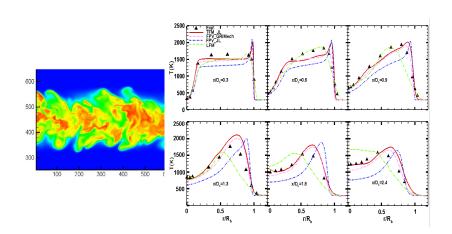
The T_{max} plot:

- One of the most convincing testing cases.
- Difference and transition position are clearly revealed[3].



Standard case

Comparsion between experimental data, which is widely used as benchmark[4].



Numerical Results

Reference

- [1] Charles David Pierce and Parviz Moin. "Progress-variable approach for large-eddy simulation of turbulent combustion". Thesis. 2001 (cit. on p. 3).
- [2] Wang Lipo. "Analysis of the filtered non-premixed turbulent flame". In: <u>Combustion and Flame</u> 175 (2017), pp. 259–269 (cit. on p. 6).
- [3] Joseph F Grear. "The Twopnt program for boundary value problems". In:

 Sandia National Laboratories Report SAND91-8230 (1992)
 (cit. on p. 13).
- [4] P. E. Dimotakis. "The mixing transition in turbulent flows". In: <u>Journal of Fluid Mechanics</u> 409 (2000), pp. 69–98. <u>ISSN</u>: 0022-1120. DOI: 10.1017/s0022112099007946 (cit. on p. 14).