

# Numerical Simulation of Turbulent Combustion Using the Turbulent Flamelet Model

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Yu Cang

# INTRODUCTION

part1.1
part1.2
part1.3

## Backgound

Turbulent combustion is encountered in most practical combustion system such as rocket, ICE, and aircraft engien.





Meaningful to practical systems:

- (a) Improve efficiency, meet demanding standards.
- (b) Reduce pollution, environment friendly.

# Research aspects

Why is numerical simulation adopted?

- (a) Analytical techniques are difficult to handle.
- (b) Experiments are too expensive to be widely used.

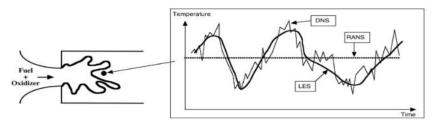
What are the key points in numerical practice?

- Interaction between turbulence and combustion.
- Proper flame model.

#### Numerical simulation of flowfield

#### Based on CFD, there're 3 possible approaches:

- (a) Direct Numerical Simulation(DNS)
  - -Precise, but costly(Tremendous memory and CPU). X
- (b) Large Eddy Simulation(LES)
  - -Compromise between accuracy and computational cost. ✓
- (c) Reynolds Averaged Navier-Stokes(RANS)
  - -Inaccurate for combustion phenomenon. X



#### Numerical simulation of flame

Although based on LES, traditional flame models are inadequate:

- Distribution of flame properties are mannually assumed.
- Parameter tuning may be unphysical.

These drawbacks are overcome by the turbulent flamelet model to be introduced:

- Designed especially for LES, with thewer approximation.
- Relations are provided through scaling law, which is based on DNS database.

TURBULENT FLAMELET MODEL

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## Theory

Original G.E. in the context of LES:

$$\frac{\partial \bar{\rho} \tilde{Z}}{\partial t} + \nabla \cdot (\bar{\rho} \tilde{Z} \tilde{\vec{u}}) = \nabla \cdot [\bar{\rho} (D + D_T) \nabla \tilde{Z}]$$
 (1)

$$\frac{\partial \bar{\rho} \tilde{Y}_{i}}{\partial t} + \nabla \cdot (\bar{\rho} \tilde{Y}_{i} \tilde{\vec{u}}) = \nabla \cdot [\bar{\rho} (D + D_{T}) \nabla \tilde{Y}_{i}] + \overline{\omega_{i}}$$
 (2)

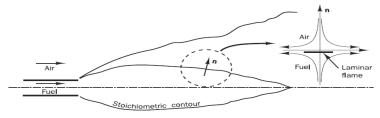
After coordinate transformation( $x_1, x_2, x_3, t$ )  $\rightarrow$  ( $Z, Z_2, Z_3, \tau$ ):

$$\bar{\rho} \frac{\partial \tilde{Y}_{i}}{\partial \tau} + \bar{\rho} \left( \tilde{\vec{u}} \cdot \nabla_{\perp} \tilde{Y}_{i} + \frac{\partial \tilde{Y}_{i}}{\partial Z_{2}} \frac{\partial Z_{2}}{\partial t} + \frac{\partial \tilde{Y}_{i}}{\partial Z_{3}} \frac{\partial Z_{3}}{\partial t} \right) = \frac{\bar{\rho} \chi}{2 Le_{T}} \frac{\partial^{2} \tilde{Y}_{i}}{\partial^{2} \tilde{Z}} \\
+ \frac{\partial \tilde{Y}_{i}}{\partial \tilde{Z}} \nabla \cdot \left[ \bar{\rho} (\mathcal{D}_{T,i} - \mathcal{D}_{T}) \vec{n} \cdot \frac{\partial \tilde{Z}}{\partial \vec{n}} \right] + \nabla \cdot (\bar{\rho} \mathcal{D}_{T,i} \nabla_{\perp} \tilde{Y}_{i}) + \bar{\omega}_{i}$$
(3)

# Laminar Flamelet assumption

Locally, the characteristic timescale of chemical reaction is much smaller that that of flow( $t_c \ll t_f$ ).

Thus, local flame structure can be described by the difffusion flame under counterflow configuration.

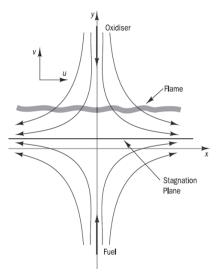


Each micro flamelet can be described by Z and  $\chi$ :

- Z describes chemical reaction.
- $\blacksquare \chi$  indicates turbulence stretching effect.

Thus, a database can be pre-computed for later looking-up.

#### Turbulent Flamelet



Unlike the laminar flamelet introduced above, G.E. of the counterflow flame is slightly modified by our turbulet flamelet model from

$$\rho \frac{\mathrm{DY_i}}{\mathrm{Dt}} = \mathcal{D}_i \frac{\partial^2 Y_i}{\partial^2 x} + \omega_i (T, \vec{Y})$$
(4)

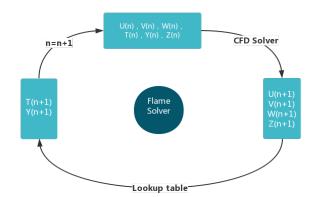
to

$$\bar{\rho} \frac{D\tilde{Y}_{i}}{Dt} = \mathcal{D}_{i} \frac{\partial^{2} \tilde{Y}_{i}}{\partial^{2} x} + \tilde{\omega}_{i} (\tilde{T}, \tilde{\vec{Y}})$$
(5)

The two equations share similar form, but have totally different meanings.

# Solution procedure

Based on the filtered turbulent flamelet database generated in the way descirbed above, the full solution procedure that incorporates a CFD solver can be described as follows:

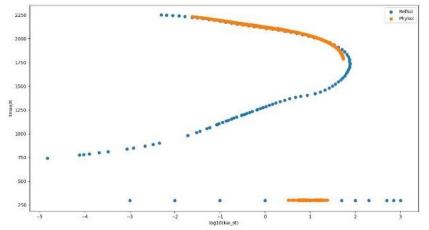


# NUMERICAL RESULTS

# Comparsion of "S" curve

#### The $T_{max}$ plot:

- One of the most convincing testing cases.
- Difference and transition position are clearly revealed!



#### Standard case

Comparsion between experimental data, which is widely used as benchmark.

