Helicopter Tuning Results Document

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Abstract

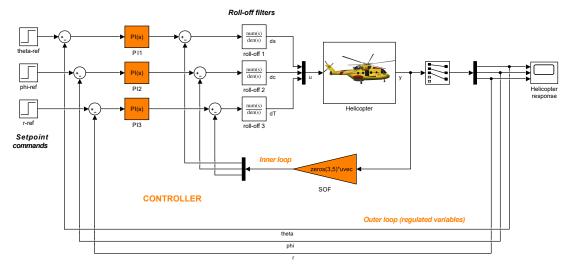
Helicopter tuning example

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Chapter 1. Helicopter Control System Model

The following model contains the Helicopter Control system and model used for tuning the gains. This is a continuous model will linearized as part of the tuning me that be



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Chapter 2. Code: Heli_tuning_script

MATLAB Tuning Code. The following code uses Simulink Control Design to tune the inner and outer loop gains for the helicopter control system. The inner loop control is a mulitvariable control loop and the outer loop consists of three PI controllers, one for each axis.

```
% Helicopter tuning example
% Copyright 2016 The MathWorks, Inc.
% first, test for a Simulink Control Design, Control System Toolbox
 and
% Robust Control Toolbox licenses
if license('test','simulink_control_design')&&...
 license('test','control_toolbox')&&license('test','robust_toolbox')
    % bring command window to the front
    commandwindow;
    ST0 = slTuner('rct_helico_Sdomain',{'PI1','PI2','PI3','SOF'});
    addPoint(STO, {'theta-ref', 'phi-ref', 'r-ref'}); % setpoint
    addPoint(STO, {'theta', 'phi', 'r'});
                                                      % corresponding
 outputs
    addPoint(ST0, {'u', 'y'});
    % Less than 20% mismatch with reference model 1/(s+1)
    TrackReq = TuningGoal.StepTracking({'theta-ref','phi-ref','r-
ref'},{'theta','phi','r'},1);
    TrackReq.RelGap = 0.2;
    % Gain and phase margins at plant inputs and outputs
    MarginReq1 = TuningGoal.Margins('u',5,40);
    MarginReq2 = TuningGoal.Margins('y',5,40);
    % Limit on fast dynamics
    MaxFrequencv = 25;
    PoleReq = TuningGoal.Poles(0,0,MaxFrequency);
    AllReqs = [TrackReq,MarginReq1,MarginReq2,PoleReq];
    [ST1,fSoft,~,Info] = systune(ST0,AllReqs);
    T1 = getIOTransfer(ST1, {'theta-ref', 'phi-ref', 'r-ref'},
{'theta','phi','r'});
    step(T1,5);
    figure('Position',[100,100,900,474]);
    viewSpec(AllRegs,ST1,Info);
    showTunable(ST1);
```

```
% Clean up
bdclose('all');
clearvars;

else % open the pre-existing report and post a warning
  open('Heli-Tuning-Report.pdf');
  warndlg(['There is a license missing for Simulink Control Design
'...
  'or Control System Toolbox or Robust Control Toolbox, a ' ...
  'pre-existing tuning report has been opened'],...
  'Missing Product License');
end
```

Chapter 3. Command Window Output

Gain Results From Tuning Script. When Heli_tuning_script.m is run in the workspace, the following results are displayed to the command window for the three outer loop PI controls and inner loop matrix gains:

```
Final: Soft = 1.11, Hard = -Inf, Iterations = 137
Block 1: <a href="matlab:hilite_system('rct_helico_Sdomain/</pre>
PI1', 'find'); pause(1); hilite_system('rct_helico_Sdomain/
PI1', 'none'); ">rct_helico_Sdomain/PI1</a> =
 1
Kp + Ki * ---
  with Kp = 1.13, Ki = 2.25
Name: PI1
Continuous-time PI controller in parallel form.
Block 2: <a href="matlab:hilite_system('rct_helico_Sdomain/</pre>
PI2', 'find'); pause(1); hilite_system('rct_helico_Sdomain/
PI2', 'none'); ">rct_helico_Sdomain/PI2</a> =
 Kp + Ki * ---
s
  with Kp = -0.086, Ki = -1.19
Name: PI2
Continuous-time PI controller in parallel form.
Block 3: <a href="matlab:hilite_system('rct_helico_Sdomain/</pre>
PI3', 'find'); pause(1); hilite_system('rct_helico_Sdomain/
PI3', 'none'); ">rct_helico_Sdomain/PI3</a> =
 Kp + Ki * ---
  with Kp = 0.133, Ki = -2.33
Name: PI3
Continuous-time PI controller in parallel form.
```

Command Window Output

Block 4: rct_helico_Sdomain/SOF =

D = u1 u2 u3 u4 u5 у1 2.395 -0.3609 -0.002145 0.8087 -0.0205 -0.1007 0.04573 y2 -0.1427 -1.115 -0.04318 -0.02792 y3 -0.02229 -2.025 -0.06152 0.03151

Name: SOF Static gain.

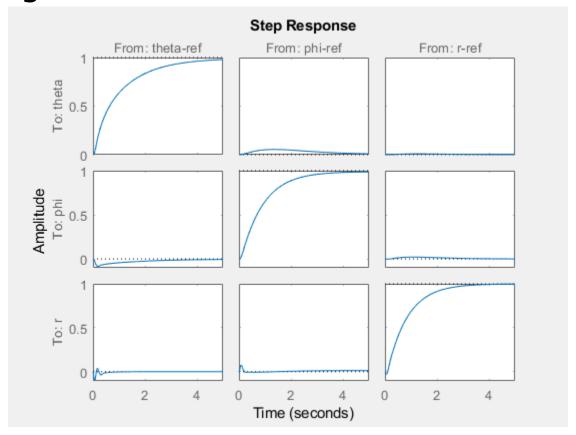
Chapter 4. Results

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Control System Responses and Stability Margins. Two figures are provided here, one that shows the step responses, and the other that shows the target responses, stability margins and closed loop pole locations. These responses are based on the slected set of gains. Running Heli_tuning_script.m creates 2 figure(s). A snapshot of each is displayed below.

4.1. Figure



4.2. Figure

