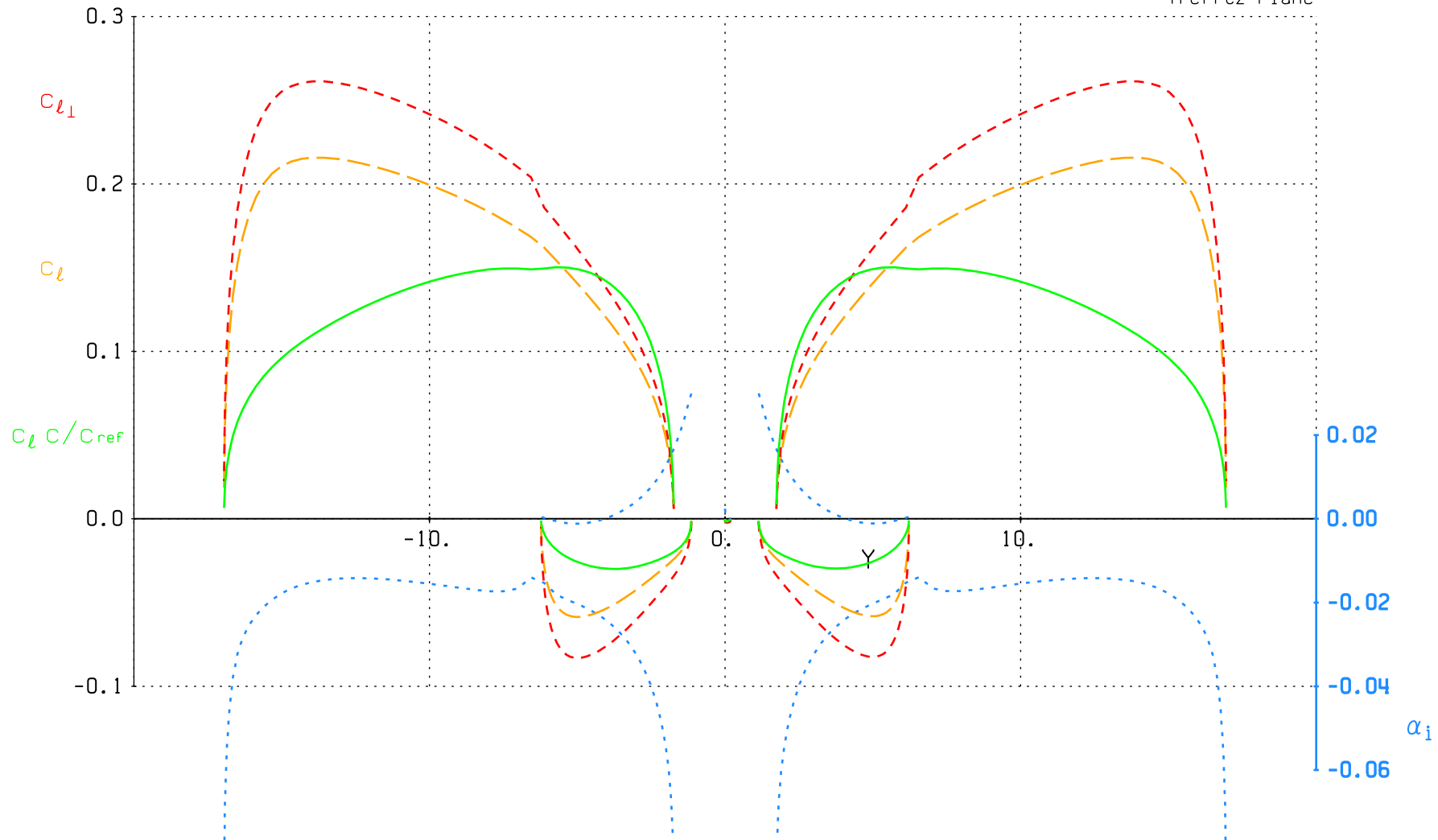


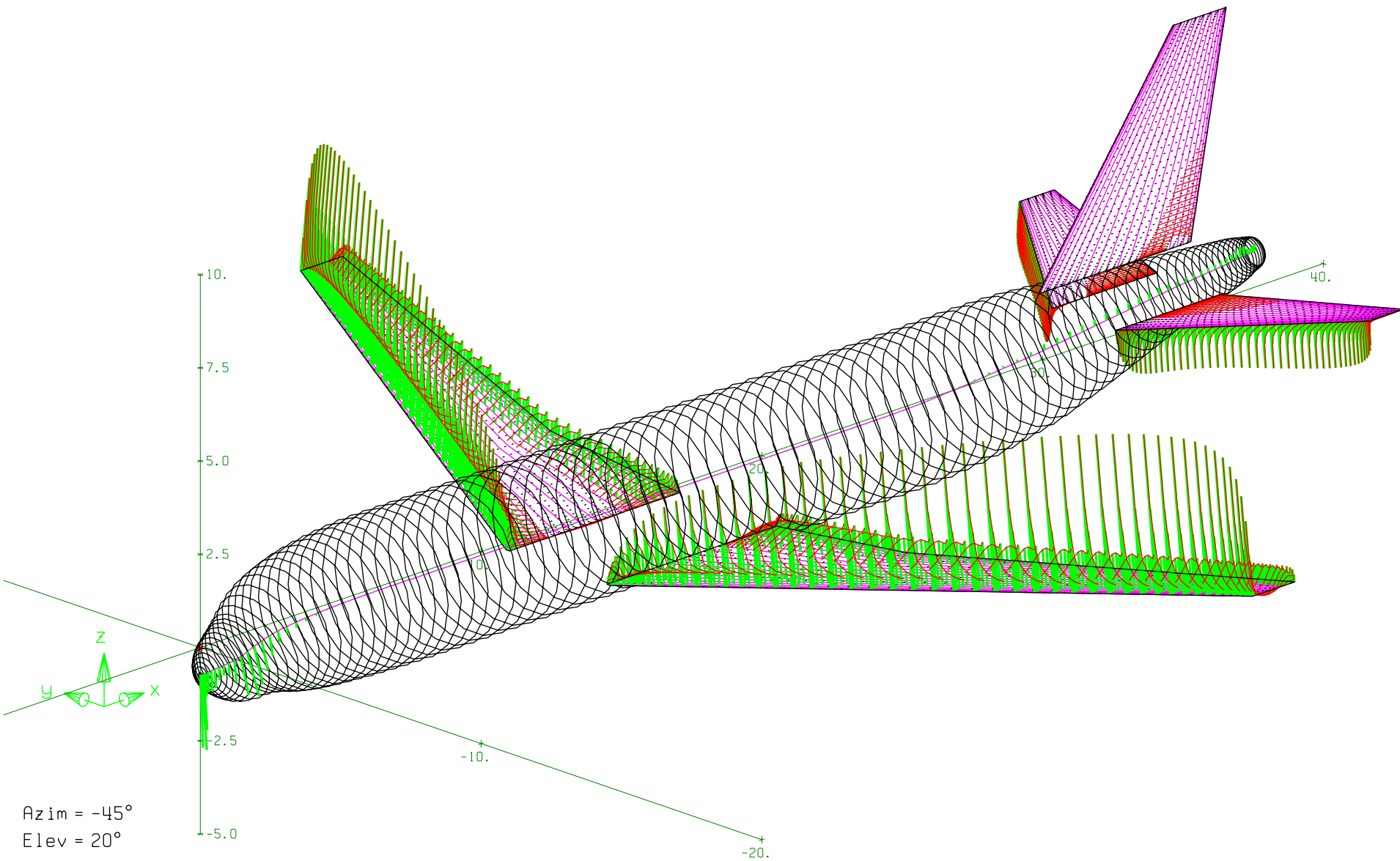
D150

$\alpha = 0.0000$	CL = 0.1109	Cl' = 0.0000	flap = 0.0000
$\beta = 0.0000$	CY = 0.0002	Cm = -0.6063	aileron = 0.0000
M = 0.300	CD = 0.00548	Cn' = -0.0003	elevator = 0.0000
pb/2V = 0.0000	CD _i = 0.00127	e = 0.5197	rudder = 0.0000
qc/2V = 0.0000	CD _p = 0.00000		
rb/2V = 0.0000			

--- C_{l_1}
 --- C_l
 --- $C_l C / C_{ref}$
 α_i

AVL 3.40
Trefftz Plane





Azim = -45°
Elev = 20°