1.14) Carbon dioxide flows through a constant area duct. At the inlet to the duct, the velocity is 120 m/s and the temperature and pressure are 200° C and 700 kPa, respectively. Heat is added to the flow in the duct and at the exit of the duct the velocity is 240 m/s and the temperature is 450° C. Find the amount of heat being added to the carbon dioxide per unit mass of gas and the mass flow rate through the duct per unit cross-sectional area of the duct. Assume that for carbon dioxide, $\gamma = 1.3$.

Solution:

Given: $\gamma = 1.3$

$$A_1=A_e=A,\,V_1=120\,{\rm m/s},\,T_1=200^\circ$$
 C = 473 K, $p_1=700$ kPa, $V_e=240\,{\rm m/s},\,T_e=450^\circ$ C = 723 K.

$$V_e = 240 \,\mathrm{m/s}, T_e = 450^{\circ} \,\mathrm{C} = 723 \,\mathrm{K}$$

To calculate: $\dot{q}/\dot{m} = ?$, $\dot{m}/A = ?$.

The schematic diagram of the problem description is shown in Fig. 1.

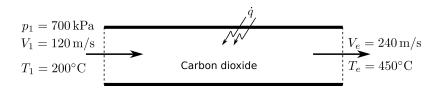


Fig. 1: Schematic diagram for problem description

Using the molar mass of cardon-dioxide (CO₂) as $12 \times 1_{\rm C} + 16 \times 2_{\rm O} = 44 \, \rm kg/kmol$ and the universal gas constant as 8314 J/kmol-K, the gas constant can be written as,

$$R = \frac{R_u}{\hat{m}} = \frac{8314}{44} = 188.95 \,\mathrm{J/kg\text{-}K}$$

The heat capacity at constant pressure can be calculated as,

$$c_p = \frac{\gamma R}{\gamma - 1} = \frac{1.3 \times 188.95}{1.3 - 1} = 818.78 \,\mathrm{J/kg\text{-}K}$$

Applying the conservation of energy principle.

$$c_p T_1 + \frac{V_1^2}{2} + \dot{q}/\dot{m} = c_p T_e + \frac{V_e^2}{2}$$

$$818.78 \times 473 + \frac{120^2}{2} + \dot{q}/\dot{m} = 818.78 \times 723 + \frac{240^2}{2}$$

We can solve for the heat added per unit mass to be,

$$\dot{q}/\dot{m} = 818.78 \times 723 + \frac{240^2}{2} - 818.78 \times 473 - \frac{120^2}{2}$$

$$\boxed{\dot{q}/\dot{m} = 226295 \,\text{J/kg}} \; .$$

The density at the inlet can be calculated using the ideal gas equation to be,

$$\rho_1 = \frac{p_1}{RT_1} = \frac{700 \times 10^3}{188.95 \times 473} = 7.8323 \,\mathrm{kg/m}^3$$

The mass flow rate,

$$\dot{m} = \rho_1 A_1 V_1$$

which gives the mass flow rate per unit area as,

$$\dot{m}/A = \rho_1 V_1 = 7.8323 \times 120$$

$$m/A = 939.876 \, \text{kg/s-m}^2$$